

A High-order Conservative Eulerian Simulation Method for Vortex Dominated Flows

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A high-order, conservative Eulerian method is presented for the simulation of vortex dominated inviscid fluid flows. The primitive variable incompressible Euler equations are recast in the velocity-vorticity form to explicitly enforce conservation of vorticity. The advection of the vorticity is then calculated via a two-step process: the velocity field is determined by evaluation of the Biot-Savart integral, and then a line-based discontinuous Galerkin (DG) Eulerian spatial discretization scheme is applied to accurately advect the vorticity field. The accuracy and convergence of this method was examined for test cases where an analytical solution exists, as well as more challenging test cases which lack an analytical solution. The convergence rate behavior is chiefly controlled by the error in the calculated velocity field. Velocity errors are due to two factors: the approximation of the Biot-Savart integral with a desingularized form, and poor quadrature convergence for the nearly singular integral. Solver parameters were chosen to balance these two effects resulting in nearly optimal convergence of the overall method in the analytical test, and high-order convergence in the qualitative test case.

I. Introduction

Direct solution of Navier-Stokes is impractical for many fluid problems and where possible simplifications should be made; one such possibility is for vortex dominated flows. It is possible to recast Navier-Stokes from a primitive variable form (u, v, p) to a velocity-vorticity (u, v, ω) form. This has several advantages: explicit conservation of vorticity, elimination of pressure terms (for incompressible flows), and reduction of required simulated degrees of freedom to just those that form the vorticity support.

Many vortex based simulation techniques use a Lagrangian approach to discretize the vorticity into a set of vortex particles,⁴ lines,⁵ sheets,⁶ or volumes.⁷ This is a natural approach to take given the material nature of the vorticity due to Helmholtz and Kelvin's theorems.

The velocity field due to the vorticity is calculated from solving the Poisson problem,²⁵ or through inversion of the Poisson problem via evaluation of the Biot-Savart integral.¹ There are several challenges with evaluation of the velocity: boundary conditions in the Poisson solution method, and singularities in the Biot-Savart volume integral method. Typically Lagrangian point vortex codes must de-singularize the kernel.^{2,3} Direct summation for the velocity field has a $\mathcal{O}(N^2)$ complexity. Tree-codes⁹ and Fast Multipole Methods (FMM)⁸ are ways of achieving a more efficient calculation.

Lagrangian point methods present several problems. Point disorganization can occur as the fluid evolves, this typically requires temporary meshing to recondition the discretization. This has been handled with various methods; recalculation of the quadrature weights at each time step,^{11,12} regridding/rezoning,¹³ and remeshing¹⁴ among them. Additionally, most Lagrangian approaches are limited to low order; careful point locations must be chosen and maintained through frequent remeshing etc. in order to achieve high-order convergence.⁸

In comparison, Eulerian approaches to vortex methods tend to be less common. Work has been done for viscous flows with solid bodies; both stationary²⁷ and moving.²⁶ Both of these solve the streamfunction-vorticity formulation, not the velocity-vorticity formulation. The velocity-vorticity formulation has been solved by an Eulerian approach by others.²⁸ A notable example of an Eulerian approach to the velocity-vorticity formulation is the work of Brown et al.¹⁵ who adapted the velocity-vorticity approach to a low order

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Finite Volume (FV) solver. Later they were able to extend their method to be accelerated via a FMM.¹⁶ However, like most FV approaches, to extend to high order solution approximations requires extended stencils that ultimately limit the geometrical freedom of the mesh. Steinhoff et al. also used an Eulerian approach, but solved a modified form of the inviscid Euler equations with “vorticity confinement” rather than the velocity-vorticity equation.⁷

The desire to resolve fine vortical structures motivates the need for a high order solver. Considering the challenges associated with achieving a high order Lagrangian method and the success of Brown et al.’s low order method, a high order Eulerian vorticity-velocity method would seem to be a possible choice. This leaves the choice of a spatial discretization.

Finite difference methods suffer from similar problems as FV with extended stencils, as well as not being explicitly conservative. A finite element approach is ill-suited to the hyperbolic nature of vorticity advection. Spectral methods are promising, but for sparse vorticity domains are less-efficient due to the global support of the harmonic bases. However, discontinuous Galerkin (DG) methods¹⁷ are a natural choice; they are conservative, able to take advantage of vorticity sparseness, are well-suited to handle advection via intelligent choice of a numerical flux function, and have bases with local/compact support.

