Particle Dynamics Project

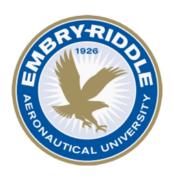
Pendulum with Drag

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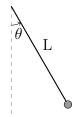
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1 Conceptualize the Problem

The first step to solving the problem is to conceptualize the system and understand what is being asked for. The idealized pendulum is a 1 degree of freedom (DOF) problem, and the best coordinate system to model the system is either normal-tangential (\hat{e}_n, \hat{e}_t) or polar $(\hat{e}_r, \hat{e}_\theta)$; this way the position of the mass is easily shown using only one variable, θ , and since most of the forces act tangentially on the mass, calculations made later will be less strenuous. We will be using a normal-tangential coordinate system.

1.1 Constants and Assumptions



Constants: Assumptions:

 $\begin{array}{lll} \text{Mass:} & \text{m} = 0.142 \text{kg} & \text{Frictionless} \\ \text{String Length:} & \text{L} = 0.5 \text{m} & \text{Released from Rest} \\ \text{Gravity:} & \text{g} = 9.81^{m}/s^{2} & \text{Particle Dynamics} \end{array}$

The system is released from rest at various initial angles of release:

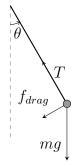
$$\theta_o = 5^{\circ}, 10^{\circ}, 15^{\circ}, 30^{\circ}, 60^{\circ}, \text{ and } 90^{\circ}$$

Drag force is defined as

$$\bar{F}_{Drag} = \frac{1}{2} \rho C_D A V^2 \left(-\hat{V} \right) = -1.65 \cdot 10^{-3} \dot{\theta} |\dot{\theta}| \hat{e}_t$$
 (1)

We are also asked to determine the following:

2 Free Body Diagram



 f_{Drag} : Drag force* acting against the velocity of the mass $(-\hat{v})$

mg: Mass · gravity, the weight of the object

T: The tension in the string

*Modeled by Eq.(1)

3 Coordinate Frame



 \hat{e}_n : + Pointing at the centroidal point about which the system rotates

 \hat{e}_t : + Pointing in the direction that represents counterclockwise tangential movement about the centroidal point

We determined the positive direction for the coordinate system based on planar dynamics conventions.

4 Sum of Forces

$$\sum \bar{F} = m\bar{a}_{cm}$$

Since we are using a normal-tangential coordinate system,

$$\bar{a}_{cm} = \begin{bmatrix} r\dot{\theta}^2 \\ r\ddot{\theta} \end{bmatrix}$$

There are no moments in this problem that are accounted for.

Without drag:

Sum of the forces in the normal direction,

$$\sum F_n = -mg\cos(\theta) + m(L\dot{\theta}^2) = 0 \tag{2}$$

Sum of the forces in the tangential direction.

$$\sum F_t = m(L\ddot{\theta}) - mg\sin(\theta) = 0 \tag{3}$$

With drag:

Sum of the forces in the normal direction,

$$\sum F_n = -mg\cos(\theta) + m(L\dot{\theta}^2) = 0 \tag{4}$$

Sum of the forces in the tangential direction.

$$\sum F_t = m(L\ddot{\theta}) - mg\sin(\theta) - F_{Drag} = 0 \tag{5}$$

Where:

 F_n : Denotes the forces acting in the normal direction.

 F_t : Denotes the forces acting in the tangential direction.

 θ : Position of the mass.

 θ : Angular velocity of the mass.

 $\ddot{\theta}$: Angular acceleration of the mass.

m, L, g: Are constants; mass, length of string, and gravity, respectively

5 Knowns and Unknowns

Knowns: Unknowns:

Mass: m=0.142 kg String Tension: T String Length: L=0.5 m Angular Acceleration: $\ddot{\theta}$

Gravity: $g = 9.81^{m/s^2}$

State Variables:

Angular Velocity: $\dot{\theta}$ Angular Acceleration: $\ddot{\theta}$

The angular velocity and position state variables are associated with the angular acceleration, so they are considered known since they can be obtained from the acceleration using integration. Therefore we have two equations of motion (2) & (3) and two unknowns (T and $\ddot{\theta}$).

6 Constraints

Because there are as many unknowns as there are equations, constraints are not needed in this problem.

