

ENAE311H – Aerodynamics I

Instructor: Prof. Christoph Brehm

3184 Glenn L. Martin Hall, #3135

cbrehm1@umd.edu

Grader: TBD

Class Schedule: Tues./Thurs., 9:30 – 10:45, ITV 1111

Office Hours: TBD

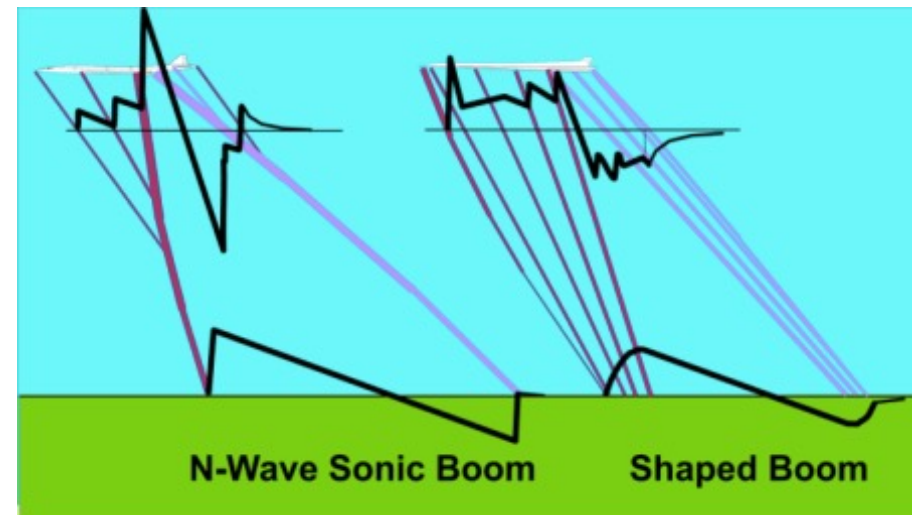
Textbooks: Fundamentals of Aerodynamics, John D. Anderson	(Required)
Introduction to Fluid Mechanics, Fox & McDonald	(FM Supplemental)
Elements of Gasdynamics, Liepmann & Roshko	(LR Supplemental)
Modern Compressible Flow with Historical Perspective, John D. Anderson	(A2 Supplemental)

Syllabus and Course Outline: uploaded on Canvas

Class Project:

Low Noise Supersonic Airplane Design Competition

- Design a supersonic airplane (based on Boom Overture)
 - Speed: Mach 1.7
 - Cruising Altitude: 60,000 ft
 - Passengers: 65 (premium-configured 1-1 cabin)
 - Full-scale length: 200ft (CFD analysis at 1:10 scale)
 - Wingspan: 60ft
 - Max takeoff weight: 170,000lb
 - Powerplant: 3 x Rolls-Royce medium-bypass turbofans without afterburners, 17,500 lbf (78kN) thrust
 - It's expected that this could cut travel time in half, with New York to London flights taking 3hr15min, Tokyo to Seattle flights taking 4hr30min, etc.
- Split up the design process into several parts
 - 2-D airfoil design
 - 3-D wing design
 - add streamlined main cabin without vertical stabilizer and no engines
- 28 students are split up into 4 teams
 - Split up team into 1 team captain, 2 CAD designers, 2 meshing and 2 CFD specialists
- Deliverables: 1 report describing the design and analysis are due at the last day of classes (limited to 20 pages) and a final presentation (20mins and 10mins Q&A)
- Evaluation Criteria: noise signature, correctness of analysis, how closely are the design parameters met?, as well as cabin comfort. (33% evaluation of presentation by your peers and 66% instructor)



