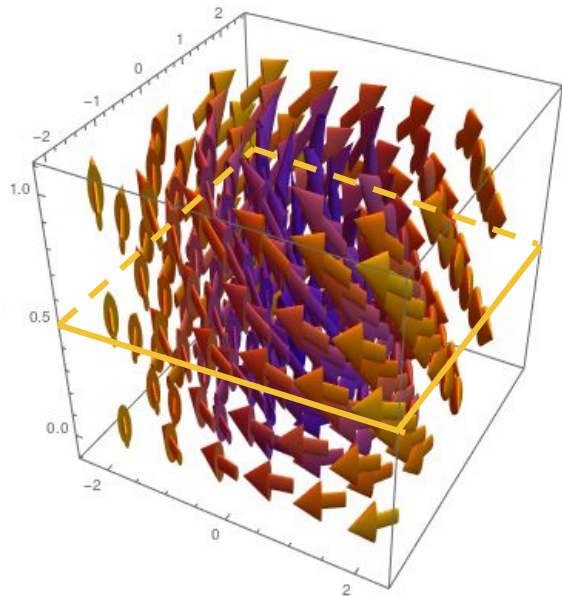
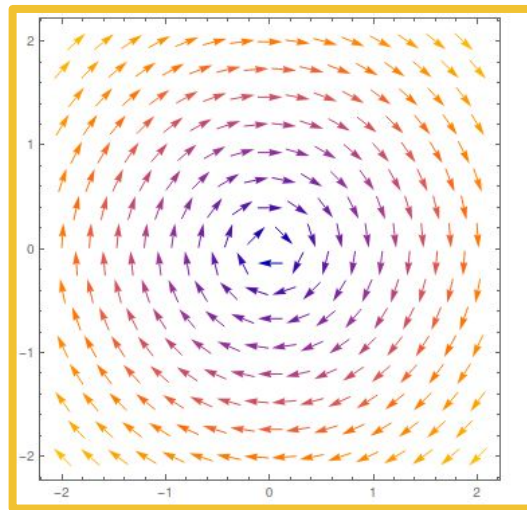


## Streamlines: Example

$$\vec{v}(\vec{x}) = \begin{bmatrix} y \\ -x \\ 0.5z \end{bmatrix}$$



$$\vec{v}(\vec{x}) = \begin{bmatrix} y \\ -x \\ 0 \end{bmatrix}$$



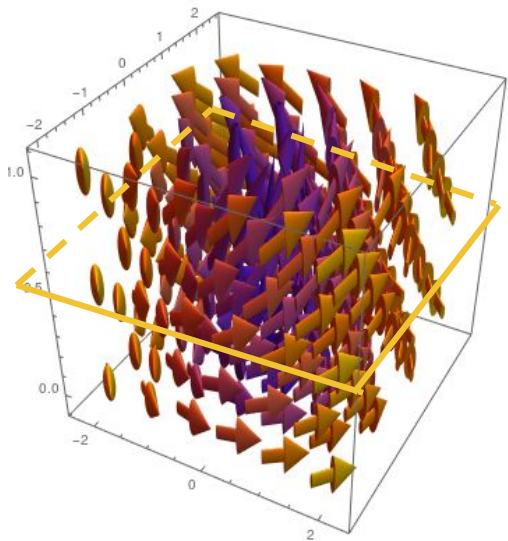
$$\frac{dy}{dx} = \frac{v}{u}$$

$$u \, dy = v \, dx$$
$$y \, dy = -x \, dx$$

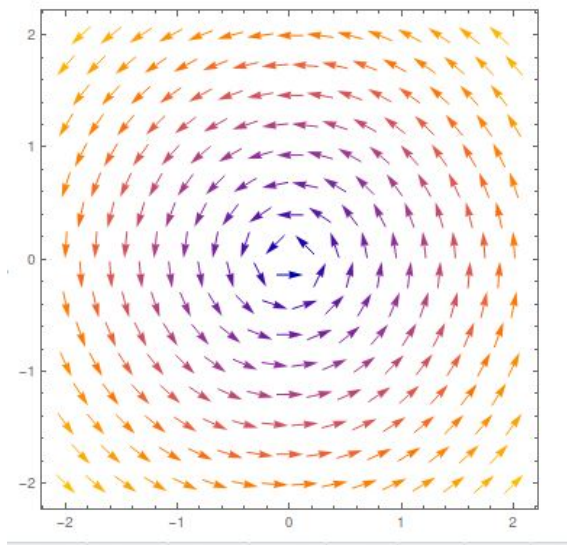
$$y^2 + x^2 = c$$

## Streamlines: Example

$$\vec{v}(\vec{x}) = \begin{bmatrix} -y \\ +x \\ 0.5z \end{bmatrix}$$



$$\vec{v}(\vec{x}) = \begin{bmatrix} -y \\ +x \\ 0 \end{bmatrix}$$



$$\frac{dy}{dx} = \frac{v}{u}$$

$$u \, dy = v \, dx$$
$$y \, dy = -x \, dx$$

$$y^2 + x^2 = c$$

## Streamlines: Example

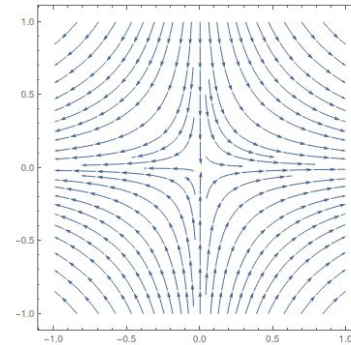
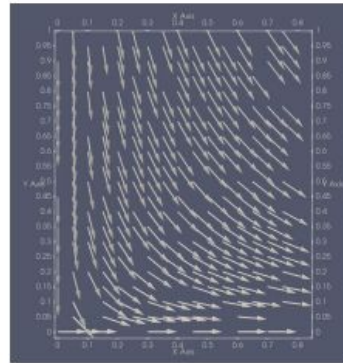
$$\frac{dy}{dx} = \frac{v}{u}$$
$$u \, dy = v \, dx$$

