

Homework 6 Due: Friday, November 20, 2015 @ 5:00:00 PM in E218

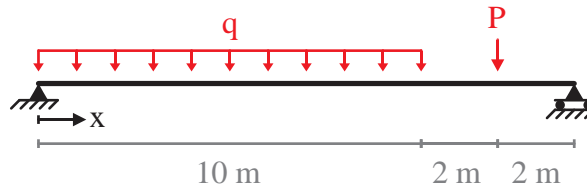
For each problem:

- i) Find the exact solution for the deflection, $w(x)$.
- ii) Define and apply Boundary Conditions to solve for d_1 and d_0 :

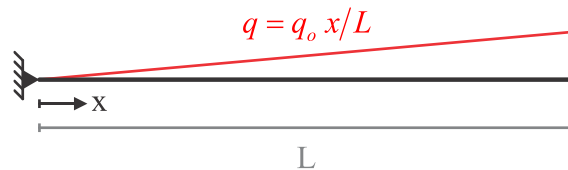
$$\bar{w}(x) = d_3 x^3 + d_2 x^2 + d_1 x + d_0$$

- iii) Solve for the remaining unknown coefficients using:
 - a. The Variational Principle
 - b. The Method of Weighted Residuals
- iv) Plot your three solutions on one graph and discuss how they compare.

1. Beam: $q = 2 \text{ kN/m}$ and $P = 10 \text{ kN}$. $E = 200,000 \text{ MPa}$, $I = 1 \times 10^9 \text{ mm}^4$. Note that the exact solution will be piecewise. The approximate solution, while not piecewise, will require piecewise integration. **25 pts.**



2. Truss: $A = 100 \text{ mm}^2$, $E = 200,000 \text{ MPa}$, $L = 10 \text{ m}$, $q_0 = 2 \text{ kN/m}$. **15 pts.**



Note: The axial deformation, w , of a truss element loaded by a distributed axial load, q , is governed by the differential equation:

$$q = -EA \frac{d^2 w}{dx^2}$$

The total potential energy for a truss element can be expressed as:

$$\Pi = \frac{1}{2} EA \int_L \left(\frac{dw}{dx} \right)^2 dx - \int_L (qw) dx$$

Homework 7 Due: Monday, December 7, 2015 @ 5:00:00 PM in E218

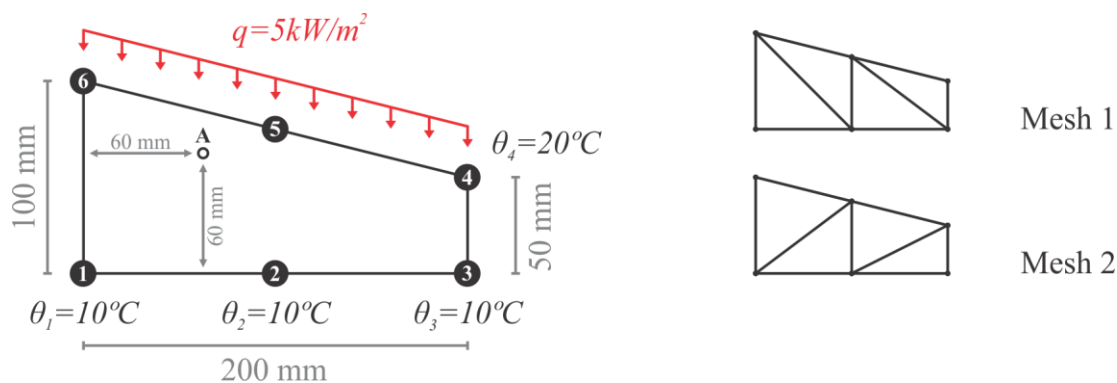
Solve **Question 1** by hand:

- Redraw the diagram and label your elements & degrees of freedom.
- Generate the force vector as well as the element and global stiffness matrices.
- Solve for the nodal temperature values and the temperature at point A.

