

# **IB Physics Topic D1 Gravitational Fields; SL & HL**

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# 1 Newton's Law of Gravitation

Newton claimed that there is always an **attractive** force between two objects, and it is quantified by

$$F_G = \frac{Gm_1m_2}{r^2} \quad (1)$$

where

- $F_G$  is the force of gravity between the two objects,
- $G = 6.67 \times 10^{-11} \text{ N m}^2 \text{ kg}^{-2}$  is the gravitational constant,
- $m_1$  and  $m_2$  are the masses of the two objects,
- $r$  is the distance of separation.

Note that this is another instance of an **inverse square law**. Assuming masses constant, when the separation is scaled by a factor of  $k$ , the force is **divided by  $k^2$** . This is a key feature to remember and use in exam questions, especially ratio questions.

An inverse square function tends to zero but does not exactly become zero as  $r$  tends to infinity. This means that no matter how far away two objects are, there is always a gravitational force between them.

## 2 Gravitational Field Strength

From now on, we will proceed with the convention that  $M$  usually denotes the mass of the more massive object (e.g. the planet, the Sun) and  $m$  denotes the mass of the smaller object (e.g. the satellite, the Moon).

The formal definition of gravitational field strength is **the force per unit mass** experienced by a small test mass placed at a particular point in a field.

This is given by

$$g = \frac{F}{m}$$

In exam questions, the test mass in question is usually under the influence of more than one gravitational fields; in this case, there is a resultant gravitational field strength, which is the vector sum of the individual field strengths. Work on the vertical and horizontal components separately when the origins of the fields are not collinear with the test mass.

Using [Equation \(1\)](#), we can rewrite the gravitational field strength as

$$g = \frac{GM}{r^2}$$

where  $M$  is the mass of the object creating the field, e.g. the planet.

In the exam, this rewritten form is often more helpful as it does not require any knowledge of the mass of the object and computes the field strength directly from  $M$  and the separation distance  $r$ .

### 3 Field Lines Around the Earth

The field lines around the Earth are **radial**, pointing towards the center of the Earth. This is because the Earth is a sphere, and the field lines are always perpendicular to the surface of the object creating the field. The field lines are also evenly spaced, indicating that the field strength is constant at any point on the same equipotential surface. However, at the surface, the field lines appear almost **uniform**, this is because the Earth's radius is big enough that the field lines are almost parallel to each other observed from a single point on the surface.

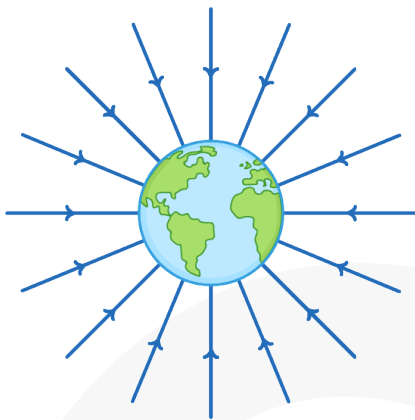


Figure 1: Radial field lines around the Earth

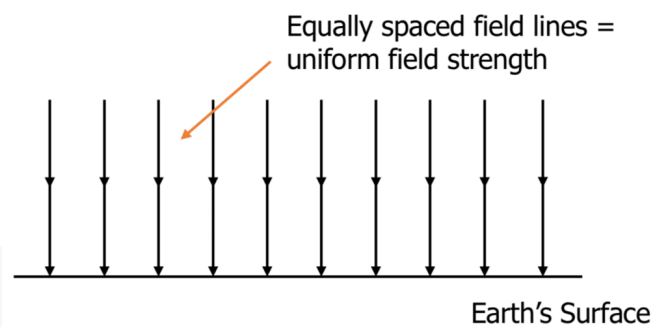


Figure 2: Uniform pattern approximation

Drawing field lines is generally trivial but nevertheless requires great care. Use a ruler.

- In a uniform field, all field lines must be parallel, equal in length, and evenly spaced. Often times, any number of lines greater than 5 should suffice.
- In a radial field, all field lines must point inwards to the center of the object, also angularly equidistant. Often times, 8 lines should suffice.

## 4 Extended Bodies and Point Masses

Extended bodies are the real-world objects; they can be treated as point masses (all mass of the object is condensed at its center of gravity) when and only when the following criteria are met:

- The field is uniform.
- Its size is small compared to the distances under study.

