

PUMped Liquid System for Amateur Rockets

Project PikaPump

Final Design Review

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James Beeman, Lamya Bhat, John Clark,
Max Cohen, Fred Gouronc, Tom Neidlein



PULSAR

Background

- Liquid rockets are fed two propellants to provide thrust
- Key stakeholders:



STUDENTS



AMATEUR ROCKET
BUILDERS



EDUCATIONAL
INSTITUTIONS

- Value Added: We aim to provide the resources in one location for other people to design and build their own electric pumps



Project Requirements

Pressure	<i>Liquid methane pressure output of 240 psia</i>
Flow	<i>Liquid methane flow rate of 1.19 lbm/s</i>
Weight	<i>Total system weight below 12.5 lbs</i>
Sizing	<i>6.6" Max diameter, 16" Maximum height</i>
Temperature	<i>Can withstand temperatures of -190 °C</i>
Corrosion	<i>Resistant to corrosive cryogenic fuels</i>
Reusability	<i>Will be able to operate for at least 2 flights</i>
Measurement	<i>Able to record outputs using sensors</i>
Safety	<i>Automatic and manual emergency shutoff</i>
Test Ready	<i>System is cryogenic test ready</i>



Business Case



\$10,000

**AVERAGE BUDGET OF
STUDENT ROCKET TEAMS**



Multi-Use

**ELECTRIC PUMPED SYSTEMS
HAVE GREATER REUSABILITY**

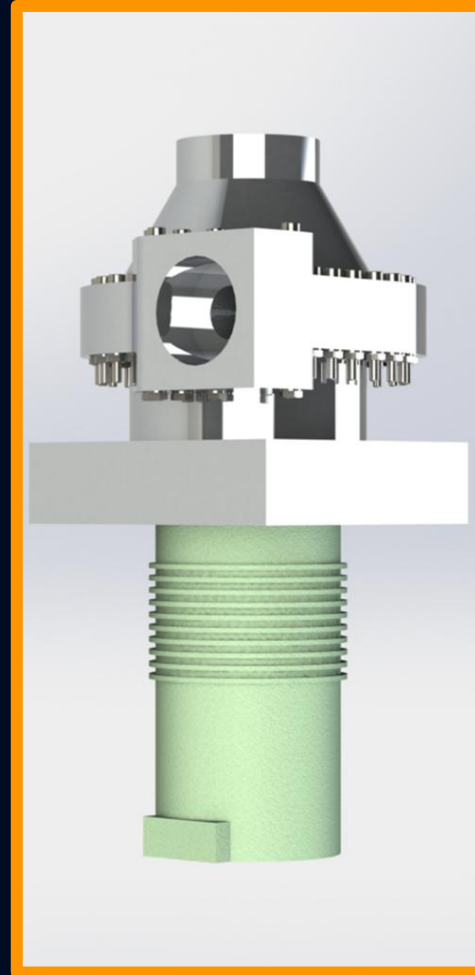
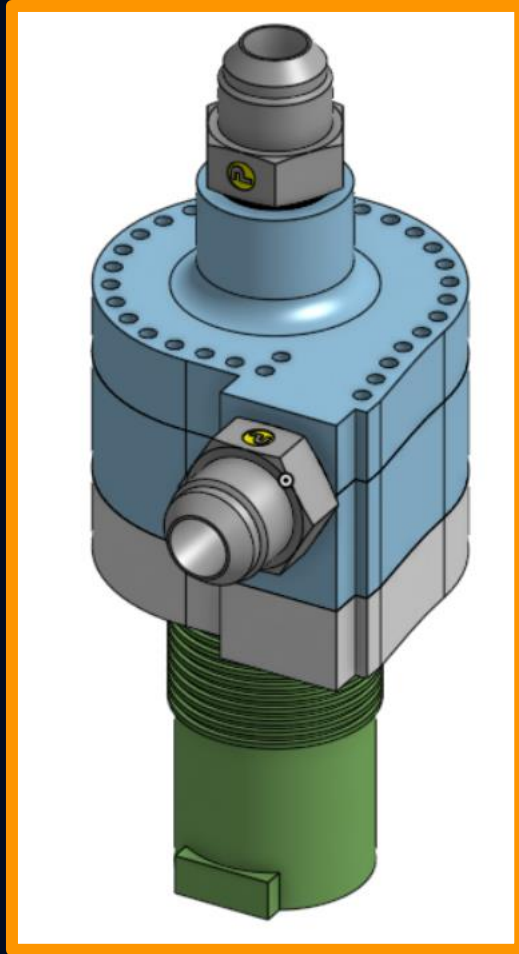


