
EEEN 460

Optimal Control

Spring 2020

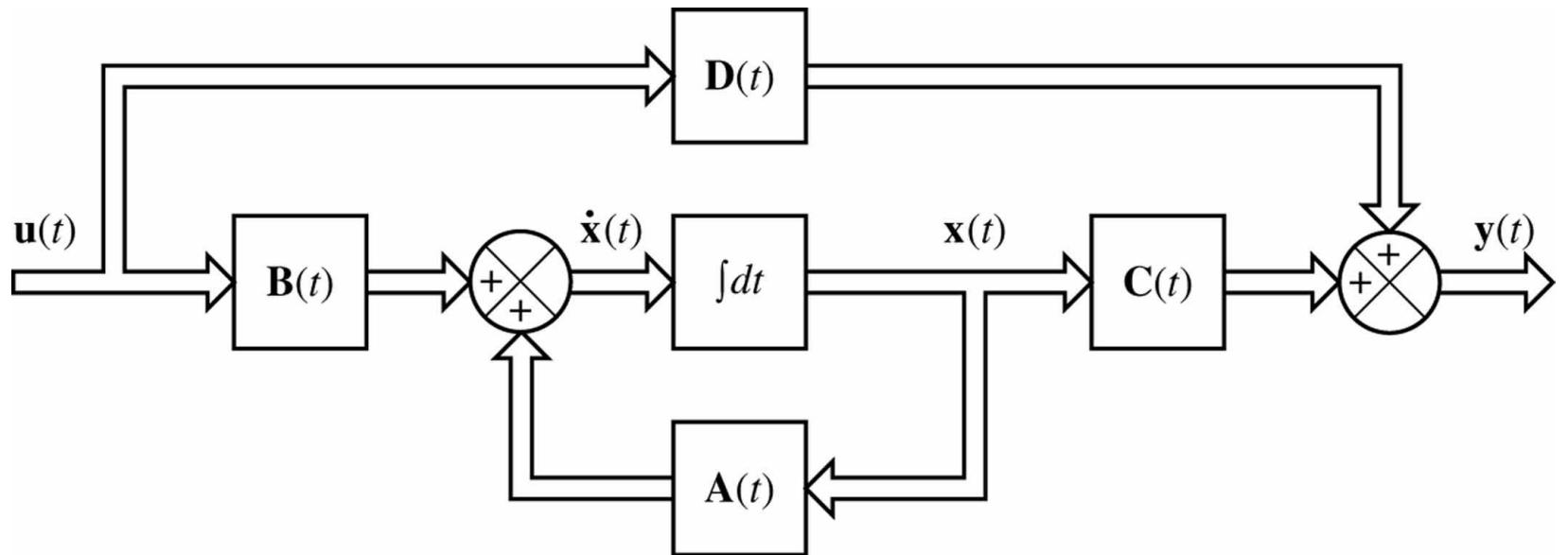
Lecture 6

Controller Design by Pole Placement

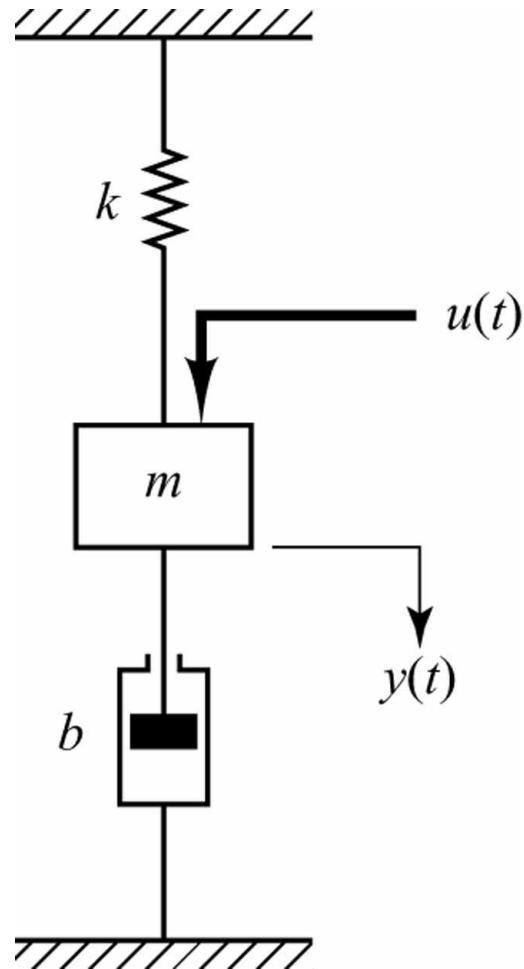
Controller Design by LQR

State space Representation

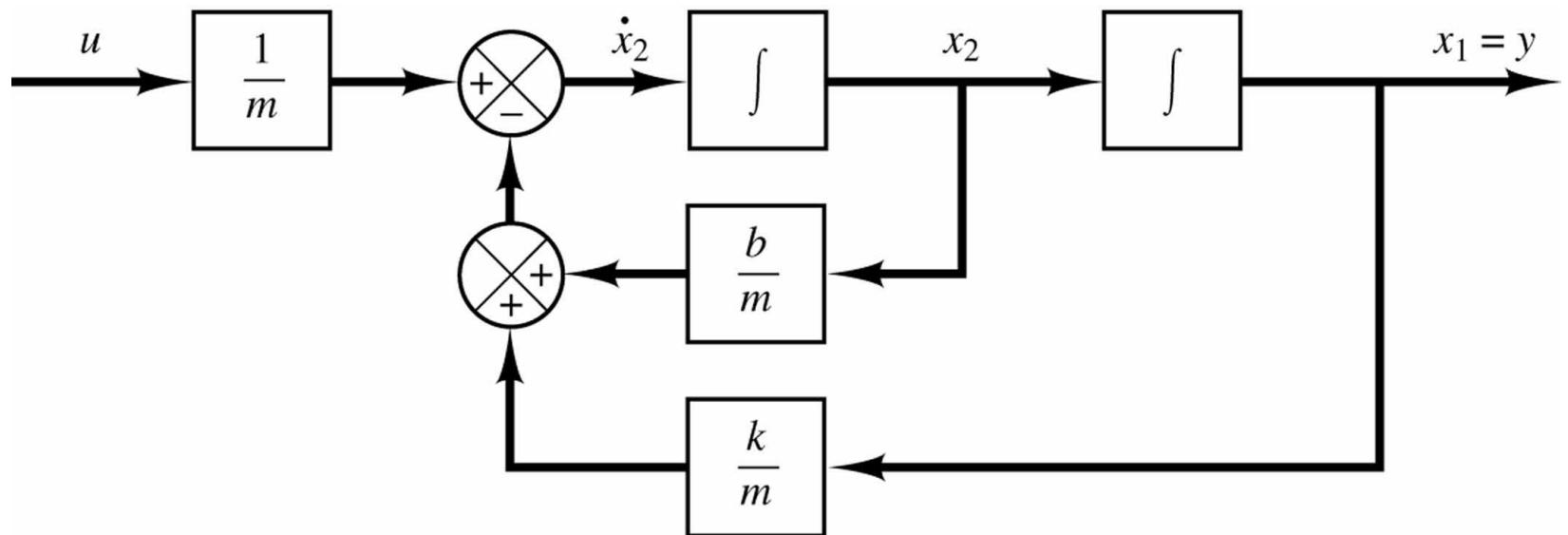
Block diagram of the linear, continuous-time control system represented in state space.



Spring-Mass-Damper system

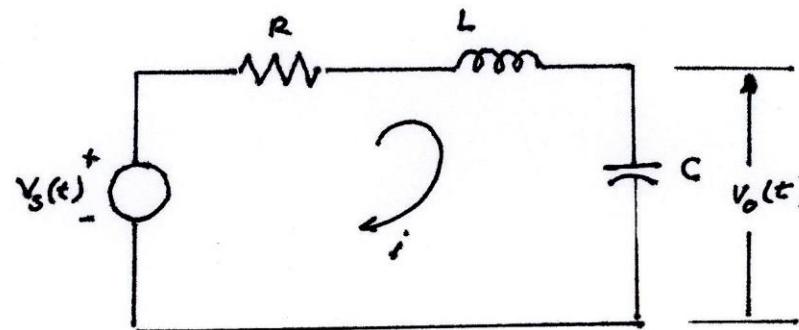


State-Space Representation of the Spring-Mass-Damper system



STATE - SPACE METHOD

Example 1.



$$\begin{aligned} R &= 1 \Omega \\ L &= 1/4 H \\ C &= 4/3 F \end{aligned}$$

$$i = i_L = i_C = C \frac{dv}{dt} \quad] \text{ state equations}$$

$$Ri_L + L \frac{di_L}{dt} + V_C = V_s$$

$$V_o = V_C \quad] \text{ output equation}$$

$$\frac{di_L}{dt} = -\frac{R}{L} i_L - \frac{1}{L} V_C + \frac{1}{L} V_s$$

$$\frac{dV_C}{dt} = \frac{1}{C} i_L$$

$$\begin{array}{c|c|c|c|c|c|c|c} \left| \frac{di_L}{dt} \right| & | & -\frac{R}{L} & -\frac{1}{L} & | & i_L & | & \frac{1}{L} \\ \left| \frac{dv_C}{dt} \right| & = & \frac{1}{C} & 0 & | & v_C & + & 0 \\ v_o & = & 0 & 1 & | & i_L & | & v_S \end{array}$$