## 5 Kepler's Laws

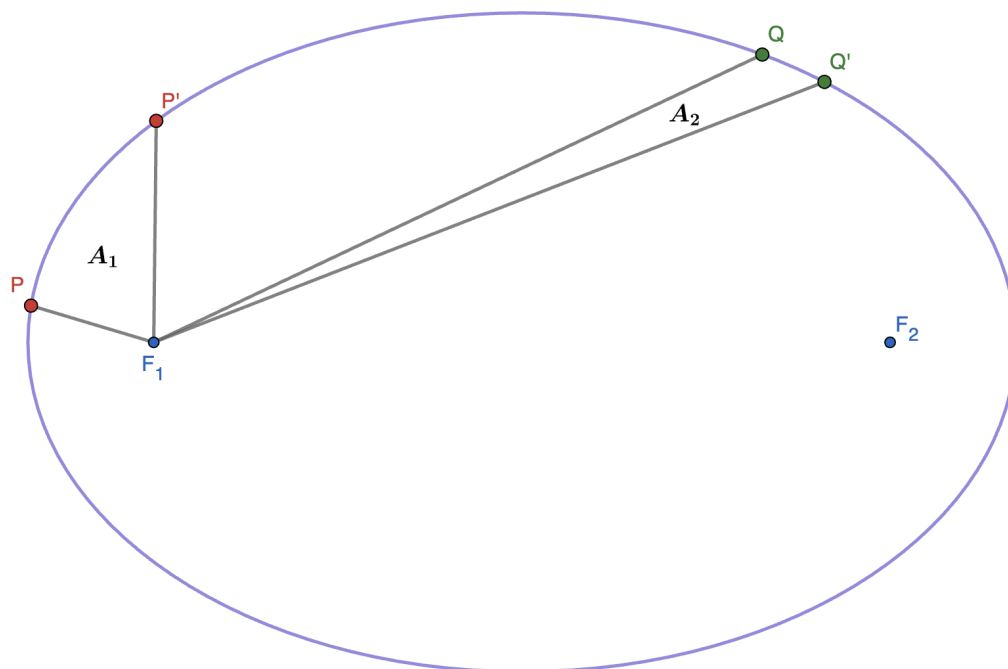


Figure 3: An elliptical orbit

- **Kepler's First Law:** The orbit of a planet is an ellipse with the Sun at one of the two foci.
  - The semi-major axis is the average distance of the planet from the Sun.
- **Kepler's Second Law:** A line segment joining a planet and the Sun sweeps out equal areas during equal intervals of time.
  - $A_1 = A_2$ , if the time intervals between  $PP'$  and  $QQ'$  are equal.
  - A consequence is that the average velocity of the planet between  $PP'$  is greater than that between  $QQ'$ . This is because at Q, the area is much sensitive to a change in the position of the planet, and so to keep the area swept out equal, the planet must move a lot slower at Q than at P.

The mark scheme style answer is given as



1. The areas swept out in unit time by the Sun-comet line are the same at P and Q
  2. At Q, the distance is greater hence the orbital speed/distance moved in unit time is lower so that the area remains the same
  3. A decrease in speed means that the kinetic energy also decreases
- **Kepler's Third Law:** The square of the orbital period of a planet is directly proportional to the cube of the semi-major axis of its orbit.

$$T^2 \propto r^3 \quad \text{or} \quad \left(\frac{T_1}{T_2}\right)^2 = \left(\frac{r_1}{r_2}\right)^3$$

- Directly use this relation in ratio questions; forget about the constant of proportionality.

## 5.1 Quantifying Kepler's Third Law

In exams, you are likely asked to derive this yourself.

1. Equate the gravitational force with the centripetal force.

$$\frac{GMm}{r^2} = m\omega^2 r$$

2. Make a substitution using the definition of angular velocity

$$\omega = \frac{2\pi}{T}$$

3. Rearrange to obtain

$$T^2 = \left(\frac{4\pi^2}{GM}\right) r^3 \quad (2)$$

which is Kepler's Third Law.

## 6 Gravitational Potential Energy

The IGCSE equation  $E = mgh$  does not work if the distance over which the change in potential energy occurs is large such that the gravitational field strength changes significantly.

The formal definition of GPE is **the work done in moving a mass from infinity to a point in a gravitational field**. Strictly speaking, GPE is a **property of the system** and not of the object itself, although it is the same for both objects and so is often treated as a property of an object in a field.

It is given by

$$U = -\frac{GMm}{r} = -mgr$$

where  $U$  is the gravitational potential energy,  $M$  is the mass of the object creating the field, and  $r$  is the distance of separation. The interpretation of the negative sign is as follows

- To separate the objects (thus increase  $r$ ), work must be done on the system to overcome the attractive force. This means that the change in GPE is positive, and the resulting GPE is “less negative”. In the case of a satellite, if the orbital radius is increased, the work done by the engine must be positive.
- To bring the objects closer together (thus decrease  $r$ ), energy is released by the system. This means that the change in GPE is negative, and the resulting GPE is “more negative”. In the case of a satellite, if the orbital radius is decreased, the work done by the engine must be negative.

## 7 Gravitational Potential

The gravitational potential is similar to the gravitational field strength in the sense that both quantities do not depend on the mass of the test object but only on the  $M$  and  $r$ . It is defined as **work done on unit mass** and is given by

$$V = -\frac{GM}{r} \quad \text{J kg}^{-1}$$

where  $V$  is the gravitational potential. The negative sign is interpreted in the same way as in the GPE equation. It can be obtained in the following ways, depending on what you are given in the question

- If you are given the GPE, divide by the mass of the object  $m$  to obtain the potential.

$$V = \frac{U}{m} = \frac{-\frac{GMm}{r}}{m} = -\frac{GM}{r}$$

- If you are given the field strength, multiply by the distance  $r$  to obtain the potential.

$$V = -g \cdot r = -\frac{GM}{r^2} \cdot r = -\frac{GM}{r}$$

### 7.1 Change in Gravitational Potential Energy

The change in GPE is given by

$$\Delta V_g = \frac{W}{m} \quad (3)$$

- $\Delta V_g$  is the change in gravitational potential energy,
- $W$  is the work done in moving the mass  $m$  from one point to another
- $m$  is the mass of the object

This equation uses the assumption that the change happened at a constant speed so that no work done is distributed to KE.

