

Measuring large scale flow structures behind a bluff body using a hot-wire rake



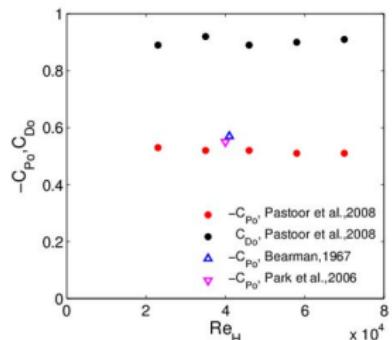
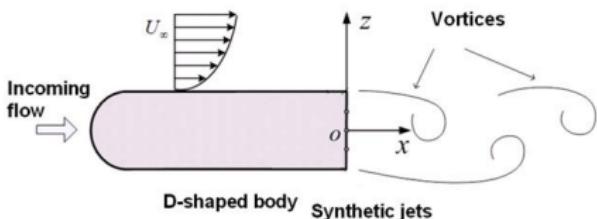
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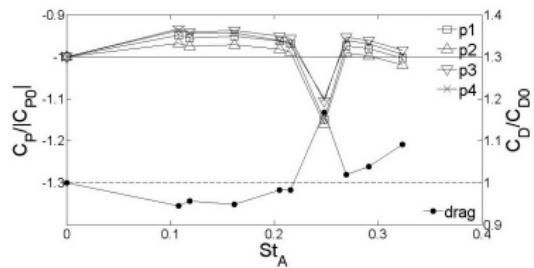
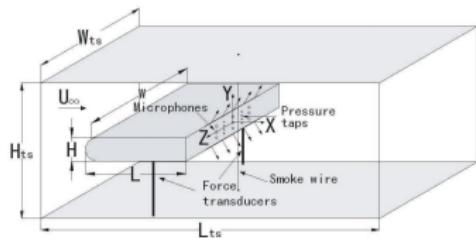
D-shaped bluff body

- Two-dimensional body with a streamlined front and a blunt trailing edge.
- Shear layers separated from upper and lower edges rolled up forming alternating vortices.
- Natural vortex shedding frequency $St_0 \approx 0.23$.



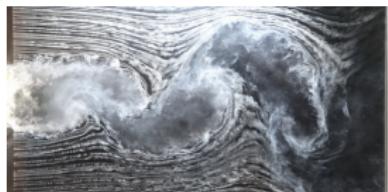
Active control

- Pastoor *et al.* (2008) in phase control of the two synthetic jets
 - 15% drag reduction when $St_A \sim 0.16$ and $C_\mu > 0.5\%$
 - Delayed transition from shear layer mode to asymmetric wake mode
- Li *et al.* (2015)
 - 5% drag reduction when $St_A \sim 0.16$ and $C_\mu \sim 0.1\%$, two jets act in phase.



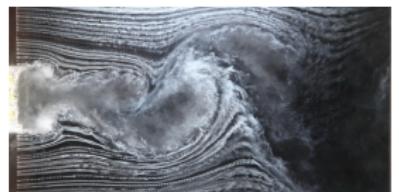
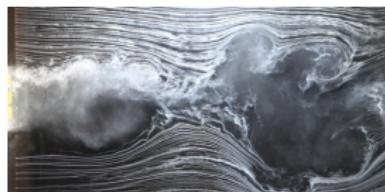
Li et.al(2015)

Visualizations for $St_A \sim 0.16$ and $C_\mu \sim 0.1\%$ (Li et al., 2015)



unforced flow

UP: Typical Kármán vortex street of un-forced flow behind a bluff body visualised by smoke wire.