Streamlines are tangent to the velocity at every point in the flow-field

$$\vec{v} = a x \hat{e}_x - a y \hat{e}_y$$

$$\frac{dy}{dx} = \frac{v}{u} = -\frac{y}{x}$$

$$xy = k$$



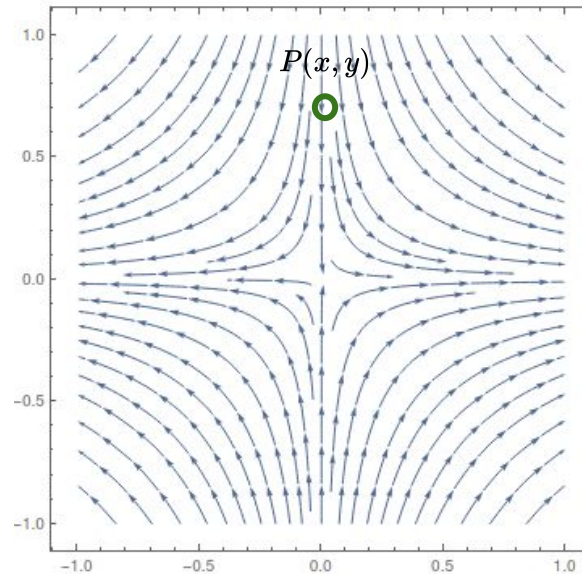
## Streamlines:Example

$$\frac{dy}{dx} = \frac{v}{u}$$

$$u dy = v dx$$

$$\vec{v} = a x \hat{e}_x - a y \hat{e}_y$$

Streamlines are tangent to the velocity at every point in the flow-field



Note:

1. Along the y-axis the fluid particle decelerate as it approaches the origin
2. The origin is the stagnation point
3. Track fluid-particles (a dyed fluid parcel)-Lagrangian approach
4. Measure velocities at a point (x,y,z;t)-Eulerian approach

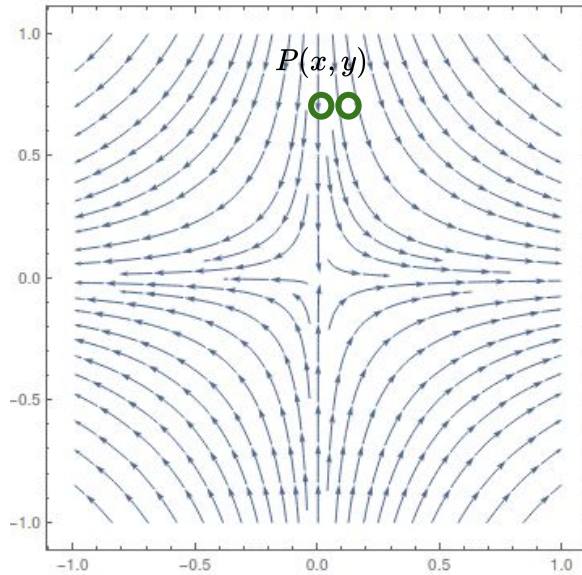
## Pathline & Streamline

$$\vec{v} = a x \hat{e}_x - a y \hat{e}_y$$

$$a = 0.3 \text{ s}^{-1}$$

Streamlines

$$\frac{dy}{dx} = \frac{v}{u}$$
$$u dy = v dx$$



Determine the velocity of fluid particle at point (2,8)

$$\vec{v} = 0.6 \hat{e}_x - 2.4 \hat{e}_y$$

If the particle is passing through point (2,8) at  $t = 0$ , determine the location of the particle at  $t = 6$  s.

$$v_x = ax \quad v_y = -ay$$

$$dx/dt = ax \quad dy/dt = -ay$$

$$x = x_o \exp(at) \quad y = y_o \exp(-at)$$

$$x = 12.1 \text{ m} \quad y = 1.32 \text{ m}$$

If the particle is passing through point (0,8) at  $t = 0$ , determine the location of the particle at  $t = 6$  s.

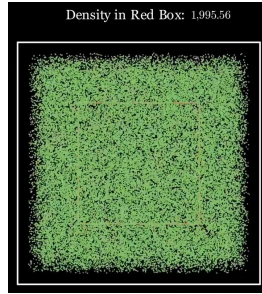
$$x = 0 \quad y = 1.32 \text{ m} \quad v_y = -0.3 \times 1.32 \approx -0.4 \text{ m/s}$$

If the particle is passing through point (0,8) at  $t = 0$ , determine the location of the particle at  $t = 60$  s.

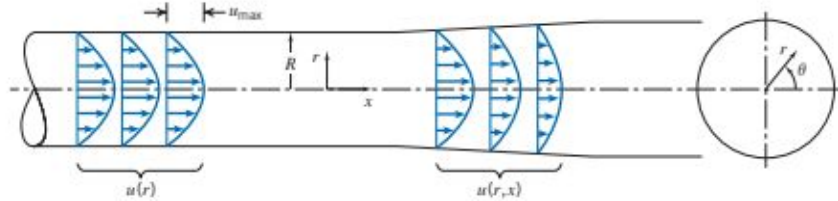
$$x = 0 \quad y = 1.2 \times 10^{-7} \text{ m} \quad v_y = -0.3 \times 1.2 \times 10^{-7} \text{ m/s}$$

## Point Properties

Density  $\rho = \frac{\Delta m}{\Delta V}$



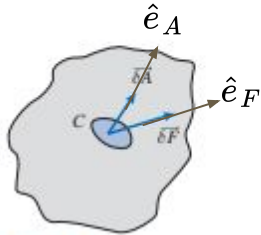
Velocity



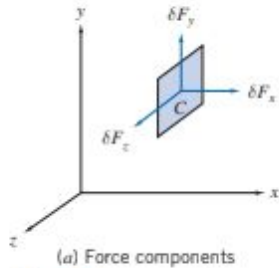
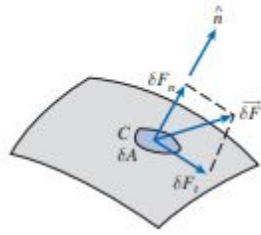
Pressure (Stress)

# Point Properties: Example

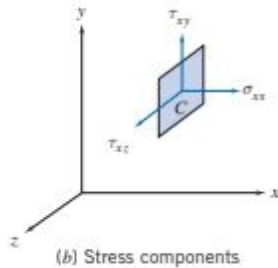
$$\bar{\sigma} = \lim_{\delta A \rightarrow 0} \frac{\delta F}{\delta A} \hat{e}_F \hat{e}_A$$



**Fig. 2.6** The concept of stress in a continuum.



(a) Force components



(b) Stress components

**Fig. 2.7** Force and stress components on the element of area  $\delta A_x$ .

1. Surface forces on a fluid particle leads to stress
2. The concept of stress is useful for describing how the forces acting on the boundaries of a control volume is transmitted through the volume
3. Stress is developed if there is a relative motion between fluid particles
4. Outward normal to the surface is considered to be positive (convention followed)

$$\begin{bmatrix} \sigma_{xx} & \sigma_{xy} & \sigma_{xz} \\ \sigma_{yx} & \sigma_{yy} & \sigma_{yz} \\ \sigma_{zx} & \sigma_{zy} & \sigma_{zz} \end{bmatrix}$$

1. 9 components
2. Symmetric
3. 6 independent components

# Point Properties: Example

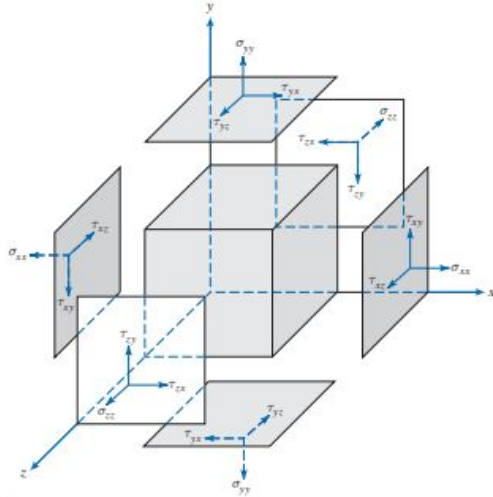


Fig. 2.8 Notation for stress.

