# Computational Aircraft Prototype Syntheses



# Training Session 3.4 CAPS Geometry

#### Marshall Galbraith

Bob Haimes

galbramc@mit.edu haimes@mit.edu Massachusetts Institute of Technology

John F. Dannenhoffer, III

jfdannen@syr.edu Syracuse University

#### caps Overview

- Loading and viewing geometry via pyCAPS
  - loadCAPS
- Accessing/modifying DESPMTR
  - set/getGeometryVal
  - saveGeometry
- Accessing SET and @values using OUTPMTR
  - getGeometryOutVal
- Directing bodies to AIMs
  - Attribute capsAIM
  - Attribute capsIntent
- Suggested Exercises

#### F-118 Boxter Geometry

#### # F-118A Boxster

```
# wing design parameters
DESPMTR
                           4240
          wing:area
                                   # area
DESPMTR
         wing:aspect
                           9.00
                                   # aspect ratio
                                   # thickness ratio
DESPMTR
         wing:thick
                           0.10
DESPMTR
         wing:xroot
                           54.0
                                   # xloc at root LE
DESPMTR
          wing:zroot
                           -5.0
                                   # zloc at root LE
```

#### 

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DESPMTR	htail:aspect	4.15	# htail aspect ratio
DESPMTR	htail:thick	0.08	# htail thickness
DESPMTR	htail:xroot	145	# xloc of root LE
DESPMTR	htail:zroot	5	# zloc of root LE

#### # vertical tail design parameters

DESPMTR	vtail:area	610	# vtail area
DESPMTR	vtail:aspect	1.80	# vtail aspect ratio
DESPMTR	vtail:thick	0.08	# vtail thickness
DESPMTR	vtail:xroot	150	# xloc of root LE
DESPMTR	vtail:zroot	9	# zloc of root LE

#### # fuselage design parameters

DESPMTR	fuse:length	180 #	fuselage length		
DESPMTR	fuse:width	20 #	width	of	fuselage
DESPMTR	fuse:height	20 #	height	of	mid fuselage



```
# set available output parameters
OUTPMTR wing:wet
OUTPMTR wing:volume
OUTPMTR htail:wet
OUTPMTR htail:volume
OUTPMTR vtail:wet
OUTPMTR vtail:volume
OUTPMTR fuse:wet
OUTPMTR fuse:volume
# Wing
                      sgrt(wing:aspect*wing:area)
SET
         wing:span
                        wing:area/wing:span
SET
         wing:chord
```

```
BIP has been initialized and in attached to 'exceeding
esp
                         Hillions, our has been loaded
```

```
BOX wing:xroot -wing:span/2 wing:zroot wing:chord wing:span wing:chord*wing:thick
SELECT body
   ATTRIBUTE _name
                   $Wing
```

```
SET wing:wet
              @area
SET wing:volume @volume
```

#### session3.4/f118\_1\_Geom.py

```
# Allow print statement to be compatible between Python 2 and 3
from __future__ import print_function
# Import pyCAPS class
from pvCAPS import capsProblem
# Initialize capsProblem object
myProblem = capsProblem()
```

- capsProblem provides the context for an CAPS session
  - Multiple capsProblems may be instantiated, but cannot interact

- Geometry loaded with loadCAPS
- Returns pvCAPS.capsGeometry instance
- Visualize with capsViewer using viewGeometry

#### session3.4/f118\_1\_Geom.py

```
# Initialize capsProblem object
mvProblem = capsProblem()
# Load geometry [.csm] file
# loadCAPS returns a class allowing interaction with bodies on the stack
# The geometry is not built with loadCAPS
filename = "f118-A.csm"
print ('\n==> Loading geometry from file "'+filename+'"...')
f118 = myProblem.loadCAPS(filename)
# The same geometry instance is available via myProblem.geometry
assert(f118 == mvProblem.geometry)
# Build and view the geometry with the capsViewer
print ('\n==> Bulding and viewing geometry...')
f118.viewGeometrv()
# Close CAPS
mvProblem.closeCAPS()
```

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• DESPMTR are modified/accessed with set/getGeometryVal

#### session3.4/f118-A.csm

```
# fuselage design parameters
          fuse:length
                                   # fuselage length
DESPMTR
                            180
DESPMTR
          fuse:width
                             20
                                   # width of fuselage
                                   # height of mid fuselage
DESPMTR.
         fuse:height
                             20
```

#### session3.4/f118\_2\_DESPMTR.py

```
# Load geometry [.csm] file
filename = "f118-A.csm"
print ('\n==> Loading geometry from file "'+filename+'"...')
f118 = myProblem.loadCAPS(filename)
# Set wide fuselage
f118.setGeometryVal("fuse:width", 60)
```