RIGHT: Visualization of many different wake types forced at actuation frequency
 $St_A \sim 0.16$ and momentum ratio
 $C_\mu \sim 0.1\%$

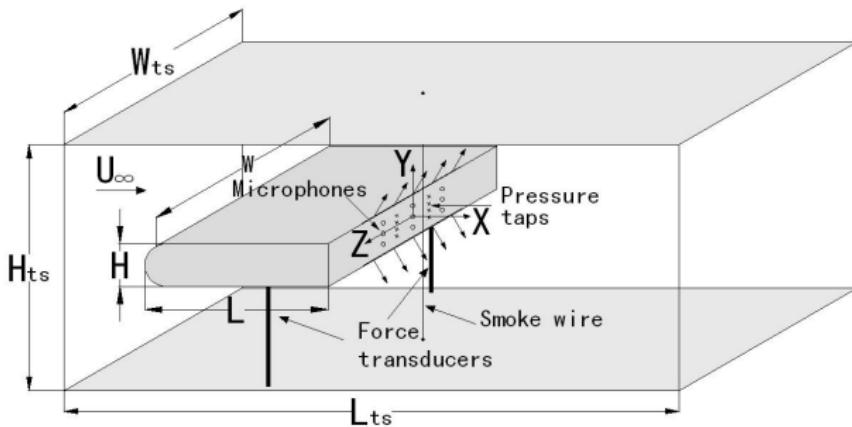


Objective

- Quantitatively determine the changes of wake structures under small amplitude periodic perturbations

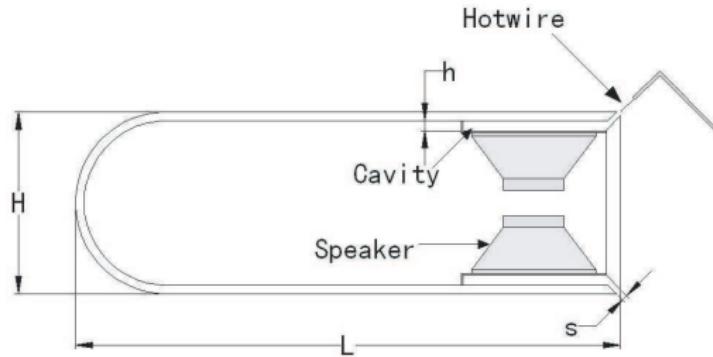
Experimental setup

- Suction type, open loop tunnel, 300x300mm test section;
- Model size $H = 63\text{mm}$, $L \sim 3H$, $W \sim 4.5H$;
- Incoming velocity $U_\infty = 9.2\text{m/s}$, corrected due to blockage to $U_{\infty,c} = 11.7\text{m/s}$, $Re_H = 47000$.



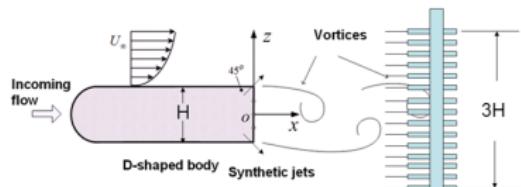
Actuators

- 2x3 array of speakers, $4.5W$, 8Ω , $45mm$, sinusoidal signal from a signal generator: $g_1(t) = A\sin(2\pi f_A t)$ and $g_2(t) = A\sin(2\pi f_A t + \phi)$
- Strength of actuation characterized using: $C_\mu = 2\frac{S}{H} \frac{u_A^2}{U_{\infty,c}^2} \cdot S = 2mm$
- Parameters
 - Momentum ratio $C_\mu = 0.1\%$
 - Actuation frequency, $St_A(f_A H/U_{\infty,c}) = 0.16$ and 0.24 ;
 - Two actuators act in phase $\phi = 0$.

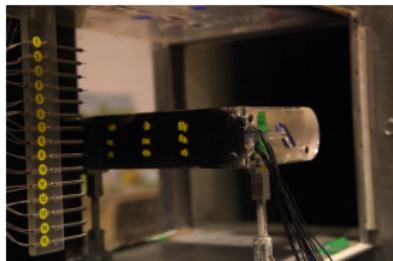


Hot-wire rake

- A rake of 15 single hot-wire probes, in house anemometry system;
- $5\mu\text{m}$ tungsten wire, 1.5mm long, calibrated *in-situ*;
- tuned to respond up to 6000Hz;
- 4kHz sampling frequency, sampling time 150s.



