



## Modelling the High Altitude Environment

### Parameters used by one or more of these models:

Temperature at sea level:	$T_0 = 288 \text{ K} = 15^\circ\text{C} = 59^\circ\text{F} = 518 \text{ R}$
Air density at sea level:	$\rho_0 = 1.225 \text{ kg/m}^3 = 0.07648 \text{ lb/ft}^3 = 0.0023769 \text{ slug/ft}^3$
Air pressure at sea level:	$P_0 = 101325 \text{ N/m}^2 = 2116 \text{ lb/ft}^2 = 14.69 \text{ lb/in}^2$
Effective height of atmosphere:	$a = 8.42 \text{ km}$ . This is the height the atmosphere would have to have if its density were <i>constant</i> with altitude, and equal to $\rho_0$ , in order to maintain pressure $P_0$ at sea level.

### Simple Exponential Atmosphere Model:

If you assume that the atmosphere consists of an ideal gas (a reasonable assumption) at constant temperature (not so reasonable), then you come up with the following model of pressure and density vs. altitude:

Temperature:	$T = T_0$ , independent of $h$ .
Pressure:	$P = P_0 e^{-h/a}$
Density:	$r = r_0 e^{-h/a}$

## 1976 Standard Atmosphere Model:

For altitudes between 32 and 47 km, NASA, NOAA, and USAF use this empirical model:

Temperature:	$T = T_0 (0.482561 + h / 102910 \text{ m})$
Pressure:	$P = P_0 (0.898309 + h / 55282 \text{ m})^{-12.20114}$
Density:	$r = r_0 (0.857003 + h / 57947 \text{ m})^{-13.20114}$

## Models' Predictions for the Environment at Float Height:

	<b>Exponential Model</b>	<b>1976 Empirical Model</b>
Air Temperature:	288 K = 15 C	251 K = -22 C
Air Pressure:	$P = 876 \text{ N/m}^2$	$P = 277 \text{ N/m}^2$
Air Density:	$r = 0.0106 \text{ kg/m}^3$	$r = 0.0039 \text{ kg/m}^3$

So, which model did the GLAST Balloon Flight designers use to design their hardware and plan its course? And what conditions did the GLAST Balloon actually encounter at this height? Check the [GLAST Flight Parameters](#) page to see what the planners

expected, or the [GLAST Balloon Flight environmental data](#) page to find out what the balloon actually measured...