

ICN-88277-G1228625-001-01

29. (A) Press **SKE** key.

- **SKE SET UP 1/2** is displayed.

NOTE

Pressing 1L LS key toggles between **OFF**, **STANDBY**, **XMIT** Station Keeping Equipment (SKE) modes.

NOTE

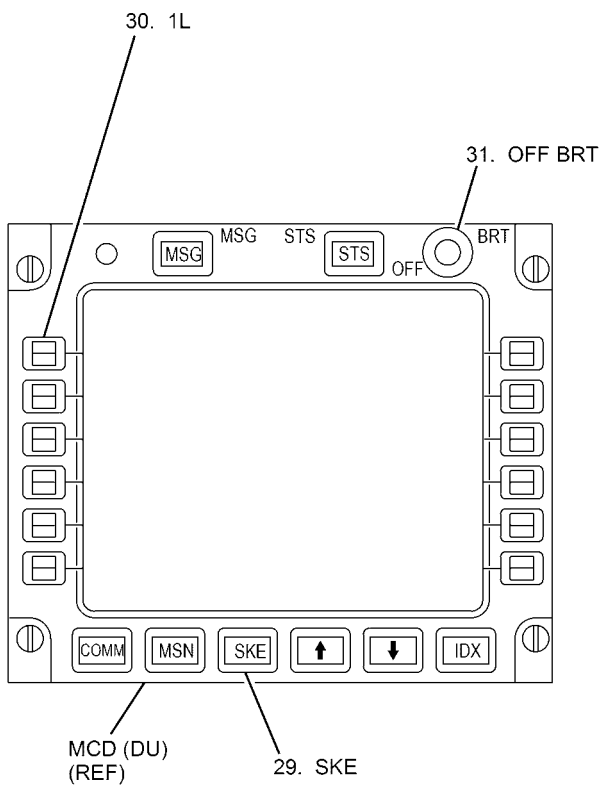
Pressing 1L LS key toggles between **STANDBY**, **NORMAL**, **MAX** and **SILENT** modes.

30. (A) Press 1L LS key until **STANDBY** is displayed.

- **STANDBY** is displayed.

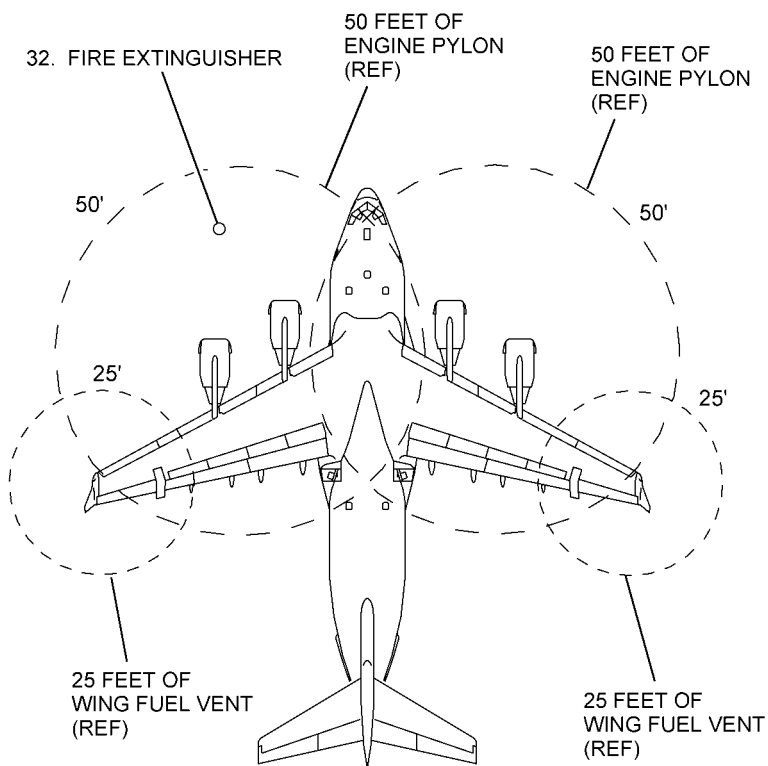
31. (A) Rotate **OFF BRT** knob to **OFF**.

- Characters on **MCD (DU)** go off.



ICN-88277-G1228626-001-01

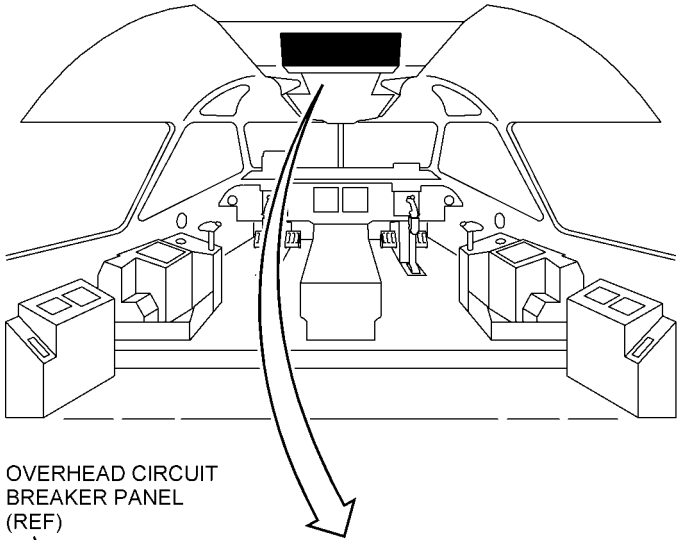
32. (A) Place fire extinguisher within 50 feet of engine pylon being drained, but outside 25 feet wing fuel vent Flight Servicing Safety Zone (FSSZ).



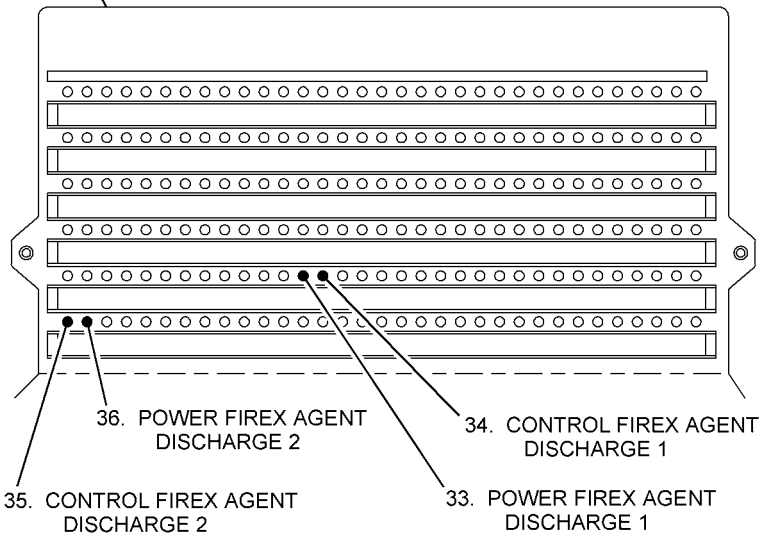
ICN-88277-G1228653-001-01

TO 1300i-2-12JG-28-2

33. (A) Open **POWER FIREX AGENT DISCHARGE 1** circuit breaker on overhead circuit breaker panel, row **E**, column **13**, and attach warning tag.
34. (A) Open **CONTROL FIREX AGENT DISCHARGE 1** circuit breaker on overhead circuit breaker panel, row **E**, column **14**, and attach warning tag.
35. (A) Open **CONTROL FIREX AGENT DISCHARGE 2** circuit breaker on overhead circuit breaker panel, row **F**, column **1**, and attach warning tag.
36. (A) Open **POWER FIREX AGENT DISCHARGE 2** circuit breaker on overhead circuit breaker panel, row **F**, column **2**, and attach warning tag.



OVERHEAD CIRCUIT
BREAKER PANEL
(REF)

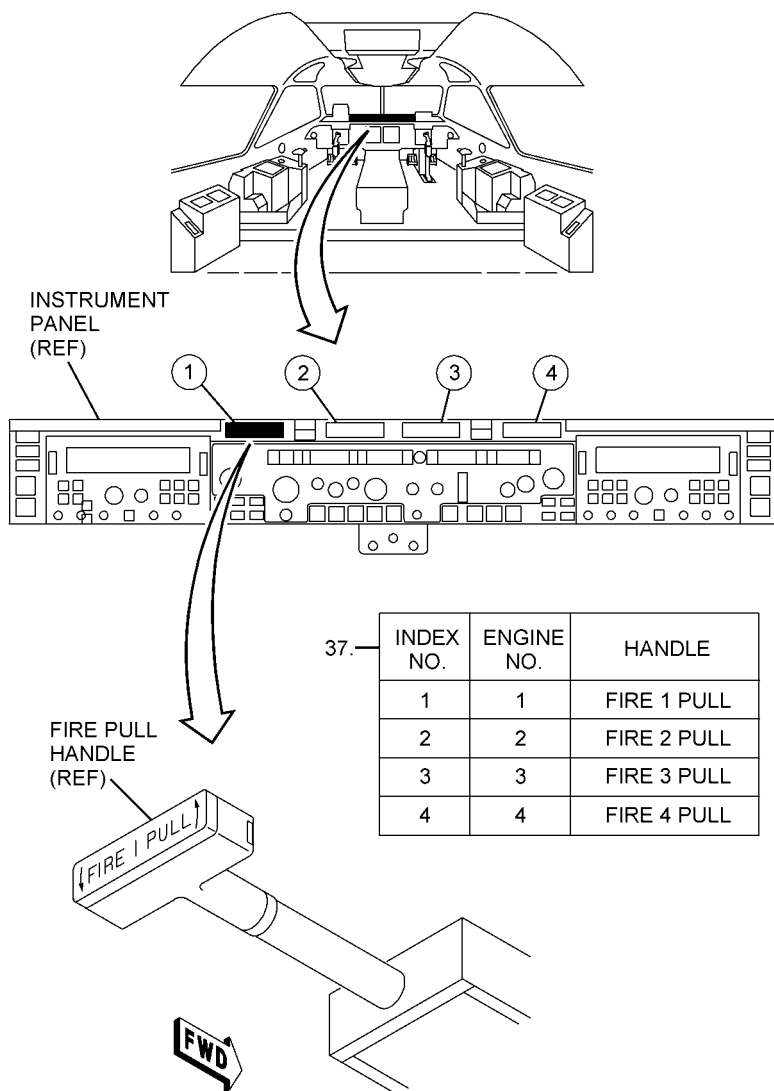


ICN-88277-G1228630-001-01

WARNING

The applicable **ENGINE FUEL & HYD FIRE SHUTOFF VALVE** circuit breaker (F/22, F/23, F/24, and F/25) on overhead circuit breaker panel shall be closed prior to pulling **FIRE PULL** handle. An open circuit breaker will disable the **FIRE PULL** handle. Failure to comply may cause injury to personnel.

37. (A) Pull **FIRE PULL** handle on instrument panel for tank being drained and attach warning tag.

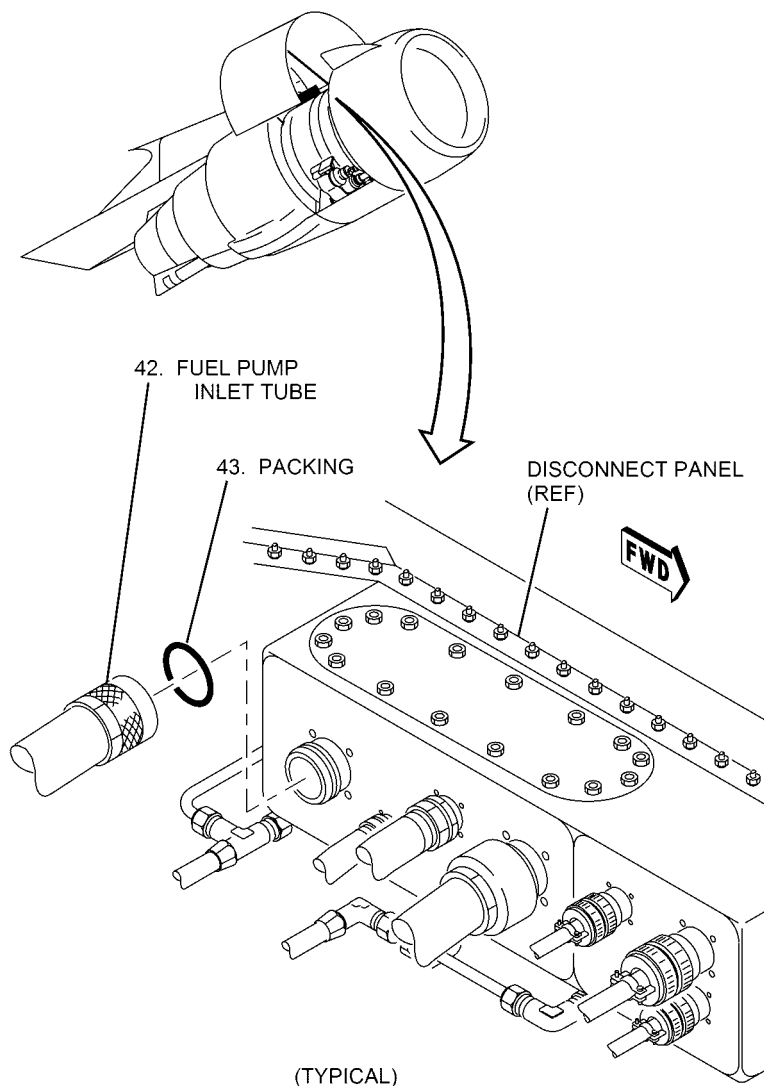


(TYPICAL)

ICN-88277-G1228631-001-01

TO 1300i-2-12JG-28-2

38. Disconnect external electrical power (10-61-01, task 01-2).
39. (A) Ensure **BATT** switch on **ELECTRICAL** panel is **OFF** and attach warning tag.
40. (A) Unlatch and open electrical ground power access door (122ARD); attach warning tag to external electrical power receptacle.
41. Open engine accessory compartment door assembly (54-14-01, task 01-1).
42. (A) Disconnect fuel pump inlet tube from disconnect panel.
43. (A) Remove and discard packing from fuel pump inlet tube.

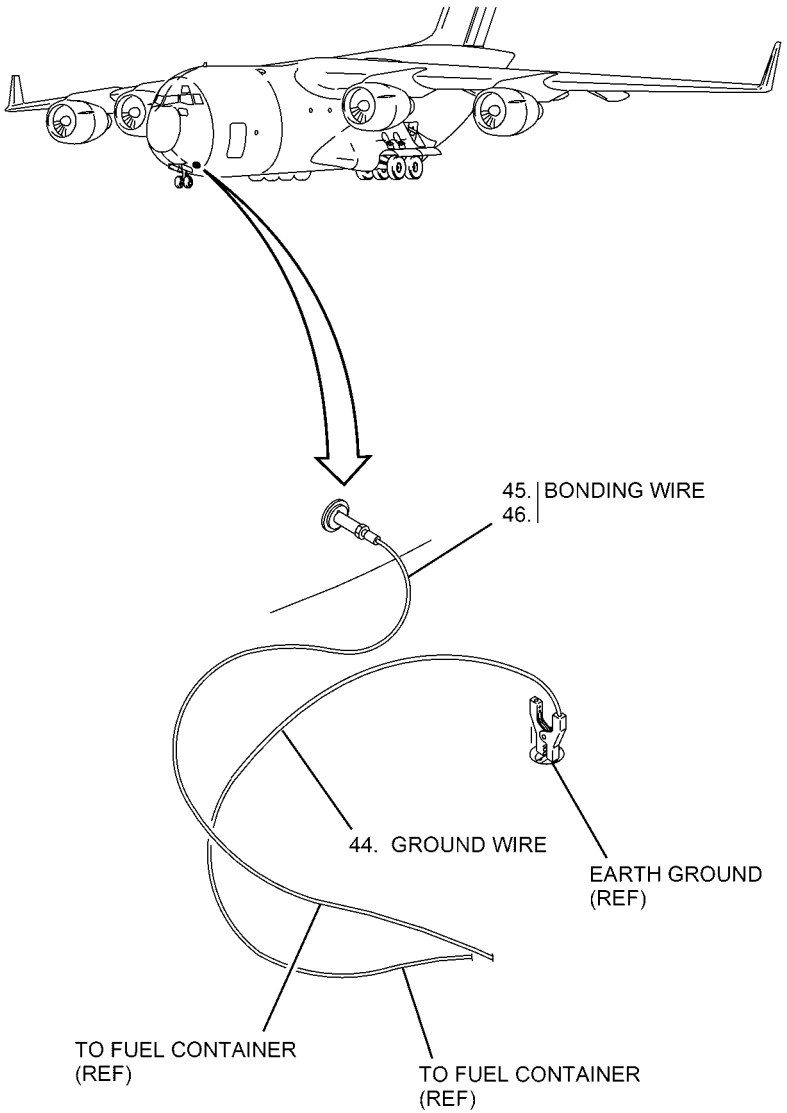


ICN-88277-G1228632-001-01

NOTE

Fuel equipment may be positioned forward or aft of the wing to facilitate a safe defuel operation.

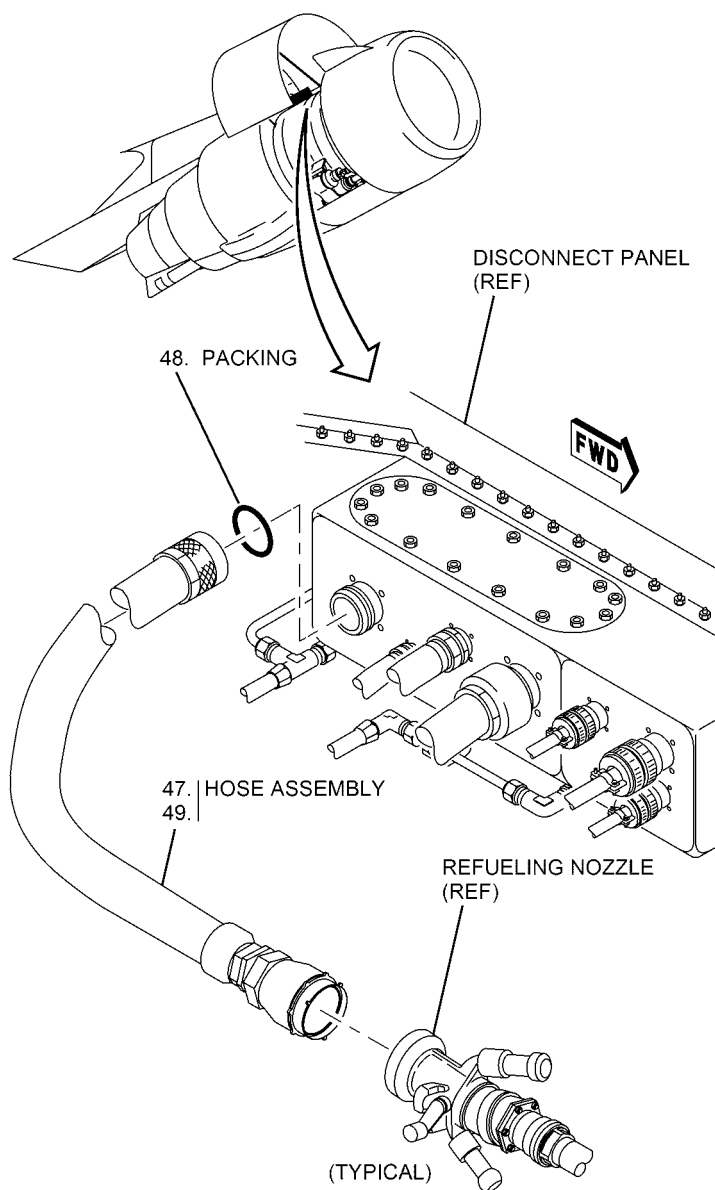
- 44. (C) Position fuel equipment and connect ground wire from fuel equipment to earth ground.
- 45. (C) Connect bonding wire from fuel equipment to aircraft.
- 46. (C) Connect bonding wire from aircraft to maintenance stand.



ICN-88277-G1228633-001-01

TO 1300i-2-12JG-28-2

47. (A) Connect lower end of hose assembly to refueling nozzle of fuel equipment.
48. (A) Apply technical petrolatum to new packing and install packing.
49. (A) Connect upper end of hose assembly to disconnect panel.



ICN-88277-G1228634-001-01

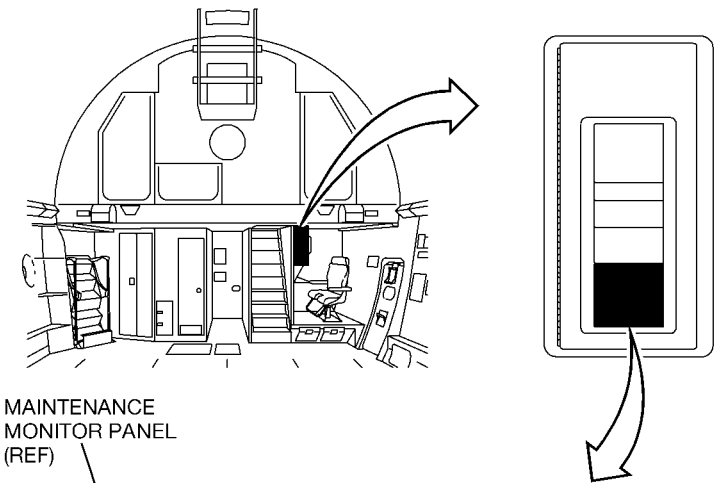
02-3. SINGLE POINT DEFUELING.

NOTE

Ⓐ → Ⓒ BEFORE 1616

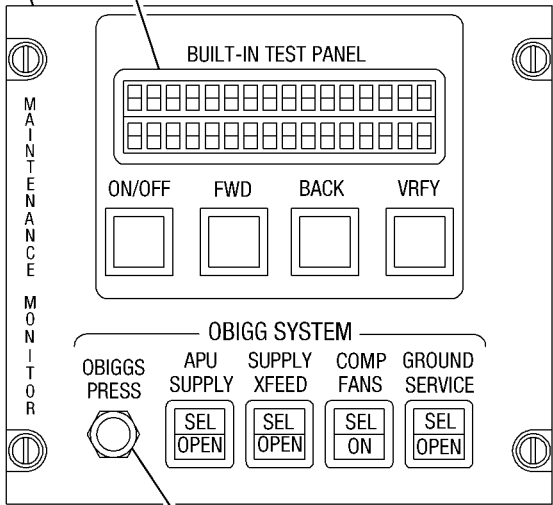
Ⓒ → Ⓕ BEFORE 1617 When OnBoard Inert Gas Generating System (OBIGGS) pressure is needed, OBIGGS shall indicate a minimum pressure of 100 psi to ensure proper inerting.

1. Ⓐ → Ⓒ BEFORE 1616 Ⓒ → Ⓕ BEFORE 1617 (A)
Press **OBIGGS PRESS** switch on **MAINTENANCE MONITOR** panel until reading is displayed.
 - Display reads **L OBIGGS** and **R OBIGGS** system pressure.
1. Ⓐ → Ⓒ AFTER 1616 Ⓒ → Ⓕ AFTER 1617 Ⓖ →
No action required.



MAINTENANCE
MONITOR PANEL
(REF)

DISPLAY
(REF)



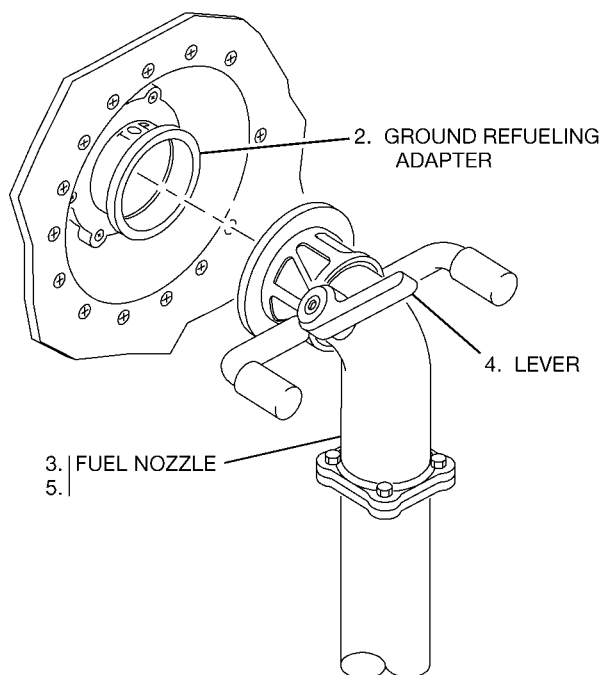
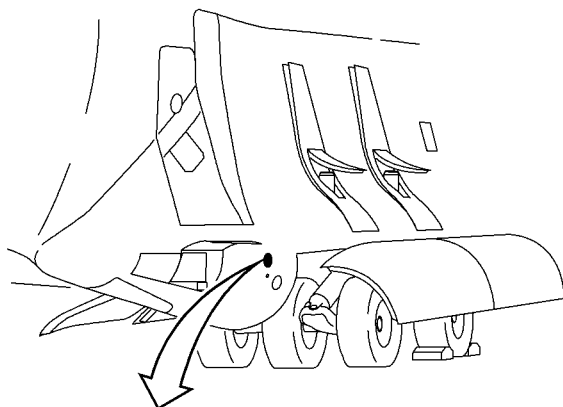
1. OBIGGS PRESS

AA → CX BEFORE 1616 CY → FW BEFORE 1617

ICN-88277-G1228331-004-01

TO 1300i-2-12JG-28-2

2. (B) Visually check ground refueling adapter for cracks, distortion, and excessive wear.
3. (B) Connect fuel nozzle to ground refueling adapter.
4. (B) Move lever to open position.
5. (B) Perform fuel nozzle lock-check by attempting to remove fuel nozzle with lever in open (flow) position.



(TYPICAL)

ICN-88277-G1228031-012-01

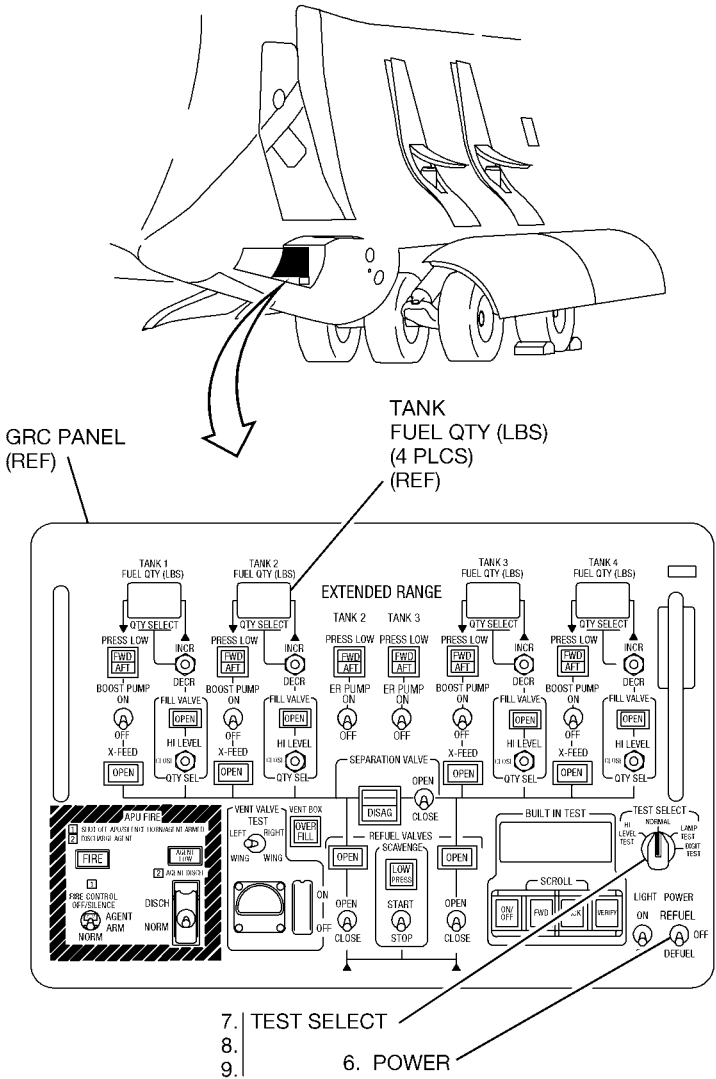
TO 1300i-2-12JG-28-2

6. (B) Set **POWER** switch on Ground Refueling Control (GRC) panel to **DEFUEL**.
 - **X-FEED OPEN** lights come on.
7. (B) Set **TEST SELECT** switch to **LAMP TEST**.
 - All panel lights come on.

NOTE

This sequence will repeat itself until **TEST SELECT** switch is set to **NORMAL**.

8. (B) Set **TEST SELECT** switch to **DIGIT TEST**.
 - All **TANK FUEL QTY (LBS)** indicators comes on for 3 seconds.
 - All **TANK FUEL QTY (LBS)** indicators display all 8's for 3 seconds.
 - All **TANK FUEL QTY (LBS)** indicators go off for 2 seconds.
9. (B) Set **TEST SELECT** switch to **NORMAL**.
 - All **TANK FUEL QTY (LBS)** indicators display fuel quantity in each tank.



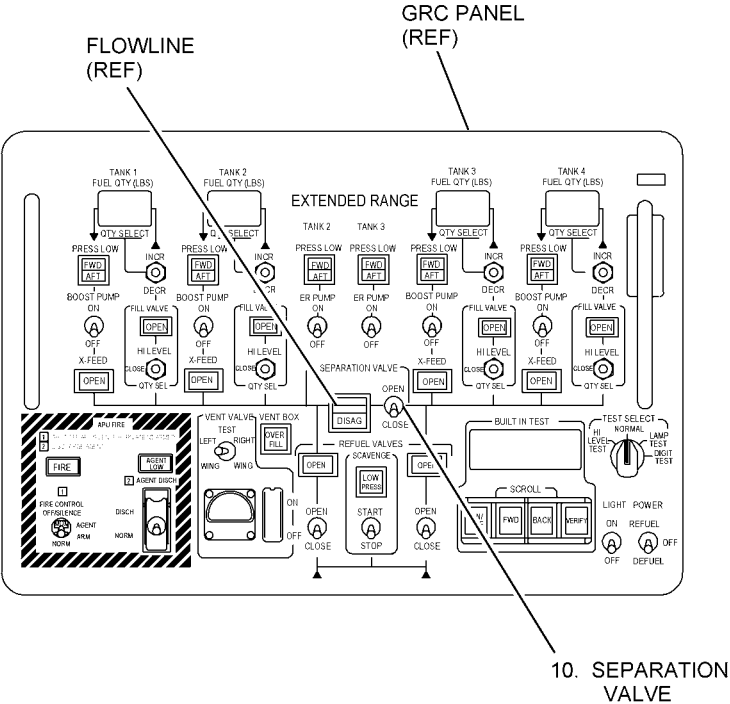
ICN-88277-G1228032-009-01

WARNING

When using both fuel receptacles, **SEPARATION VALVE** switch shall be set in the **CLOSE** position. Failure to close separation valve when using both receptacles may result in fuel spillage causing injury to personnel or damage to aircraft.

10. (B) Set **SEPARATION VALVE** switch as follows:

Quantity of Fuel Receptacles Used	Separation Valve Switch Position	Flowline Status
One	OPEN	On
Two	CLOSE	Off



ICN-88277-G1228033-009-01

NOTE

⬡AA⬡ → ⬡CX⬡ AFTER 1616 ⬡CY⬡ → Extended Range (ER) pumps shall only be operated when total aircraft fuel quantity on gauges read 165,000 lbs or more.

11. (B) Set **PUMP** switches to **ON**.

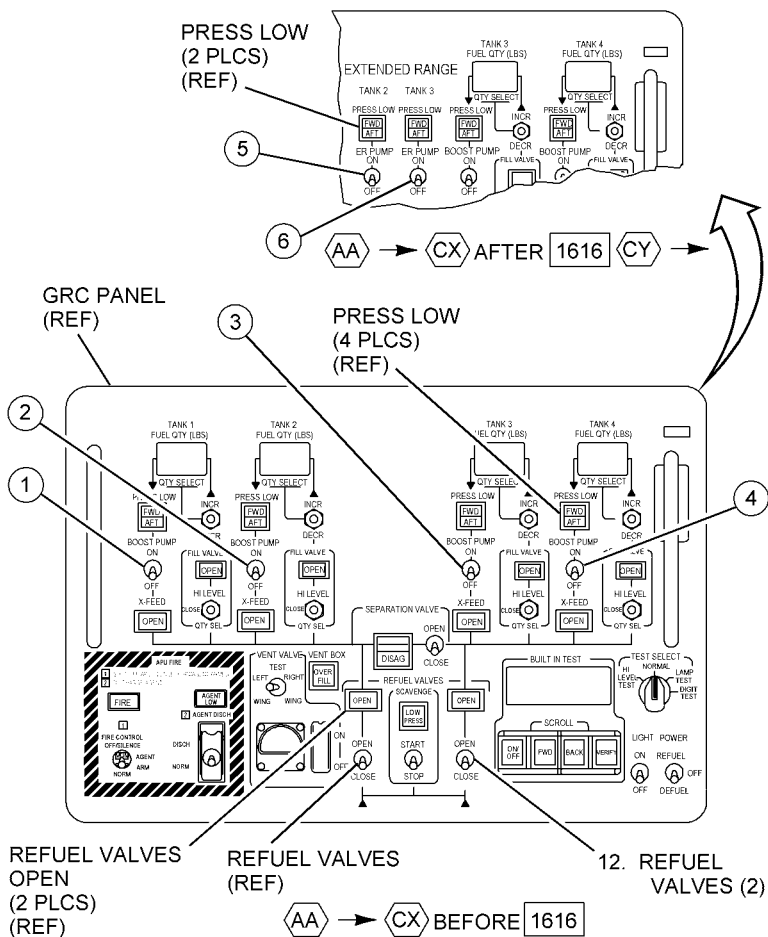
- **PRESS LOW** light comes on momentarily.

NOTE

Left **REFUEL VALVES** switch correlates to the forward fuel receptacle. Right **REFUEL VALVES** switch correlates to the aft fuel receptacle.

12. (B) Set **REFUEL VALVES** switches to **OPEN**.

- **REFUEL VALVES OPEN** light comes on.



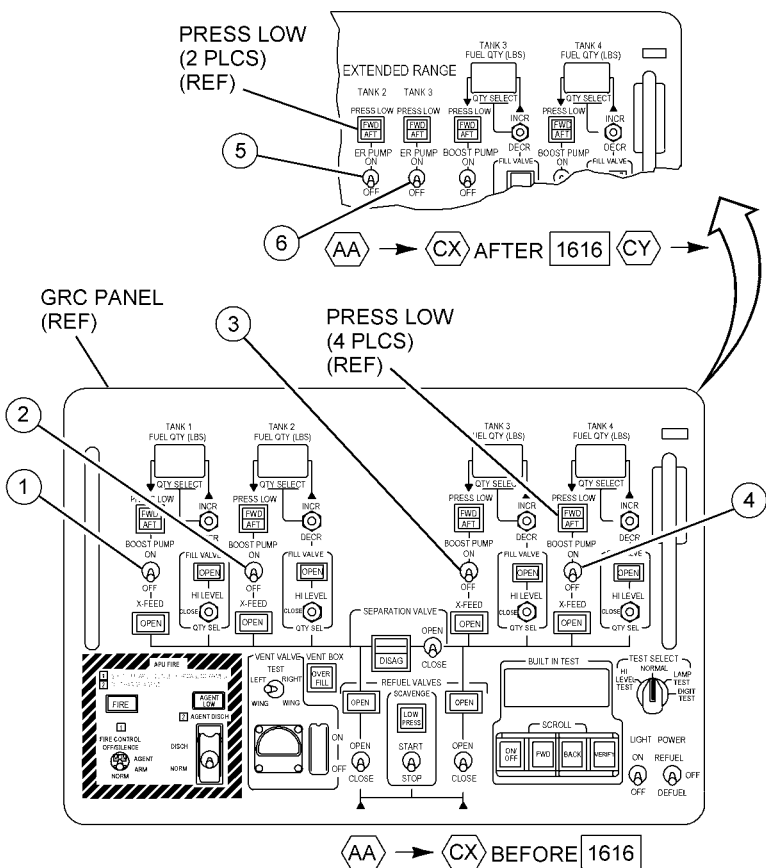
11.	TANK	INDEX NO.	SWITCH	TANK	INDEX NO.	SWITCH
	1	1	BOOST PUMP	ER 2	2	NO. 2 TANK BOOST PUMP
	2	2	BOOST PUMP		5	ER PUMP
	3	3	BOOST PUMP	ER 3	3	NO. 3 TANK BOOST PUMP
	4	4	BOOST PUMP		6	ER PUMP

ICN-88277-G1228034-012-01

WARNING

Do not operate pumps for longer than 5 minutes with **PRESSURE LOW** light on continuously. Failure to comply may cause injury to personnel and damage to aircraft.

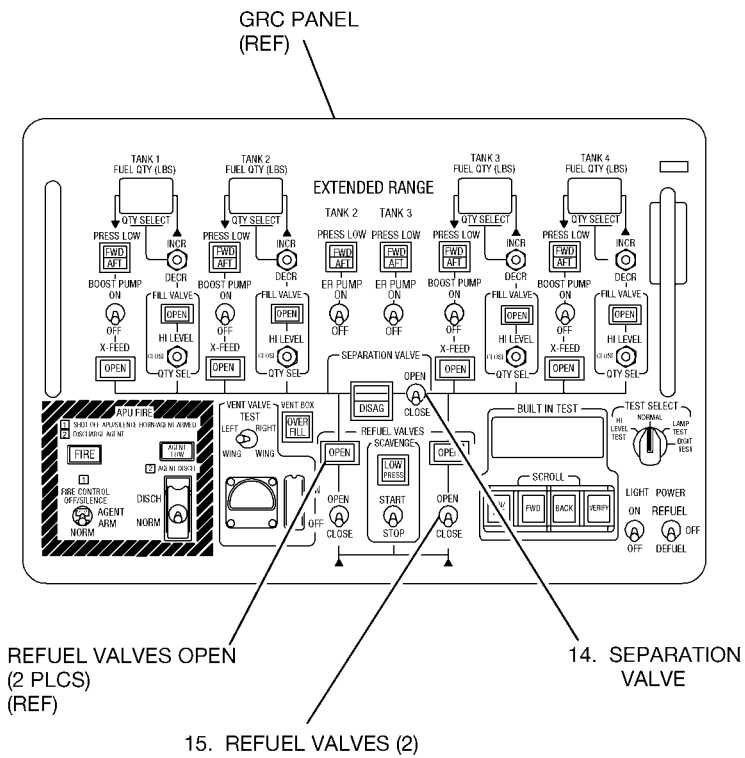
13. (A,B) Monitor defueling operation until required level of fuel is reached, then set **PUMP** switches to **OFF**.
 - **PRESS LOW** light goes off.



13.	TANK	INDEX NO.	SWITCH	TANK	INDEX NO.	SWITCH
	1	1	BOOST PUMP	ER 2	2	NO. 2 TANK BOOST PUMP
	2	2	BOOST PUMP		5	ER PUMP
	3	3	BOOST PUMP	ER 3	3	NO. 3 TANK BOOST PUMP
	4	4	BOOST PUMP		6	ER PUMP

TO 1300i-2-12JG-28-2

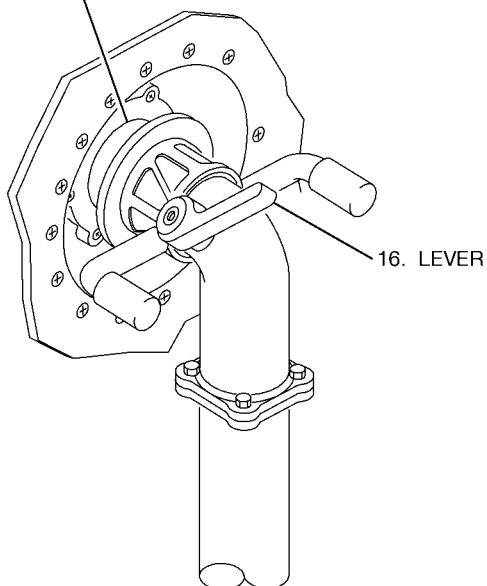
14. (B) Set **SEPARATION VALVE** switch to **CLOSE**.
15. (B) Set **REFUEL VALVES** switches to **CLOSE**.
 - **REFUEL VALVES OPEN** light goes off.
 - Fuel flow stops.



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16. (B) Shutdown fuel servicing equipment and move fuel nozzle lever to closed position.

GROUND REFUELING ADAPTER
(REF)



(TYPICAL)

ICN-88277-G1228035-010-01

17. (B) Set **POWER** switch on GRC panel to **REFUEL**.

- **X-FEED OPEN** lights go off.

NOTE

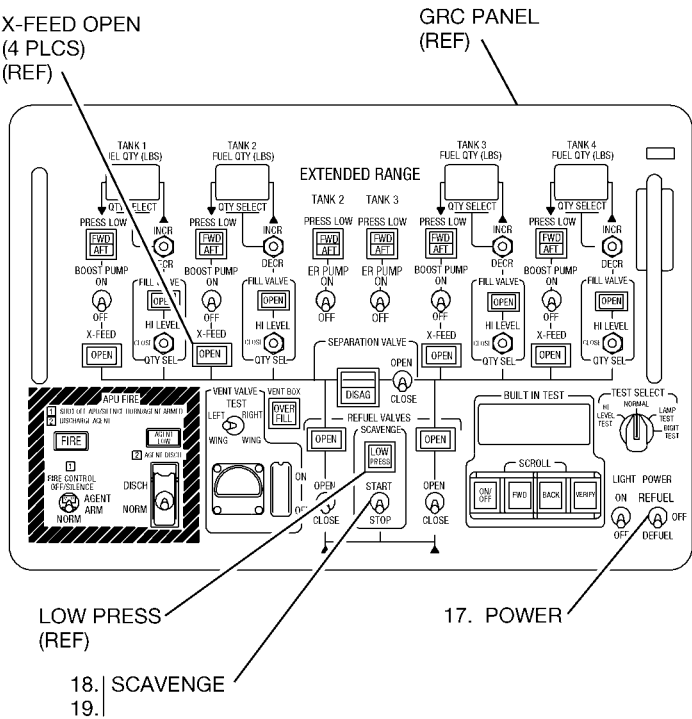
SCAVENGE LOW PRESS light will intermittently come on for approximately 30 seconds and then come on steady when fuel scavenge is complete.

18. (B) Set **SCAVENGE** switch to **START**.

- **LOW PRESS** light comes on steady for one minute.

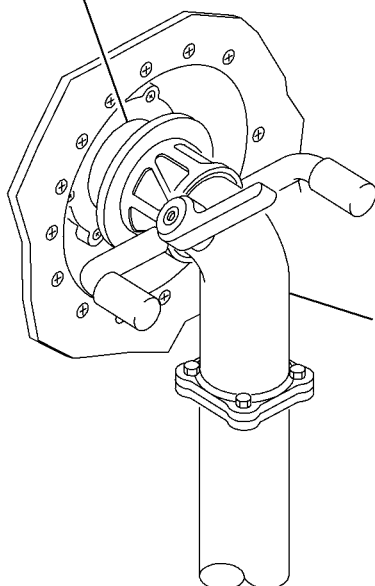
19. (B) Set **SCAVENGE** switch to **STOP**.

- **LOW PRESS** light goes off.



20. (B) Disconnect fuel nozzle from ground refueling adapter.

GROUND REFUELING ADAPTER
(REF)

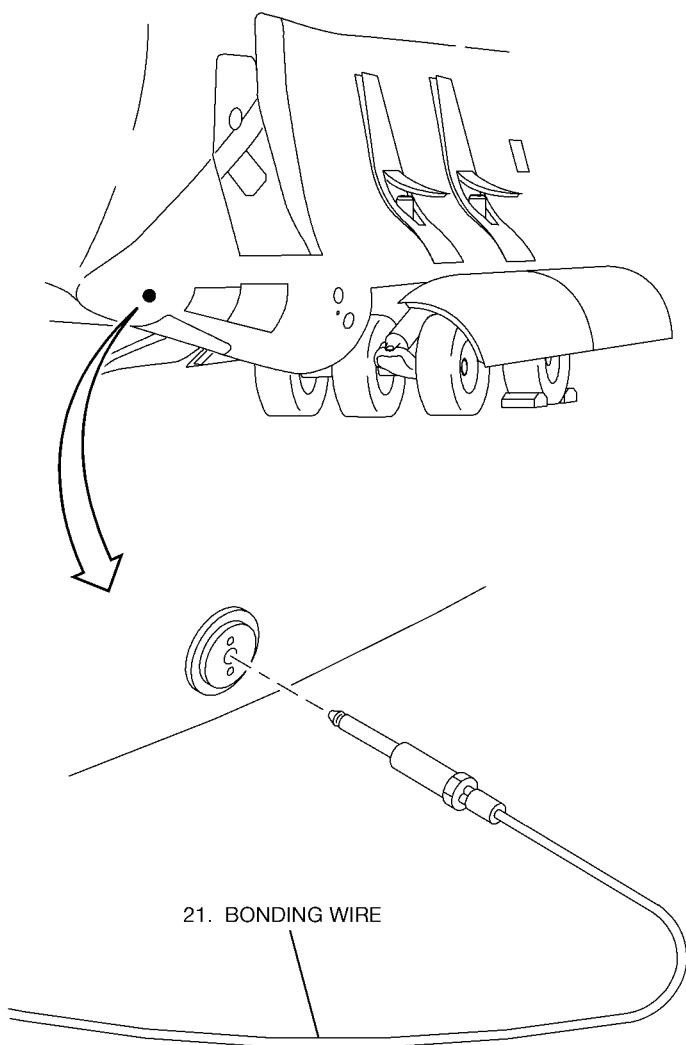


20. FUEL NOZZLE

(TYPICAL)

ICN-88277-G1228467-001-01

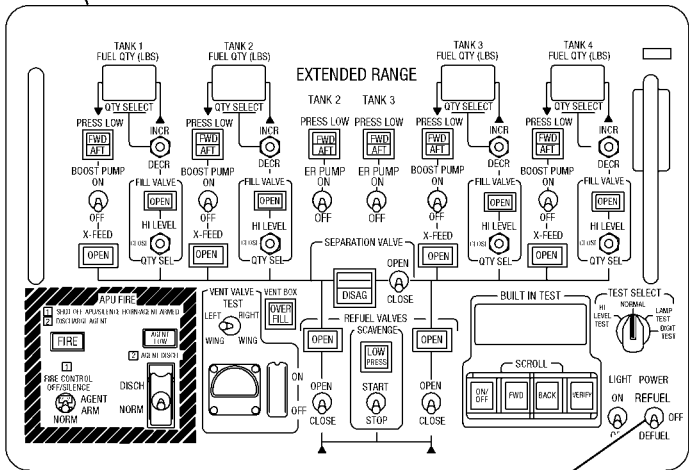
21. (A) Disconnect fuel servicing equipment bonding wire from aircraft and remove fuel servicing equipment.



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22. (B) Set **POWER** switch on GRC panel to **OFF**.

GRC PANEL
(REF)

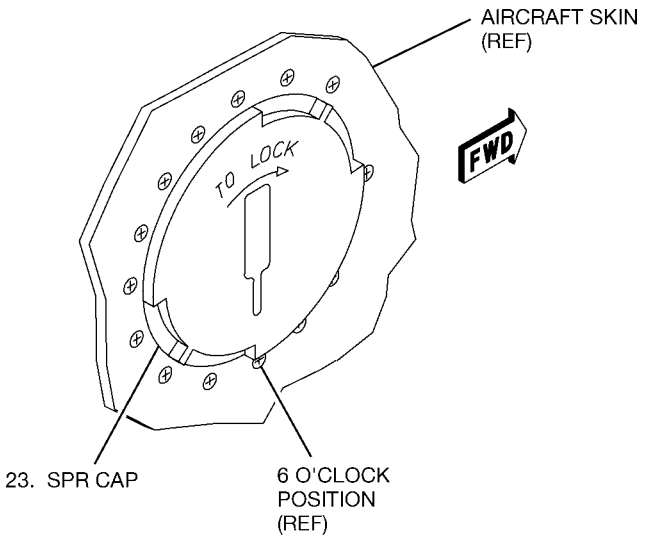


22. POWER

ICN-88277-G1228466-001-01



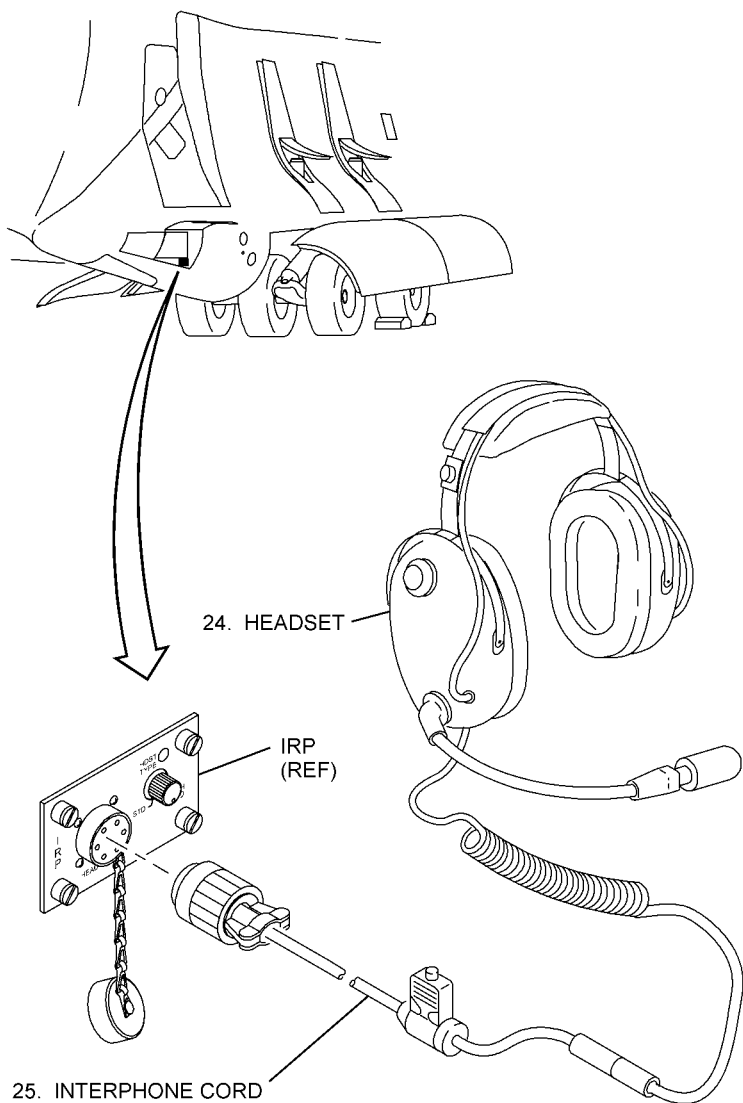
- Ensure lanyard is attached to the aircraft and fuel cap(s). Failure to comply may cause loss of fuel cap(s) and damage the aircraft.
 - Ensure fuel cap(s) are flush with the aircraft when installed. Failure to comply may cause loss of fuel cap(s) and damage to the aircraft.
 - Ensure fuel cap lanyard(s) are not exposed after fuel cap(s) installation. Failure to comply may cause loss of fuel cap(s) and damage to aircraft.
23. (B) Install Single Point Refueling (SPR) cap(s) with thin tab locked at the 6 o'clock position.



ICN-88277-G1228458-001-01

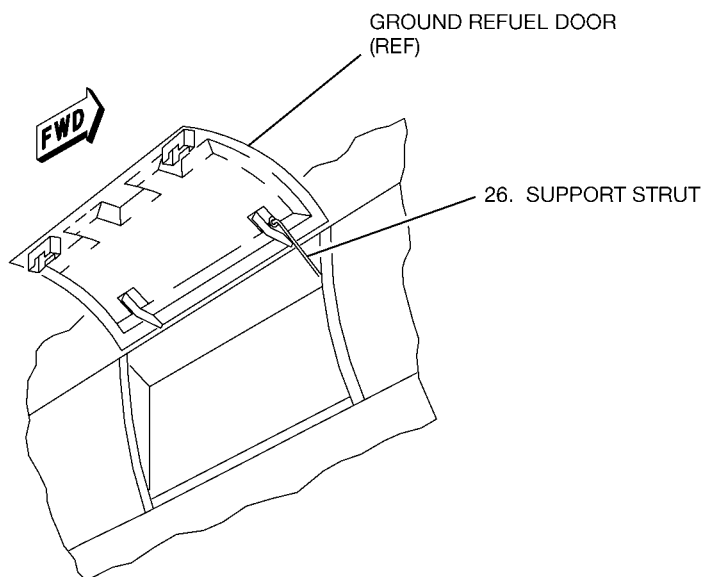
TO 1300i-2-12JG-28-2

24. (B) Disconnect headset from interphone cord.
25. (B) Disconnect interphone cord from IRP.



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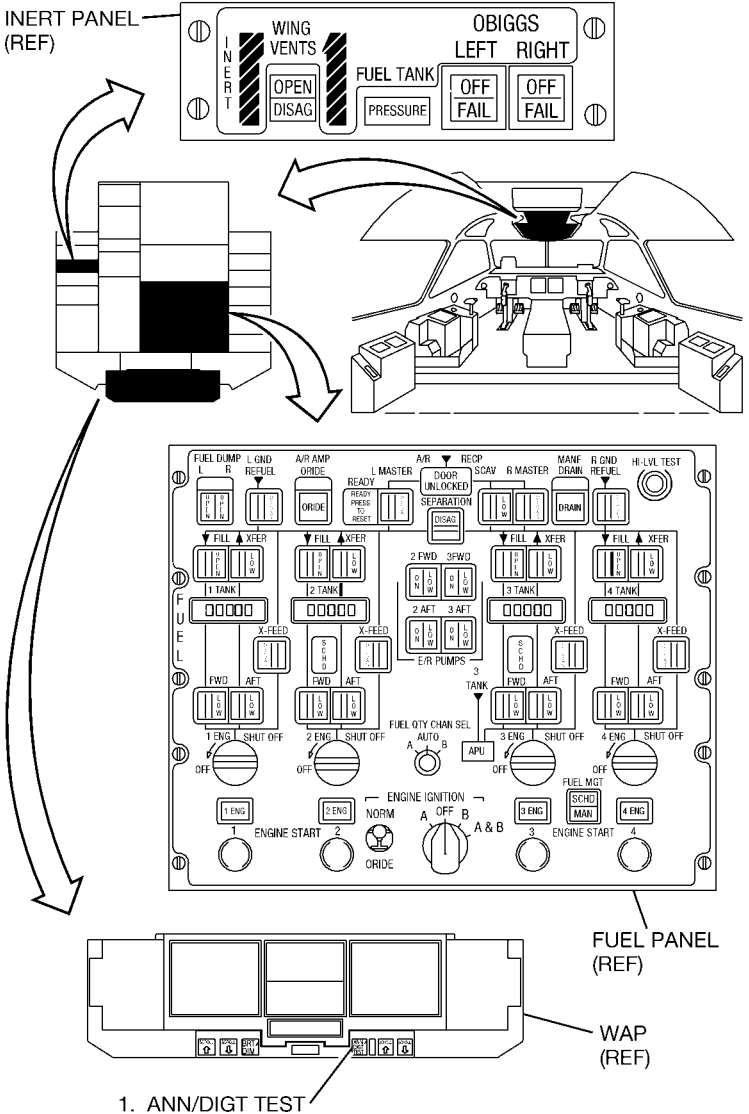
26. (B) Remove support strut; close and latch ground refuel door (183JRD).



ICN-88277-G1228464-001-01

02-4. SINGLE POINT DEFUELING - ALTERNATE METHOD.

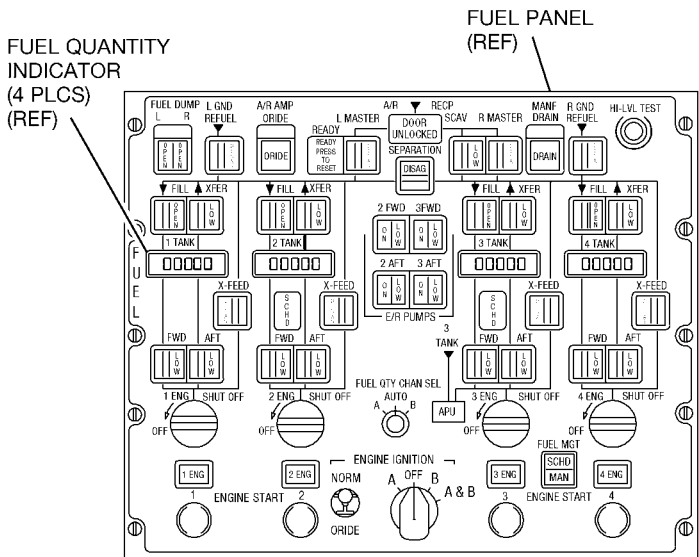
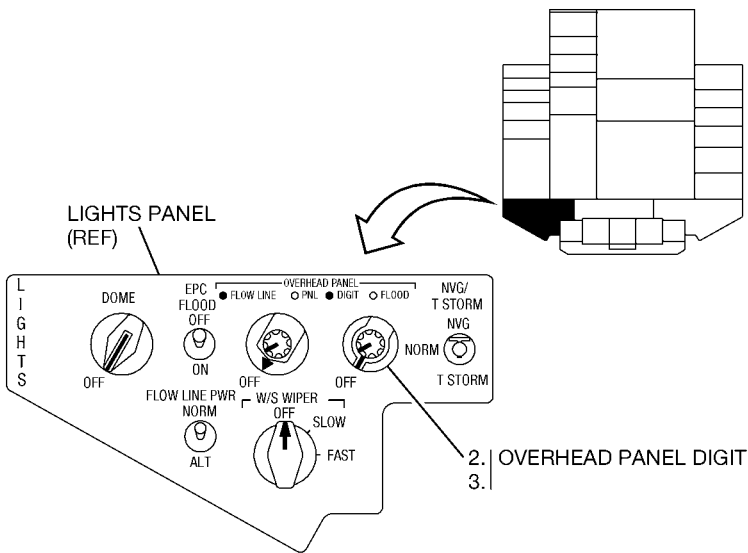
1. (B) Press and hold **ANN/DIGT TEST** switch on Warning and caution Annunciator Panel (WAP).
 - **FUEL** panel lights come on.
 - **INERT** panel lights come on.



ICN-88277-G1228357-004-01

TO 1300i-2-12JG-28-2

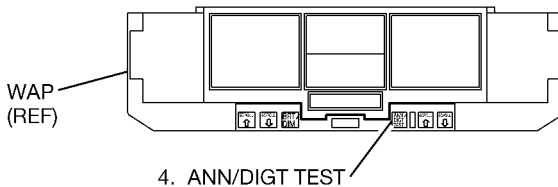
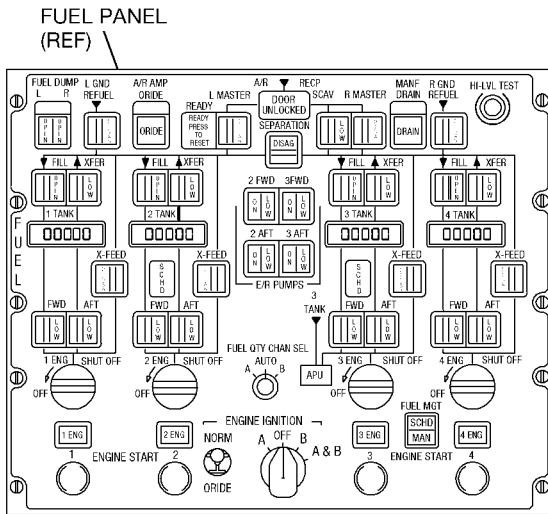
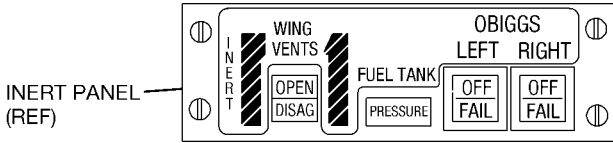
2. (B) Rotate **OVERHEAD PANEL DIGIT** knob on **LIGHTS** panel fully clockwise.
 - All fuel quantity indicators display all 8's.
3. (B) Rotate **OVERHEAD PANEL DIGIT** knob fully counter clockwise.
 - All 8's displayed on fuel quantity indicators go off.



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4. (B) Release **ANN/DIGT TEST** switch on WAP.

- **FUEL** panel lights go off.
- **INERT** panel lights go off.



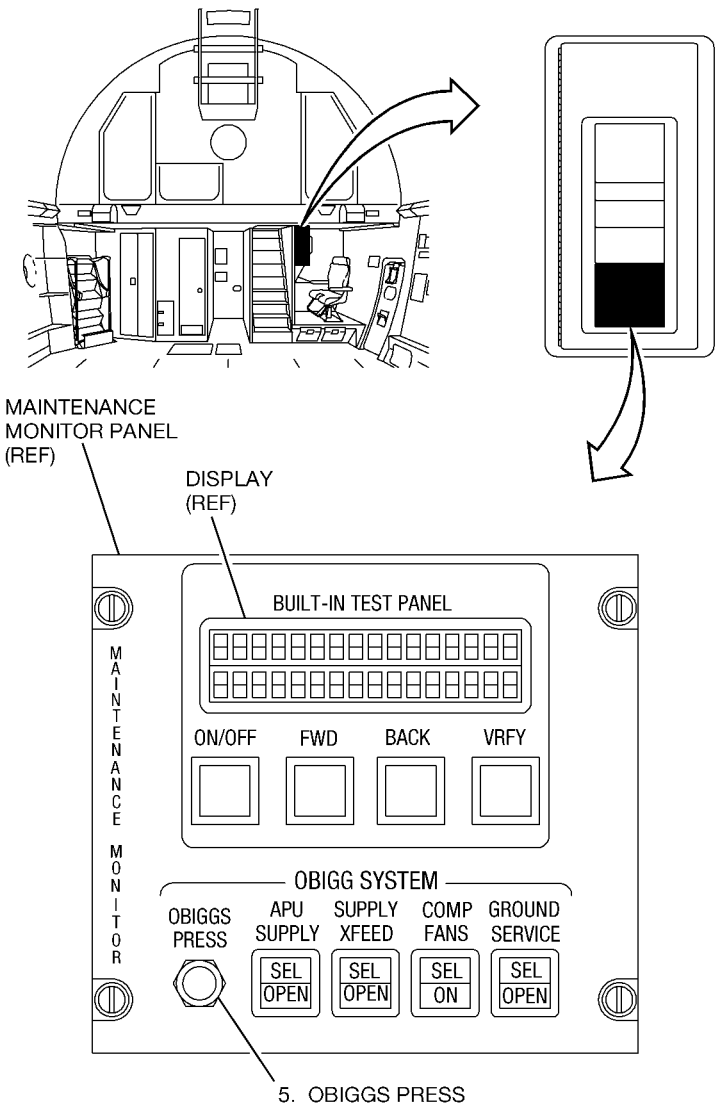
ICN-88277-G1228359-004-01

NOTE

⟨AA⟩ → ⟨CX⟩ BEFORE 1616

⟨CY⟩ → ⟨FW⟩ BEFORE 1617 When OnBoard Inert Gas Generating System (OBIGGS) pressure is needed, OBIGGS shall indicate a minimum pressure of 100 psi to ensure proper inerting.

5. ⟨AA⟩ → ⟨CX⟩ BEFORE 1616 ⟨CY⟩ → ⟨FW⟩ BEFORE 1617 (B)
Press **OBIGGS PRESS** switch on **MAINTENANCE MONITOR** panel until reading is displayed.
 - Display reads **L OBIGGS** and **R OBIGGS** system pressure.
5. ⟨AA⟩ → ⟨CX⟩ AFTER 1616 ⟨CY⟩ → ⟨FW⟩ AFTER 1617 ⟨FX⟩ →
No action required.

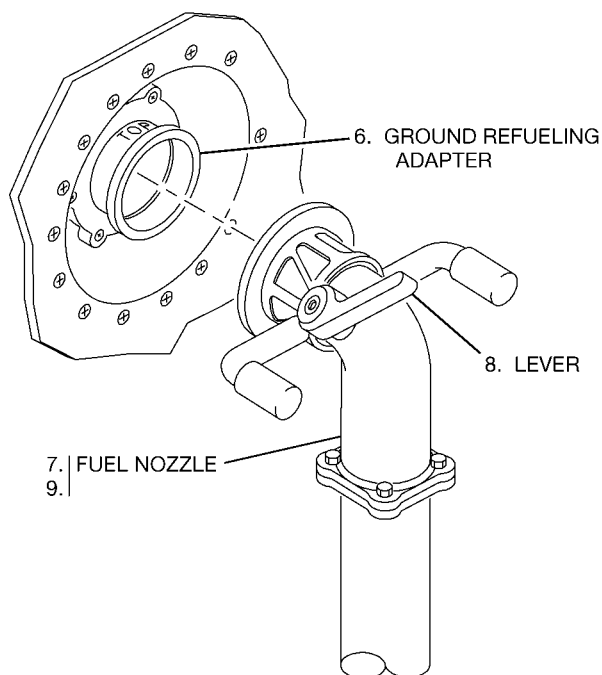
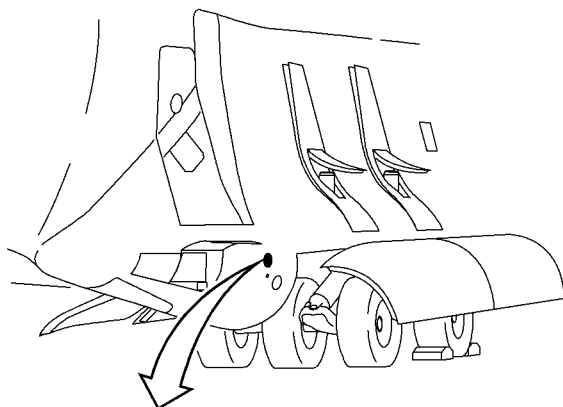


AA → CX BEFORE 1616 CY → FW BEFORE 1617

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TO 1300i-2-12JG-28-2

6. (A) Visually check ground refueling adapter for cracks, distortion, and excessive wear.
7. (A) Connect fuel nozzle to ground refueling adapter.
8. (A) Move lever to open position.
9. (A) Perform fuel nozzle lock-check by attempting to remove fuel nozzle with lever in open (flow) position.



(TYPICAL)

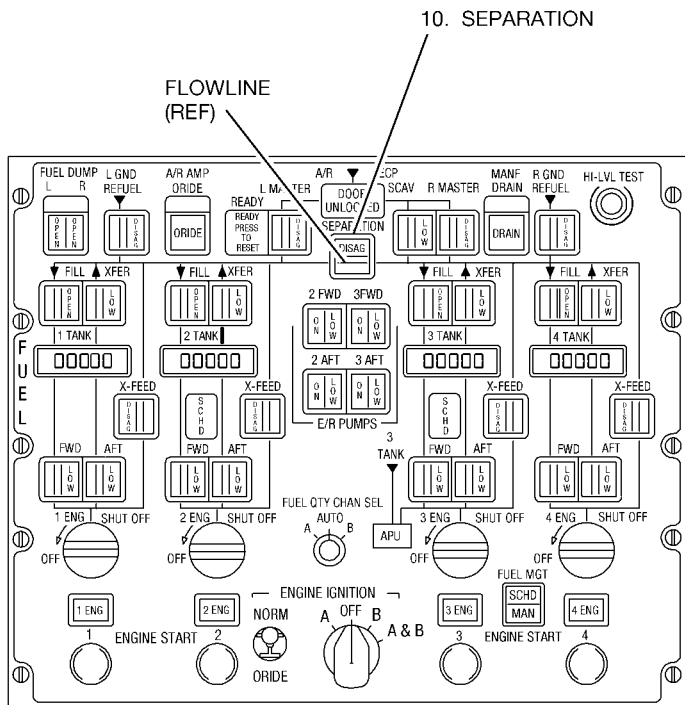
ICN-88277-G1228355-006-01

WARNING

When using both fuel receptacles, **SEPARATION VALVE** switchlight shall be set in the **CLOSE** position. Failure to close separation valve when using both receptacles may result in fuel spillage causing injury to personnel or damage to aircraft.

10. (B) Set **SEPARATION VALVE** switchlight on **FUEL PANEL** as required.

Quantity of Fuel Receptacles Used	Separation Valve Switchlight Position	Flowline Status
One	OPEN (Depressed)	On
Two	CLOSE (Raised)	Off



FUEL PANEL
(REF)

11. (B) Press **X-FEED** switchlight for tank to be defueled.

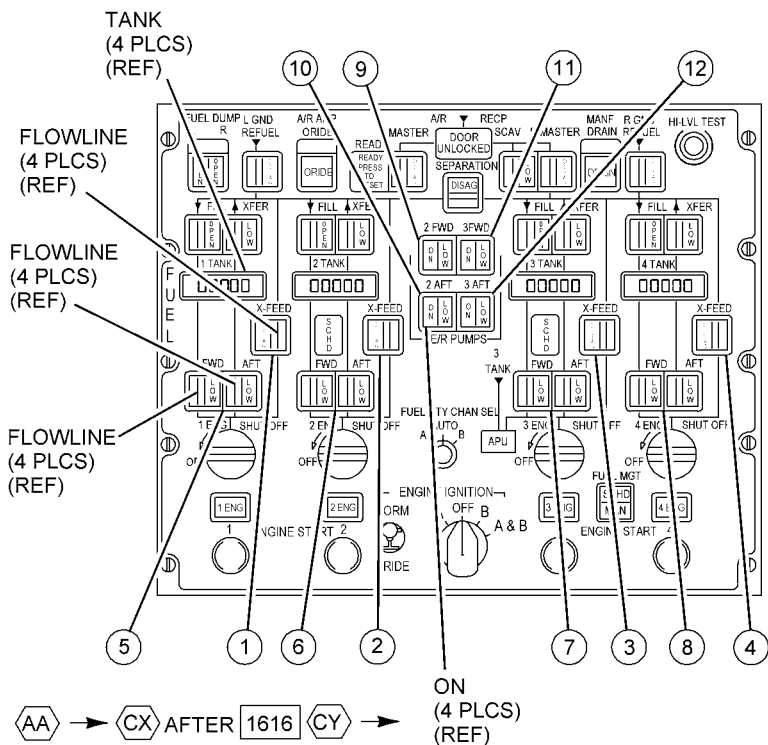
- Flowline comes on.

NOTE

⟨AA⟩ → ⟨CX⟩ AFTER 1616 ⟨CY⟩ → Extended Range (ER) pumps shall only be operated when total aircraft fuel quantity reads 165,000 lbs or more on gauges.

12. (B) Press **FWD** and **AFT** switchlights for tank to be defueled.

- **TANK** flowline comes on.
- ⟨AA⟩ → ⟨CX⟩ AFTER 1616 ⟨CY⟩ → **FWD** and **AFT** extended range pump **ON** light comes **ON**.



11.

INDEX NO.	TANK	SWITCHLIGHT
1	1	X-FEED
2	2	X-FEED
3	3	X-FEED
4	4	X-FEED

12.

INDEX NO.	TANK	SWITCHLIGHT
5	1	FWD AFT
6	2	FWD AFT
7	3	FWD AFT
8	4	FWD AFT
9	2 ER	FWD
10	2 ER	AFT
11	3 ER	FWD
12	3 ER	AFT

(TYPICAL)

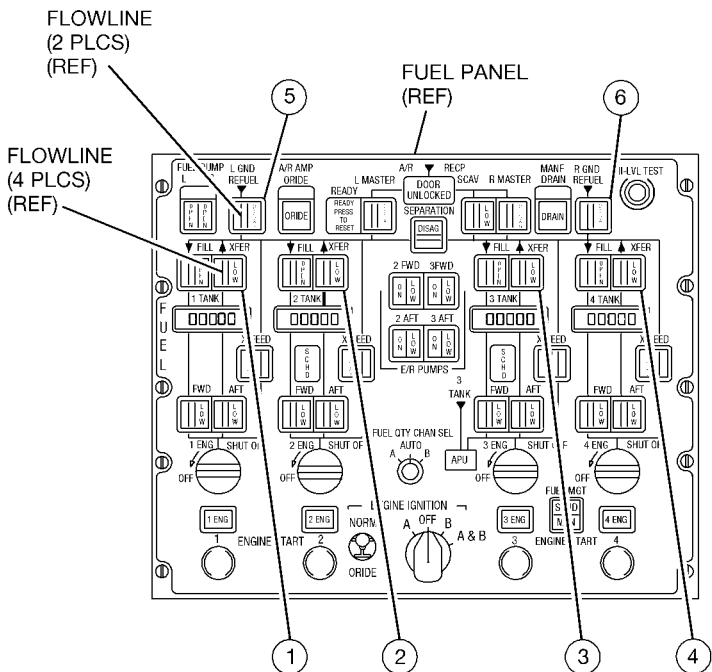
ICN-88277-G1228345-006-01

13. (B) Press **XFER** switchlight for tank to be defueled.
 - Flowline comes on.

NOTE

L GND REFUEL switchlight correlates to the forward fuel receptacle. **R GND REFUEL** switchlight correlates to the aft fuel receptacle.

14. (B) Press **GND REFUEL** switchlight.
 - Flowline comes on.



14.

INDEX NO.	SWITCHLIGHT
5	L GND REFUEL
6	R GND REFUEL

13.

INDEX NO.	TANK	SWITCHLIGHT
1	1	XFER
2	2	XFER
3	3	XFER
4	4	XFER

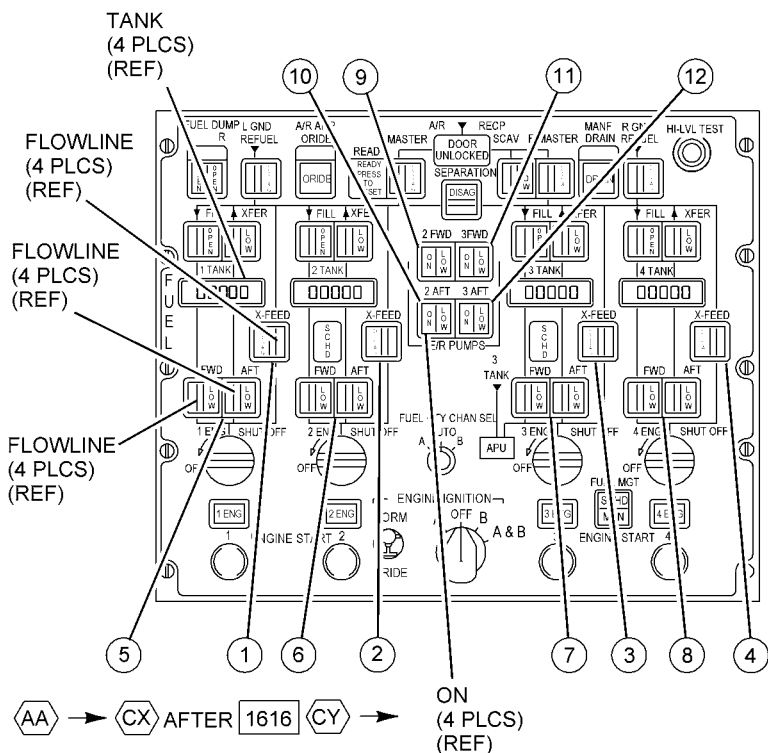
(TYPICAL)

ICN-88277-G1228360-003-01

WARNING

Do not operate pumps for longer than 5 minutes with **PRESS LOW** light on continuously. Failure to comply may cause injury to personnel and damage to aircraft.

15. (B) Monitor defueling operation until required quantity of fuel is reached, then press **FWD** and **AFT** switchlights.
 - Flowline goes off.
 - **AA** → **CX** AFTER **1616** **CY** → **FWD** and **AFT** extended range pump **ON** light goes off.
16. (B) Press **X-FEED** switchlight for tank being defueled.
 - Flowline goes off.



16.	INDEX NO.	TANK	SWITCHLIGHT
	1	1	X-FEED
	2	2	X-FEED
	3	3	X-FEED
	4	4	X-FEED

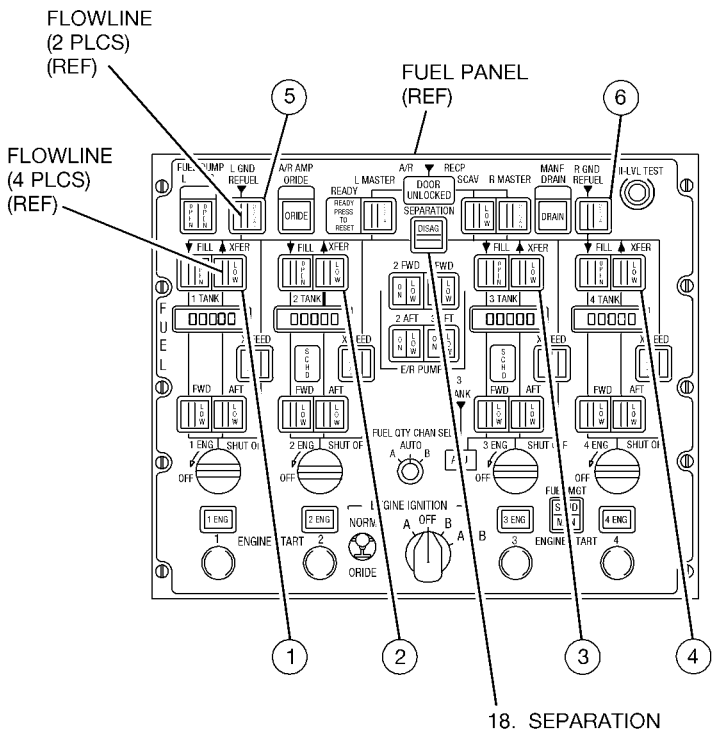
15.	INDEX NO.	TANK	SWITCHLIGHT
	5	1	FWD AFT
	6	2	FWD AFT
	7	3	FWD AFT
	8	4	FWD AFT
	9	2 ER	FWD
	10	2 ER	AFT
	11	3 ER	FWD
	12	3 ER	AFT

(TYPICAL)

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TO 1300i-2-12JG-28-2

17. (B) Press **XFER** switchlight for tank being defueled.
 - Flowline goes off.
18. (B) Press **SEPARATION** switchlight as required.
 - Flowline goes off.
19. (B) Press **GND REFUEL** switchlight.
 - Flowline goes off.

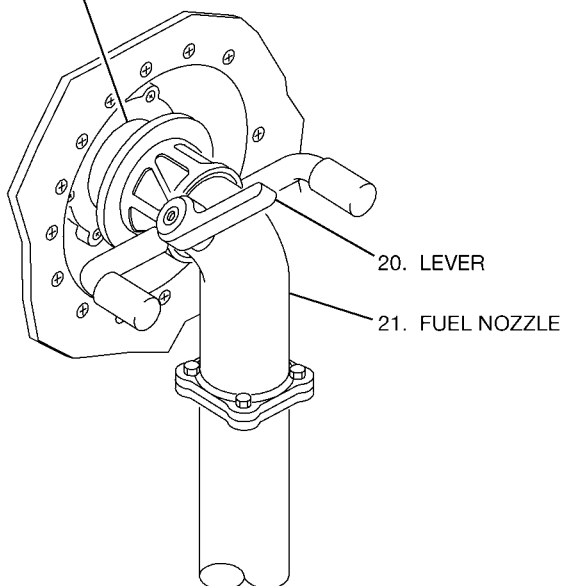


(TYPICAL)

TO 1300i-2-12JG-28-2

20. (A) Shutdown fuel servicing equipment and move fuel nozzle lever to closed position.
21. (A) Disconnect fuel nozzle from ground refueling adapter.

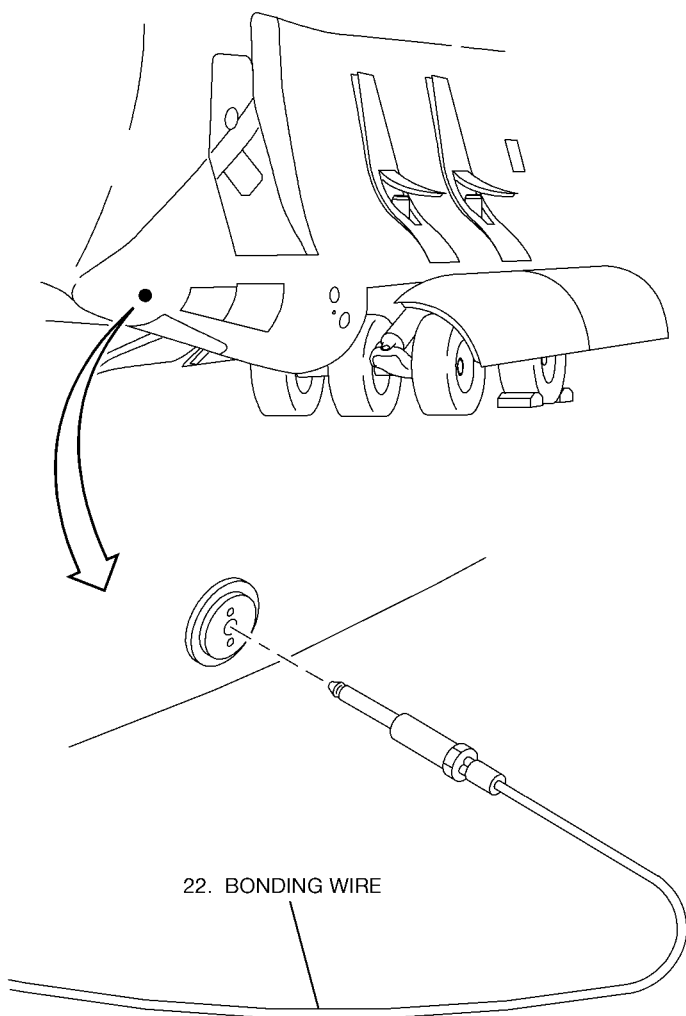
GROUND REFUELING ADAPTER
(REF)



(TYPICAL)

ICN-88277-G1228363-007-01

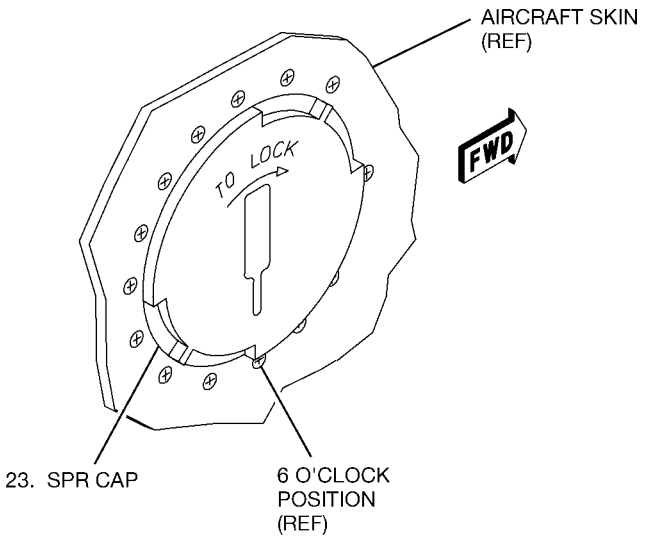
22. (A) Disconnect fuel servicing equipment bonding wire from aircraft and remove fuel servicing equipment.



ICN-88277-G1228364-004-01



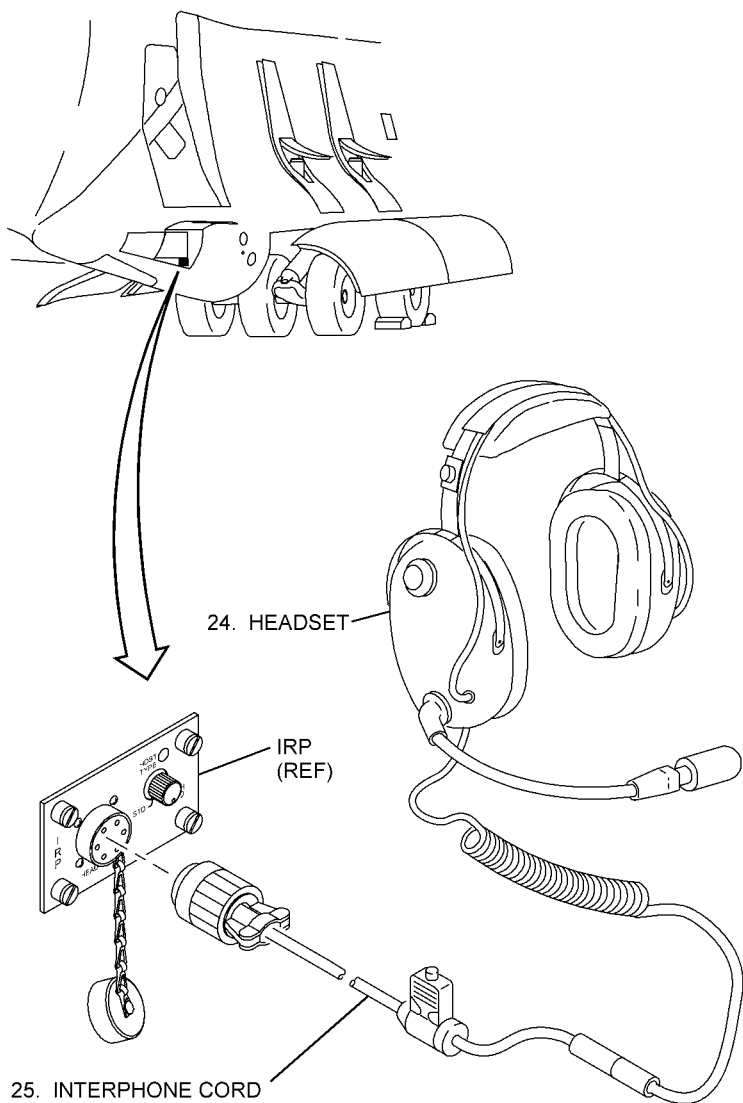
- Ensure lanyard is attached to the aircraft and fuel cap(s). Failure to comply may cause loss of fuel cap(s) and damage the aircraft.
 - Ensure fuel cap(s) are flush with the aircraft when installed. Failure to comply may cause loss of fuel cap(s) and damage to the aircraft.
 - Ensure fuel cap lanyard(s) are not exposed after fuel cap(s) installation. Failure to comply may cause loss of fuel cap(s) and damage to aircraft.
23. (B) Install Single Point Refueling (SPR) cap(s) with thin tab locked at the 6 o'clock position.



ICN-88277-G1228470-001-01

TO 1300i-2-12JG-28-2

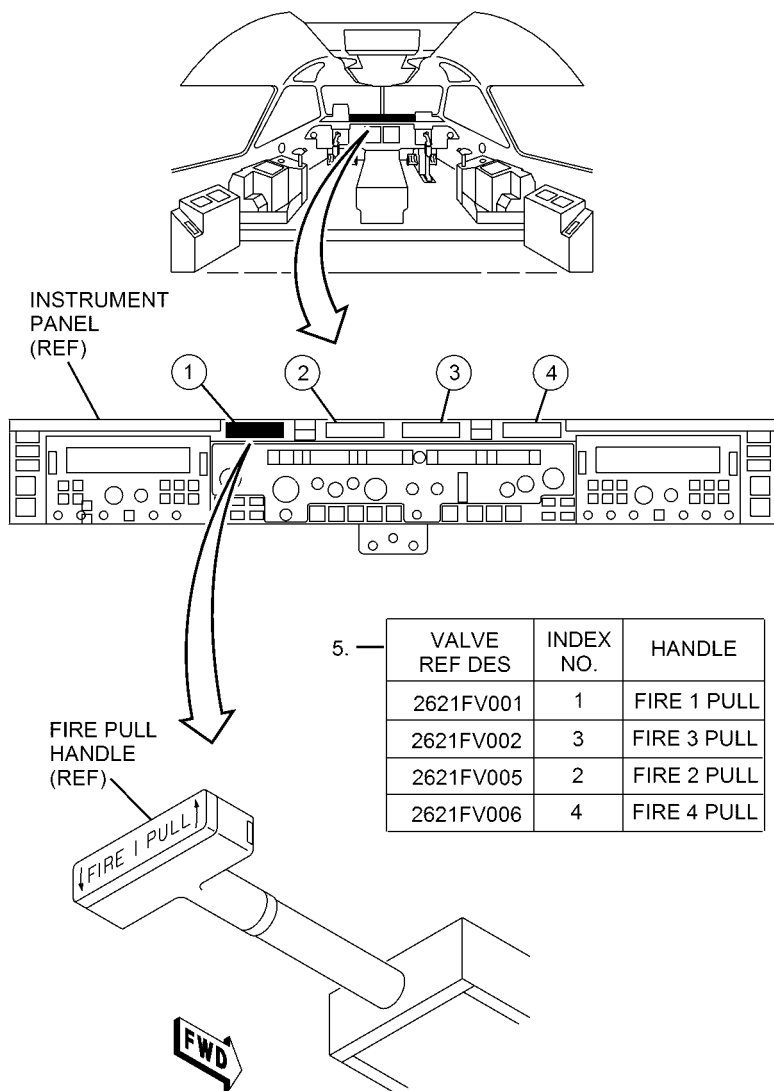
24. (B) Disconnect headset from interphone cord.
25. (B) Disconnect interphone cord from IRP.



ICN-88277-G1228469-004-01

02-4A. DEFUEL THROUGH THE ENGINE PYLON DISCONNECT PANEL.

1. (A) Remove warning tag from external electrical power receptacle in access door (122ARD).
2. (A) Remove warning tag from electrical panel and ensure **BATT** switch is **ON**.
3. Connect external electrical power (10-61-01, task 01-1).
4. Establish aircraft interphone in flight deck.
5. (A) Open engine fire shut off gate valve (ref des 2621FV001, 2621FV002, 2621FV005, or 2621FV006) on the fuel tank being drained by pushing in corresponding **FIRE PULL** handle.

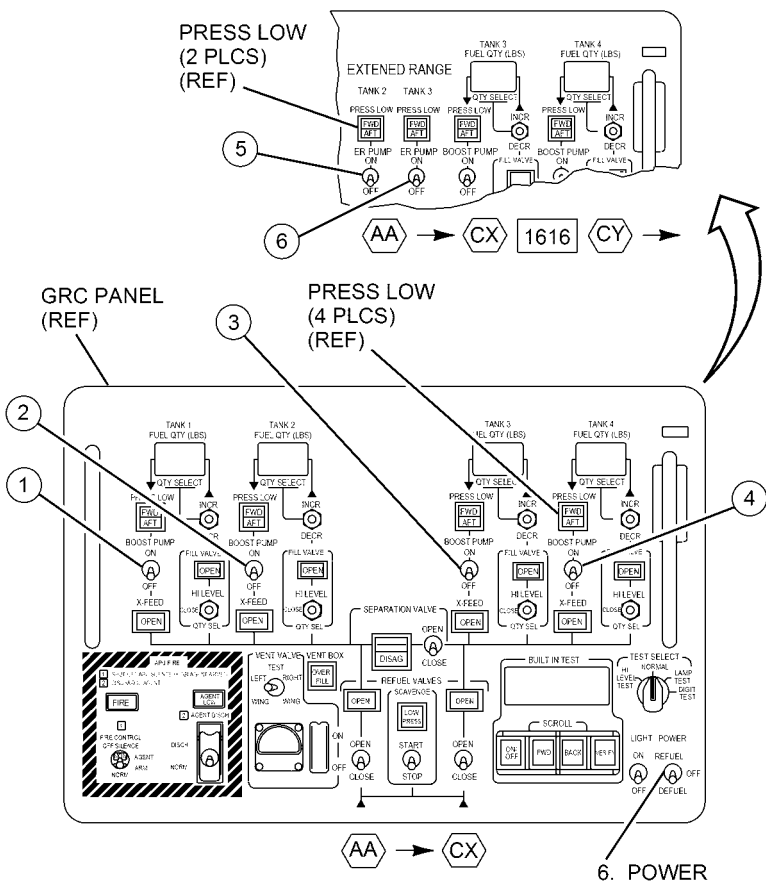


(TYPICAL)

ICN-88277-G1228651-001-01

TO 1300i-2-12JG-28-2

6. (B) Set **POWER** switch on GRC panel to **DEFUEL**.
7. (B) Set BOOST PUMP switches **FWD** and **AFT** to **ON** for tank to be defueled.
 - **PRESS LOW** light comes on momentarily.

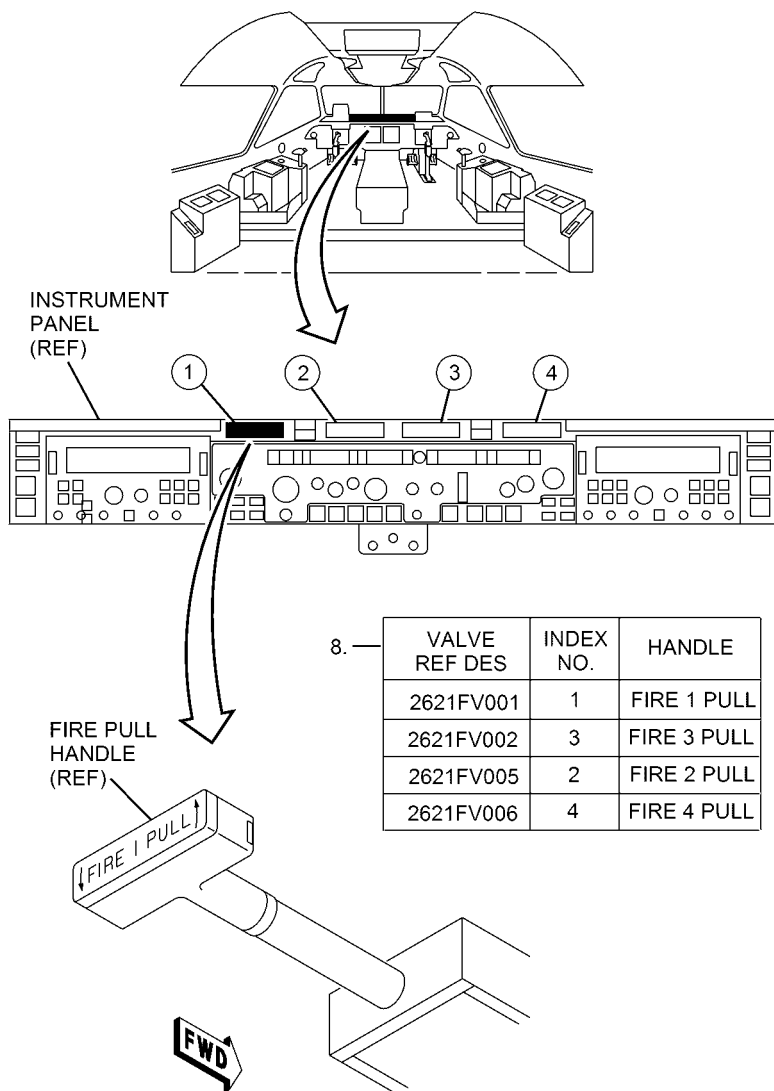


7.	TANK	INDEX NO.	SWITCH	TANK	INDEX NO.	SWITCH
	1	1	BOOST PUMP	ER 2	2	NO. 2 TANK BOOST PUMP
	2	2	BOOST PUMP		5	ER PUMP
	3	3	BOOST PUMP	ER 3	3	NO. 3 TANK BOOST PUMP
	4	4	BOOST PUMP		6	ER PUMP

ICN-88277-G1228637-001-01

TO 1300i-2-12JG-28-2

8. (A) Close engine fire shut off gate valve (ref des 2621FV001, 2621FV002, 2621FV005, or 2621FV006) by pulling corresponding fire pull handle.
9. Disconnect external electrical power (10-61-01, task 01-1).
10. (A) Ensure **BATT** switch on **ELECTRICAL** panel is **OFF** and attach warning tag.

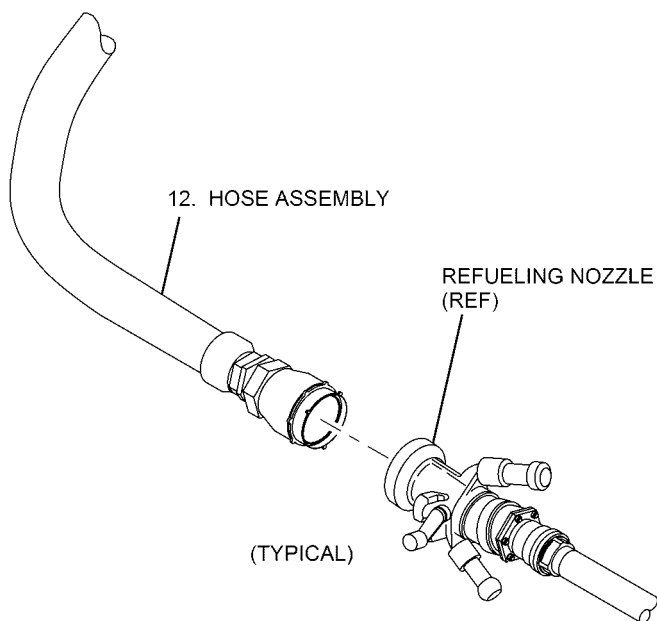


(TYPICAL)

ICN-88277-G1228652-001-01

TO 1300i-2-12JG-28-2

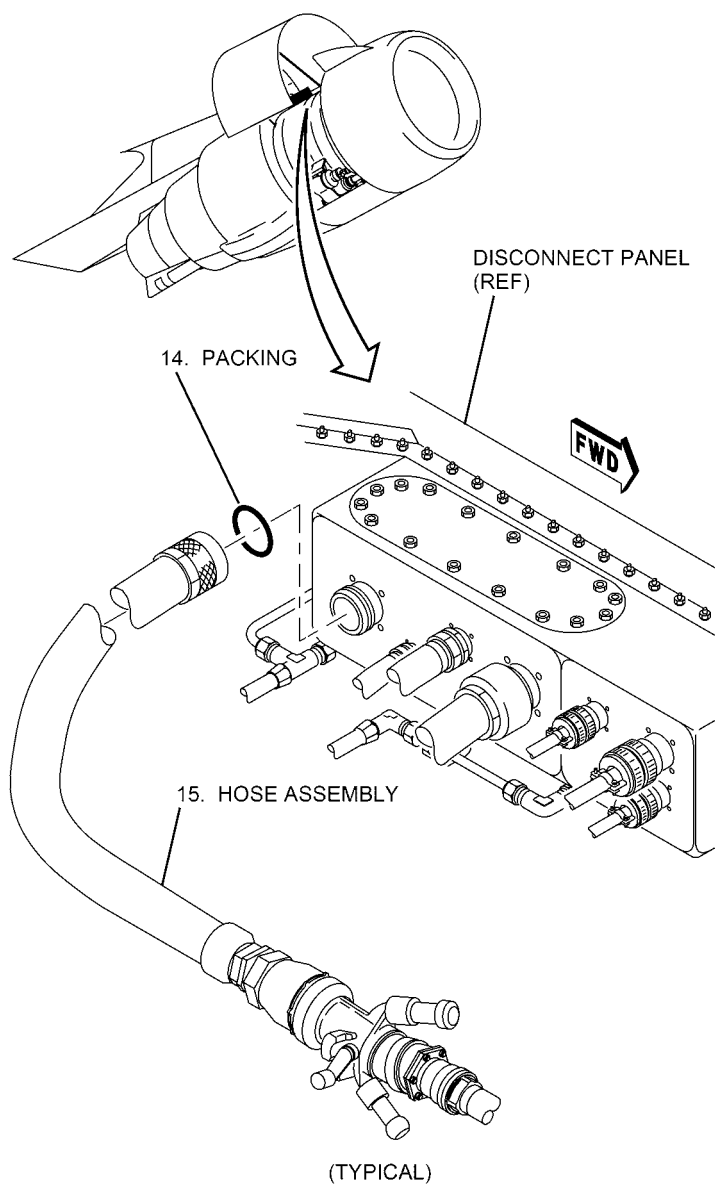
11. (A) Unlatch and open electrical ground power access door (122ARD); attach warning tag to external electrical power receptacle.
12. (B) Disconnect lower end of hose assembly from refueling nozzle of fuel equipment.



ICN-88277-G1228646-001-01

TO 1300i-2-12JG-28-2

13. (A) Disconnect upper end of hose assembly from the pylon disconnect panel.
14. (A) Remove the O-ring packing and discard.
15. (A) Drain adapter hose into bowser.



ICN-88277-G1228641-001-01

02-5. FOLLOW-ON MAINTENANCE.

1. (A) Position wheel chocks against main landing gear tires.

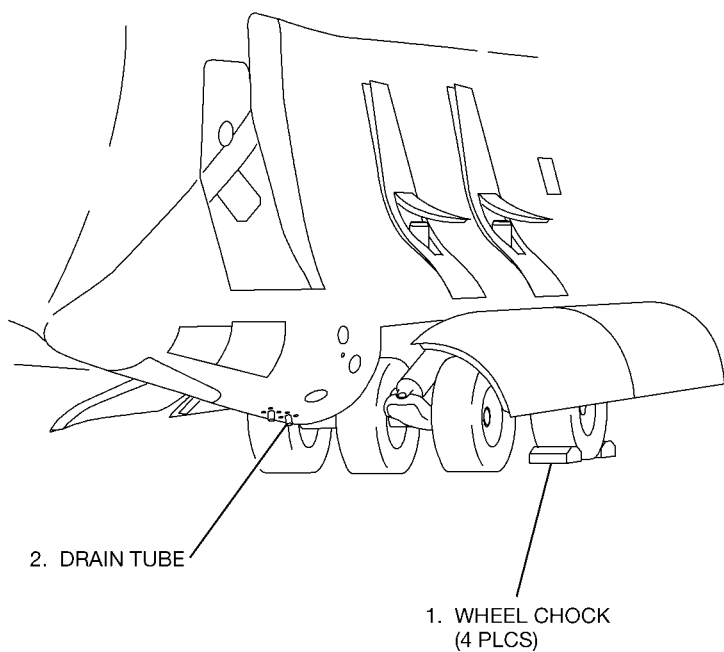
WARNING

Metal fuel containers used for draining must be bonded to aircraft prior to fuel draining.

NOTE

Manifold may contain approximately 1 gallon of fuel, when using the alternate defuel method the manifold contains approximately 45 gallons of fuel.

2. (B) Place fuel container under forward drain tube.



ICN-88277-G1228039-001-01

3. (A,B) Set **FUEL MANIFOLD DRAIN** valve to **MANIFOLD DRAIN OPEN** position.

- Fuel drains from forward drain tube.

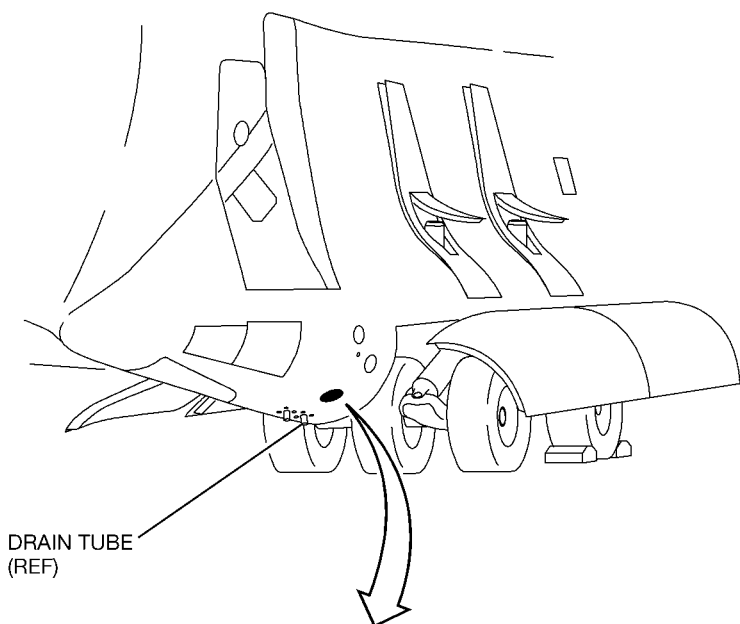
NOTE

- Scavenge lines may contain approximately 1 pint of fuel.
- If alternate method using overhead fuel panel was performed, scavenge drain not required.

4. (A,B) Set **FUEL MANIFOLD DRAIN** valve to **SCAVENGE DRAIN OPEN** position.

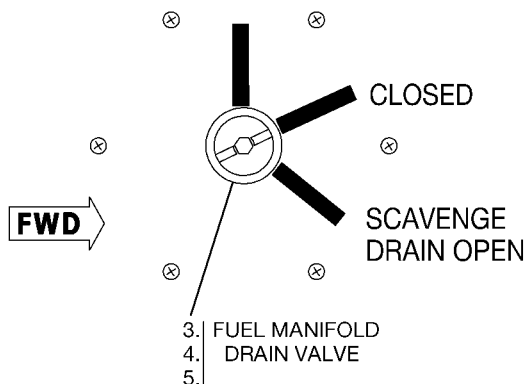
- Fuel drains from forward drain tube.

5. (A,B) Set **FUEL MANIFOLD DRAIN** valve to **CLOSED** position.



FUEL MANIFOLD DRAIN

MANIFOLD
DRAIN OPEN



ICN-88277-G1228463-001-01

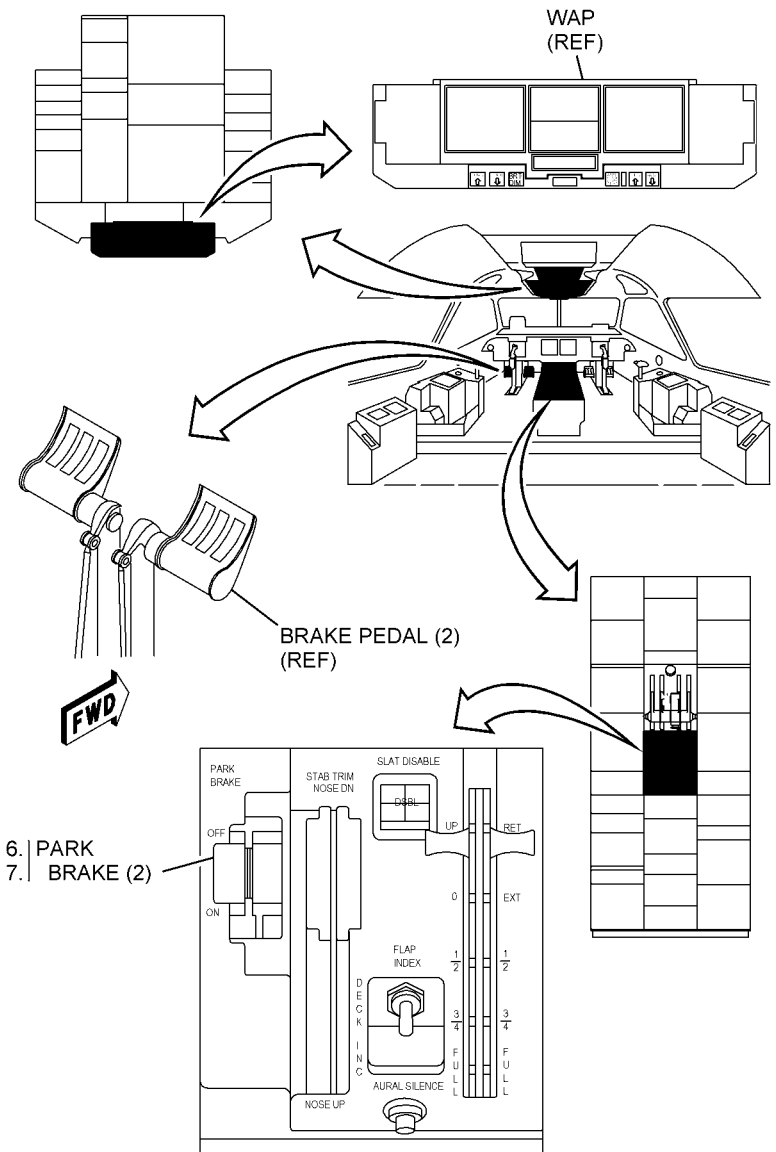
WARNING

Before releasing parking brakes, make sure personnel, stands, vehicles, or obstructions are clear of nose landing gear, main landing gear, and aircraft. Aircraft will roll FWD or AFT depending on the apron surface and weather conditions. Failure to comply may cause injury to personnel or damage to aircraft.

CAUTION

Personnel shall prevent park brake levers from moving to the forward position on their own during brake release. Failure to comply may cause damage to aircraft.

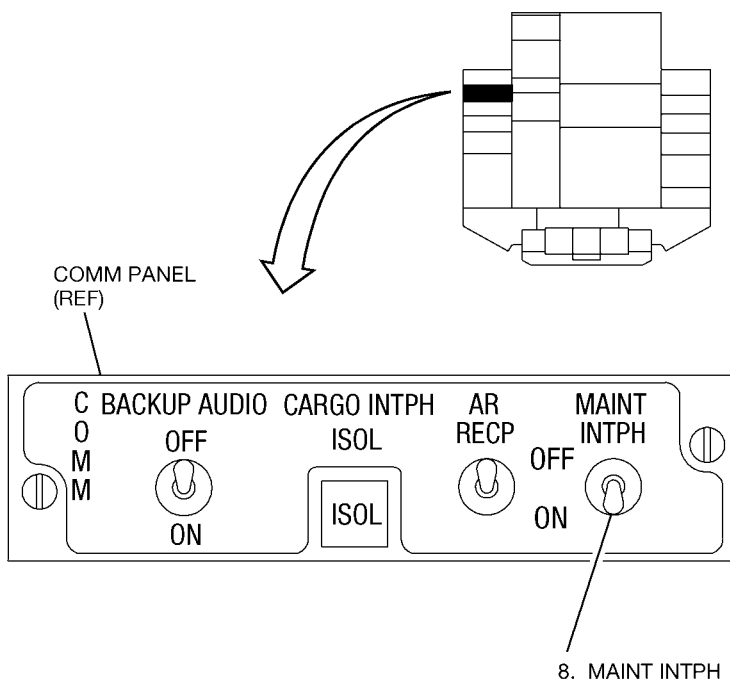
6. (B) While supporting **PARK BRAKE** levers, press and release brake pedals.
7. (B) Move **PARK BRAKE** levers to OFF.
 - **PARK BRAKE ON L R** display on Warning and caution Annunciator Panel (WAP) goes off.



ICN-88277-G1228432-004-01

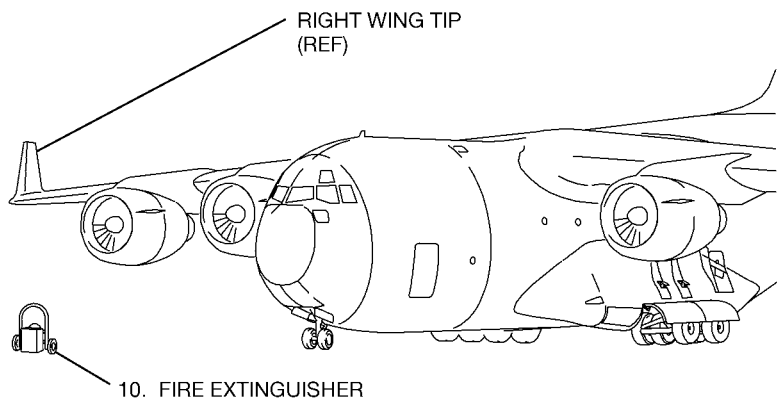
TO 1300i-2-12JG-28-2

8. (B) Set **MAINT INTPH** switch on **COMM** panel to **OFF**.
9. Disconnect external electrical power (10-61-01, task 01-2).



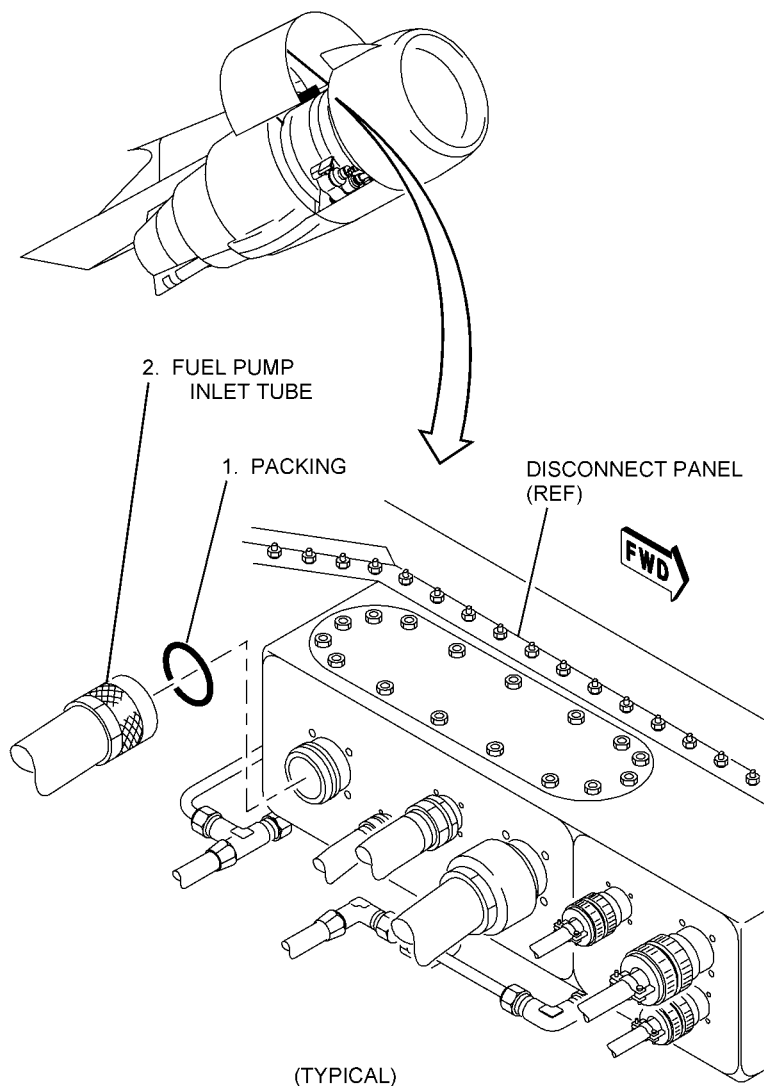
ICN-88277-G1228490-001-01

10. (A) Position fire extinguisher off right wing tip.



**02-5A. FOLLOW-ON MAINTENANCE DEFUEL
THROUGH THE ENGINE PYLON
DISCONNECT PANEL.**

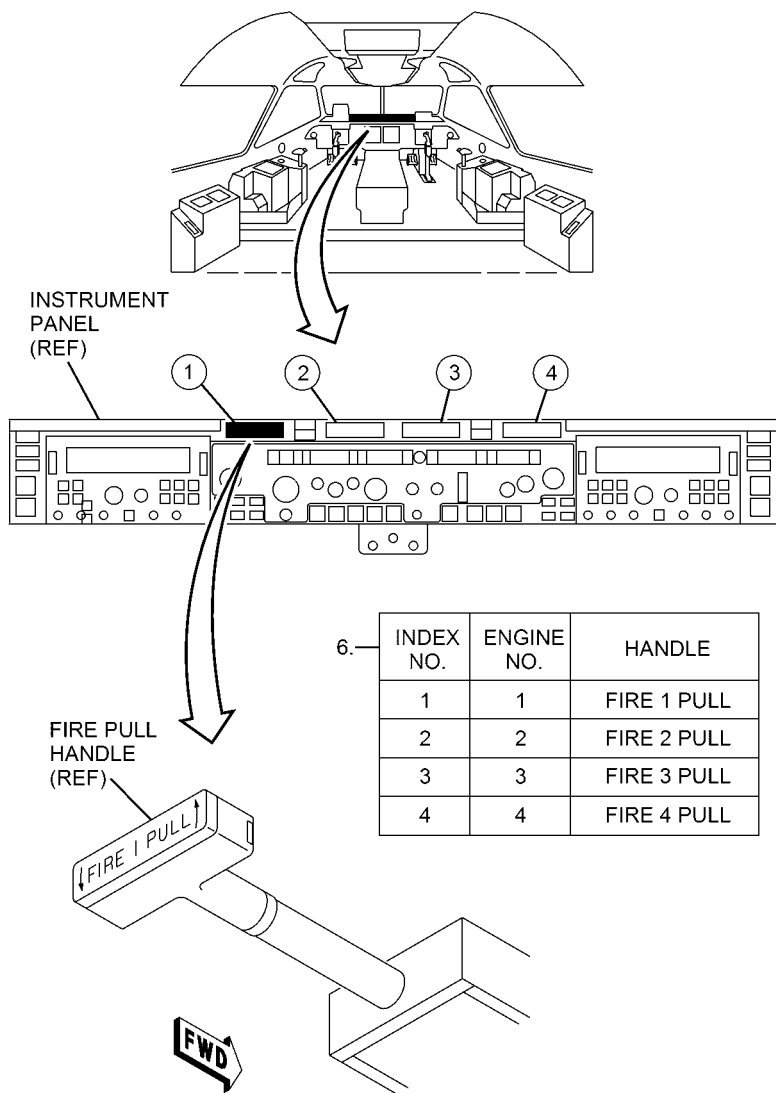
1. (A) Apply technical petrolatum to new packing and install packing.
2. (A) Connect fuel pump inlet tube at disconnect panel.



ICN-88277-G1228643-001-01

TO 1300i-2-12JG-28-2

3. (A) Remove warning tag from external electrical power receptacle in access door (122ARD).
4. (A) Remove warning tag from electrical panel and ensure **BATT** switch is **ON**.
5. (A) Connect external power (10-61-01, task 01-2).
6. (A) Remove warning tag and push in **FIRE PULL** handle on instrument panel.

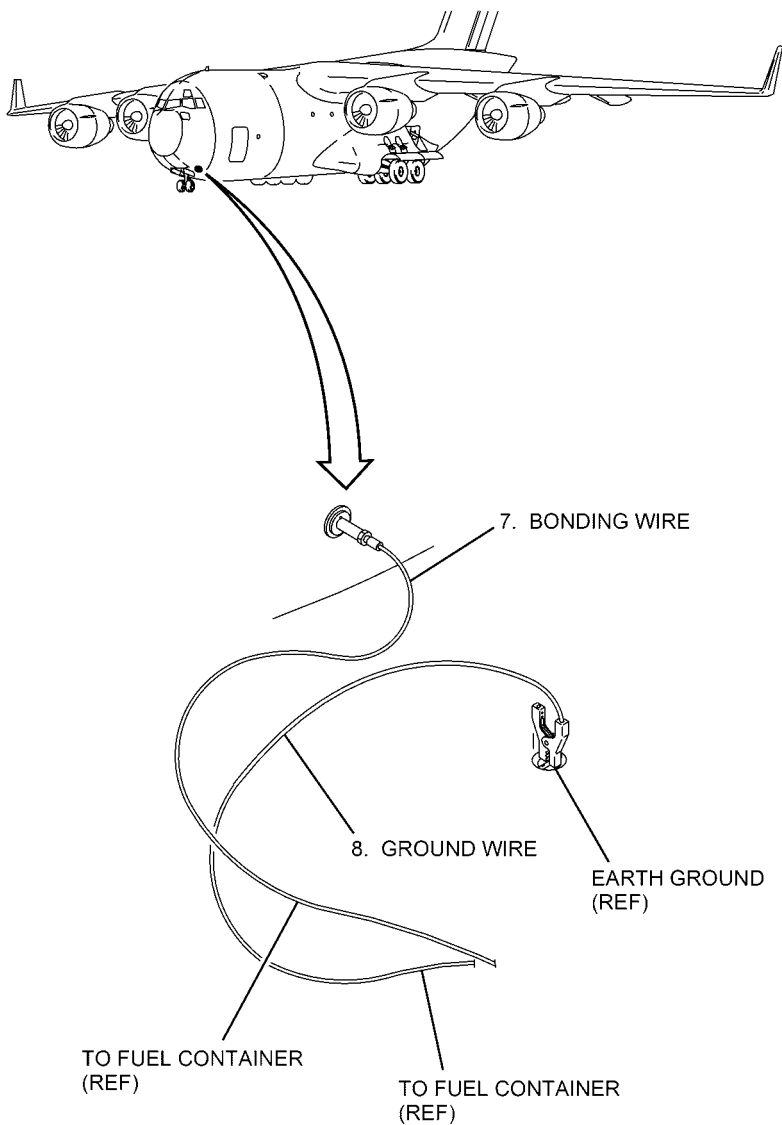


(TYPICAL)

ICN-88277-G1228644-001-01

TO 1300i-2-12JG-28-2

7. (C) Disconnect bonding wire from fuel equipment container to aircraft.
8. (C) Disconnect ground wire from fuel equipment earth ground and remove fuel equipment.
9. (C) Disconnect bonding wire from aircraft to maintenance stand.



ICN-88277-G1228645-001-01

TO 1300i-2-12JG-28-2

10. (A) Remove warning tag and close **POWER FIREX AGENT DISCHARGE 2** circuit breaker on overhead circuit breaker panel, row **F** column **2**.
11. (A) Remove warning tag and close **CONTROL FIREX AGENT DISCHARGE 2** circuit breaker on overhead circuit breaker panel, row **F** column **1**.
12. (A) Remove warning tag and close **CONTROL FIREX AGENT DISCHARGE 1** circuit breaker on overhead circuit breaker panel, row **F** column **14**.
13. (A) Remove warning tag and close **POWER FIREX AGENT DISCHARGE 1** circuit breaker on overhead circuit breaker panel, row **F** column **13**.
14. Perform power plant operational checkout - checklist (TO 1300i-2-71JG-00-1, 71-00-01, tasks 01-1, 01-2, 01-3, 01-4, 01-6, and 01-7).
15. Close engine accessory compartment door assembly (54-14-01, task 01-2).