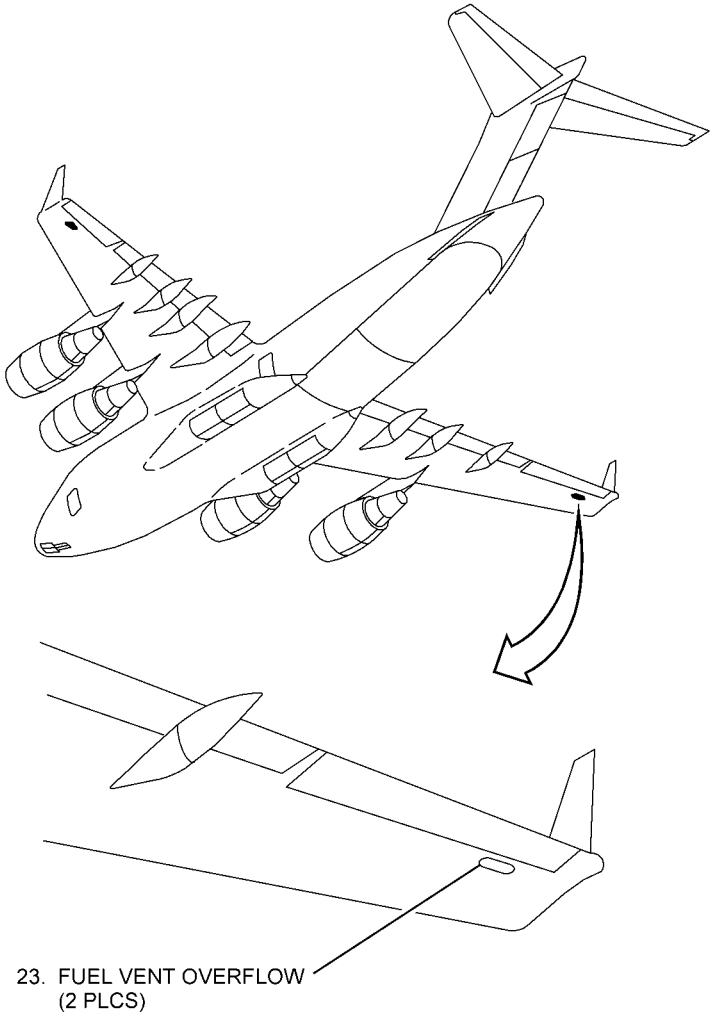


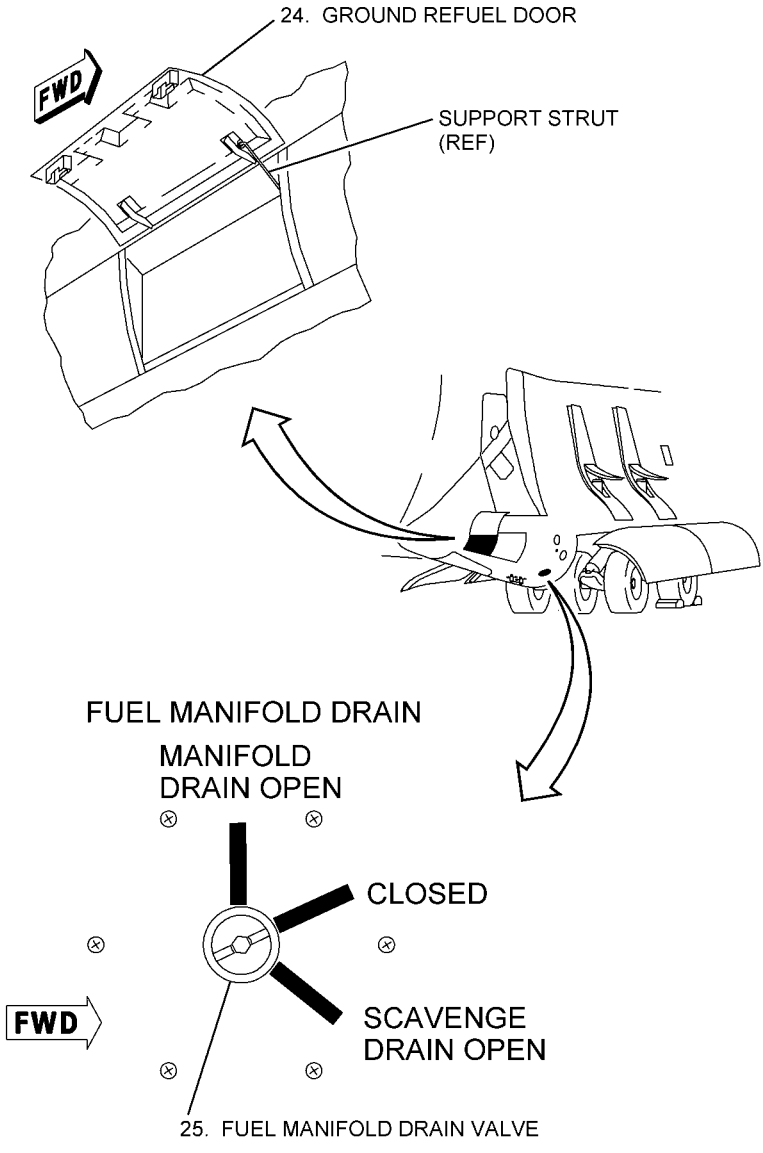
(TYPICAL)

ICN-88277-G1228419-004-01

24. (B) Unlatch and open ground refuel door (183JRD); install support strut.

25. (B) Ensure **FUEL MANIFOLD DRAIN** valve is in **CLOSED** position.

26. Perform fuel team briefing (TO 1300i-2-12JG-28-1, 12-28-00, para 1-5).



ICN-88277-G1228420-004-01

01-3. SINGLE POINT REFUELING - QUANTITY SELECT METHOD.

1. (B) Set **POWER** switch on Ground Refueling Control (GRC) panel to **REFUEL**.

NOTE

GRC Panel lamp test lights will not illuminate when the APU is running with the forward or aft boost pumps operating.

2. (B) Set **TEST SELECT** switch to **LAMP TEST**.

   - All panel lights come on.

NOTE

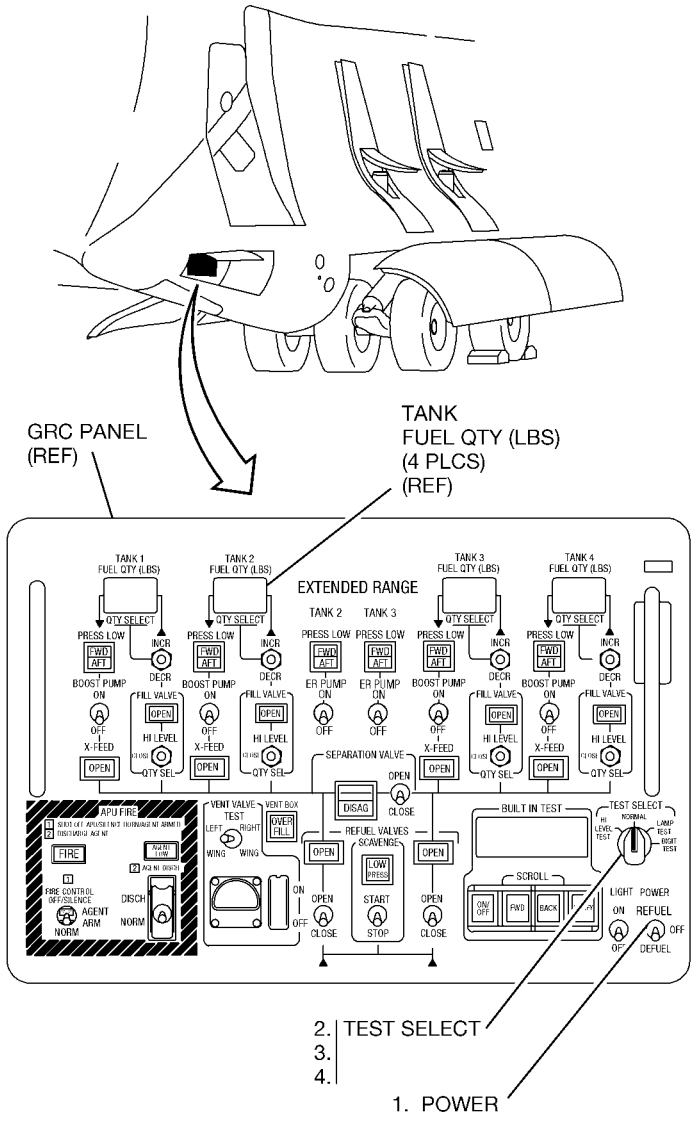
Before the **TEST SELECT** switch is placed back to **NORMAL**, the **DIGIT TEST** will cycle through the 3 bullets in step 3 continuously.

3. (B) Set **TEST SELECT** switch to **DIGIT TEST**.

   - All **TANK FUEL QTY (LBS)** indicators come on for 3 seconds.
   - All **TANK FUEL QTY (LBS)** indicators display all 8's for 3 seconds.
   - All **TANK FUEL QTY (LBS)** indicators go off for 2 seconds.

4. (B) Set **TEST SELECT** switch to **NORMAL**.

   - All **TANK FUEL QTY (LBS)** indicators display fuel quantity in each tank.



ICN-88277-G1228280-001-01

CAUTION

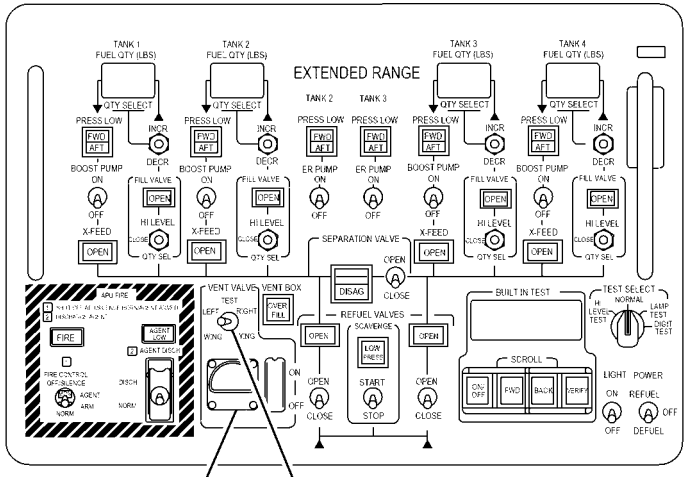
Right wing and left wing pressure shall be monitored during refueling operations and shall not exceed 1.7 psig. When pressure exceeds 1.7 psig, stop refueling. Failure to comply may cause damage to aircraft.

NOTE

The primary climb-dive valve normally prevents the wing pressures from exceeding 1.3 psig. When the primary valve fails, the secondary climb-dive valve should prevent the wing pressures from exceeding 1.7 psig. It is permissible to continue refueling with pressures above 1.3 psig, not to exceed 1.7 psig.

5. (B) Set **VENT VALVE TEST** switch for **LEFT WING** or **RIGHT WING** as required.

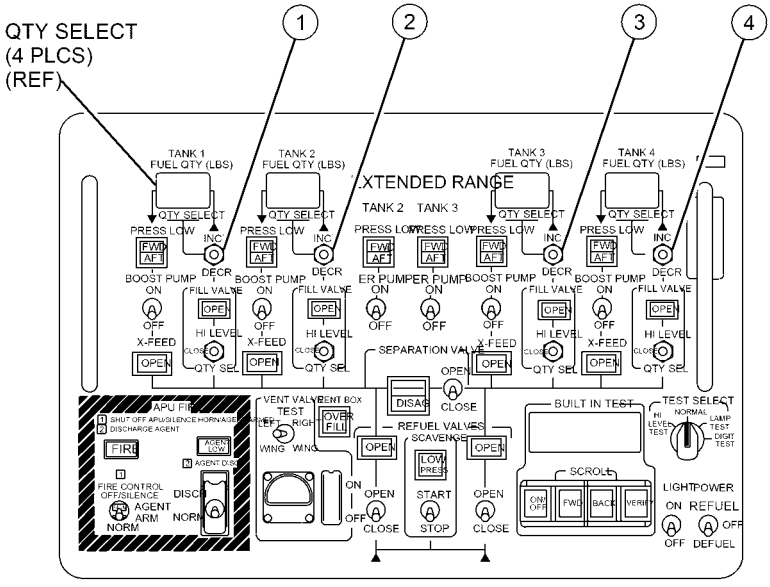
   - Pressure gauge reads 0.3-1.3 psig.



ICN-88277-G1228281-003-01

6. (B) Set **INCR/DECR** switch to required quantity.

    - **QTY SELECT** indicator displays quantity.



6.

INDEX NO.	TANK	SWITCH
1	1	INCR/DECR
2	2	INCR/DECR
3	3	INCR/DECR
4	4	INCR/DECR

ICN-88277-G1228282-001-01

NOTE

When multiple trucks are required to refuel, steps 7 through 11 shall be performed for both forward and aft locations prior to performing step 12.

7. (B) Press lock tab and remove fuel cap from ground refueling adapter.

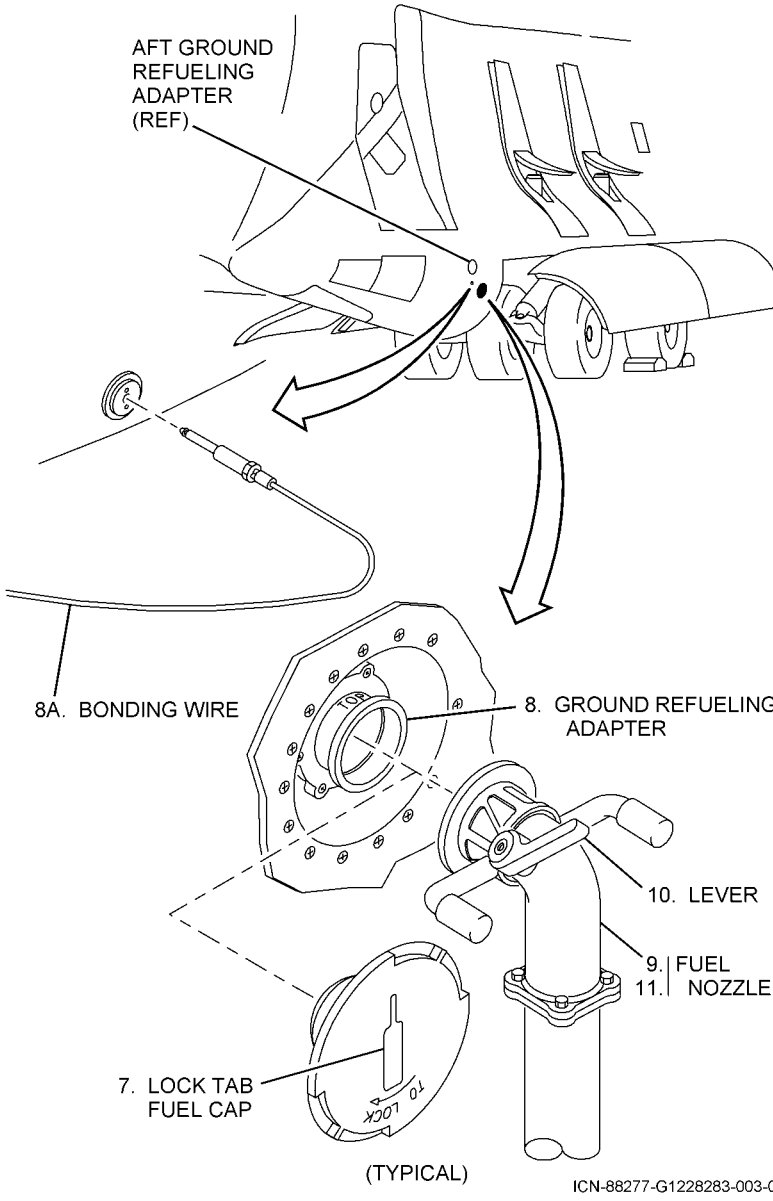
8. (B) Visually check ground refueling adapter for cracks, distortion, and excessive wear.