Risk

**LARGEST OBSTACLE IS
TECHNICAL RISK**

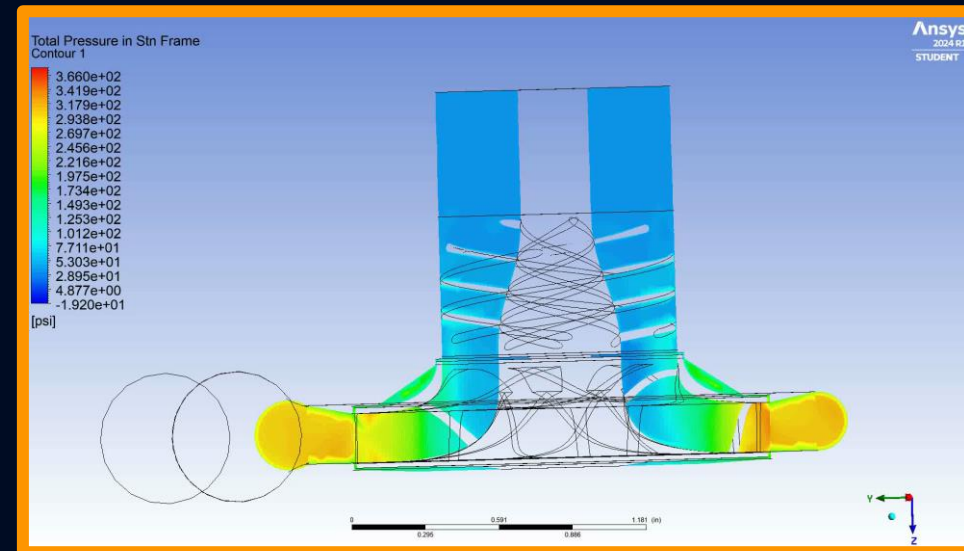
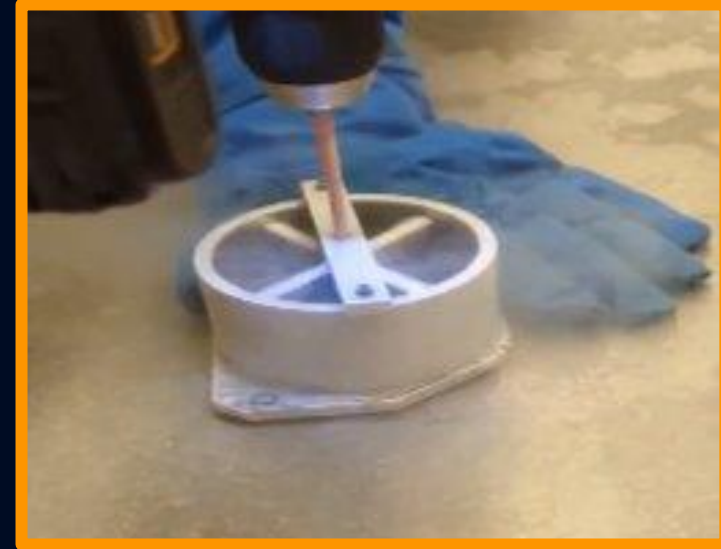


