

$$y(t) = \left(\frac{\beta_1}{\beta_0}\right) \dot{y}_1(t) + y_1(t)$$

or equivalently:

$$y(t) = y_1(t) - \left(\frac{1}{z_1}\right) \dot{y}_1(t) \quad (z_1 = -\beta_0/\beta_1)$$

Where $y_1(t)$ is the "ideal" (no zero) step response

The total response $y(t)$ is the sum of the ideal response, and a fraction of the derivative of this response.

Suppose 1st $z_1 < 0$ (LHP zero)

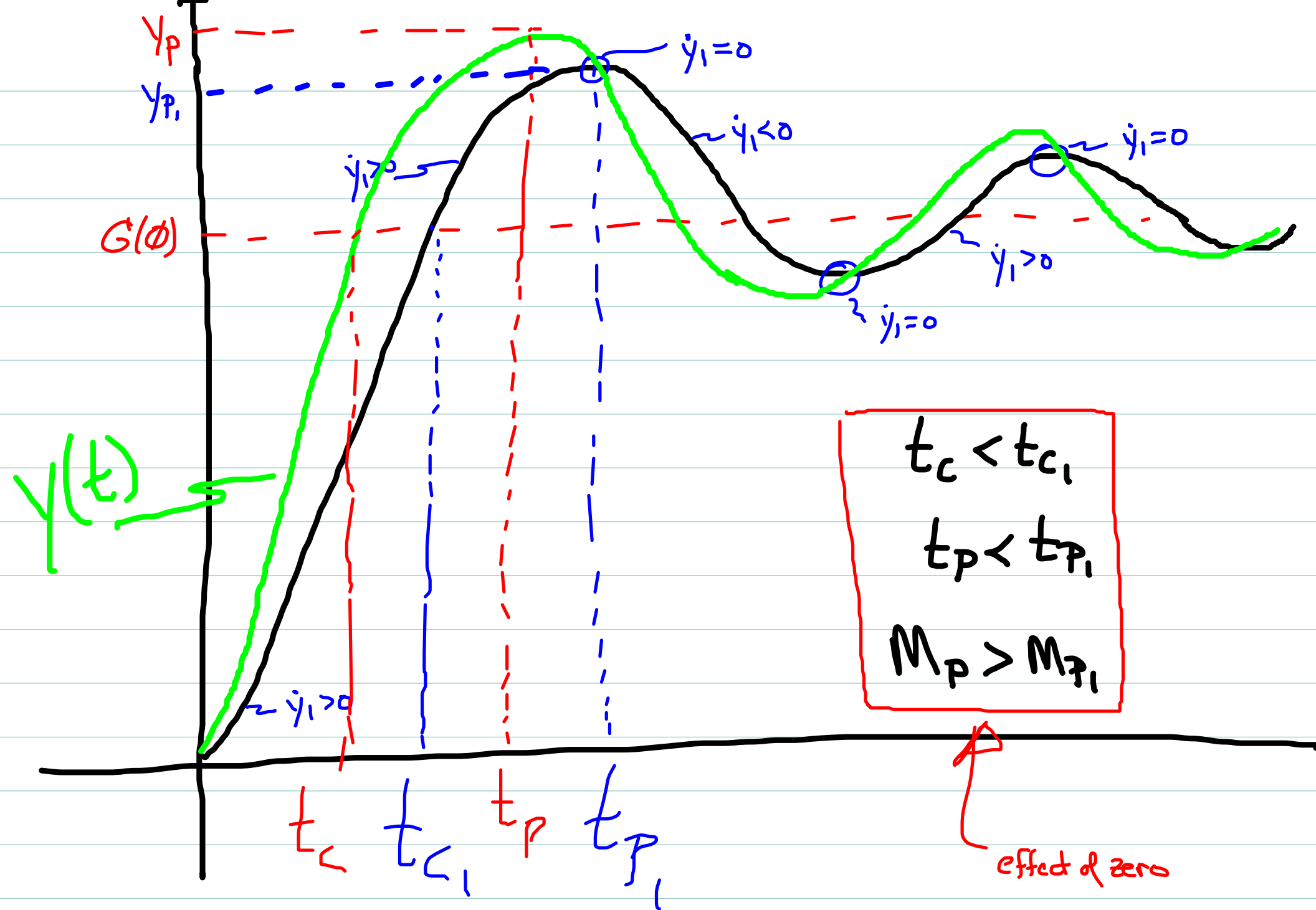
then $z_1 < 0$ and $(-\frac{1}{z_1}) > 0$ so we can write

$$y(t) = y_1(t) + \left(\frac{1}{|z_1|}\right) \dot{y}_1(t)$$

Derivative adds to total response. To understand effect of this, must examine behavior of $\dot{y}_1(t)$

Note that $\dot{y}_1(t) \rightarrow 0$ as $t \rightarrow \infty$, so the steady-state of the new response will be the same as the ideal response

$$y_{ss} = G(0)$$



Summary of observations

A LHP zero changes a 2^{nd} order step response by:

\Rightarrow Increasing overshoot y_p and M_p

\Rightarrow decreasing t_c and t_p

In a sense, system "responds" faster (crosses y_{ss} more quickly), but price is greater overshoot.

