

$$l_{plate}^2 = l_i^2 + l_{motor}^2 - 2l_i l_{motor} \cos \alpha$$

$$\alpha = \cos^{-1} \left(\frac{l_i^2 + l_{motor}^2 - l_{plate}^2}{2l_i l_{motor}} \right)$$

Figure 1: Cosine law.

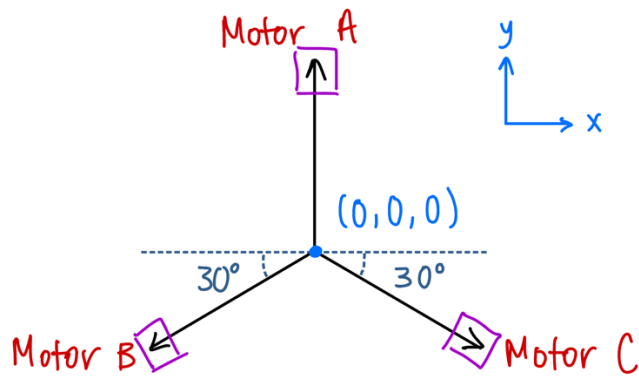


Figure 2: Motor orientations.

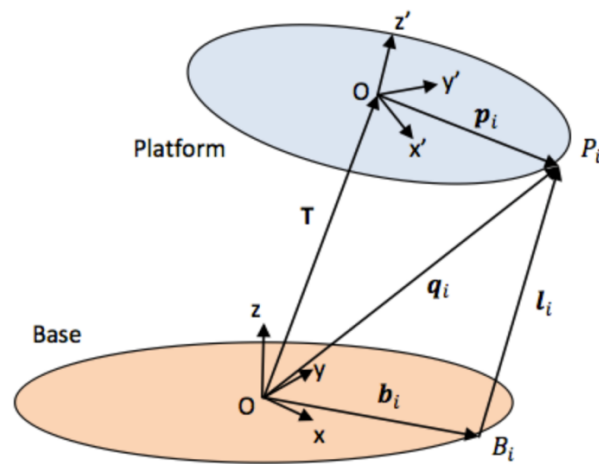


Figure 3: Diagram describing the plate vectors.

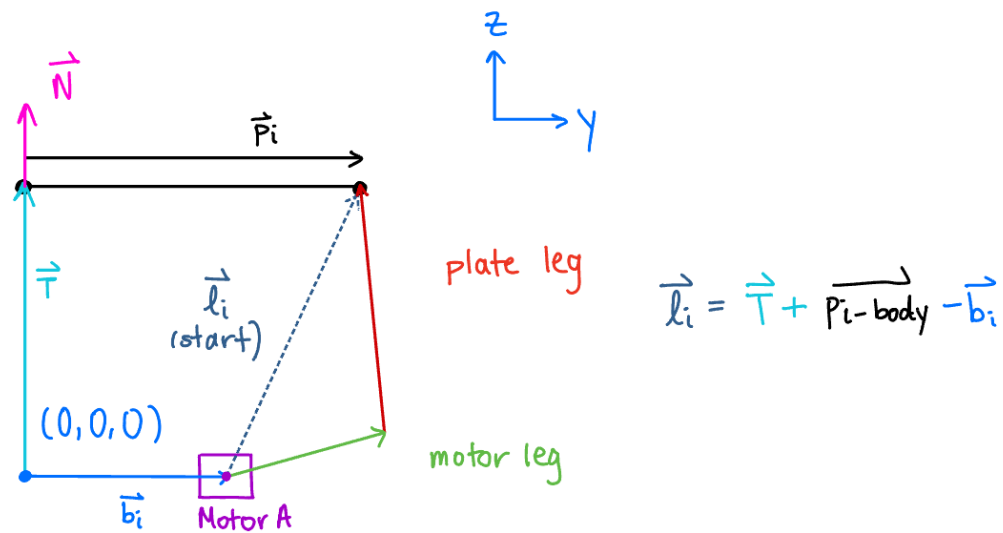


Figure 4: Motor A at rest.

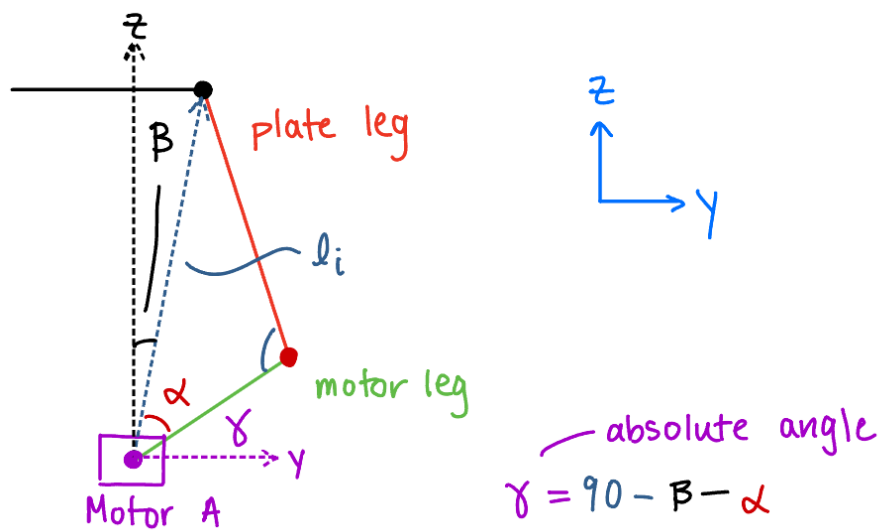


Figure 5: Gamma absolute motor angle calculation.