1. 9 components
2. Symmetric
3. 6 independent components

$$\bar{\sigma} = \lim_{\delta A \rightarrow 0} \frac{\delta F}{\delta A} \hat{e}_F \hat{e}_A$$

$$\begin{bmatrix} \sigma_{xx} & \sigma_{xy} & \sigma_{xz} \\ \sigma_{yx} & \sigma_{yy} & \sigma_{yz} \\ \sigma_{zx} & \sigma_{zy} & \sigma_{zz} \end{bmatrix}$$

$$\sigma_{xx} = \lim_{(\delta A_x \rightarrow 0)} \frac{\delta F_x}{\delta A_x}$$

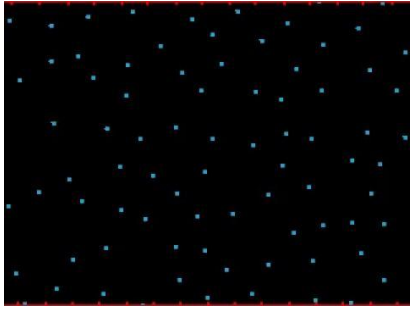
$$\tau_{xy} = \lim_{(\delta A_x \rightarrow 0)} \frac{\delta F_y}{\delta A_x}$$

$$\tau_{xz} = \lim_{(\delta A_x \rightarrow 0)} \frac{\delta F_z}{\delta A_x}$$

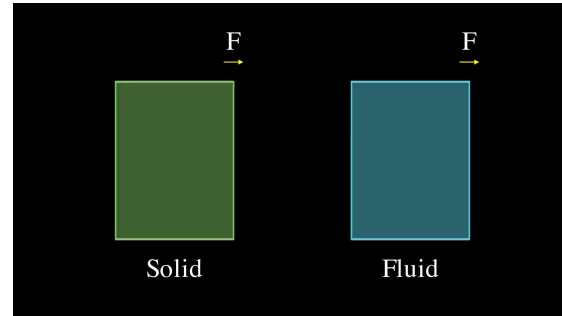
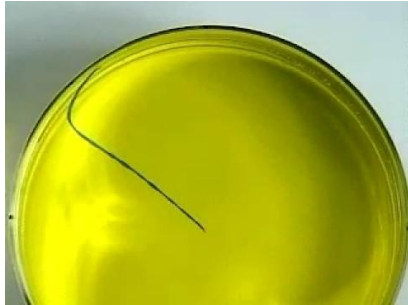
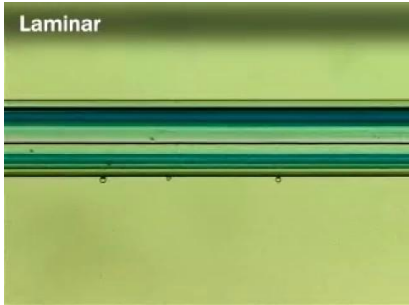
1. Outward normal to the surface is considered to be positive (convention followed)
2. Shear stress is zero when fluid is at rest



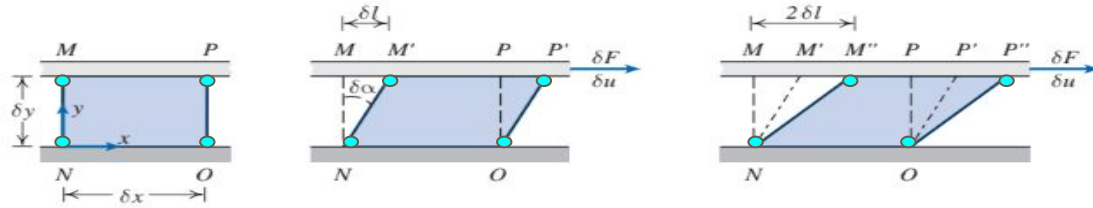
## Stress and Strain rate



Stress in fluid is generated if there is relative motion between fluid particles



## Rate of deformation - One dimensional flow



$$\text{Rate of Deformation} = \frac{d\alpha}{dt} = \lim_{\Delta t \rightarrow 0} \frac{\Delta\alpha}{\Delta t}$$

$$\text{Rate of Deformation} = \frac{d\alpha}{dt} = \lim_{\Delta t \rightarrow 0} \frac{\frac{\Delta l}{\Delta y}}{\Delta t}$$

$$\text{Rate of Deformation} = \frac{d\alpha}{dt} = \lim_{\Delta t \rightarrow 0} \frac{\frac{\Delta u \cancel{\Delta t}}{\Delta y}}{\cancel{\Delta t}}$$

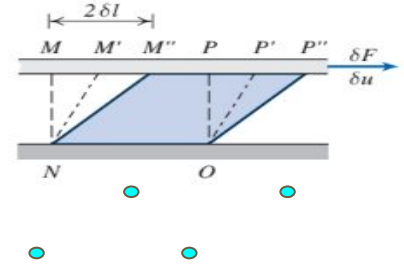
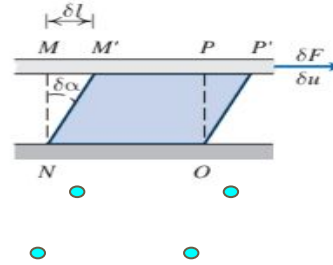
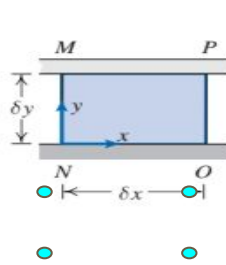
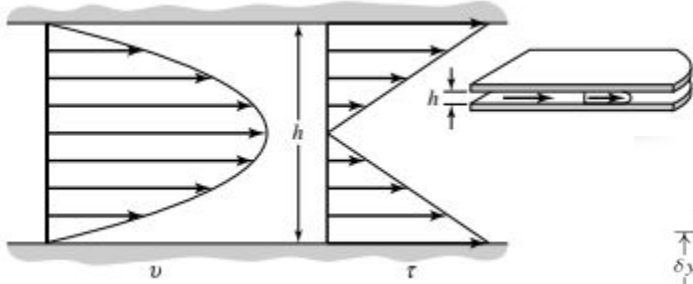
$$\text{Rate of Deformation} = \frac{d\alpha}{dt} = \frac{du}{dy} = \lim_{\Delta y \rightarrow 0} \frac{\Delta u}{\Delta y}$$