7 Solve for the Equations of Motion

The equations of motion for a simple pendulum are (from Eq. (2) and (3))

$$\ddot{\theta} = -\frac{g}{L}\sin(\theta) \tag{6}$$

and

$$T = mg\cos(\theta) + mL\dot{\theta}^2 \tag{7}$$

Whereas with drag, the only difference is in Eq. (6); plugging in the drag force representation (from Eq. (1))

$$\ddot{\theta} = -\frac{g}{L}\sin(\theta) - \frac{1.65 \cdot 10^{-3}\dot{\theta}|\dot{\theta}|}{mL} \tag{8}$$

8 Solve the Equations of Motion

8.1 Without Drag

Taking Equation (6),

$$\ddot{\theta} = -\frac{g}{L}\sin(\theta)$$

We can integrate the differential equation using MATLAB and plot the behavior of θ versus time for various release angles. (Figure (1))

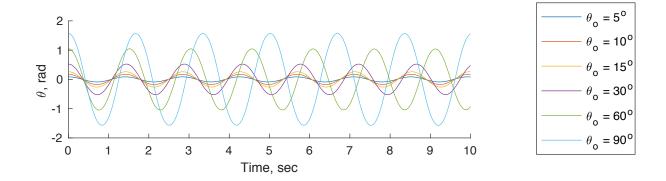


Figure 1: θ vs Time For Various Initial Release Angles

Using the small angle approximation $(\sin(\theta) \approx \theta)$, we can rewrite Equation (6) as

$$\ddot{\theta} = -\frac{g}{L}\theta$$

Letting $\theta(t) = A \cdot e^{st}$, then $\ddot{\theta} = s^2 \cdot A \cdot e^{st}$; solving the second order differential equation analytically yields

$$\theta(t) = \theta_o \cos(\sqrt{\frac{g}{L}}t)$$

Where

$$\sqrt{\frac{g}{L}} = \omega_n$$

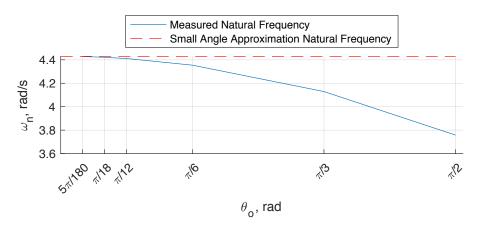


Figure 2: Natural Frequency (ω_n) vs Initial Angle of Release (θ_o)

As shown in Figure (2), using the small angle approximation has very little effect on the calculation of the natural frequency of the system, especially when θ_o is smaller in magnitude; later on when the small angle approximation is used to represent the system with drag incorporated, the initial angle of release, θ_o , is $\pi/12$. As seen in Figure (2), the value of ω_n at $\pi/12$ using the approximation is very close to the actual frequency of the system.

8.2 With Drag

Using Equation (8)

$$\ddot{\theta} = -\frac{g}{L}\sin(\theta) - \frac{1.65 \cdot 10^{-3}\dot{\theta}|\dot{\theta}|}{mL}$$

We can integrate and plot the differential equation over a 100 second period.

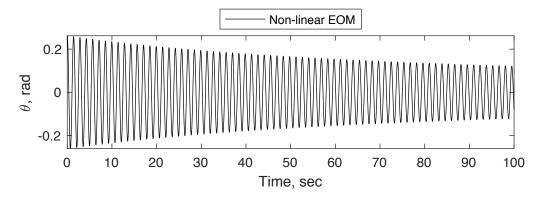


Figure 3: θ vs Time with Drag

We can see the amplitude of the pendulum oscillations decreasing over time. We then determine the second order differential equation that correctly models the system with drag to be.

$$\ddot{\theta} + 2\zeta\omega_n\dot{\theta} + \omega_n^2\theta = 0$$

Solving the second order differential equation we produce,

$$\theta(t) = e^{-\zeta \omega_n t} (A \cos(\omega_d t) + B \sin(\omega_d t)) \tag{9}$$

When $\theta(0)$ and $\theta(0)$ are evaluated they yield the equation

$$-\zeta \omega_n B + \omega_d A = 0 \quad \therefore \quad A = \frac{\zeta \omega_n B}{\omega_d}$$

Plugging A and B into (9), the equation can be rewritten as

$$\theta(t) = \frac{\pi}{12} e^{-\zeta \omega_n t} \left(\frac{\zeta \omega_n}{\omega_d} \sin(\omega_d t) + \cos(\omega_d t) \right)$$
 (10)

We then compare the plots of θ vs. Time from (Eq. 10) and our EOM.