\Rightarrow Note: tricky to quantify exact changes to t_c, t_p, y_p based on z_1

\Rightarrow However, note change from "ideal" response is proportional to $\frac{1}{|z_1|}$

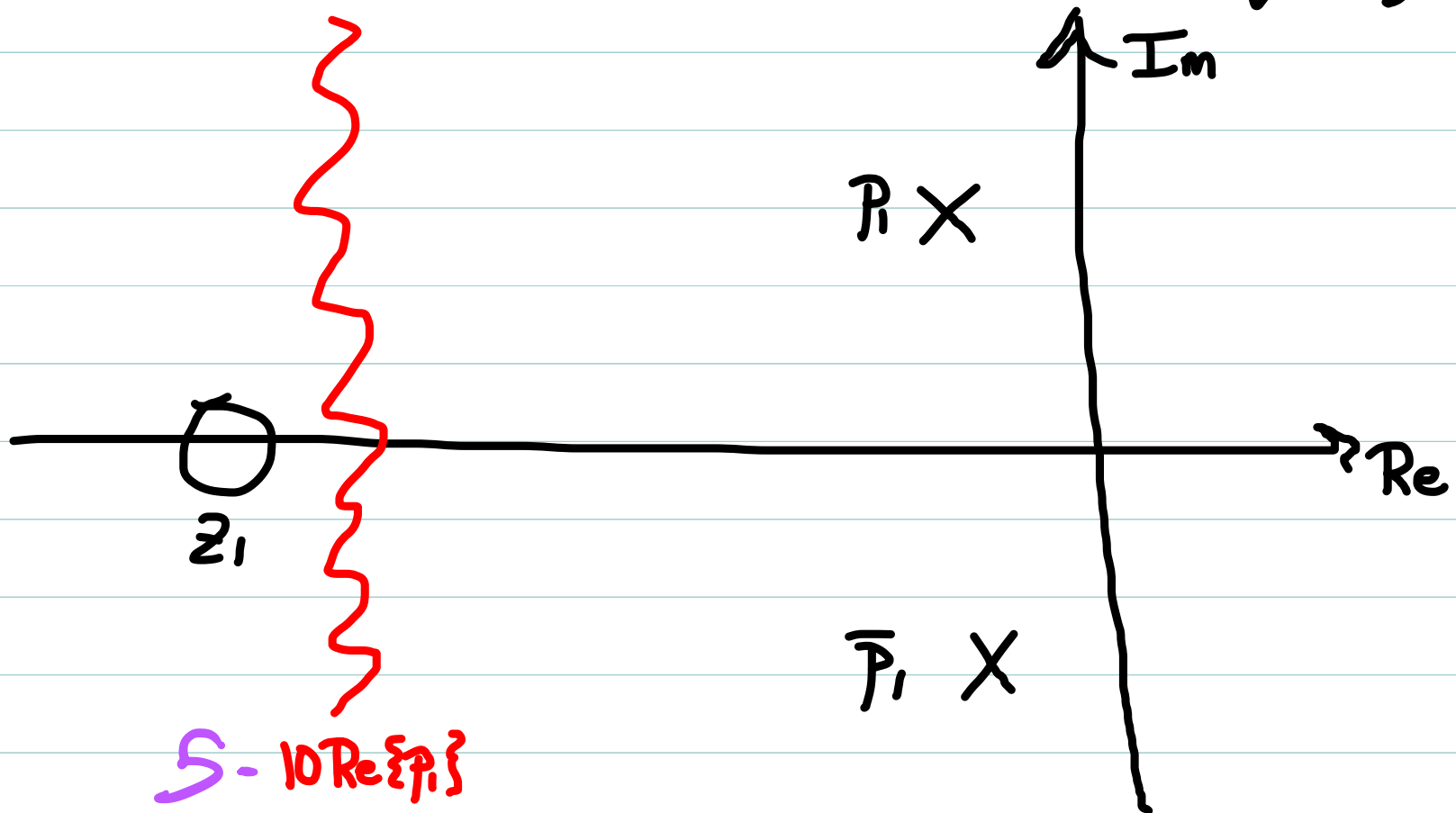
\Rightarrow The further z_1 is from imag Axis, the smaller the effect

Rule of Thumb

Effect of zero in this case is negligible if

$$|z_1| > \cancel{10} | \operatorname{Re}\{p_1\} |$$

i.e. zero is 10 times further into LHP than complex poles.



Question

\Rightarrow A zero increases (amplifies) the overshoot of a 2nd order system with $\xi < 1$ (complex poles).

\Rightarrow Can it actually create overshoot in a system with 2 real poles ($\xi \geq 1$)?