$$\Delta\alpha = \frac{\Delta l}{\Delta y}$$

$$\Delta l = \Delta u \Delta t$$

$$\text{Rate of Deformation} = \frac{d\alpha}{dt} = \frac{du}{dy}$$

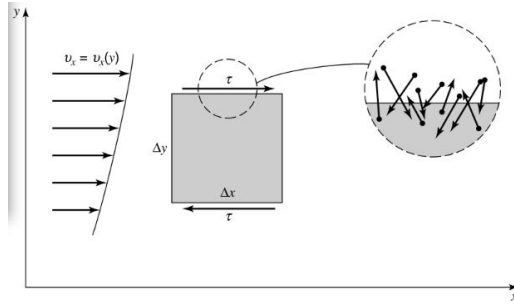
## Shear Stress and Strain rate: Newtonian Fluid



$$\tau_{xy} = \mu \frac{du}{dy}$$

Newtonian Fluid, One dimensional flow

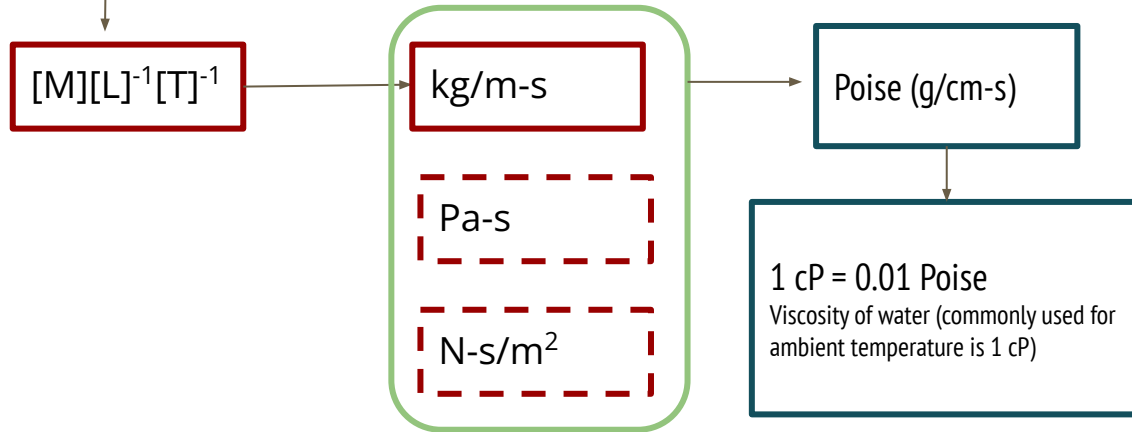
## Coefficient of Viscosity - Some points



1. Dependence of coefficient of viscosity on temperature is determined by the nature of intermolecular interaction of the fluid
2. In case of gas, the viscosity increases with increase in temperature; for ideal gas the dependence is the square root of the temperature (for real gas power may vary between 0.6 - 1)
3. Viscosity does not depend on pressure
4. In case of liquid, viscosity decreases with increase in temperature

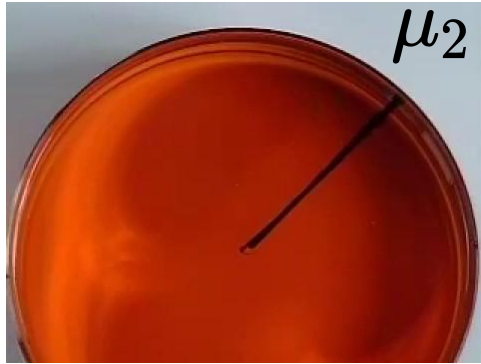
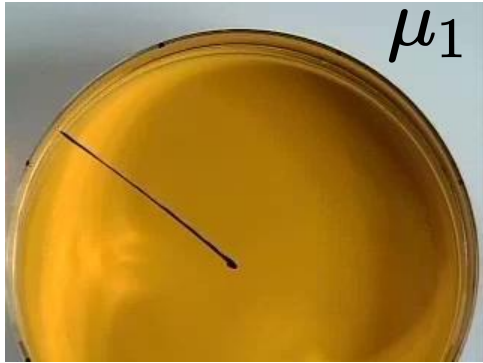
## Units of viscosity

$$\tau_{xy} = \mu \frac{du}{dy}$$



## Test Yourself

$$\tau_{xy} = \mu \frac{du}{dy}$$

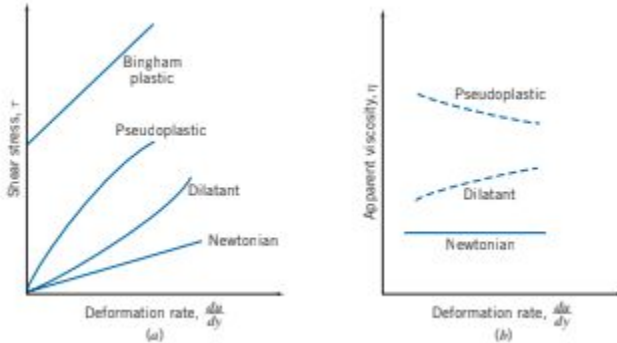
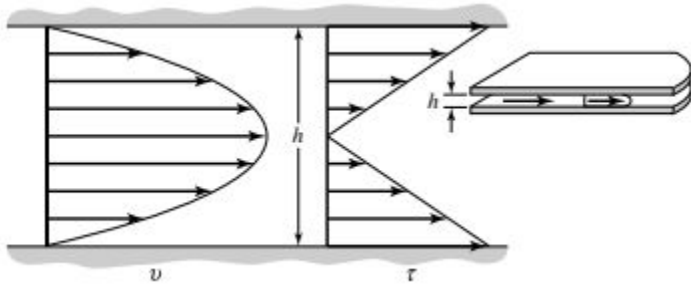