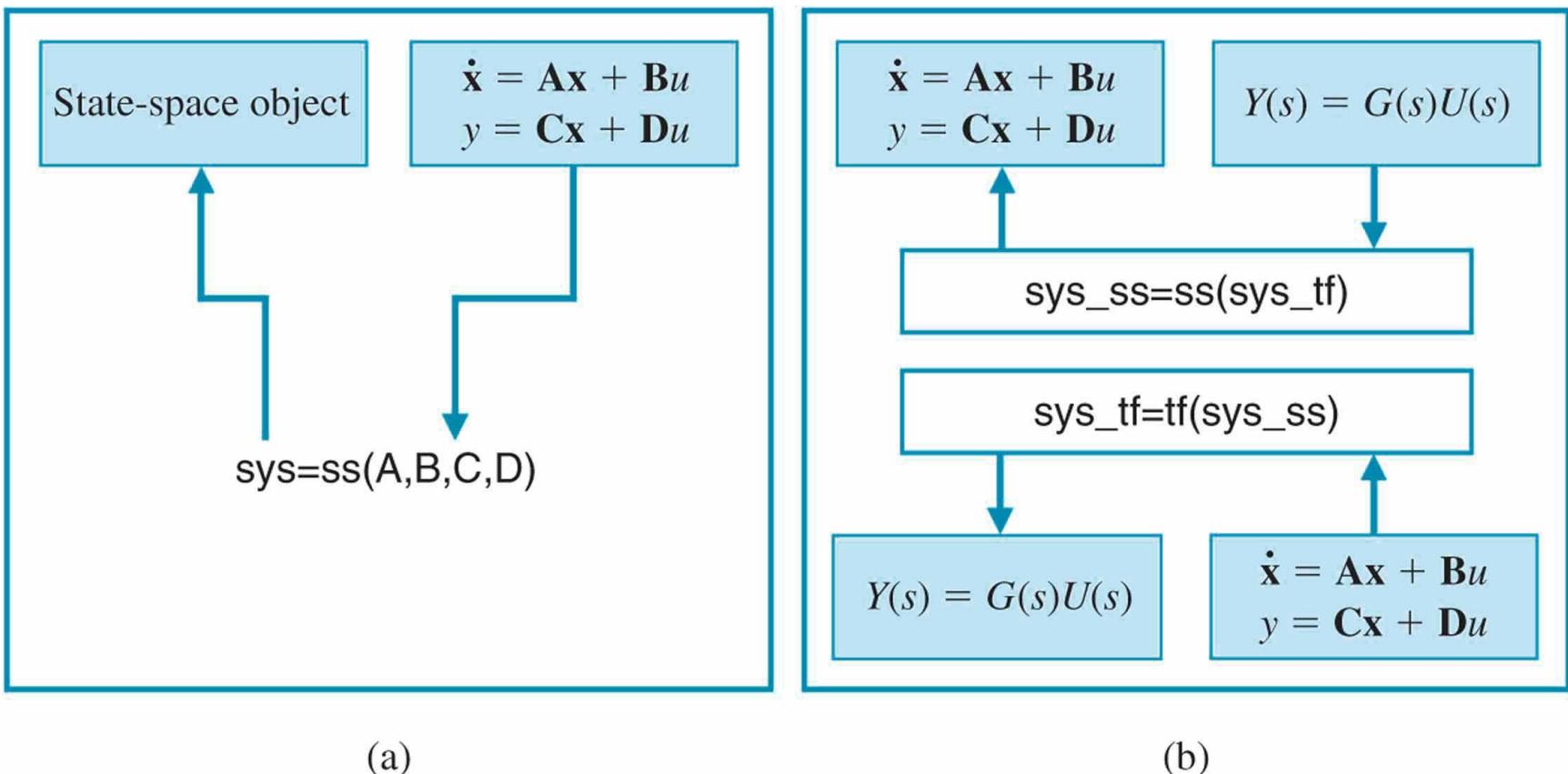
$$i_L = x_1$$

$$v_C = x_2$$

$$\begin{array}{c|c|c|c|c|c|c|c} \dot{x}_1 & | & -4 & -3 & | & x_1 & | & 4 \\ \dot{x}_2 & = & \frac{3}{4} & 0 & | & x_2 & + & 0 \\ y = & | & 0 & 1 & | & \begin{pmatrix} x_1 \\ x_2 \end{pmatrix} & + & \begin{pmatrix} 0 \end{pmatrix} u_o \end{array}$$

Conversion between linear system models

(a) The **ss** function. (b) Linear system model conversion.



(a)

(b)

Conversion from transfer function to a state-space representation

convert.m

```
% Convert G(s) = (2s^2+8s+6)/(s^3+8s^2+16s+6)
% to a state-space representation
%
num=[2 8 6]; den=[1 8 16 6]; sys_tf=tf(num,den);
sys_ss=ss(sys_tf);
```

(a)

```
>>convert
a =
          x1           x2           x3
x1      -8           -4         -1.5
x2       4            0           0
x3       0            1           0

b =
          u1
x1       2
x2       0
x3       0

c =
          x1           x2           x3
y1        1            1         0.75

d =
          u1
y1        0
```

(b)

Conversion from state-space representation to a transfer function

Find the transfer function of the
following system

$$\begin{vmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{vmatrix} = \begin{vmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -80 & -28 & -7 \end{vmatrix} \begin{vmatrix} x_1 \\ x_2 \\ x_3 \end{vmatrix} + \begin{vmatrix} 0 \\ 1 \\ -14 \end{vmatrix} \begin{vmatrix} u_1 \\ u_2 \\ u_3 \end{vmatrix}$$

$$y = \begin{vmatrix} 1 & 1 & 0 & 0 \end{vmatrix} \begin{vmatrix} x_1 \\ x_2 \\ x_3 \end{vmatrix}$$

$$[num, den] = ss2tf(A, B, C, D)$$

Conversion from state-space representation to a transfer function

```
%Finding the transfer function  
%System is represented in State-Space  
%**** Enter matrices A,B,C and D  
A=[0 1 0;0 0 1;-80 -28 -7];  
B=[0; 1;-14];  
C=[1 0 0];  
D=[0];  
[num,den]=ss2tf(A,B,C,D)
```

```
num =  
  
          0      0    1.0000   -7.0000  
  
den =  
  
     1.0000   7.0000   28.0000   80.0000
```

Computing the step response

Obtain the unit step response
of the system

$$\begin{vmatrix} \dot{x}_1 \\ \dot{x}_2 \end{vmatrix} = \begin{vmatrix} 0 & 1 \\ -1 & -0.5 \end{vmatrix} \begin{vmatrix} x_1 \\ x_2 \end{vmatrix} + \begin{vmatrix} 0 \\ 1 \end{vmatrix} u$$

$$y = \begin{vmatrix} 1 & 0 \end{vmatrix} \begin{vmatrix} x_1 \\ x_2 \end{vmatrix}$$

We put this system into following
form

$$\begin{vmatrix} \dot{x}_1 \\ \dot{x}_2 \end{vmatrix} = \underbrace{\begin{vmatrix} 0 & 1 \\ -1 & -0.5 \end{vmatrix}}_A \begin{vmatrix} x_1 \\ x_2 \end{vmatrix} + \underbrace{\begin{vmatrix} 0 & 0 \\ 1 & 0 \end{vmatrix}}_B \begin{vmatrix} u_1 \\ u_2 \end{vmatrix}$$

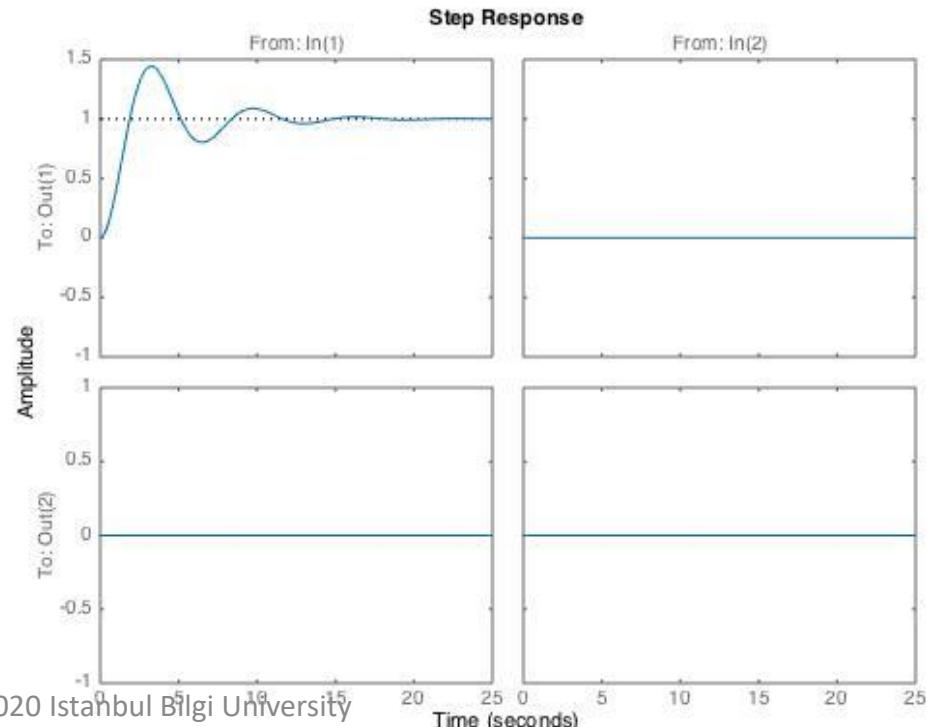
$$y = \underbrace{\begin{vmatrix} 1 & 0 \\ 0 & 0 \end{vmatrix}}_C \begin{vmatrix} x_1 \\ x_2 \end{vmatrix} + \underbrace{\begin{vmatrix} 0 & 0 \\ 0 & 0 \end{vmatrix}}_D \begin{vmatrix} u_1 \\ u_2 \end{vmatrix}$$

step (A, B, C, D)

