APOLLO G&N Specification
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PROCUREMENT SPECIFICATION PRODUCT CONFIGURATION AND ACCEPTANCE TEST REQUIREMENTS FIXED MEMORY MODULE B1-B6 ASSEMBLY DRAWING NO. 2003972

Record of Revisions

Date	Revision TDRR		Pages Revised -	Appro	Approvals	
	Letter 1	No.	Revised	MIT .	NASA	
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This specification consists of pages 1 to 21 inclusive.

APPROVALS	AIN	ECHALL 3/6/68	• mar 68	CCB CAA	F. Moddond BSPelows Mass
	NASA MSC		MIT/IL	MA 3/10/18	RAY

- 3.1.9 Sense Outputs. With the input and timing waveforms as specified in paragraph 4.2.9, the sense outputs shall be as specified in Figure 3 and as follows:
- 3.1.9.1 The maximum outut on HFT and sense lines, parity through sign shall be as follows:

V & V MMT 350 mv at 25 C

The minimum output on the MFT shall be as follows:

V & V MLT 145 mv at 25°C

V & V MILT 175 mv at -10°C

V_D & V_M HLT 120 mv at 70°C

The minimum V₁ output on sense lines -10°C = 50 MV, +25°C = 45MV +70°C = 45MV.

Test Sequence:

- A. Ambient Electrical Test Number One
- B. Thermal Conditioning per 3.1.1.
- C. Vibration per 3.1.11
- D. Thermal Extremes Electrical Test per 3.1.10
- 23. Continuity and Isolation Tests per 3.1.5 and 3.1.3
- L Ambient Temperature Test Humber Two.

The MFT V_p minimum and V_M minimum shall not change by more than \pm 25 mV from Ambient Test Number One to Ambient Test Number Two.

3.1.9.2 Sense line zero noise (V₀ measured between 1.1.=0.05 page and 2.3 -0.05 page after the start of Logic Reset (FR) shall be MMT 25 mv. See Figure 3.

3.1.9.3 An all zero output shall occur when Logic STAGAT is zero. V_0 max = NMT 5 mv, 1.1 to 2.3 µsec after RSET.

TABLE IV

Reg	uirements	Verification
	3.1.1	4.2.2
	3.1.2	4.2.3
	3.1.3	4.2.4
	3.1.4	4.2.5
	3.1.5.1	4.2.6.1
	3.1.5.2	4.2.6.2
	3.1.5.3	4.2.6.3
	3.1.7	New 2018 4.2.7 교육
	3.1.8	4.2.8
	3.1.9.1	4.2.9.1
	3.1.9.2	4.2.9.2
	3.1.9.3	4.2.9.3
	3.1.9.4	4.2.9.4
	3.1.10.1	4.3.1.1
*-1 _E	3.1.10.2	4.3.1.2
	3.1.11	4.3.2
	3.2.1	4.2.1
	3.2.2	4.2.10

4.1.1 Test Conditions

4.1.1.1 Evironmental. Unless otherwise specified, the assemblies shall be tested under the following ambient conditions:

a.	Temperature	25°C ± 10C°
b.	Relative Humidity	90% max
C.	Barometric Pressure	28 to 32 inches of Hg.

- 4.1.2 Monconforming Units. Failure of the unit to pass any examination or test of this specification shall automatically classfy the unit as nonconforming. Each nonconforming unit corrected by the contractor shall be reinspected. Reinspection may be limited to the test or examination which defined the nonconformance, or, when directed by the cognizant inspector, a complete retest and reexamination may be required. Monconforming units which have not been corrected will be considered for acceptance only upon formal application by the contractor to the cognizant NASA representative.
- 4.1.3 Input Voltage. Input voltage, when required, shall be as specified in Paragraph 3.1.3.

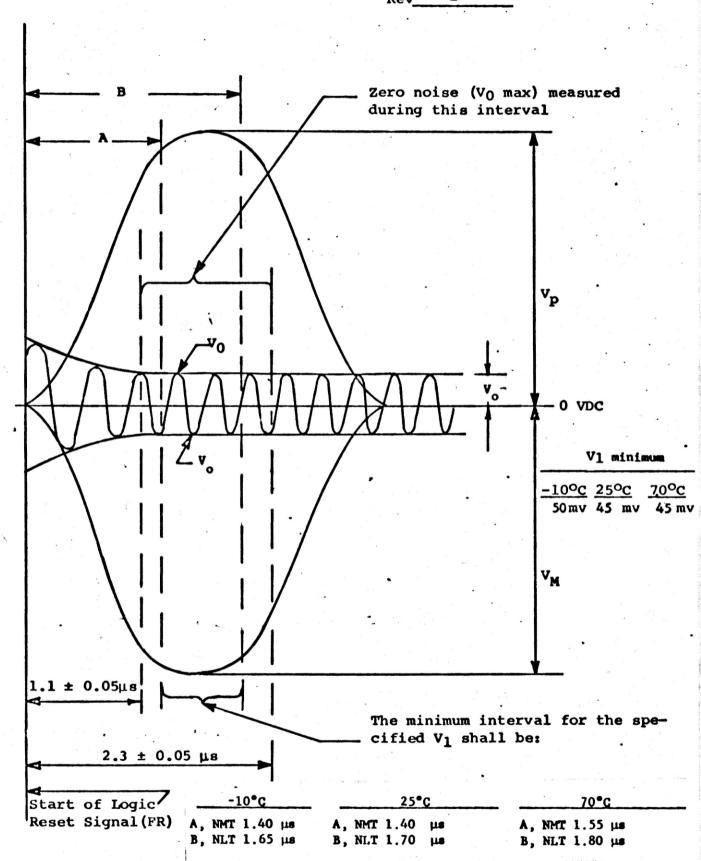


FIGURE 3 ROPE OUTPUT WAVEFORM CHARACTERISTICS .