Define the rotation matrix for a tilt about the X and Y axis.

$${}^{Body}_{plat}R = Y_{rot} * X_{rot} \rightarrow \text{Rotation of platform w. r. t. body}$$

$$Y_{rot} = \begin{bmatrix} \cos \theta & 0 & \sin \theta \\ 0 & 1 & 0 \\ -\sin \theta & 0 & \cos \theta \end{bmatrix}$$

$$X_{rot} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \phi & -\sin \phi \\ 0 & \sin \phi & \cos \phi \end{bmatrix}$$

$${}^B_pR = \begin{bmatrix} \cos \theta & 0 & \sin \theta \\ 0 & 1 & 0 \\ -\sin \theta & 0 & \cos \theta \end{bmatrix} * \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \phi & -\sin \phi \\ 0 & \sin \phi & \cos \phi \end{bmatrix}$$

$${}^B_pR = \begin{bmatrix} \cos \theta & \sin \theta \sin \phi & \sin \theta \cos \phi \\ 0 & \cos \phi & -\sin \phi \\ -\sin \theta & \cos \theta \sin \phi & \cos \theta \cos \phi \end{bmatrix}$$

Backwards-solve the rotation matrix that will transform a unit \hat{k} vector to the plane's normalized normal vector.

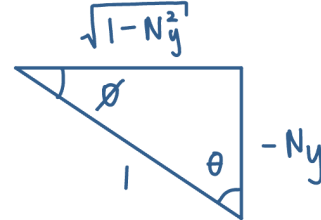
$$\begin{bmatrix} \cos \theta & \sin \theta \sin \phi & \sin \theta \cos \phi \\ 0 & \cos \phi & -\sin \phi \\ -\sin \theta & \cos \theta \sin \phi & \cos \theta \cos \phi \end{bmatrix} \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} = \hat{N} = \begin{bmatrix} N_x \\ N_y \\ N_z \end{bmatrix}$$

$$\begin{bmatrix} \sin \theta \cos \phi \\ -\sin \phi \\ \cos \phi \cos \theta \end{bmatrix} = \begin{bmatrix} N_x \\ N_y \\ N_z \end{bmatrix}$$

$$\phi = \sin^{-1}(-N_y)$$

$$\theta = \sin^{-1}\left(\frac{N_x}{\cos \phi}\right)$$

$$\theta = \sin^{-1}\left(\frac{N_x}{\sqrt{1 - N_y^2}}\right)$$



Once that is solved, substitute in ϕ and θ into the matrix. Then, apply that rotation to the $\overrightarrow{p_{l-plane}}$ vector to solve for the magnitude and direction of the rotated vector $\overrightarrow{p_{l-body}}$ to use in the \vec{l}_i equation.

$$\overrightarrow{p_{l-body}} = {}^{Body}_{plat}R * \overrightarrow{p_{l-plat}}$$

$$\overrightarrow{p_{l-body}} = \begin{bmatrix} \cos \theta & \sin \theta \sin \phi & \sin \theta \cos \phi \\ 0 & \cos \phi & -\sin \phi \\ -\sin \theta & \cos \theta \sin \phi & \cos \theta \cos \phi \end{bmatrix} * \overrightarrow{p_{l-plat}}$$

For angles $\theta_y = 5^\circ$ and $\phi_x = 5^\circ$, we can calculate \vec{N} as follows.

$$\begin{bmatrix} \sin 5^\circ \cos 5^\circ \\ -\sin 5^\circ \\ \cos 5^\circ \cos 5^\circ \end{bmatrix} = \begin{bmatrix} N_{x-norm} \\ N_{y-norm} \\ N_{z-norm} \end{bmatrix} = \begin{bmatrix} 0.08682 \\ -0.08716 \\ 0.99240 \end{bmatrix}$$

For angles $\theta_y = 5^\circ$ and $\phi_x = 5^\circ$, we can calculate \vec{l}_a as follows for Motor A. Given $\vec{p}_{a-plat} = \begin{bmatrix} 0 \\ 15 \\ 0 \end{bmatrix}$,

$$\vec{p}_{a-body} = \begin{bmatrix} \cos 5^\circ & \sin 5^\circ \sin 5^\circ & \sin 5^\circ \cos 5^\circ \\ 0 & \cos 5^\circ & -\sin 5^\circ \\ -\sin 5^\circ & \cos 5^\circ \sin 5^\circ & \cos 5^\circ \cos 5^\circ \end{bmatrix} * \begin{bmatrix} 0 \\ 15 \\ 0 \end{bmatrix} = \begin{bmatrix} 0.11394 \\ 14.94293 \\ 1.30236 \end{bmatrix}$$