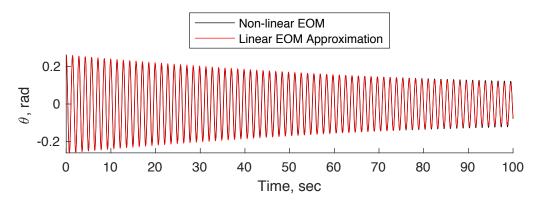


Figure 4: θ vs Time with Drag

Examining the plot more closely we can see that the Linear EOM Approximation is very close to the plot generated by the non-linear EOM. Plotting the last 5 seconds produces figure 5.

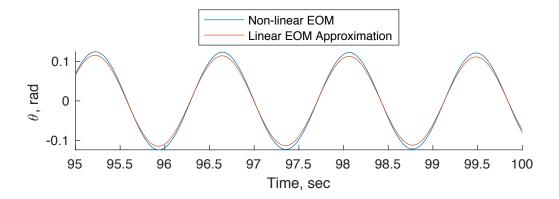


Figure 5: θ vs Time with Drag

9 Does it Make Sense?

9.1 Units

EOM without drag (from (6)):

$$\ddot{\theta} = -\frac{g}{L}\sin(\theta) = \frac{\frac{pr}{s^2}}{pr} \cdot rad = \frac{rad}{s^2}$$

Similarly, with the small angle approximation,

$$\ddot{\theta} = -\frac{g}{L}(\theta) = \frac{\frac{pr}{s^2}}{pr} \cdot rad = \frac{rad}{s^2}$$

Looking at the EOM with drag (from (8)), since the final units still need to come out to rad/s^2 , the drag term must have the units kg · m in order to cancel out:

$$\ddot{\theta} = -\frac{g}{L}\sin(\theta) - \frac{1.65 \cdot 10^{-3} \dot{\theta} |\dot{\theta}|}{mL} = \frac{\frac{\mathcal{M}}{s^2}}{\mathcal{M}} \cdot rad - \frac{\cancel{kg} \cdot \mathcal{M} \cdot rad/s^2}{\cancel{kg} \cdot \mathcal{M}} = \frac{rad}{s^2}$$

Checking with the MATLAB symbolic units tool (from Section 10.3):

```
1 % Checking EOM Units
2 u = symunit;
3 m = m*u.kg;
4 g = g*u.m/u.s^2;
5 L = L*u.m;
6 fdrag = fdrag*u.kg*u.m;
7 theta = 'theta';
8 thetadot = 'thetadot'/u.s;
9 thetaddot = 'thetaddot'/u.s^2;
10 T = T*u.N;
11 eqn = subs(eqn);
12 checkUnits(eqn)
13 EOM = subs(x.thetaddot);

unitCheck = struct with fields:
```

Compatible: [1 1]

9.2 Magnitude

Without Drag:

Looking at Figure (1), the amplitude of the oscillations show no deterioration, which is consistent for a system with no outside forces (such as drag). As the initial angle of realease (θ_o) and the period of oscillations increase, the natural frequency (ω_n) decreases, which makes physical sense. Also, as seen in Figure (2), the natural frequency of the system with small initial angles show that the small angle approximation $(\sin(\theta) \approx \theta)$ is valid, because they approach $\sqrt{\frac{g}{L}}$ $(\omega_n$ with $\sin(\theta) \approx \theta$).

With Drag:

Now, looking at Figure (3), we can see that the amplitude of the oscillations is slowly decreasing over time, showing a loss in energy due to the drag component included the system. The natural frequency (ω_n) and the damped natural frequency (ω_d) are very close in magnitude. This makes physical sense because the damping ratio (ζ) is very small, and will only affect the system in small increments.

Signs:

Given the coordinate system discussed in Section (5) and the equations of motion with and without drag, (6) & (8),

When $\theta_o > 0$, the initial angular acceleration $(\ddot{\theta_o})$ is negative, or swinging towards the position with the least energy $(\theta = 0)$.

When $\theta_o < 0$, the initial angular acceleration $(\ddot{\theta}_o)$ is positive, or swinging towards the position with the least energy $(\theta = 0)$.

Both of these situations make physical sense.