CDR to FDR Model

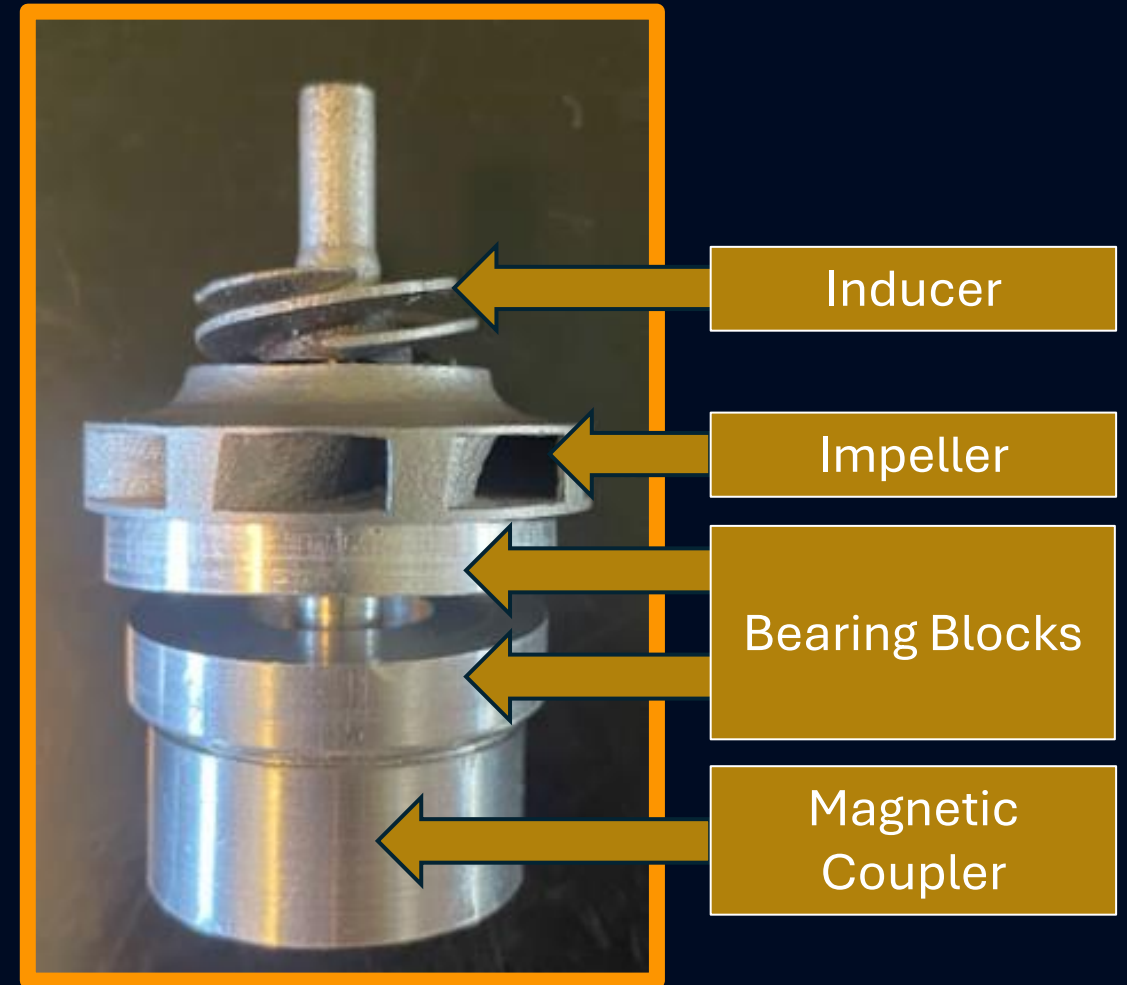
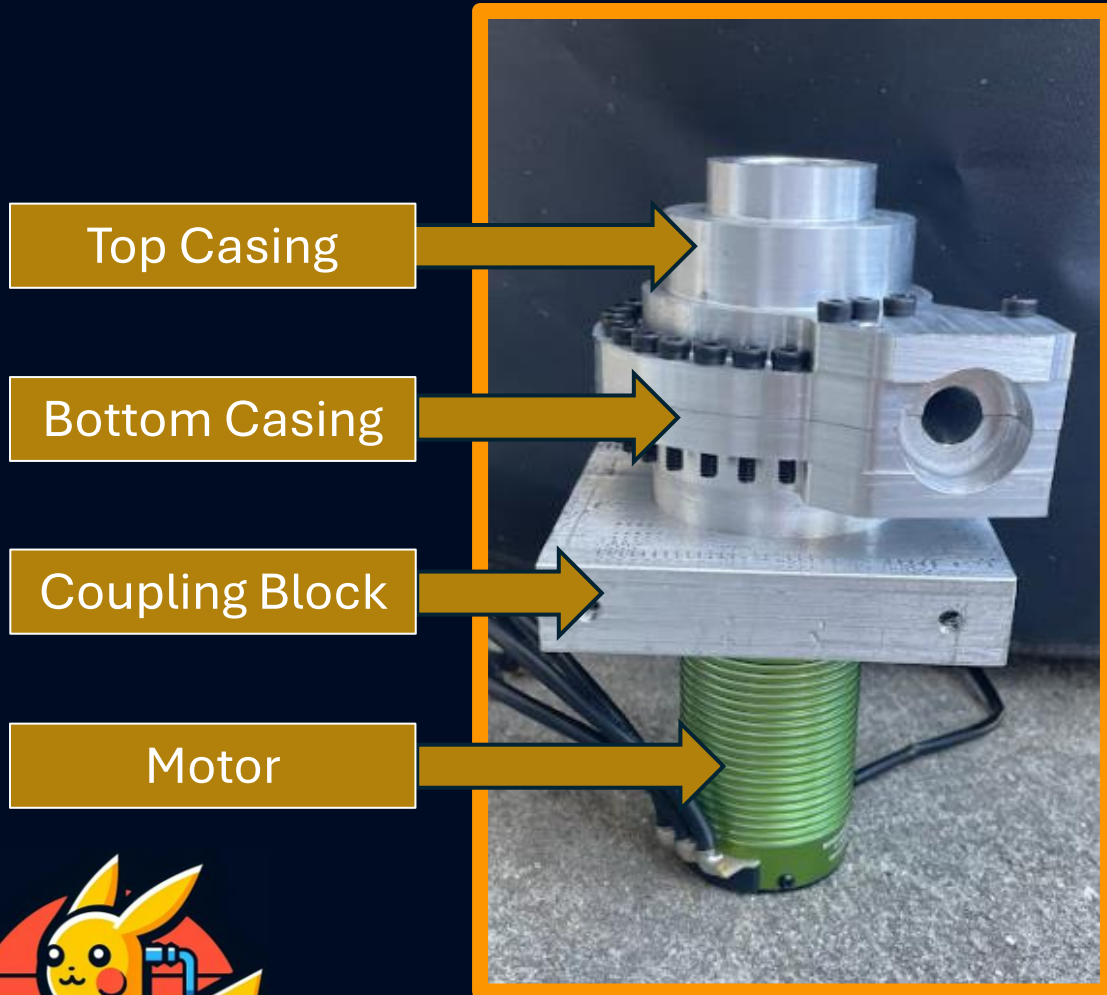