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Computing the step response

```
%Unit Step Response  
%System is represented in State-Space  
%**** Enter matrices A,B,C and D  
A=[0 1;-1 -0.5];  
B=[0 0;1 0];  
C=[1 0;0 0];  
D=[0 0;0 0];  
%Calculate the response and plot  
step(A,B,C,D);
```



Computing the step response

%Unit Step Response

%System is represented in State-Space

%**** Enter matrices A,B,C and D

A=[0 1;-1 -0.5];

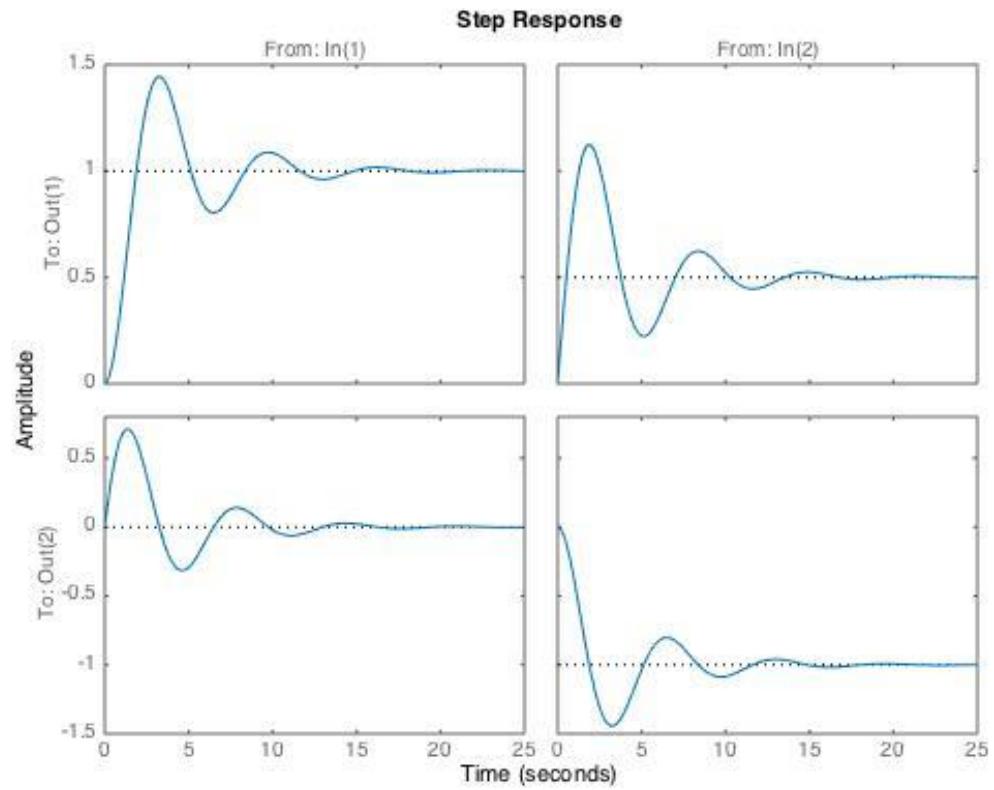
B=[0 1;1 0];

C=[1 0;0 1];

D=[0 0;0 0];

%Calculate the response and plot

step(A,B,C,D);



Plotting the root locus diagram

Plotting the Root Locus

$$\begin{vmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{vmatrix} = \begin{vmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -80 & -28 & -7 \end{vmatrix} \begin{vmatrix} x_1 \\ x_2 \\ x_3 \end{vmatrix} + \begin{vmatrix} 0 \\ 1 \\ -14 \end{vmatrix} \begin{vmatrix} u_1 \\ u_2 \\ u_3 \end{vmatrix}$$

$$y = \begin{vmatrix} 1 & 0 & 0 \end{vmatrix} \begin{vmatrix} x_1 \\ x_2 \\ x_3 \end{vmatrix}$$

The root locus command

`rlocus (A, B, C, D)`

```
%Root Locus
```

```
%System is represented in State-Space
```

```
%**** Enter matrices A,B,C and D
```

```
A=[0 1 0;0 0 1;-80 -28 -7];
```

```
B=[0; 1;-14];
```

```
C=[1 0 0];
```

```
D=[0];
```

```
%K varies from 0 to 400
```

```
K=0:0.1:400;
```

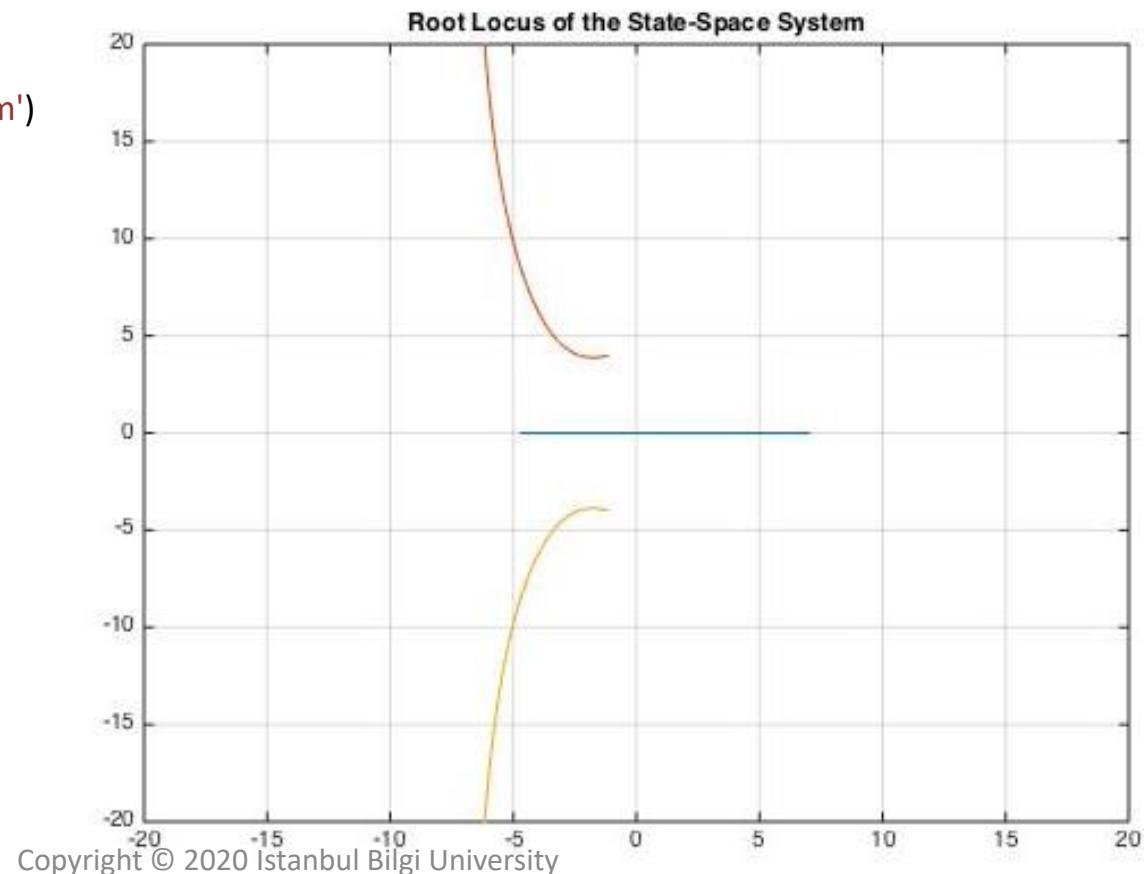
```
% plot the root locus
```

```
r=rlocus(A,B,C,D);
```

```
plot(r,'-');v=[-20 20 -20 20];axis(v)
```

```
grid
```

```
title('Root Locus of the State-Space System')
```



