

## AERO483/ENGR6461 – Homework 2

### Problem 1

For the system in figure 1, where all coordinates are in kilometers, the satellites A, B, and C have pseudorange measurements of:

$$\rho_A = 5 \text{ Km}$$

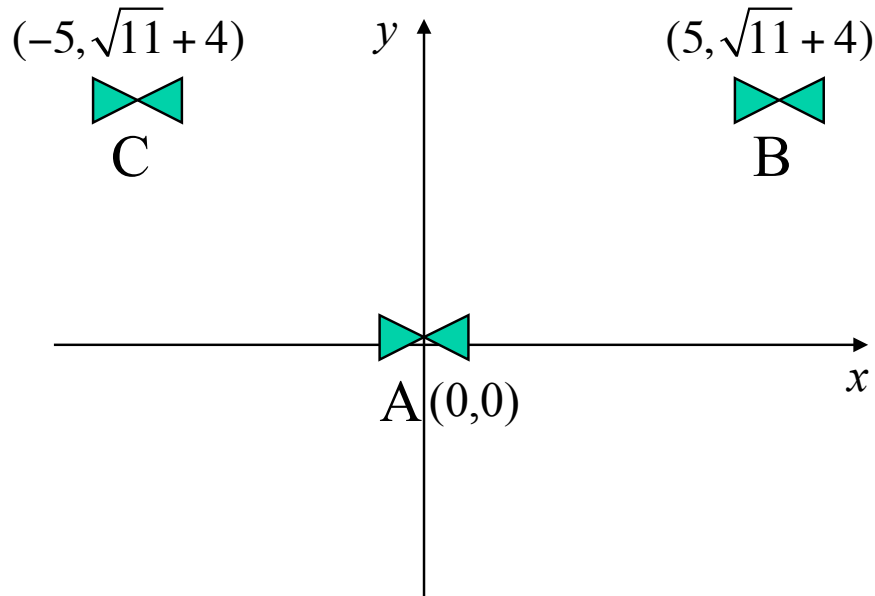
$$\rho_B = 7 \text{ Km}$$

$$\rho_C = 6 \text{ Km}$$

1. Using the linearized GPS equations, compute the position of the user assuming that initially the user is 4Km away from the origin on the positive y-axis.
2. For the same conditions as in item 1. compute the regularized least squares solution with  $L = 2I, S = 10I$ . Compare the solution to the one obtained in 1.

How different would be the affine minimum variance estimator solution?

3. Assuming a GPS measurement covariance matrix of  $R = 100I$  compute the affine minimum variance estimate of the user's position if the position of the user is a random vector with a-priori mean and covariance matrix given by  $\delta \bar{r} = \begin{bmatrix} 0 & 4 & 0 \end{bmatrix}^T, E[(\delta r - \delta \bar{r})(\delta r - \delta \bar{r})^T] = 100I$



**Figure 1 - Problem 1**

### Problem 2

Consider an INS for a navigation platform on top of a vehicle flying above a curved Earth as depicted in figure 2. Neglect the rotational motion of the Earth. Attach an inertial reference frame to Earth and a body reference frame to the vehicle as shown in the figure. Assume that  $x$  represents the distance travelled by the vehicle,  $v$  represents the speed and  $a$  represents the tangential acceleration. Define  $\psi = \phi - \theta$ . Let  $r$  be the radius of the Earth and let  $h$  be the constant altitude above the Earth. Answer the following questions:

1. Write an equation for  $\dot{x}$ , an equation for  $\dot{v}$ , and an equation for  $\dot{\psi} = \dot{\phi} - \dot{\theta}$ . Just use  $x$  to get  $a$ .
2. Write the mechanization equations for  $\hat{x}$ ,  $\hat{v}$ ,  $\hat{\phi}$  as a function of the measurements  $f_{meas}$  and  $\psi_{meas}$ .
3. Write the equation for the acceleration  $a_{xb}$  along the longitudinal x-body axis. Pull the extra term from the lecture notes form
4. Assume that the measurements of the acceleration and gyro have a bias. Assume you have an estimate of both biases. Under these assumptions, write down the equations for the measured quantities  $f_{meas}$  and  $\psi_{meas}$ .
5. Define the errors as  $\delta x = \hat{x} - x$ ,  $\delta v = \hat{v} - v$ ,  $\delta \phi = \hat{\phi} - \phi$  and write down the linearized error equations for each one of these errors.

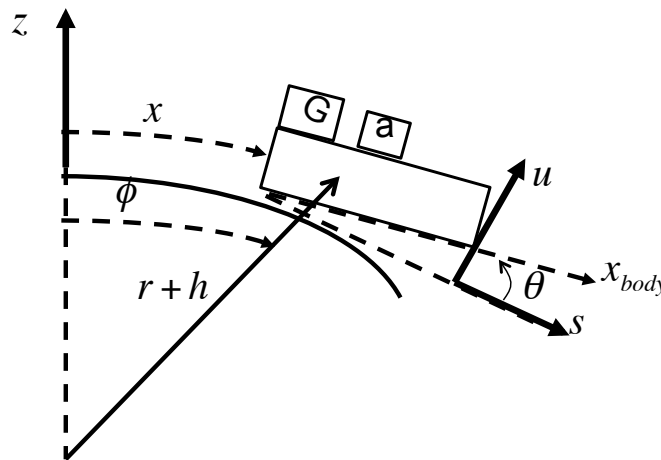


Figure 2 - Problem 2

### Problem 3

Let a measurement be described by the equation

$$y = x + b + n = f(a) + b + n$$

$x = f(a)$   
 $E[f(a)] = f(a)$   
 $E[y] = f(a)$

where  $a$  is a constant vector,  $b$  is a zero mean Gaussian random vector with covariance matrix  $P_b$ ,  $n$  is a zero mean Gaussian random vector with covariance matrix  $P_n$ , and the two random vectors are uncorrelated with each other.  $E[b] = 0$   
 $E[n] = 0$   
The function  $f$  is an invertible and continuously differentiable function. Let  $g(y) = f^{-1}(y)$ .  $f(y)$  not given, keep in function form

1. Compute the mean and the covariance matrix of  $y$ . Should be ok
2. Linearize  $g(y)$  around the point  $f(a)$  and use this linearization to obtain an expression for  $a$  as a function of  $g(y)$ . 1st order Taylor Series
3. Let  $\delta a = a - \hat{a} = a - g(y)$ . Compute the expected value and the covariance matrix of  $\delta a$  using the expression from item 1. Easy once you get 2
4. Compute the expected value and covariance matrix of the vector  $\delta c = \begin{bmatrix} \delta a^T & \delta b^T \end{bmatrix}^T$ .
5. Let  $\rho = Aa$ . Compute the expected value and the covariance matrix of  $\rho$ .

3, 4, 5 is the same procedure from class repeated 3x