## Mark Scheme Gravity Fields Past Paper Questions Jan 2002—Jan 2010 (old spec)

2

(a) period = 24 hours or equals period of Earth's rotation ✓ remains in fixed position relative to surface of Earth ✓ equatorial orbit ✓ same angular speed as Earth or equatorial surface ✓ (2)

(b)(i) 
$$\frac{GMm}{r^2} = m\omega^2 r \checkmark$$

$$T = \frac{2\pi}{\omega} \checkmark$$

$$r \left( = \frac{GMT^2}{4\pi^2} \right)^{1/3} = \left( \frac{6.67 \times 10^{-11} \times 6 \times 10^{24} \times (24 \times 3600)^2}{4\pi^2} \right)^{1/3} \checkmark$$

(gives 
$$r = 42.3 \times 10^{3}$$
 km)  
(b)(ii)  $\Delta V = GM \left(\frac{1}{R} - \frac{1}{r}\right) \checkmark$   
 $= 6.67 \times 10^{-11} \times 6 \times 10^{24} \times \left(\frac{1}{6.4 \times 10^{6}} - \frac{1}{4.23 \times 10^{7}}\right) = 5.31 \times 10^{7} \text{ (J kg}^{-1)} \checkmark$   
 $\Delta E_{\rm p} = m\Delta V (= 750 \times 5.31 \times 10^{7}) = 3.98 \times 10^{10} \text{ J} \checkmark$   
(allow C.E. for value of  $\Delta V$ )

[alternatives:

calculation of 
$$\frac{GM}{R}$$
 (6.25 × 10<sup>7</sup>) or  $\frac{GM}{r}$  (9.46 × 10<sup>6</sup>)  $\checkmark$  or calculation of  $\frac{GMm}{R}$  (4.69 × 10<sup>10</sup>) or  $\frac{GMm}{r}$  (7.10× 10<sup>9</sup>) calculation of both potential energy values  $\checkmark$  subtraction of values or use of  $m\Delta V$  with correct answer  $\checkmark$ 

(<u>6</u>) (8)

(a) work = force × distance moved in direction of force ✓
 (in circular motion) force is perpendicular to displacement ✓
 no movement in direction of force ✓ (hence no work)
 [or speed of body remains constant (although velocity changes) ✓
 kinetic energy is constant ✓
 potential energy is constant ✓

[or gravitational force acts towards the Earth  $\checkmark$  Moon remains at constant distance/radius from Earth  $\checkmark$  since radius is unchanged, gravitational force does no work or  $E_p$  of Moon is constant  $\checkmark$ ]

(3)

- (b)(i) any suitable example of circular motion ✓
  - (ii) any SHM example at maximum displacement ✓[or any other suitable example, e.g. car starts from rest](2)

<u>(5)</u>

Quality of Written Communication: Q3(a) and Q4(a)  $\checkmark\checkmark$  (2)

(2)

## **Question 3**

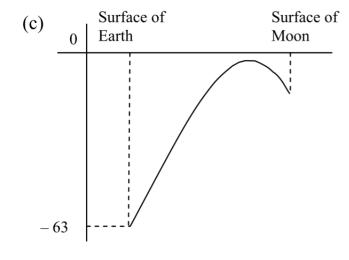
(a) work done/energy change (against the field) per unit mass ✓ when moved from infinity to the point ✓ (2)

Q3 Jan 2005

(b) 
$$V_{\rm E} = -\frac{GM_{\rm E}}{R_{\rm E}} \text{ and } V_{\rm M} = -\frac{GM_{\rm M}}{R_{\rm M}} \checkmark$$

$$V_{\rm M} = -G \times \frac{M_{\rm E}}{81} \times \frac{3.7}{R_{\rm E}} = \frac{3.7}{81} V_{\rm E} \checkmark$$

$$= 4.57 \times 10^{-2} \times (-63) = -2.9 \,\text{MJ kg}^{-1} \checkmark \qquad (2.88 \,\text{MJ kg}^{-1}) \qquad (3)$$



limiting values  $(-63, -V_{\rm M})$  on correctly curving line  $\checkmark$  rises to value close to but below zero  $\checkmark$  falls to Moon  $\checkmark$  from point much closer to M than E  $\checkmark$ 

 $\max(3)$ (8)

Question 4	Q4 Jun 2005	
(a)	attractive <b>force</b> between point masses ✓ proportional to (product of) the masses ✓ inversely proportional to square of separation/distance apart ✓	3
(b)	$m\omega^{2}R = (-)\frac{GMm}{R^{2}}\left(\text{ or } = \frac{mv^{2}}{R}\right) \checkmark$ (use of $T = \frac{2\pi}{\omega}$ gives) $\frac{4\pi^{2}}{T^{2}} = \frac{GM}{R^{3}}$ $\checkmark$ $G$ and $M$ are constants, hence $T^{2} \propto R^{3}$ $\checkmark$	3
(c) (i)	(use of $T^2 \propto R^3$ gives) $\frac{365^2}{(1.50 \times 10^{11})^3} = \frac{T_m^2}{(5.79 \times 10^{10})^3} \checkmark$ $T_m = 87(.5) \text{ days } \checkmark$	
(ii)	$\frac{1^2}{(1.50 \times 10^{11})^3} = \frac{165^2}{R_N^3} \checkmark \text{(gives } R_N = 4.52 \times 10^{12} \text{ m)}$ $\text{ratio} = \frac{4.51 \times 10^{12}}{1.50 \times 10^{11}} = 30(.1) \checkmark$	4

Question 4	Q4 Jan 2006	
(a)	orbits (westwards) over Equator ✓ maintains a fixed position relative to surface of Earth ✓ period is 24 hrs (1 day) or same as for Earth's rotation ✓ offers uninterrupted communication between transmitter and receiver ✓ steerable dish not necessary ✓	Max 3
(b) (i)	$G\frac{Mm}{(R+h)^2} = m\omega^2 (R+h) \checkmark$	
(ii)	use of $\omega = \frac{2\pi}{T} \checkmark$ gives $\frac{GM}{(R+h)^3} = \frac{4\pi^2}{T^2}$ , hence result $\checkmark$	6
(iii)	limiting case is orbit at zero height i.e. $h = 0$ $T^{2} = \left(\frac{4\pi^{2}R^{3}}{GM}\right) = \frac{4\pi^{2} \times (6.4 \times 10^{6})^{3}}{6.67 \times 10^{-11} \times 6.0 \times 10^{24}} \checkmark$ $T = 5090 \text{ s } \checkmark (= 85 \text{ min})$	
(c)	speed increases $\checkmark$ loses potential energy but gains kinetic energy $\checkmark$ [or because $v^2 \propto \frac{1}{r}$ from $\frac{GMm}{r^2} = \frac{mv^2}{r}$ ] [or because satellite must travel faster to stop it falling inwards when gravitational force increases]	2
	Total	11

Question 4		
(a)	force per unit mass $\checkmark$ [or force on a 1 kg mass or $g = F/m$ with terms explained] vector $\checkmark$	2
(b) (i) (ii)	$F(=\frac{GMm}{r^2} = \frac{6.67 \times 10^{-11} \times 6.00 \times 10^{24} \times 2.5 \times 10^3}{\left(1.6 \times 10^7\right)^2} \checkmark Q4 Jan 2007$ $= 3900 \text{ N } (3910) \checkmark$ $V_{\text{orbit}} \left( = -\frac{GM}{r} \right) = -\frac{6.67 \times 10^{-11} \times 6.00 \times 10^{24}}{1.6 \times 10^7} \checkmark$ $= -25 (\text{MJ kg}^{-1}) (-25.0) \checkmark$ $[\text{or } \frac{V_{\text{orbit}}}{V_{\text{surface}}} = \frac{r_{\text{surface}}}{r_{\text{orbit}}} \checkmark$ $\text{gives } V_{\text{orbit}} = -\left(\frac{6.4 \times 10^6}{1.6 \times 10^7}\right) \times 63 = -25 (\text{MJ kg}^{-1}) (-25.2) \checkmark$ $\Delta V = (63 - 25) \times 10^6 = 38 \times 10^6 (\text{J kg}^{-1}) \checkmark$ $\Delta E_p (= m \Delta V) = 2.5 \times 10^3 \times 38 \times 10^6 = 9.5 \times 10^{10} \text{J} \checkmark$	max 5
(c)	line starts at ( <i>R</i> , −62.5) and ends at a finite value ✓ curve of decreasing positive gradient ✓ correct (1/ <i>r</i> ) relationship shown by axis values ✓ $0  R  2R  3R  4R  r$ $0  V/MJ kg^{-1}  -16$	3
	Total	10

Question 5		
(a)	$\frac{GMm}{r^2} = m\omega^2 r \left( \text{or} = \frac{mv^2}{r} \right) \checkmark $ Q5 Jan 2008	
	correct application of $T = \frac{2\pi}{\omega}$ (or $v = \frac{2\pi r}{T}$ ) $\checkmark$	3
	(gives $\frac{GMm}{r^2} = \frac{4\pi^2 mr}{T^2}$ and $T^2 = \frac{4\pi^2 r^3}{GM}$ )	
	in which m has cancelled or does not appear in expression ✓	
(b) (i)	$\omega \left( = \frac{2\pi}{T} \right) = \frac{2\pi}{7.15 \times 24 \times 3600} = 1.0(2) \times 10^{-5} \text{ (rad) s}^{-1} \checkmark$	
(ii)	$\omega^2 r (1.02 \times 10^{-5})^2 \times 1.07 \times 10^9 = 0.11(1) \mathrm{m  s^{-2}} \checkmark$	
(iii)	centripetal acceleration = $g$ (or $\alpha = \frac{GM}{r^2}$ ) $\checkmark$	
	$M\left(=\frac{gr^2}{G}\right) = \frac{0.111 \times (1.07 \times 10^9)^2}{6.67 \times 10^{-11}} \checkmark = 1.9(1) \times 10^{27} \text{kg} \checkmark$	max 5
	[or use of $T^2 = \frac{4\pi^2 r^3}{GM}$ with $T = 6.18 \times 10^5 \text{ s}$	
	gives $M = \frac{4\pi^2 r^3}{GT^2} = \frac{4\pi^2 \times (1.07 \times 10^9)^3}{6.67 \times 10^{-11} \times (6.18 \times 10^5)^2} \checkmark$	
	$= 1.9(0) \times 10^{27} \text{kg } \checkmark]$	
	Total	8

Que	stion 3		
(a)	(i)	gravitational force (or field) decreases as $r$ increases $\checkmark$	
		gravitational force (or field strength) is proportional to $(1/r^2)$	
		[award both marks for second statement alone] Q3 Jun 2008	
	(ii)	mass of Moon M $\left(=\frac{Fr^2}{Gm}\right) = \frac{1600 \times (1.75 \times 10^6)^2}{6.67 \times 10^{-11} \times 1000}$	4
		= 7.3(5) × 10 <sup>22</sup> kg ✓	
		[or by use of any other consistent values of F and r]	
(b)	(i)	$E_{\rm P}$ lost = area under graph $\checkmark$	
		acceptable method for finding area and values ✓	
		acceptable value for E <sub>P</sub> lost ✓ [allow (2.8 ± 1.0) × 10 <sup>9</sup> J]	
		[alternative mark scheme, for candidates who use values from the graph:	
		potential of Moon's surface	
		$= \left(-\frac{GM}{r}\right) = -\frac{6.67 \times 10^{-11} \times 7.35 \times 10^{22}}{1.75 \times 10^{6}} = -2.80 \times 10^{6} (\text{J kg}^{-1}) \checkmark$	
		change in potential $\Delta V = (-2.80 \times 10^6) - 0$	
		$= (-)2.80 \times 10^6 (J \text{ kg}^{-1}) \checkmark$	
		potential energy lost (= $m \Delta V$ ) = 1000 × 2.80 × 10 <sup>6</sup>	5
		= 2.80 × 10 <sup>9</sup> J ✓]	
	(ii)	$1/2 mv^2 = 2.8 \times 10^9$ (or the $E_P$ value from (b) (i)) $\checkmark$	
		gives escape speed $v = 2370 \mathrm{ms^{-1}}$ (or a consistent value) $\checkmark$	
		[alternative mark scheme, for candidates who use gravitational potential equation:	
		$1/2 mv^2 = \frac{GMm}{r}$ gives $v = \sqrt{\frac{2GM}{r}}$	
		$= \sqrt{\frac{2 \times 6.67 \times 10^{-11} \times 7.35 \times 10^{22}}{1.75 \times 10^{6}}} \checkmark$	
		= 2370 m s <sup>-1</sup> ✓]	
		Total	9

Que	stion 3		
(a)	(i)	gradient $\left(=\frac{5.9 \times 10^8}{6000}\right) = 9.83 \times 10^4 (\text{m s}^{-2/3}) \checkmark$	
		(for 9.83 allow 9.7 to 10.0)	
	(ii)	cube root of equation is $R = \left(\frac{GM}{4\pi^2}\right)^{1/3} T^{2/3}$	
		(or equation predicts $R \propto T^{2/3}$ ) $\checkmark$	
		$R \propto T^{2/3}$ confirmed by graph as a straight line through $(0, 0)$ (or a line of constant gradient through $(0, 0)$ ) $\checkmark$	6
	(iii)	use of gradient of graph as $\left(\frac{GM}{4\pi^2}\right)^{1/3}$ or $\left(\frac{R}{T^{2/3}}\right)$ ✓	
		$\left(\frac{GM}{4\pi^2}\right)^{1/3} = 9.83 \times 10^4 \text{ gives } \left(\frac{GM}{4\pi^2}\right) = 9.50 \times 10^{14} \text{ (m}^3 \text{ s}^{-2}) \checkmark$	
		mass of Saturn $M = \frac{9.50 \times 10^{14} \times 4\pi^2}{6.67 \times 10^{-11}} = 5.62 \times 10^{26} \text{ kg} \checkmark$	
(b)		similarity:	
		graph would also be a straight line (through $(0, 0)$ because $R \propto T^{2/3}$ (or $R^3 \propto T^2$ ) always applies to any satellite $\checkmark$	2
		difference:	
		gradient would be <i>larger</i> because mass of Sun > mass of Saturn ✓	
		Total	8