Plotting the Nyquist diagram

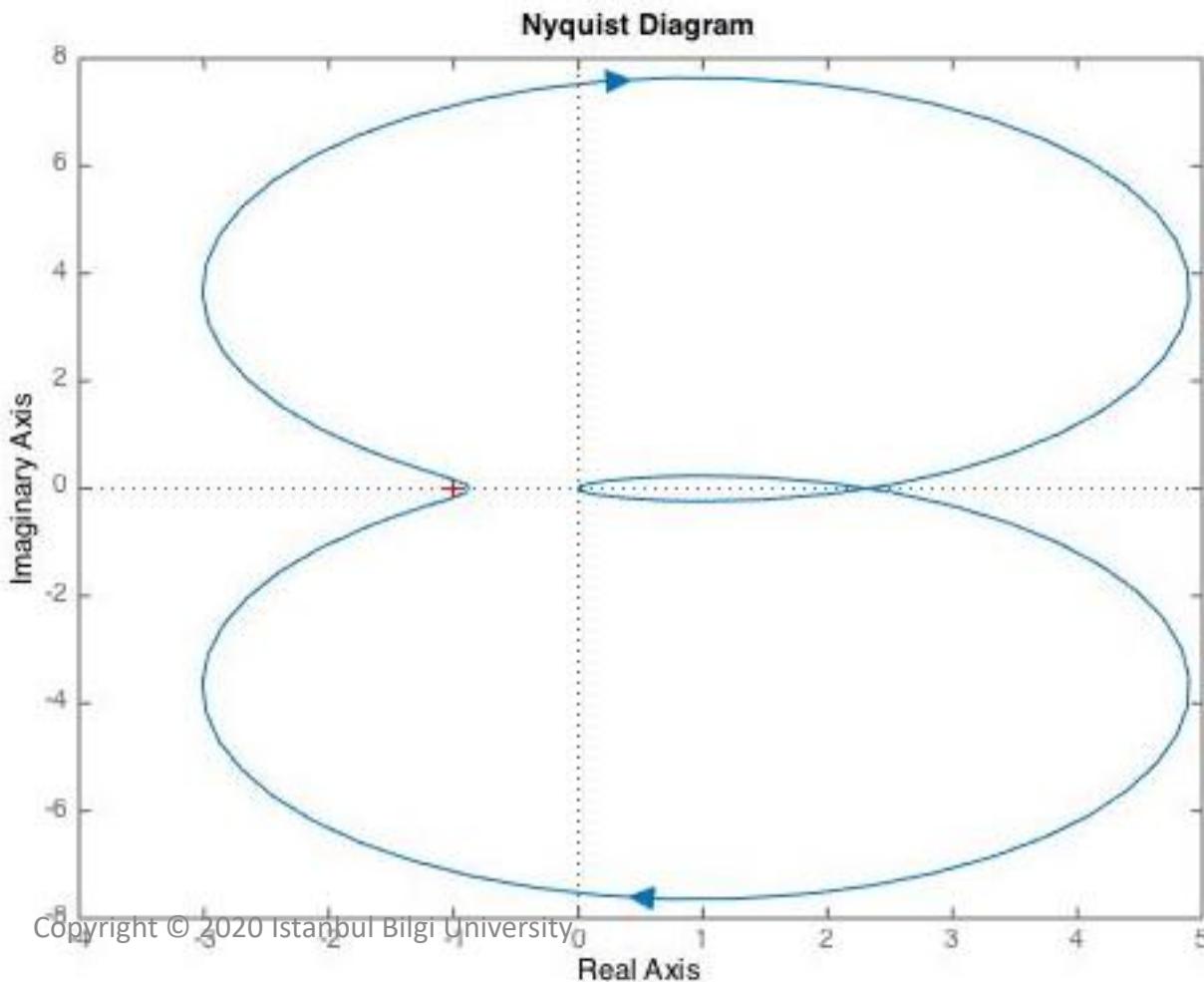
Plotting the Nyquist Diagram

$$\begin{vmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{vmatrix} = \begin{vmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -8 & -2 & -7 \end{vmatrix} \begin{vmatrix} x_1 \\ x_2 \\ x_3 \end{vmatrix} + \begin{vmatrix} 0 \\ 1 \\ -14 \end{vmatrix} \begin{vmatrix} u_1 \\ u_2 \\ u_3 \end{vmatrix}$$

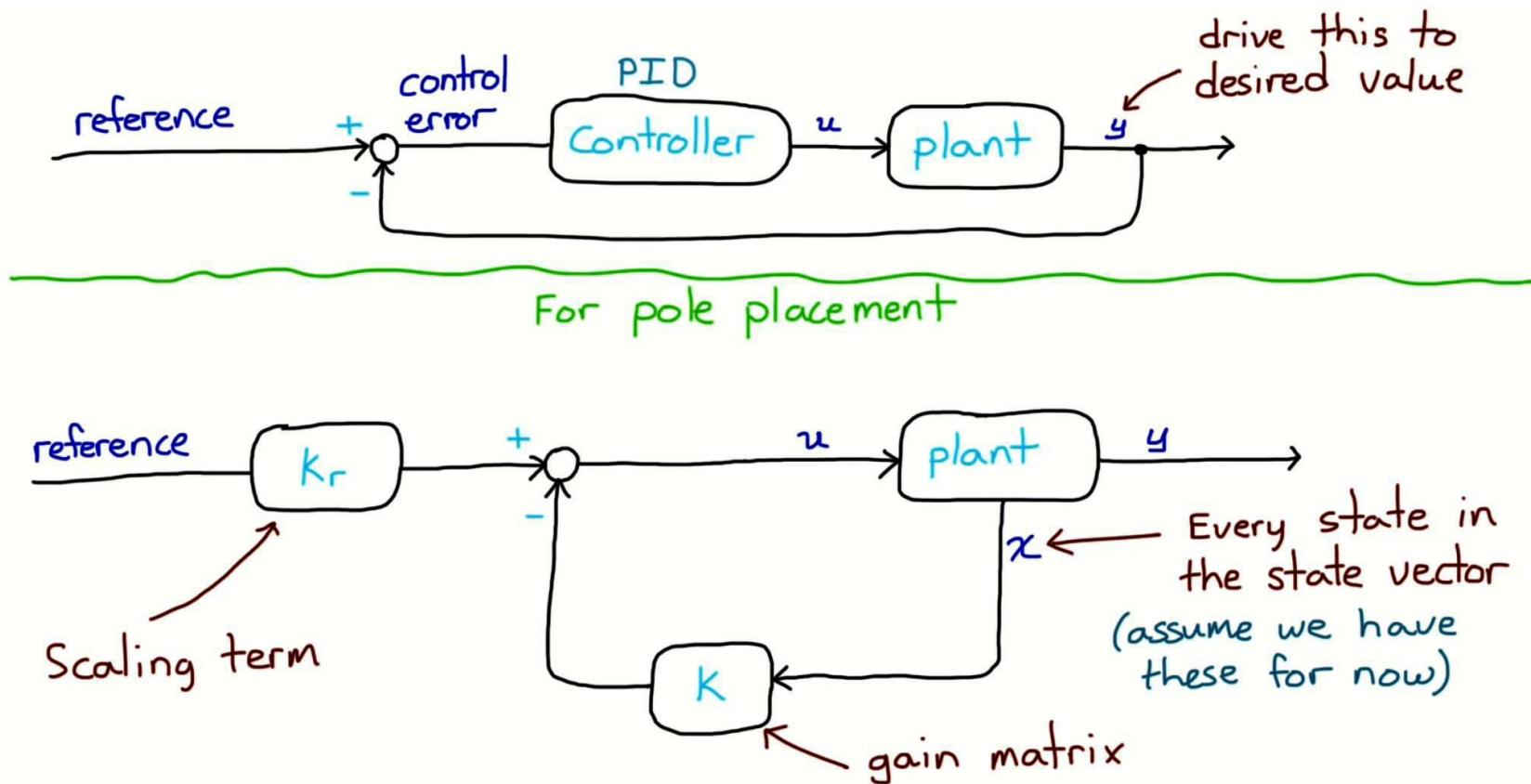
$$y = \begin{vmatrix} 1 & 0 & 0 \end{vmatrix} \begin{vmatrix} x_1 \\ x_2 \\ x_3 \end{vmatrix}$$

nyquise (A, B, C, D)

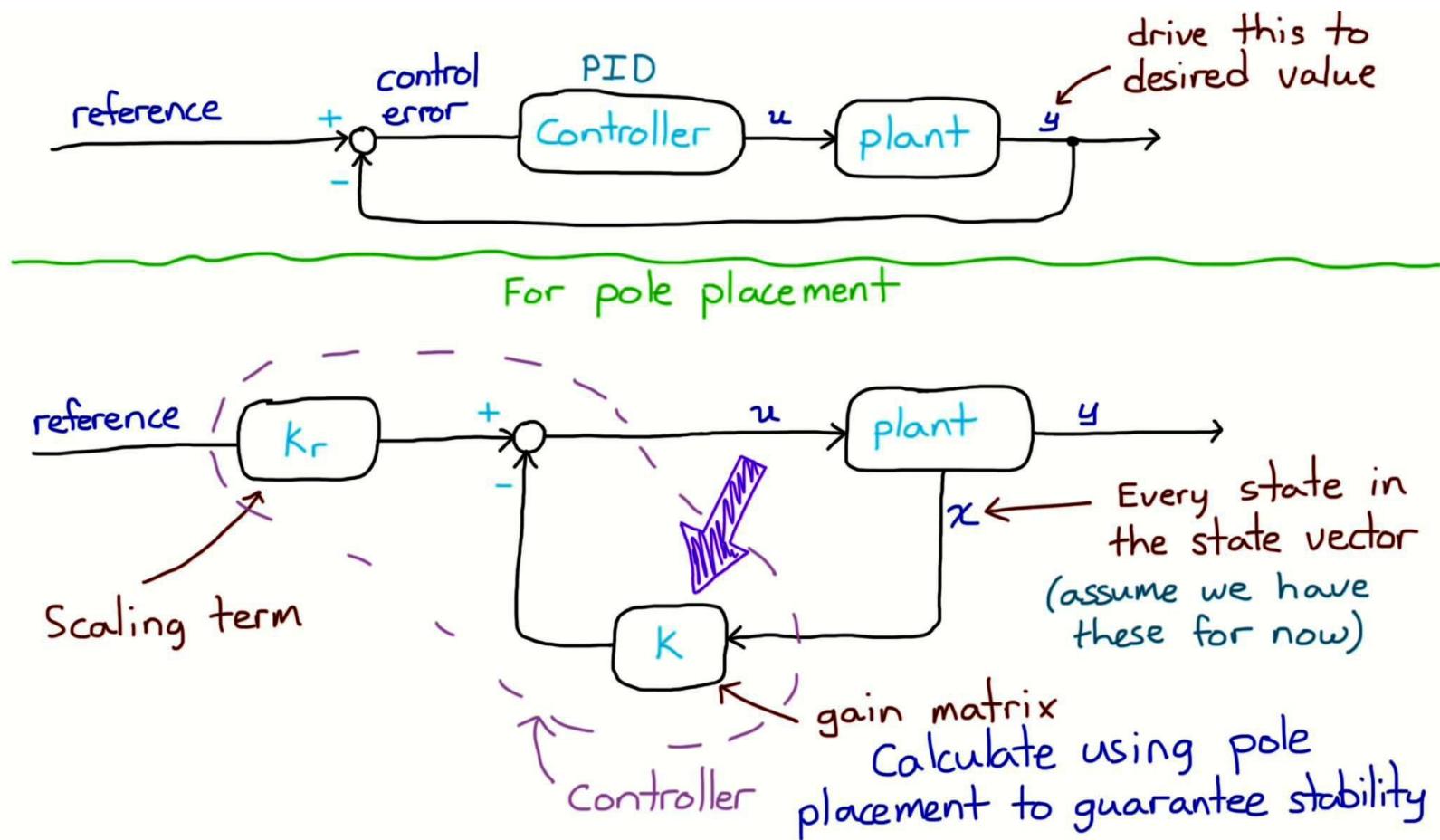
```
%Finding the transfer function  
%System is represented in State-Space  
%**** Enter matrices A,B,C and D  
A=[0 1 0;0 0 1;-8 -2 -7];  
B=[0; 1;-14];  
C=[1 0 0];  
D=[0];  
nyquist(A,B,C,D)
```



