Astrodynamics HW 3

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1.

$$\mathbf{r}_{in} = 7000km$$

Kepler's 3rd law

$$P^2=rac{4\pi^2}{\mu}a^3$$

$$\mu = 398600.4418 km^3/s^2 \ P_{fin} = 12 hours imes 60 min imes 60 sec$$

In [14]: Rfin =
$$cbrt(mu*Pfin^2/4/pi^2)$$
;

$$\mathbf{r}_{fin}=26610.2228km$$

$$a_{trans} = rac{r_{in} + r_{fin}}{2} \ = 16805.1114 km$$

using vis-viva equation

$$\Delta V = \sqrt{rac{2\mu}{r} - rac{\mu}{a}}$$

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by vis-viva equation

$$\mathbf{V}_{in} = \sqrt{rac{2\mu}{\mathbf{r}_{in}} - rac{\mu}{\mathbf{a}_{in}}}, \mathbf{V}_{fin} = \sqrt{rac{2\mu}{\mathbf{r}_{fin}} - rac{\mu}{\mathbf{a}_{fin}}}$$

$$\mathbf{V}_{trans,a} = \sqrt{rac{2\mu}{\mathbf{r}_{in}} - rac{\mu}{\mathbf{a}_{trans}}}, \mathbf{V}_{trans,b} = \sqrt{rac{2\mu}{\mathbf{r}_{fin}} - rac{\mu}{\mathbf{a}_{trans}}}$$

$$\Delta \mathbf{V}_a = \sqrt{rac{\mu}{\mathbf{r}_{in}}} - \sqrt{rac{2\mu}{\mathbf{r}_{in}} - rac{\mu}{\mathbf{a}_{trans}}}$$

$$\Delta \mathbf{V}_b = \sqrt{rac{\mu}{\mathbf{r}_{fin}}} - \sqrt{rac{2\mu}{\mathbf{r}_{fin}} - rac{\mu}{\mathbf{a}_{trans}}}$$

In [10]: Vin = sqrt((2mu/Rin)-(mu/Rin));

In [44]: Vfin = sqrt((2*mu/Rfin)-(mu/Rfin));

In [45]: Vtrans_a = sqrt((2mu/Rin)-(mu/Atrans));

In [46]: Vtrans_b = sqrt((2mu/Rfin)-(mu/Atrans));

In [50]: deltaVa = Vtrans_a - Vin;

In [52]: deltaVb = Vfin - Vtrans_b;

$$\left\{egin{aligned} \Delta V_a = 1.9496 km/s \ \Delta V_b = 1.3724 km/s \end{aligned}
ight.$$

In [53]: deltaV = deltaVa + deltaVb;

$$\Delta V = \Delta V_a + \Delta V_b \ = 3.3220 km/s$$

2.

$$Period = 318min$$

In [14]: P = 318*60;

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In [18]: mu = 398600.4418;
In [28]: a = cbrt(P^2*mu/4/pi^2);
In [56]: incl = 10*2pi/360;
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In identical equatorial plane, we can do inclination-only maneuver on two nodes which is ascending and descending nodes.

$$P^2 = \frac{4\pi^2}{\mu} a^3$$

$$a = 15433km, e = 0.5$$

$$egin{aligned} r &= rac{a(1-e^2)}{1+ecos
u}(km) \ &= rac{15432.8071(1-0.5^2)}{1+0.5cos
u}(km) \ &= rac{11574.6053}{1+0.5cos
u}(km) \end{aligned}$$

by vis-viva equation

$$v=\sqrt{rac{2\mu}{r}-rac{\mu}{a}}$$

$$=\sqrt{rac{\mu}{r}(2-rac{1-e^2}{1+0.5cos
u})}$$

at descending node,

$$V_{descending} = 4.3239 km/s$$

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at ascending node,

$$V_{ascending} = 8.2096 km/s$$

$$\Delta V_{i,only}^{
ightarrow} = \stackrel{
ightarrow}{V_{fin}} - \stackrel{
ightarrow}{V_{in}}$$