8A. (B) Ensure fuel servicing equipment is bonded to aircraft.

9. (B) Connect fuel nozzle to ground refueling adapter.

10. (B) Move lever to open position.

11. (B) Perform fuel nozzle lock-check by attempting to remove fuel nozzle with lever in open (flow) position.


AFT GROUND
REFUELING
ADAPTER
(REF)
8A. BONDING WIRE
8. GROUND REFUELING
ADAPTER
10. LEVER
9. FUEL
11. NOZZLE
7. LOCK TAB
FUEL CAP
TO LOCK
(TYPICAL)
ICN-88277-G1228283-003-01

NOTE

Left **REFUEL VALVES** switch correlates to the forward fuel receptacle. Right **REFUEL VALVES** switch correlates to the aft fuel receptacle.

12. (B) Set **REFUEL VALVES** switches to **OPEN**.

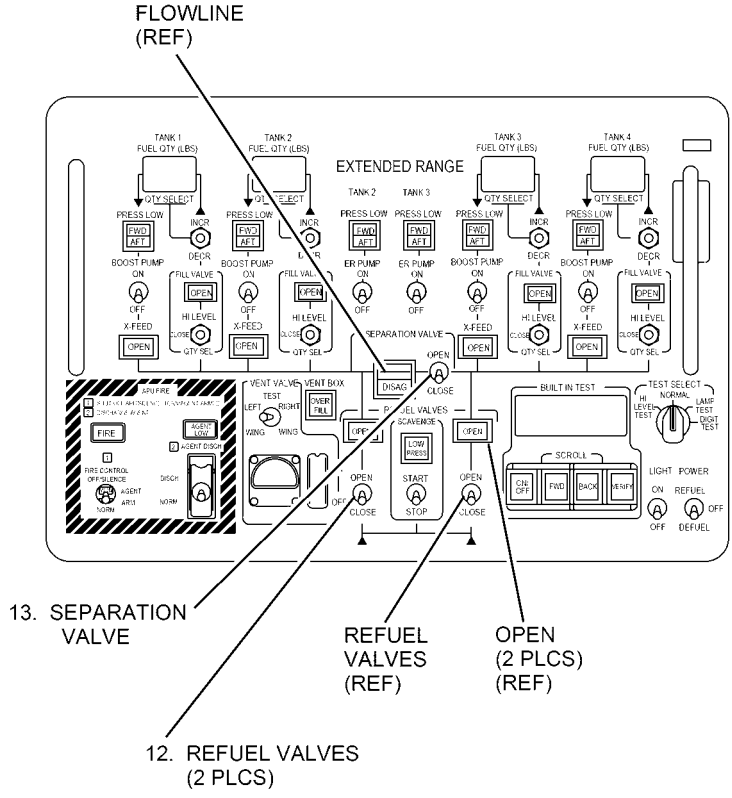
- **REFUEL VALVES OPEN** light comes on.

WARNING

When using both fuel receptacles, **SEPARATION VALVE** switch shall be set in the **CLOSE** position. Failure to close separation valve when using both receptacles may cause fuel spillage causing injury to personnel or damage to aircraft.

13. (B) Set **SEPARATION VALVE** switch as follows:

Quantity of Fuel Receptacles Used	Separation Valve Switch Position	Flowline Status
One	OPEN	On
Two	CLOSE	Off



ICN-88277-G1228284-003-01

14. (B) Set **FILL VALVE** switch to **HI LEVEL** and apply 50-55 psi fuel pressure.

    - Fuel flow starts.
    - **OPEN** light comes on.

15. (B) Set **TEST SELECT** switch to **HI LEVEL TEST**.

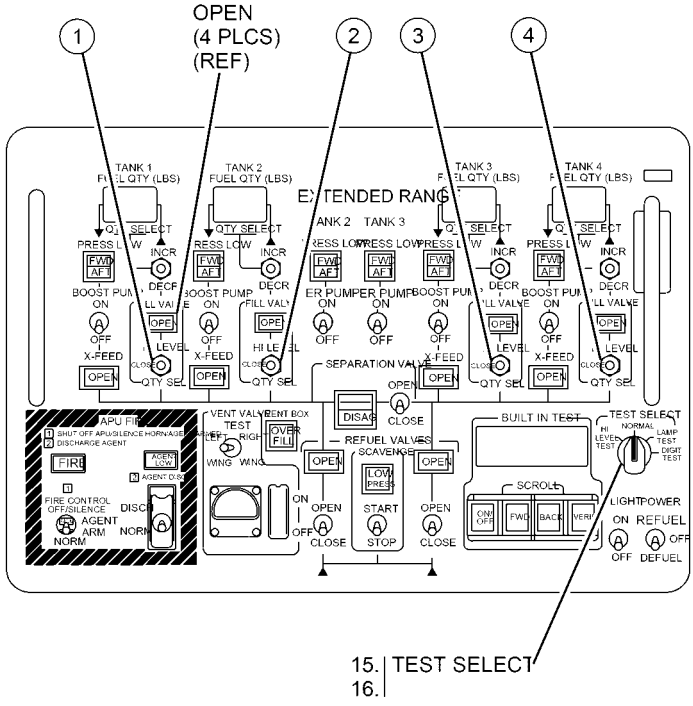
    - **OPEN** light goes off.
    - Fuel flow stops.

16. (B) Set **TEST SELECT** switch to **NORMAL**.

    - Test cycle stops.
    - **OPEN** light comes on.
    - Fuel flow starts.

17. (B) Set **FILL VALVE** switch to **CLOSE**.

    - Fuel flow stops.
    - **OPEN** light goes off.



14.
17.

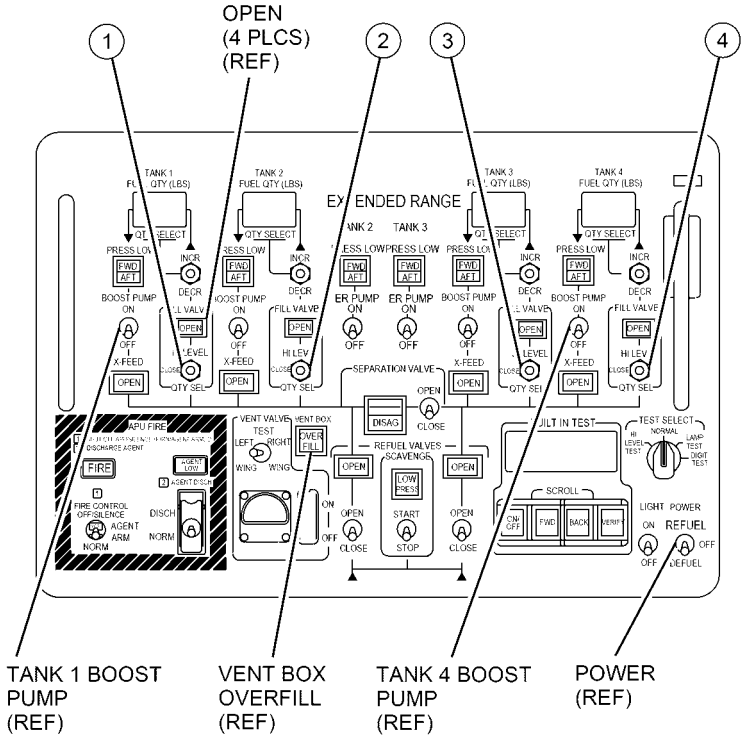
INDEX NO.	TANK	SWITCH
1	1	FILL VALVE
2	2	FILL VALVE
3	3	FILL VALVE
4	4	FILL VALVE

ICN-88277-G1228011-010-01

18. (B) Set **FILL VALVE** switch to **QTY SEL**.

- **OPEN** light comes on.
- Fuel flow starts.

SPECIAL INSTRUCTION. When **VENT BOX OVERFILL** light on GRC panel illuminates, immediately shutdown refueling operations. Set **POWER** switch to the **DEFUEL** position. Set **TANK 1** and **TANK 4 BOOST PUMP** switches to **ON** and allow pumps to run for ten minutes after **VENT BOX OVERFILL** light is extinguished. Boost pumps will not operate using battery power. Each vent box may contain approximately 35 gallons of fuel. Discontinue fuel servicing until cause of overfill can be determined and corrected.



18.

INDEX NO.	TANK	SWITCH
1	1	FILL VALVE
2	2	FILL VALVE
3	3	FILL VALVE
4	4	FILL VALVE

ICN-88277-G1228012-010-01

CAUTION

All **FILL VALVES** shall be closed (open lights off) and fuel flow stopped prior to closing the **SEPARATION** and **REFUEL VALVES**. Failure to comply may cause damage to the aircraft.

19. (A,B) Monitor **TANK FUEL QTY (LBS)** indicator and fueling operation until fuel tank(s) reach quantity selected.

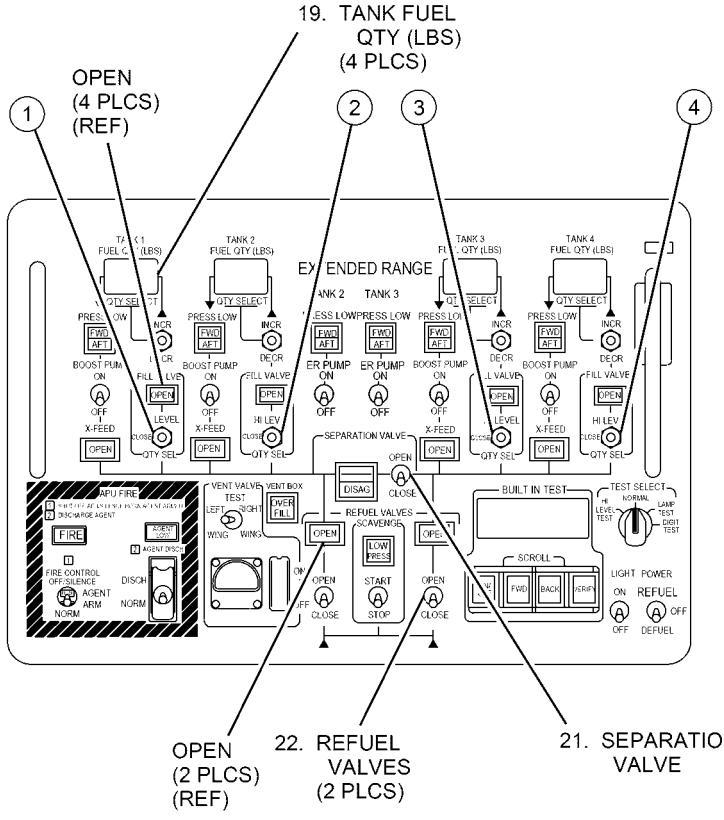
    - **OPEN** light goes off.
    - Fuel flow stops.

20. (B) Set **FILL VALVE** switch to **CLOSE**.

21. (B) Set **SEPARATION VALVE** switch to **CLOSE**.

22. (B) Set **REFUEL VALVES** switches to **CLOSE**.

    - **OPEN** lights goes off.

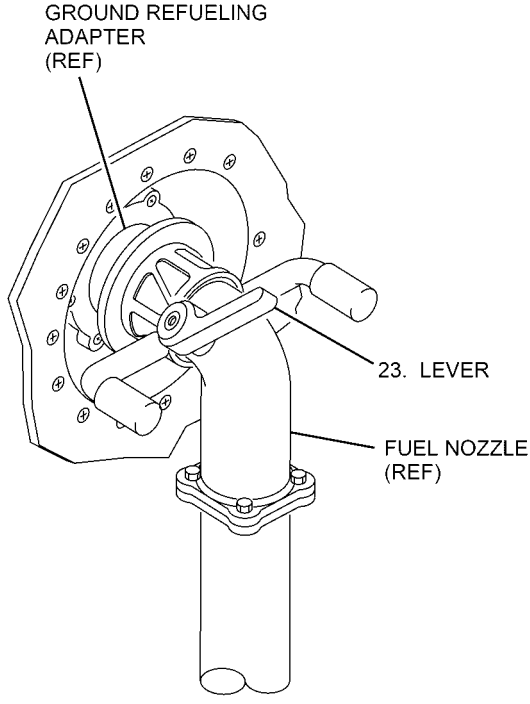


20.

INDEX NO.	TANK	SWITCH
1	1	FILL VALVE
2	2	FILL VALVE
3	3	FILL VALVE
4	4	FILL VALVE

ICN-88277-G1228028-012-01

23. (B) Shutdown fuel servicing equipment and move fuel nozzle lever to closed position.



