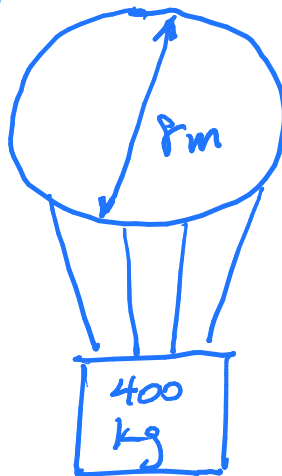
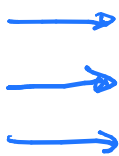


1. [Chapter 15] An 8-m-diameter hot air balloon that has a total mass of 400 kg is standing still in air on a windless day. The balloon suddenly is subjected to a 50 km/hr wind. Determine the initial acceleration of the balloon in the horizontal direction.

$$U_{\infty} = 50 \text{ km/hr} = 13.9 \text{ m/s}$$



Air @ 20°C

$$\rho = 1.204 \frac{\text{kg}}{\text{m}^3}$$

$$\mu = 1.825 \times 10^{-5} \frac{\text{kg}}{\text{m} \cdot \text{s}}$$

- Drag force acting on the balloon:

$$F_D = C_D \frac{1}{2} \rho U_{\infty}^2 A$$

Frontal area: $\frac{\pi D^2}{4}$

- Model balloon as a sphere $\Rightarrow C_D$

- Need Reynolds number to determine C_D

$$Re = \frac{\rho U_{\infty} D}{\mu} = \frac{(1.204)(13.9)(8)}{1.825 \times 10^{-5}} = 7.3 \times 10^6$$

\Rightarrow turbulent

From Table 15-2 ($Re \geq 2 \times 10^6$) $\Rightarrow C_D = 0.2$

$$F_D = (0.2) \frac{1}{2} (1.204) (13.9)^2 \frac{\pi (8)^2}{4} = 1167 \text{ N}$$

- To get initial acceleration: $F = ma$

$$a = \frac{F_D}{m} = \frac{1167}{400} = 2.9 \frac{\text{m}}{\text{s}^2}$$

2. [Chapter 15] To reduce the drag coefficient and thus improve the fuel efficiency of cars, the design of side rearview mirrors has changed dramatically in recent decades from a simple circular plate to a streamlined shape. Determine the amount of fuel and money saved per year as a result of replacing a 15-cm-diameter flat mirror by one with a hemispherical back. Assume the car is driven 25,000 km a year at an average speed of 95 km/hr. Take the density and price of gasoline to be 0.75 kg/L and \$0.90/L, respectively; the heating value of gasoline to be 44,000 kJ/kg and the overall efficiency of the engine to be 30 percent.

$$U_{\infty} = 95 \text{ km/hr} = 26.4 \text{ m/s}$$

$$D = 15 \text{ cm} = 0.15 \text{ m}$$

$$\text{Air @ } 20^{\circ}\text{C} \Rightarrow \begin{cases} \rho = 1.2 \frac{\text{kg}}{\text{m}^3} \\ \mu = 1.83 \times 10^{-5} \frac{\text{kg}}{\text{m}\cdot\text{s}} \end{cases}$$

$$Re = \frac{\rho U_{\infty} D}{\mu} = 2.6 \times 10^5$$

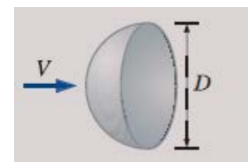
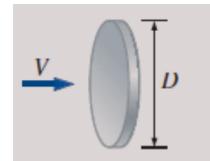
- Drag force Frontal area: $\frac{\pi D^2}{4}$

$$F_D = C_D \frac{1}{2} \rho U_{\infty}^2 A$$

- C_D values from Table 15-2:

Thin circular disk: $C_D = 1.1$

Hemisphere: $C_D = 0.4$



- Energy lost to drag (per year):

$$E_{\text{drag}} = (F_D) \left(\frac{\text{distance traveled}}{\text{year}} \right) \quad \left\{ \frac{\text{kJ}}{\text{yr}} \right\}$$

- Fuel energy required to overcome drag:

$$E_{\text{required}} = \frac{E_{\text{drag}}}{\eta_{\text{engine}}} \quad 30\% \quad \left\{ \frac{\text{kJ}}{\text{yr}} \right\}$$