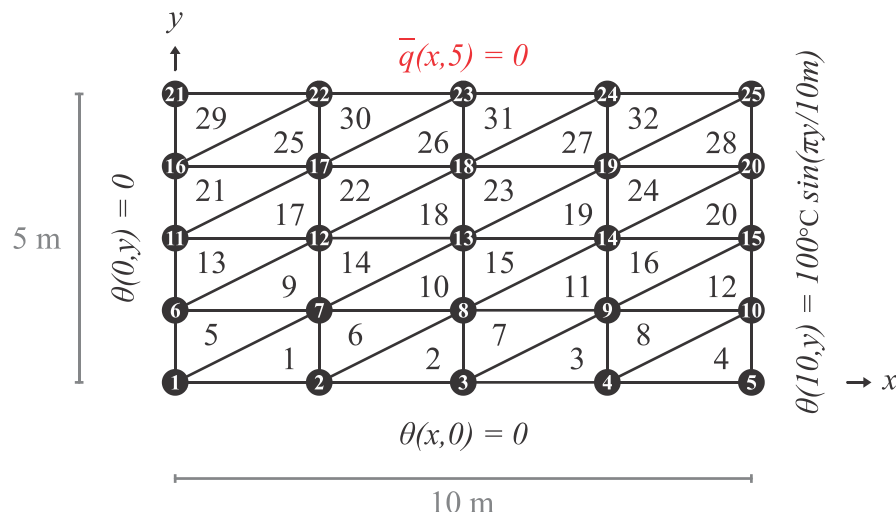
Code in the heat triangle element into MATLAB and perform an analysis for **Questions 1 and 2**:

- Provide labeled images of your input geometry and a plot of your temperature distribution.
- Compare (tabulate and discuss) your hand calculations with your MATLAB results.
- Provide a short description of your code changes and attach a printout only of the parts of the code that you changed.

1. 2D heat: Perform all analyses for both meshes. $t = 10$ mm. $k_x = k_y = 50$ W/m°C. $Q = 0$. **25 pts**



2. 2D heat: Instead of hand calculations, compare your answer to the exact solution. $Q = 0$. **15 pts**



Exact solution:
$$\theta(x, y) = \frac{100^\circ\text{C}}{\sinh(\pi)} \sin\left(\frac{\pi y}{10\text{m}}\right) \sinh\left(\frac{\pi x}{10\text{m}}\right)$$

Homework 8 Due: Monday, December 7, 2015 @ 5:00:00 PM in E218

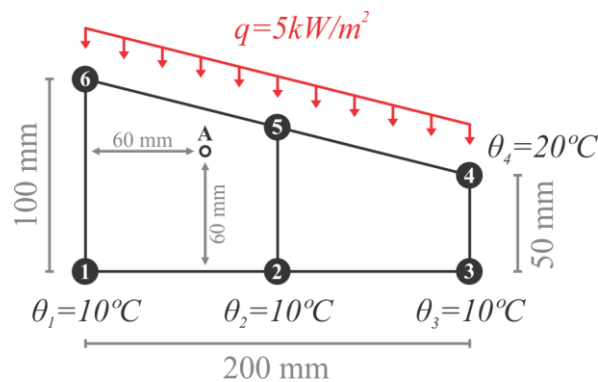
Solve **Question 1** by hand:

- Redraw the diagram and label your elements & degrees of freedom.
- Generate the force vector as well as the element and global stiffness matrices.
- Solve for the nodal temperature values and the temperature at point A.

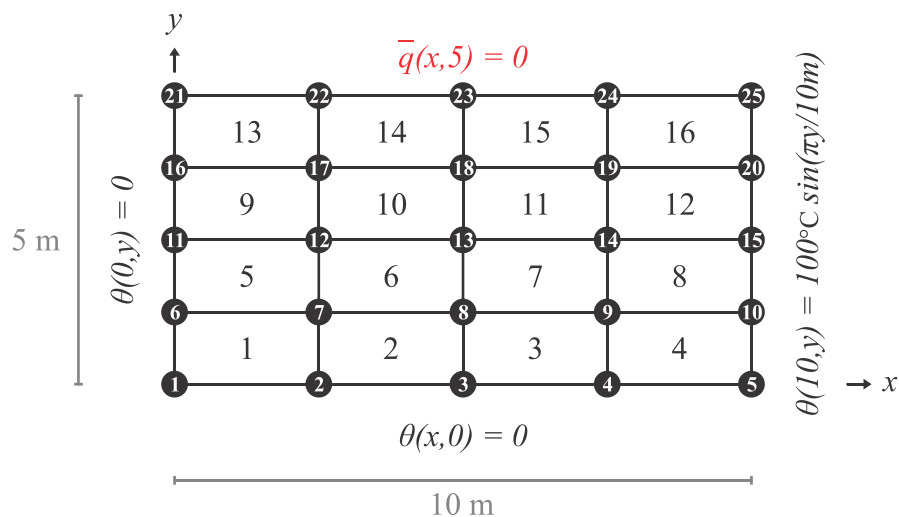
Code in the heat quad element into MATLAB and perform an analysis for **Questions 1 and 2**:

- Provide labeled images of your input geometry and a plot of your temperature distribution.
- Compare (tabulate and discuss) your hand calculations with your MATLAB results as well as your calculations from Homework 7.
- Provide a short description of your code changes and attach a printout only of the parts of the code that you changed.

