

<b>BRDF</b>	Bidirectional Reflection Density Function
<b>DPSSL</b>	Diode Pumped Solid State Laser
<b>laser</b>	Light Amplification by Stimulated Emission of Radiation
<b>LDA</b>	Laser Diode Array
<b>LiDAR</b>	Light Detection And Ranging
<b>TRL</b>	Technology Readiness Level

## 0.1 Detailed Design Optical Emitting Payload

LiDAR is a remote sensing system comprising an optical emitting device, used to acquire topographic data, e.g. surface elevation gradients or ground composition by evaluating the BRDF, considering multi-angular measurement are taken. For the generation of optical pulses, a highly efficient, diode-pumped, solid-state Nd-YAG laser is considered Diode Pumped Solid State Laser (DPSSL). Solid-State Light Amplification by Stimulated Emission of Radiations (lasers) have a high TRL with relatively good properties in terms of beam quality (Q-factor), efficiency and pulse manipulation. Data products for topographical missions require that the laser wave form be nearly pure Gaussian (known as transverse resonator mode  $TEM_{00}$ ), both temporally and spatially, with a uniform wave front. The digitized time of flight waveform returning provides the topographic structure [3].

### 0.1.1 Principles of *AlGaAs* Laser Diodes

laser diodes are electrically pumped semiconductor lasers, in which the gain is generated by an electrical current flowing through a *p-n junction* or (more frequently) a *p-i-n structure*[7]. In such a heterostructure, excitons dynamics can occur (electrons and holes can recombine), releasing the energy portions as photons. This process can be spontaneous, but can also be stimulated by incident photons, in effect leading to optical amplification. Most higher-power laser diodes, however, exhibit a relatively poor beam quality, combined with other non-favorable properties, such as a large beam divergence, high asymmetry of beam radius and beam quality between two perpendicular directions, and astigmatism (property of rays to exhibit different foci in different symmetrical planes). Especially considering the long distances used in LiDAR missions, these properties degrade the potential data quantity, as well as quality.

A quantum well is a thin layer which can confine (quasi-)particles (typically electrons or holes) in the dimension perpendicular to the layer surface, whereas the movement in the other dimensions is not restricted. A quantum well is often realized with a thin layer of a semiconductor medium, embedded between other semiconductor layers of wider bandgap. The thickness of such a quantum well is typically  $\sim 5 - 20$  [nm].

A major challenge is to reach the laser threshold, because the optical gain for the intracavity laser beam occurs only on a very small distance (in one or several quantum wells). It is therefore necessary to realize a laser resonator with very low losses, i.e., *Bragg mirrors* with high reflectivity.

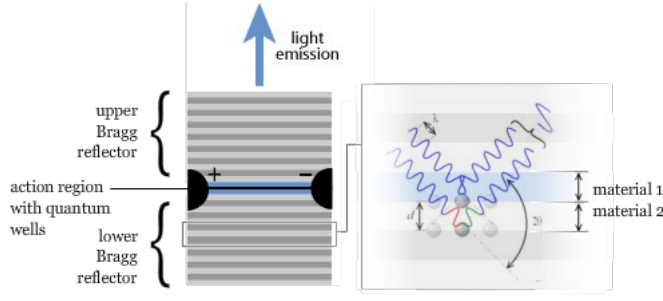


Figure 1: Basic diode laser configuration [8]. Each layer boundary causes a partial reflection of an optical wave. For waves whose wavelength is close to four times the optical thickness of the layers, the many reflections combine with constructive interference, and the layers act as a high-quality reflector. The range of wavelengths that are reflected is called the *photonic stopband*. Bragg's law describes the condition for constructive interference from successive crystallographic planes of the crystalline lattice according to  $n \lambda = 2d \cdot \sin(\theta)$ .

A Bragg mirror (also called *distributed Bragg reflector*) is a structure which consists of an alternating sequence of layers of two different optical materials. The principle of operation can be understood as follows. Each interface between the two materials contributes a Fresnel reflection. For the design wavelength, the optical path length difference between reflections from subsequent interfaces is half the wavelength; in addition, the reflection coefficients for the interfaces have alternating signs. Therefore, all reflected components from the interfaces interfere constructively, which results in a strong reflection. The reflectivity achieved is determined by the number of layer pairs and by the refractive index contrast between the layer materials.

Individual laser diodes normally generate quasi-continuous waves with powers  $\sim 1 - 10$  [mW]. To be able to generate higher power ( $\sim 1 - 10$  [W]) *laser diodes arrays* or *laser diode stacks* can be created, simply by combining multiple individual laser diodes. High-power laser diode arrays (! (!)LDAs) are used for a variety of space-based remote sensor laser programs as an energy source for DPSSLs. Laser Diode Arrays (LDAs) have been flown on NASA missions including MOLA, GLAS and MLA and have continued to be viewed as an important part of the laser-based instrument component suite [6].

Laser diode bars have many single emitters arranged side-by-side and spaced approximately 0.5 [mm] apart, on a single slab of semiconductor material measuring approximately 0.5 [mm] x 10 [mm] in size. The individual emitters are connected in parallel which keeps the required voltage low at 2V, but increases the required current to 50 A/bar to 100 A/bar. Stacking these laser diode bars 2 to 20+ slabs high yields high power LDAs capable of emitting several hundreds of Watts. Electrically, the bars are wired in series increasing the voltage by 2 V/bar while maintaining the total current at 50 A to 100 A. These arrays are one of the enabling technologies for efficient, high power solid-state lasers.

Traditionally these arrays are operated in QCW (Quasi Continuous Wave) mode with pulse widths of  $\sim 50$  s to 200 s and repetition rates of  $\sim 10$  Hz to 200 Hz. In QCW mode, the wavelength and the output power of the laser reaches steady-state, but the temperature does not. The advantage is a substantially higher output power than in CW mode, where the output power would be limited

by the internal heating and the heat sinking properties of the device. The disadvantage is a much higher thermally induced mechanical stress caused by the constant heating and cooling cycle of the QCW operational mode, considering non-conductive cooling configurations.

Considering the fact that Nd:YAG is considered as gain medium (0.1.2), the existence of strong  $Nd^{3+}$  absorption near 808 [nm] permits efficient pumping with a GaAlAs (Gallium-Aluminium-Arsenide) diode lasers for the  $F_{3/2} \rightarrow I_{9/2}$  transition. The direct band gap crystal AlGaAs is often used for laser diodes with wavelengths between 750nm and 880nm.  $Al_xGa_{(1-x)}As$ , through changing the x, the ratio of the aluminum to gallium can be adjusted to vary the band gap and thereby control the wavelength. In the double heterostructure, stimulated emission occurs only within a thin active layer of GaAs, which is sandwiched between p- and n- doped AlGaAs layers that have a wider band gap. Laser diodes use heterojunctions to achieve simultaneous carrier and photon confinement in the active region.