Analysis Performed

- Initial CFD sims
- Material Analysis tests
- Driveshaft design

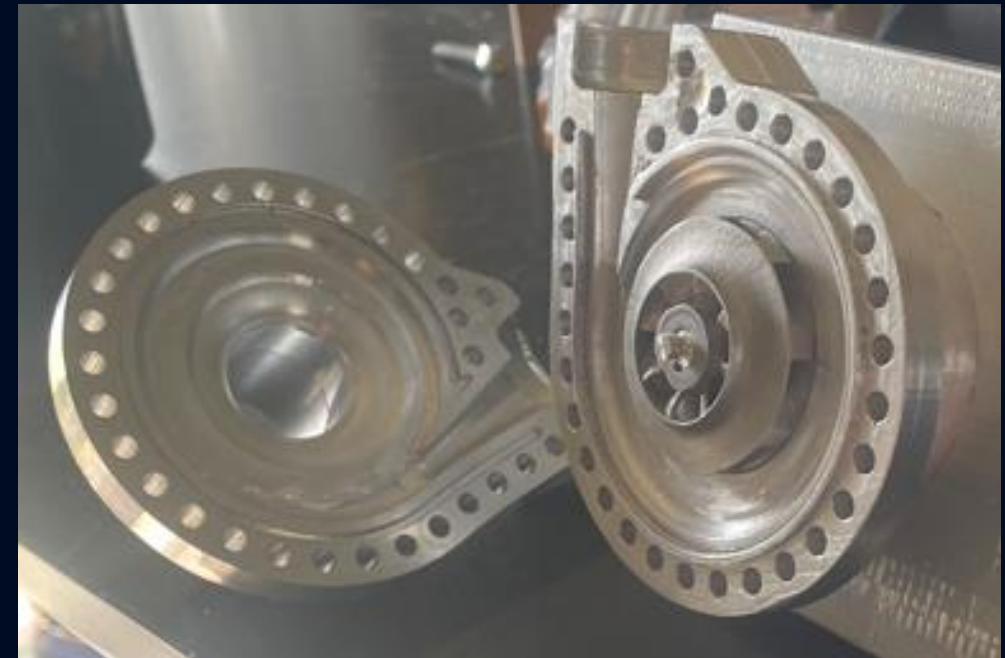
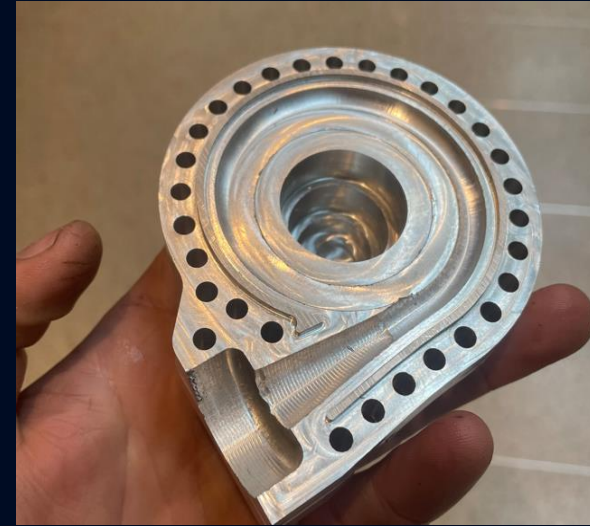


Overall Assembly



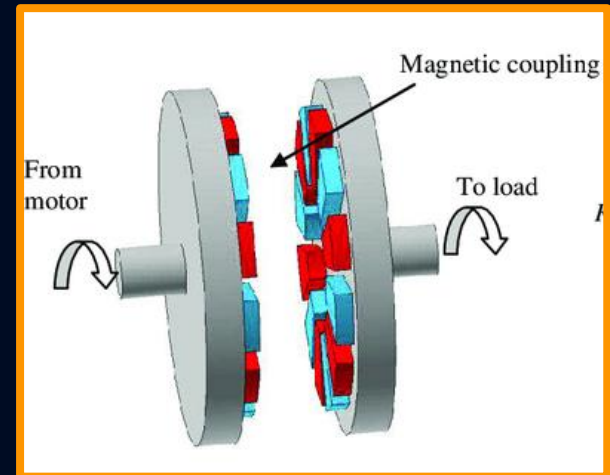
Top and Bottom Casing

- Machined on the MB-15 CNC mill
 - High precision
 - Purchased 1/16" endmill for seal
- Post processing
 - Bolted together and tapped
 - Adjusted clearances



Magnetic Coupler

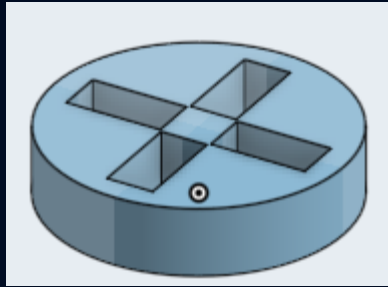
- Torque transmission with no leakage
- Custom-made couplers
 - Neodymium magnets packed into an aluminum case



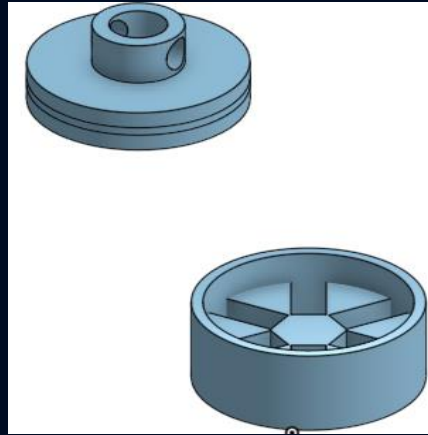
<https://www.stanfordmagnets.com/magnetic-coupling.html>