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In [62]: cosfpa1 = sqrt((1-e^2)*a^2/r/(2a-r));
In [63]: cosfpa2 = sqrt((1-e^2)*a^2/r2/(2a-r2));
In [61]: delV_a = 2cosfpa1*Vasc*sin(incl/2);
In [64]: delV_d = 2cosfpa2*Vdesc*sin(incl/2);
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1)ascending node

$$egin{aligned} sin(5^\circ) &= rac{\Delta V_{i,only,ascending}/2}{2cos(\phi_{fpa})V_{ascending}} \ &\Rightarrow \Delta V_{i,only,ascending} = 2cos(\phi_{fpa})V_{ascending}sin(5^\circ) \ &= 1.3846km/s \end{aligned}$$

2)descending node

$$egin{aligned} sin(5^\circ) &= rac{\Delta V_{i,only,descending}/2}{2cos(\phi_{fpa})V_{descending}} \ &\Rightarrow \Delta V_{i,only,descending} = 2cos(\phi_{fpa})V_{descending}sin(5^\circ) \ &= 0.6613km/s \end{aligned}$$

3.

$$altitude_{sat1} = 191.344km$$
 $altitude_{GEO} = 35781.35km$

Note. the initial and final orbits are circular, whereas the transfer orbits are elliptical.

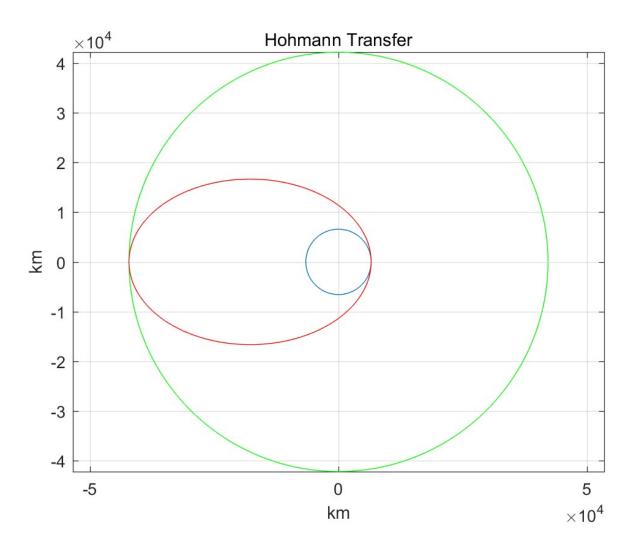
For the One-tangent burn transfer, it is given as follows.

$$u_{trnans,b} = 160^{\circ}$$

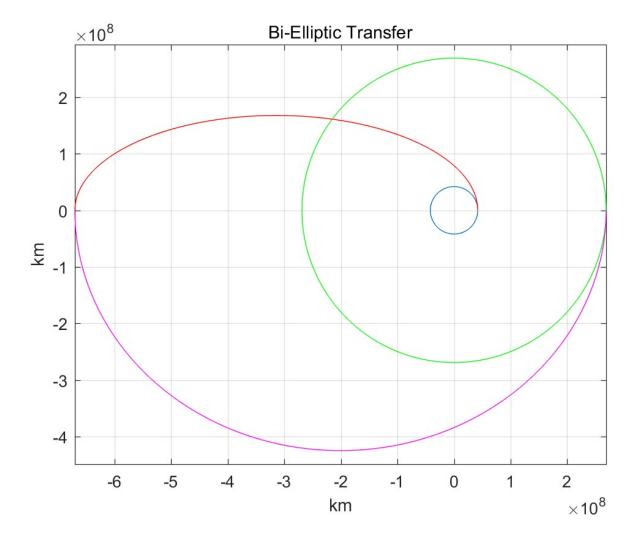
the total change in velocity, transfer time and comparing the results from each transfer.

hohmann(191.344+re, 35781.35+re,0,0,0,pi)