$$\mu_3 > \mu_2 > \mu_1$$

$$\mu_3 < \mu_1 < \mu_2$$

$$\mu_3 < \mu_2 < \mu_1$$

# Shear Stress and Strain rate



Type	Description	Example
Real (Bingham) Plastic	Deforms linearly after the yield stress $\tau_{xy} = \tau_o + \mu \frac{du}{dy}$	Toothpaste
Pseudoplastic (shear-thinning)	“Viscosity” decreases as strain rate increases ( $n < 1$ )	Paint, cream, hair gel
Dilatant (shear thickening)	“Viscosity” increases as strain rate increases ( $n > 1$ )	Sand, silly putty

$$\tau_{xy} = \mu \frac{du}{dy} \left| \frac{du}{dy} \right|^{(n-1)}$$

## Point to point variation

Consider a function  $f(x, y)$

$$df = \left( \frac{\partial f}{\partial x} \right)_y dx + \left( \frac{\partial f}{\partial y} \right)_x dy$$

$$\frac{df}{ds} = \left( \frac{\partial f}{\partial x} \right)_y \frac{dx}{ds} + \left( \frac{\partial f}{\partial y} \right)_x \frac{dy}{ds}$$

$$\frac{df}{ds} = \left( \frac{\partial f}{\partial x} \right)_y \cos \alpha + \left( \frac{\partial f}{\partial y} \right)_x \sin \alpha$$

along an isoline  $\frac{df}{ds} = 0$ ; along a direction of maximum change

$$\tan \alpha^0 = - \frac{\left( \frac{\partial f}{\partial x} \right)_y}{\left( \frac{\partial f}{\partial y} \right)_x}$$

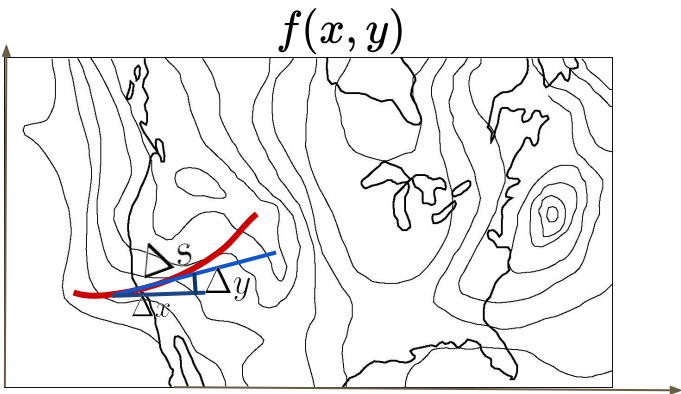
$$\frac{d}{d\alpha} \frac{df}{ds} = 0;$$

$$\tan \alpha^m = + \frac{\left( \frac{\partial f}{\partial y} \right)_x}{\left( \frac{\partial f}{\partial x} \right)_y}$$

$$\tan \alpha^0 = - \cot \alpha^m$$

$$\tan \alpha^0 = - \tan (90^\circ - \alpha^m)$$

$$\boxed{\alpha^0 - \alpha^m = 90^\circ}$$



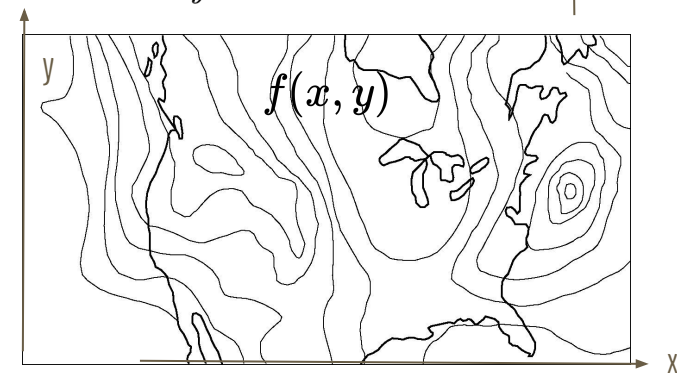
Isolines are normal to the direction of the maximum changes



## Point to point variation

Consider a function  $f(x, y)$

$$\frac{df}{ds} = \left(\frac{\partial f}{\partial x}\right)_y \cos \alpha + \left(\frac{\partial f}{\partial y}\right)_x \sin \alpha$$



direction of maximum change

$$\tan \alpha^m = + \frac{\left(\frac{\partial f}{\partial y}\right)_x}{\left(\frac{\partial f}{\partial x}\right)_y}$$

$$\frac{df}{ds} \Big|_{\max} = \left(\frac{\partial f}{\partial x}\right)_y \cos \alpha^m + \left(\frac{\partial f}{\partial y}\right)_x \sin \alpha^m$$

$$\frac{df}{ds} \Big|_{\max} = \sqrt{\left(\frac{\partial f}{\partial x}\right)_y^2 + \left(\frac{\partial f}{\partial y}\right)_x^2}$$

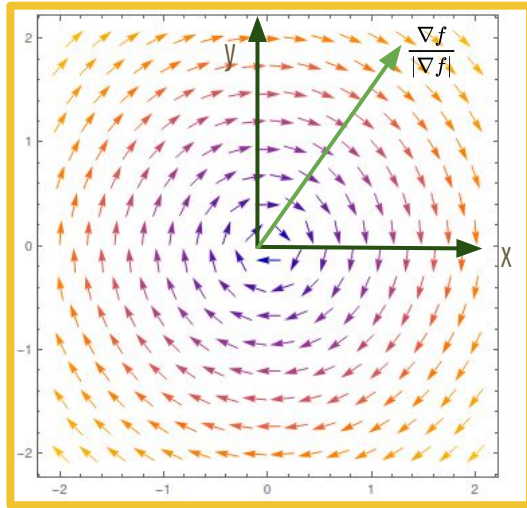
$$\nabla f = \frac{\partial f}{\partial x} \hat{e}_x + \frac{\partial f}{\partial y} \hat{e}_y$$

$$|\nabla f| = \sqrt{\left(\frac{\partial f}{\partial x}\right)_y^2 + \left(\frac{\partial f}{\partial y}\right)_x^2}$$

$$\sin \alpha^m = \frac{\tan \alpha^m}{\sqrt{1 + \tan^2 \alpha^m}} \quad \sin \alpha^m = \frac{\left(\frac{\partial f}{\partial y}\right)_x}{\sqrt{\left(\frac{\partial f}{\partial x}\right)_y^2 + \left(\frac{\partial f}{\partial y}\right)_x^2}}$$
$$\cos \alpha^m = \frac{1}{\sqrt{1 + \tan^2 \alpha^m}} \quad \cos \alpha^m = \frac{1}{\sqrt{\left(\frac{\partial f}{\partial x}\right)_y^2 + \left(\frac{\partial f}{\partial y}\right)_x^2}}$$