For domains free of impinging bodies a hexahedral mesh with a tensor product grid of interpolation points is convenient to implement. This permits the use of a line-DG²⁴ approach that considerably simplifies multi-dimensional cases by allowing reuse of 1D methods. Choosing a 2D domain permits evaluation of whether the method is practical, as well as to limit solution times to those that are reasonable on a workstation. It also has the advantage of removing the vortex stretching term and reducing the vorticity to a scalar. The resultant partial differential equation (PDE) takes on the familiar form of a scalar conservation law. To maintain maximum flexibility for investigative purposes and to remove as much approximation error as possible the velocity field for validation of the underlying method is calculated via direct evaluation of the Biot-Savart integral.

II. Governing Equations and Discretization

A. Velocity-Vorticity Formulation

The Navier-Stokes momentum equation is

$$\rho \left(\frac{\partial \mathbf{u}}{\partial t} + \mathbf{u} \cdot \nabla \mathbf{u} \right) = -\nabla p + \mu \nabla^2 \mathbf{u} + \frac{1}{3} \mu \nabla (\nabla \cdot \mathbf{u}) \quad (1)$$

where u is the velocity field, p is the pressure, and ρ is the density. If we restrict ourselves to incompressible flows and define the quantity *vorticity* as

$$\omega = \nabla \times \mathbf{u} \quad (2)$$

Then the traditional form of the Navier-Stokes equations can be recast

$$\frac{\partial \omega}{\partial t} + \mathbf{u} \cdot \nabla \omega - \omega \cdot \nabla \mathbf{u} = S(x, t) \quad (3)$$

where we collect viscous generation of vorticity in S .

For 2-D distributions of vorticity, several simplifications can be made. The originally vectorial vorticity becomes a scalar quantity, all vorticity is directed normal to the plane. As a result, the vortex stretching term in Eqn. (3) becomes zero. The only non-zero component of ω is in the z-direction, however the gradient of the velocity field is zero in the z-direction, so the product is therefore zero. The result is

$$\frac{\partial \omega}{\partial t} + u \cdot \nabla \omega = S(x, t) \quad (4)$$

or if instead the second term is expressed in terms of the flux of the vorticity (where $f_i(\omega) = u_i \omega$):

$$\frac{\partial \omega}{\partial t} + \frac{\partial f}{\partial x_i} = S(x, t) \quad (5)$$

For an incompressible 2D or 3D flow we can relate the velocity and vorticity by:

$$\nabla^2 \mathbf{u} = -\nabla \times \omega \quad (6)$$

If inverted, the Biot-Savart integral is obtained

$$u(x^*) = \int_{\Omega} K(x^*, x) \times \omega(x) dx \quad (7)$$

where x^* is the point we wish to evaluate the velocity, x is the coordinate in regions of non-zero vorticity, and $K(x^*, x)$ is the singular Biot-Savart kernel⁷

$$K(x^*, x) = \frac{-1}{2\pi} \frac{x^* - x}{|x^* - x|^2} \quad (8)$$

The high-order Eulerian approach taken here means that while the Biot-Savart integral converges, a singularity is always present within any of the extended vorticity patches thanks to self-influence. Conceptually this doesn't present an impossible problem, the integral *does* converge in an analytical sense; practically speaking however a singularity may cause numerical integration procedures to diverge, or at the very least converge quite slowly. The approach taken by Brown¹⁶ was to use the Rosenhead-Moore kernel, choosing a core size such that the maximum velocity occurred on the face of the finite volume unit. This can be constructed *a priori* because the vorticity is taken as constant across the volume, as is typical in a FV approach. If the vorticity is spatially varying however, the choice of core size is more troublesome.

One may attempt to desingularize the Biot-Savart kernel by introducing a core function $\eta(z/\delta)$, with characteristic cutoff radius δ . Traditionally the core function is convolved with the Biot-Savart kernel to yield a desingularized kernel K_{δ} .

$$K * \eta(r) = K_{\delta} \quad (9)$$

The choice of a core function and a characteristic radius has important implications on the accuracy and convergence of a Lagrangian point vortex method. Choosing a cutoff radius too small and there is insufficient smoothing, too large and the vorticity discretization is spatially smeared. In a Lagrangian method the convolution with a core function has a physical heuristic: the point vortex is replaced by a finite size vortex blob described by $\eta(r)$, with characteristic radius δ .

