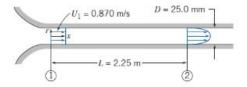
### Recapitulation: Momentum Conservation (Integral Approach)

Air enters a duct, of diameter D=25.0 mm, through a well-rounded inlet with uniform speed,  $U_1=0.870$  m/s. At a downstream section where L=2.25 m, the fully developed velocity profile is

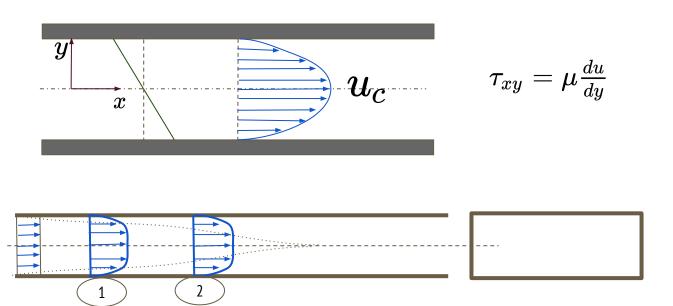
$$\frac{u(r)}{U_c} = 1 - \left(\frac{r}{R}\right)^2 \tag{6}$$

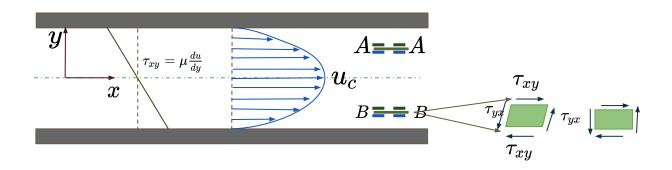
The pressure drop between these sections is  $p_1-p_2 = 1.92 \text{ N/m}^2$ . Find the total force of friction exerted by the tube on the air.

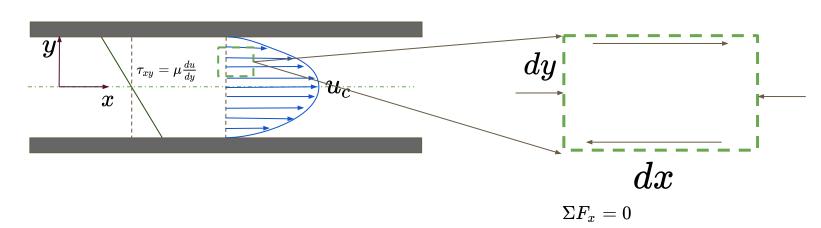


Question: How to determine the velocity profile of the fully developed flow: Differential Approach

Momentum Conservation: Differential Approach





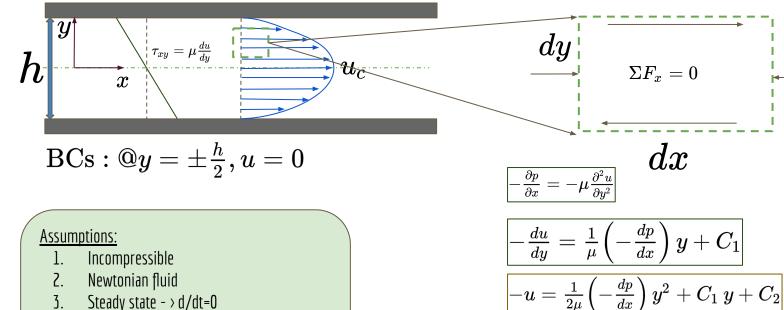


- 1. Incompressible
- 2. Newtonian fluid
- 3. Steady state  $\rightarrow$  d/dt=0
- 4. Fully developed  $\rightarrow du_x/dx = 0$
- 5. Two dimensional geometry ->d/dz=0
- 6. Laminar-> $u_y = 0$ ,  $u_z = 0$
- 7.  $g_x = 0$

$$\left[\left(p|_{x}-p|_{x+\Delta x}
ight)dy(1)+\left(- au_{xy}|_{y}+ au_{xy}|_{y+\Delta y}
ight)dx(1)=0
ight]$$

$$-rac{\partial p}{\partial x}+rac{\partial au_{xy}}{\partial y}=0$$

$$oxed{-rac{\partial p}{\partial x}\,=\,-rac{\partial au_{xy}}{\partial y}} egin{array}{c} au_{xy} = \mu rac{du}{dy} \ \hline -rac{\partial p}{\partial x} = -\mu rac{\partial^2}{\partial y} \end{array}$$