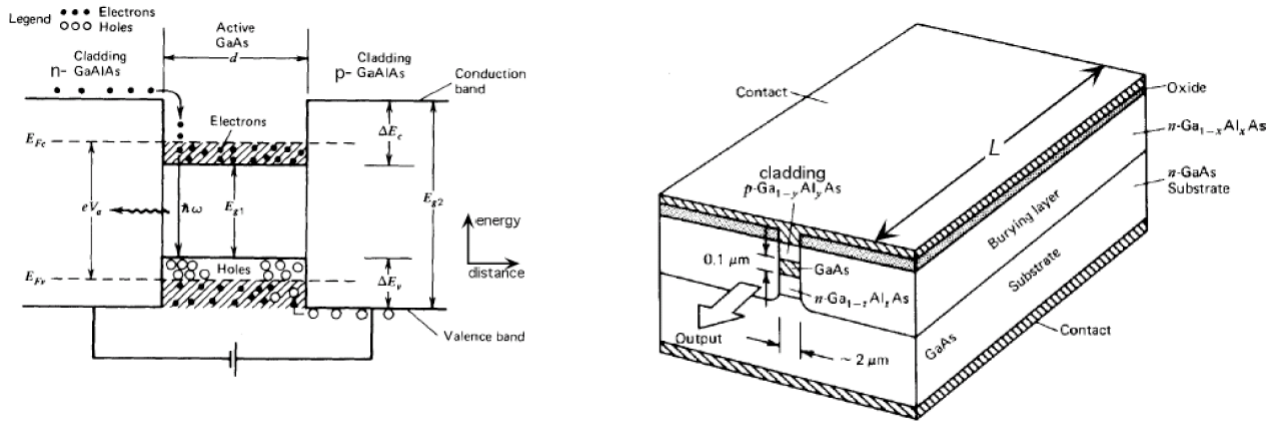


Figure 2: Basic AlGaAs diode laser configuration [5].

A high laser efficiency demands that the light and injected charge carriers be confined as closely as possible to the same volume. The AlGaAs laser diode consists of a double heterojunction formed by an undoped (or lightly p-doped) active region surrounded by high bandgap p and n  $Al_xGa_{(1-x)}As$  cladding layers [5]. The surrounding cladding layers provide an energy barrier to confine carriers to the active region. The actual operation wavelengths may range from 750 to 880 [nm] due to the effects of dopants, the size of the active region, and the compositions of the active and cladding layers. When a certain parameter is fixed, the wavelength can vary in several (sub)nanometers due to other variables. For example, when the active layer has an energy gap  $E_g = 1.424[eV]$ , the nominal emission wavelength is  $\lambda = hc/E_g = 871[nm]$ . When a bias voltage is applied in the forward direction, electrons and holes are injected into the active layer. Since the band gap energy is larger in the cladding layers than in the active layer, the injected electrons and holes are prevented from diffusing across the junction by the potential barriers formed between the active layer and cladding layers. The electrons and holes confined to the active layer create a state of population inversion, allowing the amplification of light by stimulated emission.

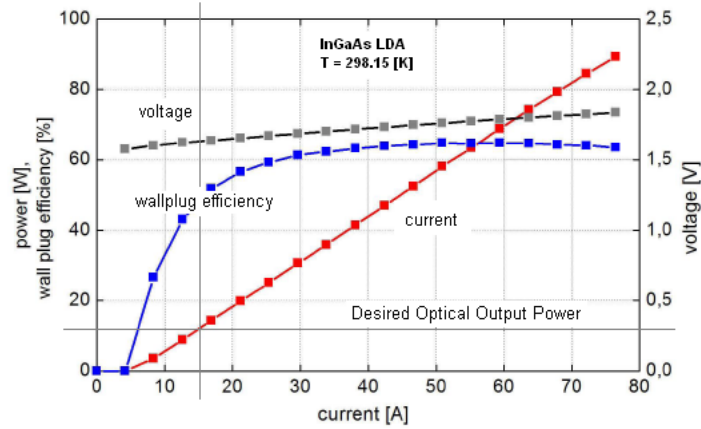


Figure 3: Typical laser diode characteristics; power and current information at constant temperature. Parameters decrease with higher temperatures.

### 0.1.2 Diode Pumped Solid-State Laser Configuration

Nd-YAG ( Neodymium Yttrium Aluminum Garnet $\text{Y}_3\text{Al}_5\text{O}_{12}$ )	
Nd Concentration	0.2 – 1.4 %
Diameter	0.5 – 15.0 mm
Length	1.0 – 220.0 mm
Damage Threshold	> 20 J/cm <sup>2</sup>
Refractive Index (n)	1.8169 @ 1,064 nm
Thermal Conductivity	0.129 W/cm.K
Specific Heat	0.59 J/ g. K
Density	4.55 gm/cm <sup>3</sup>
Tensile Strength	280 MPa
Young's Modulus	282 GPa
dn/dT	+8.9 · 10 <sup>-6</sup> K <sup>-1</sup>

Figure 4: Basic (simplified) configuration of the laser with dimensions. Individual components will be explained later in this section.

### Nd-YAG Laser Characteristics

*Yttrium Aluminum Garnet* has emerged as the most widely produced laser gain host and has enjoyed recent popularity as a substrate material for optical components. The YAG host is a stable compound, mechanically robust, physically hard, optically isotropic, and transparent from below 300 to beyond 4,000 [nm]. YAG single crystals are able to accept trivalent laser activator ions from both the rare Earth and transition metal groups, and can be grown with very low strain.

For applications where  $TEM_{00}$  single mode operation is required, it is necessary to reduce

or eliminate the variations in the bulk material and in the absorption of the pumping radiation throughout the component. In addition, wavefront distortions due to geometric imperfections and thermal gradient effects such as thermal lensing must be minimized. In this case, Neodymium concentration in the 0.4 to 0.8% range is typically specified.