Design by Pole Placement



Design by Pole Placement



Design by Pole Placement

$$\dot{x} = \underbrace{Ax}_{\text{Dynamics captured here}} + \underbrace{Bu}_{\text{How the system responds to inputs}}$$

↑ Special? Yes!

A controller has to modify the A matrix to change dynamics

eigenvalues (A) = poles of the system

↑ location dictates stability

Moving the poles \Rightarrow choose system stability

Design by Pole Placement

$$\dot{x} = \begin{bmatrix} 0 & 1 \\ 2 & -1 \end{bmatrix} x$$

A

can be written as

$$\begin{aligned}\dot{x}_1 &= x_2 \\ \dot{x}_2 &= 2x_1 - x_2\end{aligned}$$

transform using

```
>> A=[0 1;2 -1];
>> eig(A)

ans =
    1
   -2

>> [X,D]=eig(A)

X =
    0.7071   -0.4472
    0.7071    0.8944
```

D =

1 0
0 -2

↓ to

$$\dot{z} = \tilde{A}z$$

eigenvalues, λ

$$\dot{z} = \begin{bmatrix} 1 & 0 \\ 0 & -2 \end{bmatrix} z$$

can be written as

$$\begin{aligned}\dot{z}_1 &= z_1 \text{ function of its own state} \\ \dot{z}_2 &= -2z_2\end{aligned}$$

Design by Pole Placement

$$\dot{\tilde{z}} = \tilde{A}\tilde{z}$$

\downarrow

$$\dot{\tilde{z}} = \begin{bmatrix} 1 & 0 \\ 0 & -2 \end{bmatrix} z$$

eigenvalues, λ

can be written as

$$\dot{z}_1 = z_1 \text{ function of its own state}$$
$$\dot{z}_2 = -2z_2$$

Also,

$$\dot{\tilde{z}} = \tilde{A}\tilde{z}$$

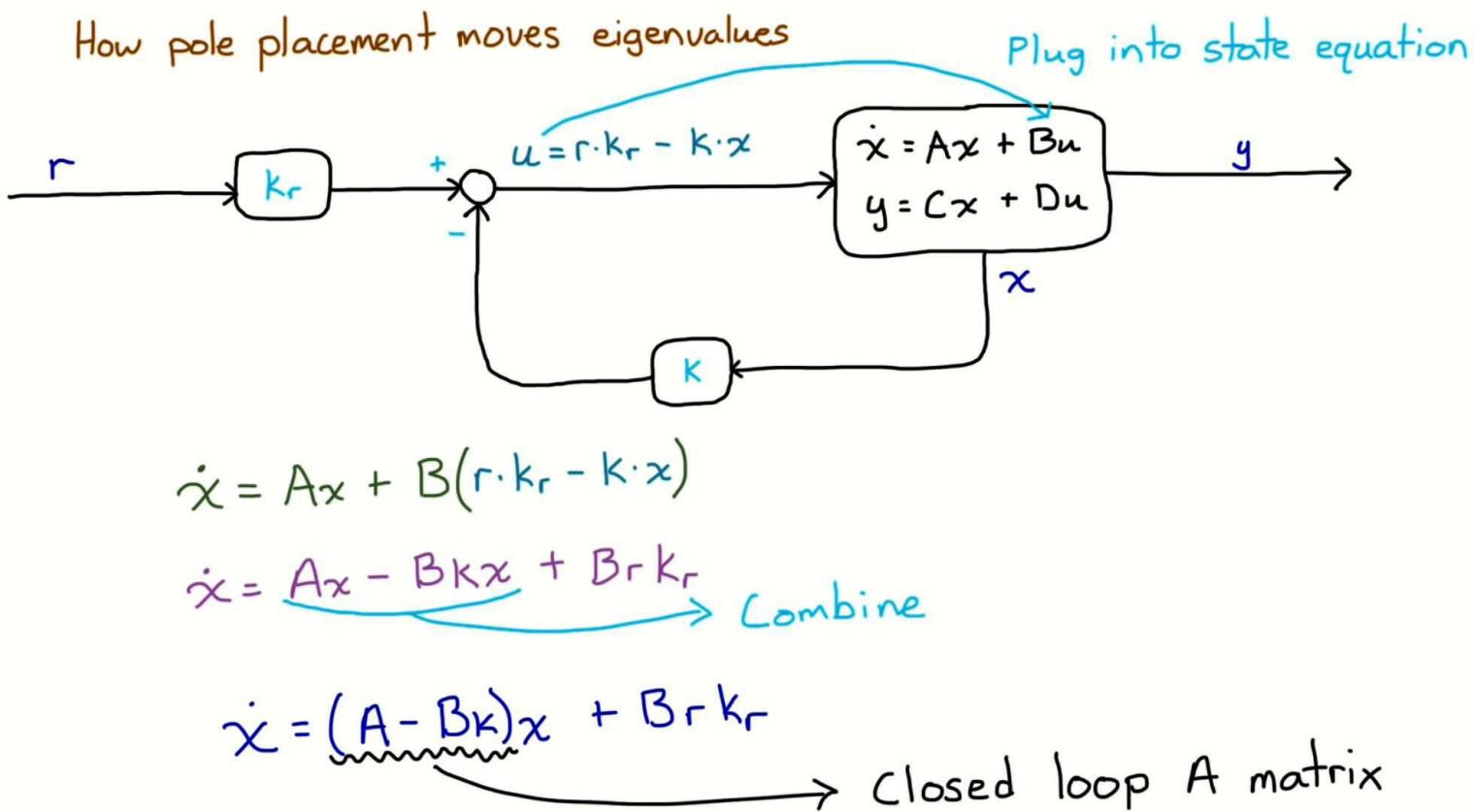
eigenvalues, λ

$$\dot{\tilde{z}} = \begin{bmatrix} -2 & 0 \\ 0 & 1 \end{bmatrix} z$$

can be written as

$$\dot{z}_1 = -2z_1$$
$$\dot{z}_2 = z_2 \text{ function of its own state}$$

Design by Pole Placement



Design by Pole Placement

example

$$\dot{x} = \begin{bmatrix} 0 & 1 \\ 2 & -1 \end{bmatrix}x + \begin{bmatrix} 1 \\ 0 \end{bmatrix}u$$

$$y = [1 \ 0]x$$

find eigenvalues

$$\det(A - \lambda I) = 0$$

$$\det\left(\begin{bmatrix} -\lambda & 1 \\ 2 & -1-\lambda \end{bmatrix}\right) = 0 \Rightarrow \lambda^2 + \lambda - 2 = 0$$

$$\lambda = -2, +1$$

unstable

Solve for λ

pole placement

$$A_{cl} = A - B \cdot K = \begin{bmatrix} 0 & 1 \\ 2 & -1 \end{bmatrix} - \begin{bmatrix} 1 \\ 0 \end{bmatrix} \begin{bmatrix} k_1 & k_2 \end{bmatrix}$$

$$A_{cl} = \begin{bmatrix} -k_1 & 1-k_2 \\ 2 & -1 \end{bmatrix}$$

Let's place poles at $\lambda = -2, -1$

$$(\lambda+2)(\lambda+1) = 0$$

find eigenvalues of A_{cl}

$$\det(A_{cl} - \lambda I) = 0$$

$$\det\left(\begin{bmatrix} -k_1 - \lambda & 1 - k_2 \\ 2 & -1 - \lambda \end{bmatrix}\right) = 0$$