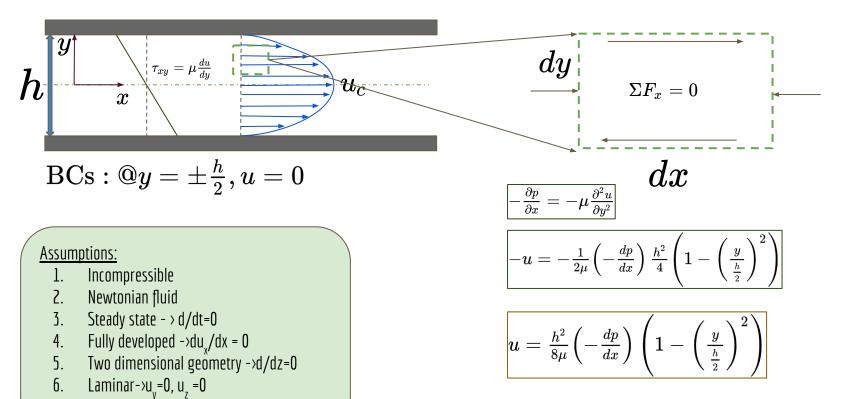


- Steady state > d/dt=0
- Fully developed  $\rightarrow du_v/dx = 0$
- Two dimensional geometry ->d/dz=0
- Laminar-> $u_v = 0$ ,  $u_z = 0$
- $g_x = 0$

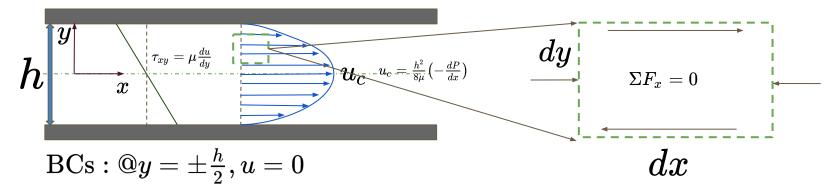
$$C_2 = -rac{1}{2\mu} \Big( -rac{dp}{dx} \Big) rac{h^2}{4} igg| C_1 = 0$$

$$-u=-rac{1}{2\mu}\Bigl(-rac{dp}{dx}\Bigr)\,rac{h^2}{4}\Biggl(1-\left(rac{y}{rac{h}{2}}
ight)^2\Bigr)$$

 $g_x = 0$ 



### Flow between parallel plates (channel flow): Centreline Velocity



- 1. Incompressible
- 2. Newtonian fluid
- 3. Steady state  $\rightarrow$  d/dt=0
- 4. Fully developed  $\rightarrow du_y/dx = 0$
- 5. Two dimensional geometry ->d/dz=0
- 6. Laminar-> $u_v = 0$ ,  $u_z = 0$
- 7.  $g_x = 0$

$$u=rac{h^2}{8\mu}\Bigl(-rac{dp}{dx}\Bigr)\left(1-\left(rac{y}{rac{h}{2}}
ight)^2
ight)$$