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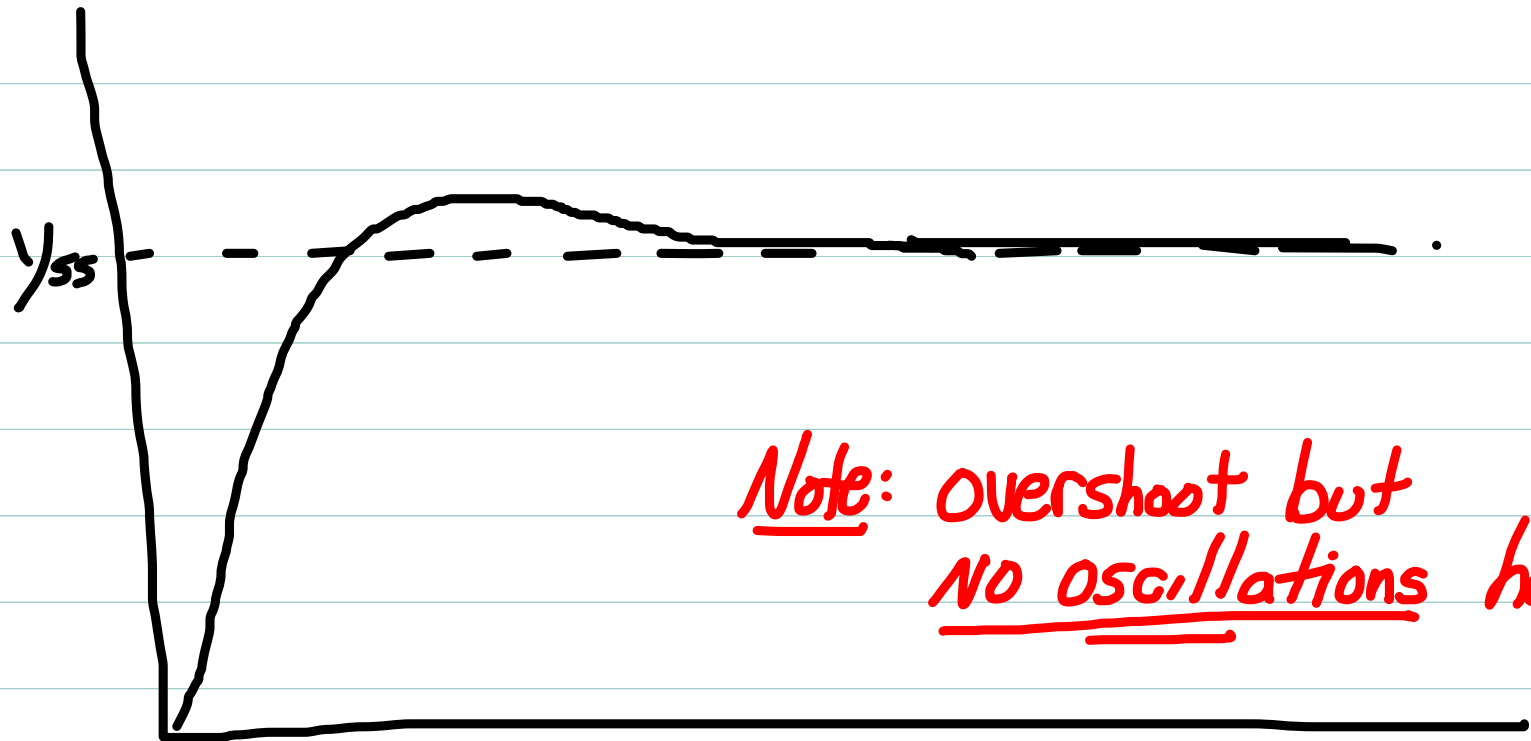
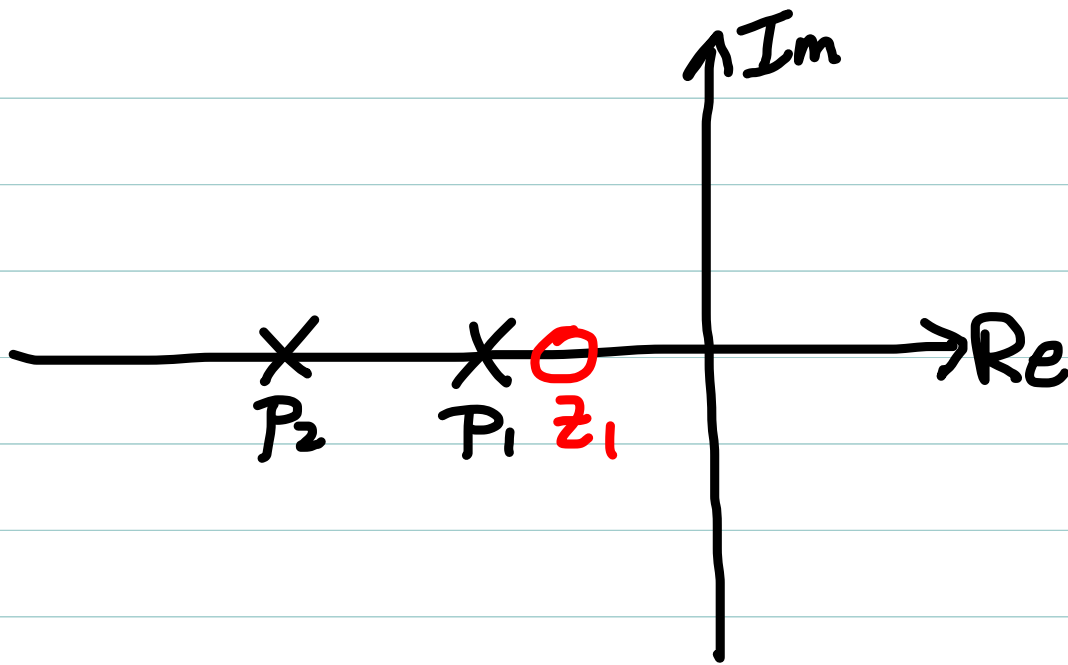
\Rightarrow Can it actually create overshoot in a system with 2 real poles ($\xi \geq 1$)?

\Rightarrow **Yes!**

\Rightarrow With 2 real poles p_1 and p_2 , $y_p > y_{ss}$ if

$$|z| < \min(|p_1|, |p_2|)$$

i.e. if zero is closer to imag axis than ^{both} ~~either~~ of the two poles.