10 Appendix

10.1 Attributions

Jeffrey Chen Conceptualization, Solving Damped/Natural Frequencies

Thorne Wolfenbarger Solving Damped/Natural Frequencies, Write-up Trey Dufrene Free Body/Coordinate System, Step 9, Write-up

Joint Effort MATLAB Coding, Solving EOM's

10.2 Analytical Solution

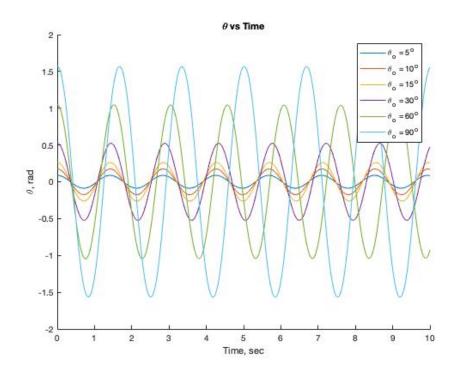
10.3 Numerical Solution

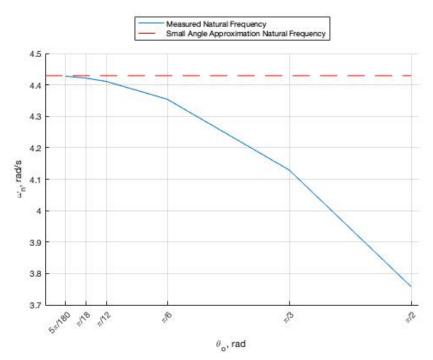
Without Drag

```
1 % Declaring Constants
2 \text{ c.g} = 9.81; \% \text{ ms/s}^2
3 \text{ c.m} = 0.142; \% \text{ kg}
4 \text{ c.L} = .5; \% \text{ m}
5
6 options = odeset('Events', @event);
7
8 % Declaring Variables
9 syms m g L theta thetadot thetaddot T
10
11 % Our Equations of Motion (EOM)
12 eqn(1) = m*(L*thetadot^2) == T - m*g*cos(theta);
13
14 eqn(2) = (thetaddot) == (-m*g*sin(theta))/(m*L);
15
16 % Solve our EOM and integrate it in ode45
17 x = solve(eqn,[T,thetaddot]);
18
19 syms theta(t) thetadot(t)
20 thetaEOM = subs(x.thetaddot,{'theta','thetadot'},...
21
                   {theta,thetadot});
22 eom = odeFunction([thetadot;thetaEOM],[theta;thetadot],g,L);
23
24 hold on;
25 % Plotting our Data for Theta Time History for System With
      Drag at
26 % different angles
27 nat freq = zeros(1,6);
28 releaseAngle = [5,10,15,30,60,90];
29 \text{ for } i = 1:6
       j = releaseAngle(i);
30
31
       [Time,S,TE,SE,IE] = ode45(Q(t,s)eom(t,s,c.g,c.L),linspace
          (0,10,1001),[(j*pi/180),0],options);
32
       plot(Time,S(:,1),'DisplayName', ['\theta o = ' num2str(j)
            '^o']);
       xlabel('Time, sec');
33
34
       ylabel('\theta, rad');
       nat freq(i) = 2*pi / (TE(2))
36 end
37 title('\theta vs Time');
38 legend('show')
39
40 %Graph of natural frequencies
41 figure (2)
42 hold on
```

```
43 plot(releaseAngle*pi/180,nat_freq, 'DisplayName', 'Measured
                    Natural Frequency')
44 line([0 90].*pi/180, [sqrt(c.g/c.L),sqrt(c.g/c.L)],'Color','
                    red','LineStyle','--',...
                         'DisplayName',['Small Angle Approximation Natural
45
                                 Frequency'])
46 xlabel('\theta_o, rad');
47 ylabel('\omega n, rad/s');
48 xticks([5*pi/180 pi/18 pi/12 pi/6 pi/3 pi/2])
49 xticklabels(\{ '5\pi/180', '\pi/18', '\pi/12', '\pi/6', '\pi/3', '\pi/8', '\pi/8
                    pi/2'})
50 xtickangle (45)
51 grid on
52 legend('show')
53 legend('location', 'northoutside')
54
55 set(gcf, 'PaperPositionMode', 'manual');
56 set(gcf, 'PaperUnits', 'inches');
57 set(gcf, 'PaperPosition', [1 1 6 2.5]);
58 \text{ fig = gcf};
59 print('BestFitFigure','-dpdf');
60
61
62 % Event function
63 function [value isterminal direction] = event(t,s)
                        value = s(2);
64
                        isterminal(1) = false;
65
                        direction(1) = -1;
66
67 end
          nat freq =
```