In the present method the core function is a means to an end; attempting to numerically integrate the singular Biot-Savart integral will result in spurious values, so the desingularized kernel is used as an approximation. A number of kernels can be chosen in a Lagrangian method, we found the following kernel¹⁰ generally had the lowest approximation error:

$$K_{spectral} = \frac{z}{2\pi|z|^2} (1 - J_0(\frac{z}{\delta})) \quad (10)$$

where we have substituted $z = x^* - x$, and J_{α} is a Bessel function of the first kind.

B. Discontinuous Galerkin Spatial Discretization

In order to solve Eqn. (4) we adopt a method-of-lines approach.¹⁸ We will first spatially discretize the system to obtain the semi-discrete system, then we use an explicit time discretization method to march forward in time. Note that Eqn. (4) has the form of a scalar conservation law, with ω being the conserved quantity. The velocity field that advects the conserved quantity has been calculated by evaluation of the Biot-Savart integral for the current timestep.

We use a standard nodal DG formulation¹⁷ for the 1D problem, resulting in the variational form defined on a particular element:

$$\int_{\Omega} \frac{\partial \tilde{\omega}}{\partial t} \phi_j dx + \int_{\Omega} \frac{\partial f(\tilde{\omega})}{\partial x} \phi_j dx = 0 \quad \text{for all } j \quad (11)$$

where $\tilde{\omega}$ is the solution approximation for the vorticity, and ϕ_j is the j^{th} test function. Both the solution approximation bases and the test functions are apart of the same polynomial vector space.

We take the set of Lagrange polynomials as the basis for our polynomial vector space. The interpolatory property of the Lagrange basis means that $\ell_i(x_j) = \delta_{ij}$, so that the value of the function at the interpolation points forms the expansion coefficients. Therefore the local Mth order approximation to vorticity takes the form:

$$\omega(x, t) \approx \tilde{\omega}(x, t) = \sum_{i=0}^M a_i(t) \psi_i(x) \quad (12)$$

where a_i is the value of the vorticity interpolation at the i^{th} node, and ψ_i is the i^{th} solution approximation basis.

The spatial discretization thus described only concerns the solution within an element. Continuity across elements is not enforced in DG, the solution is multiply defined at coincident nodes from neighboring elements. To recover the global solution, flux functions are usually employed to couple local elemental solutions to one another. We use an upwind flux, where defining the average as $\{\{\omega^+\}\} = \frac{\omega^+ + \omega^-}{2}$ and the jump as $[[\omega]] = \omega^+ - \omega^-$,¹⁷ yields

$$\hat{f}_{upwind}(x^+, x^-) = u\{\{\tilde{\omega}\}\} + \frac{|u|}{2}[[\tilde{\omega}]] \quad (13)$$

Continuing to follow a standard nodal DG approach and substitution of these choices of bases functions and flux functions yields the elemental form:

$$\frac{\Delta x}{2} \sum_{i=0}^M \left[\frac{da_i}{dt} \int_{-1}^1 \psi_i \phi_j dX \right] + \hat{f} \phi_j \Big|_{x_L}^{x_R} - \int_{-1}^1 f(\tilde{\omega}) \frac{d\phi_j}{dX} dX = 0 \quad (14)$$

where x_R and x_L are the element bounds, we have mapped to a computational element $X \in [-1, 1]$ via the mapping $X = G(x) = \frac{2(x-x_L)}{\Delta x} + 1$, and $\Delta x = x_R - x_L$ is the element size.

Thus far only a 1D solution has been discussed. The 2D solution basis is formed from the tensor product of two orthogonal 1D solutions, each along a coordinate axis. An important consequence of the tensor basis and hexahedral mesh is that the two basis directions are not coupled except at the interpolation nodes at the intersection of each direction's basis. This is the central idea in the line-DG²⁴ approach, a 2D problem is transformed into two 1D problems of the form of Eqn. (14). The 2D time evolution of the overall system is the linear combination of the 1D evolutions; notably the instantaneous rate of change of a particular node is the sum of the rates from each direction.