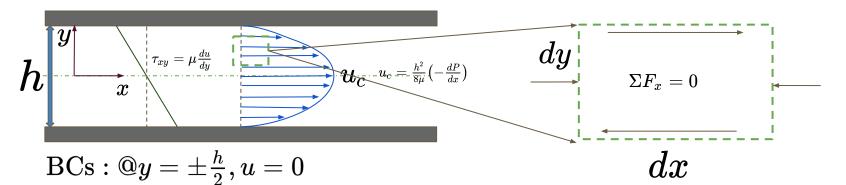
$$u_{
m max} = u_c = rac{h^2}{8\mu}ig(-rac{dP}{dx}ig)$$

$$u(y) = u_c \left(1 - \left(rac{y}{rac{h}{2}}
ight)^2
ight)$$

$$u_c=rac{3}{2}u_{
m av}$$

$$egin{align} u_{\mathrm{avg}} &= rac{1}{rac{h}{2}} \int_0^rac{h}{2} \, u(y) dy(1) \ &= rac{1}{rac{h}{2}} \int_0^rac{h}{2} \, u_c \left(1 - \left(rac{y}{rac{h}{2}}
ight)^2
ight) dy \ &= rac{2}{3} u_c \end{split}$$

## Flow between parallel plates (channel flow): Pressure Drop



- 1. Incompressible
- 2. Newtonian fluid
- 3. Steady state  $\rightarrow$  d/dt=0
- 4. Fully developed  $\rightarrow du_x/dx = 0$
- 5. Two dimensional geometry -xd/dz=0
- 6. Laminar-> $u_y = 0$ ,  $u_z = 0$
- 7.  $g_x = 0$

$$\overline{u_{
m max} = u_c = rac{h^2}{8\mu}ig(-rac{dP}{dx}ig)}$$

$$u_c=rac{3}{2}u_{ ext{avg}}$$

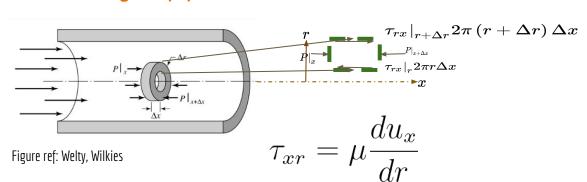
$$\left(-\frac{dP}{dx}\right) = \frac{\Delta P}{L}$$

$$\frac{\Delta P}{L} = \frac{8\mu u_c}{h^2}$$

$$rac{\Delta P}{L} = rac{12 \mu u_{
m avg}}{h^2}$$

### Flow through a pipe with circular cross-section

convention



#### <u>Assumptions:</u>

- Incompressible
- 2. Newtonian fluid
- . Steady state  $\rightarrow$  d/dt=0
- Fully developed  $\rightarrow du_x/dx = 0$
- 5. Axisymmetric  $\rightarrow$ d/d $\Theta$ =0 5. Laminar->u=0 u $\Theta$ =0
- Laminar-> $u_r=0$ ,  $u\Theta=0$  $g_x=0$

$$0 = \bigoplus_{\substack{\text{Outward} \\ \text{normal positive}}} P 2\pi r \Delta r|_{x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal positive}}} P 2\pi r \Delta r|_{x+\Delta x} \bigoplus_{\substack{x \in \mathcal{X} \\ \text{ormal p$$

### Solution

$$\frac{d}{dr}(r\tau_{xr}) = -r\left(-\frac{dp}{dx}\right) - \tau_{xr} = \mu \frac{du_x}{dr}$$

$$\frac{d}{dr}\left(r\mu\frac{du_x}{dr}\right) = -r\left(-\frac{dp}{dx}\right)$$

$$\left(r\,\mu\frac{du_x}{dr}\right) = -\frac{r^2}{2}\left(-\frac{dp}{dx}\right) + c_1$$

$$\frac{du_x}{dr} = -\frac{r}{2\mu} \left( -\frac{dp}{dx} \right) + \frac{c_1}{r}$$

