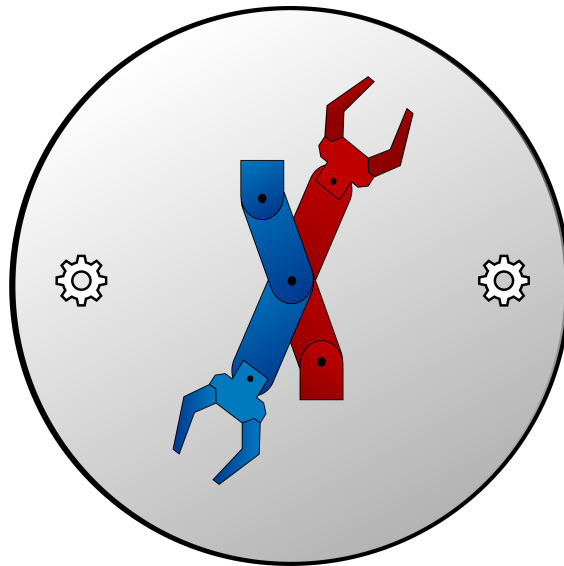


FINAL REPORT

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ME 407

Preliminary Design of Robotic Systems

Embry-Riddle Aeronautical University



Meiosis

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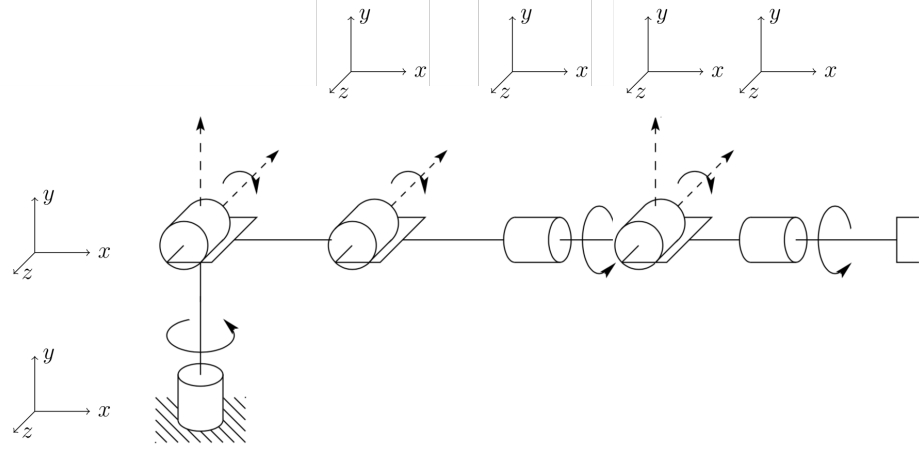


Figure 1: Coordinate Systems

$${}^I_B r_1 = \begin{bmatrix} 0 \\ 0 \\ \ell_b \end{bmatrix} \quad {}^1_1 r_2 = \begin{bmatrix} 0 \\ 0 \\ \ell_1 \end{bmatrix} \quad {}^2_2 r_3 = \begin{bmatrix} 0 \\ \ell_2 \\ 0 \end{bmatrix} \quad {}^3_3 r_4 = \begin{bmatrix} 0 \\ \ell_3 \\ 0 \end{bmatrix} \quad {}^4_4 r_5 = \begin{bmatrix} 0 \\ \ell_4 \\ 0 \end{bmatrix} \quad {}^5_5 r_6 = \begin{bmatrix} 0 \\ \ell_5 \\ 0 \end{bmatrix}$$

$${}^I_B r_1 = \begin{bmatrix} 0 \\ 0 \\ \ell_b \end{bmatrix} \quad {}^I_B r_2 = \begin{bmatrix} 0 \\ 0 \\ \ell_b + \ell_1 \end{bmatrix} \quad {}^I_B r_3 = \begin{bmatrix} -\ell_2 c_{\theta_2} s_{\theta_1} \\ \ell_2 c_{\theta_{12}} \\ \ell_b + \ell_1 + \ell_2 s_{\theta_2} \end{bmatrix} \quad {}^I_B r_4 = \begin{bmatrix} -s_{\theta_1}(\ell_3 c_{\theta_{23}} + \ell_2 c_{\theta_2}) \\ c_{\theta_1}(\ell_3 c_{\theta_{23}} + \ell_2 c_{\theta_2}) \\ \ell_1 + \ell_b + \ell_3 s_{\theta_{23}} + \ell_2 s_{\theta_2} \end{bmatrix}$$

$${}^I_B r_5 = \begin{bmatrix} -s_{\theta_1}(\ell_3 c_{\theta_{23}} + \ell_4 c_{\theta_{23}} + \ell_2 c_{\theta_2}) \\ c_{\theta_1}(\ell_3 c_{\theta_{23}} + \ell_4 c_{\theta_{23}} + \ell_2 c_{\theta_2}) \\ \ell_1 + \ell_b + \ell_3 s_{\theta_{23}} + \ell_4 s_{\theta_{23}} + \ell_2 s_{\theta_2} \end{bmatrix} \quad {}^I_B r_6 = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$