Note: overshoot but
no oscillations here

Back to 2nd order ($\zeta < 1$ case)

Suppose $z_1 > 0$, i.e. z_1 in RHP, then

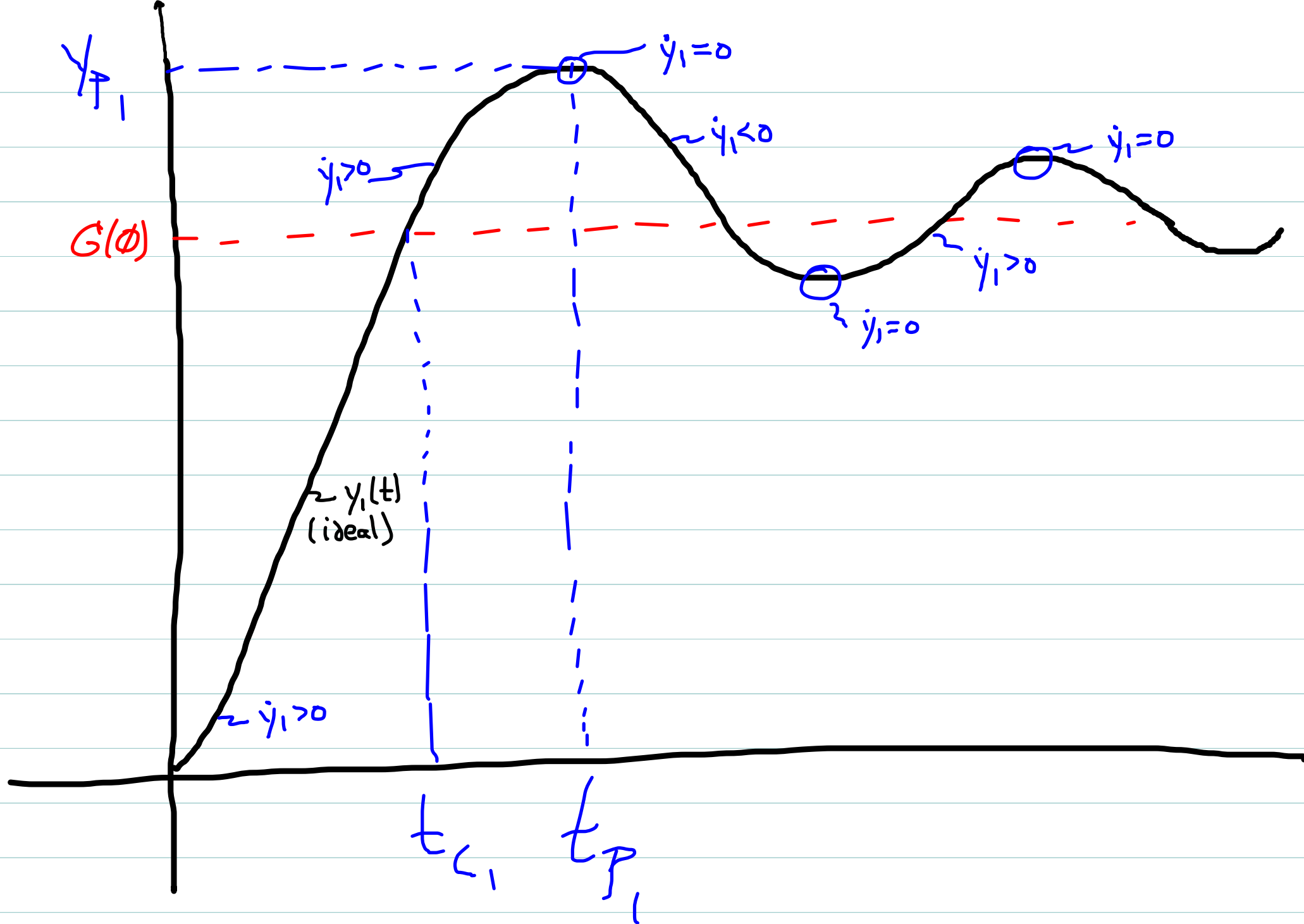
$$y(t) = y_1(t) - \left(\frac{1}{z_1}\right) \dot{y}_1(t)$$

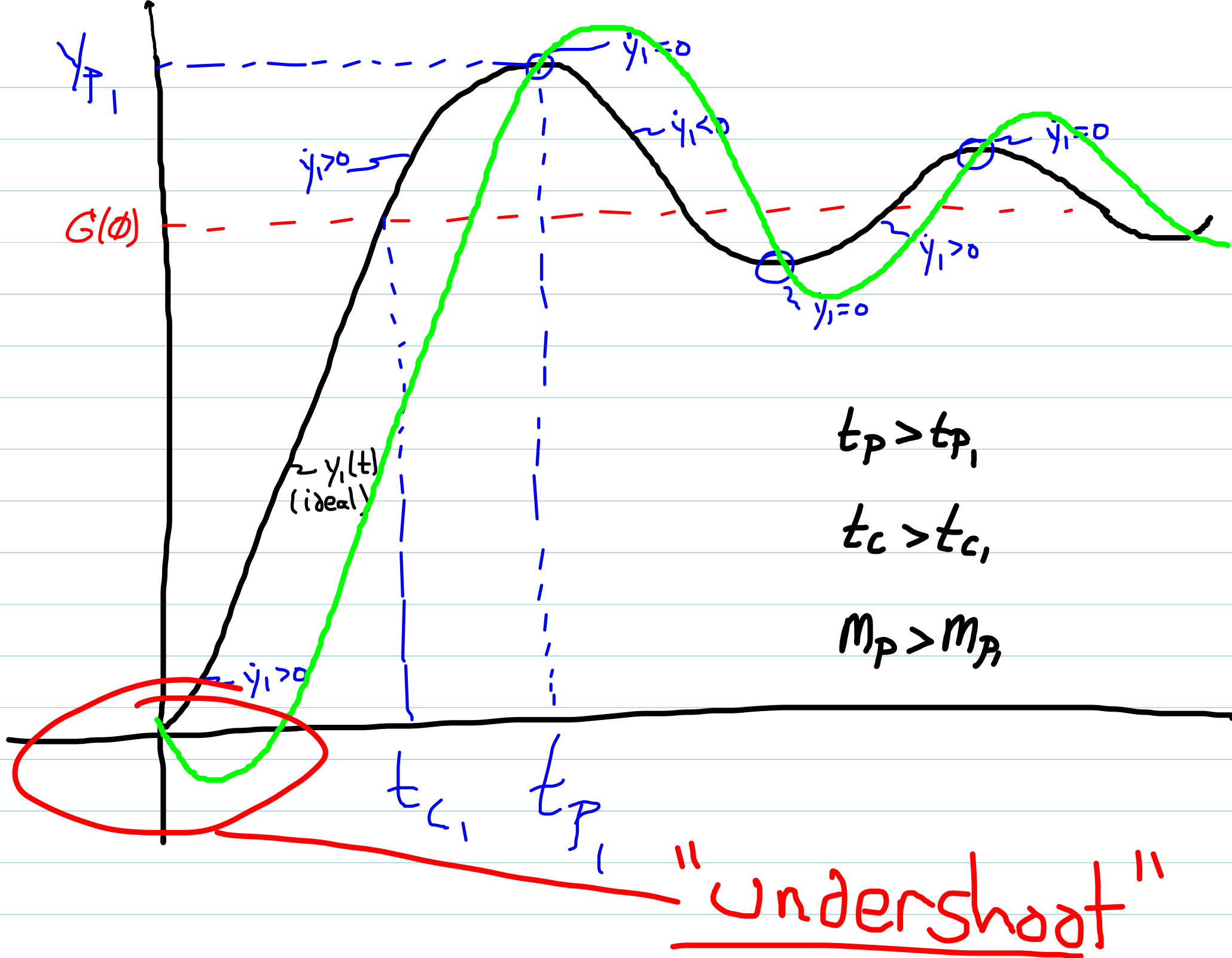
So we are subtracting the derivative from the ideal response.