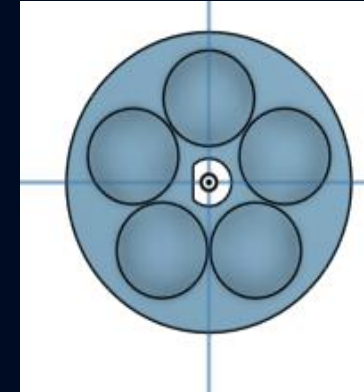
Coupler Iteration



Stage 1



Stage 2- More magnets



Final Design



Impeller and Inducer

- Core internal pump components
- Could not be easily made in-house
 - Geometries impractical for machining
- Parts were metal 3D printed
 - Sanded down to ensure smoothness

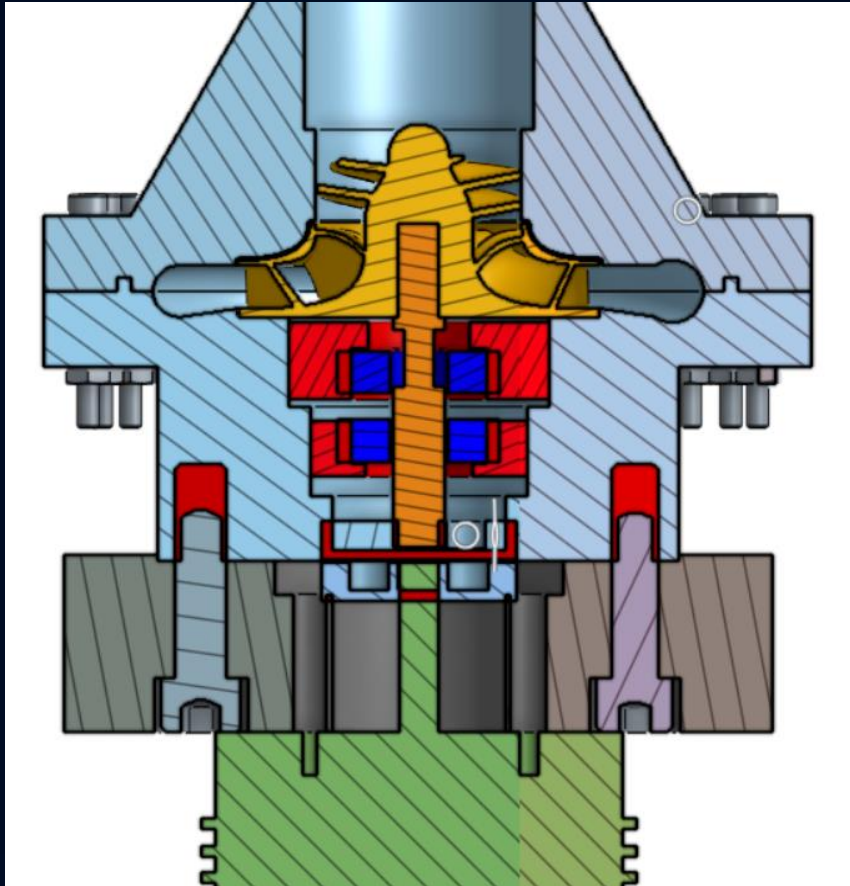


Other Major Components

- Pump Driveshaft
 - Connects magnetic coupler to pump impeller
 - Uses left hand threading to prevent unscrewing
- Coupling Block
 - Holds magnetic coupler in place
 - Supports pump and motor
 - Bolted to test setup using a welded steel H-bracket



Drivetrain Iteration

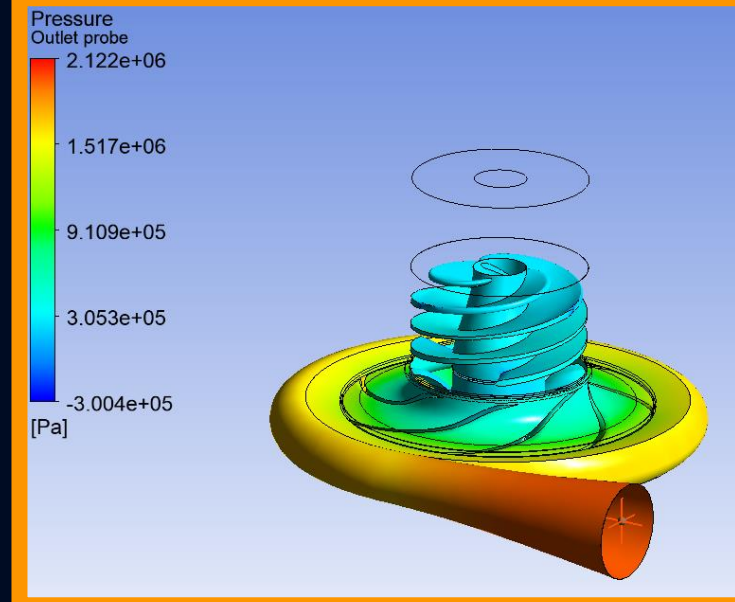
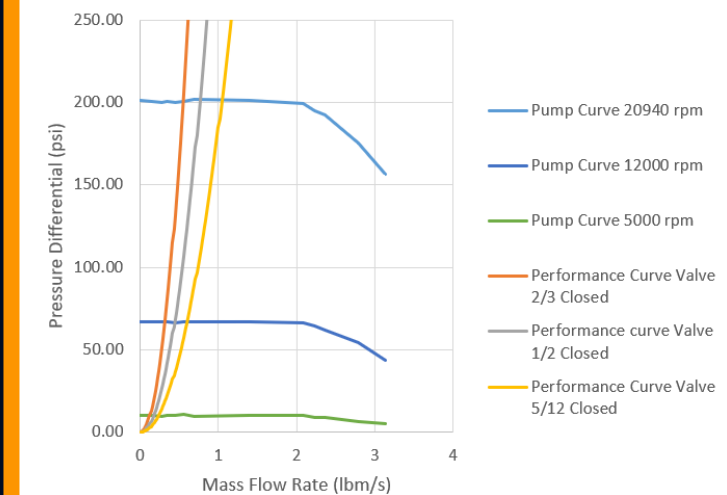


