△ № Lunar Scout: Task 1A - Practical

Sep 2023

Task Instructions

In this task, we will be deriving the state space model of the Rotary Inverted Pendulum system and implementing the LQR control scheme on the model. You will be required to find out the equations of motion for the physical system and modify code as instructed to find the correct mathematical model & find optimal controller gains.

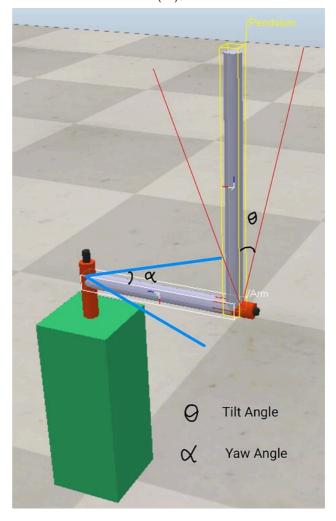
The Mathematical Modeling process is already demonstrated in the previous pages of this doc. Additionally a similar process for a two wheeled self balancing robot was shown during e-Yantra's technical sessions. So those who missed that, can refer it here:

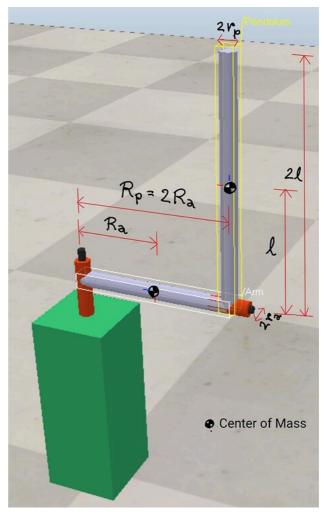
- https://www.youtube.com/watch? v=nOHky2f58ME
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Download this **Task1A.zip** (**UPDATED**) file and extract. For every Physical System that we will discuss here, you will find following files in the folder:

- 1. Task_1A.m
- 2. task1a.pyc

Physical System - Rotary Inverted Pendulum





Open the file Task_1A.m using Octave.

In this file you will find the following functions defined:

- Jacobian A B()
- lqr_Rotary_Inverted_Pendulum()
- Rotary_Inverted_Pendulum_main()

1. Jacobian _A_B():

This function is used to define the dynamics of the system by using the equations of motion you have derived for this system, finally to get the **State Space Model** of the system.

```
Jacobian _A_B()
```

- 1. function [A,B] = Jacobian_A_B(Mp,l,g,Ma
- 2. alpha = sym('alpha');
- 3. theta = sym('theta');
- 4. theta_dot = sym('theta_dot');
- 5. alpha_dot = sym('alpha_dot');
- 6. u = sym('u');
- 7. cos_theta = cos(theta);
- 8. sin_theta = sin(theta);
- 9. cos_alpha = cos(alpha);
- 10. sin_alpha = sin(alpha);

Steps:

- 1. Define equations of motion (4-state
- 2. Partial Differentiation of equation
- 3. Linearization by substituting equil

NOTE ### : Sequence of states should

SEQUENCE OF STATES : [alpha_dot; alpha; Example:

A = [x x x x; # correspo

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```
x x x x; # ...alpha
x x x x x; # ...theta
x x x x x] # ...theta
B = [x; # ...alpha_dot
x; # ...alpha
x; # ...theta_dot
x] # ...theta
```

- 11. A = ; # A should be (double) datatype
- 12. B = ; # B should be (double) datatype
- 13. endfunction

#}

Lets now understand this function:

- Line 2 to Line 6 are the representation of states in symbols
- Line 7 to Line 10 are the symbolic representation of cos and sin angle required in forming dynamic equations of the system
- There are HINTS given as comments in the function to help with the steps.
- Use the same sequence of states in the model as given in comments. Use "0" degree angle as the reference equilibrium point for tilt angle.
- Notice the Shape of B matrix given in comments. It says that you have to derive model considering it as single input system. (Underactuated !! (**))
- This function is intended to work as a helper to deal with long mathematical expressions in a simplified manner. You can choose to do the math partially by-hand/on-paper and feed in the values here. It will be considered valid as long as it passes the evaluator's tests without modifying anything else.

2. lqr_Rotary_Inverted_Pendulum():

You do not have to modify anything in this function. This function demonstrates the use of Octave to apply LQR control and find optimal gains using performance specified by Q & R matrices.

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It will be used in Task 1B.

lqr_Rotary Inverted Pendulum()

1. function K = lqr_Rotary_Inverted_Pendul

```
2. C = [1 0 0 0;
0 1 0 0;
0 0 1 0;
0 0 0 1]; ## Initialise
```

3. D =
$$[0;0;0;0]$$
; ## Initiali

6. sys =
$$ss(A,B,C,D)$$
; ## State Sp

8. endfunction



Note- You are not allowed to make any changes to this function.

3. Rotary_Inverted_Pendulum_main():

This function defines the physical parameters of system. It uses state space model generated and finds optimal gain using respective functions.

Rotary_Inverted_Pendulum_main()

function Rotary_Inverted_Pendulum_main(

2.
$$Mp = 0.5$$
; # mass of t

3.
$$l = 0.15$$
; # length fr

6.
$$r_a = 0.01$$
; # radius of

```
7. r_p = 0.01;  # radius of
8. Rp = 0.2;  # length fr
9. Ra = 0.1;  # length f
10. I_arm = ;  # Moment o
11. I_pendulum_theta = ;  # Moment o
12. I_pendulum_alpha = ;  # Moment o
13. [A,B] = Jacobian_A_B(Mp,l,g,Ma,Rp,Ra,I)
14. K = lqr_Rotary_Inverted_Pendulum(A,B)
endfunction
```

- You have to use the dimensions and other
 physical parameters of the model to compute
 respective moments of inertia for the system.
 You can feed it to use while deriving dynamics
 equations.
- Do not change the arguments or return values of any function.

Testing your solution:

 Your solution can be tested using task1a.pyc provided in the Task1A.zip



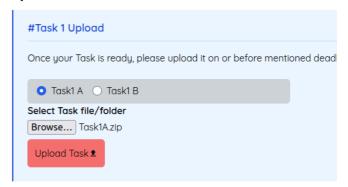
python task1a.pyc

- Feed respective TEAM ID and Octave executable path when prompted by it.
- Results will be printed at the last and an encrypted task1a_output.txt file will be generated in same folder.

For your reference, the evaluator checks format of your state space model, A, B matrices, open loop and closed loop eigen values/poles and stability, controlability, open loop response and LQR gains, etc.

Submission Instructions (**UPDATED**):

- Once you get your output, Task1A folder will contain the following.:
 - o Task 1A.m
 - o task1a.pyc
 - task1a output.txt
- Right click on the Task1A folder and compress as Task1A.zip.
- Upload the Task1A.zip file on the portal.
- In the Task 1 Upload section, click on Task 1A bullet and select Choose file button to upload the Task1A.zip file. From the dialogue box, select the file and click Open.
- You shall see the file name Task1A.zip in textbox besides the Choose file button. Click on Upload Task button to submit the file.



Task 1A is complete!!

Congrats!!

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𝚱 [Announcement] Task 1 Released!

Unlisted on Sep 17, 2023

Closed on Sep 19, 2023

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