But recall $\dot{y}_1(t) \geq 0$ for $0 < t < t_p$,

And $y_1(t) \approx 0$ for t close to zero

Seems to suggest that $y(t)$ may become negative for times near $t=0$...?





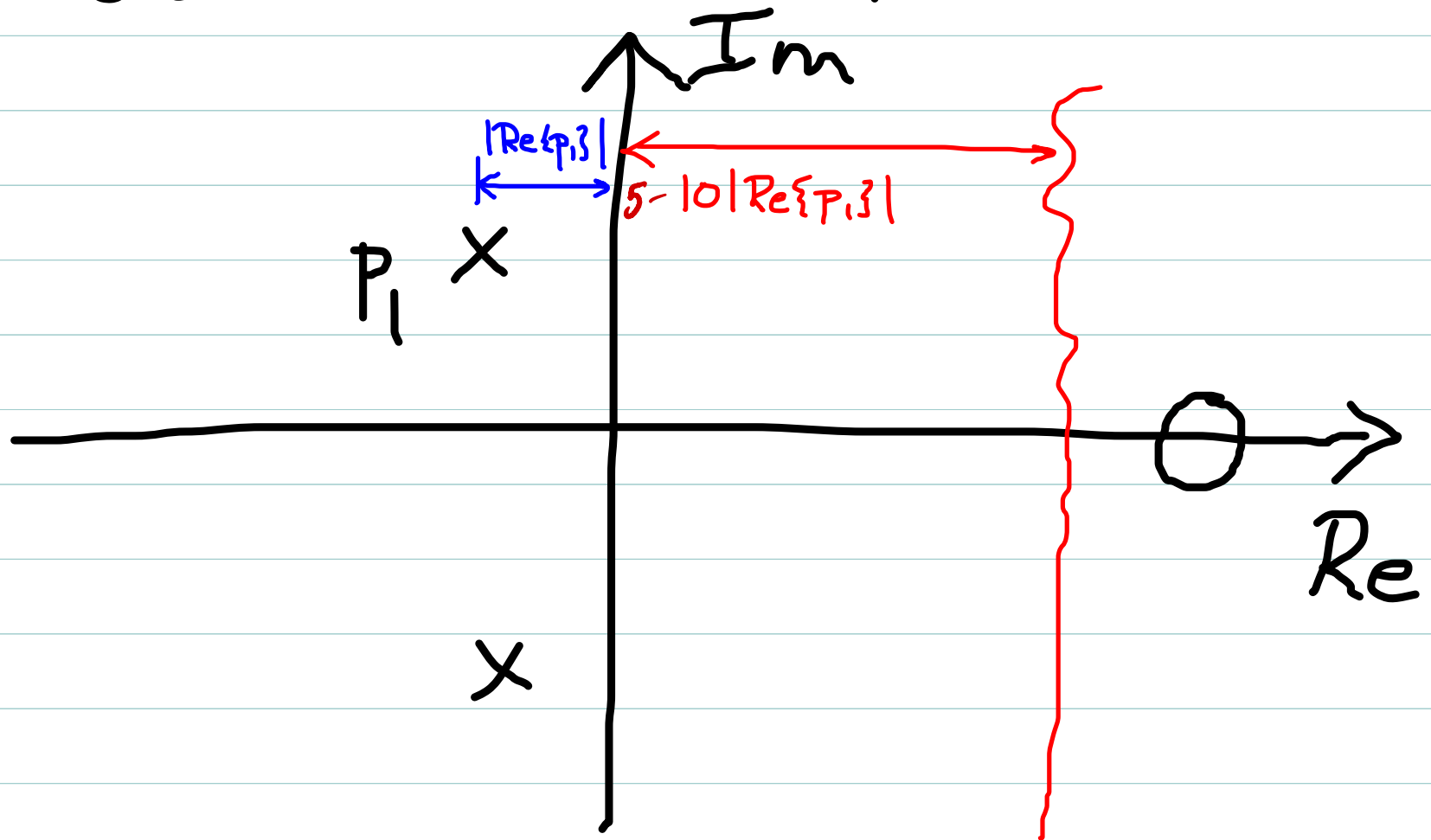
Observations (RHP zero)

- \Rightarrow Again, the peak response is greater
- \Rightarrow However, t_c and t_p have increased
- \Rightarrow Appearance of a new feature: "undershoot"
- \Rightarrow Response initially heads "in wrong direction"
before ultimately returning to the same steady-state
- \Rightarrow Such behavior is Not UNstable
- \Rightarrow It is, however, very tricky to design controllers
for such systems.