Problem	Solution
Coupler must be larger	Increase bearing block size
Can't assemble driveshaft	Add "coupler adaptor" part
Threads unscrew when system spins	Use left hand threading
Bolt assembly interferes w/ motor	Decrease bolt size and move outward
Bearing blocks can't be removed after press fit	Change interference fit to transition fit
Cryogenic bearings don't spin well at low RPMs	Use cheaper bearings for testing in water
Impeller/couplers bind with the casing	Increase clearances until it runs smoothly



CFD Simulations

- Simulate a wide range of operating conditions
- Both Water and Cryogenic fluids
- Need to be validated with experiments



Validation Plan

Verifying the design to ensure it meets expected parameters

- Create test setup
- Record parameters
- Data Analysis



Data Acquisition



1 High pressure sensor

Gstatic



1 Tachometer

Tachometer



1 Flow meter

Flow meterd



Arduino Uno

Arduino

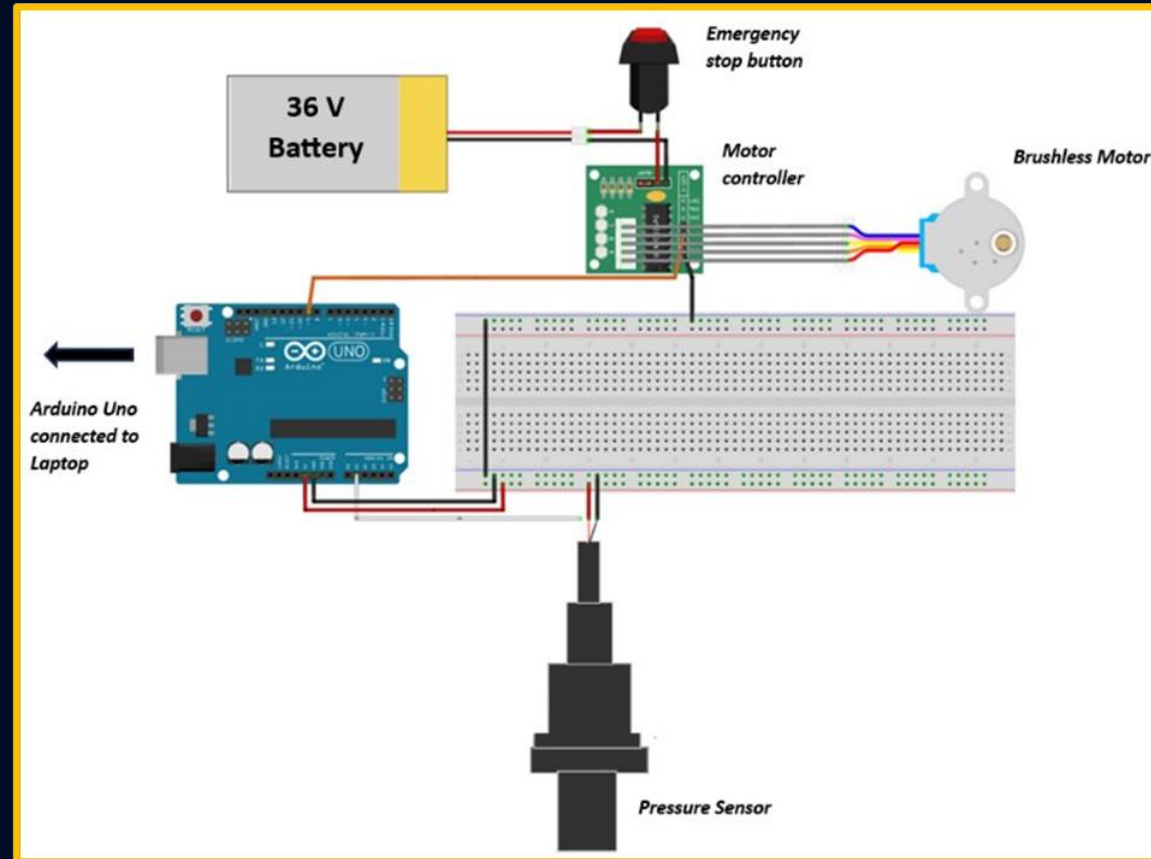


Laptop

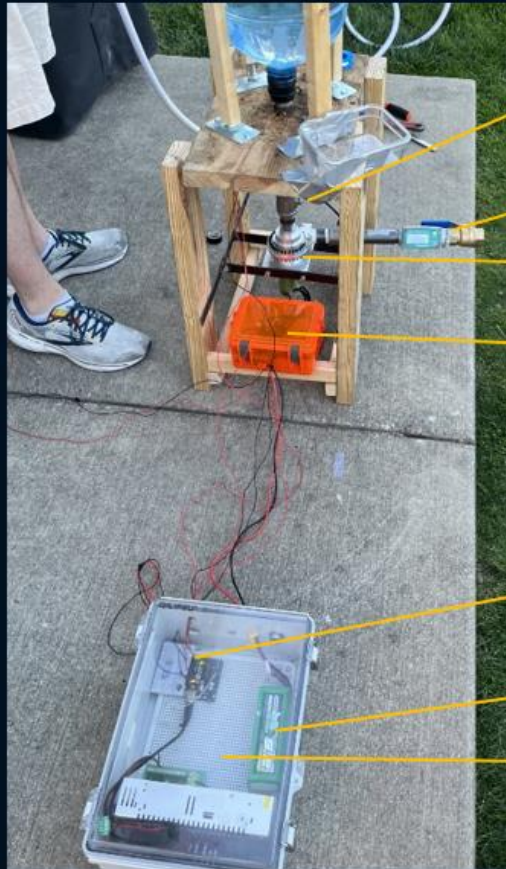
Laptop



Electrical CAD



Test Setup



Pressure sensor

Flow meter

Pump housing

Motor controller

Arduino & Breadboard

Battery

Water proof box



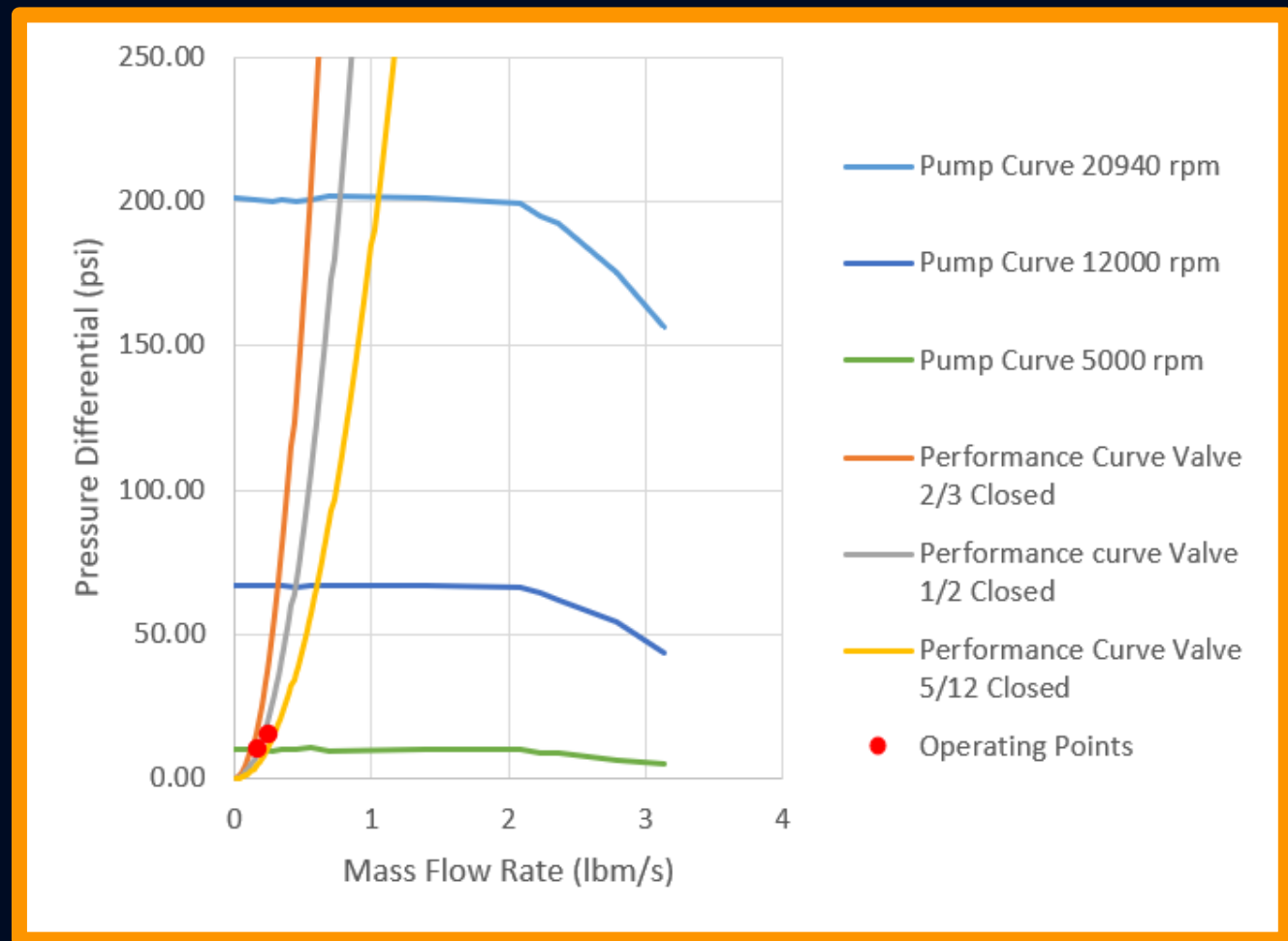
Test Results

- Ran test up ~7000 rpms
- Pressure rise ~15 psi
- Flow rate of 0.55 lbm/s
- Any higher the coupler was not strong enough