(TYPICAL)

ICN-88277-G1228013-012-01

NOTE

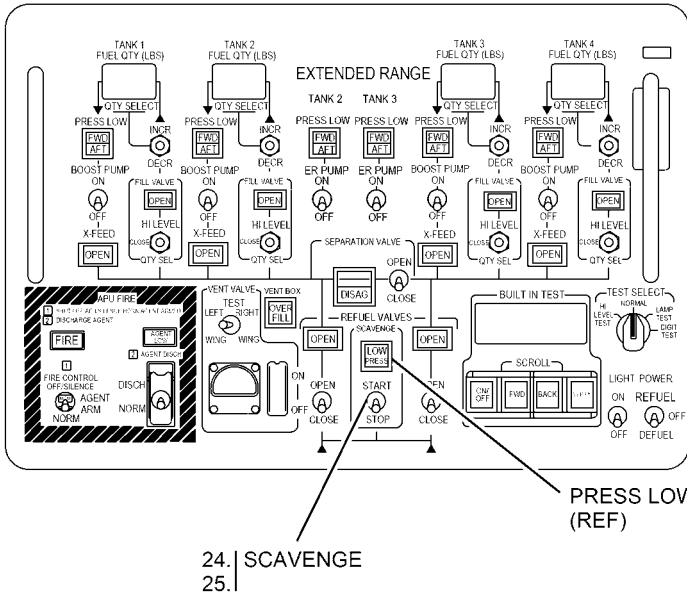
SCAVENGE PRESS LOW light will intermittently come on for approximately 30 seconds and then come on steady when fuel scavenge is complete.

24. (B) Set **SCAVENGE** switch on GRC panel to **START**.

    - **PRESS LOW** light comes on steady for one minute.

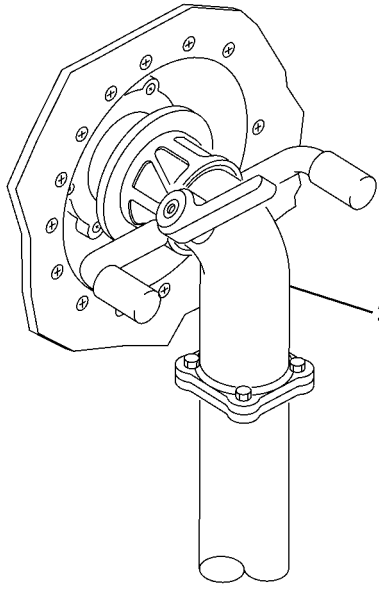
25. (B) Set **SCAVENGE** switch to **STOP**.

    - **PRESS LOW** light goes off.



ICN-88277-G1228015-015-01

26. (B) Disconnect fuel nozzle from ground refueling adapter.

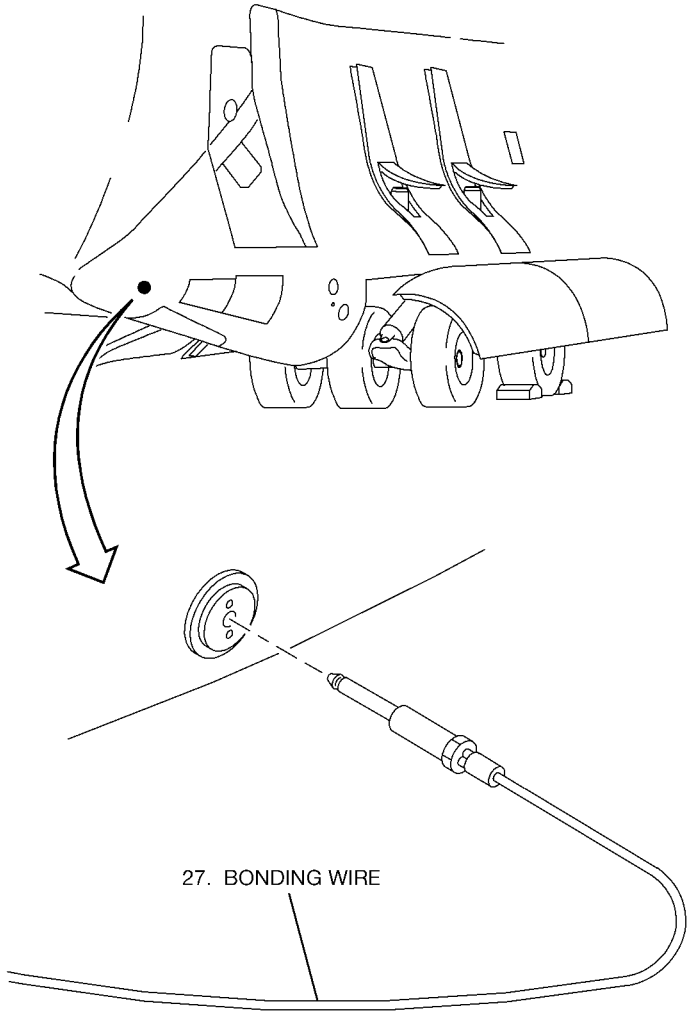


(TYPICAL)

ICN-88277-G1228446-001-01

27. (A) Disconnect fuel servicing equipment bonding wire from aircraft and remove fuel servicing equipment.

28. (A,B) Perform follow-on maintenance (task 01-6 or 01-7), as applicable.



ICN-88277-G1228014-010-01

01-4. SINGLE POINT REFUELING - HIGH LEVEL METHOD.

1. (B) Press lock tab and remove fuel cap from ground refueling adapter.

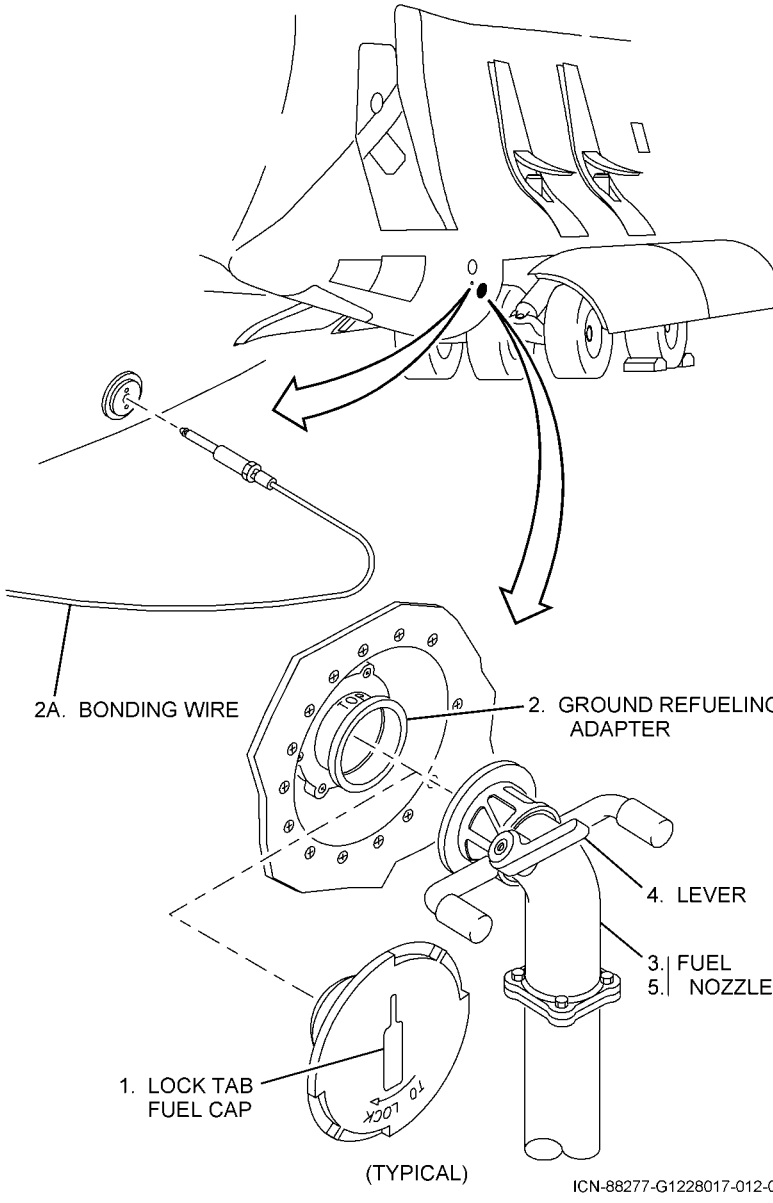
2. (B) Visually check ground refueling adapter for cracks, distortion, and excessive wear.

2A. (B) Ensure fuel servicing equipment is bonded to aircraft.

3. (B) Connect fuel nozzle to ground refueling adapter.

4. (B) Move lever to open position.

5. (B) Perform fuel nozzle lock-check by attempting to remove fuel nozzle with lever in open (flow) position.


2A. BONDING WIRE
2. GROUND REFUELING ADAPTER
4. LEVER
3. FUEL
5. NOZZLE
1. LOCK TAB
FUEL CAP
TO LOCK
(TYPICAL)
ICN-88277-G1228017-012-01

6. (B) Set **POWER** switch on Ground Refueling Control (GRC) panel to **REFUEL**.

7. (B) Set **TEST SELECT** switch to **LAMP TEST**.

   - All panel lights come on.

NOTE

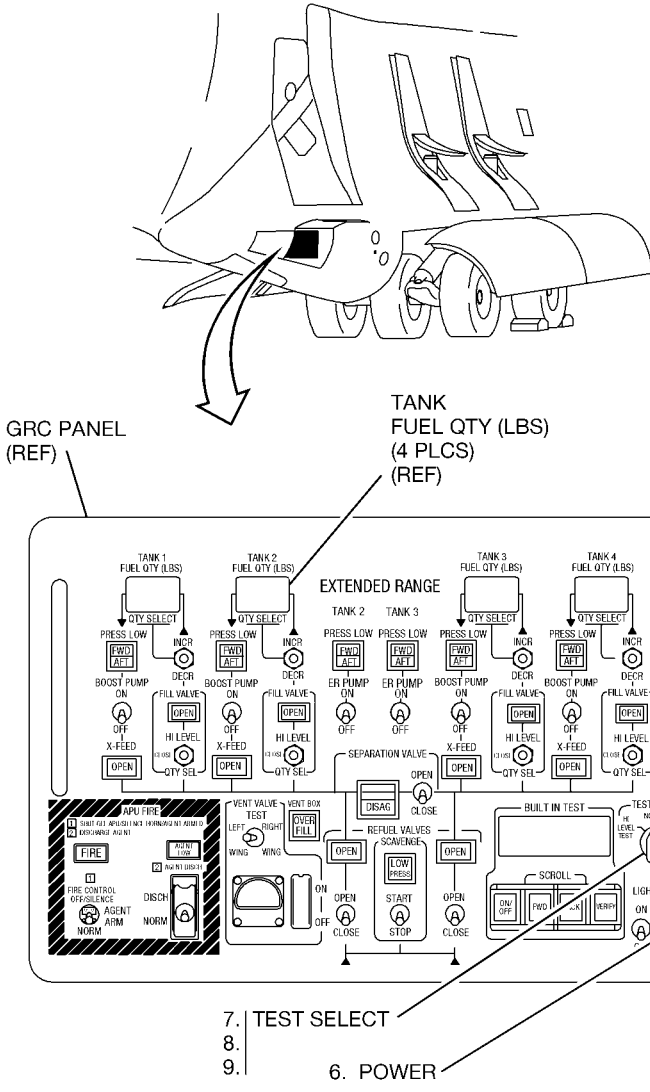
This sequence will repeat itself until **TEST SELECT** switch is set to **NORMAL**.

8. (B) Set **TEST SELECT** switch to **DIGIT TEST**.

   - All **TANK FUEL QTY (LBS)** indicators come on for 3 seconds.
   - All **TANK FUEL QTY (LBS)** indicators display all 8's for 3 seconds.
   - All **TANK FUEL QTY (LBS)** indicators go off for 2 seconds.

9. (B) Set **TEST SELECT** switch to **NORMAL**.

   - All **TANK FUEL QTY (LBS)** indicators display fuel quantity in each tank.



ICN-88277-G1228018-012-01

CAUTION

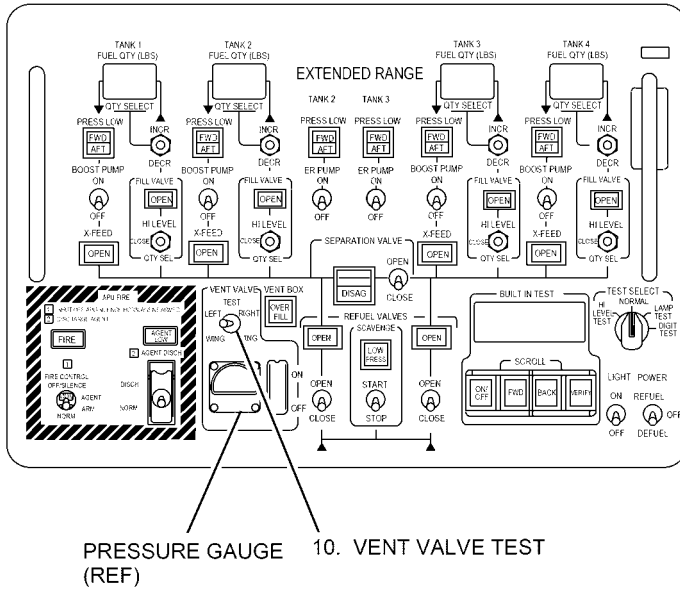
Right wing and left wing pressure must be monitored during refueling operations and shall not exceed 1.3 psig. When pressure exceeds 1.3 psig, stop refueling. Failure to comply may cause damage to aircraft.

NOTE

When the primary climb and dive valve has failed, it is permissible to continue refueling when wing pressure does not exceed 1.8 psig and the secondary climb and dive valve is functional.

10. (B) Set **VENT VALVE TEST** switch for **LEFT WING/RIGHT WING** as required.

    - Pressure gauge reads 0.3-1.3 psig.



PRESSURE GAUGE (REF)

10. VENT VALVE TEST

ICN-88277-G1228149-008-01

NOTE

Left **REFUEL VALVES** switch correlates to the forward fuel receptacle. Right **REFUEL VALVES** switch correlates to the aft fuel receptacle.

11. (B) Set **REFUEL VALVES** switches to **OPEN**.

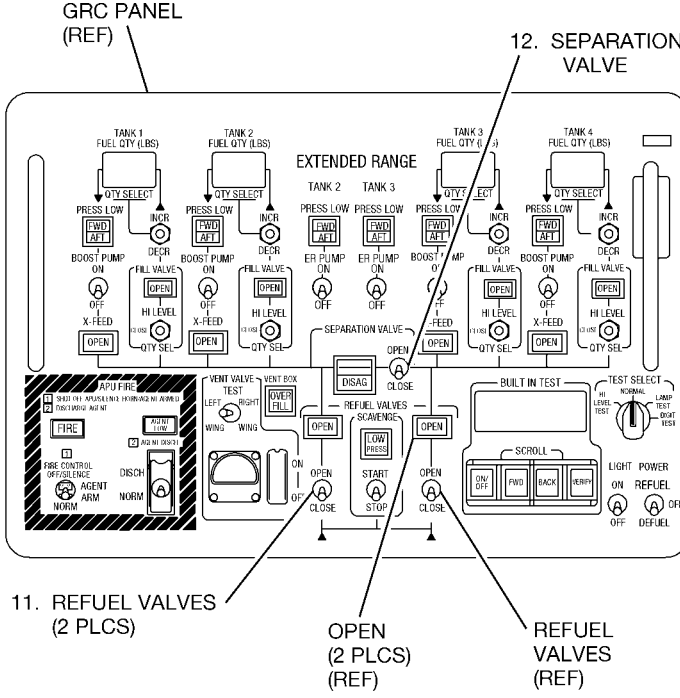
- **OPEN** light comes on.