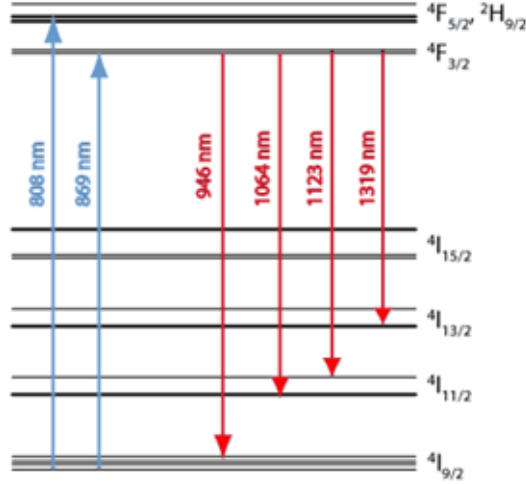


Figure 5: Quantum energy level and transitions for the Nd:YAG crystal, including the used  $F_{3/2} \rightarrow I_{9/2}$  transition.

## Second Harmonic Generation

Since Nd-YAG has no principle absorption peak at the desired wavelength for the LiDAR mission, the frequency should be altered from the original 946 [nm]. This can be done using *second harmonic generation* or *frequency doubling* in nonlinear crystals. The physical mechanism behind frequency doubling can be understood as follows. Due to the (2) nonlinearity, the fundamental (pump) wave generates a nonlinear polarization wave which oscillates with twice the fundamental frequency. According to Maxwell's equations, this nonlinear polarization wave radiates an electromagnetic field with this doubled frequency. Due to phase-matching issues, the generated second-harmonic field propagates dominantly in the direction of the nonlinear polarization wave. The latter also interacts with the fundamental wave, so that the pump wave can be attenuated (pump depletion) when the second-harmonic intensity develops: energy is transferred from the pump wave to the second-harmonic wave.  $\beta - BaB_2O_4$  is used for second, third, fourth and fifth harmonic generation of Nd doping lasers. Typical dimensions of these crystal are  $\sim 0.05 - 10[mm]$ .

## Pulse Generation

The generation and manipulation of pulses can highly influent the data in LiDAR missions. To be able to transform the quasi-continuous wave into a pulsed wave, Q-switching is applied. Q-switching is a technique for obtaining energetic short pulses from a laser by modulating the intracavity losses and thus the Q-factor (a measure of the damping of resonator modes) of the laser resonator. The

technique is mainly applied for the generation of nanosecond pulses of high energy and peak power with solid-state bulk lasers. For *active Q-switching*, the losses are modulated with an active control element typically either an acousto-optic or electro-optic modulator. Both techniques rely on the fact that the optical properties within a nonlinear crystal change on the occurrence of an induced sound wave (acousto-optic) or electric field (electro-optic). There are also mechanical, less viable for space missions, Q-switches such as spinning mirrors, used as end mirrors of laser resonators. In any case, the achieved pulse energy and pulse duration depends on the energy stored in the gain medium, i.e. on the pump power and the pulse repetition rate. A Pockels cell is a device consisting of an electro-optic crystal including electrodes through which an electromagnetic beam can propagate. Dependent on the configuration, the phase delay or polarization state in the crystal (due to the *Pockels effect*) can be modulated by applying a flux electric voltage (typical second harmonic generation characteristics:  $\sim 40,000V/0.1[mA]$ ) Hence, for short periods ( $dt$ ) the polarization state of the incoming electromagnetic radiation can be altered. If a *polarizer disk* is used after the Pockel cell, the generation of pulses will begin, since the polarizer disk transmits certain polarized states only, deflecting the rest (acting like a 'polarize filter'). Pulses in the order of nanoseconds could be created this way. Care should be taken at the fact that the peak power after the Pockel cell is increased in several orders, due to the conversion from continuous to pulsed waves. Hence, the polarizer disk should be able to cope with these stresses. According to data posted after the GLAS-mission (using the same sort of laser, an optical induced layer was formed at the non-linear crystal, probably induced by the high peak powers of the pulsed waves. In this configuration, the pulsed wave is created after the  $\beta - BaB_2O_4$  - crystal, reducing the risk of the creation of critical optical damage (COD).

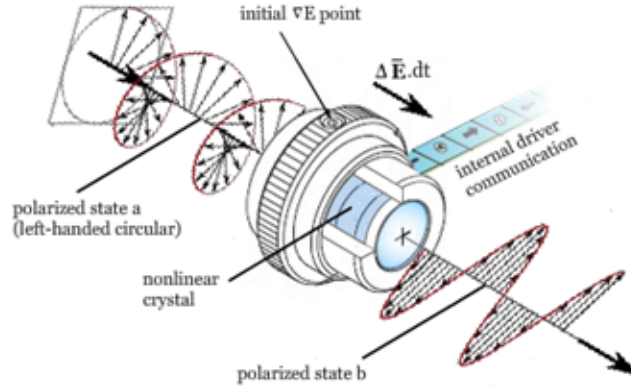


Figure 6: Change in polarized state of electromagnetic radiation when passed through a Pockel cell. Communication with Pockel cell driver should be applied to determine magnitude of electric field flux and  $dt$ . Considering the fact that the value of  $f_{rep}$  could be changed in-mission (for example, if the pulse rate should be higher due to lower reflectivity on specific areas) communication with ground-station should be applied.

### 0.1.3 Optical Characteristics

Considering for a moment, the radius of the beam equals  $400 [\mu\text{m}]$ . Considering a  $TEM_{00}$  transverse mode of electromagnetic radiation this results in a area equal to

$$A_{beam} = \pi \cdot r^2 = \pi \cdot 400^2 = 502,654 \mu\text{m}^2 = 0.50265 \text{cm}^2 \quad (1)$$

The pulse energy  $E_p$  [J] (maximum optical power of a pulse) is determined using the simulator. Sufficient energy should be present within the electromagnetic radiation to ensure the optimum path from the transmitter towards the receiver. Lowering the value of  $E_p$  below this threshold energy can lead to atmospheric and surface absorption or translational mismatching due to incorrect scattering. The value of  $E_p$  of this particular mission is determined to be  $\sim 1$  [mJ] (see ??). Pulse repetition rate  $f_{rep}$  [Hz], i.e. the number of pulses emitted per second, is an important parameter for the altimetry mission. Again, using the results from the simulator, the quantity of this parameter can be determined to be  $\sim 5000$  [Hz] ( $\Delta t = 0.0002$  [s] with a pulse duration  $t_p \sim 10$  [ns]). Using the value of  $f_{rep}$ , the spatial resolution of the pulses along-track and in the nadir-direction can be calculated, considering the orbital velocity to be fixed at the determined altitude.

$$d_{along} = \Delta t \cdot v = 0.0002[s] \cdot 7,617[m/s] = 1.5234[m] \quad (2)$$

$$d_{nadir} = \Delta t \cdot c = 0.0002[s] \cdot 299,792,458[m/s] = 59,958.49[m] \quad (3)$$