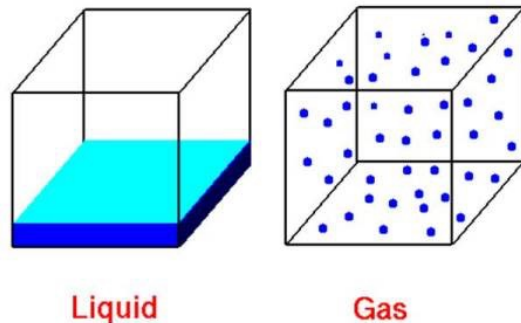
Introduction

ENAE311H Aerodynamics I

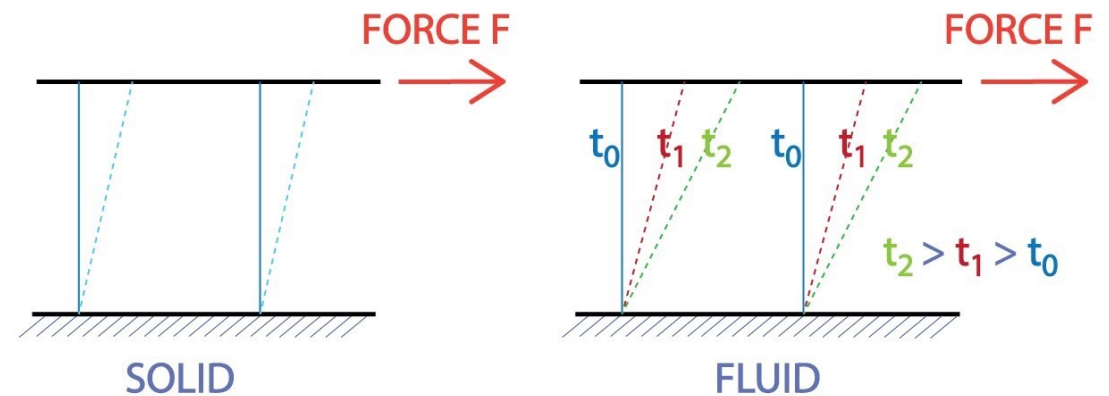
Christoph Brehm

Definitions

- *What is aerodynamics?*
 - A branch of fluid mechanics dealing with the motion of air (or other gases) and its interactions with solid surfaces
 - **External aerodynamics:** forces and moments (also heat transfer at high speeds) acting on bodies moving through air
 - **Internal aerodynamics:** flow of air through ducts, pipes, etc.
- *What is a fluid?*
 - A fluid is a substance that continuously deforms in response to an applied force (c.f. a solid, which undergoes a finite deformation in response to constant force)
 - **Liquid:** essentially incompressible (constant density) because of strong intermolecular forces
 - **Gas:** compressible because intermolecular forces are very weak



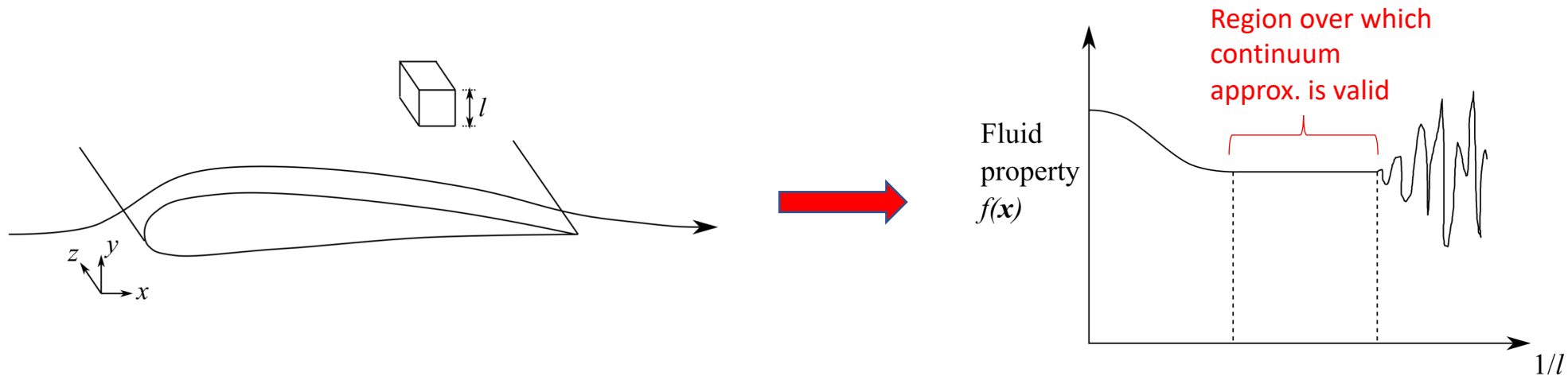
From <https://www.grc.nasa.gov/www/k-12/airplane/state.html>



From <https://www.youtube.com/watch?v=DL6QRTySWGs>

The continuum approximation

- Almost all fluid-dynamical theories treat a gas as a continuum (whereas we know it is composed of discrete molecules)
- Possible because, in macroscopic aerodynamics, typically there will be a huge number of molecules even in a volume corresponding to the smallest length scale of interest
 - For air at STP conditions, there are 2.7×10^7 particles in box with 1-micron sides
- This means all flow conditions are well-defined and smoothly varying (except, e.g., at shock)