## 7.2 Equipotential Surfaces

These are imaginary circular surfaces around a point mass; each ring represents the locus of points where the gravitational potential is the same. An important concept is that **no work is done in moving an object along an equipotential surface**, as the potential energy is constant.

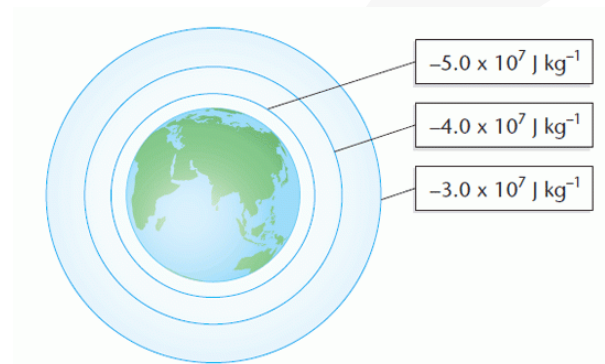


Figure 4: Equipotential surfaces around the Earth

A quick observation to make is that the separation between the rings occupying potentials at fixed intervals (-3, -4, -5 in the diagram) is not constant, this is because  $V_g \propto \frac{1}{r}$  is an inversely proportional relation.

The field lines and field strengths are both always normal to the equipotential surfaces.

## 7.3 Linking to Gravitational Field Strength

We have previously mentioned the relation

$$V = -g \times r$$

we can now write this in a differential form as follows

$$g = -\frac{dV}{dr}$$

This is the tangential gradient of the potential/distance graph at a point.

## 8 Orbital Motion Equations

The first equation is the orbital speed — the linear velocity at which a planet circles the Sun; again, it is derived by equating the gravitational force with the centripetal force.

$$\begin{aligned}\frac{GMm}{r^2} &= \frac{mv^2}{r} \\ v^2 &= \frac{GM}{r} \\ v_{\text{orbital}} &= \sqrt{\frac{GM}{r}}\end{aligned}\tag{4}$$

We can then convert this to angular velocity using  $\omega = \frac{v}{r}$  to obtain

$$\omega_{\text{orbital}} = \sqrt{\frac{GM}{r^3}}\tag{5}$$

The time period  $T$  can be found using the previously derived Kepler's Third Law, [Equation \(2\)](#).

## 9 Escape Velocity

To escape from the gravitational field of a planet, work must be done to take the object from the surface to infinity. This work is in fact the initial kinetic energy upon launch and equal to the GPE at the surface. Thus

$$KE + GPE = 0$$

Substitution and rearrangement gives

$$v_{\text{escape}} = \sqrt{\frac{2GM}{r}} = \sqrt{2gr} = \sqrt{-2V_{\text{surface}}} \quad (6)$$

Observations:

- Independent of the object's mass
- $v_{\text{escape}} = v_{\text{orbital}}\sqrt{2} > v_{\text{orbital}}$

In fact, this works for an object at any orbital height, not just the surface. For an object above the surface, the orbital speed must be increased to  $\sqrt{2}v_{\text{orbital}}$  to escape, analogously, the kinetic energy must be raised to twice the original (since velocity is squared when computing KE, the factor of  $\sqrt{2}$  is squared to give 2).

## 10 Total Energy of a Satellite

The total energy of a satellite in orbit is given by the sum of the kinetic and potential energies.

$$\Sigma E = KE + GPE = \frac{1}{2}mv^2 - \frac{GMm}{r}$$

Knowing the orbital speed formula we can rewrite KE:

$$KE = \frac{1}{2}m \left( \sqrt{\frac{GM}{r}} \right)^2 = \frac{1}{2} \times \frac{GMm}{r} = \frac{1}{2}GPE$$

This means that the KE is always half the magnitude of GPE, assuming that the satellite is in a circular orbit at a constant speed. We can make a substitution back to the expression for  $\Sigma E$  to obtain

$$\Sigma E = \frac{1}{2}GPE = -\frac{1}{2} \frac{GMm}{r} = -KE \quad (7)$$

Let's now consider what happens when the satellite moves to a different orbital height. Consider moving the satellite to a height closer to the planet.

- Suppose the kinetic energy is increased by  $\Delta E > 0$ . It is positive because moving it closer increases the centripetal force and thus angular speed.
- Then, the GPE must be made more negative by  $2\Delta E$ , hence giving a GPE change of  $-2\Delta E$ .
- The net change is  $\Delta E - 2\Delta E = -\Delta E < 0$ .

Hence, when the system loses (becoming more negative)  $\Delta E$  joules of energy, the KE must have increased by  $\Delta E$  joules, and the GPE must have decreased by  $2\Delta E$  joules.



## 10.1 Impact of Viscous Drag

The syllabus also requires us to consider the impact of viscous drag on the satellite.

At low heights, the satellite collides with air molecules in the atmosphere, thus experiencing a drag force opposing the circular motion.

1. The total energy of the satellite is decreased as a portion is lost to friction
2. The satellite will undergo **orbital decay**, where the altitude and analogously the orbital radius of the satellite decreases, leading to an **increase in the kinetic energy and hence the speed**.
3. The decrease in potential energy must be larger than increase in kinetic energy for total energy to decrease.