## Point to point variation

$$y^2 + x^2 = f(x, y)$$



Streamline Equation :  $y^2 + x^2 = c$

$$\nabla f = 2x\hat{e}_x + 2y\hat{e}_y \quad \text{Radial Direction}$$

# Classification of Fluid Motion

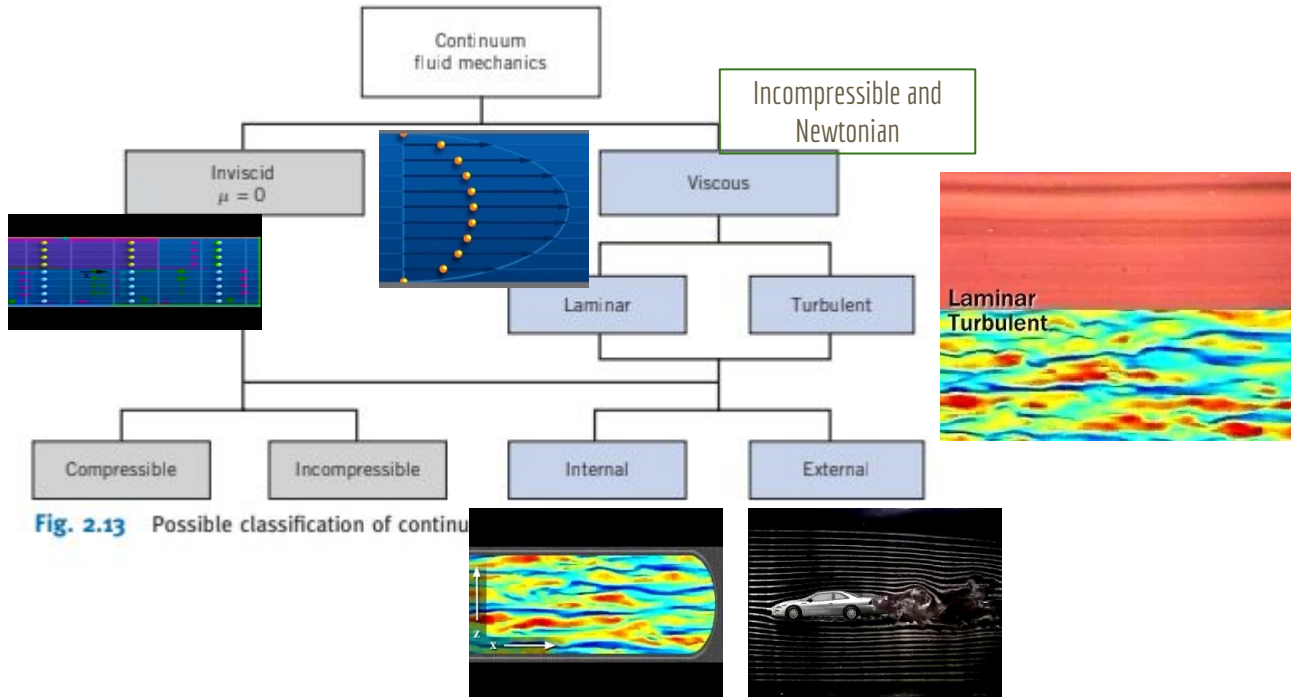
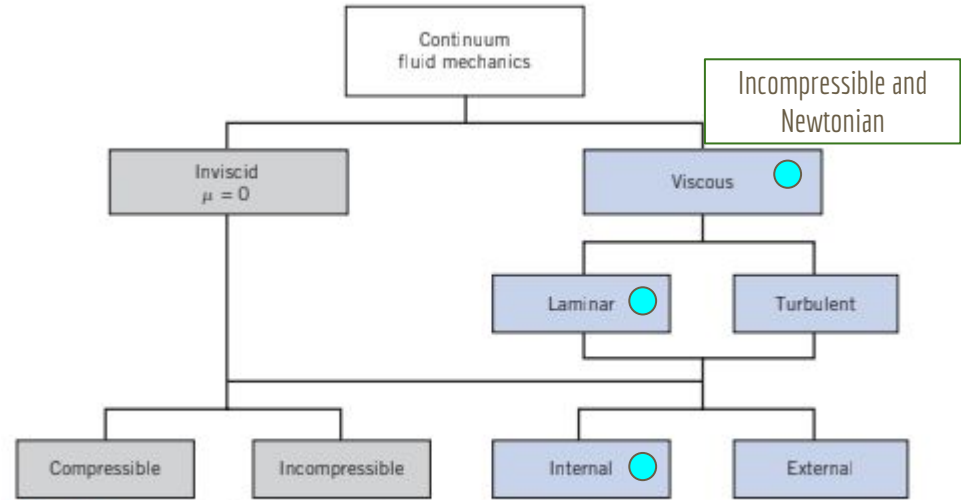


