

Computational Aircraft Prototype Syntheses



Training Session 2

CAPS Geometry

ESP v1.18

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- Loading and viewing geometry via pyCAPS
 - loadCAPS
- Accessing/modifying DESPMTR
 - set/getGeometryVal
 - saveGeometry
- Accessing SET and @values using OUTPMTR
 - getGeometryOutVal
- Directing bodies to AIMs
 - Attribute capsAIM
 - Attribute capsIntent
- Suggested Exercises

session02/f118-A.csm

F-118A Boxster

wing design parameters

DESPMTR	wing:area	4240	# area
DESPMTR	wing:aspect	9.00	# aspect ratio
DESPMTR	wing:thick	0.10	# thickness ratio
DESPMTR	wing:xroot	54.0	# xloc at root LE
DESPMTR	wing:zroot	-5.0	# zloc at root LE

horizontal tail design parameters

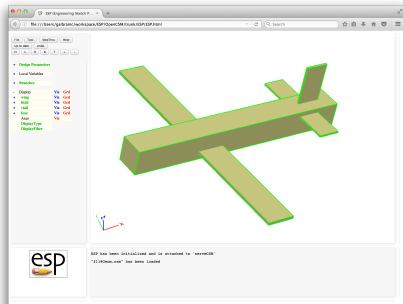
DESPMTR	htail:area	1210	# htail area
DESPMTR	htail:aspect	4.15	# htail aspect ratio
DESPMTR	htail:thick	0.08	# htail thickness
DESPMTR	htail:xroot	145	# xloc of root LE
DESPMTR	htail:zroot	5	# zloc of root LE

vertical tail design parameters

DESPMTR	vtail:area	610	# vtail area
DESPMTR	vtail:aspect	1.80	# vtail aspect ratio
DESPMTR	vtail:thick	0.08	# vtail thickness
DESPMTR	vtail:xroot	150	# xloc of root LE
DESPMTR	vtail:zroot	9	# zloc of root LE

fuselage design parameters

DESPMTR	fuse:length	180	# fuselage length
DESPMTR	fuse:width	20	# width of fuselage
DESPMTR	fuse:height	20	# height of mid fuselage



session02/f118-A.csm

```

#-----
# set available output parameters
OUTPMTR wing:wet
OUTPMTR wing:volume

OUTPMTR htail:wet
OUTPMTR htail:volume

OUTPMTR vtail:wet
OUTPMTR vtail:volume

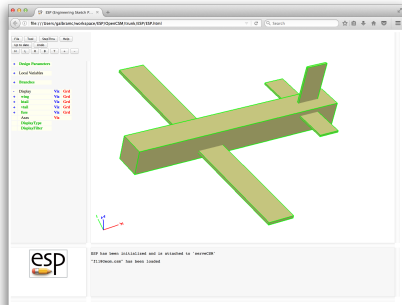
OUTPMTR fuse:wet
OUTPMTR fuse:volume

#=====
# Wing
SET wing:span      sqrt(wing:aspect*wing:area)
SET wing:chord     wing:area/wing:span

BOX wing:xroot -wing:span/2 wing:zroot wing:chord wing:span wing:chord*wing:thick
SELECT body
  ATTRIBUTE _name $Wing

SET wing:wet @area
SET wing:volume @volume

```



session02/f118_1_Geom.py

```
#-----#  
  
# Allow print statement to be compatible between Python 2 and 3  
from __future__ import print_function  
  
# Import pyCAPS class  
from pyCAPS import capsProblem  
  
#-----#  
  
# Initialize capsProblem object  
myProblem = capsProblem()
```

- capsProblem provides the context for an CAPS session
 - Multiple capsProblems may be instantiated, but cannot interact