We can now use the developed 1D basis in a tensor product to form our 2D basis

$$f(x, y) \approx \left[\sum_{t=0}^M z_t \ell_t(y) \right] \times \left[\sum_{s=0}^M z_s \ell_s(x) \right] = \sum_{t=0}^M \sum_{s=0}^M z_{st} \ell_t \ell_s = \sum_{t=0}^M z_{st} \ell_t \sum_{s=0}^M \ell_s \quad (15)$$

where s and t are node numbers along the x and y directions respectively.

We now substitute our basis into Eqn. (14) to get a particular direction's PDE

$$\frac{\Delta x}{2} \sum_{i=0}^M \left[\frac{dz_{ij}}{dt} \int_{-1}^1 \ell_i \ell_j dX \right] + \hat{f} \ell_j \Big|_{x_L}^{x_R} - \int_{-1}^1 f(\tilde{\omega}) \ell_j' dX = 0 \quad (16)$$

We solve the PDE "line-wise"; we form the tensor product of the chosen 1D interpolation nodes along each direction (in 2D there are x-line bases and y-line bases). The rate of change at a particular node x_{st} is:

$$\frac{\partial \omega_{st}}{\partial t} = \left(\frac{\partial \omega_{st}}{\partial t} \right)_{x-line} + \left(\frac{\partial \omega_{st}}{\partial t} \right)_{y-line} \quad (17)$$

III. Implementation

A. Solver Overview

```

Define problem parameters
Define solver parameters
Calculate derived solver parameters
Setup initial conditions
Initialize solver
%Time stepping
for t=0 to end
    if datalog?=yes
        save system state to file and plot
    end

```

```

%Loop through RK stages
for s=1 to last_stage
    %For elements above threshold
    for each vorticity source
        calculate velocity contributions
    end
    %Calculate semi-discrete system terms
    interpolate element_boundary_vorticity
    calculate numerical_fluxes
    calculate total_surface_flux
    calculate internal_stiffness_flux

    vorticity_rate_of_change = ...
        internal_stiffness_flux - total_surface_flux

    RK_stage = (RK_coeff_a * RK_stage) + ...
        (time_step * vorticity_rate_of_change)
    vorticity = vorticity + RK_coeff_b * RK_stage
end
end

```

B. Vortex Dominated Flow: Diagnostics

Two diagnostics are useful to evaluate the evolution of vortex dominated flows. The first is the vorticity moments of the system, which should be conserved,²³ and are given by:

$$J_{mn} = \int \int \omega(x, y) x^m y^n dx dy \quad (18)$$

We shall consider in particular the linear impulse, J_{01} and J_{10} .

The other diagnostic considered is the effective aspect ratio.²³ This is particularly useful when considering the evolution of an elliptical vortex, which will typically undergo some degree of axisymmetrization. The effective aspect ratio is:

$$\lambda_{eff}^2 = \frac{J + R}{J - R} \quad (19)$$

where $J = J_{20} + J_{02}$, $D = J_{20} - J_{02}$, and $R^2 = D^2 + 4J_{11}^2$.

C. Discrete L^2 error

It is possible to calculate the L^2 error of the solution approximation v compared to an analytical solution or some other approximation, u via:

$$||u - v||_2 = \sqrt{\int (u - v)^2 dx} \quad (20)$$

A discrete version needs to perform numerical quadrature to calculate the integral of the error. In situations where the calculated error is obtainable at the solution points, we have ready-made quadrature points that we can collocate such that we obtain the discrete form:

$$||u - v||_2 = \sqrt{[w_i]^T (u_{ij} - v_{ij})^2 [w_j]} \quad (21)$$

where w_i and w_j are the appropriate quadrature weights associated with each direction of the tensor product of quadrature points, and u_{ij} and v_{ij} are the values at the interpolation nodes. If two tests do not share a common set of solution points, we interpolate the results to a shared set of equispaced points and calculate the error by:

$$||u - v|| = \sqrt{\frac{1}{M \times N} \sum_{i=1}^M \sum_{j=1}^N (u_{ij} - v_{ij})^2} \quad (22)$$

IV. Results

In this section we will introduce some figures and tables. It can be seen in figure 1 that magnetization is a function of applied field. Sometimes writing meaningless text can be quiet easy, but other times one is hard