WARNING

When using both fuel receptacles, **SEPARATION VALVE** switch shall be set in the **CLOSE** position. Failure to close separation valve when using both receptacles may cause fuel spillage causing injury to personnel or damage to aircraft.

12. (B) Set **SEPARATION VALVE** switch as follows:

Quantity of Fuel Receptacles Used	Separation Valve Switch Position	Flowline Status
One	OPEN	On
Two	CLOSE	Off



ICN-88277-G1228019-007-01

13. (B) Set **FILL VALVE** switch to **HI LEVEL** and apply 50-55 psi fuel pressure.

    - Fuel flow starts.
    - **OPEN** lights comes on.

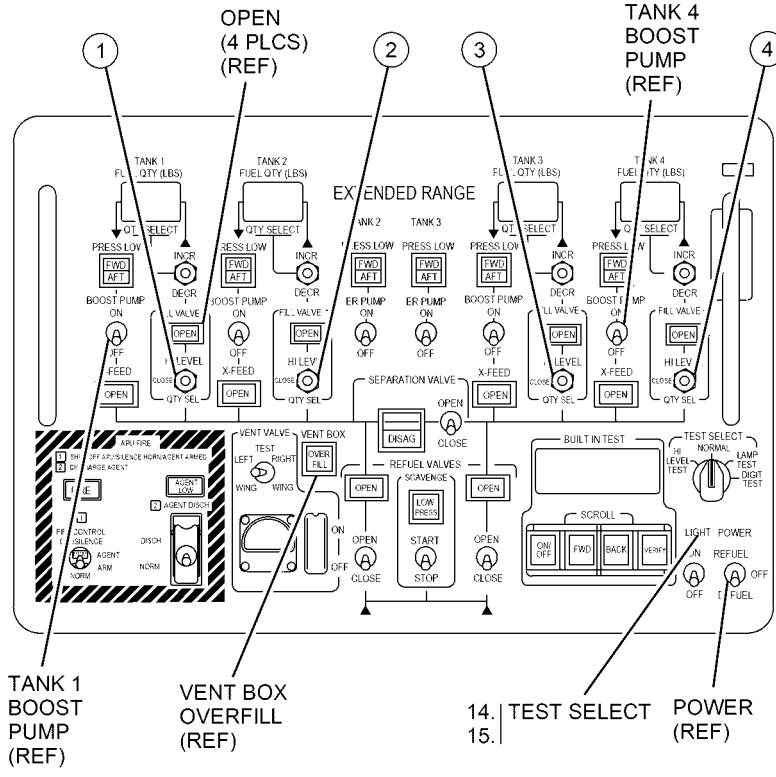
14. (B) Set **TEST SELECT** switch to **HI LEVEL TEST**.

    - **OPEN** lights goes off.
    - Fuel flow stops.

15. (B) Set **TEST SELECT** switch to **NORMAL**.

    - Test cycle stops.
    - **OPEN** lights comes on.
    - Fuel flow starts.

SPECIAL INSTRUCTION. When **VENT BOX OVERFILL** light on GRC panel illuminates, immediately shutdown refueling operations. Set **POWER** switch to the **DEFUEL** position. Set **TANK 1** and **TANK 4 BOOST PUMP** switches to **ON** and allow pumps to run for ten minutes after **VENT BOX OVERFILL** light is extinguished. Boost pumps will not operate using battery power. Each vent box may contain approximately 35 gallons of fuel. Discontinue fuel servicing until cause of overfill can be determined and corrected.



13.

INDEX NO.	TANK	SWITCH
1	1	FILL VALVE
2	2	FILL VALVE
3	3	FILL VALVE
4	4	FILL VALVE

ICN-88277-G1228020-009-01

CAUTION

All **FILL VALVES** shall be closed (open lights off) and fuel flow stopped prior to closing the **SEPARATION** and **REFUEL VALVES**. Failure to comply may cause damage to the aircraft.

16. (A,B) Monitor **TANK FUEL QTY (LBS)** indicator and fueling operation until fuel tanks reach high level shut off.

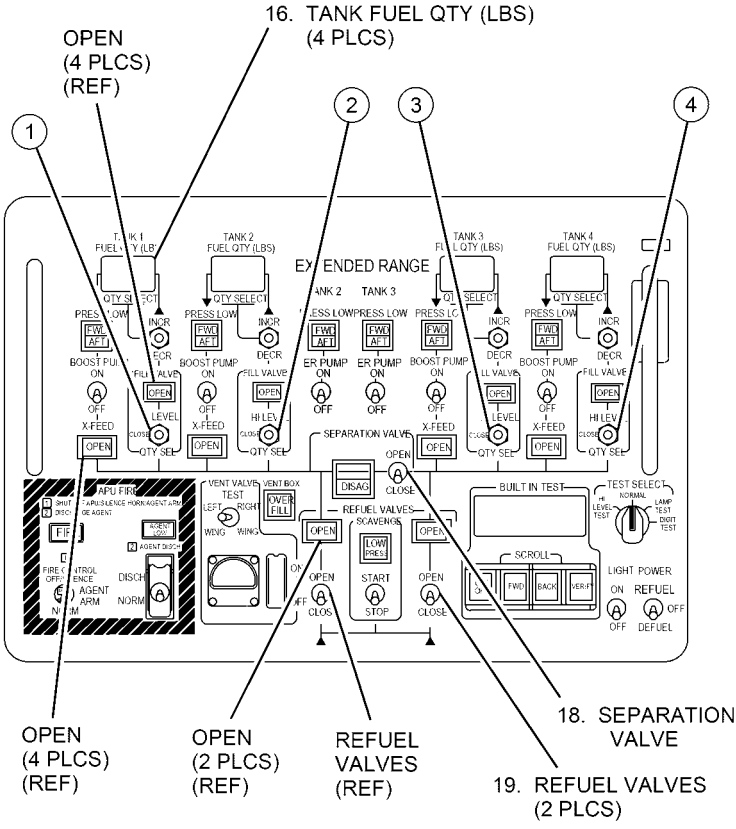
    - Fuel flow stops.
    - **OPEN** lights go off.

17. (B) Set **FILL VALVE** switches to **CLOSE**.

18. (B) Set **SEPARATION VALVE** switch to **CLOSE**.

19. (B) Set **REFUEL VALVES** switches to **CLOSE**.

    - **OPEN** lights go off.

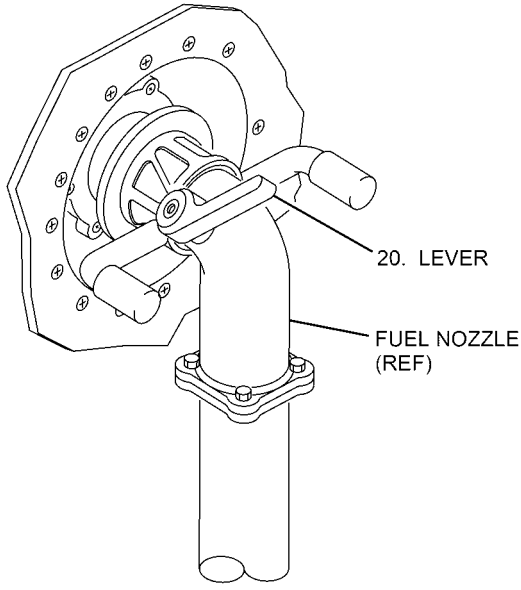


17.

INDEX NO.	TANK	SWITCH
1	1	FILL VALVE
2	2	FILL VALVE
3	3	FILL VALVE
4	4	FILL VALVE

ICN-88277-G1228021-011-01

20. (B) Shutdown fuel servicing equipment and move fuel nozzle lever to closed position.



(TYPICAL)

ICN-88277-G1228454-001-01

NOTE

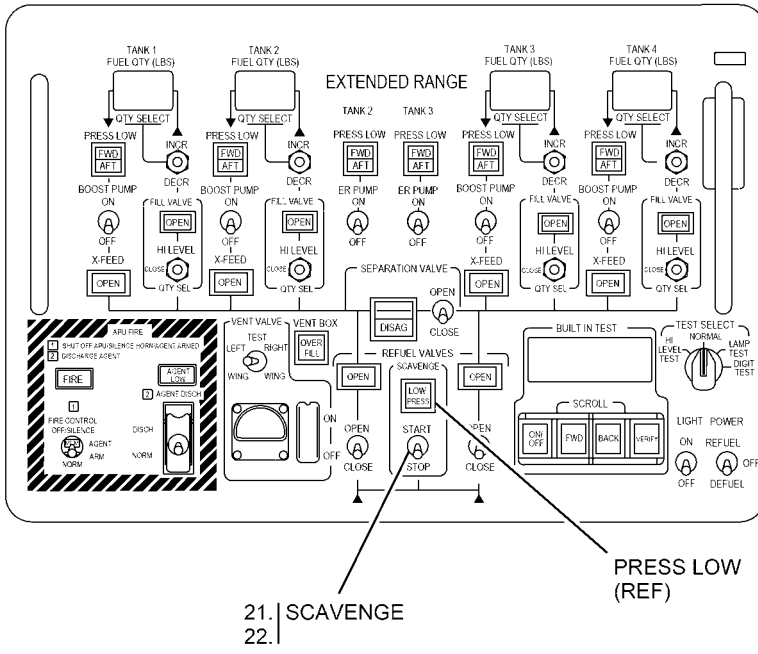
SCAVENGE PRESS LOW light will intermittently come on for approximately 30 seconds and then come on steady when fuel scavenge is complete.

21. (B) Set **SCAVENGE** switch on GRC panel to **START**.

    - **PRESS LOW** light comes on steady for one minute.

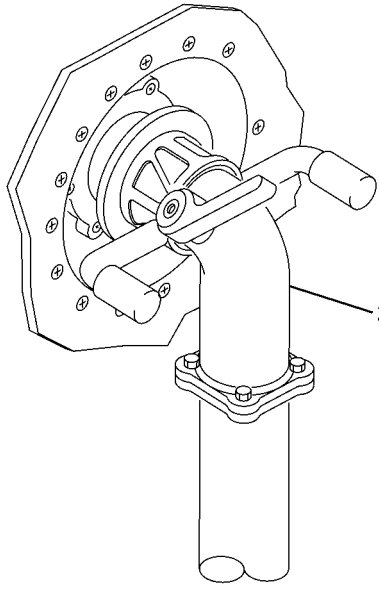
22. (B) Set **SCAVENGE** switch to **STOP**.

    - **PRESS LOW** light goes off.



ICN-88277-G1228024-011-01

23. (B) Disconnect fuel nozzle from ground refueling adapter.

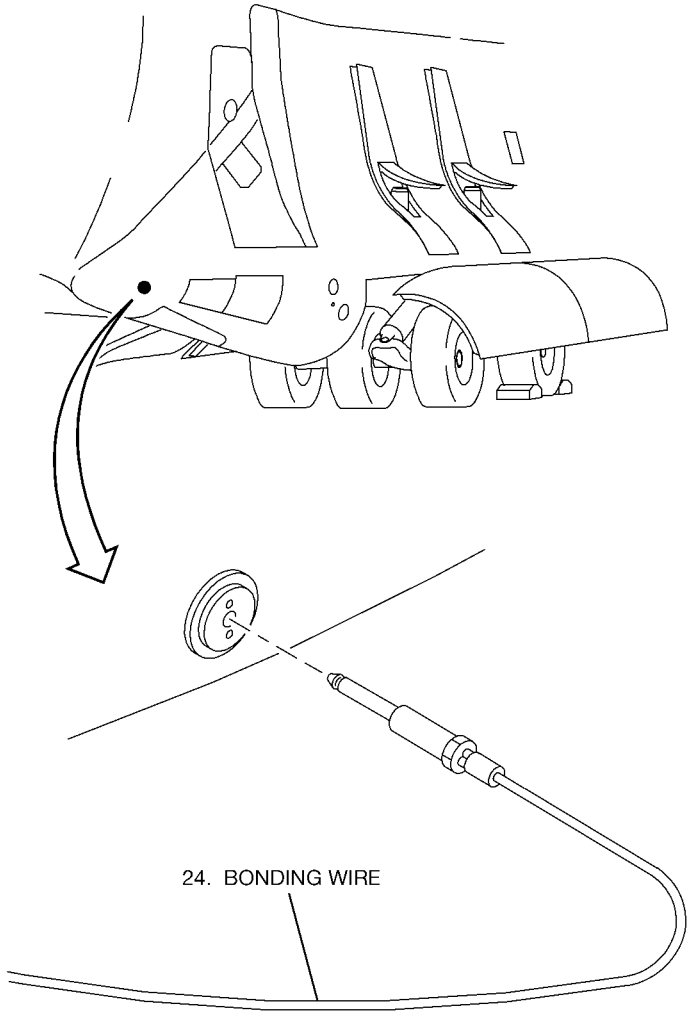


(TYPICAL)

ICN-88277-G1228022-010-01

24. (A) Disconnect fuel servicing equipment bonding wire from aircraft and remove fuel servicing equipment.

25. (A,B) Perform follow-on maintenance (task 01-6 or 01-7), as applicable.