Considering the value of  $E_p$  to be  $1$  [mJ] with a  $f_{rep}$  of  $5,000$  [Hz], the total power that should be induced within the electromagnetic wave can be calculated.

$$P_{output} = E_p \cdot f_{rep} = 0.001[J] \cdot 5,000[1/s] = 5.0[W] \quad (4)$$

The *total* electrical-to-optical power efficiency of a laser system ( $\eta_{wp}$ ), i.e. the *wall plug efficiency*, typically is  $\sim 10\%$ , however, linear interpolations of the current data, considering the large amount of research done on this subject, shows that  $\eta_{wp}$  increases with one percent point every year (on average) from 2004, giving an wall plug efficiency of  $\sim 16\%$  in 2010 and  $>20\%$  in 2015 [3]. A higher value of  $\eta_{wp}$  reduces the electrical power consumption and also the amount of heat which has to be removed. Simulation shows a power needed for succesfull photon detection in this altimetry mission of  $\sim 6[W]$ , ending up with a total power need of  $60$  [W] ( $\eta_{wp} = 0.1$ ) or (more realistic)  $30$  [W] ( $\eta_{wp} = 0.2$ ).

The pulse peak intensity equals  $E_p/t_p = 0.001[J]/10 \cdot 10^{-9}[s] = 100,000W$ . The intensity turns out to be  $\frac{E_p/t_p}{A_{beam}} = \frac{100,000[W]}{0.50265[cm^2]} = 198,950.6[W/cm^2](0.00199[J/cm^2/10ns])$ . The standard damage threshold energy  $E_{p,damage}$  for dielectric components equals  $0.5 - 10[J/cm^2/10ns]$ . Considering the lowest value  $I_{p,damage}$ , hence,  $0.5[J/cm^2/10ns]$ , and converting this to the appropriate dimensions, shows the intensity created within the electromagnetic pulses should do no harm to the dielectric components. Especially the polarizer disk (with the lowest  $I_{p,damage}$ ) is vulnerable for peak power caused by pulsed electromagnetic radiation.

### 0.1.4 Gaussian Beam Propagation And Diffraction

Collimated plane wave propagation (uniform  $\mathbf{k}$ -vector distribution) in optical systems would give rise to discrete and accurate calculations. However, due to optical distortions and modifications, the  $\mathbf{k}$ -vector distribution can change, hence, altering the wave propagation.

*Gaussian Beams* For the analysis of the laser beam intensity profile, a Gaussian profile (transverse resonator mode  $TEM_{00}$ ) is considered corrected by the  $M^2$  factor for optical distortion. The  $M^2$  factor is a common measure of the beam quality of a laser beam. The electric field distribution for a Gaussian beam is represented as:

$$E(r, z) = E_0 \cdot \frac{w_0}{w(z)} \cdot \exp \left[ \frac{-r^2}{w(z)} \right] \cdot \exp \left[ -i(kz - \arctan \left( \frac{z}{z_r} \right) + \frac{kr^2}{2R(z)}) \right] \quad (5)$$

$$E(x, y, z) = \exp \left[ \frac{-i(kz + \psi(z))}{w(z)} \right] \exp \left[ \frac{-(x^2 + y^2)}{w^2(z)} - ik \frac{(x^2 + y^2)}{2R(z)} \right] \quad (6)$$

The main point of this section is to determine the Gaussian beam propagations dependency on diffraction phenomenon. To characterize the Gaussian beam in more details, the following equations are used to be able to describe the propagation.

$$\theta = \frac{\lambda}{\pi \cdot w_0} \quad (7)$$

$$w_R(z) = w_{0R} \sqrt{1 + \left( \frac{z(\theta + \Delta\alpha)M^2}{w_{0R}} \right)^2} \quad (8)$$

$$w_R(z) = w_{0R}(z) \sqrt{1 + \left( \frac{z\lambda M^2}{\pi w_{0R}^2} \right)^2} \quad (9)$$

$$R_R(z) = z \left[ 1 + \left( \frac{w_{0R}}{z\theta M^2} \right)^2 \right] \quad (10)$$

$$R_R(z) = z \left[ 1 + \left( \frac{w_{0R}}{z(\theta + \Delta\alpha)M^2} \right)^2 \right] \quad (11)$$

*Fraunhofer Diffraction* Diffraction is a fundamental characteristic of all wave fields. The effect of diffraction is typically manifested when an obstacle is placed in the path of a beam. On an observation screen some distance away from the obstacle, one observes a rather complicated modulation of the time-average intensity in the vicinity of the boundary separating the illuminated region from the geometrical shadow cast by the obstacle. With the use of high-power lasers, diffraction of radiation beams (cavity oscillating in the fundamental transverse Gaussian  $TEM_{00}$  mode) with finite transverse dimensions has significant consequences. The Fresnel number  $F = a^2/\lambda \cdot R$ , where



a is the characteristic size ("radius") of the aperture,  $\lambda$  is the wavelength, and R is the distance from the aperture, determines the diffraction regime that should be considered (Fj 1: Fraunhofer (far-field); F >1, Fresnel). The far-field light field is the Fourier transform of the apertured field. The far-field light field is the Fourier transform of the apertured field.

$$E(k_x, k_y) = \mathcal{F} \left\{ \overbrace{t(x, y)}^{\text{Transmission function}} \cdot E(x, y) \right\} = \iint (\exp(-i(k_x x + k_y y)) \cdot t(x, y) \cdot E(x, y) dx dy) \quad (12)$$

$$E(x, y, z) = \frac{\exp[-i(kz + \psi(z))]}{w(z)} \cdot \exp \left[ \frac{-(x^2 + y^2)}{w^2(z)} - \frac{ik(x^2 + y^2)}{2R(z)} \right] \quad (13)$$

The lens incorporates a phase delay to the outgoing electromagnetic field. For the entire derivation of this equation, [8] should be evaluated.

$$t_{lens} = \exp \left\{ -i \left( (n-1) \left( \frac{k}{2R} (x^2 + y^2) \right) \right) \right\} \quad (14)$$

Combining the above calculations, calculations for the Fraunhofer diffraction can be conducted, which shows the dependency on divergence.

$$\mathcal{F} \left\{ \left( \exp \left\{ -i \left( (n-1) \left( \frac{k}{2R(z)} (x^2 + y^2) \right) \right) \right\} \right) \otimes \left( \frac{\exp[-i(kz + \psi(z))]}{w(z)} \cdot \exp \left[ \frac{-(x^2 + y^2)}{w^2(z)} - \frac{ik(x^2 + y^2)}{2R(z)} \right] \right) \right\} \quad (15)$$

