# **Technical Appendix - VTOL Precision Landing Simulator**

## 1) Notation (units)

z: altitude above landing pad (m).

W: camera image width in pixels (px). H: height in px. HFOV: horizontal field of view (deg).

f px: focal length in pixels. S: marker physical size (m). px: marker pixel span (px).

x, y: lateral pad-relative coordinates (m). dt: time step (s), default dt = 1.

g: process noise scaling for the constant-velocity model. sigma: measurement std dev (m).

## 2) Camera geometry (pinhole, nadir)

Focal length in pixels: f px = W / (2 \* tan(HFOV / 2)).

Marker pixel size model:  $px = f_px * S / max(z, eps)$ . We use eps = 1e-6 to avoid division by zero.

Field-of-view gate (approx.): radial  $\leq z * tan(HFOV / 2)$ , where radial  $= sqrt(x^2 + y^2)$ .

#### 3) Vision detection and lock

Base detection probability (logistic): base =  $1/(1 + \exp(-k * (px - thresh_px)))$ . We use k = 0.25. Modifiers: light\_boost = 0.6 + 0.4 \* illum; blur\_penalty = (1 - 0.6 \* blur); backend\_boost: 1.0 for ArUco, 1.1 for AprilTag.

Final probability: p det = clip(base \* light boost \* blur penalty \* backend boost, 0, 1).

Dwell and lock logic: detected -> dwell += 1, else dwell = 0. If dwell >= N, lock = True.

Unlock: if locked and i > 1/3 of the approach, unlock with small probability p ~ 0.05 when not detected.

#### 4) Beacon correction (when locked)

When locked, pull raw XY toward the pad center (beacon-like correction):  $pos_raw = pos_raw + (-g * pos_raw)$ , where g in [0.0, 0.8].

## 5) Constant-velocity Kalman filter (2D)

State:  $X = [x, y, vx, vy]^T$ . Measurement:  $z_meas = [x, y]^T$ .

Dynamics (dt = 1): A = [[1,0,1,0],[0,1,0,1],[0,0,1,0],[0,0,0,1]]. H = [[1,0,0,0],[0,1,0,0]].

Process noise: Q = q \* [[1/4,0,1/2,0],[0,1/4,0,1/2],[1/2,0,1,0],[0,1/2,0,1]].

Measurement noise:  $R = diag([sigma^2, sigma^2])$ .

Predict: X = A X;  $P = A P A^T + Q$ . Update:  $K = P H^T (H P H^T + R)^{-1}$ ; X = X + K (z - H X); P = (I - K H) P.

# **Technical Appendix (continued)**

## 6) Measurement noise scheduling (R)

Unlocked: sigma = kf r base (GPS  $\sim$ 1.0 m; RTK  $\sim$ 0.03 to 0.10 m).

Locked: sigma = clip(0.8 / max(px est, 1.0), 0.02, 0.20).

Interpretation: larger marker in image -> smaller sigma -> tighter R -> more measurement trust.

#### 7) Landing cone and corridor

Allowed radius:  $r_allowed = (z / z_{top}) * r_{top}$ . In app:  $z_{top} = 10 \text{ m}$ ,  $r_{top} = 1.0 \text{ m}$ . Cone violation per frame: radial > r allowed. Violation rate is mean over the run.

#### 8) Metrics

Pad accuracy (XY):  $e_xy = sqrt(x_TD^2 + y_TD^2)$  using final Kalman position.

Touchdown vertical speed: v = max(0, (z[-k-1] - z[-1]) / (k \* dt)), e.g., k = 5.

Cone violation rate: mean(radial > r\_allowed). Lock stability: mean(locked over last 30 percent of frames).

## 9) Scoring (0 to 100)

 $score = 100 * (0.40 * exp(-e_xy / 0.20) + 0.20 * exp(-max(0, v_z - 0.5) / 0.5) + 0.20 * exp(-5 * viol_rate) + 0.20 * lock_stability).$ 

Goal: encourage pad accuracy and soft touchdown, reward cone compliance and stable final lock.

# 10) Numerical stability and reproducibility

Use eps = 1e-6 in denominators; clip probabilities to [0,1]; clip sigma to [0.02,0.20] when locked. Set a random seed for reproducible tuning; keep dt consistent across A, Q, and v z computation.

## 11) Parameter ranges (typical)

RTK sigma: 0.02..0.10 m; GPS sigma: 0.5..2.0 m.

Lock threshold: 20..40 px; dwell: 6..12 frames; marker size: 0.3..0.8 m.

Kalman q: ~3e-5..1e-2; beacon gain q: 0.2..0.6.

# 12) Data export schema (CSV)

t (frame), x\_raw (m), y\_raw (m), x\_kf (m), y\_kf (m), z\_agl (m), detected (0/1), locked (0/1), px\_est (px).