Brushed DC Motor Model Derivation State Space Form Transfer Functions Simulated Response

Lecture Module - Electrical Systems

ME3050 - Dynamic Modeling and Controls

Mechanical Engineering
Tennessee Technological University

Topic 4 - Example: DC Motor



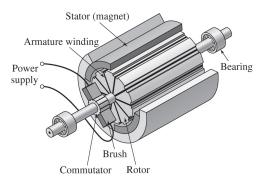
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Electrical Systems

- Brushed DC Motor
- Model Derivation
- State Space Form
- Transfer Functions
- Simulated Response

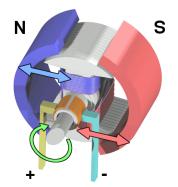
Model Derivation

Armature Controlled Brushed DC Motor



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Brushed DC Motor

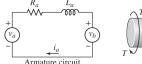


Animation on Web

Model Derivation

Armature Controlled Brushed DC Motor







 v_a : armature voltage (input)

 R_a : armature resistance Torque on armature

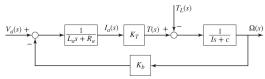
$$T = (nBLi_a) r = (nBLr) i_a = K_T i_a$$

Back EMF (electromotive force) voltage

$$v_b = nBLv = (nBLr)\omega = K_b\omega$$

Model Derivation

Armature Controlled Brushed DC Motor



Kirchoff's Voltage Law

$$v_a - R_a i_a - L_a \frac{di_a}{dt} - K_b \omega = 0$$

Newtons's Second Law

$$I\frac{d\omega}{dt} = T - c\omega - T_L = K_T i_a - c\omega - T_L$$

Image: System Dynamics, Palm, 4th, Pg. 376-378



State Space Form

State-Variable (State-Space) form

$$\frac{di_a}{dt} = \dot{x}_1 = \frac{1}{L_a} \left(v_a - R_a i_a - K_b \omega \right) = \begin{bmatrix} -\frac{R_a}{L_a} & -\frac{K_b}{L_a} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} \frac{1}{L_a} & 0 \end{bmatrix} \begin{bmatrix} v_a \\ T_L \end{bmatrix}$$

$$\frac{d\omega}{dt} = \dot{x}_2 = \frac{1}{L} \left(K_T i_a - c\omega - T_L \right) = \begin{bmatrix} -\frac{K_T}{L} & -\frac{c}{L} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 & -\frac{1}{L} \end{bmatrix} \begin{bmatrix} v_a \\ T_L \end{bmatrix}$$

Write the state equation in matrix form with states $x_1=i_a$, and $x_2=\omega$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -\frac{R_a}{L_a} & -\frac{K_b}{L_a} \\ \frac{K_T}{I} & -\frac{c}{I} \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} \frac{1}{L_a} & 0 \\ 0 & -\frac{1}{I} \end{bmatrix} \begin{bmatrix} v_a \\ T_L \end{bmatrix}$$

Transfer Functions

The input-output relationships can be represented by the following transfer functions.

Armature Current to Armature Voltage

$$\frac{I_a(s)}{V_a(s)} = \frac{Is + c}{L_a I s^2 + (R_a I + c L_a) s + c R_a + K_b K_T}$$

Armature Current to External Load

$$\frac{I_{a}(s)}{T_{L}(s)} = \frac{K_{b}}{L_{a}Is^{2} + (R_{a}I + cL_{a})s + cR_{a} + K_{b}K_{T}}$$



Transfer Functions

Armature Angular Velocity to Armature Voltage

$$\frac{\Omega(s)}{V_a(s)} = \frac{K_T}{L_a I s^2 + (R_a I + c L_a) s + c R_a + K_b K_T}$$

Armature Angular Velocity to External Load

$$\frac{\Omega\left(s\right)}{T_{L}\left(s\right)} = \frac{L_{a}s + R_{a}}{L_{a}Is^{2} + \left(R_{a}I + cL_{a}\right)s + cR_{a} + K_{b}K_{T}}$$

Transfer Functions

Final Value Theorem: To find the value of a function x(t) as $t \to \infty$

$$x\left(\infty\right)=\lim_{x\to\infty}f(x)$$

Use final value theorem to find steady state value to step input on V_a , T_L

$$i_a = \frac{cV_a + K_b T_L}{cR_a + K_b K_T}$$

$$\omega = \frac{K_T V_a - R_a T_L}{cR_a + K_b K_T}$$

Simulated Response

The following MATLAB code defines a state space system object and simulates the system response to various inputs.

Simulated Response

```
% define components of the state equation
A=[-Ra/La -Kb/La
  KT/I - c/I:
% B matrix is 2x2 because u vector is 2x1
B=[1/La\ 0]
  0 - 1/I;
% use first two states as outputs
C=[1 0
  0 1];
% the D matrix shape of B matrix
D = [0 \ 0]
  0 0];
```

Simulated Response

```
\% calculate the steady state step response
Va=12;
TL=0:
ia_ss=(c*Va+Kb*TL)/(c*Ra+Kb*KT)
% create a state space model object
sys1=ss(A,B,C,D);
% simulate a step response
figure(1)
time=0:0.001:1;
opts=stepDataOptions('StepAmplitude', Va);
step(sys1,time,opts); grid on
```