ICN-88277-G1228023-009-01

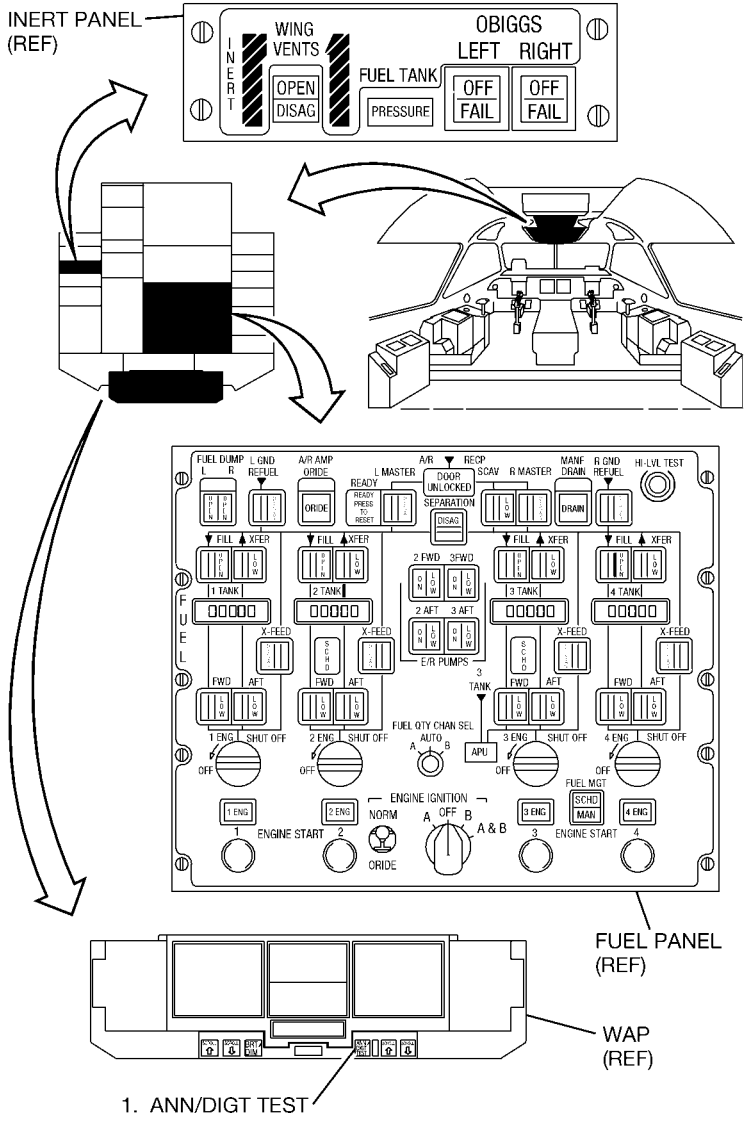
01-5. SINGLE POINT REFUELING - ALTERNATE METHOD USING OVERHEAD FUEL PANEL.

NOTE

This task shall be used when ground refueling panel is inoperative.

1. (B) Press and hold **ANN/DIGT TEST** switch on Warning and caution Annunciator Panel (WAP).

   - **FUEL** panel lights come on.
   - **INERT** panel lights come on.



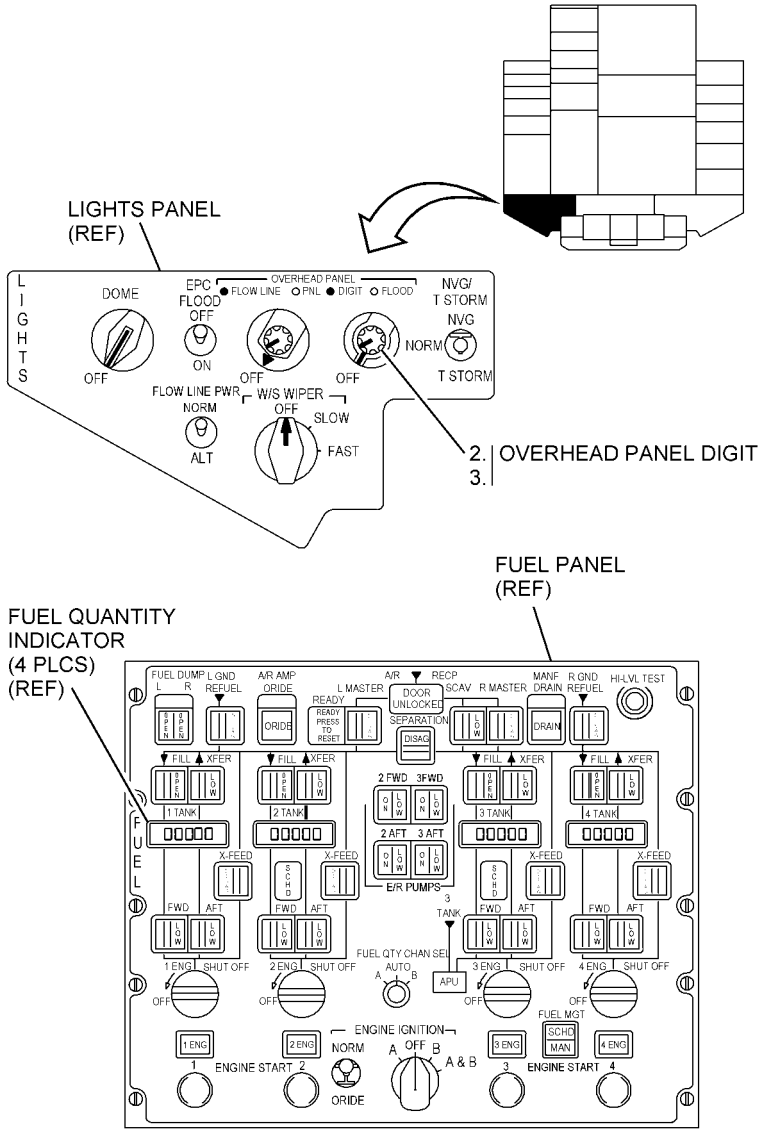
ICN-88277-G1228151-007-01

2. (B) Rotate **OVERHEAD PANEL DIGIT** knob on **LIGHTS** panel fully clockwise.

    - All fuel quantity indicators display all 8's.

3. (B) Rotate **OVERHEAD PANEL DIGIT** knob fully counter clockwise.

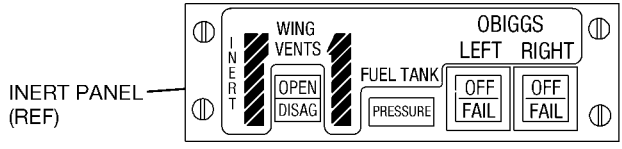
    - All 8's displayed on fuel quantity indicators go off.

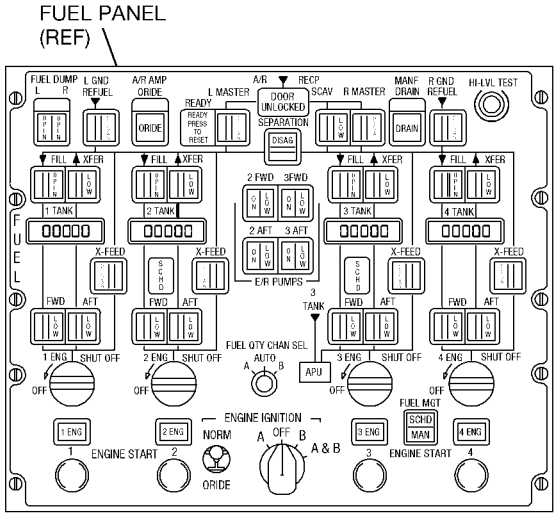


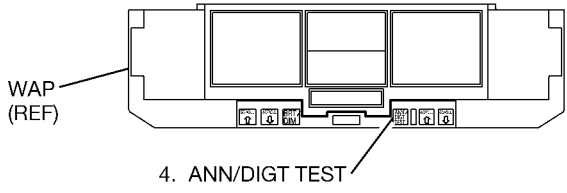
ICN-88277-G1228152-007-01

4. (B) Release **ANN/DIGT TEST** switch on WAP.

   - **FUEL** panel lights go off.
   - **INERT** panel lights go off.







ICN-88277-G1228153-007-01

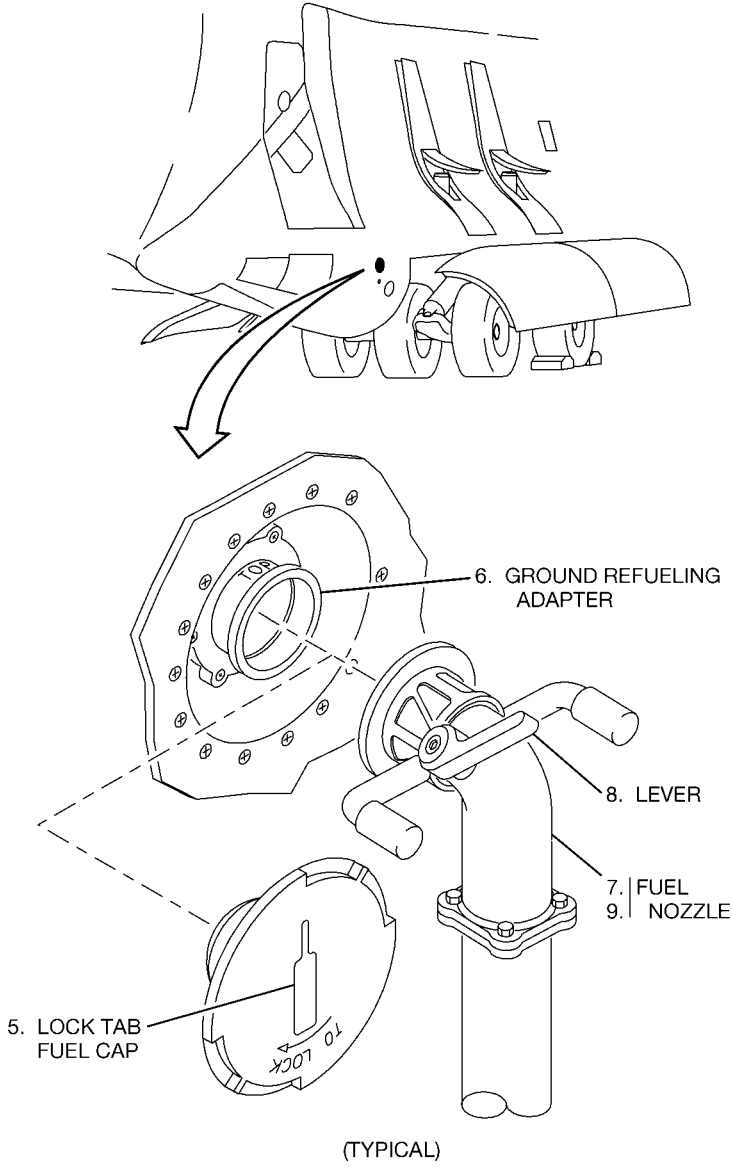
5. (A,C) Press lock tab and remove fuel cap from ground refueling adapter.

6. (A,C) Visually check ground refueling adapter for cracks, distortion, and excessive wear.

7. (A,C) Connect fuel nozzle to ground refueling adapter.

8. (A,C) Move lever to open position.

9. (A,C) Perform fuel nozzle lock-check by attempting to remove fuel nozzle with lever in open (flow) position.



ICN-88277-G1228155-008-01

10. (B) Press **L GND REFUEL** switchlight on **FUEL** panel.

    - Flowline comes on.

11. (B) Press **R GND REFUEL** switchlight.

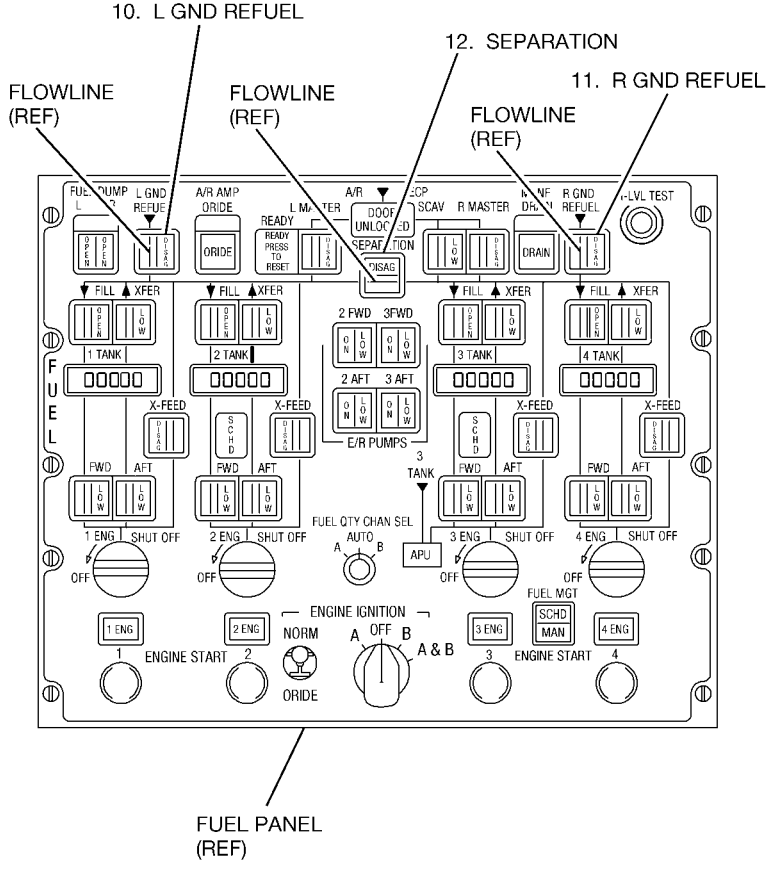
    - Flowline comes on.

WARNING

When using both fuel receptacles, the **SEPARATION VALVE** switchlight shall be set in the closed position. Failure to close **SEPARATION VALVE** when using both receptacles may cause fuel spillage causing injury to personnel or damage to aircraft.

12. (B) Set **SEPARATION VALVE** switchlight on **FUEL PANEL** as required.

Quantity of Fuel Receptacles Used	Separation Valve Switchlight Position	Flowline Status
One	OPEN (Depressed)	On
Two	CLOSE (Raised)	Off



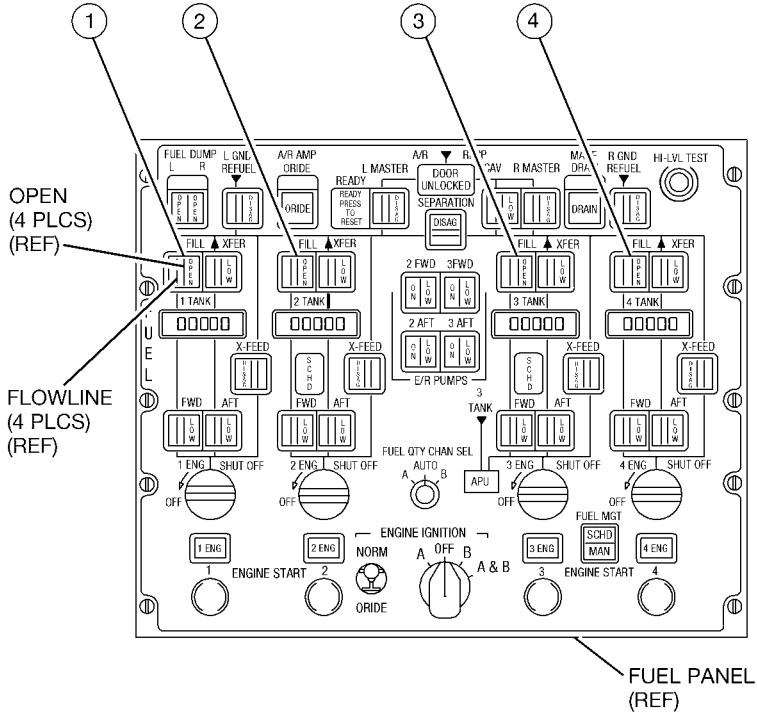
ICN-88277-G1228156-005-01

NOTE

As each fuel tank is filled to high level shut off, the fuel fill shut off valve will close automatically and the **OPEN** light will go off. In order to refuel a tank to a quantity other than full, the **FILL** switchlight must be manually pressed to the closed position to stop fuel flow once the desired fuel quantity is achieved.