6 Motor Dynamics

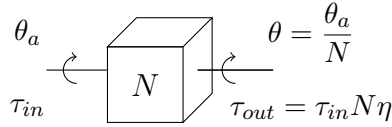
Given robot dynamics described by $H(\gamma)\ddot{\gamma} + n(\gamma, \dot{\gamma}) = \tau$, the torque, τ , provided by the servo motors is necessary to solve the closed loop dynamics of the system. Assuming the servo is driven by a D.C. motor with proportional derivative control,

$$\tau_a = Ki_a = J_a\ddot{\theta}_a + b_a\dot{\theta}_a + \tau_L \quad (1)$$

Where τ_a is the actuator torque, K is the back-EMF constant, i_a is the motor current, J_a is the armature inertia, θ_a , $\dot{\theta}_a$, $\ddot{\theta}_a$ is the motor position and it's first and second time derivatives, respectively, b_a is the viscous friction coefficient, and τ_L is the torque available for the actuator to do work. The basic equation for a motor is known to be:

$$V_a = i_a R_a + K\dot{\theta}_a \quad (2)$$

Where V_a is the voltage applied to the actuator and R_a is the armature resistance. Given a gearbox with in/out ratio N and efficiency η ,



The motor equation (1) can be expressed in the output coordinates:

$$Ki_a = J_a N \ddot{\theta} + b_a N \dot{\theta} + \frac{\tau}{N\eta}$$

Substituting into equation (2) and solving for i_a :

$$\begin{aligned} i_a &= \frac{J_a N}{K} \ddot{\theta} + b_a N \dot{\theta} + \frac{\tau}{N\eta} \\ V_a &= \frac{R_a J_a N}{K} \ddot{\theta} + \frac{R_a b_a N}{K} \dot{\theta} + \frac{R_a}{K N \eta} \tau + K N \dot{\theta} \end{aligned} \quad (3)$$

Assuming P.D. control, $V_a = K_p(\theta - \theta_d) + K_d\dot{\theta}$, where θ_d is the desired orientation of the actuator, the following solution is found by setting the P.D. solution equal to (3). After collecting like terms:

$$\frac{R_a J_a N}{K} \ddot{\theta} + \left(\frac{R_a J_a N}{K} - K_d + K N \right) \dot{\theta} - K_p \theta = -K_p \theta_d - \frac{R_a}{K N \eta} \tau \quad (4)$$

The following parameters of the system can be obtained by applying a step input to the system with $\tau = 0$ and measuring the characteristics of its response. Denoting ζ as the damping ratio and ω_n as the natural frequency of the system,

$$\% \text{ Overshoot} = \left(\frac{\theta_{max} - \theta_{ss}}{\theta_{ss}} \right) \times 100, \quad \zeta = \frac{-\ln(\%OS/100)}{\sqrt{\pi^2 + \ln^2(\%OS/100)}}, \quad \omega_n = \frac{\pi}{T_p \sqrt{1 - \zeta^2}}$$

Given θ_{max} , θ_{ss} , and T_p as measured parameters of the system's max output, steady state, and time to peak, respectively.

Refactoring equation (4) and equating with the general solution for a second order system given by $\ddot{\theta} + 2\zeta\omega_n\dot{\theta} + \omega_n^2\theta = \omega_n^2\theta_d$,

$$2\zeta\omega_n = \frac{b_a}{J_a} - \frac{KK_d}{R_aJ_aN} + \frac{K^2}{R_aJ_a}, \quad \omega_n^2 = \frac{-KK_p}{R_aJ_aN} \quad (5)$$

Performing a similar experiment as previously described, except with a known inertial load $\tau = J_m\ddot{\theta}$, the following parameters can be found:

$$\alpha_m \equiv 2\zeta\omega_n = \frac{R_ab_aN^2\eta - KK_dN\eta + K^2N^2\eta}{R_aJ_aN^2\eta + R_aJ_m}, \quad \beta_m \equiv \omega_n = -\frac{KK_pN\eta}{R_aJ_aN^2\eta + R_aJ_m} \quad (6)$$

$$\begin{bmatrix} 1 & -(\alpha_1J_1 + \beta_1J_1) \\ 1 & -(\alpha_2J_2 + \beta_2J_2) \\ \vdots & \vdots \end{bmatrix} \begin{bmatrix} \frac{R_ab_aN^2\eta - KK_dN\eta + K^2N^2\eta - KK_pN\eta}{R_aJ_aN^2\eta} \\ 1 \\ \frac{1}{J_aN^2\eta} \end{bmatrix} = \begin{bmatrix} \alpha_1 + \beta_1 \\ \alpha_2 + \beta_2 \\ \vdots \end{bmatrix} \quad (7)$$

Finally, with multiple datasets (varying inertial loads, J_m) the coefficients of the second order system equation can be found using the least squares solution of (7).

Acknowledgements & Attributions

We would like to acknowledge the following people for their contributions to creating this report?

— Dr. Isenberg

I Appendix

i Drawings

ii Salient Code