A different point of view, conveniently in the sense of the LiDAR mission, considers the use of focal lengths to change the Gaussian beam diffraction, giving the same result as the above Fourier transform, i.e. the divergences influence the intensity profile.

$$E(x_1, y_1) = \iint \left[ \exp \left( ik \left( \frac{-2xx_1 - 2yy_1}{2z} + \frac{x^2 + y^2}{2z} \cdot t_{lens}(x, y) \cdot E(x, y) \right) \right) \right] dx dy \quad (16)$$

Avoiding the quadratic terms and using the following relationship, the Fraunhofer diffraction pattern can be conducted.

$$\frac{k}{2z} = (n-1) \frac{k}{2R_1} \quad (17)$$

Adjusting the lens formula gives:

$$\frac{1}{f} = (n-1) \left[ \frac{1}{R_1} - \frac{1}{R_2} \right] \quad (18)$$

The final result shows the dependency of focus length to the Fourier transform of the aperture field.

$$E(x_1, y_1) = \iint \exp \left[ -i \frac{k}{f} (xx_1 + yy_1) \cdot t(x, y) \cdot E(x, y) \right] dx dy \quad (19)$$

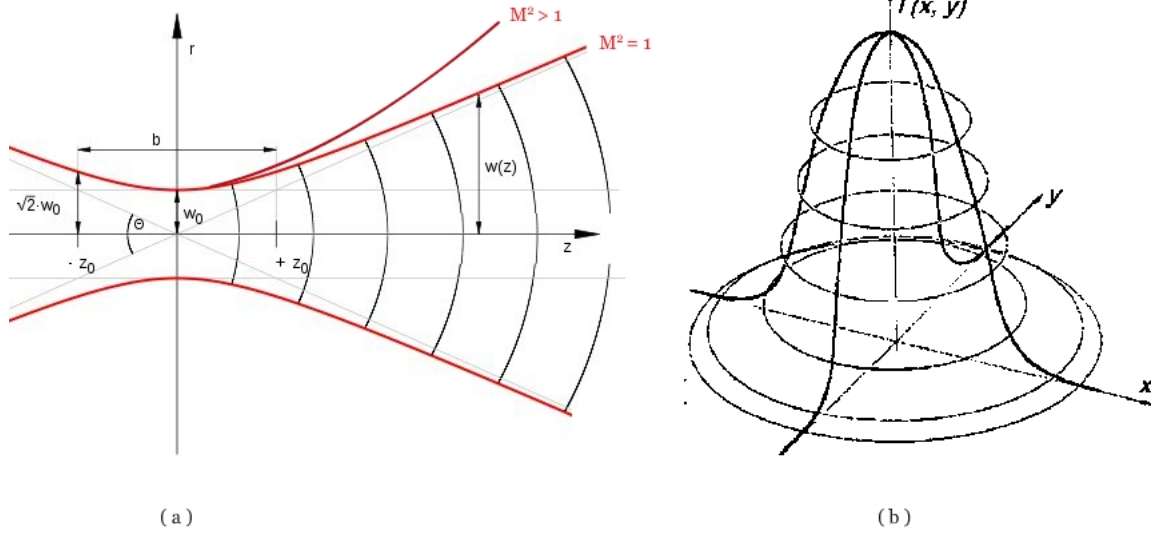


Figure 7

### 0.1.5 Thermal Control

Basically, there are three critical parts (LDA, Nd:YAG laser crystal and optical components after polarizer disk) of the laser configuration in terms of thermal control. All of these components shall be considered in this subsection.

*LDA.* The constituent parts and materials of a typical LDA are the diode die (laser bar) and the mechanical structure. The packaging design and materials enable the array of laser bars to stay together in a stack, to be energized electrically (with a relatively high drive current), to pass the heat generated out of the unit to the mounting surface (thermal path, heat sinking), to be sufficiently rugged against mechanical insults, to provide a standard mounting interface (screws or clamps) and to be as small as possible. The active region of the LDA, where heat is generated, is only about 1 micron wide, located about 3 microns from the P-side of the bar. The bars are about 0.1 [mm] wide and typically spaced about 0.5 [mm] from each other. Waste energy in the form of heat must be conductively transferred into the solder material and from there into the heat sink material (typically BeO or CuW) as rapidly as possible. The solder material of choice is a soft Indium alloy for its ductile property allowing the bar and the heat sink to expand or contract at different rate with temperature. The LDA manufacturers try to use materials which possess higher thermal conductivity and a relatively comparable coefficient of thermal expansion (CTE) in order to minimize the thermal resistance of the device and the induced mechanical stresses. Additionally important to reducing mechanical stress is consideration of the use of soft solders

which are highly pliable with a relatively low melting point ( $\sim 160$  C). Post life test analysis indicates that solder deformation caused solder roll-over, in turn creating voids, which increase thermal resistance. When coupled with built-in stress due to fabrication, such roll over, in time often obstructs emitters, leading to increased heating, or extends across the bar from anode to cathode causing bar shorts which eventually result in contaminations to the emitter face and localized hot spots, further degrading performance. Excessive heating and thermal cycling of the LDA active regions plays a key role in limiting the reliability and lifetime of LDAs operated in the QCW mode, particularly where pulse widths are long. To improve the assemblies heat extraction performance, advanced materials are being considered for packaging LDAs, which have high thermal conductivity and a CTE (Coefficient of Thermal Expansion) that matches that of the laser bars to compensate inhomogeneous thermal strain.