1. 2D heat: $t = 10 \text{ mm}$. $k_x = k_y = 50 \text{ W/m}^\circ\text{C}$. $Q = 0$. **25 pts**



2. 2D heat: Instead of hand calculations, compare your answer to the exact solution. $Q = 0$. **15 pts**



Exact solution:
$$\theta(x, y) = \frac{100^\circ\text{C}}{\sinh(\pi)} \sin\left(\frac{\pi y}{10m}\right) \sinh\left(\frac{\pi x}{10m}\right)$$

Homework 9 Due: Friday, December 11, 2015 @ 5:00:00 PM in E218

For both **Question 1 and 2**:

- Redraw the diagram and label your elements & degrees of freedom.
- State whether the problem is Plane Stress or Plane Strain (and why).
- Calculate the applied force vector.

Solve **Question 1** by hand:

- Generate all element stiffness matrices as well as $[K_{UU}]$ and $[K_{US}]$ without generating $[K_G]$.
- Solve for the nodal displacements.
- Find the displacements, $\{\epsilon\}$, $\{\sigma\}$, σ_1 , σ_2 , and θ_p at point A.

Code the elastic triangle into MATLAB and perform an analysis for **Questions 1 and 2**:

- Provide images of the input and deformed geometry as well as plots of σ_x , σ_y , τ_{xy} , σ_1 & σ_2 .
- Provide a description of your code changes and only the modified parts of your code.

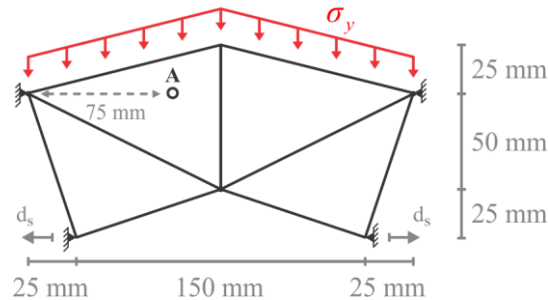
Perform an analysis with SAP2000 for **Questions 1 and 2**:

- Provide images of the input and deformed geometry as well as plots of σ_x , σ_y , τ_{xy} , σ_1 & σ_2 .

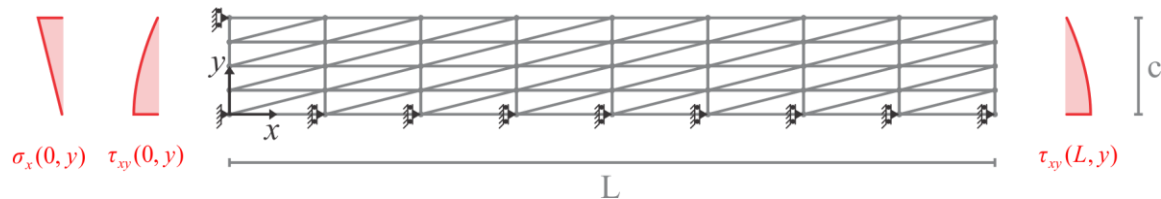
Finally, for both **Questions 1 and 2**:

- Tabulate the displacements from your MATLAB, SAP2000, and hand calculations.
- Compare and discuss your results based on the tabulated displacements and the stress plots.

1. 2D Elasticity: $t = 10$ m, $E = 200\,000$ MPa, $\nu = 0.3$, $d_s = 0.1$ mm, and $\sigma_y = 50$ MPa down. **35 pts**



2. 2D Elasticity: $L = 16$ m, $c = 2$ m, $t = 10$ mm, $E = 10\,000$ MPa, $\nu = 0.3$, and $P = -10$ kN. **25 pts**



The non-zero applied surface tractions are defined in global coordinates as:

$$\sigma_x(0, y) = -\frac{3PL}{2tc^3} y \quad \tau_{xy}(0, y) = \tau_{xy}(L, y) = \frac{3P}{4tc^3} (c^2 - y^2)$$

Use the exact solution, instead of hand calculations, for your comparison:

$$\begin{Bmatrix} u_x \\ u_y \end{Bmatrix} = \frac{P}{4Etc^3} \begin{Bmatrix} y(3x^2 - 6Lx + 2.3c^2 - 2.3y^2) \\ (L-x)^3 - L^3 + x(5.5c^2 + 3L^2) + 0.9(L-x)y^2 \end{Bmatrix} \quad \begin{aligned} \sigma_x &= -3P(L-x)y/2tc^3 \\ \sigma_y &= 0 \\ \tau_{xy} &= 3P(c^2 - y^2)/4tc^3 \end{aligned}$$

Homework 10 Due: Friday, December 18, 2015 @ 5:00:00 PM in E218

For both **Question 1 and 2**:

- i) Redraw the diagram and label your elements & degrees of freedom.

