

0.1 High level goals

0.2 Mission drivers

Given the nature of the mission, being JUNO an interplanetary mission starting from a distance of around 1 AU, with an final nominal distance from the Sun of 5.2 AU, and operating in an highly radiation intense environment, the following drivers have been identified:

- 1. Using proven technologies** HO AGGIUNTO QUESTO, CONTROLLARE CON CONDIZIONI NON NOMINALI, THE JUNO Mission
The total cost of the program is financed with 1.1 Billion \$ for 74 months from the launch date and includes development of the S/C, science instruments, launch services, mission operations, science processing and relay support. The simplicity and the need of proven technologies was thus fundamental. The S/C is maintained stable during the maneuvers thanks to its spin, raised to 5 RPM from 2, nominal condition during science operations.
- 2. Providing enough electricity during the duration of the mission**
The journey of JUNO is long and passes through different regions of the solar system. Solar panels were chosen to provide electric energy across the mission over a nuclear source, since it has been decided that it was better to advance technology of solar cells rather than developing a new reactor. The system needed is thus oversized at 1 AU: the radiation on Jupiter is in fact up to 96% lower on Jupiter than on Earth. Furthermore the operations are scheduled to begin around 5 years into the mission, so degradation of the solar cells must be taken into account. The final design consists in 3 solar arrays, 9 by 2.65 m each, resulting in about 60 m² and granting a maximum power of 14 KW around Earth and up to 500 W around Jupiter. It was the first S/C to operate with solar panel at such distance from the Sun. They are mounted on the side of hexagonal body of the S/C at 120° one by the other and are deployed after booster separation. The spacecraft is spinning at around 1.4 RPM during this phase and the deployment of the solar arrays slows it down. A crucial element must be considered during the cruise phases: nominally the high gain antenna, that is mounted with the same positive direction of the solar arrays, is doing Earth pointing to communicate data but, at some point during the cruise phase, due to the proximity of the Sun, thermal requirements dictate a change in attitude and thus the spacecraft won't be pointing directly the Sun. Moreover, since the fly-by around Earth is done to gain ΔV , the S/C will be in an eclipse for around 20 minutes: batteries are thus needed. Two lithium-ions battery of 55 Ah each are present. The nominal polar orbit around Jupiter allows Sun pointing during the whole nominal science operation Phase (NOME DELLA FASE A CUI CI RIFERIAMO). INSERIRE RIFERIMENTI: VIDEO, SLIDES LAVAGNA, JUNO Mission
- 3. Providing the correct temperature range across the instruments during operation and the journey**
The delicate suites of instrument present onboard JUNO require a very narrow range of temperature to operate, as low as $\pm 1K$. The MWR instruments in fact need it to measure with high accuracy the microwave emission from Jupiter. During the cruise phase JUNO will pass as close as 0.88 AU from the Sun, so the S/C will need to protect the P/L from the incoming heat. Passive solutions were preferred to maintain simplicity and reliability during the whole mission lifetime. Juno's thermal control subsystem uses a design with heaters and louvers. It consists of an insulated louvered electronics vault atop a heated propulsion module. LINK NASA JPL
- 4. Shielding the instruments from the harsh environment of Jupiter**
To accomplish its goals, JUNO will need to cross the Jupiter radiation belts: a heavy shielding structure is needed. The magnetosphere represents a great challenge for JUNO: the value of the magnetic field measured at its perijove is 776 μT , 50 % higher than expected. The main issue with JUNO orbit is represented by the ionizing particles present in the belt around Jupiter: with measured value up to of tens and hundreds of MeV ions located between 2 and 4 R_J , order of GeV were expected under 2 R_J , where JUNO should pass through to reach a lower altitude, thanks to its highly elliptical orbit, where radiations are lower, to perform science. The vault in which all the electronics is preserved is cubed shaped and it is made of 1 cm thick titanium alloy, 144 Kg in total. The top deck of Juno is planned to receive a radiation dose of 22 Mrad. Moreover, star trackers are also heavily shielded. All these attentions are needed to allow science operation and to preserve the integrity of the S/C during its lifetime.
- 5. Maintaining communication during the journey and the science operations**
Given the amount of data collected by the instruments, a continuous link with ground operations is needed. The attitude of the S/C is so defined. This configuration, given the distance from the Sun and the Earth, grants a sufficient inclinations of the solar panels with respect to the Sun to provide enough electric power [2].

- 0.3 Functional analysis**
- 0.4 Main mission phases**
- 0.5 ConOps**
- 0.6 Payload analysis**
- 0.7 Mission analysis**