Solution has to be finite at r = 0

$$u_x(r) = -\frac{r^2}{4\mu} \left( -\frac{dp}{dx} \right) + c_2 \qquad c_2 = \frac{R^2}{4\mu} \left( -\frac{dp}{dx} \right)$$

no slip bc :  $u_x(r=R)=0$ 

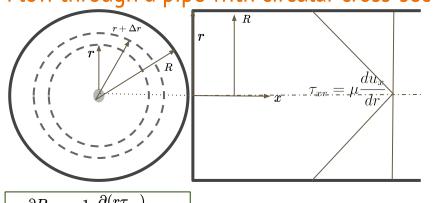
$$u_x(r) = \frac{R^2}{4\,\mu} \left( -\frac{dp}{dx} \right) \, \left( 1 - \frac{r^2}{R^2} \right)$$

- 1. Incompressible
- 2. Newtonian fluid
- 3. Steady state > d/dt=0
- Fully developed ->du/dx = 0
- 5. Axisymmetric ->d/d $\Theta$ =0
- Laminar-> $u_r=0$ ,  $u\Theta=0$
- q x = 0

$$rac{dP}{dx} = rac{1}{r}rac{d}{dr}(r au_{rx})$$

- 1. LHS is function of "r" alone
- 2. P is linear in "x"
- . dP/dx is independent of "r"
- from "r" momentum conservation

## Flow through a pipe with circular cross-section & a channel



$$\left[ -\frac{\partial P}{\partial x} + \frac{1}{r} \frac{\partial (r\tau_{rx})}{\partial r} = 0 \right]$$

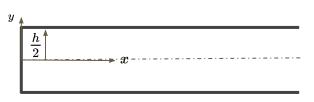
$$u_x(r) = \frac{R^2}{4\pi} \left( -\frac{dp}{dx} \right) \left( 1 - \frac{r^2}{P^2} \right)$$

$$u_x(r) = \frac{1}{4\mu} \left( -\frac{1}{dx} \right) \left( 1 - \frac{1}{R^2} \right)$$

$$u_{
m max} = u_c = rac{R^2}{4\mu} ig( -rac{dP}{dx} ig)$$

$$egin{aligned} u_x(r) &= u_c \left(1 - \left(rac{r}{R}
ight)^2
ight) \ u_{ ext{avg}} &= rac{1}{\pi R^2} \int_0^R u(r) 2\pi r dr \ &= rac{2u_c}{r} \int_0^R \left(1 - \left(rac{r}{R}
ight)^2
ight) r dr \end{aligned}$$

$$egin{align} u_{
m avg} &= u_c \left(1 - \left(rac{R}{R}
ight)
ight) \left|u_{
m avg} &= rac{1}{\pi R^2} \int_0^{\infty} u(r) 2\pi r dr \ &= rac{2u_c}{R^2} \int_0^R \left(1 - \left(rac{r}{R}
ight)^2
ight) r dr \ &= rac{1}{\pi} u_a \end{split}$$



$$au_{xy} = \mu rac{du}{dy}$$

$$rac{\partial au_{xy}}{\partial au_{xy}}$$

$$u=rac{h^2}{8\mu}\Bigl(-rac{dp}{dx}\Bigr)\left(1-\left(rac{y}{rac{h}{2}}
ight)^2
ight)$$

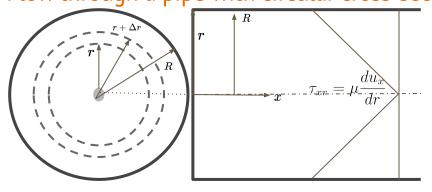
$$u_{
m max} = u_c = rac{h^2}{8\mu}ig(-rac{dP}{dx}ig)$$

$$u(y) = u_c \left(1 - \left(rac{y}{rac{h}{2}}
ight)^2
ight)$$
 ,  $u_{ ext{avg}} = rac{1}{rac{h}{2}} \int_0^rac{h}{2} \, u(y) dy(1)$ 

$$u_{
m avg}$$

$$\left| u_c = rac{3}{2} u_{ ext{avg}} 
ight| = rac{1}{rac{h}{2}} \int_0^{rac{h}{2}} u_c \left( 1 - \left( rac{y}{rac{h}{2}} 
ight)^2 
ight) dy$$