- Geometry loaded with loadCAPS
- Returns pyCAPS.capsGeometry instance
- Visualize with capsViewer using viewGeometry

session02/f118_1_Geom.py

```
# Initialize capsProblem object
myProblem = capsProblem()

# Load geometry [.csm] file
# loadCAPS returns a class allowing interaction with bodies on the stack
# The geometry is not built with loadCAPS
filename = "f118-A.csm"
print ('\n==> Loading geometry from file "' + filename + '"...')
f118 = myProblem.loadCAPS(filename)

# The same geometry instance is available via myProblem.geometry
assert(f118 == myProblem.geometry)

# Build and view the geometry with the capsViewer
print ('\n==> Building and viewing geometry...')
f118.viewGeometry()

# Close CAPS (optional)
myProblem.closeCAPS()
```

- Loading and viewing geometry via pyCAPS
 - loadCAPS
- Accessing/modifying DESPMTR
 - set/getGeometryVal
 - saveGeometry
- Accessing SET and @values using OUTPMTR
 - getGeometryOutVal
- Directing bodies to AIMs
 - Attribute capsAIM
 - Attribute capsIntent
- Suggested Exercises

- DESPMTR are modified/accessed with set/getGeometryVal

session02/f118-A.csm

```
# fuselage design parameters
DESPMTR   fuse:length      180   # fuselage length
DESPMTR   fuse:width       20    # width of fuselage
DESPMTR   fuse:height      20    # height of mid fuselage
```

session02/f118_2_DESPMTR.py

```
# Load geometry [.csm] file
filename = "f118-A.csm"
print ('\n==> Loading geometry from file "' + filename + '"...')
f118 = myProblem.loadCAPS(filename)

# Set wide fuselage
f118.setGeometryVal("fuse:width", 60)
```

- DESPMTR are modified/accessed with set/getGeometryVal

session02/f118-A.csm

```
# horizontal tail design parameters
DESPMTR   htail:area      1210   # htail area
DESPMTR   htail:aspect    4.15   # htail aspect ratio
DESPMTR   htail:thick     0.08   # htail thickness
DESPMTR   htail:xroot     145    # xloc of root LE
DESPMTR   htail:zroot     5      # zloc of root LE
```

session02/f118_2_DESPMTR.py

```
# Double the htail:area
htail_area = f118.getGeometryVal("htail:area")
f118.setGeometryVal("htail:area", htail_area*2)

print ("--> old htail:area = ", htail_area)
print ("--> new htail:area = ", f118.getGeometryVal("htail:area"))

# Build and view the geometry with the capsViewer
print ('\n==> Building and viewing geometry...')
f118.viewGeometry()
```

- DESPMTR are modified/accessed with set/getGeometryVal

session02/f118_2_DESPMTR.py

```
# Build the Canard variant

# Reset the fuselage
f118.setGeometryVal("fuse:width", 20)

htail_area = f118.getGeometryVal("htail:area")
wing_area  = f118.getGeometryVal("wing:area")

# Swap wing and htail area
f118.setGeometryVal("htail:area", wing_area)
f118.setGeometryVal("wing:area", htail_area/2)

# Rebuild and view geometry
print ('\n==> Bulding and viewing geometry...')
f118.viewGeometry()
```

- Modified geometry can be saved with saveGeometry
 - Available extensions: .egads .stp .step .igs .iges .brep

session02/f118_2_DESPMTR.py

```
# Build and view the geometry with the capsViewer
print ('\n==> Bulding and viewing geometry...')
f118.viewGeometry()
```

session02/f118_3_Save.py

```
# Build and save geometry
print ('\n==> Bulding and saving geometry...')
f118.saveGeometry("f118_3_Save_Wide.egads")
```

- View geometry with:
 - serveCSM f118_3_Save_Wide.egads
 - serveCSM f118_3_Save_Canard.egads

- Loading and viewing geometry via pyCAPS
 - loadCAPS
- Accessing/modifying DESPMTR
 - set/getGeometryVal
 - saveGeometry
- Accessing SET and @values using OUTPMTR
 - getGeometryOutVal
- Directing bodies to AIMs
 - Attribute capsAIM
 - Attribute capsIntent
- Suggested Exercises