The satellite will eventually burn up in the atmosphere. This is why it requires a boost in altitude every few years to maintain its orbit.

## 11 Exam Tips and Notes

### 11.1 The Radius

In questions, always bear in mind that the radius is the distance from the center of the planet (or any other object creating the field) to the satellite, **not the distance from the surface of the planet to the satellite**. Consider the following

A satellite is in an orbit at a height  $h$  above the surface of Earth. The gravitational potential at its location is  $V$ .

The satellite is transferred to a different orbit where the gravitational potential is  $\frac{V}{2}$ . What can be deduced about the new height of this satellite from the surface of Earth?

One may easily fall into the pitfall of thinking that, since  $V = -\frac{GM}{r}$ , if  $V$  is halved, then  $r$  is doubled, and so the new height is  $2h$ . However, this is incorrect; the radius is the distance from the center of the Earth to the satellite! Let  $R$  be the radius of the Earth; the old **orbital radius** is  $R + h$  and the new is  $2(R + h)$ , so the new height is  $2R + 2h - R = R + 2h > 2h$ .

### 11.2 Some Geography

Consider the axis line through the two poles of the Earth. This is the axis of rotation of the Earth. The equator is a circle of latitude that divides the Earth.

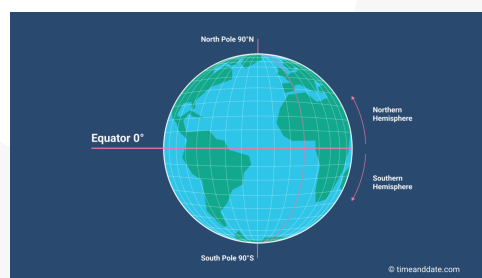


Figure 5: Geography

## 12 Exam Questions

### 12.1 Finding the Esc. Velocity with V

A spherical planet has a radius  $R_0$ . The graph shows the variation of the gravitational potential due to the planet with distance  $r$  from the center of the planet.

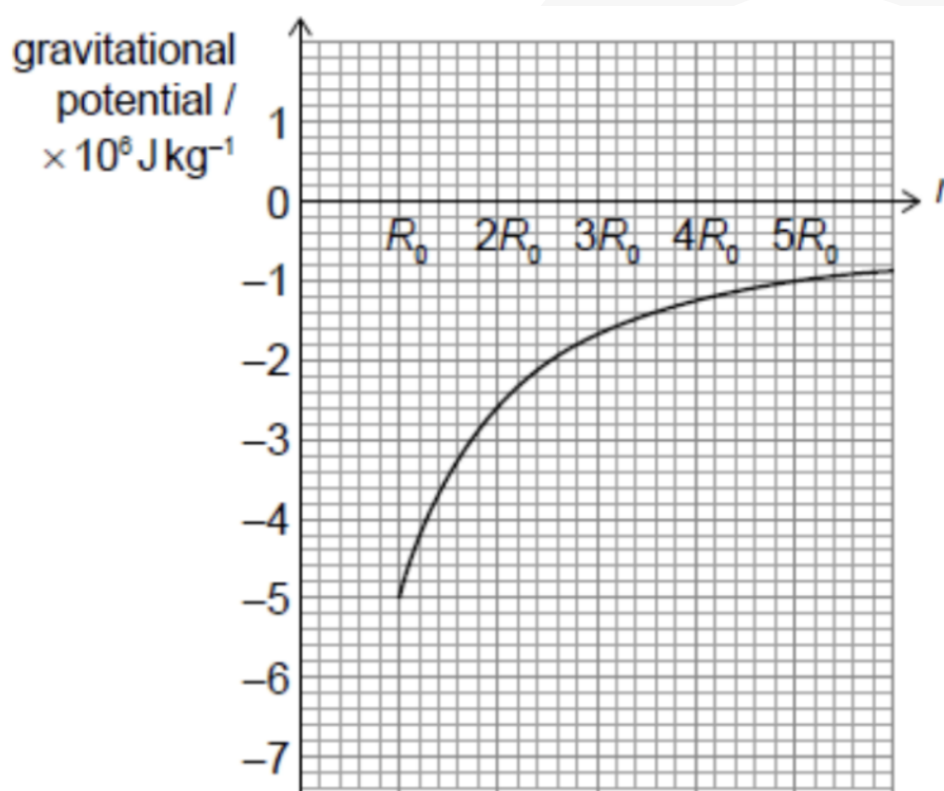


Figure 6: Graph of gravitational potential

What is the escape speed from the surface of the planet?

$$\begin{aligned}
 V_{\text{escape}} &= \sqrt{-2V_{\text{surface}}} \\
 &= \sqrt{2 \times 5 \times 10^6} \\
 &= 3.2 \times 10^3 \text{ m s}^{-1}
 \end{aligned}$$

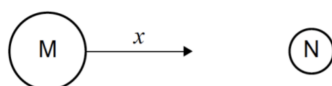
**12.2 Quick-Fire MCQ # 1**

A space probe moves in a circular orbit around Earth. The kinetic energy of the probe is  $E$ . The probe will reach the escape speed when its **kinetic** energy is increased at least to?