Results vs. Simulation

- Point falls on the performance curves as predicted
- Validates simulation
- Would be better to test at higher rpms



Requirements Assessment

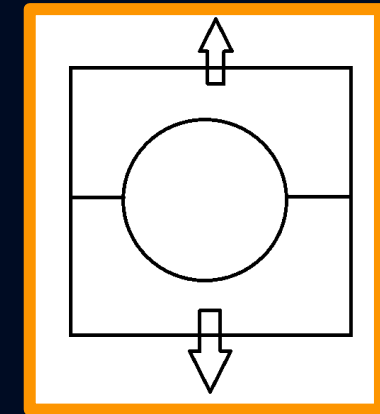
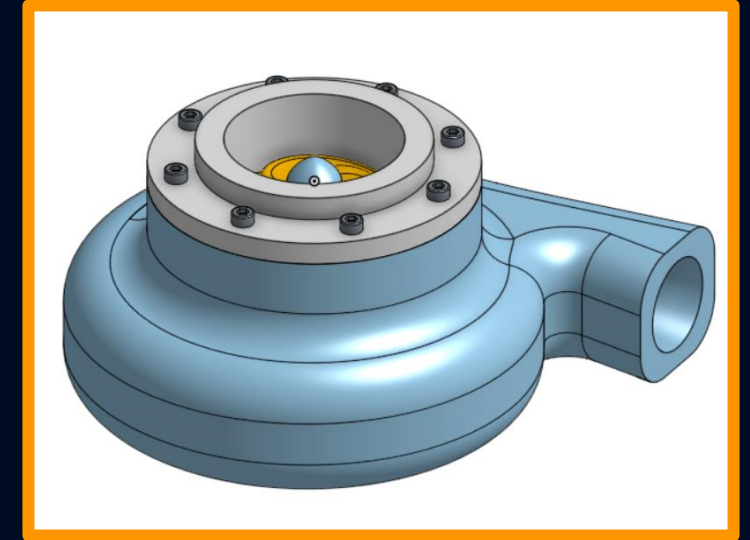
<u>Requirement</u>	<u>Description</u>	<u>Rqmt Met?</u>	<u>Reasoning</u>
Pressure	<i>Liquid methane pressure output of 240 psia</i>	No	No cryo test
Flow	<i>Liquid methane flow rate of 1.19 lbm/s</i>	No	No cryo test
Weight	<i>Total system weight below 12.5 lbs</i>	Yes	~4 lbs
Sizing	<i>6.6" Max diameter, 16" Maximum height</i>	Yes	4" Dia., 10" Tall
Temperature	<i>Can withstand temperatures of -190 °C</i>	Yes*	Full compliance
Corrosion	<i>Resistant to corrosive cryogenic fuels</i>	Yes*	Full compliance
Reusability	<i>Will be able to operate for at least 2 flights</i>	Yes	Multiple tests
Measurement	<i>Able to record outputs using sensors</i>	Yes	Data acquired
Safety	<i>Automatic and manual emergency shutoff</i>	Yes	E-stop works
Test Ready	<i>System is cryogenic test ready</i>	Yes*	Full compliance

*No tests were performed with cryogenics, but all components are designed for cryogenic compatibility



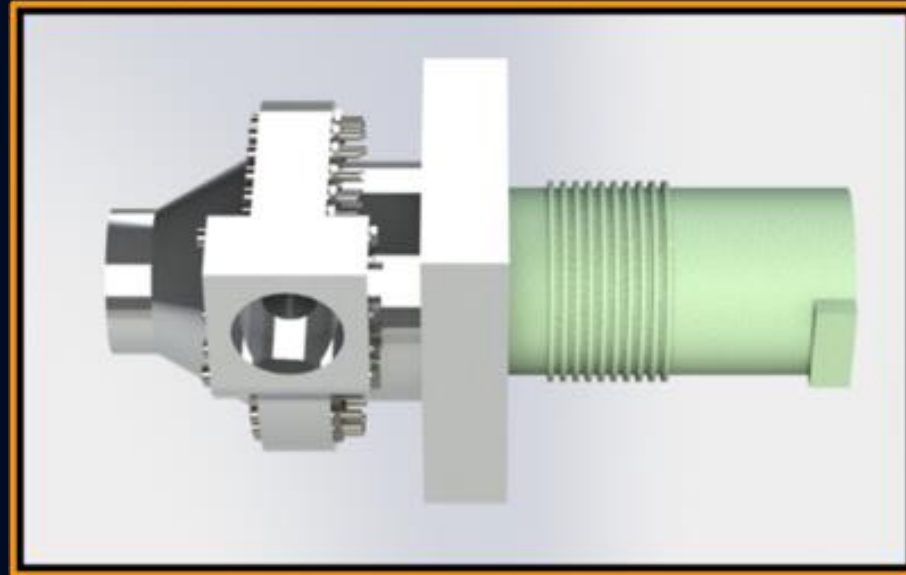
Design Improvements

- Single piece pump casing
- Coupler magnet placement and strength
- Better machining tolerances and finish
- Sensor and data acquisition improvements



Moving Forward

- Provided roadmap, documentation, CAD, and code on GitHub so others can iterate on the design



Thank you!



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