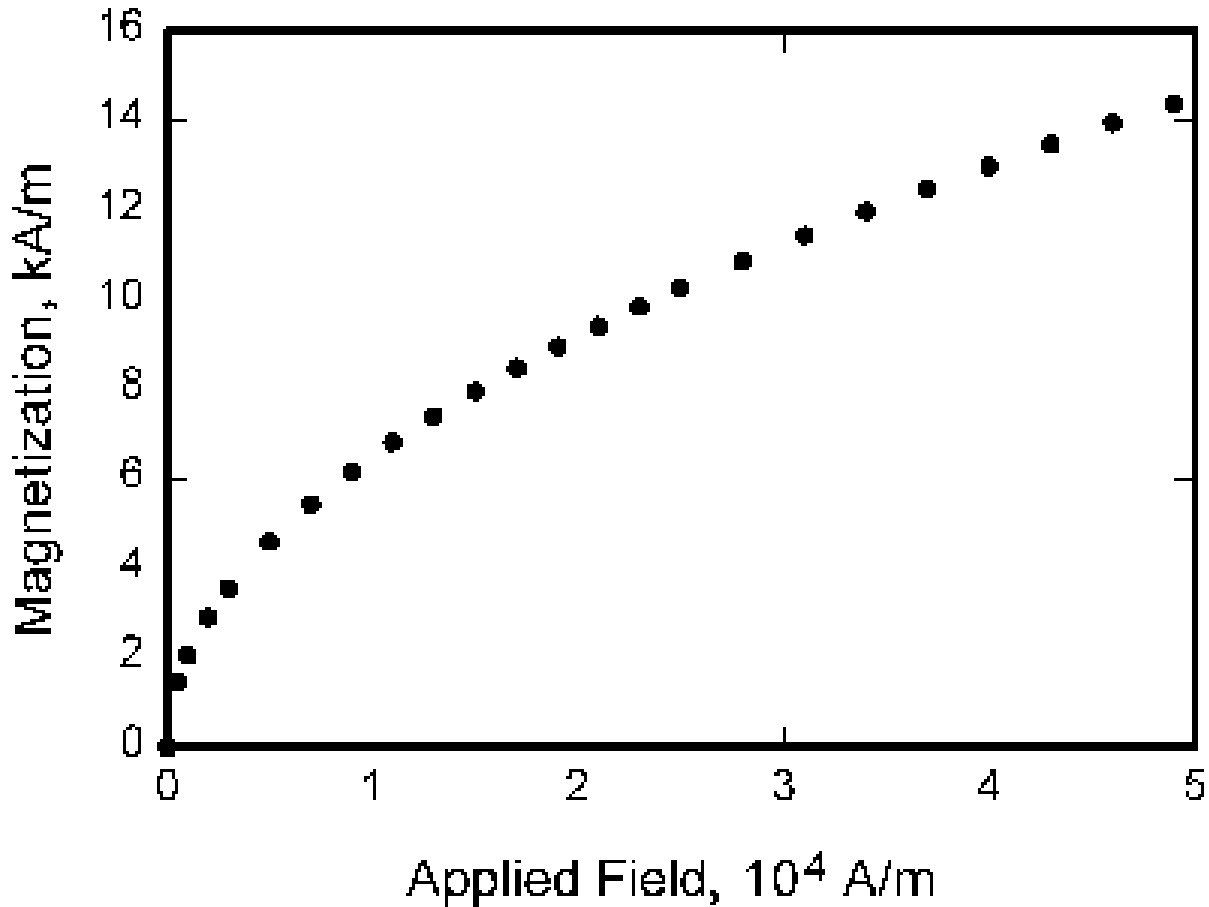


Figure 1. Magnetization as a function of applied field, which has borders so thick that they overwhelm the data and for some reason the ordinate label is rotated 90 degrees to make it difficult to read. This figure also demonstrates the dangers of using a bitmap as opposed to a vector image.

pressed to keep the words flowing.^a Meanwhile back in the other world, table 1 shows a nifty comparison.

V. Validation

VI. Conclusion

After much typing, the paper can now conclude. Four rocks were next to the channel. This caused a few standing waves during the rip that one could ride on the way in or jump on the way out.

Acknowledgments

A place to recognize others.

^aAnd sometimes things get carried away in endless detail.

Table 1. Variable and Fixed Coefficient Runge-Kutta Schemes as a Function of Reynolds Number

Re	Vary	Fixed
1	868	4,271
10	422	2,736
25	252	1,374
50	151	736
100	110	387
500	85	136
1,000	77	117
5,000	81	98
10,000	82	99

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