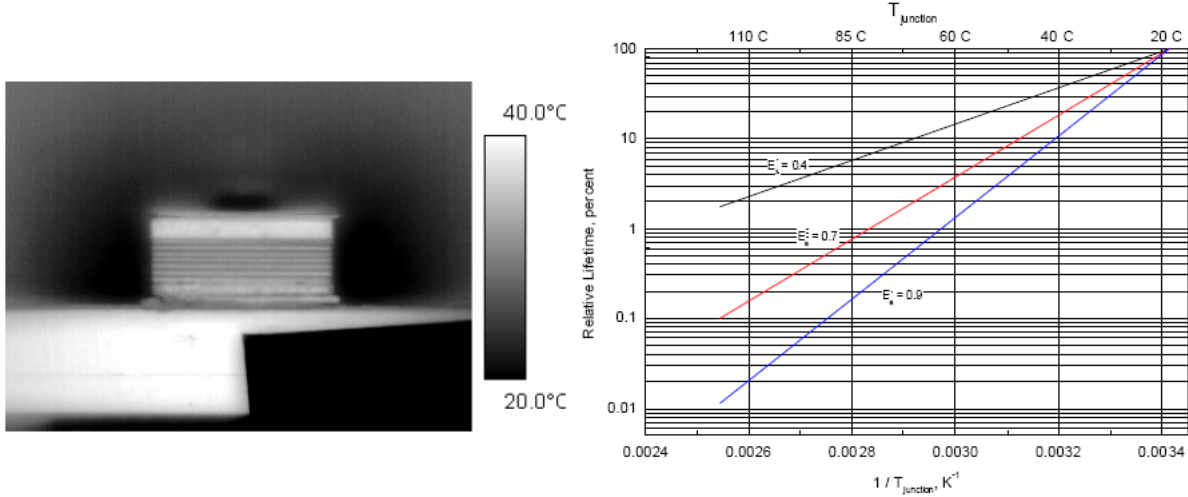


Figure 8: Thermal simulation of diode laser producing quasi-continuous waves (40[W],  $f_{rep}$  2,500 [Hz]). Graph shows a decrease of diode lifetime as function of junction temperature.[2]

**Nd:YAG! slab.** The Nd:YAG slab plays a central role in the laser configuration. To be able to cope with thermal stresses induced by the wave formation, the slab is thermally bonded to a molybdenum copper block in order to match the CTE.

*Dielctrical Component Temperature Dependency* Optical misalignment is a serious issue with the laser configuration, both in the manufacturing phase, as well during the actual mission. Refractive indices of dielectric components alter the beam translation and should be considered. Given the fact that the refractive index is a material parameter with a direct dependency of temperature (temperature-dependent Sellmeier equation), beam propagation can change unwillingly during mission.

### 0.1.6 Laser Lifetime Expectance

Multiple aspects influence the expected lifetime of the optical emitter device, such as power, temperature interval, repetition rate and intracavity properties. Since most lasers have a non-continuous

mode of operation (i.e. the duty factor is lower than 100%), reliability data for long-term cycles are not abundant available.

For damage-free operation in a harsh, hands-off, environment such as space, a major form of damage risk reduction is the creation of a large single intracavity mode to reduce peak fluence. Since resonator efficiency depends strongly on the inversion density of the gain medium, it is advantageous to confine the desired cavity mode as close as possible. To accomplish this, the 808 [nm] light from the diode arrays should be collimated by a single plano-convex cylindrical lens (for maximal efficiency, made of undoped YAG). By doing this, the probability of the existence of thermal lensing is reduced, increasing the beam quality and the lifetime.

Considering a constant value of  $f_{rep}$  of 5,000 [Hz], the total number of pulses equals  $788.4 \cdot 10^9$  [pulses/5 years]. All optical components should be able to cope with the large amount of pulses and the peak power implied by these pulses, i.e. the energy damage threshold of the dielectric components should be higher than the incoming energy of the electromagnetic radiation. Since  $I_{p,damage}$  is given with a temporal resolution in the order of a single pulse width ( $\sim 10$  [ns]), individual pulses can be analyzed. Stationary calculations can be conducted with the information based on the electromagnetic radiation energy and hence, the proper optical elements could be chosen ( $I_{p,damage} > I_p$ ).

[3] shows an experimental set-up, where the lifetime of a DPSSL is investigated, using approximately the same laser configuration with  $f_{rep} = 242$  [Hz] and  $E_p = 0.0150$  [J]. The pulse energy is much larger than the value of  $E_p$  in the case of the LiDAR mission described in this report ( $\sim 0.001$  [J]). ?? shows the results. After  $2.4 \cdot 10^9$  shots, there was no damage found in any of the cavity optics, but inspection of the diodes revealed that a single bar was lost on one array. After the first year, the pump pulse length was increased from 89 [ $\mu$ m] to 105 [ $\mu$ m] to restore the output energy to 15 [mJ]. This roughly simulated the procedure that would be performed in space in order to maintain an altimetry link. The final result was that after more than  $4.8 \cdot 10^9$  10 - 15 [mJ] laser pulses, there was no optical damage present in the system [3]. This clearly indicates that the LDAs lifetime considerations are important for the entire laser system. AlGaAs lasers can suffer from catastrophic optical damage (COD), rapid degradation, and gradual degradation. These phenomena are due to darkline defect propagation and a high surface recombination rate [4].

[3] shows an experimental set-up, where the lifetime of a DPSSL is investigated, using approximately the same laser configuration with  $f_{rep} = 242$ [Hz] and  $E_p = 0.0150$ [J]. The pulse energy is much larger than the value of  $E_p$  in the case of the LiDAR mission described in this report ( $\sim 0.001$  [J]). ?? shows the results. After  $2.4 \cdot 10^9$  shots, there was no damage found in any of the cavity optics, but inspection of the diodes revealed that a single bar was lost on one array. After the first year, the pump pulse length was increased from 89 [ $\mu$ m] to 105 [ $\mu$ m] to restore the output energy to 15 [mJ]. This roughly simulated the procedure that would be performed in space in order to maintain an altimetry link. The final result was that after more than  $4.8 \cdot 10^9$  10 - 15[mJ] laser pulses, there was no optical damage present in the system [3]. This clearly indicates that the LDAs lifetime considerations are important for the entire laser system.

Taken into account the fact that the total number of shots in five years exceed the number of total shots delivered by a single LDA without considerable loss in power and beam quality, the obvious consequence is that multiple LDAs should be implemented within the structure. Given the fact that individual laser diode has dimensions  $\sim 0.01$ [m], multiple diodes could be added to form