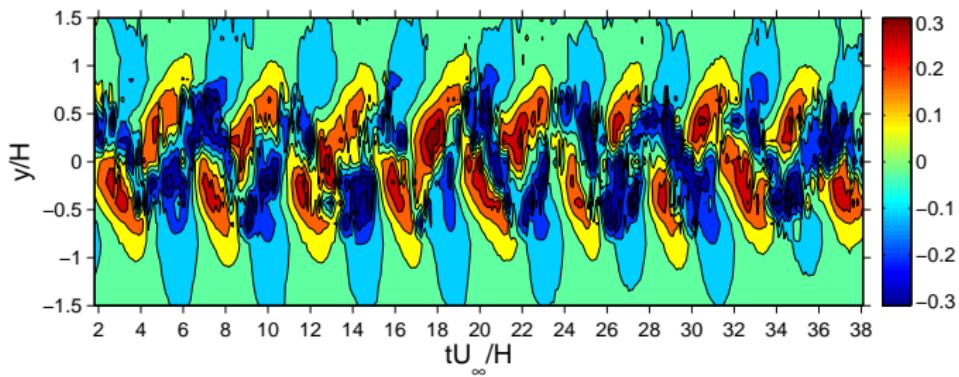
8-channel hotwire anemometry



Hot-wire array

Results

The instantaneous velocity for un-forced flow at $x/H = 2$



Proper Orthogonal Decomposition(POD) of velocity signals

- The instantaneous velocity $u(y, t)$ can be decomposed into 15 modes using POD:

$$u(y, t) = \sum_{n=1}^{15} a_n(t) \phi^{(n)}(y)$$

- Coefficients for mode n were obtained by projecting the velocity $u(y, t)$ on to a set of orthogonal basis:

$$a_n(t) = \int_{-1.5H}^{1.5H} u(y, t) \phi^{(n)} dy$$

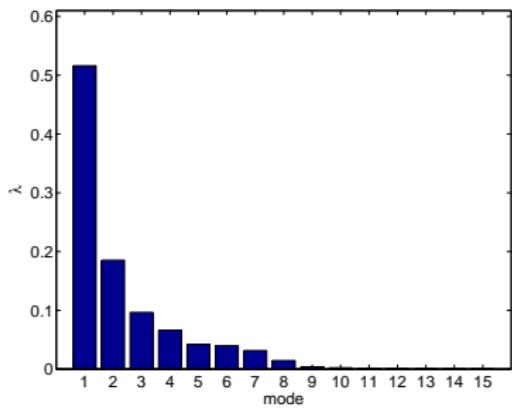
- The orthogonal basis $\phi(n)$ was obtained by computing the eigenvectors of the correlation matrix:

$$\int R_{uu}(y, y', \tau = 0) \phi^{(n)}(y') dy' = \Lambda^{(n)} \phi^{(n)}(y)$$

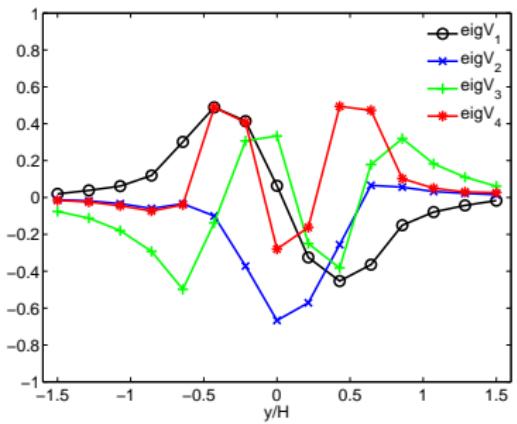
- The motion for the n -th mode can be re-constructed using $a_n(t)$ and $\phi^{(n)}(y)$:

$$u_n(y, t) = a_n(t) \phi^{(n)}(y)$$

Results decomposition of un-forced flow at $x/H = 2$

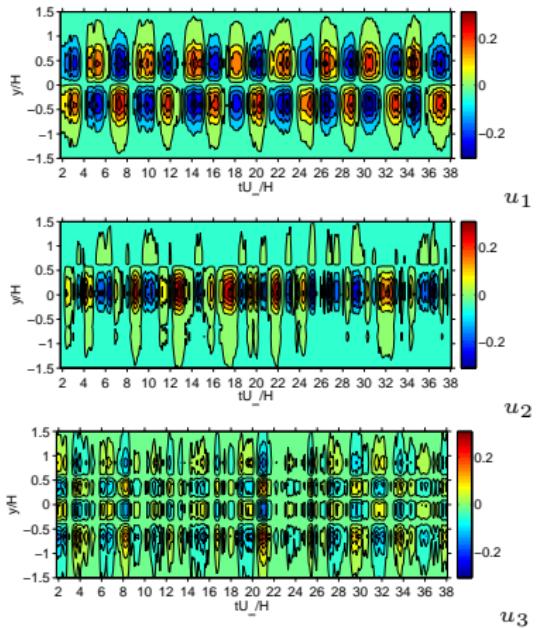
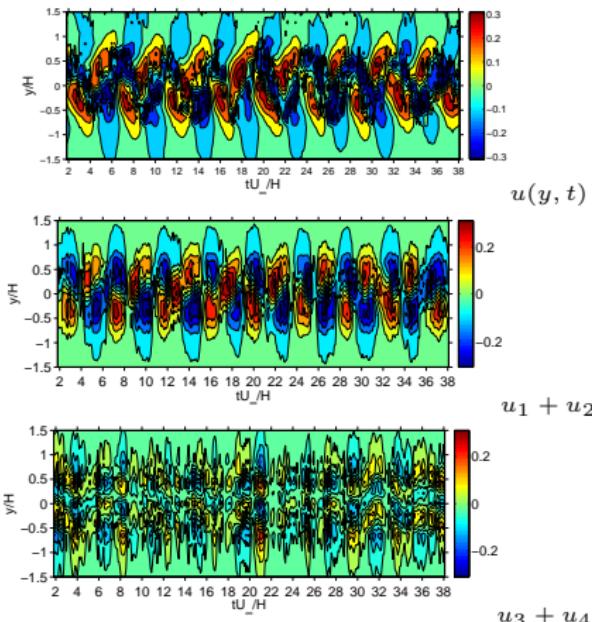


Λ^n
eigenvalues for each mode

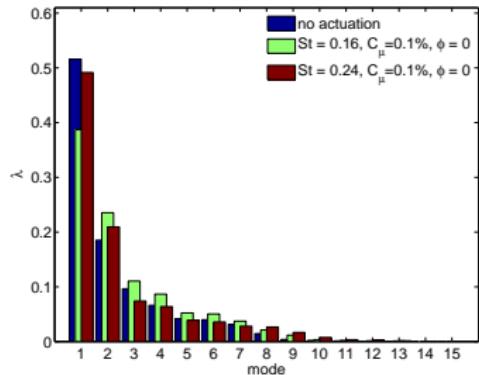


ϕ^n
eigenfunctions for each mode

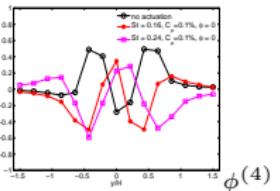
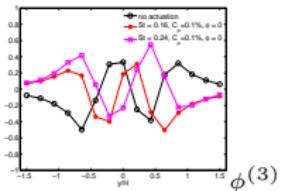
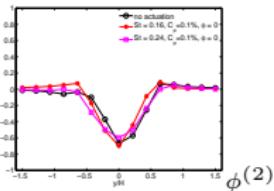
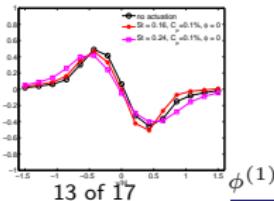
Results re-constructed velocity for un-forced flow at $x/H = 2$



Results effect of forcing frequency on modal energy, $x/H = 2$



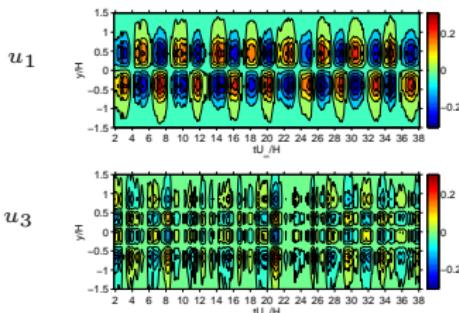
Eigenvalues ($\lambda^{(n)}$) for the three different cases