13. (B,C) Press **FILL** switchlight and apply 50-55 psi fuel pressure.

- **OPEN** light comes on.
- Flowline comes on.
- Fuel flow starts.



13.

INDEX NO.	TANK	SWITCHLIGHT
1	1	FILL
2	2	FILL
3	3	FILL
4	4	FILL

ICN-88277-G1228157-005-01

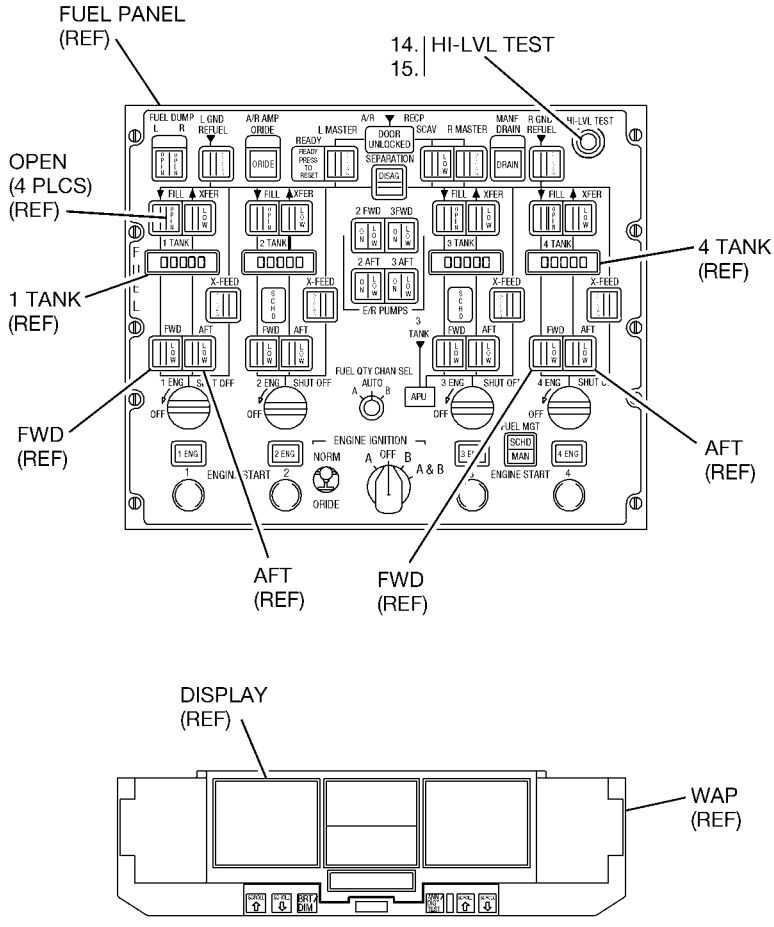
14. (B) Press and hold **HI-LVL TEST** switch.

    - **OPEN** lights go off.
    - Fuel flow stops.

15. (B) Release **HI-LVL TEST** switch.

    - **OPEN** lights come on.
    - Fuel flow starts.

    SPECIAL INSTRUCTION. When **FUEL OVERFILL** is displayed on WAP, immediately shutdown refueling operations. Press **1 TANK** and **4 TANK FWD** and **AFT** switchlights on **FUEL** panel and allow pumps to run for ten minutes after **FUEL OVERFILL** display on WAP goes off. Each vent box may contain approximately 35 gallons of fuel. Discontinue fuel servicing until cause of overfill can be determined and corrected.



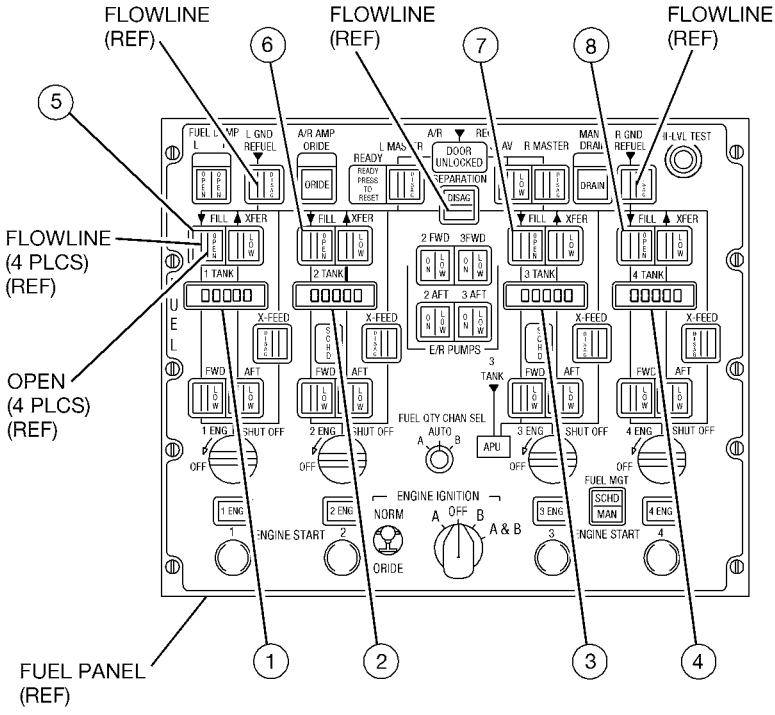
ICN-88277-G1228158-007-01

16. (A,B,C) Monitor **TANK** fuel quantity indicators and fueling operation until high level shut off or desired quantity is achieved.

All **FILL VALVES** shall be closed (open lights off) and fuel flow stopped prior to closing the **SEPARATION** and **REFUEL VALVES**. Failure to comply may cause damage to the aircraft.

17. (B) Press **FILL** switchlight.

- **OPEN** light goes off.
- Flowline goes off.
- Fuel flow stops.



16.

INDEX NO.	INDICATOR
1	1 TANK
2	2 TANK
3	3 TANK
4	4 TANK

17.

INDEX NO.	TANK	SWITCHLIGHT
5	1	FILL
6	2	FILL
7	3	FILL
8	4	FILL

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18. (B) Press **SEPARATION** switchlight as required.

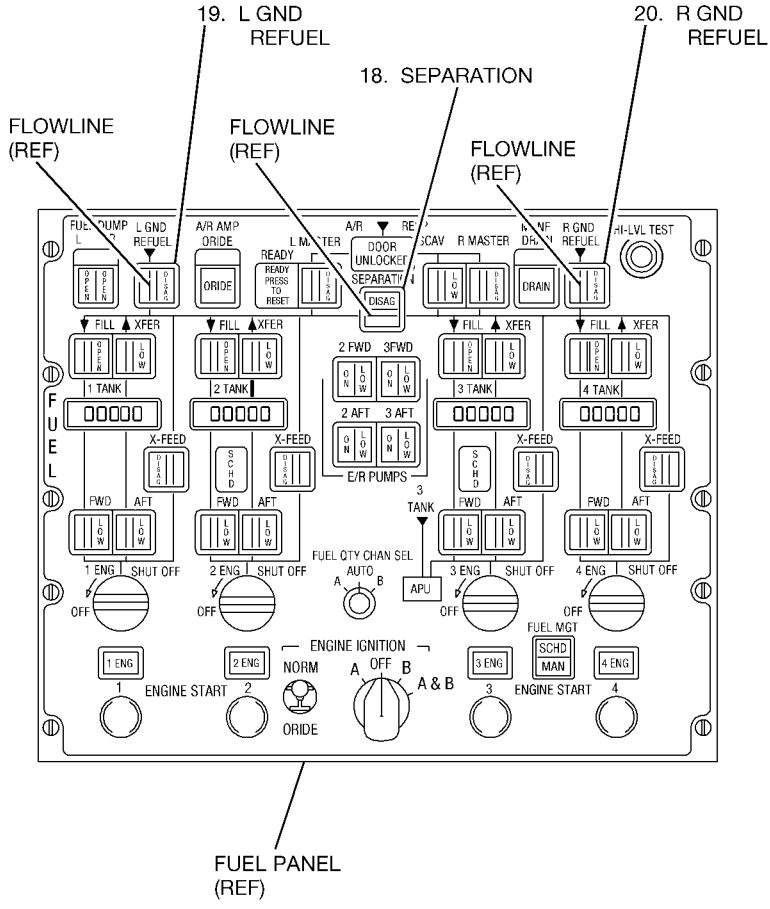
    - Flowline goes off.

19. (B) Press **L GND REFUEL** switchlight.

    - Flowline goes off.

20. (B) Press **R GND REFUEL** switchlight.

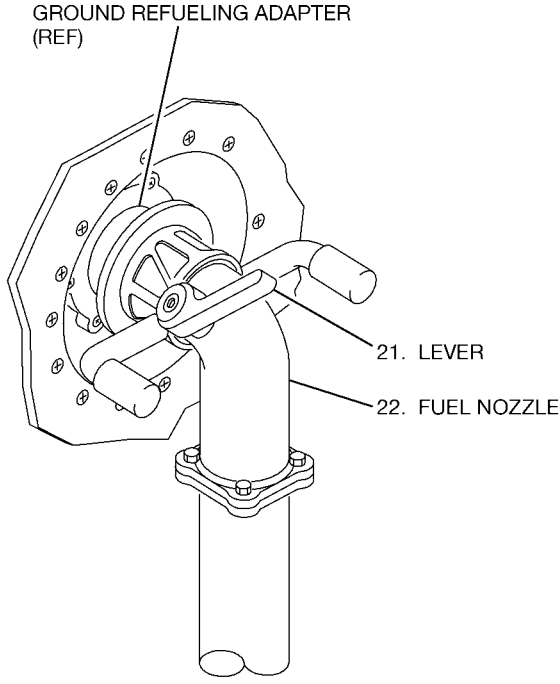
    - Flowline goes off.



ICN-88277-G1228374-005-01

21. (A) Shutdown fuel servicing equipment and move fuel nozzle lever to closed position.

22. (A) Disconnect fuel nozzle from ground refueling adapter.



(TYPICAL)

ICN-88277-G1228161-008-01

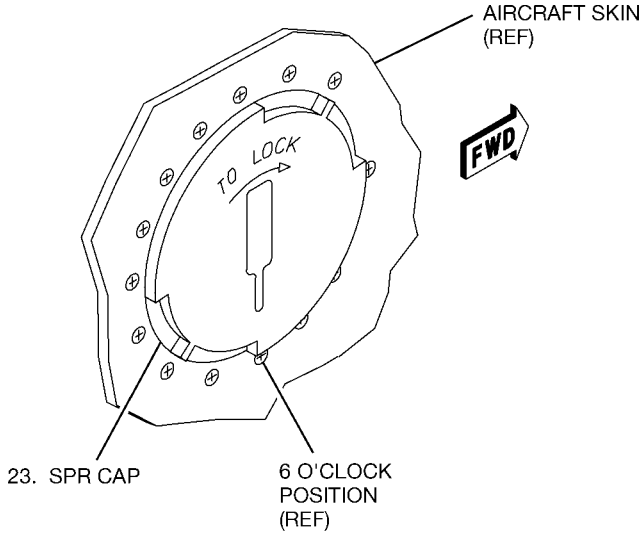
CAUTION

- Ensure lanyard is attached to the aircraft and fuel cap(s). Failure to comply may cause the loss of fuel cap(s) and damage to aircraft.

- Ensure fuel cap(s) are flush with the aircraft when installed. Failure to comply may cause the loss of fuel cap(s) and damage to aircraft.

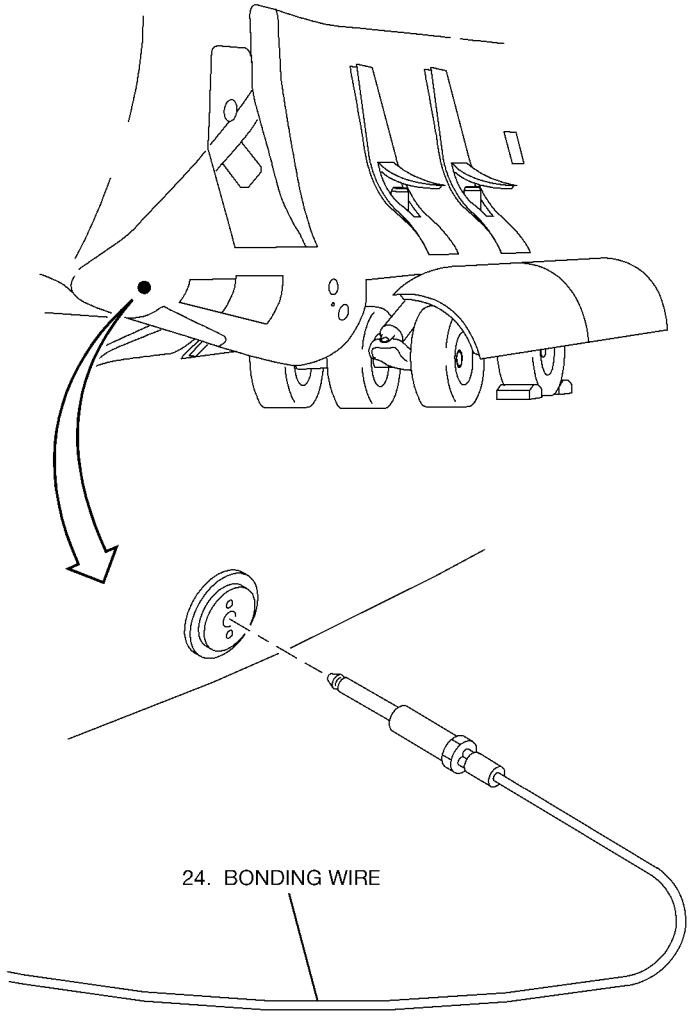
- Ensure fuel cap lanyard(s) are not exposed after fuel cap(s) installation. Failure to comply may cause loss of fuel cap(s) and damage to aircraft.