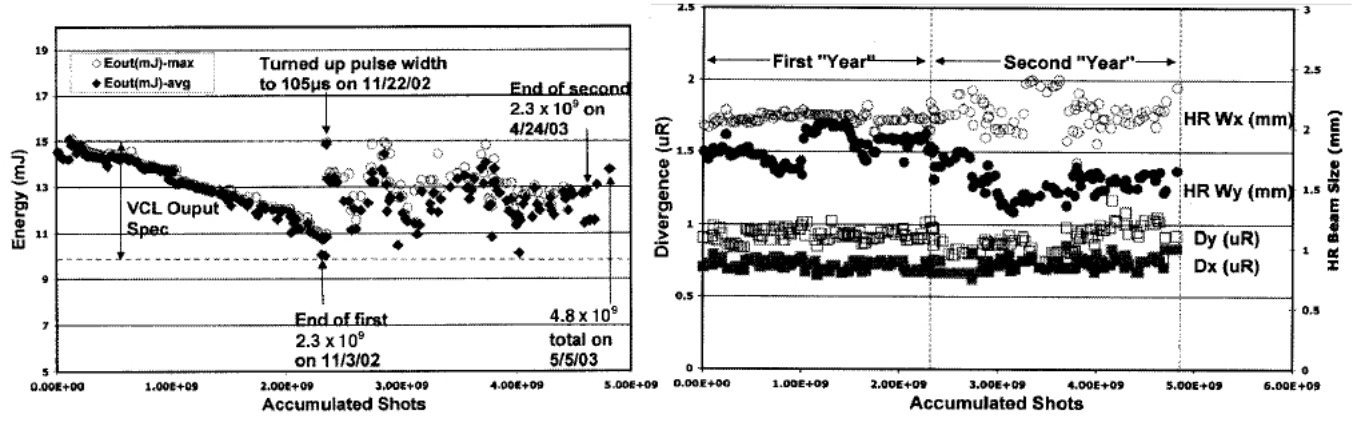


Figure 9: Results of the conducted experiments. Results shows a steady decay in output power. After one year without any modifications, the system was reintegrated and inspected. The pulse length decrease are used to compensate for the loss in pulse power. After two years and multiple quasi-continuous shots, no optical failure occurred , except single diode on the LDA failure.

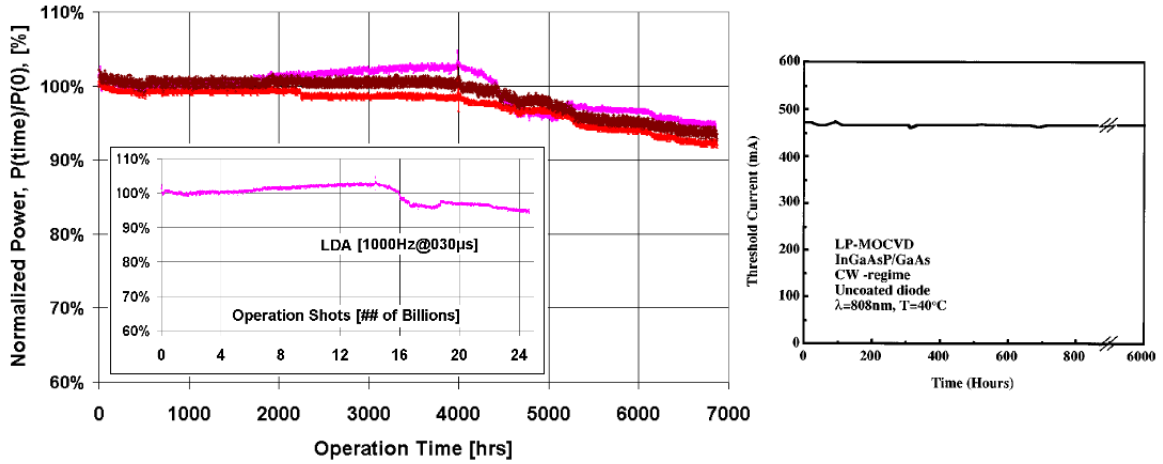


Figure 10: Space-graded conductively cooled expected diode lifetime in terms of output pulses and lifetime hours (100% duty cycle)

a LDA matrix.

### 0.1.7 laser Focus Calculation

The figure 12 on page 15 gives a overview of the emitter optics. In order to diverge or focus the laser beam, it is possible to move the parabolic mirror up or down from the exact focus position. In this case, the divergent angle  $\gamma$  needs to be calculated, which can be verified or optimized later

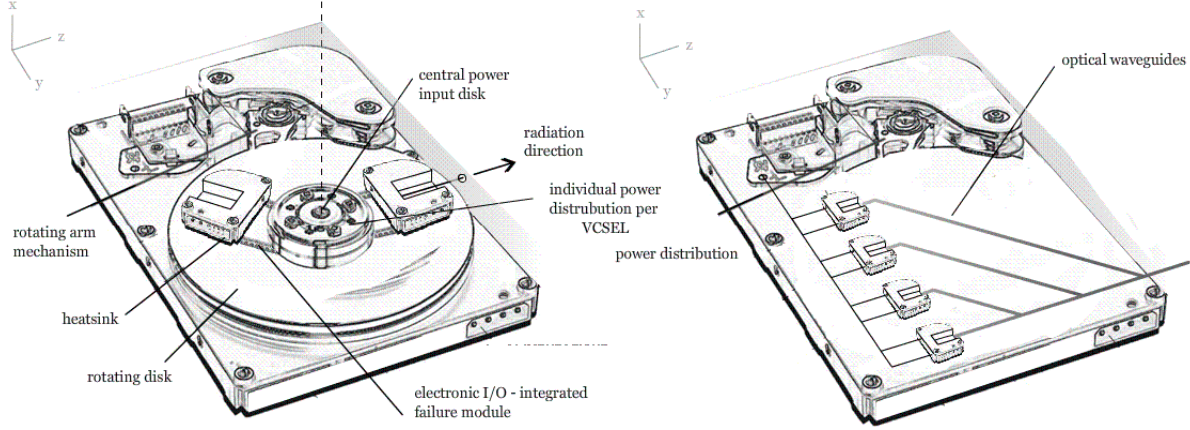


Figure 11: Two possible design options for the LDA matrix. For both cases, the individual LDAs should have communication within the subsystem, to be able to react on the failure of a single LDA. Left: Rotating mechanism (downside: extra translation, hard to produce accurately, increased power budget), right: using optical waveguides to translate the electromagnetic radiation towards the Nd:YAG laser

on to obtain the desired footprint size. The calculation drawing is shown in figure 13 on page 16.