Fig. 2.13 Possible classification of continuum

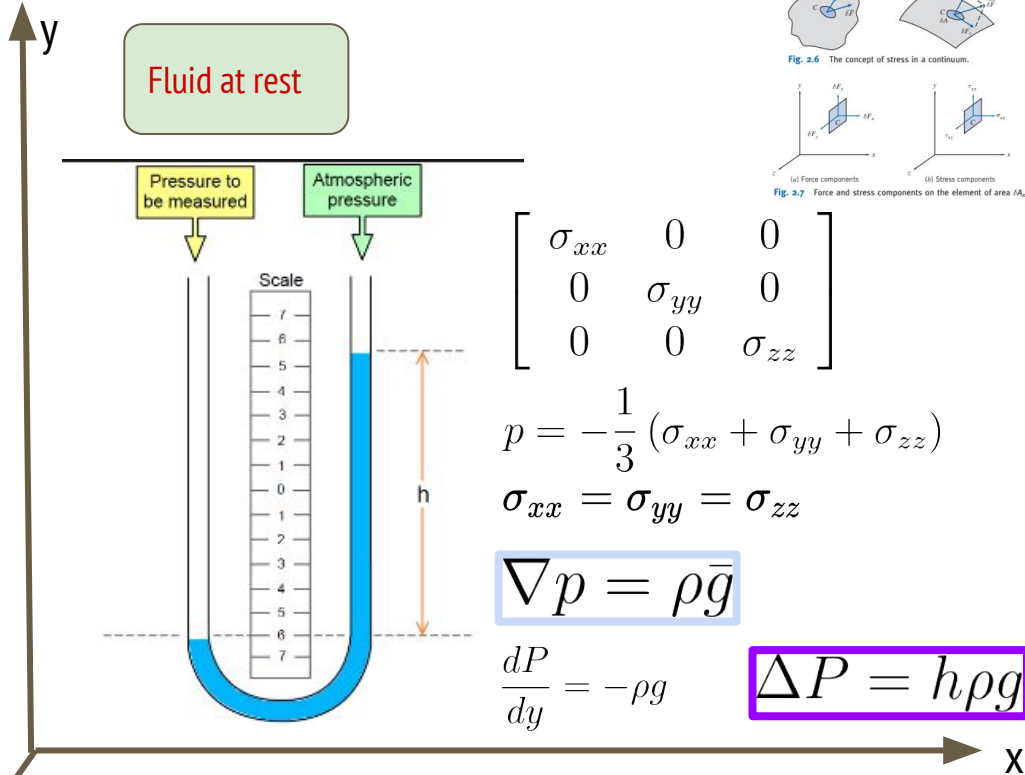
# Summary and Way Forward

1. Fluids and their Properties:
  - a. What is a fluid?
  - b. Difference between Solid and Fluid
  - c. System and Control Volume
  - d. Units
  - e. Fluid and Continuum
  - f. (In)compressibility
  - g. Properties at a point
  - h. Classification
  - i. Point to point variation
  - j. Static Pressure
2. Fluid in motion



**Fig. 2.13** Possible classification of continuum fluid mechanics.

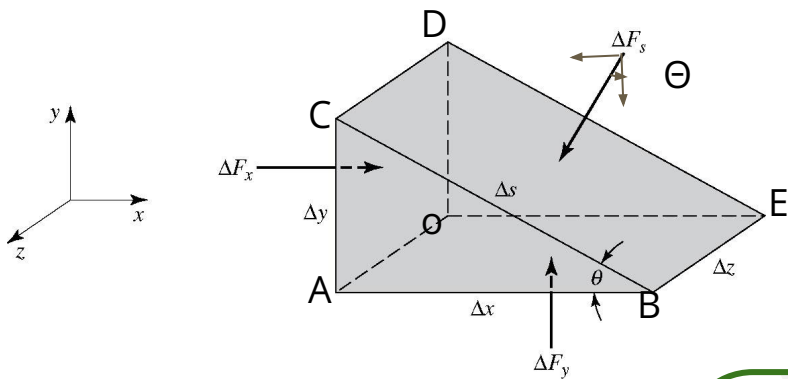
# Point Properties: Pressure



$$\bar{\sigma} = \lim_{\delta A \rightarrow 0} \frac{\delta F}{\delta A} \hat{e}_F \hat{e}_A$$

1. Pressure is defined in compressive sense
2. Pressure is isotropic (*Derivation on next slide as additional learning material*)
3. Pressure is a scalar quantity
4. For incompressible fluid density does not depend on pressure
5. Principle of hydrostatics is important in:
  - a. Computing forces on submerged bodies
  - b. Developing instruments to measure pressure
  - c. Determining forces on hydraulic systems

# Pressure is isotropic



1. Newton's equation is applied to a fluid element as the fluid element approaches to infinitesimally small volume
2. Fluid at rest, so shear force is zero, only normal force acts on a surface
3. Force Balance on the differential fluid element ABCDEO

$$\Sigma F_x = 0; \quad \Sigma F_y = 0$$

$$\begin{aligned} \Delta F_x - \Delta F_s \sin \theta &= 0 \\ \Delta F_x - \Delta F_s \frac{\Delta y}{\Delta s} &= 0 \\ \frac{\Delta F_x}{\Delta y \Delta z} - \frac{\Delta F_s}{\Delta s \Delta z} &= 0 \end{aligned}$$

$$\sigma_{xx} = \sigma_{ss}$$

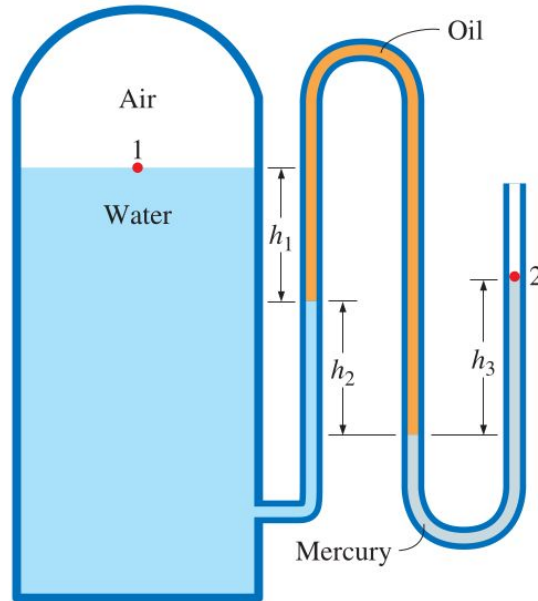
$$\begin{aligned} \Delta F_y - \Delta F_s \cos \theta - \rho g_y \frac{1}{2} (\Delta x \Delta y \Delta z) &= 0 \\ \Delta F_y - \Delta F_s \frac{\Delta x}{\Delta s} - \rho g_y \frac{1}{2} (\Delta x \Delta y \Delta z) &= 0 \\ \frac{\Delta F_y}{\Delta x \Delta z} - \frac{\Delta F_s}{\Delta s \Delta z} - \rho g_y \frac{1}{2} (\Delta y) &= 0 \end{aligned}$$

$$\sigma_{yy} = \sigma_{ss}$$

Vanishes as  $\Delta y \rightarrow 0$

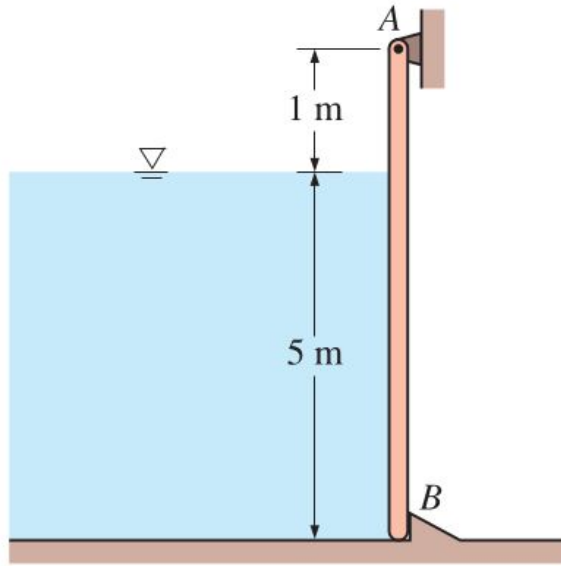
## Practice Problem 1: Hydrostatic

The water in a tank is pressurized by air, and the pressure is measured by a multifluid manometer. The tank is located on a mountain at an altitude of 1400 m where the atmospheric pressure is 85.6 kPa. Determine the air pressure in the tank if  $h_1 = 0.1$  m,  $h_2 = 0.2$  m, and  $h_3 = 0.35$  m. Take the densities of water, oil, and mercury to be  $1000 \text{ kg/m}^3$ ,  $850 \text{ kg/m}^3$ , and  $13,600 \text{ kg/m}^3$ , respectively. (~130 kPa)