• DESPMTR are modified/accessed with set/getGeometryVal

#### session3.4/f118-A.csm

```
# horizontal tail design parameters
DESPMTR.
         htail:area
                         1210
                                # htail area
                         4.15 # htail aspect ratio
DESPMTR
       htail:aspect
DESPMTR
       htail:thick
                         0.08 # htail thickness
DESPMTR
       htail:xroot
                         145 # xloc of root LE
DESPMTR
       htail:zroot
                                # zloc of root LE
```

#### session3.4/f118\_2\_DESPMTR.py

```
# Grow the htail:area
htail_area = f118.getGeometryVal("htail:area")
f118.setGeometryVal("htail:area", htail_area*2)
print ("--> old htail:area = ", htail_area)
print ("--> new htail:area = ", f118.getGeometryVal("htail:area"))
# Build and view the geometry with the capsViewer
print ('\n==> Bulding and viewing geometry...')
f118.viewGeometrv()
```

• DESPMTR are modified/accessed with set/getGeometryVal

#### session3.4/f118\_2\_DESPMTR.py

```
# Build the Canard variant
# Reset the fuselage
f118.setGeometryVal("fuse:width", 20)
htail_area = f118.getGeometryVal("htail:area")
wing area = f118.getGeometryVal("wing:area")
# Swap wing and htail area
f118.setGeometrvVal("htail:area", wing area)
f118.setGeometryVal("wing:area", htail area/2)
# Rebuild and view geometry
print ('\n==> Bulding and viewing geometry...')
f118.viewGeometry()
# Close CAPS
myProblem.closeCAPS()
```

## Saving Geometry with pyCAPS

- Modified geometry can be saved with saveGeometry
  - Available extensions: .egads .stp .step igs .iges .brep

#### session3.4/f118\_2\_DESPMTR.py

```
# Build and view the geometry with the capsViewer
print ('\n==> Bulding and viewing geometry...')
f118.viewGeometry()
```

#### session3.4/f118\_3\_Save.py

```
# Build and save geometry
print ('\n==> Bulding and saving geometry...')
f118.saveGeometry("f118_3_Save_Wide.egads")
```

- View geometry with:
  - serveCSM f118\_3\_Save\_Wide.csm
  - serveCSM f118\_3\_Save\_Canard.csm

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# Accessing Geometry OUTPMTR with pyCAPS

• OUTPMTR values are accessed with getGeometryOutVal

### session3.4/f118-A.csm

```
# set available output parameters
OUTPMTR wing:wet
OUTPMTR wing:volume
BOX wing:xroot -wing:span/2 wing:zroot wing:chord wing:span wing:chord*wing:thick
SELECT body
   ATTRIBUTE _name $Wing
SET wing:wet
             @area
SET wing:volume @volume
```

#### session3.4/f118\_4\_OUTPMTR.py

```
# Load geometry [.csm] file
filename = "f118-A.csm"
print ('\n==> Loading geometry from file "'+filename+'"...')
f118 = myProblem.loadCAPS(filename)
# Build and print all available output parameters
print ("--> wing:wet =", f118.getGeometryOutVal("wing:wet"
print ("--> wing:volume =", f118.getGeometryOutVal("wing:volume")))
```

# Accessing Geometry OUTPMTR with pyCAPS

#### • OUTPMTR not set return None

#### session 3.4/f118-A.csm

OUTPMTR fuse:wet.

```
OUTPMTR fuse:volume

BOX 0 -fuse:width/2 -fuse:height/2 fuse:length fuse:width fuse:height

SELECT body

ATTRIBUTE name $Fuselage
```

# fuse:wet and fuse:volume not set

#### session3.4/f118\_4\_OUTPMTR.py

```
# Accessing OUTPMTR that has not been set
print ("--> fuse:wet =", fil8.getGeometryOutVal("fuse:wet" ) )
print ("--> fuse:volume =", fil8.getGeometryOutVal("fuse:volume" ) )
```

# Accessing Geometry OUTPMTR with pyCAPS

• Accessing non-OUTPMTR gives CAPS\_NOTFOUND error

#### session 3.4/f118-A.csm

```
# set available output parameters
OUTPMTR wing:wet
OUTPMTR wing:volume
OUTPMTR
        htail:wet
       htail:volume
OUTPMTR.
OUTPMTR
       vtail:wet
OUTPMTR
        vtail:volume
OTITPMTR
       fuse:wet
OUTPMTR fuse:volume
# Wing
SET
          wing:span
                         sqrt(wing:aspect*wing:area)
                         wing:area/wing:span
SET
          wing:chord
```

#### session3.4/f118\_4\_OUTPMTR.py

```
# Attempt to get a SET value not defined as OUTPMTR
print ("--> wing:span =", f118.getGeometryOutVal("wing:span" ))
```

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  - Attribute capsIntent
- Suggested Exercises

- - capsAIM attribute
    - String semicolon separated AIM names
    - AIMs suitable to use the body
  - capsIntent attribute
    - Optional string used to direct bodies to AIMs
    - String semicolon separated AIM names
    - Multiple bodies may have the same capsIntent

