

# Stagnation Thermal Boundary Layer

## 1 Overview

In fluid dynamics, stagnation properties are associated with the stagnation point of a fluid flow. The stagnation point is a location in a flow field where the velocity of the fluid is zero. The stagnation properties, often referred to as total or stagnation quantities, represent the conditions a fluid element would experience if it were brought to rest (stagnated) isentropically, i.e., without friction and heat transfer, from the free stream conditions.

## 2 Governing Equation

### 2.1 Derivation

Recall some stagnation properties:

$$h_0 = h + \frac{1}{2}u_i u_i \quad (1)$$

$$h_0 = C_p T_0 \quad (2)$$

Simplified Enthalpy Equation:

$$\rho C_p \left( u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} \right) = u \frac{\partial P}{\partial x} + k \frac{\partial^2 T}{\partial y^2} + \mu \left( \frac{\partial u}{\partial y} \right)^2 \quad (3)$$

Ignoring the forces, the kinetic energy equation is:

$$\begin{aligned} \rho u_i \frac{Du_i}{Dt} &= u_i \frac{\partial \tau_{ji}}{\partial x_j} \\ &= u_i \frac{\partial}{\partial x_j} \left( -P \delta_{ji} + 2\mu S_{ji} - \frac{2}{3} \Delta S_{ji} \right) \\ &= -u_i \frac{\partial P}{\partial x_i} + 2\mu u_i \frac{\partial}{\partial x_j} S_{ji} \end{aligned} \quad (4)$$

Expand:

$$2\mu u_i \frac{\partial}{\partial x_j} S_{ji} = \mu u_i \frac{\partial}{\partial x_j} \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right) \quad (5)$$

Based on the boundary layer approximation, only  $u$  and  $\partial u/\partial y$  matter:

$$2\mu u_i \frac{\partial}{\partial x_j} S_{ji} = \mu u \frac{\partial}{\partial y} \left( \frac{\partial u}{\partial y} \right) \quad (6)$$

Add the enthalpy equation and the kinetic equation together to get the stagnation enthalpy equation:

$$\rho \frac{Dh}{Dt} + \rho \frac{D(\frac{1}{2}u_i u_i)}{Dt} = u \frac{\partial P}{\partial x} + k \frac{\partial^2 T}{\partial y^2} + \mu \left( \frac{\partial u}{\partial y} \right)^2 - u \frac{\partial P}{\partial x} + \mu u \frac{\partial}{\partial y} \left( \frac{\partial u}{\partial y} \right) \quad (7)$$

$$\rho C_p \left[ u \frac{\partial T_0}{\partial x} + v \frac{\partial T_0}{\partial y} \right] = k \frac{\partial^2 T}{\partial y^2} + \mu \frac{\partial}{\partial y} \left( u \frac{\partial u}{\partial y} \right) \quad (8)$$

## 2.2 Unit Prandtl Number Assumption

If  $Pr = 1$ , then:

$$Pr = \frac{\mu C_p}{k} = 1 \quad (9)$$

$$\mu = \frac{k}{C_p} \quad (10)$$

Divide the previous equation by  $C_p$ :

$$\rho \left[ u \frac{\partial T_0}{\partial x} + v \frac{\partial T_0}{\partial y} \right] = \mu \frac{\partial^2 T}{\partial y^2} + \frac{\mu}{C_p} \frac{\partial}{\partial y} \left( u \frac{\partial u}{\partial y} \right) = \mu \frac{\partial^2}{\partial y^2} \left[ T + \frac{\frac{1}{2}u^2}{C_p} \right] \quad (11)$$

Based on the definition of stagnation temperature:

$$T_0 = T \left[ 1 + \frac{1}{2C_p} (u^2 + v^2) \right] \quad (12)$$

From boundary layer assumption,  $u \gg v$ , then:

$$T_0 \approx T \left[ 1 + \frac{u^2}{2C_p} \right] \quad (13)$$

Then with the **unit Prandtl number assumption**:

$$\rho \left[ u \frac{\partial T_0}{\partial x} + v \frac{\partial T_0}{\partial y} \right] = \mu \frac{\partial^2 T_0}{\partial y^2} \quad (14)$$

In this case, the wall can be isothermal (specified temperature) or adiabatic (no heat transferred). From observation,  $T_0 = \text{const}$  is a solution, then:

$$T + \frac{u^2}{2C_p} = \text{const} \quad (15)$$

Differentiating in  $y$ :

$$\frac{\partial T}{\partial y} + \frac{1}{C_p} \left( u \frac{\partial u}{\partial y} \right) = 0 \quad (16)$$

$$C_p \frac{\partial T}{\partial y} = -u \frac{\partial u}{\partial y} \quad (17)$$

At the wall, due to no-slip condition,  $u = 0$ , so  $\partial T / \partial y = 0$ . This solution corresponds to an adiabatic wall.

### 3 Recovery Factor

If  $T_0$  constant and zero heat flux, there is an adiabatic compression at the wall, all kinetic energy is converted into heat. However, the energy conversion is not 100% in real life. Therefore, we define the recovery factor as:

$$r = \frac{T_{ad,w} - T_\infty}{T_{0,\infty} - T_\infty} = \frac{T_{ad,w} - T_\infty}{U_\infty^2 / (2C_p)} \quad (18)$$

Some remarks:

1.  $T_{ad,w}$  is the adiabatic wall temperature, or temperature of the adiabatic wall
2.  $T_{0,\infty}$  is the free-stream stagnation temperature
3. If  $Pr = 1$ ,  $r = 1$ ,  $T_{ad,w} = T_{0,\infty}$