23. (C) Install Single Point Refueling (SPR) cap(s) with thin tab locked at the 6 o'clock.



ICN-88277-G1228445-001-01

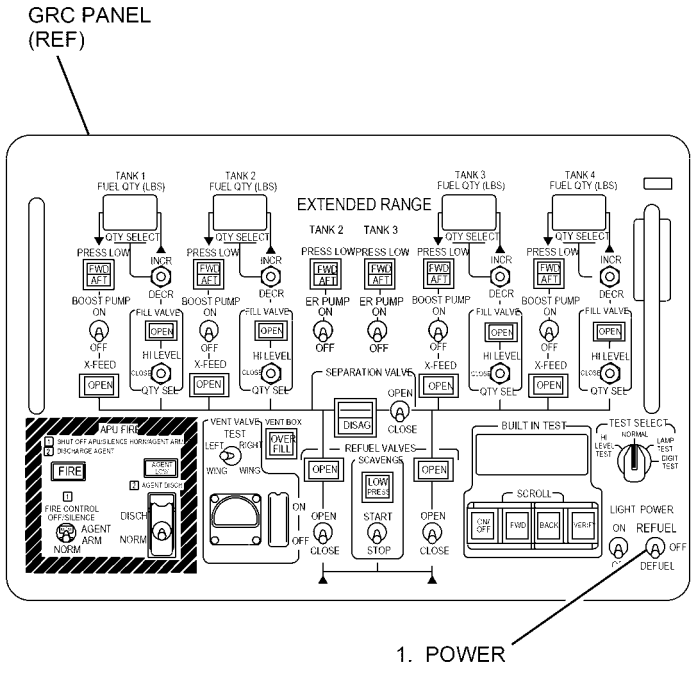
24. (A) Disconnect fuel servicing equipment bonding wire from aircraft and remove fuel servicing equipment.



ICN-88277-G1228162-006-01

01-6. FOLLOW-ON MAINTENANCE FOR SINGLE POINT REFUELING USING EXTERNAL OR AUXILIARY POWER UNIT POWER.

1. (B) Set **POWER** switch on Ground Refueling Control (GRC) panel to **OFF**.

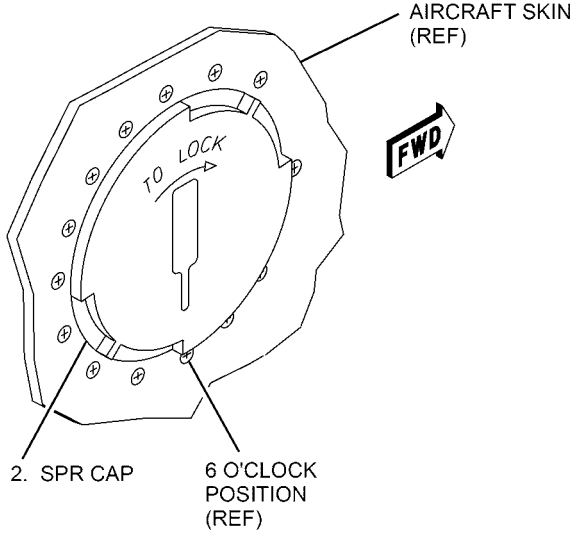


ICN-88277-G1228667-001-01

CAUTION

- Ensure lanyard is attached to the aircraft and fuel cap(s). Failure to comply may cause the loss of fuel cap(s) and damage to aircraft.
- Ensure fuel cap(s) are flush with the aircraft when installed. Failure to comply may cause the loss of fuel cap(s) and damage to aircraft.
- Ensure fuel cap lanyard(s) are not exposed after fuel cap(s) installation. Failure to comply may cause loss of fuel cap(s) and damage to aircraft.

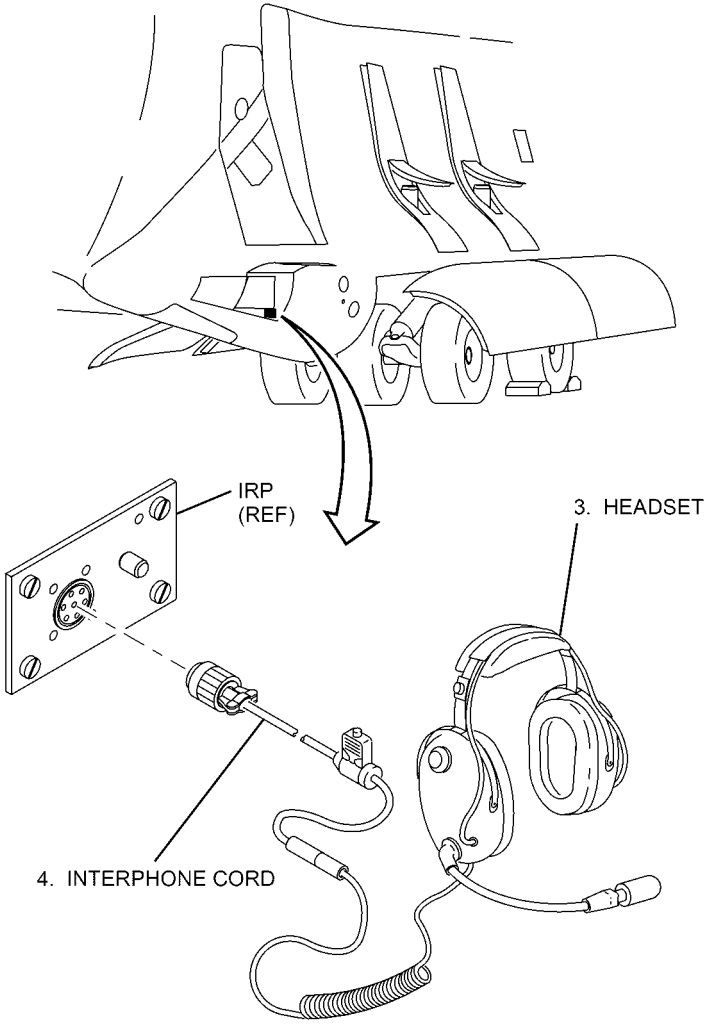
2. (B) Install Single Point Refueling (SPR) cap(s) with thin tab locked at the 6 o'clock.



ICN-88277-G1228668-001-01

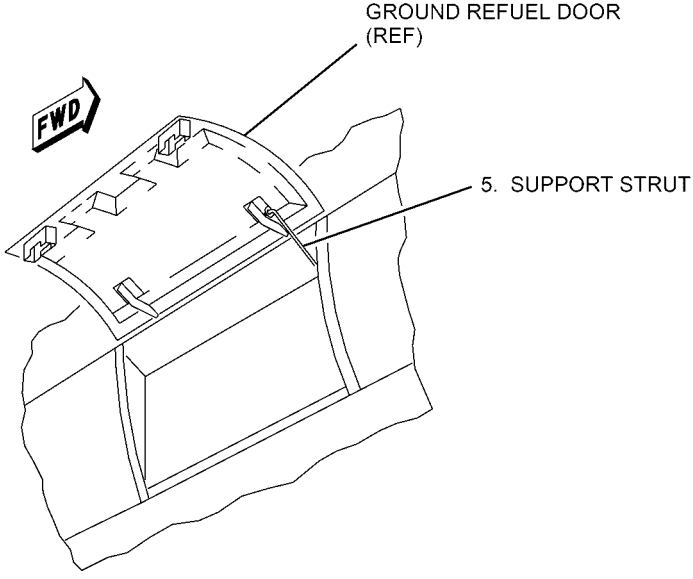
3. (B) Disconnect headset from interphone cord.

4. (B) Disconnect interphone cord from **IRP**.



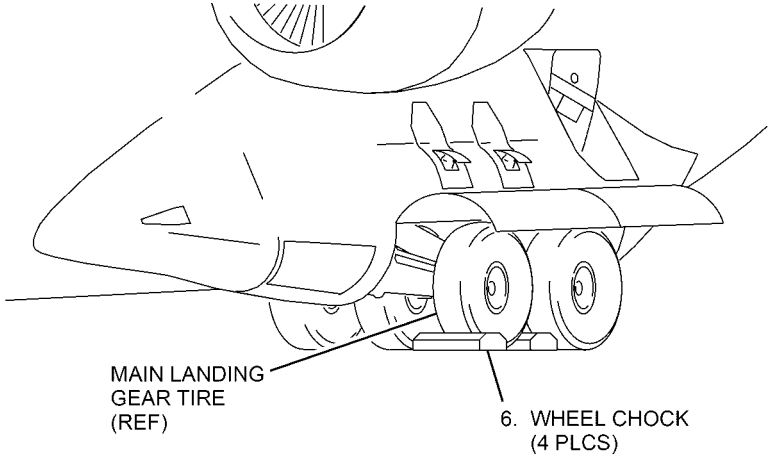
ICN-88277-G1228669-001-01

5. (B) Remove support strut; close and latch ground refuel door (183JRD).



ICN-88277-G1228670-001-01

6. (A) Position wheel chocks against main landing gear tires.



ICN-88277-G1228438-003-01

WARNING

- When drain is not conducted following refuel, visual inspection of the scavenge and manifold drain lines in the aft pod shall be conducted to ensure no presence of fuel (leakage) or fuel odor prior to aircraft departure. Failure to comply may cause injury to personnel and damage to aircraft.

- Metal fuel containers used for draining must be bonded to the aircraft prior to fuel draining. Failure to comply may cause injury to personnel and damage to aircraft.

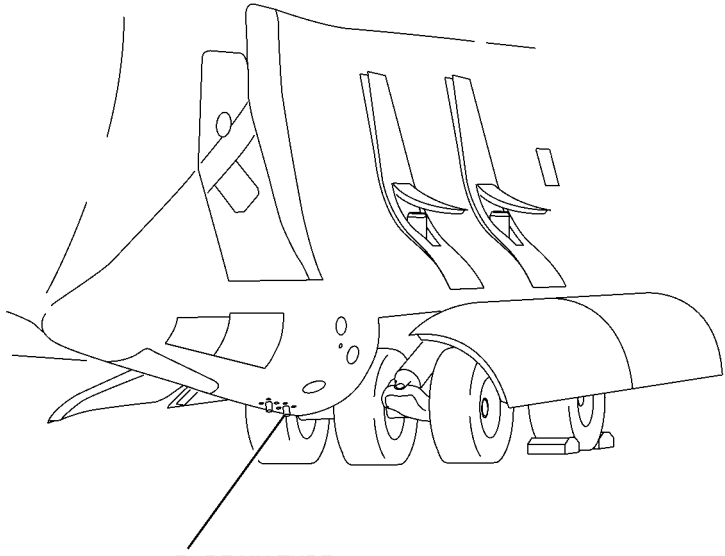
NOTE

- Personnel performing fuel servicing shall be familiar with TO 00-25-172 and AFMAN 91-203, Chapter 18.5.

- The manual fuel manifold drain procedure subjects personnel and the environment to an increased likelihood of fuel exposure. When any of the following conditions are met; approved and serviceable fuel manifold drain support equipment is not available, approved and serviceable personal protective equipment is not available or waste fuel cannot be disposed of by an approved method that meets local environmental protection requirements, steps 2 through 5 shall be delayed until the next refuel.

- Manifold may contain approximately 1 gallon of fuel, when using the alternate refuel method the ground refuel manifold contains approximately 45 gallons of fuel.

7. (B) Place fuel container under forward drain tube.



7. DRAIN TUBE

ICN-88277-G1228443-002-01

8. (A,B) Set **FUEL MANIFOLD DRAIN** valve to **MANIFOLD DRAIN OPEN** position.

   - Fuel drains from forward drain tube.

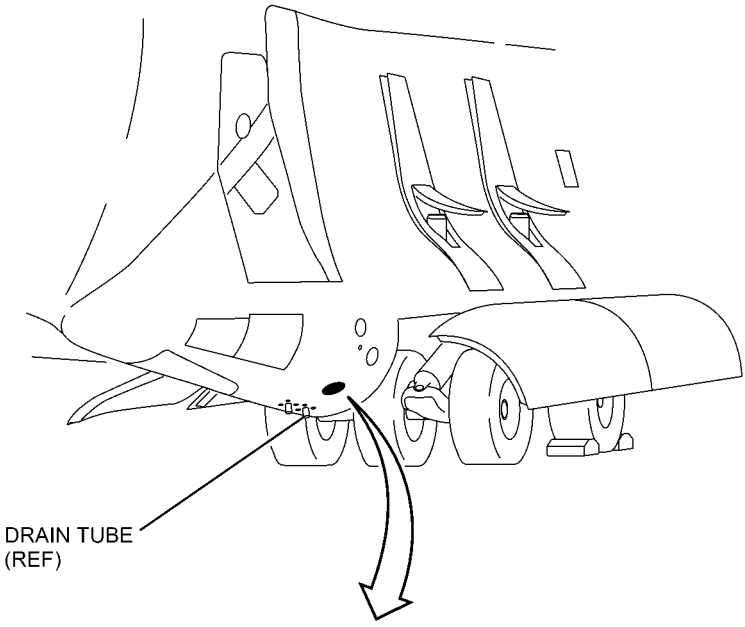
NOTE

- Scavenge lines may contain approximately 1 pint of fuel.
- When alternate method using overhead fuel panel was performed, scavenge drain not required.

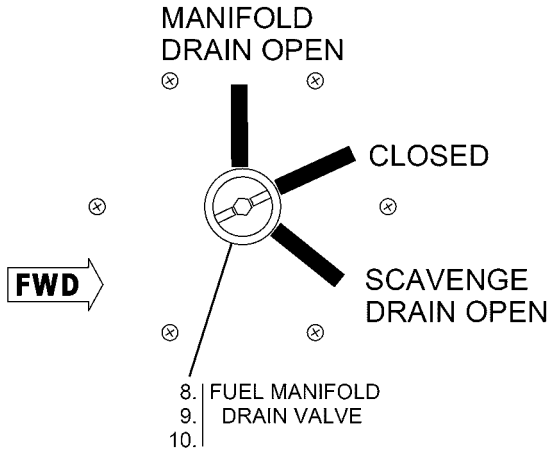
9. (A,B) Set **FUEL MANIFOLD DRAIN** valve to **SCAVENGE DRAIN OPEN** position.

   - Fuel drains from forward drain tube.

10. (A,B) Set **FUEL MANIFOLD DRAIN** valve to **CLOSED** position.