- OUTPMTR values are accessed with getGeometryOutVal

session02/f118-A.csm

```
#-----  
# set available output parameters  
OUTPMTR wing:wet  
OUTPMTR wing:volume  
  
BOX wing:xroot -wing:span/2 wing:zroot wing:chord wing:span wing:chord*wing:thick  
SELECT body  
    ATTRIBUTE _name $Wing  
  
SET wing:wet @area  
SET wing:volume @volume
```

session02/f118_4_OUTPMTR.py

```
# Load geometry [.csm] file  
filename = "f118-A.csm"  
print ('\n==> Loading geometry from file "' + filename + '"...')  
f118 = myProblem.loadCAPS(filename)  
  
# Build and print all available output parameters  
print ("--> wing:wet    =", f118.getGeometryOutVal("wing:wet"    ) )  
print ("--> wing:volume =", f118.getGeometryOutVal("wing:volume" ) )
```

- **None** returned for any OUTPMTR not SET

session02/f118-A.csm

```
OUTPMTR fuse:wet
OUTPMTR fuse:volume
```

```
BOX 0 -fuse:width/2 -fuse:height/2 fuse:length fuse:width fuse:height
SELECT body
  ATTRIBUTE _name $Fuselage
```

```
# fuse:wet and fuse:volume not set
```

session02/f118_4_OUTPMTR.py

```
# Accessing OUTPMTR that has not been set
print ("--> fuse:wet    =", f118.getGeometryOutVal("fuse:wet"    ) )
print ("--> fuse:volume =", f118.getGeometryOutVal("fuse:volume" ) )
```

- Accessing non-OUTPMTR gives CAPS_NOTFOUND error

session02/f118-A.csm

```
#-----  
# set available output parameters  
OUTPMTR wing:wet  
OUTPMTR wing:volume  
  
OUTPMTR htail:wet  
OUTPMTR htail:volume  
  
OUTPMTR vtail:wet  
OUTPMTR vtail:volume  
  
OUTPMTR fuse:wet  
OUTPMTR fuse:volume  
  
#=====
```

```
# Wing  
SET      wing:span      sqrt(wing:aspect*wing:area)  
SET      wing:chord     wing:area/wing:span
```

session02/f118_4_OUTPMTR.py

```
# Attempt to get a SET value not defined as OUTPMTR  
print ("--> wing:span   =", f118.getGeometryOutVal("wing:span" ) )
```

- Loading and viewing geometry via pyCAPS
 - loadCAPS
- Accessing/modifying DESPMTR
 - set/getGeometryVal
 - saveGeometry
- Accessing SET and @values using OUTPMTR
 - getGeometryOutVal
- Directing bodies to AIMs
 - Attribute capsAIM
 - Attribute capsIntent
- Suggested Exercises

- capsAIM attribute
 - String semicolon separated AIM names
 - AIMs suitable to use the body
- capsIntent attribute
 - Optional string used to direct bodies to AIM instance
 - String semicolon separated names
 - Multiple bodies may have the same capsIntent

session02/f118-B.csm

```
# Htail
SET      htail:span      sqrt(htail:aspect*htail:area)
SET      htail:chord     htail:area/htail:span

BOX      htail:xroot     -htail:span/2  htail:zroot  htail:chord  htail:span  htail:chord*htail:thick
SELECT   body
  ATTRIBUTE capsAIM      $masstranAIM;astrosAIM
  ATTRIBUTE capsIntent   $htail;tail
  ATTRIBUTE _name        $Htail
```

- Loading AIM with loadAIM
 - aim: The name of the AIM to load
 - analysisDir: Directory to write files if any (must be unique)
 - altName: Alternative name to track multiple instances of the same type of AIM
- No capsIntent loads the all bodies with matching capsAIM

session02/f118_5_AIM.py

```
# capsAIM == $masstranAIM
masstranAll = myProblem.loadAIM(aim = "masstranAIM",
                                analysisDir="masstranALL", altName="All")

# The AIM instance is also available in the capsProblem.analysis dict
assert(masstranAll == myProblem.analysis["All"])

# Show the geometry used by the AIM
print("==> Geometry used by masstranAll instance with no capsIntent")
masstranAll.viewGeometry()
```

