## Ground Tracks

If we have a satellite in ECI coordinate frame, the Earth is spinning in the frame.

Can't get a groundtrack!

Instead we need Earth-centered Earth-fixed (ECEF) coordinates.

OERA:

Let Tu=

$$\Theta_{FPA} = 2 \Upsilon \left(0.7790572732640 + 1.00273781191135448 Tu\right)$$

To go from ECI to ECEF, we need to rotate about 2 by DERA

what is the rotation matrix?

Now we have a vector in ECEF!

Latitude: 0=

Longitude

What if you don't have FECT?

Need some spherical geometry

Latitude:

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We can either continuously find  $\theta_{ERA}$  or define  $\theta_{ERA}$  at Epoch and

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Wa = 7.2921151467 × 10-5 rad/s

## South East Zenith

While having a ground-track is important, it's just as useful to determine important, it's just as useful to determine where a satellite is located based on a ground station

("î") South You need to know has, las, das,

Need Fosic in SEZ.

To find the azimuth and elevation of slc wrt ground station,

Azimuth

Elevation:

To find velocity of slc wrt ground Station,

Vgs sic =

## Example:

Given the Moon's Position in ECI of  $\frac{ECI}{F_{ext}} = -76,290.6 \hat{x} - 375,924\hat{y}$ -126,3062 km

Find the latitude, longitude, azimuth and elevation.

Let DERA = 3.39.1° and a

Prescott ground Station with  $\phi_{gs} = 34.54^{\circ}, \Lambda_{gs} = -112.46^{\circ}, hgs = 1.64 km$ 

## 5. Find $\Theta_A$ and $\Theta_E$

O<sub>E</sub> =

Q =