$$\lambda^2 + ((1+k_1)\lambda + (k_1 + 2k_2 - 2)) = 0$$

$$1 + k_1 = 3$$

$$k_1 = 2$$

$$k_1 - 2 + 2k_2 = 2 \quad k_2 = 1$$

Summary

Designing a Control System by Pole Placement

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

We shall choose the control as

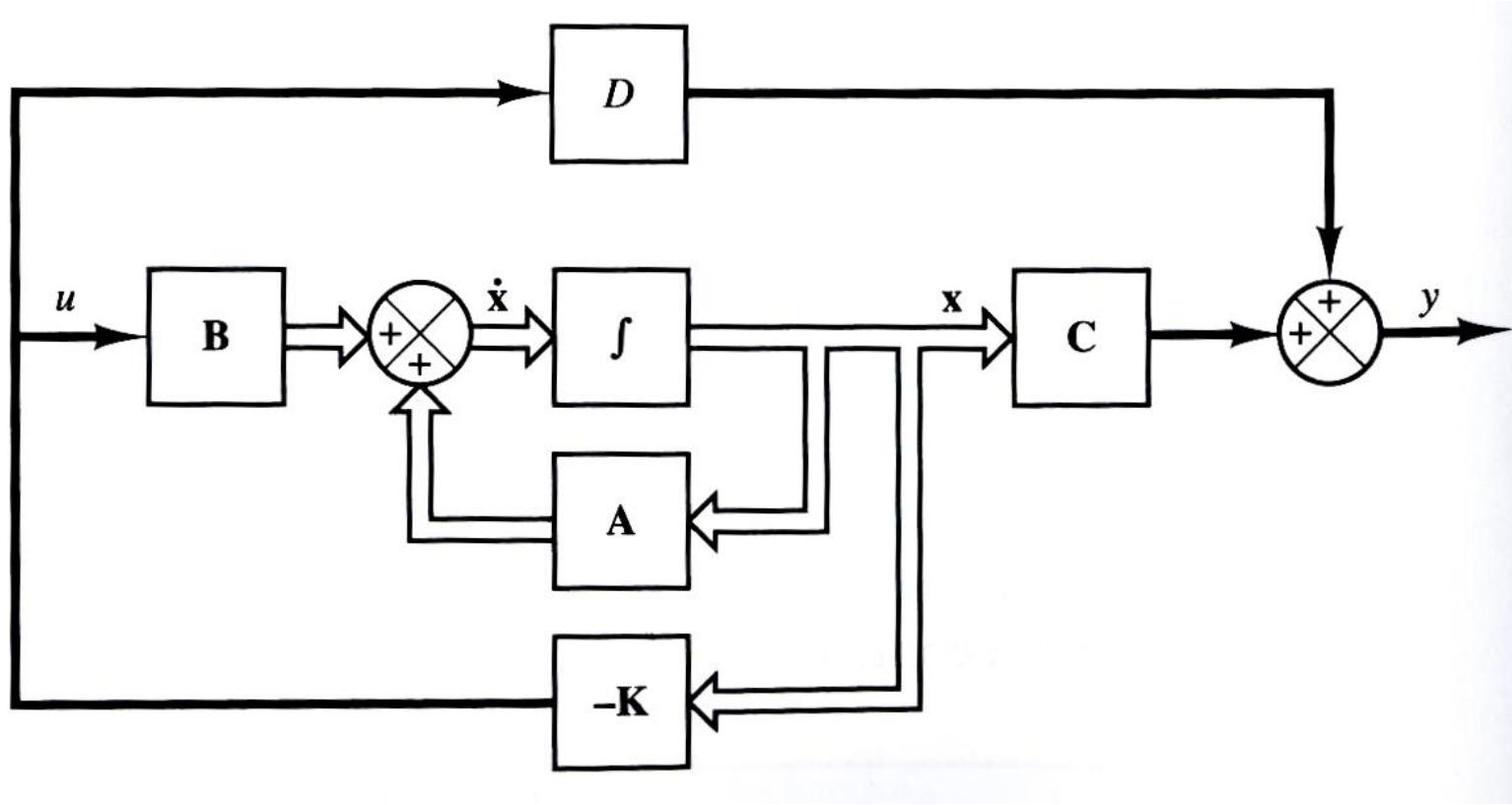
$$u = -Kx$$

$$\dot{x} = (A - BK)x$$

How should we choose K so that
the system will have desired
closed loop poles?

This can be solved by using MATLAB

Design by Pole Placement



Design by Pole Placement

Example:

Consider the system

$$\begin{vmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{vmatrix} = \begin{vmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ -1 & -5 & -6 \end{vmatrix} \begin{vmatrix} x_1 \\ x_2 \\ x_3 \end{vmatrix} + \begin{vmatrix} 0 \\ 0 \\ 1 \end{vmatrix} \begin{vmatrix} u_1 \\ u_2 \\ u_3 \end{vmatrix}$$

$$u = -[k_1 \ k_2 \ k_3] \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

determine k_1, k_2, k_3

to have the closed loop poles at

$$P_1 = -2+j4, \quad P_2 = -2-j4, \quad P_3 = -10$$

$$\mathcal{J} = [P_1 \ P_2 \ P_3]$$

$$K = \text{place}(A, B, \mathcal{J})$$

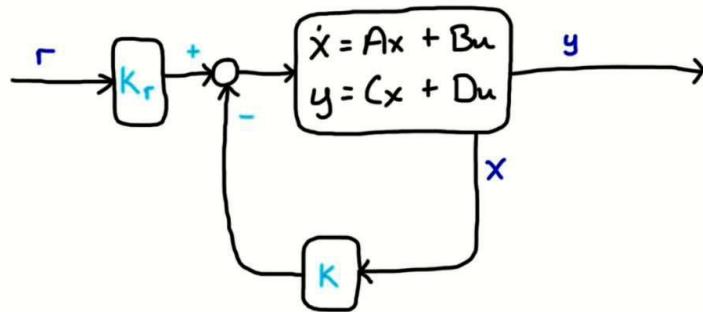
Design by Pole Placement

```
%Design by Pole_Placement  
%Enter A, B and J  
A=[0 1 0;0 0 1;-1 -5 -6];  
B=[0; 0; 1];  
J=[-2+j*4 -2-j*4 -10];  
%Compute K  
K=place(A,B,J)
```

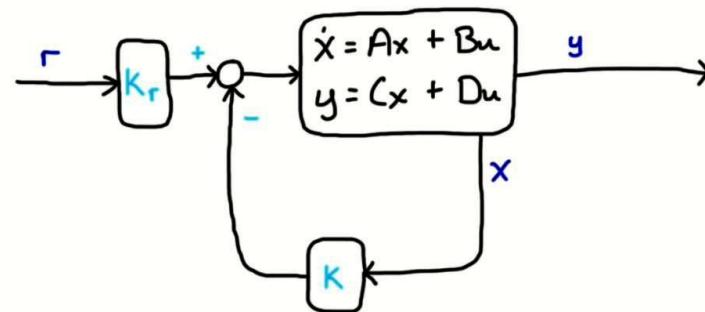
K =

199.0000 55.0000 8.0000

Pole placement
full state feedback

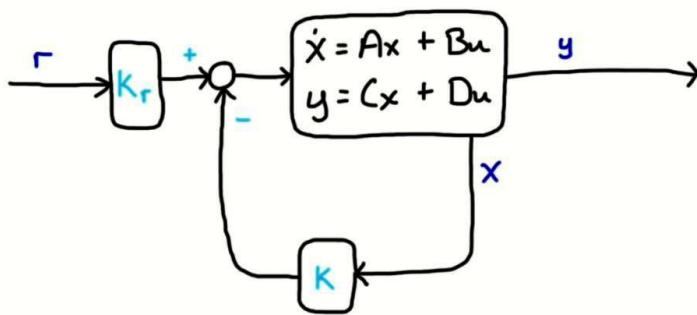


LQR
full state feedback

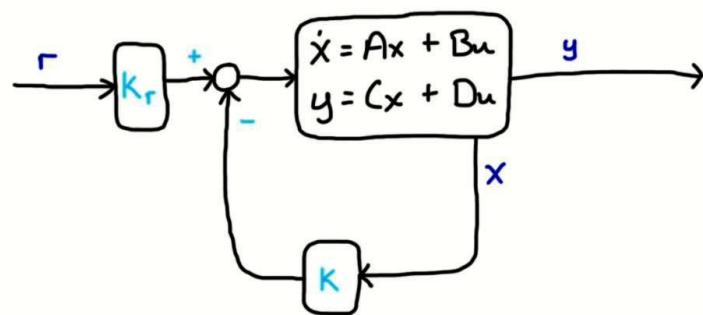


What is the Difference ?