- We know the escape speed is  $\sqrt{2} \times$  orbital speed
- $E \propto v^2$
- $v \rightarrow \sqrt{2}v \implies E \rightarrow 2E$

## 12.3 Field Strength – Two Mass System

Two spheres M and N are far from any other mass. The mass of sphere M is greater than the mass of sphere N. A distance is measured from the surface of M along the line between the centres of M and N. The variation with of the gravitational field strength  $g$  is determined.



What is the variation of  $g$  with  $x$ ?

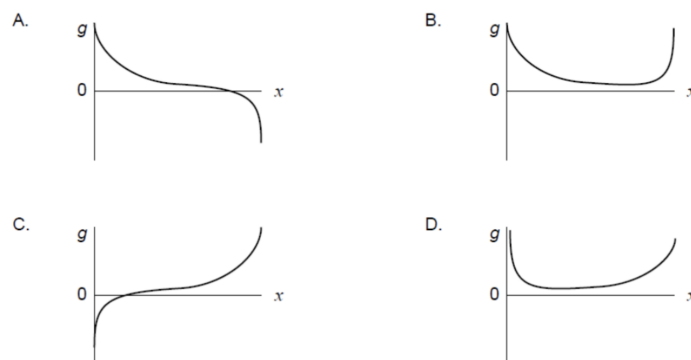


Figure 7: Field strength graph

- The key is to recognize that the question is talking about the resultant field strength, e.g. the field strength at N in M's field plus the field strength at M in N's field.
- Although not worded clearly, it is asking us to consider a point moving between the surfaces of M and N respectively.
- At  $x = 0$ , the point is at the surface of M.
- Hence, there is a point somewhere in the middle at which the two field strengths are equal (hence a root on the graph). We can already **eliminate B and D**.
- The idea is that, at this point of equilibrium, the point should be further away from M than N to balance out the greater mass of M. Hence, **A is the answer**.

## 12.4 Miscellaneous #1

At a distance  $r$  from the centre of a planet, the gravitational potential due to the planet is  $V$  and the gravitational field strength is  $g$ .

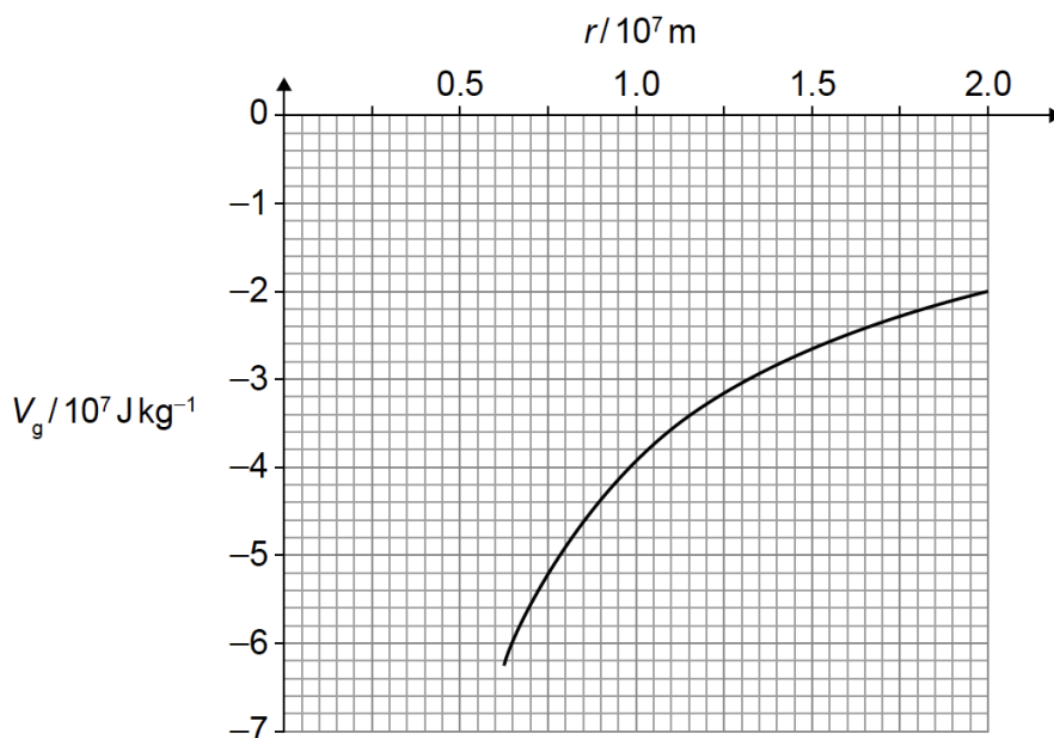


Figure 8: Graph of potential

(a) Show that  $V_g = -gr$

$$V = -\frac{GM}{r}$$

$$g = \frac{GM}{r^2}$$

$$V_g = -\left(\frac{GM}{r^2}\right)r = -gr$$

(b) A satellite of mass 750 kg orbits the Earth in a circular orbit of radius  $1.0 \times 10^7 \text{ m}$ . Calculate the magnitude of

(i) the gravitational field strength at the orbital radius of the satellite.