- Fuel required = $\frac{E_{\text{required}}}{\text{heating value}} \quad \left\{ \frac{\text{kg}}{\text{yr}} \right\}$

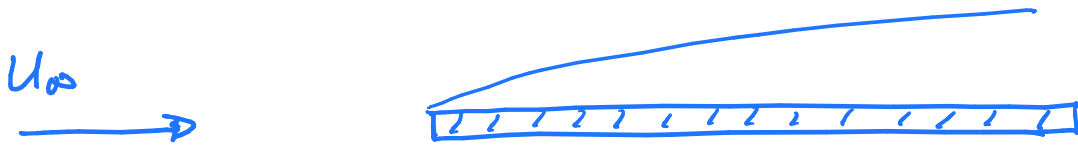
- Fuel required = $\frac{E_{\text{required}}}{(\text{heating value})(\rho_{\text{fuel}})} = \left\{ \frac{\text{L}}{\text{yr}} \right\}$

| <u>Mirror type</u> | <u>F_D (N)</u> | <u>$E_{drag} (\frac{kJ}{yr})$</u> | <u>$E_{required} (\frac{kJ}{yr})$</u> |
|--------------------|-----------------------------|--|--|
| Thin Disk | 8.12 | 2.03×10^5 | 6.77×10^5 |
| Hemisphere | 2.95 | 7.38×10^4 | 2.46×10^5 |

| <u>Mirror type</u> | <u>Fuel required ($\frac{L}{yr}$)</u> | <u>Cost ($\frac{\\$}{yr}$)</u> |
|--------------------|--|---|
| Thin Disk | 20.5 | 18.46 |
| Hemisphere | 7.46 | 6.71 |
| Savings: | 13.1 L/yr | \$11.75/yr |

\$0.90/L →

3. [Chapter 15] Consider laminar flow of a fluid over a flat plate. Now the freestream velocity is doubled. Determine the change in the drag force on the plate. Assume that the flow remains laminar.



- Drag force on a flat plate:

$$F_D = C_f \frac{1}{2} \rho U_\infty^2 A$$

- If $U_{\infty 2} = 2 U_{\infty 1}$, find change in drag force

$$\frac{F_{D2}}{F_{D1}} = \frac{C_{f2} \frac{1}{2} \rho U_{\infty 2}^2 A}{C_{f1} \frac{1}{2} \rho U_{\infty 1}^2 A} = \frac{C_{f2} (2U_{\infty 1})^2}{C_{f1} U_{\infty 1}^2} = \frac{4 C_{f2}}{C_{f1}}$$

- For laminar flow: $C_f = \frac{1.33}{Re_L^{1/2}}$

$$\text{with } Re_L = \frac{\rho U_\infty L}{\mu}$$

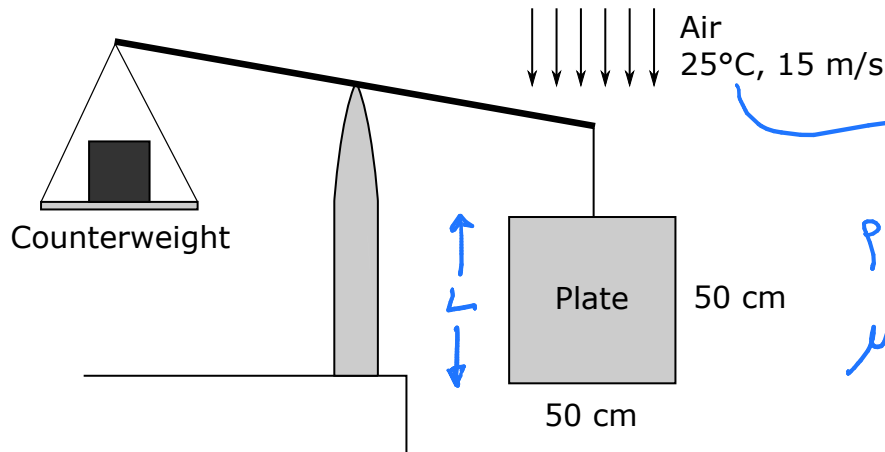
$$\frac{F_{D2}}{F_{D1}} = 4 \frac{C_{f2}}{C_{f1}} = 4 \frac{1.33/Re_{L2}^{1/2}}{1.33/Re_{L1}^{1/2}} = 4 \frac{Re_{L1}^{1/2}}{Re_{L2}^{1/2}}$$