FUEL MANIFOLD DRAIN



ICN-88277-G1228444-002-01

WARNING

Before releasing parking brakes, make sure personnel, stands, vehicles, or obstructions are clear of nose landing gear, main landing gear, and aircraft. Aircraft will roll **FWD** or **AFT** depending on the apron surface and weather conditions. Reposition chocks against main landing gear tires if necessary after releasing brakes. Failure to comply may cause injury to personnel or damage to aircraft.

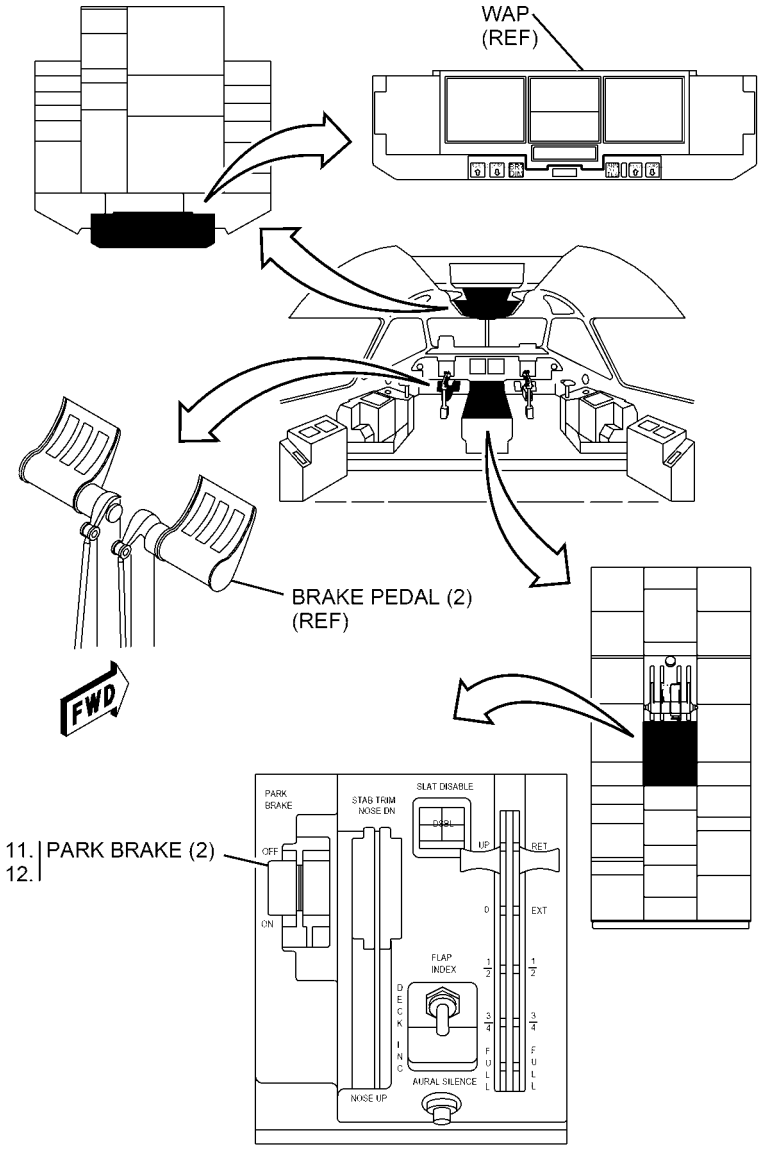
CAUTION

Personnel shall prevent park brake levers from moving to the forward position on their own during brake release. Failure to comply may cause damage to aircraft.

11. (B) While supporting **PARK BRAKE** levers, press and release brake pedals.

12. (B) Move **PARK BRAKE** levers to **OFF**.

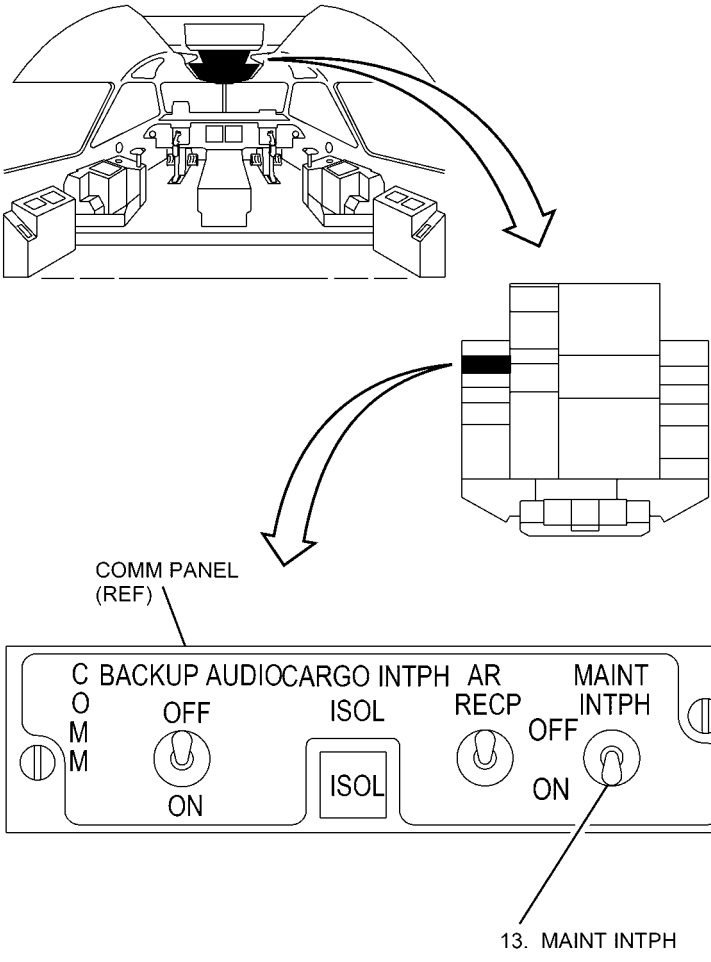
    - **PARK BRAKE L**, **R** display on Warning and caution Annunciator Panel (WAP) goes off.



ICN-88277-G1228436-005-01

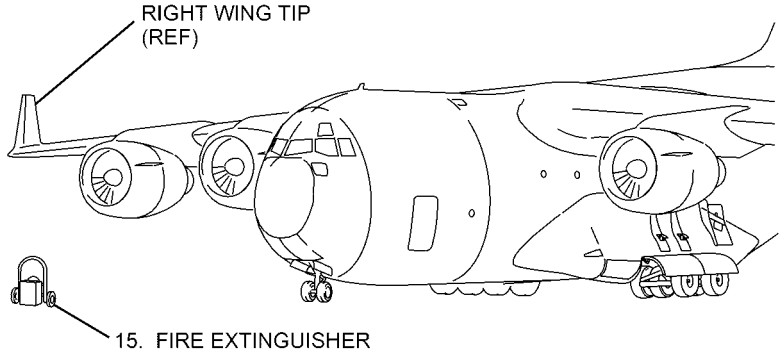
13. (B) Set **MAINT INTPH** switch on **COMM** panel to **OFF**.

14. Disconnect external electrical power (10-61-01, task 01-2).



ICN-88277-G1228481-002-01

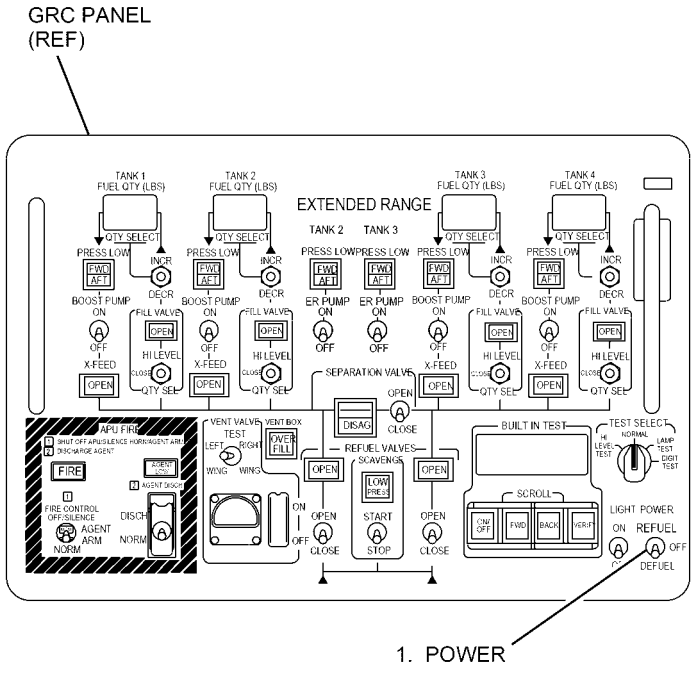
15. (A) Position fire extinguisher off right wing tip.



ICN-88277-G1228602-002-01

01-7. FOLLOW-ON MAINTENANCE FOR SINGLE POINT REFUELING USING AIRCRAFT BATTERY.

1. (B) Set **POWER** switch on Ground Refueling Control (GRC) panel to **OFF**.

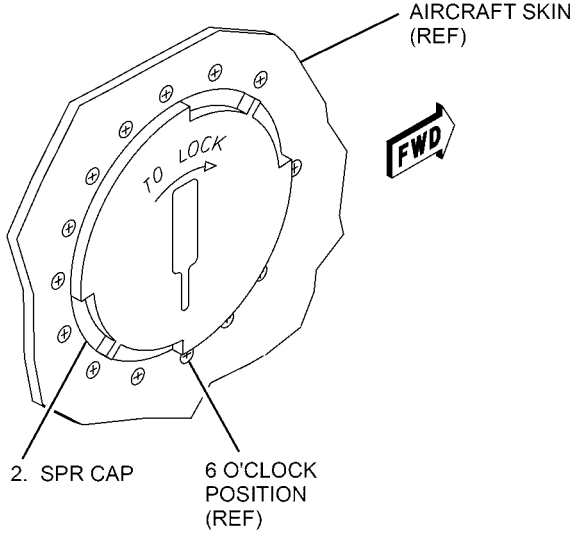


ICN-88277-G1228671-001-01

CAUTION

- Ensure lanyard is attached to the aircraft and fuel cap(s). Failure to comply may cause the loss of fuel cap(s) and damage to aircraft.
- Ensure fuel cap(s) are flush with the aircraft when installed. Failure to comply may cause the loss of fuel cap(s) and damage to aircraft.
- Ensure fuel cap lanyard(s) are not exposed after fuel cap(s) installation. Failure to comply may cause loss of fuel cap(s) and damage to aircraft.

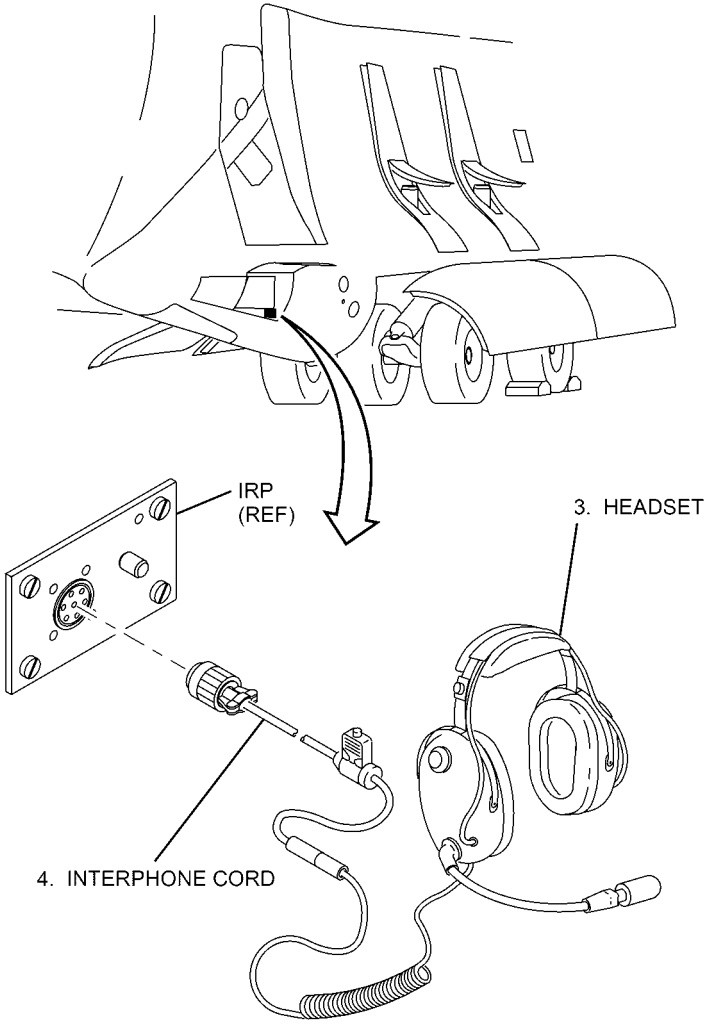
2. (B) Install Single Point Refueling (SPR) cap(s) with thin tab locked at the 6 o'clock.



ICN-88277-G1228672-001-01

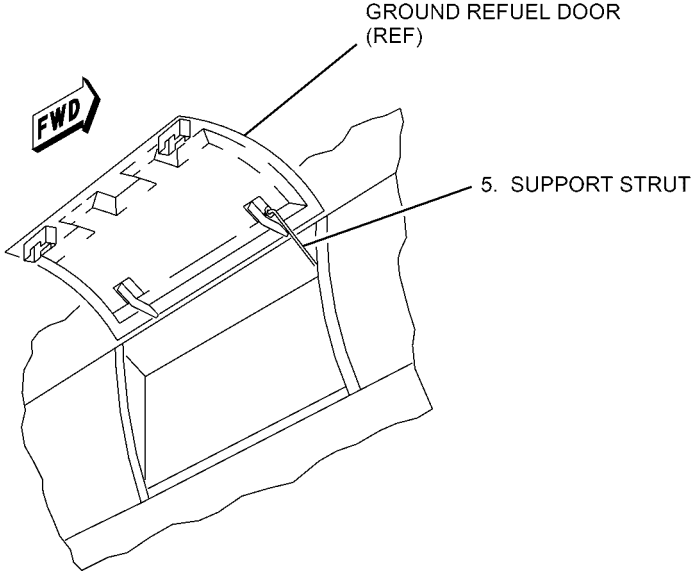
3. (B) Disconnect headset from interphone cord.

4. (B) Disconnect interphone cord from **IRP**.



ICN-88277-G1228673-001-01

5. (B) Remove support strut; close and latch ground refuel door (183JRD).



ICN-88277-G1228674-001-01

6. (A) Position wheel chocks against main landing gear tires.