Effect is still proportional to $\frac{1}{|z_1|}$

hence diminishes as z_1 moves further from Im axis

Again negligible if $|z_1| > 10|\operatorname{Re}\{p_1\}|$



Effect on settling time

How a zero, either LHP or RHP, affects t_s is difficult to predict.

\Rightarrow Often, but not always, t_s is longer with zero due to increased amplitude of transient oscillations

\Rightarrow No hard and fast rule here

\Rightarrow Primary effect is increased overshoot and:

- reduction of t_c, t_p (LHP)

- undershoot, with increase of t_c, t_p (RHP)

Performance Specifications

⇒ Step inputs representative for many desired behaviors

- Move to new pointing angle (spacecraft)
- Move to new altitude or heading (aircraft)

⇒ Required performance often specified as upper

Limits on acceptable t_s , M_p

- System must settle quickly enough, and not overshoot too much.

⇒ Recall:

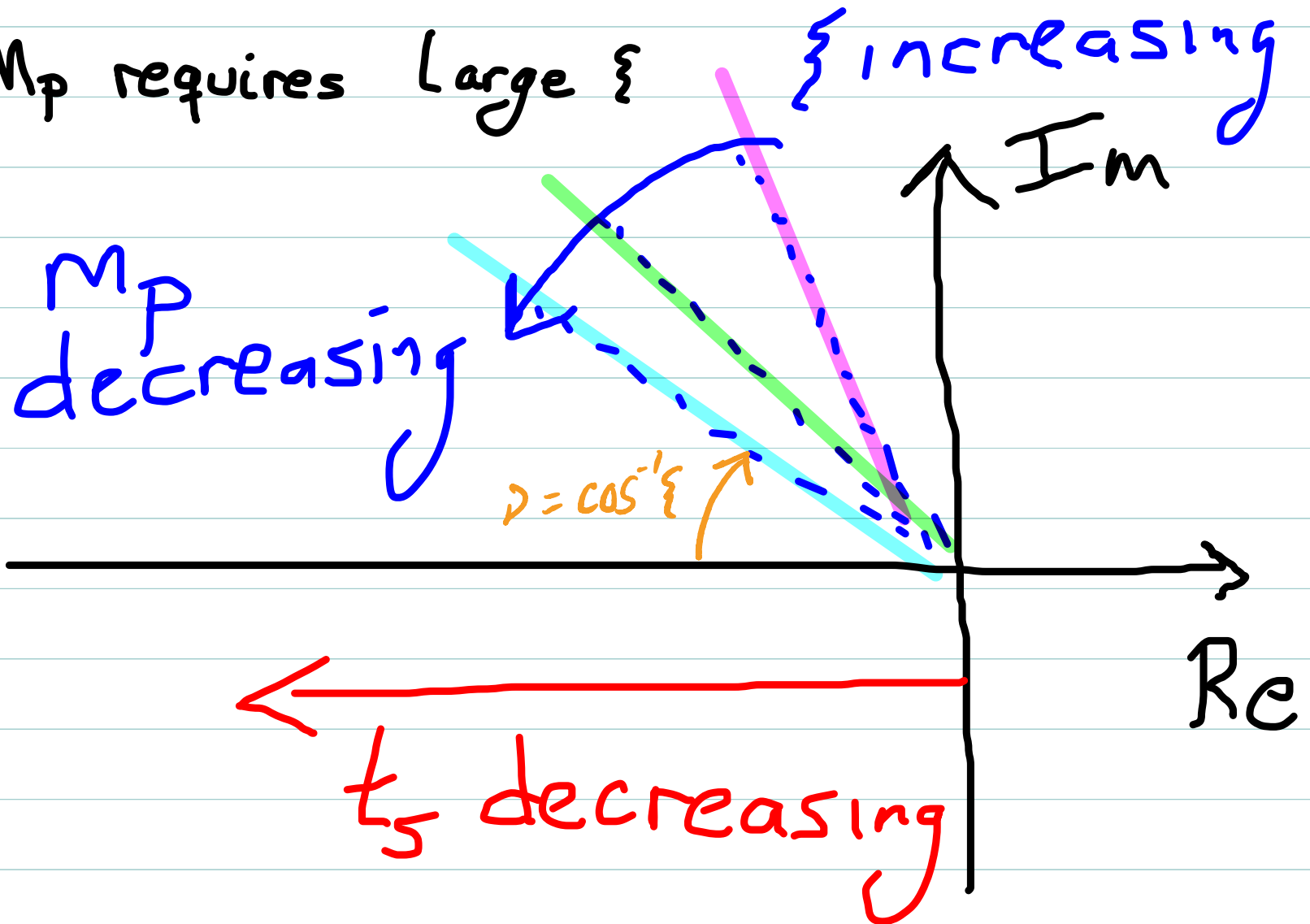
- t_s inversely proportional to $|\operatorname{Re}\{p_i\}|$
- M_p a decreasing function of ξ