In the figure 13,  $p_1$  is the parabolic mirror positioned at the exact focus point  $f$ , and  $p$  is the distance between  $f$  and origin.  $p_2$  is the parabolic mirror with the exact same shape but which is moved away from focus point with distance  $\xi$ .  $L_{in}$  indicates the incoming light.  $L_{out1}$  is the outgoing light due to  $p_1$  and  $L_{out2}$  is the outgoing light due to  $p_2$ . Meanwhile,  $r_1(x_1, y_1)$ ,  $r_2(x_2, y_2)$  are the reflected points due to  $p_1$  and  $p_2$ . The purpose of this focusing calculation is to find the divergent angle  $\gamma$  with respect to the design parameters  $p$ ,  $\xi$  and reflection point  $r_1(x_1, y_1)$ . [1]Parabolic mirror  $p_1$  has the equation 20, and  $p_2$  has the equation 21.

$$y = -\frac{1}{4p}x^2 \quad (20)$$

$$y = -\frac{1}{4p}x^2 + \xi \quad (21)$$

The equation 22 for incoming light line  $L_{in}$  can be obtained since  $r_1(x_1, y_1)$  is known in this case.

$$y = \frac{y_1 + p}{x_1}x - p \quad (22)$$

Insert equation 21 into equation 22,  $x_2$  of  $r_2(x_2, y_2)$  can be obtained as:

$$x_2 = \frac{-\frac{y_1+p}{x_1} + \sqrt{\frac{y_1+p}{x_1}^2 - \frac{\xi-p}{p}}}{\frac{1}{2p}} \quad (23)$$

Next step is to find the tangent line of  $p_2$  at  $r_2$ :

$$\left(\frac{dy}{dx}\right)_{x_2} = -\frac{1}{2p}x_2 = \tan(\mu_3) \Rightarrow \mu_3 = \arctan\left(-\frac{1}{2p}x_2\right) \quad (24)$$

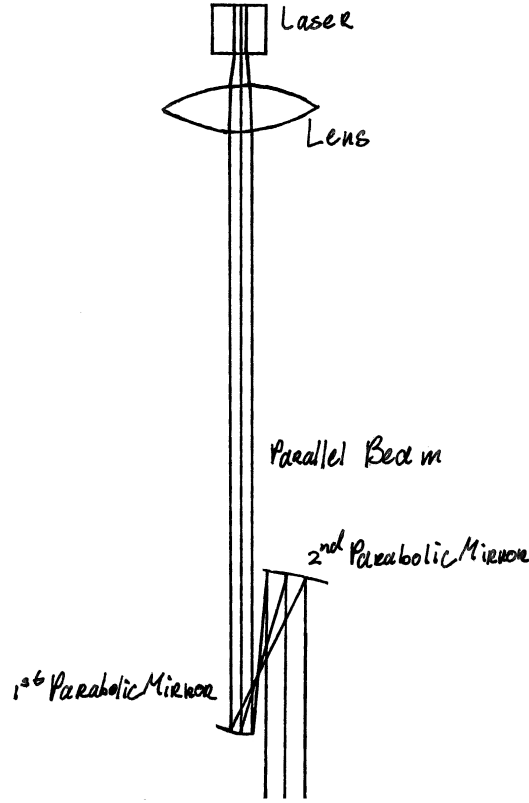


Figure 12: Emitter optics drawing

In the figure, 't' is the tangent line at point  $r_2$ , and 'N' is the normal line perpendicular to the tangent line. The normal line 'N' is also the angle bisect, and  $\mu_2$  is a half of the reflecting angle. From the drawing, these relations can be found:

$$\mu_1 + \mu_2 + \mu_3 = 90^\circ = \mu_1 + \mu_2 + \mu_4 \implies \gamma = \mu_2 - \mu_4 = \mu_2 - \mu_3 \quad (25)$$

To find  $\mu_2$ ,  $\mu_1$  need to be calculated.  $\mu_1$  is the tangent angle of  $L_{in}$  at  $r_1$  or  $r_2$ :

$$\left(\frac{dy}{dx}\right)_{x_1, y_1} = \frac{y_1 + p}{x_1} = \tan(\mu_1) \implies \mu_1 = \text{atan}\left(\frac{y_1 + p}{x_1}\right) \implies \mu_2 = 90\text{deg} - \mu_1 - \mu_3 \quad (26)$$

Insert value of  $\mu_2$  and  $\mu_3$  to equation 25, so the divergent angle  $\gamma = f(p, \xi, r_1(x_1, y_1))$  is obtained. Put these equations into Excel, and it is much easier to see how is  $\gamma$  verified. For instance, give values for  $p=350[\text{mm}]$ ,  $\xi = 5[\text{mm}]$  and  $x_1=5[\text{mm}]$ ,  $\gamma=0.01169[\text{deg}]$ , which will give the footprint size of 102 meters. By adjusting the  $\xi$ , the mirror has a divergence of  $20.4[\text{m/mm}]$  for the same  $p$  and  $x_1$ .

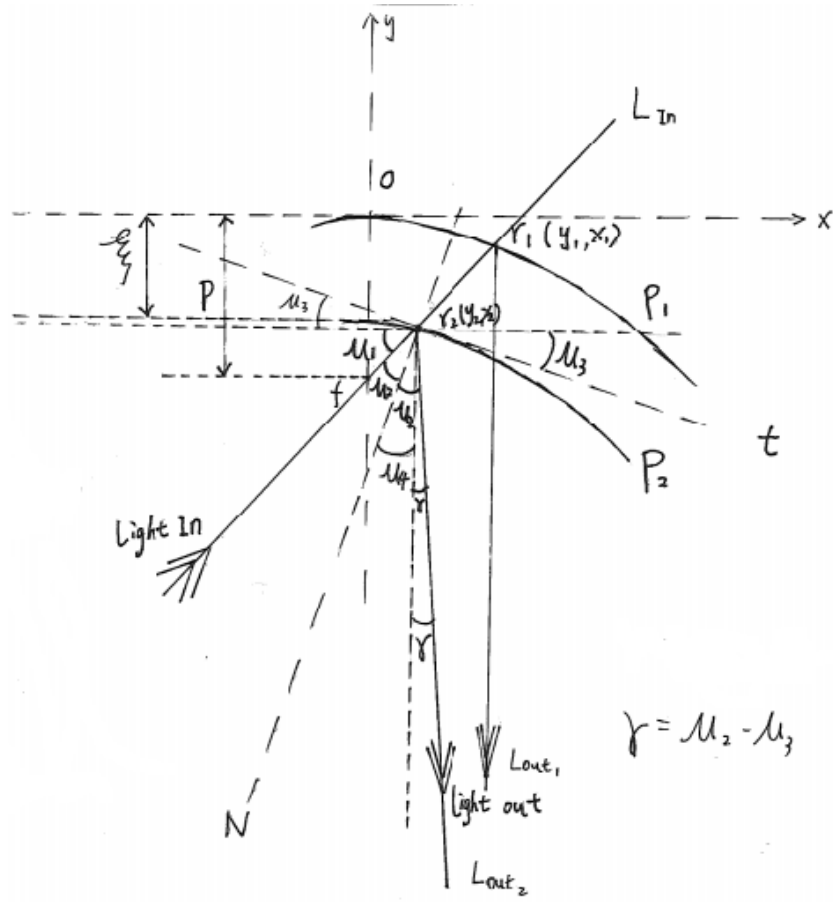


Figure 13: Focus calculation draft



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