## Practice Problem 2: Hydrostatic

A 6-m-high, 5-m-wide rectangular plate blocks the end of a 5-m-deep freshwater channel. The plate is hinged about a horizontal axis along its upper edge through a point A and is restrained from opening by a fixed ridge at point B. Determine the force exerted on the plate by the ridge. (~450 kN)





# Content - Lecture 1

1. Fluids and their Properties:
  - a. What is a fluid?
  - b. Difference between Solid and Fluid
  - c. System and Control Volume
  - d. Units
  - e. Fluid and Continuum
  - f. (In)compressibility
  - g. Properties at a point
  - h. Point to point variation
  - i. Static Pressure
2. Fluid in motion

# Test Yourself

For the velocity field

$$\vec{v}(\vec{x}) = \begin{bmatrix} Ax^2y \\ Bxy^2 \\ 0 \end{bmatrix}$$

Obtain the equation of streamlines

Plot several streamlines in the first quadrant

$A = 2 \text{ m}^{-2} \text{ s}^{-1}$ ,  $B = 1 \text{ m}^{-2} \text{ s}^{-1}$

# Test Yourself

For the velocity field

$$\vec{v}(\vec{x}) = \begin{bmatrix} Ax^2y \\ Bxy^2 \\ 0 \end{bmatrix}$$

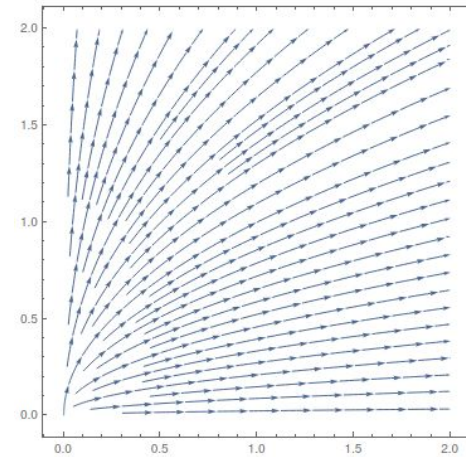
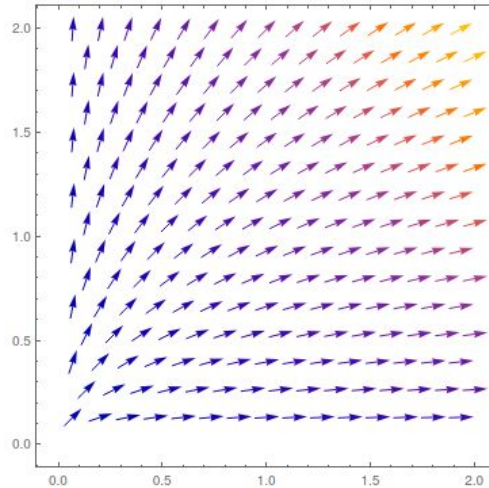
$$\frac{dy}{dx} = \frac{y}{2x}$$

$$y = cx^{\frac{1}{2}}$$

Obtain the equation of streamlines

Plot several streamlines in the first quadrant

$A = 2 \text{ m}^{-2} \text{ s}^{-1}$ ,  $B = 1 \text{ m}^{-2} \text{ s}^{-1}$



# Mass and Momentum Conservation

# Mass and Momentum Conservation

$$\left. \frac{dM}{dt} \right|_{\text{system}} = 0$$

Mass of a system can not be created or destroyed

$$\left. \frac{d\vec{P}}{dt} \right|_{\text{system}} = \Sigma \vec{F}$$

Rate of change of linear momentum is summation of forces acting on the system

Rate of change of the mass of the system

Rate of change of the amount of mass in the control volume

Net rate of inflow of mass to the control volume through the control surface

$$\left. \frac{dM}{dt} \right|_{\text{system}} = \frac{\partial}{\partial t} \int_{\text{CV}} \rho \, dV + \int_{\text{CS}} \rho (\vec{n} \cdot \vec{v}) \, dA$$

Rate of change of the momentum of the system

Rate of change of the amount of the linear momentum in the control volume

Net rate of inflow of the linear momentum to the control volume through the control surface

$$\left. \frac{d\vec{P}}{dt} \right|_{\text{system}} = \frac{\partial}{\partial t} \int_{\text{CV}} \rho (\vec{v}) \, dV + \int_{\text{CS}} \rho (\vec{v}) (\vec{n} \cdot \vec{v}) \, dA$$

# Mass and Momentum Conservation

$$\frac{dM}{dt}|_{\text{system}} = 0$$

$$\frac{d\vec{P}}{dt}|_{\text{system}} = \Sigma \vec{F}$$

System to CV:

$$\frac{dM}{dt}|_{\text{system}} = \frac{\partial}{\partial t} \int_{\text{CV}} \rho(1) dV + \int_{\text{CS}} \rho(1) (\vec{n} \cdot \vec{v}) dA$$

$$\frac{d\vec{P}}{dt}|_{\text{system}} = \frac{\partial}{\partial t} \int_{\text{CV}} \rho(\vec{v}) dV + \int_{\text{CS}} \rho(\vec{v}) (\vec{n} \cdot \vec{v}) dA$$

Body force: Acts throughout the control volume; in the context of fluid mechanics, it is usually gravity

Surface force: Acts on the control surface, e.g., normal and the shear stresses

$$\frac{\partial}{\partial t} \int_{\text{CV}} \rho dV + \int_{\text{CS}} \rho (\vec{n} \cdot \vec{v}) dA = 0$$

$$\frac{\partial}{\partial t} \int_{\text{CV}} \rho \vec{v} dV + \int_{\text{CS}} \rho \vec{v} (\vec{n} \cdot \vec{v}) dA = \Sigma \left( \vec{F}_s + \vec{F}_B \right)$$

$\vec{F}_B$  : Body Force

$\vec{F}_s$  : Surface Force

## Mass Conservation

$$\frac{\partial}{\partial t} \int_{\text{CV}} \rho dV + \int_{\text{CS}} \rho (\vec{n} \cdot \vec{v}) dA = 0$$

@ steady state :  $\int_{\text{CS}} \rho (\vec{n} \cdot \vec{v}) dA = 0$

Convention followed: outward normal: +ve