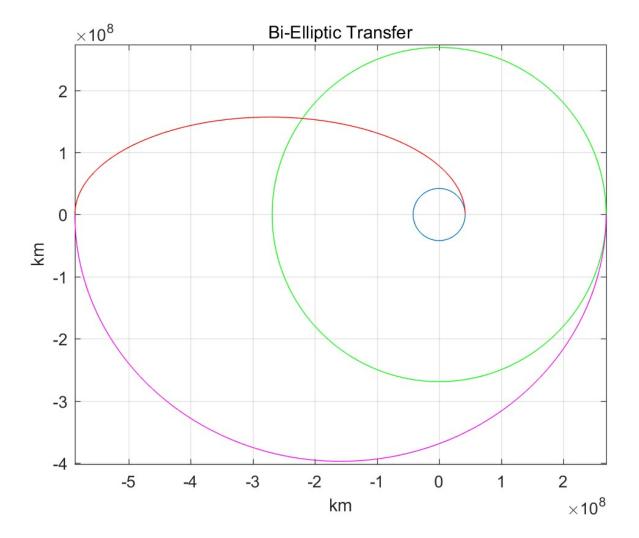
$$\left\{egin{aligned} \Delta V_{a} &= V_{trans,a} - V_{initial} \ \Delta V_{b} &= V_{final} - V_{trans,b} \ \Delta V_{total} &= \Delta V_{a} + \Delta V_{b} \end{aligned}
ight.$$



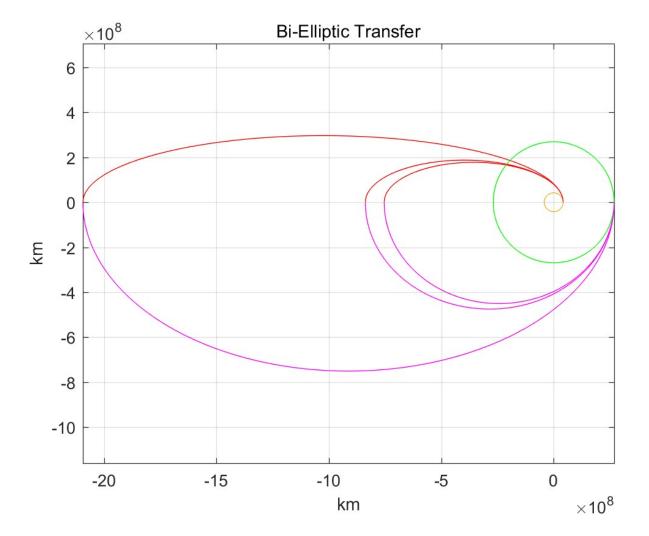
biellip(191.344+re,105112,35781.35+re,0, 0, 0, pi)



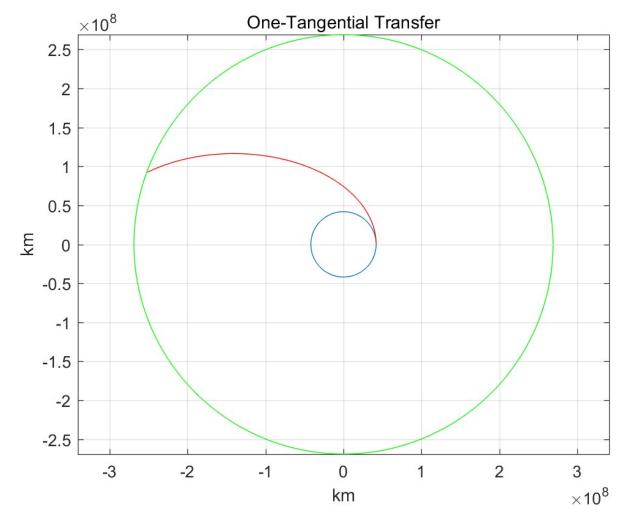
biellip(191.344+re,91973,35781.35+re,0,0,0,pi)



biellip(191.344+re,328475,35781.35+re,0, 0, 0, pi)



onetang(191.344+re,35781.35+re,0,0,0,160/180*pi)



In []: