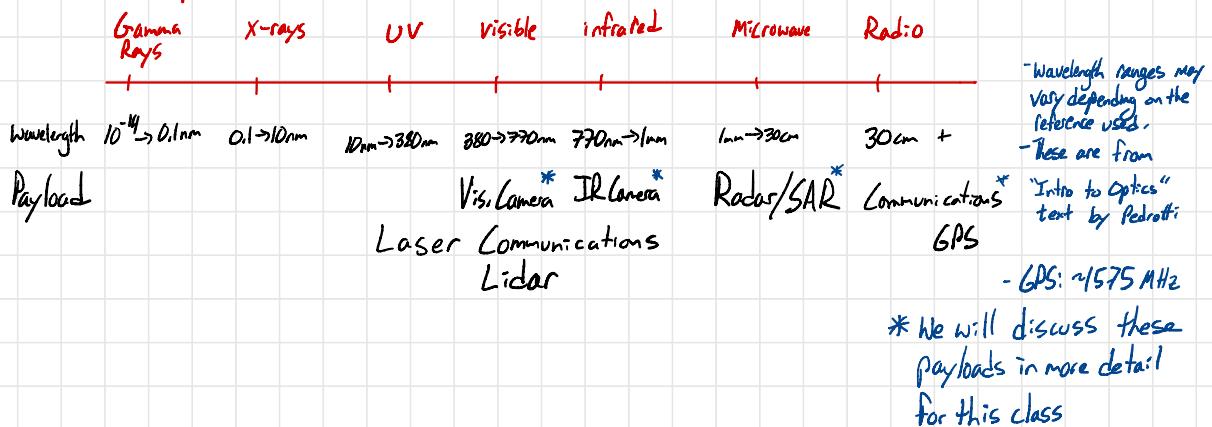


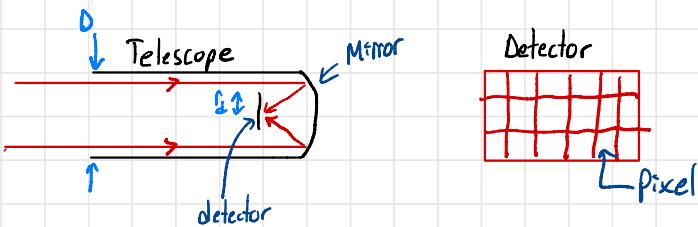
Astro 331X: Payloads

1/

EM Spectrum



Visible Cameras



- Telescope focuses incoming light (photons) onto a detector
- Detector, e.g. CCD, converts photons to electrons, charge builds up on each pixel over exposure time
- After exposure, charge of each pixel is recorded \rightarrow image
- Passive Sensor: relies on reflected EM radiation from another source (e.g. Sun)

Figures of Merit

1) Spatial Resolution: dimension of smallest identifiable object

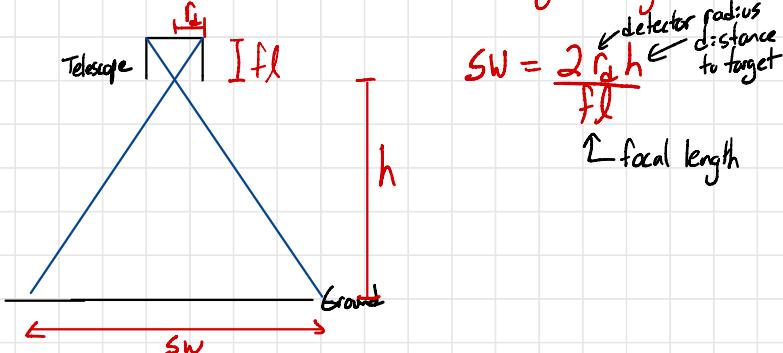
$$\text{Res} = 2.44 \frac{\lambda}{D}$$

Annotations: λ is wavelength, D is telescope aperture diameter, and distance to object.

- fix of telescope, distance and wavelength being sensed.

- Lower Res. is good
- Better w/ smaller λ , smaller D or larger distance

2) Swath Width: Distance across image on the ground



- typically a higher SW is good b/c you can see more.

- SW and Res are competing requirements
 $\rightarrow h \uparrow \Rightarrow SW \text{ better, Res worse}$

Mission Applications

- "Normal" imagery, e.g., google maps!

Satellite Design Considerations

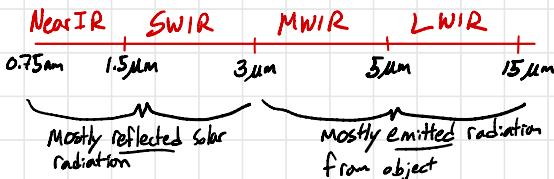
- Orbit/altitude must be chosen to balance Res. and SW requirements
- Rocket size and performance can be limiting factors
 - \rightarrow Optics are heavy!
 - \rightarrow Large telescopes yield good performance, but must fit in rocket fairing and be light enough to reach orbit.

IR Cameras

- Same basic design/ operation as visible cameras
- Passive Sensors
- Same resolution and SW eqns

- Note: IR wavelengths are larger than vis \Rightarrow IR res. will be worse

IR Spectrum



NearIR: reflected light, not heat

Shortwave IR: can travel thru smoke, clouds, etc.

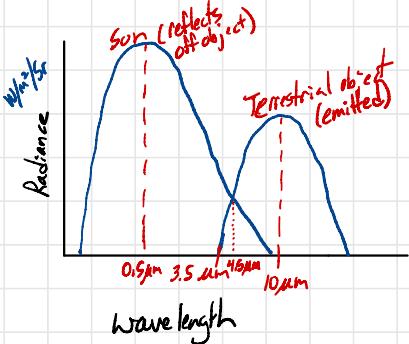
Midwave IR: high heat, rockets, fires, etc.

Longwave IR: less heat, people, ground, etc.

For terrestrial objects (300K) emitted radiation surpasses reflected solar radiation in MWIR

- objects on Earth do emit a little in SWIR, but much more energy is reflected at those wavelengths

Black Body Curves



← plot shows blackbody curves for sun ($T=5800\text{K}$) and an object @ 300K

(Fig 7.1 in Olson)

- So, for SWIR, T_{tgt} must be illuminated.
- Not so for MWIR or LWIR sensing

Mission Applications

1) Thermal Imaging

- View target by the radiation it emits
- Estimate temperature using Stephan-Boltzmann Law

$$S = \epsilon \sigma T^4$$

Estimated emissivity $\downarrow (0.9)$

Total power emitted (area under black body curve) \uparrow

Object temperature \uparrow

Constant $5.67 \times 10^{-8} \frac{\text{W}}{\text{m}^2 \text{K}^4}$

- The sensor can collect total power, or power over several bands, but doesn't necessarily know ϵ of objects. ϵ must be estimated.

Notes:

- Emitted radiation can be determined by the CCD charge ready: more power \rightarrow more photons \rightarrow higher charge

- Collecting energy over a few λ bands allows estimation of the Black-body curve \Rightarrow estimate of temp.

\rightarrow ϵ just moves curve up/down but does not change slope

2) Weather

- Surface temps
- Cloud composition
- Vegetation + soil moisture

→ 2 ways:

- 1) Dry soil heats up more quickly + is hotter
(can calibrate sensor w/ known soil moisture readings)
- 2) Water absorbs certain IR λ . \therefore by looking at reflected light in certain λ , can estimate water content.

3) Missile Warning

→ Good temporal resolution required

Design Considerations

- Materials: Windows/lenses must be transparent to IR
Detectors must absorb IR
 - different materials needed vs. visible
 - Cooling: required for 2 reasons
 - 1) Reduce Unwanted emissions (warm objects emit IR light)
 - ensure camera does not record emissions from optical components
 - 2.) Reduce noise in detector/improve sensitivity
 - Thermal vibrations in detector material releases electrons ("dark current")
 - Adds noise that could mask desired signal.

Radar / Synthetic Aperture Radar (SAR)

Radar Basics

Active Sensor: Microwave EM radiation is transmitted, hits a target and returns

Microwave radiation is produced by a magnetron.

Challenge

$$\text{Res} = \frac{2.44}{D} h$$

- One of the challenges w/ Radar is related to forces

2) Microwave 10^{-3}
Optical 10^{-7}

Relatively large $\lambda \Rightarrow$ poor resolution

\rightarrow need large operator (D)
- real or synthetic

- real or synthetic

→ need different principle for image generation

Figures of Merit

$$\text{Radar range: } R = \sqrt{\frac{P_t G^2 \lambda^2 \sigma}{P_r (4\pi)^3}}$$

P_t : transmit power (W)

P_r : received power (W)

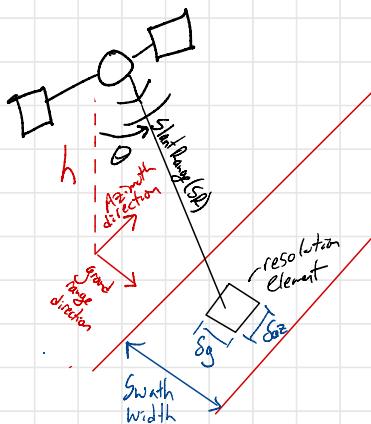
λ : wavelength (m)

σ : radar cross section (m^2)

G : receiver/transmitter gain

- use to calc. range to target given P_r , or calc. max possible range given min detectable P_r

RCS: Area which, if radiating the incident power isotropically, would produce the same received power in the radar.
Depends on Target size, shape, orientation, material and Radar wavelength
← will discuss gain later with comm

Resolutions:

Note: $SR = \frac{h}{\cos \theta}$
 θ : look angle

Radar "pulse width" T (sec)

This is the length of time the radar transmits EM radiation before stopping to listen.

Slant range resolution: along radar Line of sight

$$SR = \frac{cT}{2}$$

- we care more about ground range res. which accounts for look angle θ

Ground range resolution: 1 to ground track

$$Sg = \frac{cT}{2 \sin \theta}$$

Notes:

- both SR and Sg are indep. of range \rightarrow

- Ground range res actually improves as range \uparrow since

$$\theta, \sin \theta \downarrow$$

- these eqns imply you want T as small as possible. True, but then you have less power on target \Rightarrow less chance of detection.

Both independent of range.

Azimuth Resolution: // to ground track

Real Radar (single pulse):

$$S_{az} = \frac{h \lambda}{\cos \theta D_{az}}$$

D_{az} : Diameter of antenna in direction of travel

Ex: $h = 300 \text{ km}$

$$\lambda = 0.015 \text{ m} \Rightarrow S_{az} = 1732 \text{ m} \approx$$

$$D_{az} = 3 \text{ m}$$

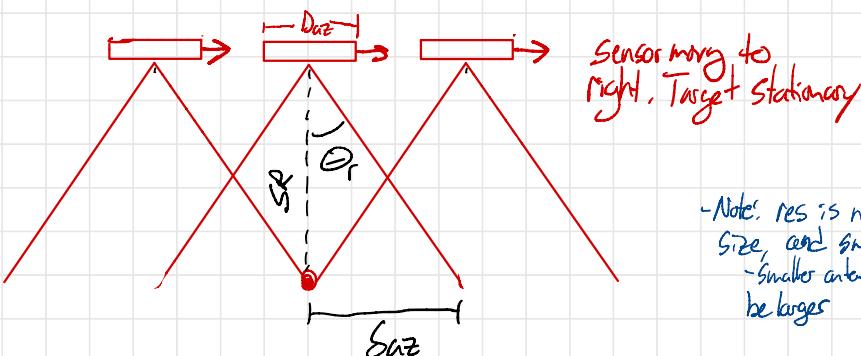
$$\Theta = 30^\circ$$

Dependant on range(h)! \Rightarrow poor resolution.

Solution: Synthetic Aperture Radar

\rightarrow Radar returns are combined while target is in view, forming a large Synthetic Aperture

Synthetic Aperture Length (D'_{az})



$$D'_{az} = 2S_{az} = 2(SR \frac{\lambda}{D_{az}})$$

$$\frac{S_{az}}{SAR} = SR \frac{\lambda}{D'_{az}} = \frac{SR \lambda}{2SR \frac{\lambda}{D_{az}}} \Rightarrow$$

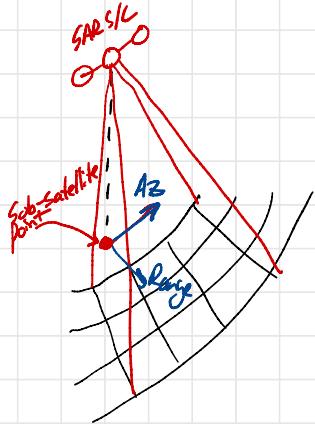
$$\boxed{\frac{S_{az}}{SAR} = \frac{D_{az}}{2}}$$

- Note: res is now only a function of antenna size, and smaller is better! Why?
- Smaller antenna, larger beam, D_{az} can be larger

SAR Az. Res. is only a function of antenna size!
Smaller D_{az} is better.
Why? Smaller antenna \rightarrow wider beam \rightarrow D_{az} is larger

SAR Image Generation: Basic Ideas

To generate an image the system must map each return to a location in both range + azimuth directions



- here we'll just try to describe the basic ideas used to generate images.
Reality it's quite a bit more complicated

Range: Determined by time from pulse transmission to return receive

Azimuth: Determined by doppler shift in return.
 → returns from targets "in front" have higher freq (pos. doppler shift)
 → returns from targets "behind" have lower freq (neg. doppler shift)

- Target location in Range/Az can be determined for each pulse
- Multiple pulse returns can be combined + Range/Az map shifted based on doppler history to generate an image.

Mission Applications

- All Weather, day/night imaging

Design Considerations

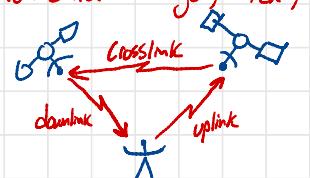
- Active System → higher power requirements compared to optical/IR
- Lots of processing required for SAR image generation

RF Communications

Transmit and/or receive data:

Satellite data: telemetry (temp, voltage, position, etc)

Mission data: images, text, notifications, etc.



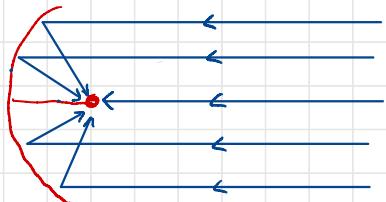
- We'll focus on comm systems using RF (larger wavelengths), but Comm using lasers is also possible, and is becoming more common.

Uses an Antenna: Any device that converts electronic signals to EM waves or vice versa

[] []
Transmit antenna Receive Antenna

- electronic signals are, e.g. Signals in wires or cables.

Antenna Gain : Dish collects more energy and focuses



Paraboloid Antenna Max Gain:

$$G = \frac{4\pi A}{\lambda^2} \epsilon$$

A: cross section of dish
 $A = \pi r^2$

ϵ : efficiency (%: 20% $\rightarrow \epsilon = 0.2$)

\rightarrow in $\epsilon \rightarrow \epsilon$ gain depends on signal!

ϵ : fxn of materials and/or manufacturer

- Gain changes based on direction

Antenna Beam Pattern:



- We saw that SNR drops w/R. If the antenna has a gain > 1 , then it does not transmit or receive energy in all directions equally

For best performance, want

Transmit antenna oriented so that receiver is in main beam

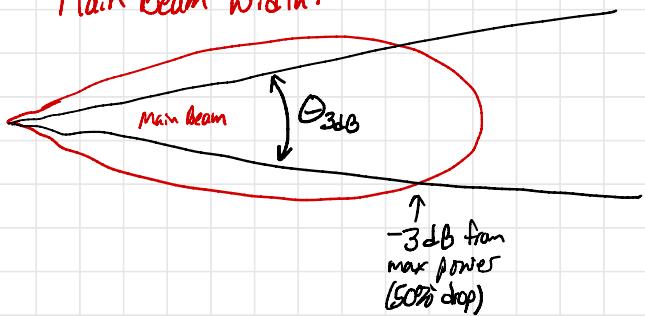
Receive antenna oriented so that transmitter of interest is in main beam

Astro 331X : Payloads

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- Typically when we are designing an antenna, or planning how to use it, we care about the main beam. We'll generally point our antenna so that the target is in the main beam b/c it's easiest to collect that way.

Main Beam Width:



$$\text{General: } \Theta_{3\text{dB}} = \beta \frac{\lambda}{D} \text{ (radians)}$$

λ : wavelength (m)

D : Antenna dimension in same plane as beam pattern (m)

β : Beamwidth fudge factor (unitless)
typically $0.5 \rightarrow 2.0$

Figure of Merit: Signal to Noise ratio (SNR)
Typically want > 1

$$\text{SNR} = \frac{P_t G_t \left(\frac{\lambda}{4\pi S} \right)^2 G_r}{K T_r B} = \frac{\text{Signal Power}}{\text{Noise Power}}$$

Annotations for the SNR equation:

- Transmit power (P_t)
- Transmit gain (G_t)
- Wavelength (λ)
- Range
- receivers gain (G_r)
- $K = 1.38 \times 10^{-23} \text{ J/K}$
- receiver temp. (T_r)
- receiver bandwidth (B)

If listening, can control: G_r, T_r, S

← what can be controlled if transmitting?

- discuss min/max range
 $R_{\text{max}} = \sqrt{(R_{\text{oth}})^2 - R_0^2}$