- We should be looking at the coordinate  $r = 1.0 \times 10^7 \text{ m}$  on the graph. This represents the position that is at the orbital radius of the planet.
- The corresponding field strength is

$$g = \frac{V_g}{r} = \frac{4.0 \times 10^7}{1.0 \times 10^7} = 4 \text{ m s}^{-2}$$

(ii) the gravitational force acting on the satellite.

$$\begin{aligned} F &= ma = mg \\ &= 750 \times 4 = 3000 \text{ N} \end{aligned}$$

The satellite is moved to a new circular orbit. The orbital radius of the satellite

The satellite is moved to a new circular orbit. The orbital radius of the satellite increases by 2.0 km.

- (c) (i) Outline why the magnitude of the gravitational field strength at the new orbit is approximately the same as at the old orbit.
- The change in the orbital radius is negligible compared to the original radius.
  - The gravitational strength is defined as  $g = \frac{GM}{r^2}$ , and since the change in  $r$  is small, the change in  $g$  is negligible.
- (ii) Determine the change in the total orbital energy of the satellite.
- Recall that the total orbital energy is given as

$$\Delta \Sigma E = \frac{1}{2} mg \Delta r = \frac{1}{2} (750)(4)(2.0 \times 10^3) = 3.0 \times 10^6 \text{ J}$$

## 12.5 Miscellaneous #2

A uniform, spherical planet P has mass  $M$  and radius  $R$ .

- (a) A projectile of mass  $m$  is launched with kinetic energy  $\frac{2GMm}{3R}$

**diagram not to scale**

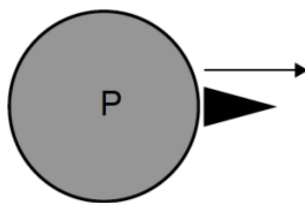


Figure 9: Diagram

Determine whether the projectile will escape the gravitational field of P.

$$\begin{aligned}\Sigma E &= E_K - E_P = \frac{2GMm}{3R} - \frac{GMm}{R} \\ &= \frac{2GMm}{3R} - \frac{3GMm}{3R} = -\frac{GMm}{3R} \\ \Sigma E &< 0\end{aligned}$$

Hence, the projectile will not escape the gravitational field of P.

- (b) The surface of P is an equipotential surface  $E_0$ . The dotted circle labeled  $E_1$  is an equipotential line at a distance  $\frac{3R}{2}$  from the center of P. The potential difference between  $E_1$  and  $E_0$  is  $2.2 \times 10^6 \text{ J kg}^{-1}$ .



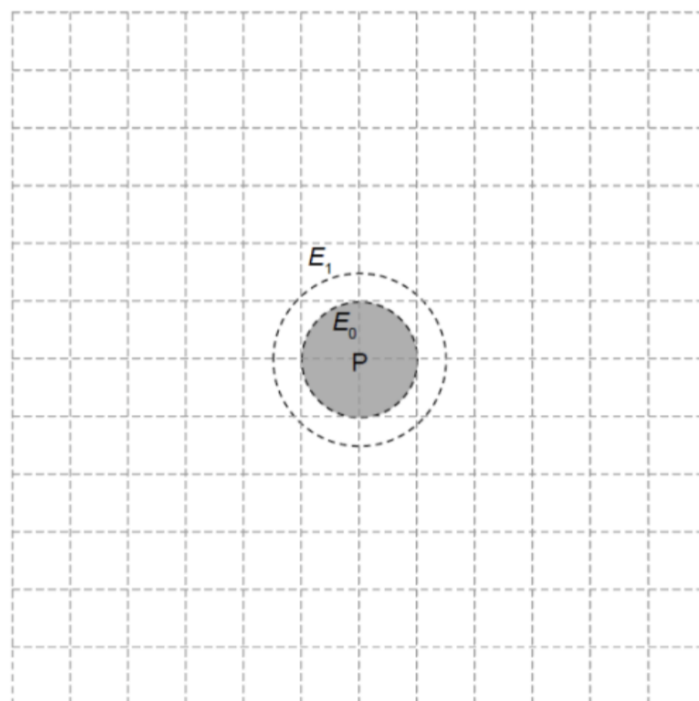


Figure 10: Diagram

- (i) The radius  $R$  of  $P$  is  $8.5 \times 10^6$  m. Estimate the magnitude of the gravitational field strength between  $E_1$  and  $E_0$ .

- The command word estimate is a hint that we have to read off the diagram.
- We are told that the radius of  $P$ , the equivalent of one square, is  $8.5 \times 10^6$  m.
- To find the field strength difference between two points, we require the formula

$$\Delta g = \frac{\Delta V}{\Delta r} = \frac{2.2 \times 10^6}{\frac{1}{2}(8.5 \times 10^6)} = 0.52 \text{ m s}^{-2}$$

- We used  $\Delta r = \frac{1}{2}(8.5 \times 10^6)$  because the distance between the two equipotential surfaces is half a square.
- (ii)  $E_2$  is an equipotential line such that the potential difference between  $E_2$  and  $E_1$  is also  $2.2 \times 10^6 \text{ J kg}^{-1}$ . Draw the equipotential line  $E_2$  on the diagram.

- To plot the equipotential ring, we need to know its radius. Let this quantity be  $x$
- From the given information, we know that

$$\begin{aligned} V_2 - V_1 &= V_1 - V_0 \\ -\frac{GM}{x} + \frac{GM}{1.5R} &= -\frac{GM}{1.5R} + \frac{GM}{R} \\ x &= 3R \end{aligned}$$

- Hence

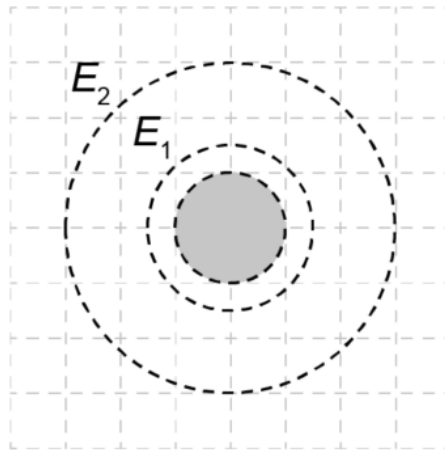


Figure 11: Solution

**12.6 Quick-Fire MCQ #2**

The escape speed from the surface of earth is  $v_{\text{esc}}$ . The radius of earth is  $R$ . A satellite of mass  $m$  is in orbit at a height  $\frac{R}{4}$  above the surface of the Earth. What is the energy required to move the satellite to infinity?

A.  $\frac{mv_{\text{esc}}^2}{5}$

B.  $\frac{2mv_{\text{esc}}^2}{5}$

C.  $mv_{\text{esc}}^2$

D.  $2mv_{\text{esc}}^2$

- The key is to recognize that this energy is the **total energy** of the satellite, namely KE + GPE, which is given by  $\Sigma E = \frac{1}{2}GMm \left( \frac{4}{5R} \right)$ .
- We know  $v_{\text{esc}}^2 = \frac{2GM}{R}$ .
- Hence  $\Sigma E = \frac{1}{5}mv_{\text{esc}}^2$  by substitution.

### 12.7 Miscellaneous #3

A moon M orbits a planet P. The gravitational field strength at the surface of P due to P is  $g_P$ . The gravitational field strength at the surface of M due to M is  $g_M$ .

For M and P:  $\frac{r_M}{r_P} = 0.27$  and  $\frac{m_M}{m_P} = 0.055$ .

- (a) Determine  $\frac{g_M}{g_P}$

$$g \propto \frac{m}{r^2}$$

$$\frac{g_M}{g_P} = \frac{m_M}{m_P} \left( \frac{r_P}{r_M} \right)^2 = 0.055 \left( \frac{1}{0.27} \right)^2 = 0.75$$

- (b) Point O lies on the line joining the centre of M to the centre of P.

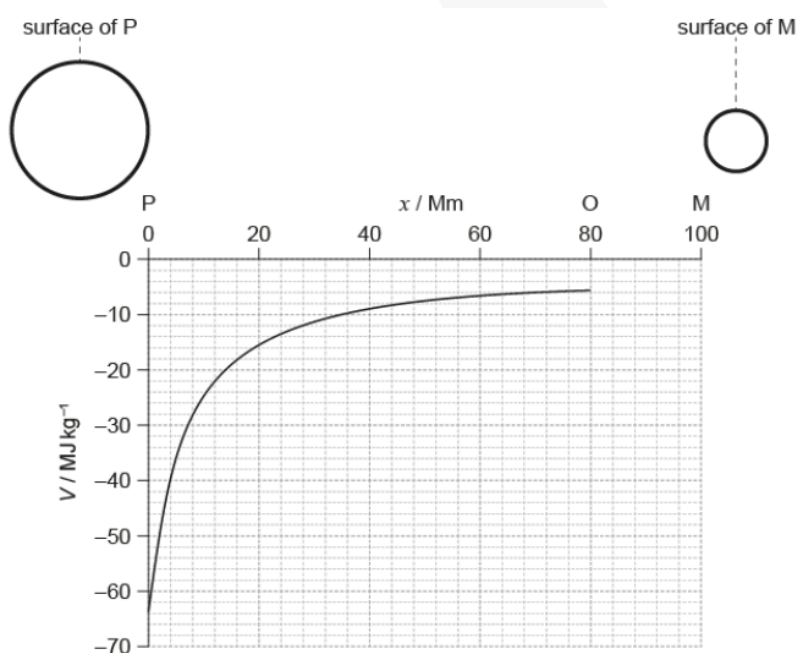


Figure 12: Graph

The graph shows the variation of gravitational potential  $V$  with distance  $x$  from the surface of P to O. The gradient of the graph is zero at point O.

- (i) State and explain the magnitude of the resultant gravitational field strength at O.

$$g = -\frac{dV_g}{dr} = -\frac{dV}{dx} = \text{gradient of graph at O} = 0$$

- (ii) Outline why the graph between P and O is negative.

- The gravitational field is attractive so that energy is required to move away from P
- the gravitational potential is defined as 0 at  $\infty$ , (the potential must be negative)

- (iii) The gravitational potential due to the mass of M at the surface of P can be assumed to be negligible.

Estimate, using the graph, the gravitational potential at the surface of M due to the mass of M.

$$\begin{aligned}\frac{V_M}{V_P} &= \left(\frac{g_M}{g_P}\right) \left(\frac{r_M}{r_P}\right) \\ V_M &= V_P(0.75)(0.27) = -6.4 \times 10^7 \times 0.75 \times 0.27 \\ &= -1.3 \times 10^7 \text{ J kg}^{-1}\end{aligned}$$

- (iv) Draw on the axes, the variation of gravitational potential between O and M.