Solve **Question 1** by hand:

- ii) Generate all required element matrices. For your first integration point, find $[D]$, $[J_e]$, $[J_{e,1}]$, $[J_{e,1}]$, $[B_{e,1}]$, and $[K_{e,1}]$. Then find $[K_{e,2}]$, $[K_{e,3}]$, and $[K_{e,4}]$ as well as the final $[K_e]$.
 iii) Find $[K_{UU}]$ and $[K_{US}]$ without solving for $[K_G]$. Then, solve for the nodal displacements.
 iv) Find the displacements, $\{\epsilon\}$, $\{\sigma\}$, σ_1 , σ_2 , and θ_p at point A.

Code the elastic quad into MATLAB and perform an analysis for **Questions 1 and 2**:

- v) Provide images of the input and deformed geometry as well as plots of σ_x , σ_y , τ_{xy} , σ_1 & σ_2 .
 vi) Provide a description of your code changes and only the modified parts of your code.

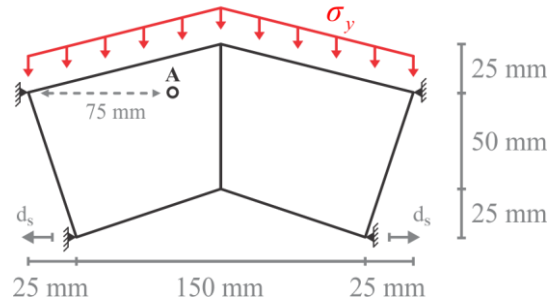
Perform an analysis with SAP2000 for **Questions 1 and 2**:

- vii) Provide images of the input and deformed geometry as well as plots of σ_x , σ_y , τ_{xy} , σ_1 & σ_2 .

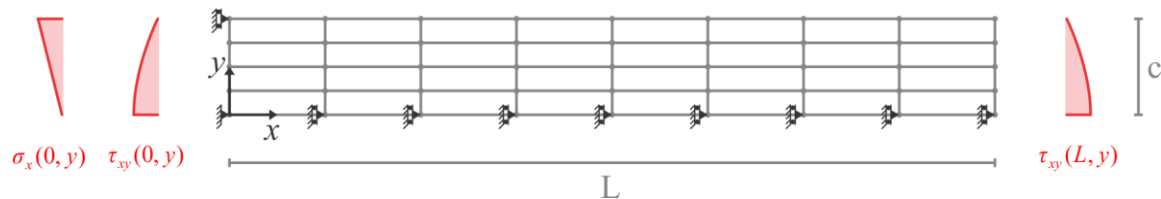
Finally, for both **Questions 1 and 2** (incorporating results from HW9):

- viii) Tabulate the displacements from your MATLAB, SAP2000, and hand calculations.
 ix) Compare and discuss your results based on the tabulated displacements and the stress plots.

1. 2D Elasticity: $t = 10$ m, $E = 200\,000$ MPa, $\nu = 0.3$, $d_s = 0.1$ mm, and $\sigma_y = 50$ MPa down. **35 pts**



2. 2D Elasticity: $L = 16$ m, $c = 2$ m, $t = 10$ mm, $E = 10\,000$ MPa, $\nu = 0.3$, and $P = -10$ kN. **20 pts**



The non-zero applied surface tractions are defined in global coordinates as:

$$\sigma_x(0, y) = -\frac{3PL}{2tc^3} y \quad \tau_{xy}(0, y) = \tau_{xy}(L, y) = \frac{3P}{4tc^3} (c^2 - y^2)$$

Use the exact solution, instead of hand calculations, for your comparison:

$$\begin{Bmatrix} u_x \\ u_y \end{Bmatrix} = \frac{P}{4Etc^3} \begin{Bmatrix} y(3x^2 - 6Lx + 2.3c^2 - 2.3y^2) \\ (L-x)^3 - L^3 + x(5.5c^2 + 3L^2) + 0.9(L-x)y^2 \end{Bmatrix} \quad \begin{aligned} \sigma_x &= -3P(L-x)y/2tc^3 \\ \sigma_y &= 0 \\ \tau_{xy} &= 3P(c^2 - y^2)/4tc^3 \end{aligned}$$