- Loading masstranAIM with bodies
capsAIM == \$masstranAIM and capsIntent == \$wing

session02/f118-B.csm

```
# Wing
SET      wing:span      sqrt(wing:aspect*wing:area)
SET      wing:chord     wing:area/wing:span

BOX      wing:xroot     -wing:span/2  wing:zroot  wing:chord  wing:span  wing:chord*wing:thick
SELECT   body
  ATTRIBUTE capsAIM      $masstranAIM;astrosAIM
  ATTRIBUTE capsIntent   $wing
  ATTRIBUTE _name        $Wing
```

session02/f118_5_AIM.py

```
# capsAIM == $masstranAIM and capsIntent == $wing
myProblem.loadAIM(aim = "masstranAIM", capsIntent="wing",
                  analysisDir="masstranWing", altName="Wing")

# Show the geometry used by the AIM
print("==> Geometry used by Wing instance with capsIntent='wing'")
myProblem.analysis["Wing"].viewGeometry()
```

- Loading masstranAIM with bodies
`capsAIM == $masstranAIM` and `capsIntent == $tail`

session02/f118-B.csm

```
BOX htail:xroot -htail:span/2 htail:zroot htail:chord htail:span htail:chord*htail:thick
SELECT body
  ATTRIBUTE capsAIM $masstranAIM;astrosAIM
  ATTRIBUTE capsIntent $htail;tail
  ATTRIBUTE _name $Htail
```

```
BOX vtail:xroot 0 vtail:zroot vtail:chord vtail:chord*vtail:thick vtail:span
SELECT body
  ATTRIBUTE capsAIM $masstranAIM;astrosAIM
  ATTRIBUTE capsIntent $vtail;tail
  ATTRIBUTE _name $Vtail
```

session02/f118_5_AIM.py

```
# capsAIM == $masstranAIM and capsIntent == $tail
masstranTail = myProblem.loadAIM(aim = "masstranAIM", capsIntent="tail",
                                analysisDir="masstranTail", altName="Tail")
```

- Loading masstranAIM with bodies
capsAIM == \$masstranAIM and
(capsIntent == \$wing or capsIntent == \$fuse)

session02/f118-B.csm

```
BOX wing:xroot -wing:span/2 wing:zroot wing:chord wing:span wing:chord*wing:thick
SELECT body
  ATTRIBUTE capsAIM $masstranAIM;astrosAIM
  ATTRIBUTE capsIntent $wing
  ATTRIBUTE _name $Wing
```

```
BOX 0 -fuse:width/2 -fuse:height/2 fuse:length fuse:width fuse:height
SELECT body
  ATTRIBUTE capsAIM $masstranAIM;astrosAIM
  ATTRIBUTE capsIntent $fuse
  ATTRIBUTE _name $Fuselage
```

session02/f118_5_AIM.py

```
# capsAIM == $masstranAIM and (capsIntent == $wing or capsIntent == $fuse)
myProblem.loadAIM(aim = "masstranAIM", capsIntent=["wing","fuse"],
  analysisDir="masstranWingFuse", altName="WingFuse")
```

Fix f118-B.csm

- SET fuse:wet and fuse:volume in session02/f118-B.csm
- Add wing:span as OUTPMTR in session02/f118-B.csm
- Rerun session02/f118_4_OUTPMTR.py

Custom f118-A.csm

- Customize the f118-A.csm with setGeometryVal
 - Start from a copy of session02/f118_2_DESPMTR.py

Custom masstran analysis

- Load wing, htail and fuselage into a masstranAIM
 - Start from a copy of session02/f118_5_AIM.py

- Create your own