- This question is worth one mark, it is awarded for the curve segment being
  - always negative
  - curving downwards
  - ending at  $y$ -coordinate between -8 and -20

## 12.8 Quick-Fire MCQ – Comparing Total Energies

Two satellites are in circular orbits around the Earth. Both satellites have the same mass and satellite X is closer to Earth than satellite Y. What is correct for the orbital periods of X and Y and the total energies of X and Y?

	Orbital periods	Total energies
A.	$X$ greater than $Y$	$X$ greater than $Y$
B.	$X$ greater than $Y$	$Y$ greater than $X$
C.	$Y$ greater than $X$	$X$ greater than $Y$
D.	$Y$ greater than $X$	$Y$ greater than $X$

1. The orbital period is easy to compare, the closer to the Earth, the shorter the period. This helps to **eliminate A and B**.
2. The total energy is given by

$$\Sigma E = -\frac{1}{2} \frac{GMm}{r}$$

- Since  $r_X < r_Y$ , the total energy of  $X$  is more negative than that of  $Y$  and so is less than  $Y$ .
- This takes us to the answer **D**.

## 12.9 Weightlessness

An astronaut is orbiting Earth in a spaceship. Why does the astronaut experience weightlessness?

The acceleration of the astronaut is the same as the acceleration of the spaceship.

## 12.10 Work Done by Gravity

P and S are two points on a gravitational equipotential surface around a planet. Q and R are two points on a different gravitational equipotential surface at a greater distance from the planet.

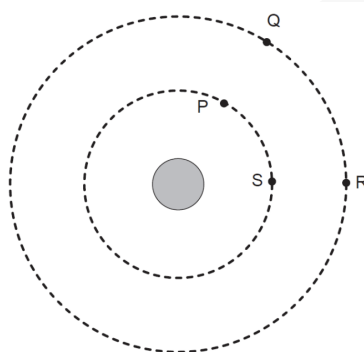


Figure 13: Diagram

The greatest work done by the gravitational force is when moving a mass from

- A. P to S
- B. Q to R
- C. R to P
- D. S to R

- Before anything one must realize the keyword “by the gravitational force” means that the work done is
  - Negative when moving away from the planet
  - Positive when moving towards the planet
- Firstly, we can **eliminate A and B**, as the work done is 0 as we move along the equipotential surfaces.
- The **answer is C**, as we are moving towards the planet.



## 12.11 Work Done by Engine

A satellite in a circular orbit around the Earth needs to reduce its orbital radius. What is the work done by the satellite rocket engine and the change in kinetic energy resulting from this shift in orbital height?

	Work done by the satellite rocket engine	Kinetic energy
A.	positive	increase
B.	positive	decrease
C.	negative	increase
D.	negative	decrease

- Here, we are moving towards the planet and so the as opposed to moving away, it is “easier” for the engine because we go “with” the gravitational force instead of “against” it. This eliminates A and B.
- Recall that

$$KE = -\frac{1}{2}GPE = \frac{GMm}{2r}$$

With a smaller orbital radius, the KE will increase, and so the answer is C.

## 12.12 Separation-Potential Graph – Turning Point

The graph shows the variation of the gravitational potential between the Earth and Moon with distance from the centre of the Earth. The distance from the Earth is expressed as a fraction of the total distance between the centre of the Earth and the centre of the Moon.

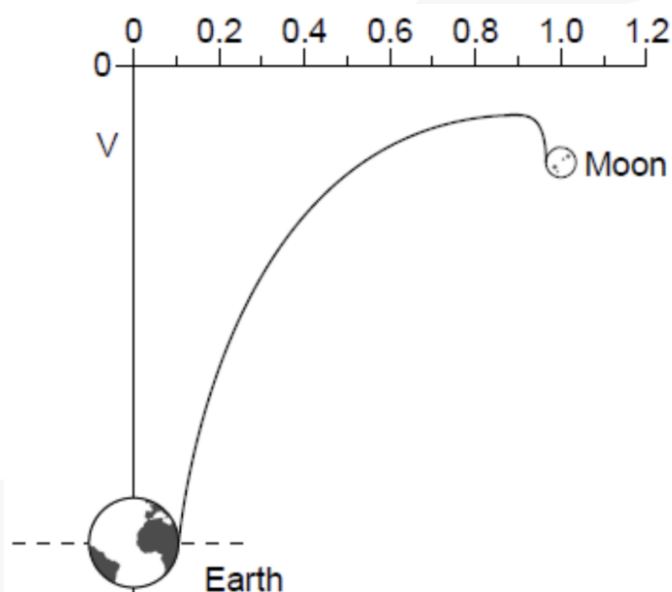


Figure 14: Graph

Given that the mass of the Earth is  $5.97 \times 10^{24}$  kg, determine, using the graph, the mass of the Moon.

- The method is to use the fact that the gravitational field strength at the turning point (which is located at  $x = 0.9$ ) is 0, hence the gravitational field strength due to the Earth and the Moon respectively are equal and opposite at this point.

$$\begin{aligned}
 g_E &= g_M \\
 \frac{M_E}{0.9^2} &= \frac{M_M}{(1 - 0.9)^2} \\
 M_M &= \frac{M_E}{0.9^2} \times 0.1^2 \\
 &= \frac{M_E}{81} = 7.4 \times 10^{22} \text{ kg}
 \end{aligned}$$