$$t_s \approx \frac{4}{|\operatorname{Re}\{p_1\}|}$$

$$M_p = \exp\left[\frac{-\xi\pi}{\sqrt{1-\xi^2}}\right]$$

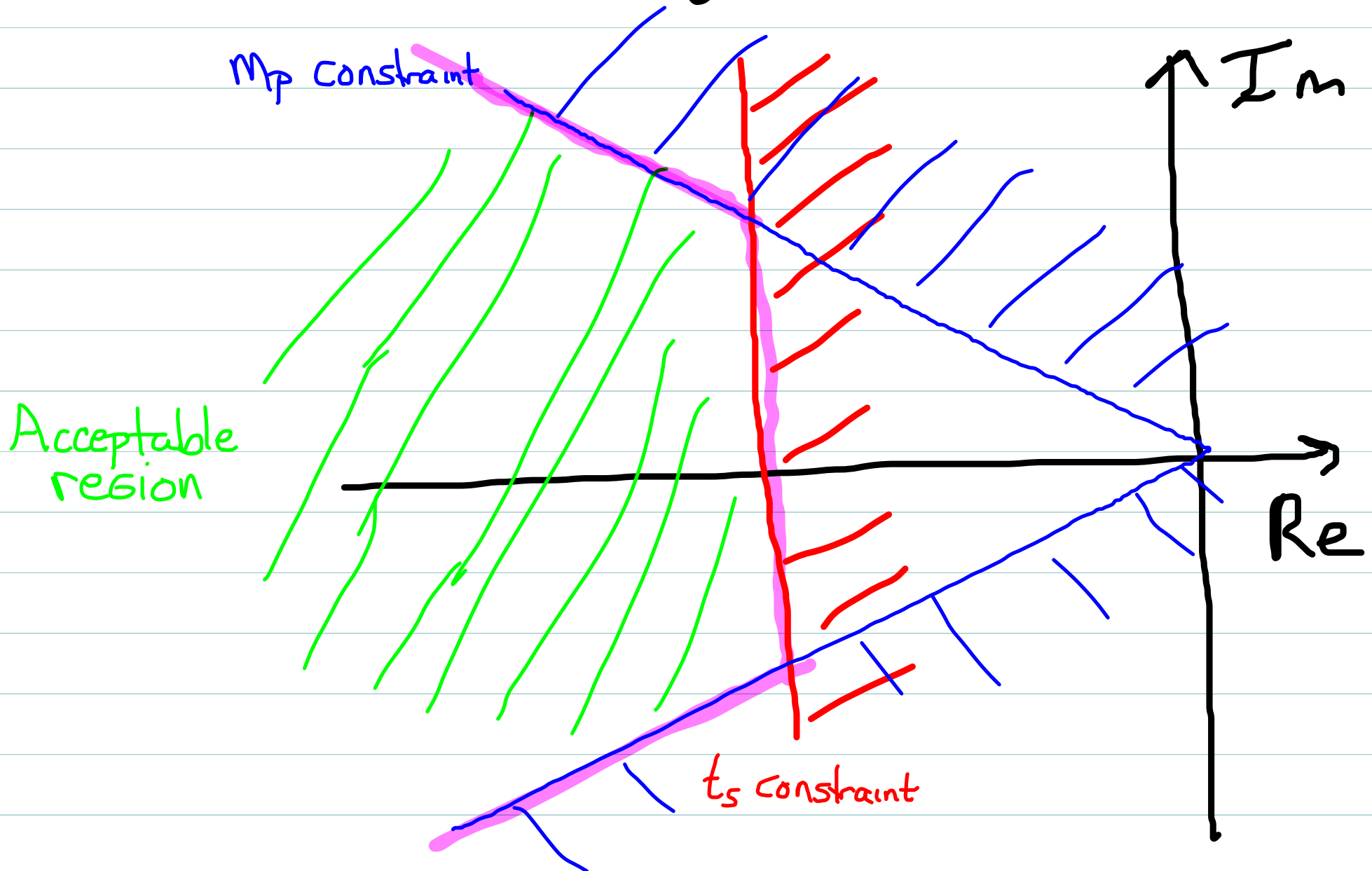
\Rightarrow Small t_s requires large $|\operatorname{Re}\{p_1\}|$

