

Last Time

Chapter 2 Rigid Body Motion

- 2 Rotational motion in \mathbb{R}^3
 - The *Exponential Map*
 - Rodrigues *Formula*
 - *Euler Angles*

Recap $R \xrightarrow{\text{SO}(3)} \text{SO}(3) \xrightarrow{\text{exp}} \text{so}(3)$

Exp Map: $e^{\hat{\omega}\theta} = I + \hat{\omega}\theta + \frac{(\hat{\omega}\theta)^2}{2!} + \frac{(\hat{\omega}\theta)^3}{3!} + \dots$

Rodsigns: $e^{\hat{\omega}\theta} = I + \hat{\omega} \sin\theta + \hat{\omega}^2 (1 - \cos\theta) \quad (\|\omega\| = 1)$

$\exp: \text{so}(3) \rightarrow \text{so}(3)$ is onto

i.e., $\forall R \in \text{SO}(3), \exists \hat{\omega} \in \text{so}(3), \theta \in \mathbb{R}$
 $\text{s.t. } e^{\hat{\omega}\theta} = R$

$$\text{tr}(R) = r_{11} + r_{22} + r_{33} = \varepsilon \lambda_i \in [-1, 3]$$

case, $\text{tr}(R) = 3 \Rightarrow \theta = 0$

Case 2 $-1 < \operatorname{tr}(R) < 3$

$$\theta = \cos^{-1} \left(\frac{\operatorname{tr}(R) - 1}{2} \right)$$

$$\omega = \frac{1}{2 \sin \theta} \begin{bmatrix} n_{32} - n_{23} \\ n_{13} - n_{31} \\ n_{21} - n_{12} \end{bmatrix}$$

Case 3 $\operatorname{tr}(R) = -1$

$$\theta = \pm \omega, \quad \omega = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} \text{ or } \begin{bmatrix} 0 \\ 1 \\ 0 \end{bmatrix} \text{ or } \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$

2.2 Rotational Motion in \mathbb{R}^3

21

□ Other Parametrizations of $SO(3)$:

- XYZ fixed angles (or Roll-Pitch-Yaw angle)

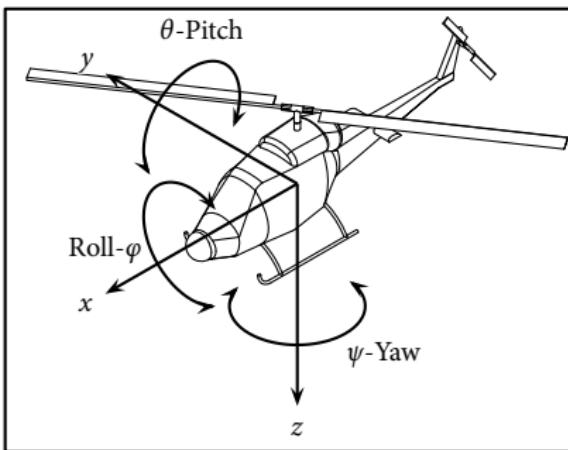


Figure 2.8

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2.2 Rotational Motion in \mathbb{R}^3

■ XYZ fixed angles (or Roll-Pitch-Yaw angle) Continued

$$R_x(\varphi) := e^{\hat{x}\varphi} = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \varphi & -\sin \varphi \\ 0 & \sin \varphi & \cos \varphi \end{bmatrix}$$

$$R_y(\theta) := e^{\hat{y}\theta} = \begin{bmatrix} \cos \theta & 0 & \sin \theta \\ 0 & 1 & 0 \\ -\sin \theta & 0 & \cos \theta \end{bmatrix}$$

$$R_z(\psi) := e^{\hat{z}\psi} = \begin{bmatrix} \cos \psi & -\sin \psi & 0 \\ \sin \psi & \cos \psi & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

$$R_{ab} = R_x(\varphi)R_y(\theta)R_z(\psi)$$

$$= \begin{bmatrix} c_\theta c_\psi & -c_\theta s_\psi & s_\theta \\ s_\varphi s_\theta c_\psi + c_\varphi s_\psi & -s_\varphi s_\theta s_\psi + c_\varphi c_\psi & -s_\varphi c_\theta \\ -c_\varphi s_\theta c_\psi + s_\varphi s_\psi & c_\varphi s_\theta s_\psi + s_\varphi c_\psi & c_\varphi c_\theta \end{bmatrix}$$

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.2 Rotational Motion in \mathbb{R}^3

23

■ ZYX Euler angle



Figure 2.9

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.2 Rotational Motion in \mathbb{R}^3

23

■ ZYX Euler angle

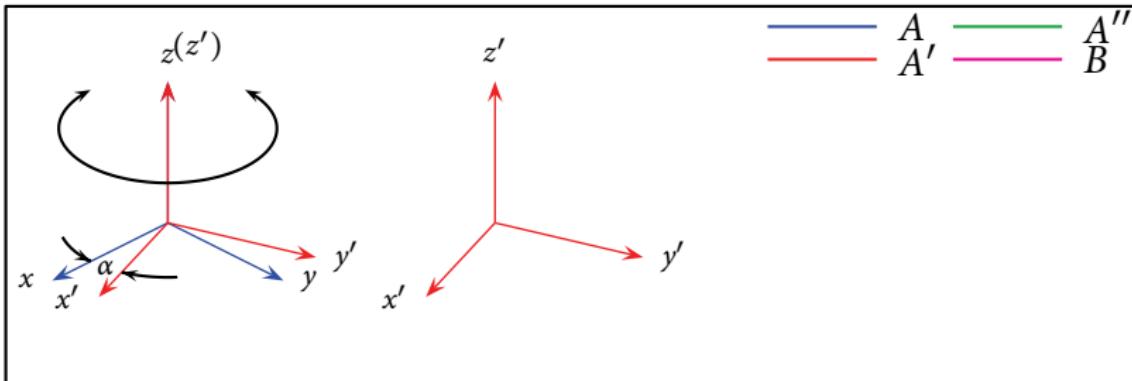


Figure 2.9

$$R_{aa'} = R_z(\alpha)$$

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.2 Rotational Motion in \mathbb{R}^3

23

■ ZYX Euler angle

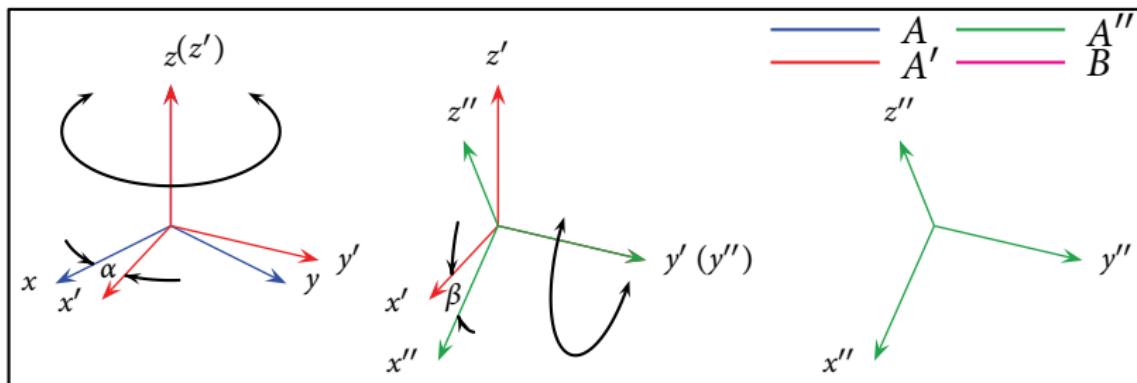


Figure 2.9

$$R_{aa'} = R_z(\alpha)$$

$$R_{a'a''} = R_y(\beta)$$

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.2 Rotational Motion in \mathbb{R}^3

23

■ ZYX Euler angle

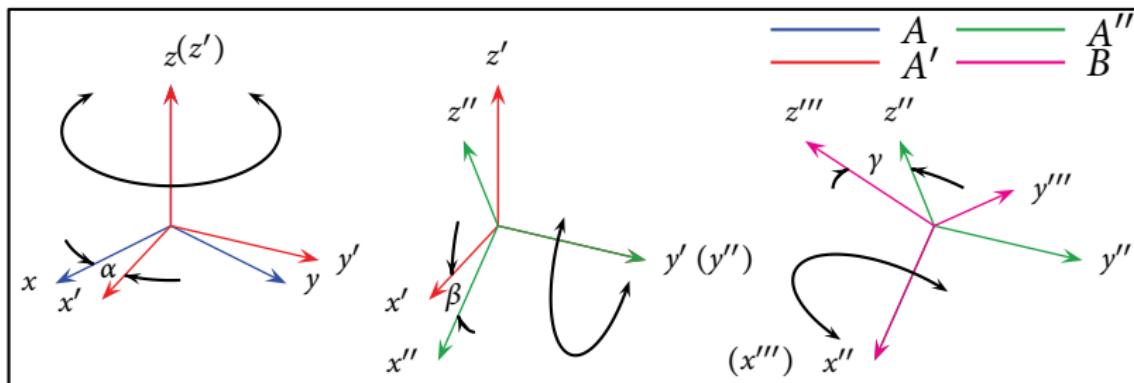


Figure 2.9

$$R_{aa'} = R_z(\alpha)$$

$$R_{a'a''} = R_y(\beta)$$

$$R_{a''b} = R_x(\gamma)$$

$$R_{ab} = R_z(\alpha)R_y(\beta)R_x(\gamma)$$

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2.2 Rotational Motion in \mathbb{R}^3

24

■ ZYX Euler angle (continued)

$$R_{ab}(\alpha, \beta, \gamma) = \begin{bmatrix} c_\alpha c_\beta & -s_\alpha c_\gamma + c_\alpha s_\beta s_\gamma & s_\alpha s_\gamma + c_\alpha s_\beta c_\gamma \\ s_\alpha c_\beta & c_\alpha c_\gamma + s_\alpha s_\beta s_\gamma & -c_\alpha s_\gamma + s_\alpha s_\beta c_\gamma \\ -s_\beta & c_\beta s_\gamma & c_\beta c_\gamma \end{bmatrix}$$

Note: When $\beta = \frac{\pi}{2}$, $\cos \beta = 0$, $\alpha + \gamma = \text{const} \Rightarrow$ singularity!

$$\beta = \text{atan2}(-r_{31}, \sqrt{r_{32}^2 + r_{33}^2})$$

$$\alpha = \text{atan2}(r_{21}/c_\beta, r_{11}/c_\beta)$$

$$\gamma = \text{atan2}(r_{32}/c_\beta, r_{33}/c_\beta)$$

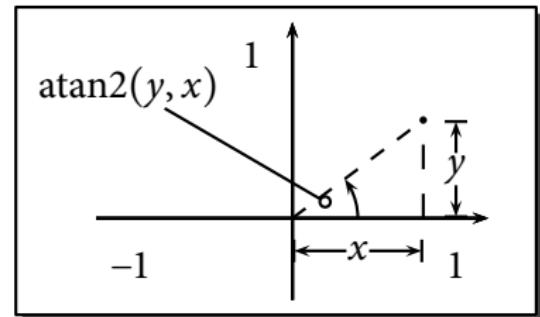


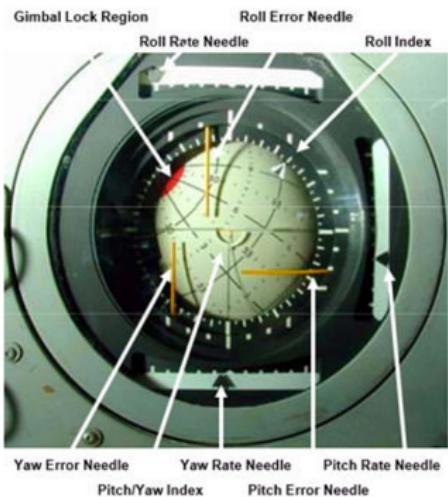
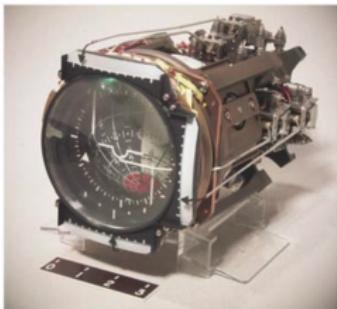
Figure 2.10

Apollo 10

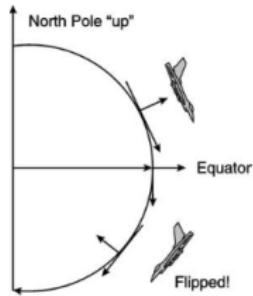


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— CSM/LM — CM/SM SEP — SIZ 8/LM —



F-16 Fly-By-Wire Fighter Jet



Today

Chapter 2 Rigid Body Motion

- 2 Rotational motion in \mathbb{R}^3
 - *Quaternions*
- 3 Rigid Motion in \mathbb{R}^3
 - SE(3)
 - As a Configuration Space
 - *Homogeneous Representation*
 - *SE(3) is a Group*
 - *SE(3) is a Rigid Body Transformation*
 - *Exponential coordinates of SE(3)*
 - *Twists*
 - *se(3)*
 - *The Exponential Map*

Today

Chapter 2 Rigid Body Motion

2 Rotational motion in \mathbb{R}^3

- The *Exponential Map*
- Rodrigues Formula
- Euler Angles
- **Quaternions**

3 Rigid Motion in \mathbb{R}^3

- SE(3)
- *Exponential coordinates of SE(3)*

Quaternions to the Rescue



$$z = e^{i\theta} = \cos \theta + i \sin \theta, \quad i^2 = -1$$

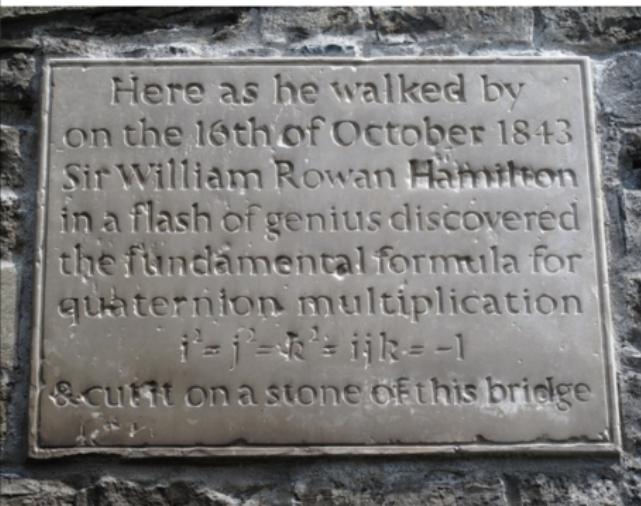
$$Q = (q_0, \vec{q}) = q_0 + i q_1$$

$\underbrace{_{\vec{q}}}_{\vec{q}}$

$$\|Q\| = q_0^2 + q_1^2 = 1$$

s^3

Hamilton's Walk



Here as he walked by
on the 16th of October 1843
Sir William Rowan Hamilton
in a flash of genius discovered
the fundamental formula for
quaternion multiplication
 $i^2 = j^2 = k^2 = ijk = -1$
& cut it on a stone of this bridge

$$Q = (q_0, \vec{q}) = q_0 + \underbrace{i q_1 + j q_2 + k q_3}_{\vec{q}}$$

$$i^2 = j^2 = k^2 = 1, \quad k = -1, \quad ij = k, \quad ji = -k, \quad kj = i, \quad ik = j$$

Rotation along axis ω by angle θ :

$$Q = \left(\cos \frac{\theta}{2}, \omega \sin \frac{\theta}{2} \right)$$

$$\|Q\| = \sqrt{q_0^2 + q_1^2 + q_2^2 + q_3^2} = 1$$

2.2 Rotational Motion in \mathbb{R}^3

25

§ Quaternions:

$$Q = q_0 + q_1 i + q_2 j + q_3 k$$

where $i^2 = j^2 = k^2 = -1, i \cdot j = k, j \cdot k = i, k \cdot i = j$

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.2 Rotational Motion in \mathbb{R}^3

25

§ Quaternions:

$$Q = q_0 + q_1 i + q_2 j + q_3 k$$

where $i^2 = j^2 = k^2 = -1, i \cdot j = k, j \cdot k = i, k \cdot i = j$

Property 1: Define $Q^* = (q_0, q)^* = (q_0, -q), q_0 \in \mathbb{R}, q \in \mathbb{R}^3$
 $\|Q\|^2 = QQ^* = q_0^2 + q_1^2 + q_2^2 + q_3^2$

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.2 Rotational Motion in \mathbb{R}^3

§ Quaternions:

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Property 1: Define $Q^* = (q_0, q)^* = (q_0, -q), q_0 \in \mathbb{R}, q \in \mathbb{R}^3$

$$\|Q\|^2 = QQ^* = q_0^2 + q_1^2 + q_2^2 + q_3^2$$

Property 2: $Q = (q_0, \vec{q}), P = (p_0, \vec{p})$

$$QP = (q_0p_0 - \vec{q} \cdot \vec{p}, q_0\vec{p} + p_0\vec{q} + \vec{q} \times \vec{p})$$

Chapter
2 Rigid Body
Motion

Rigid Body
Transformations

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.2 Rotational Motion in \mathbb{R}^3

25

§ Quaternions:

Chapter
2 Rigid Body
Motion

Rigid Body
Transformations

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

$$Q = q_0 + q_1 i + q_2 j + q_3 k$$

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Property 1: Define $Q^* = (q_0, q)^* = (q_0, -q), q_0 \in \mathbb{R}, q \in \mathbb{R}^3$

$$\|Q\|^2 = QQ^* = q_0^2 + q_1^2 + q_2^2 + q_3^2$$

Property 2: $Q = (q_0, q), P = (p_0, p)$
 $QP = (q_0p_0 - q \cdot p, q_0p + p_0q + q \times p)$

Property 3: (a) The set of unit quaternions forms a group
(b) If $R = e^{\hat{\omega}\theta}$, then $Q = (\cos \frac{\theta}{2}, \omega \sin \frac{\theta}{2})$
(c) Q acts on $x \in \mathbb{R}^3$ by QXQ^* , where $X = (0, x)$

Quaternions double cover space of Rotations

2.2 Rotational Motion in \mathbb{R}^3

26

□ Unit Quaternions:

Given $Q = (q_0, \mathbf{q})$, $q_0 \in \mathbb{R}$, $\mathbf{q} \in \mathbb{R}^3$, the vector part of QXQ^* is given by $R(Q)x$, recall that

$$q_0 = \cos \frac{\theta}{2}, \mathbf{q} = \boldsymbol{\omega} \sin \frac{\theta}{2}$$

and the Rodrigues' formula:

$$e^{\hat{\boldsymbol{\omega}}\theta} = I + \hat{\boldsymbol{\omega}} \sin \theta + \hat{\boldsymbol{\omega}}^2(1 - \cos \theta)$$

then

$$R(Q) = I + 2q_0\hat{\mathbf{q}} + 2\hat{\mathbf{q}}^2$$

$$= \begin{bmatrix} 1 - 2(q_2^2 + q_3^2) & -2q_0q_3 + 2q_1q_2 & 2q_0q_2 + 2q_1q_3 \\ 2q_0q_3 + 2q_1q_2 & 1 - 2(q_1^2 + q_3^2) & -2q_0q_1 + 2q_2q_3 \\ -2q_0q_2 + 2q_1q_3 & 2q_0q_1 + 2q_2q_3 & 1 - 2(q_1^2 + q_2^2) \end{bmatrix}$$

where $\|Q\| \triangleq q_0^2 + q_1^2 + q_2^2 + q_3^2 = 1$

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2.2 Rotational Motion in \mathbb{R}^3

27

□ Quaternions (continued):

Conversion from Roll-Pitch-Yaw angle to unit quaternions:

$$Q = \left(\cos \frac{\varphi}{2}, x \sin \frac{\varphi}{2} \right) \left(\cos \frac{\theta}{2}, y \sin \frac{\theta}{2} \right) \left(\cos \frac{\psi}{2}, z \sin \frac{\psi}{2} \right) \Rightarrow$$

$$q_0 = \cos \frac{\varphi}{2} \cos \frac{\theta}{2} \cos \frac{\psi}{2} - \sin \frac{\varphi}{2} \sin \frac{\theta}{2} \sin \frac{\psi}{2}$$

$$q = \begin{bmatrix} \cos \frac{\varphi}{2} \sin \frac{\theta}{2} \sin \frac{\psi}{2} + \sin \frac{\varphi}{2} \cos \frac{\theta}{2} \cos \frac{\psi}{2} \\ \cos \frac{\varphi}{2} \sin \frac{\theta}{2} \cos \frac{\psi}{2} - \sin \frac{\varphi}{2} \cos \frac{\theta}{2} \sin \frac{\psi}{2} \\ \cos \frac{\varphi}{2} \cos \frac{\theta}{2} \sin \frac{\psi}{2} + \sin \frac{\varphi}{2} \sin \frac{\theta}{2} \cos \frac{\psi}{2} \end{bmatrix}$$

Conversion from unit quaternions to roll-pitch-yaw angles (?)

† End of Section †

Today

Chapter 2 Rigid Body Motion

- 2 Rotational motion in \mathbb{R}^3
 - *Quaternions*
- 3 Rigid Motion in \mathbb{R}^3
 - **SE(3)**
 - As a Configuration Space
 - *Homogeneous Representation*
 - *SE(3) is a Group*
 - *SE(3) is a Rigid Body Transformation*
 - *Exponential coordinates of SE(3)*
 - *Twists*
 - *se(3)*
 - *The Exponential Map*

2.3 Rigid motion in \mathbb{R}^3

28

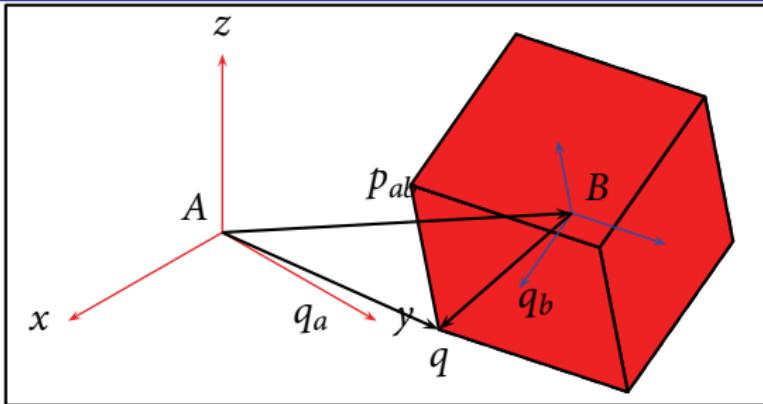


Figure 2.11

$p_{ab} \in \mathbb{R}^3$: Coordinates of the origin of \$B\$

$R_{ab} \in SO(3)$: Orientation of \$B\$ relative to \$A\$

$SE(3) : \{(p, R) | p \in \mathbb{R}^3, R \in SO(3)\}$: Configuration Space

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.3 Rigid motion in \mathbb{R}^3

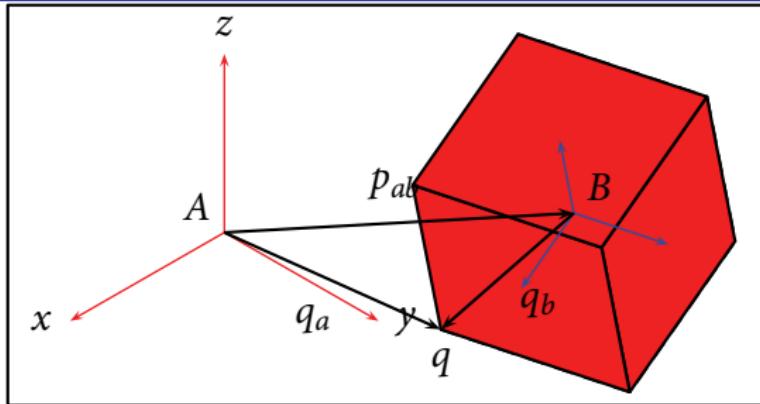


Figure 2.11

$p_{ab} \in \mathbb{R}^3$: Coordinates of the origin of B

$$R_{ab} \cdot q_{b,a} \mapsto q_b$$

$R_{ab} \in SO(3)$: Orientation of B relative to A

$$q_{b,a} = R_{ab} q_b$$

$SE(3) : \{(p, R) | p \in \mathbb{R}^3, R \in SO(3)\}$: Configuration Space

Or...as a transformation:

$$g_{ab} = (p_{ab}, R_{ab}) : \mathbb{R}^3 \mapsto \mathbb{R}^3$$

$$q_b \mapsto q_a = p_{ab} + \underline{R_{ab} \cdot q_b}$$

Today

Chapter 2 Rigid Body Motion

2 Rotational motion in \mathbb{R}^3

- *Quaternions*

3 Rigid Motion in \mathbb{R}^3

- $SE(3)$
 - As a Configuration Space
 - *Homogeneous Representation*
 - $SE(3)$ is a Group
 - $SE(3)$ is a Rigid Body Transformation
- *Exponential coordinates of $SE(3)$*
 - Twists
 - $se(3)$
 - *The Exponential Map*

2.3 Rigid motion in \mathbb{R}^3

29

□ Homogeneous Representation:

Points:

$$q = \begin{bmatrix} q_1 \\ q_2 \\ q_3 \end{bmatrix} \in \mathbb{R}^3$$


$$\bar{q} = \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ 1 \end{bmatrix} \in \mathbb{R}^4$$

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.3 Rigid motion in \mathbb{R}^3

□ Homogeneous Representation:

Points:

$$q = \begin{bmatrix} q_1 \\ q_2 \\ q_3 \end{bmatrix} \in \mathbb{R}^3$$



$$\bar{q} = \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ 1 \end{bmatrix} \in \mathbb{R}^4$$

Vectors:

$$v = p - q = \begin{bmatrix} p_1 - q_1 \\ p_2 - q_2 \\ p_3 - q_3 \end{bmatrix} = \begin{bmatrix} v_1 \\ v_2 \\ v_3 \end{bmatrix}$$



$$\bar{v} = \bar{p} - \bar{q} = \begin{bmatrix} p_1 \\ p_2 \\ p_3 \\ 1 \end{bmatrix} - \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ 1 \end{bmatrix} = \begin{bmatrix} v_1 \\ v_2 \\ v_3 \\ 0 \end{bmatrix}$$

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.3 Rigid motion in \mathbb{R}^3

□ Homogeneous Representation:

Points:

$$q = \begin{bmatrix} q_1 \\ q_2 \\ q_3 \end{bmatrix} \in \mathbb{R}^3$$



$$\bar{q} = \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ 1 \end{bmatrix} \in \mathbb{R}^4$$

Vectors:

$$v = p - q = \begin{bmatrix} p_1 - q_1 \\ p_2 - q_2 \\ p_3 - q_3 \end{bmatrix} = \begin{bmatrix} v_1 \\ v_2 \\ v_3 \end{bmatrix}$$



$$\bar{v} = \bar{p} - \bar{q} = \begin{bmatrix} p_1 \\ p_2 \\ p_3 \end{bmatrix} - \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ 1 \end{bmatrix} = \begin{bmatrix} v_1 \\ v_2 \\ v_3 \\ 0 \end{bmatrix}$$

- 1 Point-Point = Vector
- 2 Vector+Point = Point
- 3 Vector+Vector = Vector
- 4 Point+Point: Meaningless

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2.3 Rigid motion in \mathbb{R}^3

30

$$q_a = p_{ab} + R_{ab} \cdot q_b$$

$$\begin{bmatrix} q_a \\ 1 \end{bmatrix} = \underbrace{\begin{bmatrix} R_{ab} & p_{ab} \\ 0 & 1 \end{bmatrix}}_{\bar{g}_{ab}} \begin{bmatrix} q_b \\ 1 \end{bmatrix}$$

$$\bar{q}_a = \bar{g}_{ab} \cdot \bar{q}_b$$

$$g_{ab} = (p_{ab}, R_{ab})$$

$\in \mathbb{SE}(3)$



$$\bar{g}_{ab} = \begin{bmatrix} R_{ab} & p_{ab} \\ 0 & 1 \end{bmatrix}$$

$\in \mathbb{IR}^{4 \times 4}$

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.3 Rigid motion in \mathbb{R}^3

$$q_a = p_{ab} + R_{ab} \cdot q_b$$

$$\begin{bmatrix} q_a \\ 1 \end{bmatrix} = \underbrace{\begin{bmatrix} R_{ab} & p_{ab} \\ 0 & 1 \end{bmatrix}}_{\bar{g}_{ab}} \begin{bmatrix} q_b \\ 1 \end{bmatrix}$$

$$\bar{q}_a = \bar{g}_{ab} \cdot \bar{q}_b$$

$$g_{ab} = (p_{ab}, R_{ab})$$



$$\bar{g}_{ab} = \begin{bmatrix} R_{ab} & p_{ab} \\ 0 & 1 \end{bmatrix}$$

□ Composition Rule:

$$\bar{q}_b = \bar{g}_{bc} \cdot \bar{q}_c$$

$$\bar{q}_a = \bar{g}_{ab} \cdot \bar{q}_b$$

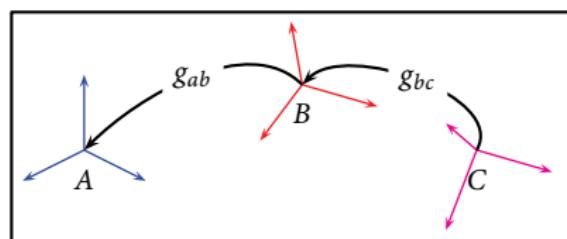


Figure 2.12

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.3 Rigid motion in \mathbb{R}^3

30

$$q_a = p_{ab} + R_{ab} \cdot q_b$$

$$\begin{bmatrix} q_a \\ 1 \end{bmatrix} = \underbrace{\begin{bmatrix} R_{ab} & p_{ab} \\ 0 & 1 \end{bmatrix}}_{\bar{g}_{ab}} \begin{bmatrix} q_{ab} \\ 1 \end{bmatrix}$$

$$\bar{q}_a = \bar{g}_{ab} \cdot \bar{q}_b$$

$$g_{ab} = (p_{ab}, R_{ab}) \in \mathfrak{so}(3)$$

$$\bar{g}_{ab} = \begin{bmatrix} R_{ab} & P_{ab} \\ 0 & 1 \end{bmatrix}$$

Composition Rule:

$$\begin{bmatrix} R_{ab} & P_{ab} \\ 0 & 1 \end{bmatrix} \begin{bmatrix} R_{bc} & P_{bc} \\ 0 & 1 \end{bmatrix}$$

$$\bar{q}_b = \bar{g}_{bc} \cdot \bar{q}_c$$

$$\bar{q}_a = \bar{g}_{ab} \cdot \bar{q}_b = \underbrace{\bar{g}_{ab} \cdot \bar{g}_{bc}}_{\bar{g}_{ac}} \cdot \bar{q}_c$$

$$\bar{g}_{ac} = \bar{g}_{ab} \cdot \bar{g}_{bc} = \begin{bmatrix} R_{ab}R_{bc} & R_{ab}p_{bc} + p_{ab} \\ 0 & 1 \end{bmatrix} \in SE(3)$$

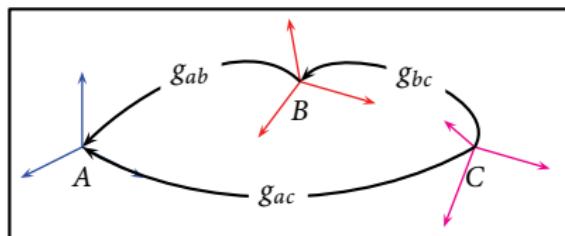


Figure 2.12

Today

Chapter 2 Rigid Body Motion

- 2 Rotational motion in \mathbb{R}^3
 - *Quaternions*
- 3 Rigid Motion in \mathbb{R}^3
 - SE(3)
 - As a Configuration Space
 - *Homogeneous Representation*
 - ***SE(3) is a Group***
 - *SE(3) is a Rigid Body Transformation*
 - *Exponential coordinates of SE(3)*
 - *Twists*
 - *se(3)*
 - *The Exponential Map*

2.3 Rigid motion in \mathbb{R}^3

31

□ Special Euclidean Group:

$$SE(3) = \left\{ \begin{bmatrix} R & p \\ 0 & 1 \end{bmatrix} \in \mathbb{R}^{4 \times 4} \middle| p \in \mathbb{R}^3, R \in SO(3) \right\}$$

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.3 Rigid motion in \mathbb{R}^3

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Property 4: $SE(3)$ forms a group.

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

2.3 Rigid motion in \mathbb{R}^3

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Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

Property 4: $SE(3)$ forms a group.

Proof : $\begin{bmatrix} R_1 & p_1 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} R_2 & p_2 \\ 0 & 1 \end{bmatrix} = \begin{bmatrix} R_1 R_2 & R_1 p_2 + p_1 \\ 0 & 1 \end{bmatrix} \in SE(3)$

1 $g_1 \cdot g_2 \in SE(3)$

2 $e = I_4$

3 $(\bar{g})^{-1} = \begin{bmatrix} R^T & -R^T p \\ 0 & 1 \end{bmatrix} \begin{bmatrix} R & p \\ 0 & 1 \end{bmatrix} = \begin{bmatrix} R^T R & R^T p - R^T p \\ 0 & 1 \end{bmatrix} = \begin{bmatrix} I & 0 \\ 0 & 1 \end{bmatrix}$

4 Associativity: Follows from property of matrix multiplication



Today

Chapter 2 Rigid Body Motion

2 Rotational motion in \mathbb{R}^3

- *Quaternions*

3 Rigid Motion in \mathbb{R}^3

- $SE(3)$
 - As a Configuration Space
 - *Homogeneous Representation*
 - *$SE(3)$ is a Group*
 - *$SE(3)$ is a Rigid Body Transformation*
- *Exponential coordinates of $SE(3)$*
 - *Twists*
 - *$se(3)$*
 - *The Exponential Map*

2.3 Rigid motion in \mathbb{R}^3

32

§ Induced transformation on vectors:

$$\bar{v} = s - r = \begin{bmatrix} v_1 \\ v_2 \\ v_3 \\ 0 \end{bmatrix}, \boxed{\bar{g}_* \bar{v}} = \bar{g}s - \bar{g}r = \begin{bmatrix} R & p \\ 0 & 1 \end{bmatrix} \begin{bmatrix} v_1 \\ v_2 \\ v_3 \\ 0 \end{bmatrix} = \boxed{\begin{bmatrix} Rv \\ 0 \end{bmatrix}}$$

The bar will be dropped to simplify notations

$$q_a = R_{ab} q_b + p_{ab}$$

Property 5: An element of $SE(3)$ is a rigid transformation.

(1) Length is preserved . $\|g q_1 - g q_2\| = \|q_1 - q_2\|$

$$\begin{aligned} g = \begin{bmatrix} R & p \\ 0 & 1 \end{bmatrix} \Rightarrow LHS &= \left\| \begin{bmatrix} p + R q_1 \\ 1 \end{bmatrix} - \begin{bmatrix} p + R q_2 \\ 1 \end{bmatrix} \right\| \\ &= \left\| \begin{bmatrix} R q_1 - R q_2 \\ 0 \end{bmatrix} \right\| = \|R(q_1 - q_2)\| \\ &= \|q_1 - q_2\| \end{aligned}$$

i) Orientation is preserved. $g_{\pi^*} \times g_{\pi^*} = g_{\pi^*}(v \times w)$

$$g_{\pi^*} v \times g_{\pi^*} w = \begin{bmatrix} Rv \\ 0 \end{bmatrix} \times \begin{bmatrix} Rw \\ 0 \end{bmatrix} = \begin{bmatrix} Rv \times Rw \\ 0 \end{bmatrix} = \begin{bmatrix} R(v \times w) \\ 0 \end{bmatrix}$$
$$= g_{\pi^*}(v \times w)$$

Today

Chapter 2 Rigid Body Motion

- 2 Rotational motion in \mathbb{R}^3
 - *Quaternions*
- 3 Rigid Motion in \mathbb{R}^3
 - $SE(3)$
 - As a Configuration Space
 - *Homogeneous Representation*
 - *$SE(3)$ is a Group*
 - *$SE(3)$ is a Rigid Body Transformation*
 - *Exponential coordinates of $SE(3)$*
 - *Twists*
 - *$se(3)$*
 - *The Exponential Map*

2.3 Rigid motion in \mathbb{R}^3

Exponential coordinates of $SE(3)$:

For rotational motion: $\hat{\omega} \times \vec{p} - \omega \times q$

$$\begin{bmatrix} \dot{p}(t) \\ \dot{p} \\ 0 \end{bmatrix} = \begin{bmatrix} \hat{\omega} & -\omega \times q \\ 0 & 0 \end{bmatrix} \begin{bmatrix} p \\ 1 \end{bmatrix}$$

$$\text{or } \dot{\bar{p}} = \hat{\xi} \cdot \bar{p} \Rightarrow \bar{p}(t) = e^{\hat{\xi}t} \bar{p}(0)$$

$$\text{where } e^{\hat{\xi}t} = I + \hat{\xi}t + \frac{(\hat{\xi}t)^2}{2!} + \dots$$

$$\begin{bmatrix} \hat{\omega} \\ \hat{\xi} \end{bmatrix} = \hat{\epsilon}$$

$$\begin{aligned} \dot{\bar{P}} &= \hat{\xi} \bar{P}(t) \\ \Rightarrow \bar{P}(t) &= e^{\hat{\xi}t} \bar{P}(0) \end{aligned}$$

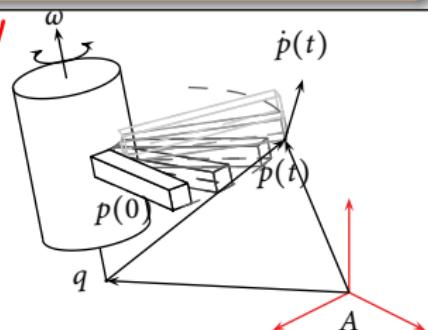


Figure 2.13

2.3 Rigid motion in \mathbb{R}^3

Exponential coordinates of $SE(3)$:

Chapter
2 Rigid Body
Motion

Rigid Body
Transforma-
tions

Rotational
motion in \mathbb{R}^3

Rigid Motion
in \mathbb{R}^3

Velocity of a
Rigid Body

Wrenches and
Reciprocal
Screws

Reference

For rotational motion:

$$\dot{\vec{p}}(t) = \omega \times (\vec{p}(t) - \vec{q})$$

$$\begin{bmatrix} \dot{\vec{p}} \\ \dot{\vec{p}} \end{bmatrix} = \begin{bmatrix} \hat{\omega} & -\omega \times \vec{q} \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \vec{p} \\ 1 \end{bmatrix}$$

$$\text{or } \dot{\vec{p}} = \hat{\xi} \cdot \vec{p} \Rightarrow \vec{p}(t) = e^{\hat{\xi}t} \vec{p}(0)$$

$$\text{where } e^{\hat{\xi}t} = I + \hat{\xi}t + \frac{(\hat{\xi}t)^2}{2!} + \dots$$

For translational motion:

$$\dot{\vec{p}}(t) = \nu$$

$$\begin{bmatrix} \dot{\vec{p}}(t) \\ 0 \end{bmatrix} = \begin{bmatrix} 0 & \nu \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \vec{p} \\ 1 \end{bmatrix}$$

$$\dot{\vec{p}}(t) = \hat{\xi} \cdot \vec{p}(t) \Rightarrow \vec{p}(t) = e^{\hat{\xi}t} \vec{p}(0)$$

$$\hat{\xi} = \begin{bmatrix} 0 & \nu \\ 0 & 0 \end{bmatrix}$$

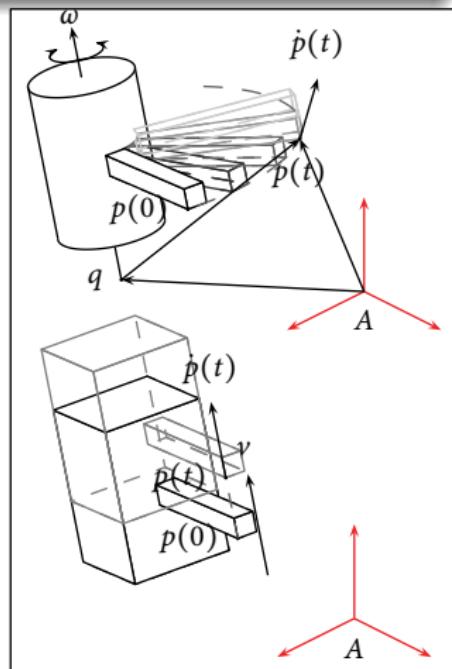


Figure 2.13