#### session3.4/f118-B.csm

```
# Htail
SET
         htail:span
                         sqrt(htail:aspect*htail:area)
         htail:chord
                         htail:area/htail:span
BOX htail:xroot -htail:span/2 htail:zroot htail:chord htail:span htail:chord*htail:thick
SELECT body
    ATTRIBUTE capsAIM
                        $masstranAIM:astrosAIM
    ATTRIBUTE capsIntent $htail; tail
    ATTRIBUTE name
                        $Htail
```

- Loading AIM with loadAIM
  - aim: The name of the AIM to load
  - analysisDir: Directory to write files if any (must be unique)
  - altName: Alternative name to track multiple instances of the same type of AIM
- No capsIntent loads the all bodies with matching capsAIM

#### session $3.4/f118_5$ \_AIM.py

```
# capsAIM == $masstranAIM
masstranAll = myProblem.loadAIM(aim = "masstranAIM",
                                analysisDir="masstranALL", altName="All")
# The AIM instance is also available in the capsProblem.analysis dict
assert(masstranAll == myProblem.analysis["All"])
# Show the geometry used by the AIM
print("==> Geometry used by masstranAll instance with no capsIntent")
masstranAll.viewGeometrv()
```

# Wing

• Loading masstranAIM with bodies capsAIM == \$masstranAIM and capsIntent == \$wing

#### session3.4/f118-B.csm

```
SET
         wing:span
                        sqrt(wing:aspect*wing:area)
SET
                        wing:area/wing:span
         wing:chord
BOX wing:xroot -wing:span/2 wing:zroot wing:chord wing:span wing:chord*wing:thick
SELECT body
   ATTRIBUTE capsAIM
                        $masstranAIM:astrosAIM
   ATTRIBUTE capsIntent $wing
   ATTRIBUTE _name
                        $Wing
```

#### session3.4/f118\_5\_AIM.py

```
# capsAIM == $masstranAIM and capsIntent == $wing
myProblem.loadAIM(aim = "masstranAIM", capsIntent="wing",
                  analysisDir="masstranWing", altName="Wing")
# Show the geometry used by the AIM
print("==> Geometry used by Wing instance with capsIntent='wing'")
myProblem.analysis["Wing"].viewGeometry()
```

• Loading masstranAIM with bodies capsAIM == \$masstranAIM and capsIntent == \$tail

#### session3.4/f118-B.csm

```
BOX htail:xroot -htail:span/2 htail:zroot htail:chord htail:span htail:chord*htail:thick
SELECT body
   ATTRIBUTE capsAIM
                        $masstranAIM:astrosAIM
   ATTRIBUTE capsIntent $htail;tail
   ATTRIBUTE _name
                        $Htail
```

```
BOX vtail:xroot 0 vtail:zroot
                                  vtail:chord vtail:chord*vtail:thick vtail:span
SELECT body
   ATTRIBUTE capsAIM
                        $masstranAIM;astrosAIM
   ATTRIBUTE capsIntent $vtail; tail
   ATTRIBUTE name
                        $Vtail
```

#### session $3.4/f118_5$ AIM.py

```
# capsAIM == $masstranAIM and capsIntent == $tail
masstranTail = myProblem.loadAIM(aim = "masstranAIM", capsIntent="tail",
                                 analysisDir="masstranTail", altName="Tail")
```

• Loading masstranAIM with bodies capsAIM == \$masstranAIM and (capsIntent == \$wing or capsIntent == \$fuse)

#### session3.4/f118-B.csm

```
BOX wing:xroot -wing:span/2 wing:zroot wing:chord wing:span wing:chord*wing:thick
SELECT body
                        $masstranAIM;astrosAIM
   ATTRIBUTE capsAIM
   ATTRIBUTE capsIntent $wing
   ATTRIBUTE name
                        $Wing
BOX 0 -fuse:width/2 -fuse:height/2
                                      fuse:length fuse:width fuse:height
SELECT body
   ATTRIBUTE capsAIM
                        $masstranAIM;astrosAIM
   ATTRIBUTE capsIntent $fuse
   ATTRIBUTE name
                        $Fuselage
```

#### session $3.4/f118_5$ -AIM.py

```
# capsAIM == $masstranAIM and (capsIntent == $wing or capsIntent == $fuse)
myProblem.loadAIM(aim = "masstranAIM", capsIntent=["wing", "fuse"],
                  analysisDir="masstranWingFuse", altName="WingFuse")
```

# Suggested Exercises

#### Fix f118-B.csm

- SET fuse:wet and fuse:volume in session3.4/f118-B.csm
- Add wing:span as OUTPMTR in session3.4/f118-B.csm
- Rerun session3.4/f118\_4\_OUTPMTR.py

#### Custom f118-A.csm

- Customize the f118-A.csm with setGeometryVal
  - Start from a copy of session3.4/f118\_2\_DESPMTR.py

## Custom masstran analysis

- Load wing, htail and fuselage into a masstranAIM
  - Start from a copy of session3.4/f118\_5\_AIM.py
- Create your own