Pole placement
full state feedback



LQR
full state feedback



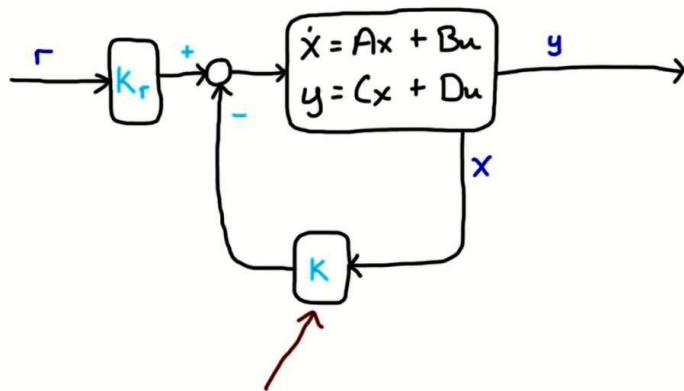
The implementation of K is the same

But how we choose K is different

Design by Pole Placement

Design by LQR

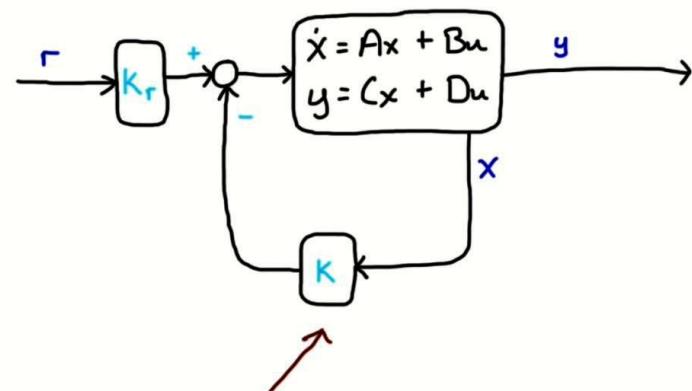
Pole placement
full state feedback



Solve for K by choosing
pole locations

not terribly intuitive

LQR
full state feedback

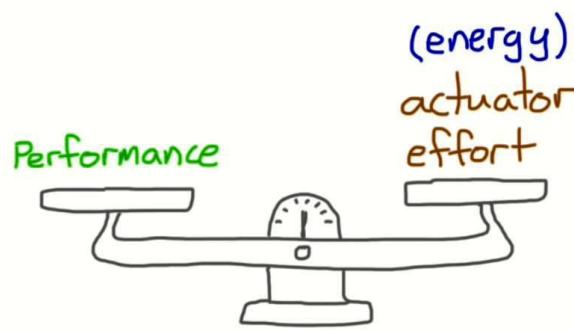
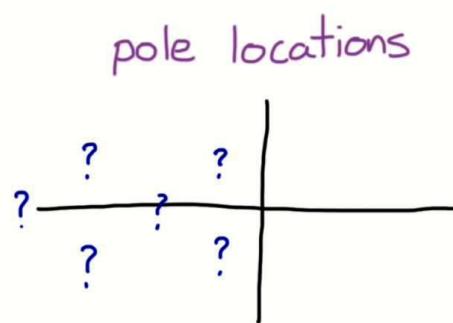