# Flow through a pipe with circular cross-section & a channel



$$\frac{h}{2}$$
  $x$ 

$$u_{
m max} = u_c = rac{R^2}{4\mu}ig(-rac{dP}{dx}ig)$$

$$u_c = 2u_{
m avg}$$

$$\left(-\frac{dP}{dx}\right) = \frac{\Delta P}{L}$$

$$rac{\Delta P}{L} = rac{4 \mu u_c}{R^2}$$

$$rac{\Delta P}{L} = rac{8 \mu u_{
m avg}}{R^2}$$

$$\frac{D}{D} = rac{32 \mu u_{
m avg}}{D^2}$$

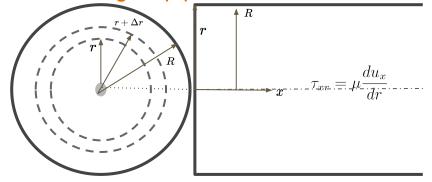
$$egin{aligned} u_{ ext{max}} &= u_c = rac{h^2}{8\mu}(-rac{dP}{dx}) \ u_c &= rac{3}{2}u_{ ext{avg}} \end{aligned}$$

$$\left(-\frac{dP}{dx}\right) = \frac{\Delta}{2}$$

$$\frac{P}{L} = \frac{8\mu u_c}{h^2}$$

$$rac{P}{h} = rac{12 \mu u_{
m avg}}{h^2}$$

Flow through a pipe with circular cross-section

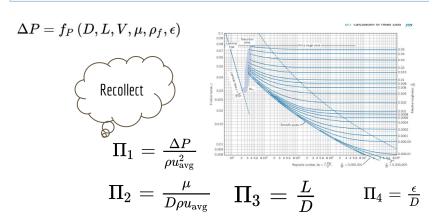


$$rac{\Delta P}{L} = rac{32 \mu u_{
m avg}}{D^2}$$

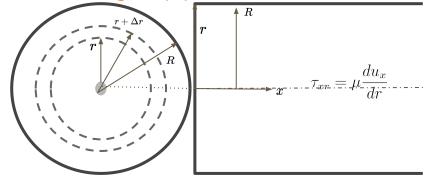
$$\frac{\Delta P}{\frac{1}{2}\rho u_{\mathrm{avg}}^2} = \frac{64\mu}{\rho u_{\mathrm{avg}}D} \frac{L}{D}$$

$$rac{\Delta P}{rac{1}{2}
ho u_{
m avg}^2} = \left(rac{64\mu}{
ho u_{
m avg}D}
ight)\left(rac{L}{D}
ight)$$

Test yourself: Pressure drop in a pipe flow



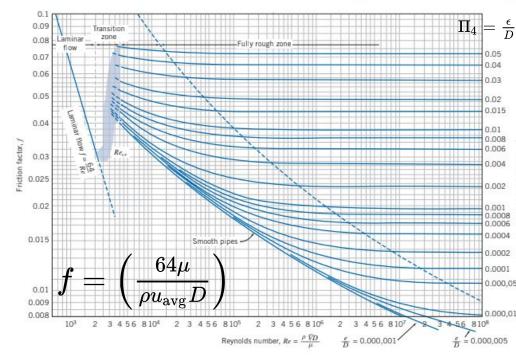
### Flow through a pipe with circular cross-section



$$rac{\Delta P}{L} = rac{32 \mu u_{
m avg}}{D^2}$$

$$\left[ rac{\Delta P}{rac{1}{2}
ho u_{
m avg}^2} = rac{64\mu}{
ho u_{
m avg} D} rac{L}{D} 
ight]$$

$$rac{\Delta P}{rac{1}{2}
ho u_{
m avg}^2} = \left(rac{64\mu}{
ho u_{
m avg}D}
ight)\left(rac{L}{D}
ight)$$



# Flow through a pipe: Pressure Drop and Frictional Losses