Recall, $\vec{l}_a = \vec{T} + \vec{p}_{a-body} - \vec{b}_a$. Given $\vec{T} = \begin{bmatrix} 0 \\ 0 \\ 8 \end{bmatrix}$ and $\vec{b}_a = \begin{bmatrix} 0 \\ 12 \\ 0 \end{bmatrix}$,

$$\vec{l}_a = \begin{bmatrix} 0 \\ 0 \\ 8 \end{bmatrix} + \begin{bmatrix} 0.11394 \\ 14.94293 \\ 1.30236 \end{bmatrix} - \begin{bmatrix} 0 \\ 12 \\ 0 \end{bmatrix} = \begin{bmatrix} 0.11394 \\ 2.94293 \\ 9.30236 \end{bmatrix}$$

For a known $\vec{l}_a = \begin{bmatrix} 0.11394 \\ 2.94293 \\ 9.30236 \end{bmatrix}$, we can calculate the relative motor angle $\alpha_{plate-leg}$ as follows. Given

$l_{motor-leg} = 3$ and $l_{plate-leg} = 6$,

$$|\vec{l}_a| = \sqrt{0.11394^2 + 2.94293^2 + 9.30236^2} = 9.7574$$

$$\alpha_{plate-leg} = \cos^{-1} \left(\frac{l_a^2 + l_{motor-leg}^2 - l_{plate-leg}^2}{2l_a l_{motor-leg}} \right) = \cos^{-1} \left(\frac{9.7574^2 + 3^2 - 6^2}{2 \times 9.7574 \times 3} \right) = 54.8268^\circ$$

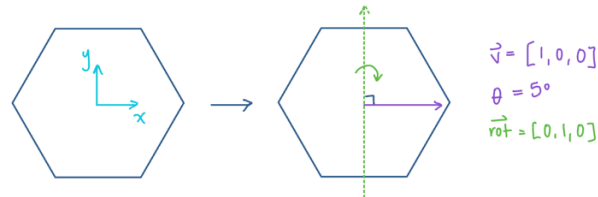
After obtaining the relative motor angle, we can find the angle β_a of \vec{l}_a with respect to the unit \hat{k} vector. See `absolute_motor_angle.png` for more details.

$$\beta_a = \cos^{-1} \left(\frac{\hat{k} \cdot \vec{l}_a}{|\hat{k}| |\vec{l}_a|} \right) = \cos^{-1} \left(\frac{\begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} \cdot \begin{bmatrix} 0.11394 \\ 2.94293 \\ 9.30236 \end{bmatrix}}{9.7574} \right) = 17.5670^\circ$$

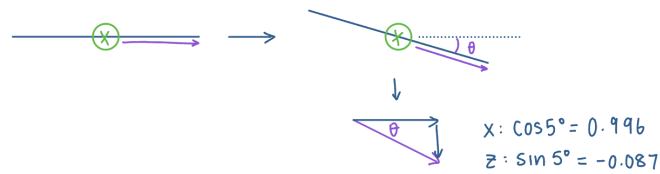
Subtracting $\alpha_{plate-leg}$ and β_a from 90° yields the absolute motor angle γ_a w. r. t. the horizontal.

$$\gamma_a = 90^\circ - \alpha_{plate-leg} - \beta_a = 90^\circ - 54.8268^\circ - 17.5670^\circ = 17.61^\circ$$

For a direction vector of $\vec{v} = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix}$ and a tilt magnitude of $\theta = 5^\circ$, the normal vector of the plate can be calculated. First, the axis of rotation is determined to be $\vec{v}_{rot-axis} = \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix}$, which is the orthogonal vector to the direction vector in the x-y plane.



Then, the plate is tilted about this axis in the direction of the direction vector, about magnitude θ . Vector decomposition is used to determine the new, rotated direction vector.



$$\vec{v}_{rot} = \begin{bmatrix} 0.996 \\ -0.087 \\ 0 \end{bmatrix}$$

Given the rotated direction vector and the axis of rotation, these are crossed to find the third orthogonal vector, which is the normal vector to the plane.

$$\vec{N} = \vec{v}_{rot-axis} \times \vec{v}_{rot} = \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix} \times \begin{bmatrix} 0.996 \\ -0.087 \\ 0 \end{bmatrix} = \begin{bmatrix} -0.087 \\ 0 \\ -0.996 \end{bmatrix}$$

The vector is multiplied by -1 to ensure the Z-direction is positive.

$$\vec{N} = \begin{bmatrix} 0.087 \\ 0 \\ 0.996 \end{bmatrix}$$