\Rightarrow Small M_p requires large ξ

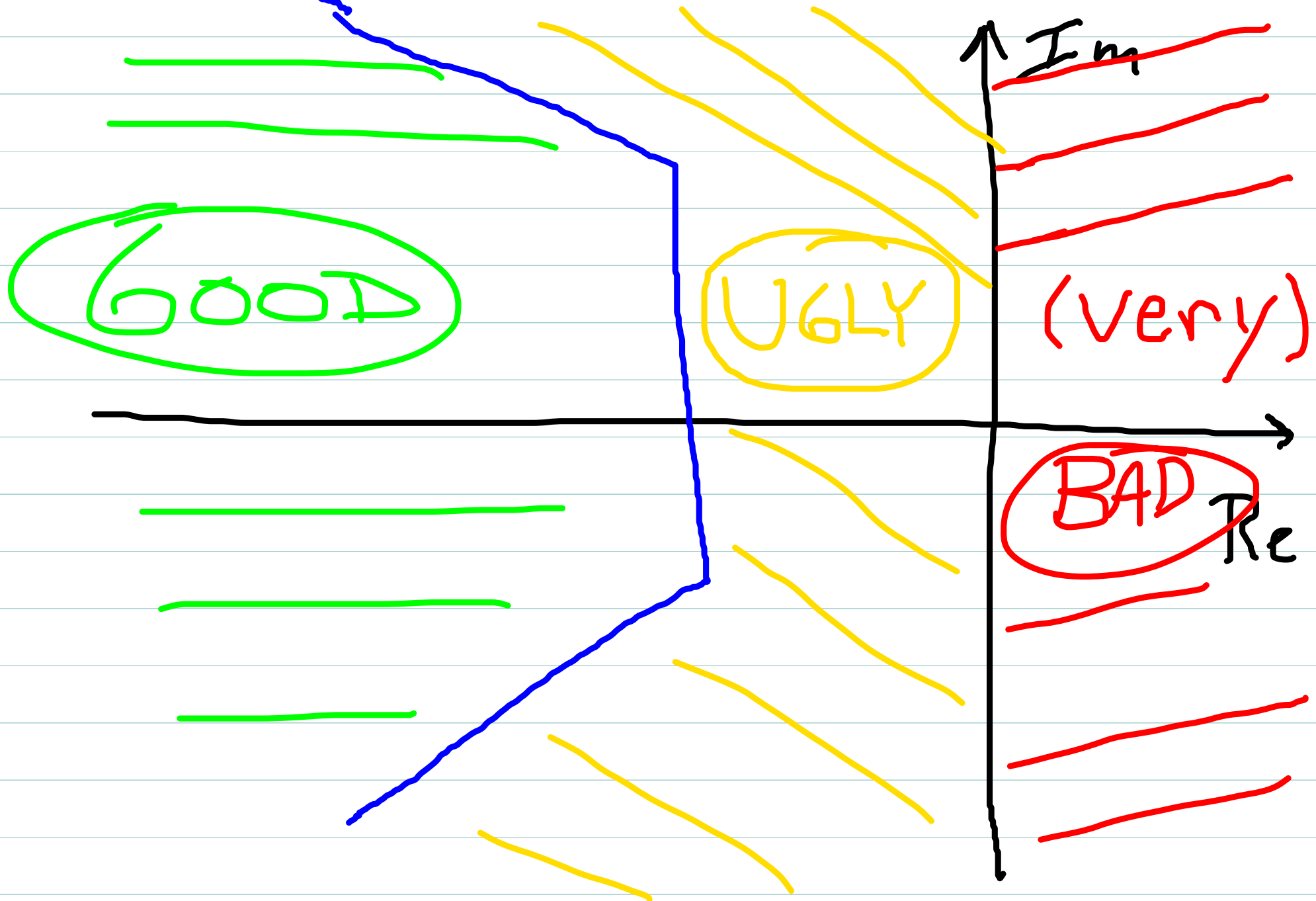


\Rightarrow Upper bound on t_s gives lower bound on $|\operatorname{Re}\{p_i\}|$

\Rightarrow Upper bound on M_p gives lower bound on ξ



Desireable Pole Locations



=> "Good" poles satisfy all transient performance constraints (upper bounds on t_s , M_p)

=> "Bad" poles are unstable

=> "Ugly" poles are stable, but have too much overshoot or take too long to settle.

=> Most aerospace system have natural dynamics which are "bad" or "ugly"

=> Goal of control is to make these systems "good"

Feedback "moves" poles

⇒ Already seen this on previous homeworks.

⇒ But it can be tricky!

$$\text{Suppose } u(t) = K(y_d(t) - y(t))$$

If system is modeled with $Y(s) = G(s)U(s)$

$$\text{where } G(s) = \frac{\beta_1 s + \beta_0}{s^2 + \alpha_1 s + \alpha_0}$$

Then poles are moved to roots of

$$r_{cl}(s) = s^2 + (\alpha_1 + K\beta_1)s + (\alpha_0 + K\beta_0)$$

=> Tricky to predict movement of poles for all possible values of $K, \alpha_0, \alpha_1, \beta_0, \beta_1$

=> Even more complicated for $G(s)$ with additional poles and/or zeros

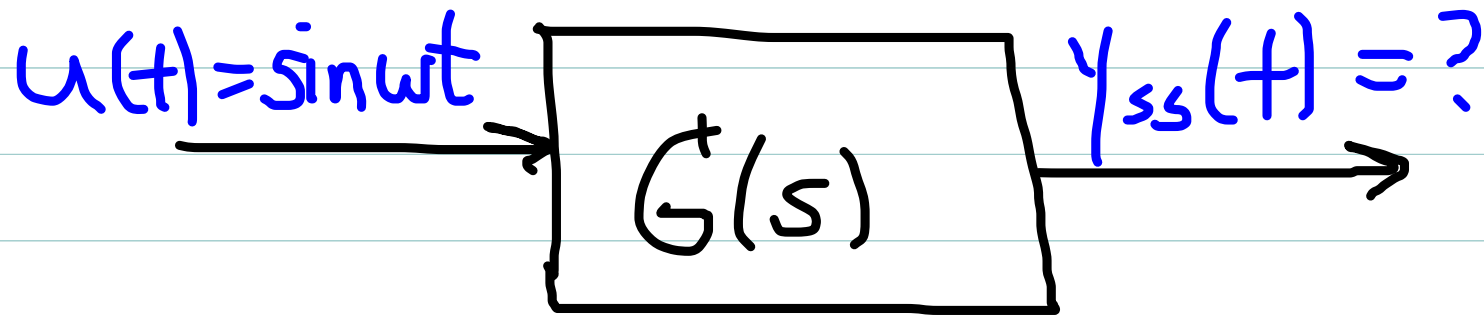
=> Need a more systematic tool to predict effectiveness of a control strategy.

=> One approach is based on a more careful analysis of the behavior of $G(j\omega)$.

Sinusoidal Response

Here we wish to understand the properties of the steady-state response of a stable system when $u(t) = \sin \omega t$.

Note: our focus is shifting (temporarily) away from the transient response



$$\Rightarrow y_{ss}(t) = \text{Im} \{ G(j\omega) e^{j\omega t} \}$$