$$= 4 \frac{(\rho U_{\infty 1} L / \mu)^{1/2}}{(\rho U_{\infty 2} L / \mu)^{1/2}} = 4 \frac{U_{\infty 1}^{1/2}}{U_{\infty 2}^{1/2}} = 4 \frac{U_{\infty 1}^{1/2}}{(2U_{\infty 1})^{1/2}}$$

$$= \frac{4}{\sqrt{2}} = 2.83$$

$$\hookrightarrow F_{D2} = 2.83 F_{D1}$$

4. [Chapter 15] The weight of a thin flat plate 50 cm x 50 cm in size is balanced by a counterweight that has a mass of 2 kg. Now a fan is turned on and air at 1 atm and 25°C flows downward over both surfaces of the plate (front and back in the sketch) with a freestream velocity of 15 m/s. Determine the mass of the counterweight that needs to be added in order to balance the plate in this case.



$$\rho = 1.184 \text{ kg/m}^3$$

$$\mu = 1.85 \times 10^{-5} \frac{\text{kg}}{\text{m}\cdot\text{s}}$$

- Need to calculate the drag force on the plate

$$F_D = C_f \frac{1}{2} \rho U_\infty^2 A$$

- To get C_f , first calculate Re to determine flow regime

$$Re_L = \frac{\rho U_\infty L}{\mu} = \frac{(1.184)(15)(\frac{50}{100})}{1.85 \times 10^{-5}} = 4.8 \times 10^5$$

laminar $\swarrow < 5 \times 10^5$

- For laminar flow over flat plate:

$$C_f = \frac{1.33}{Re_L^{1/2}} = 0.001919$$

- Drag force: $F_D = C_f \frac{1}{2} \rho U_\infty^2 A$

$$= (0.0019) \frac{1}{2} (1.18) (15)^2 \left(\frac{50}{100}\right) \left(\frac{50}{100}\right) (2)$$

$$= 0.1278 \text{ N}$$

- Additional counterweight:

$$W = mg = F_D \Rightarrow m = \frac{F_D}{g} = \frac{0.1278}{9.81} = 0.013 \text{ kg} = 13 \text{ g}$$

Both sides \swarrow

5. [Chapter 15] A jumbo jet airplane has a mass of about 400,000 kg when fully loaded with over 400 passengers and takes off at a speed of 280 km/hr. Determine the takeoff speed when the airplane has 100 empty seats assuming that each passenger with luggage is 150 kg and the wing and flap settings are maintained the same.

• At lift-off weight = lift force:

$$W = F_L = C_L \frac{1}{2} \rho v^2 A$$

$$\Rightarrow v = \sqrt{\frac{2W}{\rho C_L A}}$$

• Comparing the take-off speed under two weight scenarios:

$$\frac{v_2}{v_1} = \frac{\sqrt{\frac{2W_1}{\cancel{\rho C_L A}}}}{\sqrt{\frac{2W_2}{\cancel{\rho C_L A}}}}$$

ρ, C_L, A the same
for both cases

$$= \sqrt{\frac{W_1}{W_2}} = \sqrt{\frac{\cancel{m_1 g}}{m_2 \cancel{g}}} = \sqrt{\frac{m_2}{m_1}}$$

$$v_2 = v_1 \sqrt{\frac{m_1}{m_2}} \quad v_1 = 280 \frac{\text{km}}{\text{hr}} \quad m_1 = 400,000 \text{ kg}$$

1 passenger w/ luggage = 150 kg

100 passengers = (100)(150) = 15,000 kg

$$m_2 = m_1 - 15,000 = 385,000 \text{ kg}$$

$$v_2 = 280 \sqrt{\frac{385}{400}} = 275 \frac{\text{km}}{\text{hr}}$$

Answers

1. 2.9 m/s^2
2. 13.1 L/year, \$11.75/year
3. 2.8x increase
4. 13 g
5. 275 km/hr