4.4278 4.4215 4.4110 4.3543 4.1292 3.7578





With Drag

```
1 % Declaring Constants
2 \text{ c.g} = 9.81; \% \text{ ms/s}^2
3 \text{ c.m} = 0.142; \% \text{ kg}
4 \text{ c.L} = .5; \% \text{ m}
5 \text{ c.fdrag} = 1.65*10^-3;
7 options = odeset('Events', @event);
8
9 % Declaring Variables
10 syms m g L fdrag theta thetadot thetaddot T
11 % Our Equations of Motion (EOM)
12 eqn(1) = m*(L*thetadot^2) == T - m*g*cos(theta);
13 eqn(2) = (thetaddot)*(m*L) == (-m*g*sin(theta)) - (fdrag)*
      thetadot*(abs(thetadot));
14
15 % Solve our EOM and integrate it in ode45
16 x = solve(eqn,[T,thetaddot]);
17
18 syms theta(t) thetadot(t)
19 thetaEOM = subs(x.thetaddot,{'theta','thetadot'},...
20
                   {theta,thetadot});
21 eom = odeFunction([thetadot; thetaEOM], [theta; thetadot],g,L,m,
      fdrag);
22
23 [Time,S,TE,SE,IE] = ode45(Q(t,s)eom(t,s,c.g,c.L,c.m,c.fdrag),
      linspace(0,100,10001),[(15*pi/180),0],options);
24
25 % Plotting our Data for Theta Time History for System With
      Drag
26 figure (1)
27 plot(Time, S(:,1), '-k')
28 xlabel('Time, sec')
29 ylabel('\theta, rad')
30 legend('Non-linear EOM', 'Linear EOM Approximation')
31 legend('location', 'northoutside')
32 legend('show')
33
34 %Calculating omega_d, zeta, and omega_n
35 mean period time = mean(diff(TE));
36 omega_d = (2*pi/mean_period_time);
37 decayRate = mean(log(SE(2:end,1)./(SE(1:end-1,1)))./(TE(2:end
      )-TE(1:end-1)));
38 zeta = [];
39 for i = 1:length(TE)
40 \text{ zeta(i)} = \dots
```

```
41 sqrt((log(SE(i,1)./(15*pi/180)).^2)./((log(SE(i,1)./(15*pi
      /180)).^2)+(2*pi*i)^2) ...
42);
43 end
44 zeta = mean(zeta);
45 omega_n = omega_d/sqrt(1-zeta^2);
46 % Theta in terms of omega_d, zeta, and omega_n
47 theta func = pi/12 .* exp(-zeta*omega n.*Time) .* (zeta*
      omega n/omega d .* sin(omega d.*Time)+cos(omega d.*Time));
48 figure (2)
49 % Plotting Time vs theta
50 xlabel('Time, sec')
51 ylabel('\theta, rad')
52 hold on
53 plot(Time, S(:,1), '-k')
54 plot(Time, theta func, '-r')
55 hold off
56 legend('Non-linear EOM', 'Linear EOM Approximation')
57 legend('location', 'northoutside')
58 legend('show')
59
60 figure (3)
61 % Plotting Time vs theta on a short duration
62 xlabel('Time, sec')
63 ylabel('\theta, rad')
64 hold on
65 plot(Time(9500:end),S(9500:end,1))
66 plot(Time(9500:end), theta func(9500:end))
67 xlim([95 100])
68 hold off
69 legend('Non-linear EOM', 'Linear EOM Approximation')
70 legend('location', 'northoutside')
71 legend('show')
72
73 % Checking EOM Units
74 u = symunit;
75 m = m*u.kg;
76 \text{ g = g*u.m/u.s}^2;
77 L = L*u.m;
78 fdrag = fdrag*u.kg*u.m;
79 theta = 'theta';
80 thetadot = 'thetadot'/u.s;
81 thetaddot = 'thetaddot'/u.s^2;
82 T = T*u.N;
83 eqn = subs(eqn);
84 unitCheck = checkUnits(eqn)
85 EOM = subs(x.thetaddot);
86
```

```
87 % Event function
88 function [value isterminal direction] = event(t,s)
89
       % value is a function that is zero at the event
90
       % isterminal is 1 if you desire to terminate integration
          at the event
91
                       O if you desire to continue integration
92
       \% direction defines the slope of the function value at
         the event
              1 for positive slope, -1 for negative slope, 0 for
93
           either
94
       value = s(2);
       isterminal = false;
95
       direction = -1;
96
97 end
```

unitCheck =
 struct with fields:

Consistent: [1 1]
Compatible: [1 1]