Find the optimal K by
choosing characteristics

performance & effort

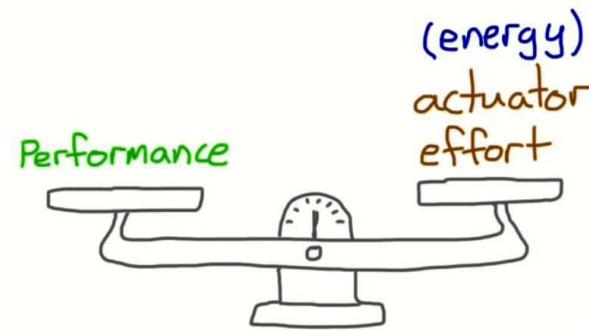
Design by Pole Placement

Design by LQR



This is how LQR approaches
finding the optimal gain

Design by LQR



This is how LQR approaches finding the optimal gain

$$J = \int_0^{\infty} (x^T Q x + u^T R u) dt \Rightarrow u = -K x \Rightarrow$$

find optimal K

$$\dot{x} = Ax + Bu$$
$$y = Cx + Du$$

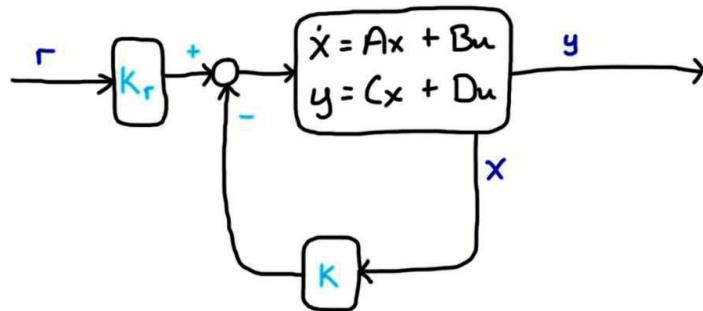
when subject to these dynamics

↑ performance ↑ effort

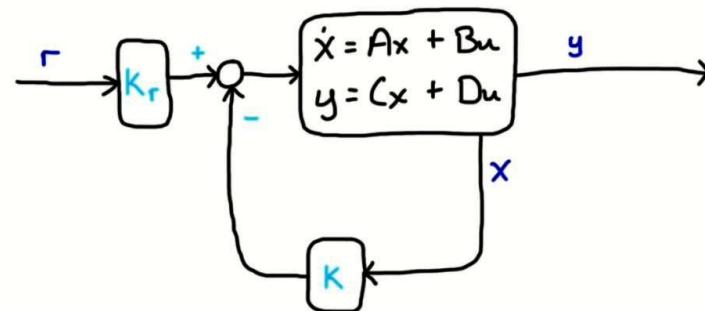
Design by Pole Placement

Design by LQR

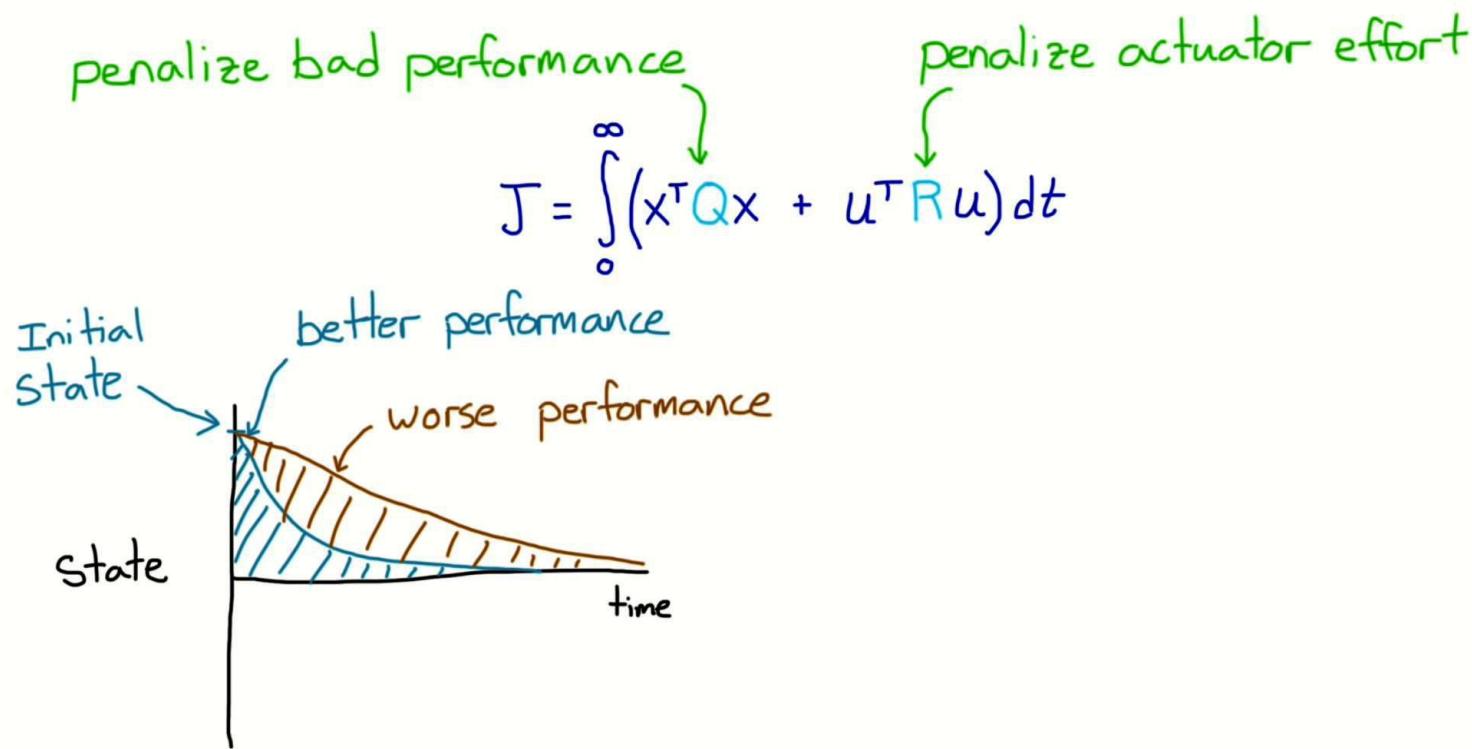
Pole placement
full state feedback



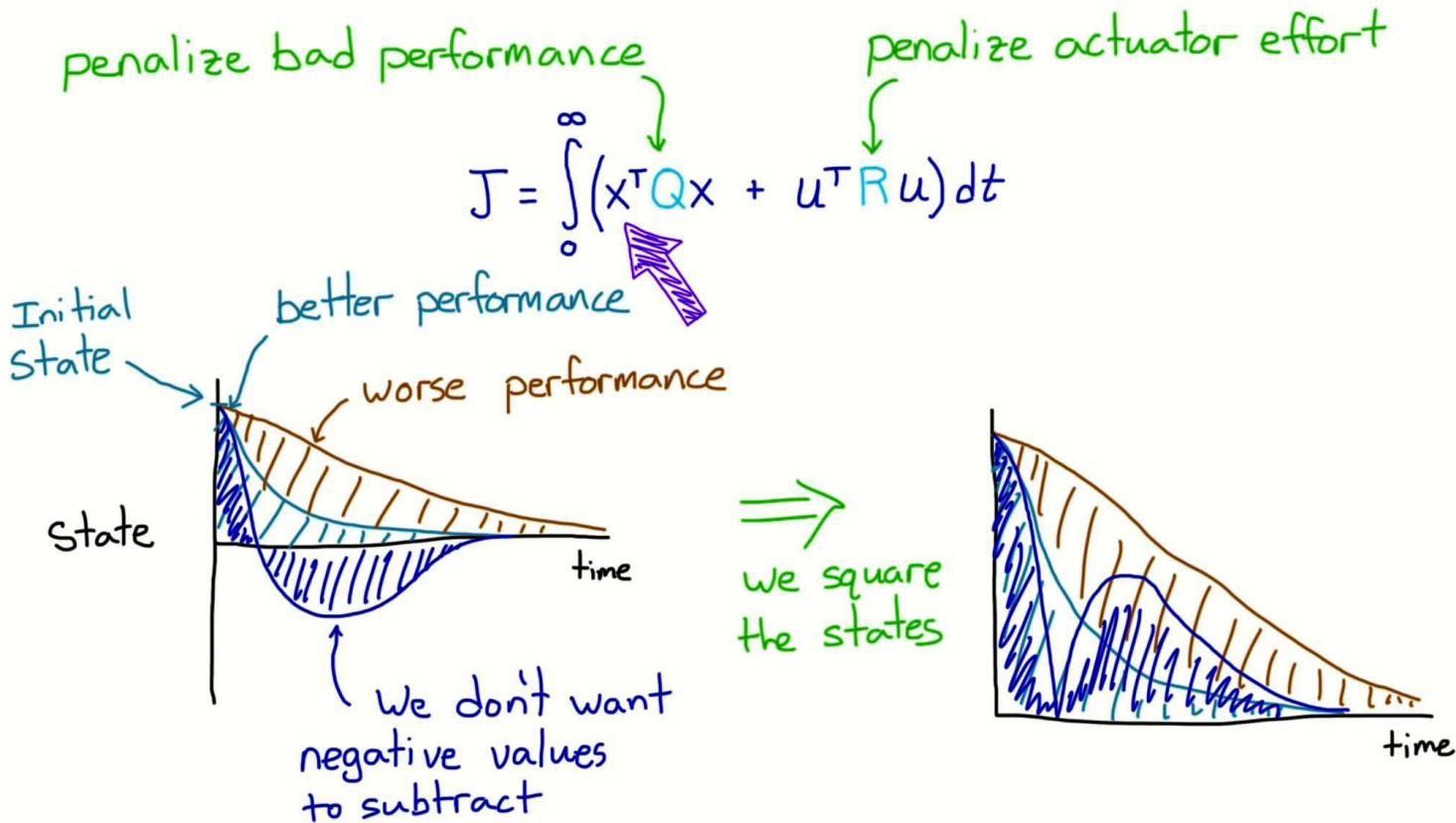
LQR
full state feedback



Design by LQR



Design by LQR

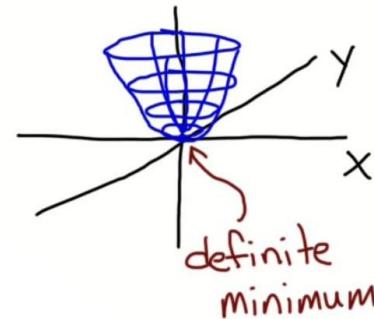


Design by LQR

penalize bad performance penalize actuator effort

$$J = \int_0^{\infty} (x^T Q x + u^T R u) dt$$

Quadratic Functions
 $z = x^2 + y^2$



penalize bad performance

penalize actuator effort

$$J = \int_0^{\infty} (x^T Q x + u^T R u) dt$$
$$[x_1 \ x_2 \dots x_n] \begin{bmatrix} Q_1 & & \\ & Q_2 & \\ & & \ddots \\ & & & Q_n \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix}$$

Positive definite

$$x^T Q x > 0$$

Design by LQR

penalize bad performance

penalize actuator effort

$$J = \int_0^{\infty} (x^T Q x + u^T R u) dt$$
$$\begin{bmatrix} x_1 & x_2 & \dots & x_n \end{bmatrix} \begin{bmatrix} Q_1 & & & \\ & Q_2 & & \\ & & \ddots & \\ & & & Q_n \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix}$$

Positive definite

$x^T Q X > 0$

Penalize x_2 by increasing Q_2

Design by LQR

penalize bad performance

$$J = \int_0^{\infty} (x^T Q x + u^T R u) dt$$

penalize actuator effort

R is similar but acts on the input vector

$$[x_1, x_2 \dots x_n] \begin{bmatrix} Q_1 & O \\ O & \ddots \\ O & O_n \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix}$$

Penalize x_2
by increasing Q_2

$$[u_1, u_2 \dots u_m] \begin{bmatrix} R_1 & O \\ O & \ddots \\ O & R_m \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ \vdots \\ u_m \end{bmatrix}$$

Expensive actuator?

Design by LQR

penalize bad performance

$$J = \int_0^{\infty} (x^T Q x + u^T R u) dt$$

$$[x_1, x_2 \dots x_n] \begin{bmatrix} Q_1 & O \\ O & \ddots & O \\ & O & Q_n \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix}$$

Low error is important?

Penalize it by increasing Q_2

penalize actuator effort

$$[u_1, u_2 \dots u_m] \begin{bmatrix} R_1 & O \\ O & \ddots & O \\ & O & R_m \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ \vdots \\ u_m \end{bmatrix}$$

Expensive actuator?

Penalize it by increasing R_2

Design by LQR

How do we solve this optimization problem?

$$J = \int_0^\infty (x^T Q x + u^T R u) dt \Rightarrow u = -K x \Rightarrow$$

find optimal K

$\dot{x} = Ax + Bu$
 $y = Cx + Du$
when subject
to these dynamics

↑
performance ↑
effort

Designing with LQR

- Develop a linear model

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

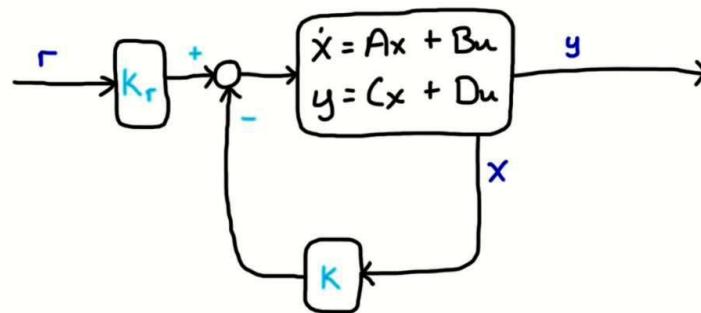
- Adjust Q and R

- Find the optimal gain set

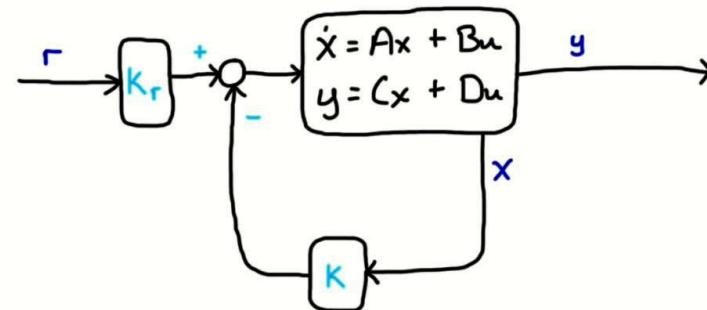
$$\gg K = lqr(A, B, Q, R)$$

Design by LQR

Pole placement
full state feedback



LQR
full state feedback



Design of Optimal Control System

Optimal Control

Given the system equation

$$\dot{x} = Ax + Bu$$

Determine K

$$u(t) = -Kx(t)$$

so as to minimize the performance index

$$J = \int_0^{\infty} (x^T Q x + u^T R u) dt$$

This problem is solved by
using the Matlab function

$$K = lqr(A, B, Q, R)$$

Design of Optimal Control System

Example

Consider the system

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

where

$$A = \begin{vmatrix} 0 & 1 & 0 \\ 0 & 0 & 1 \\ 0 & -2 & -3 \end{vmatrix}, \quad B = \begin{vmatrix} 0 \\ 0 \\ 1 \end{vmatrix}, \quad C = [1 \ 0 \ 0], \quad D = [0]$$

Determine the state-feedback gain matrix

$$K = [k_1 \ k_2 \ k_3]$$

such that the following performance index is minimized

$$J = \int_0^{\infty} (x^T Q x + u^T R u) dt$$

where

$$Q = \begin{vmatrix} q_{11} & 0 & 0 \\ 0 & q_{22} & 0 \\ 0 & 0 & q_{33} \end{vmatrix}, \quad R = r_1$$

and

$$q_{11} = 100$$

$$q_{22} = q_{33} = 0.01$$

Design of Optimal Control System

```
%Design by Pole_Placement  
%Enter A, B, Q and R  
A=[0 1 0;0 0 1;0 -2 -3];  
B=[0; 0; 1];  
Q=[100 0 0;0 1 0;0 0 1];  
R=[0.01];  
%Compute K  
K=lqr(A,B,Q,R)
```

K =

100.0000 53.1200 11.6711

End of Lecture 6