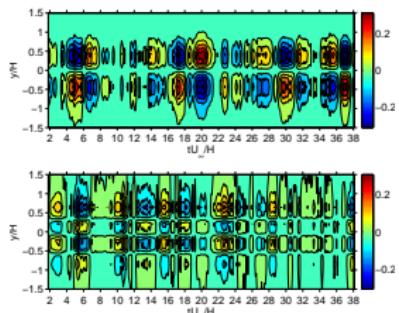
Results

re-construction velocity for POD mode 1 (u_1) and mode 3 (u_3), $x/H = 2$

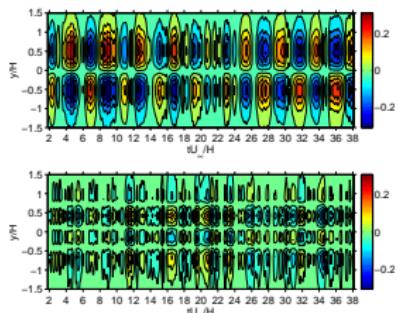
un-forced flow



in-phased forcing($\phi = 0$)
with $St_A = 0.16$



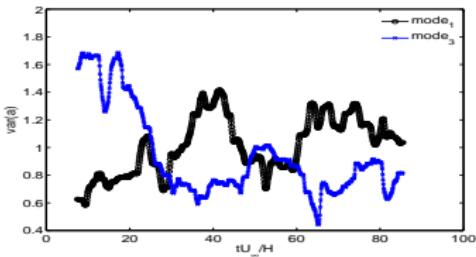
in-phased forcing($\phi = 0$)
with $St_A = 0.24$



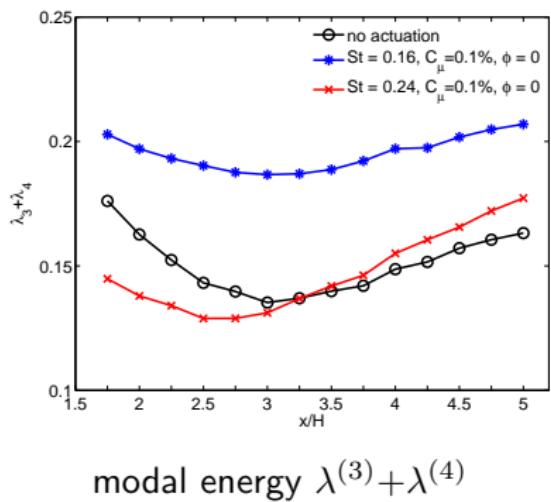
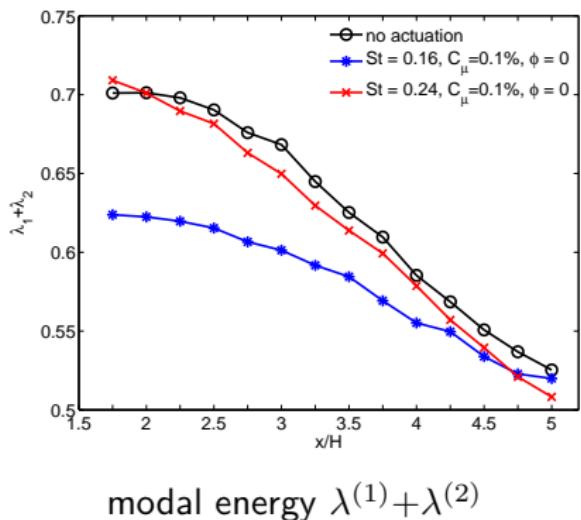
'move box variance'

$$t + \Delta T / 2 \\ \sum_{t - \Delta T / 2}^{t + \Delta T / 2} a^{(n)}(t)^2 / N,$$

$N = \Delta T \cdot f_{sampling}$, for in-phased forcing($\phi = 0$) with $St_A = 0.16$



Results Evolution of modal energy in the streamwise direction





Concluding remarks

- In-phase actuation on the trailing edges of a D-shaped body at $f_A H/U_o = 0.16$ decreased the form drag;
- In-phase actuation at $f_A H/U_o = 0.16$ was found to suppress the anti-symmetric von-Karman vortex (the lower POD modes) and promote the symmetric vortices (higher modes) ;
- the transitions between lower mode motions and higher modes seem to be sporadic, possibilities of higher mode motion were increased by the in-phased actuations at $f_A H/U_o = 0.16$.



Thank you.