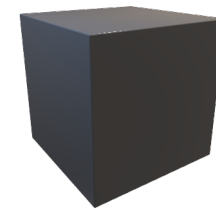


- Note, however, that transport coefficients must be derived from microscopic considerations

Density

- Density is the mass per unit volume of fluid – it is a scalar point property.
- Consider a fluid element at position \mathbf{x} ; the density, $\rho(\mathbf{x})$, is defined as

$$\rho(\mathbf{x}) = \lim_{dv \rightarrow 0} \frac{dm}{dv}$$



volume = dv

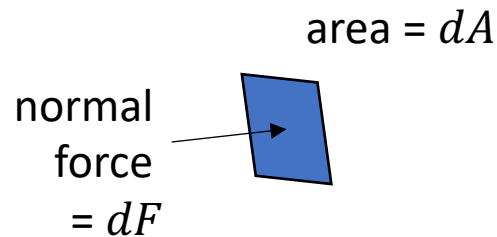
mass = dm

- At low speeds ($M < 0.3$), air can generally be considered a constant density fluid.

Pressure

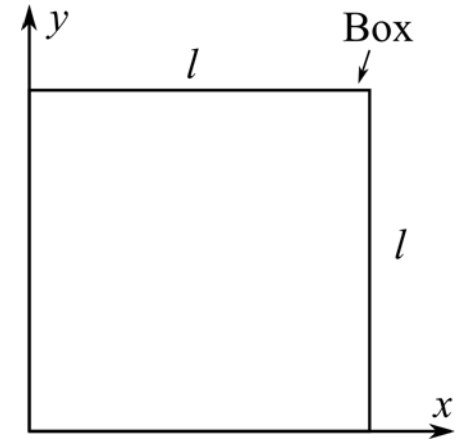
- Pressure is the force per unit area acting on a surface (real or imagined), moving at the bulk velocity of the flow – it is a scalar point property.
- Consider a fluid area element at position \mathbf{x} moving with the fluid; the pressure, $p(\mathbf{x})$, is defined as

$$p(\mathbf{x}) = \lim_{dA \rightarrow 0} \frac{dF}{dA}$$



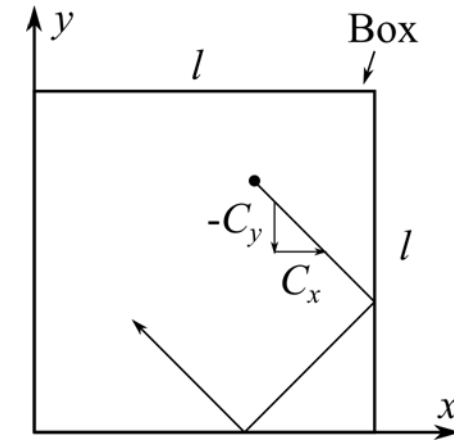
Pressure

- Pressure is related both to the density of the gas and the average random speed of the molecules that make it up. To see this, we consider a particle bouncing around inside a cubical box.



Pressure

- Pressure is related both to the density of the gas and the average random speed of the molecules that make it up. To see this, we consider a particle bouncing around inside a cubical box.
 - For simplicity, consider just a single particle colliding with x-normal wall (assume specular reflection)
 - Momentum transferred per collision: $2m|C_x|$
 - Time between collisions: $2l/|C_x|$
 - \Rightarrow Momentum deposited per unit time (force): mC_x^2/l
 - \Rightarrow Pressure from one particle: $mC_x^2/lA = mC_x^2/V$



Pressure

- Pressure is related both to the density of the gas and the average random speed of the molecules that make it up. To see this, we consider a particle bouncing around inside a cubical box.

- Summing over all particles, we have

$$p(\mathbf{x}) = \frac{1}{V} \sum_i m_i C_{xi}^2$$

- Note we can repeat for y- and z-normal walls, sum and divide by 3 to obtain

$$p(\mathbf{x}) = \frac{1}{3V} \sum_i m_i (C_{xi}^2 + C_{yi}^2 + C_{zi}^2) = \frac{1}{3V} \sum_i m_i C_i^2$$

- From our definition of density

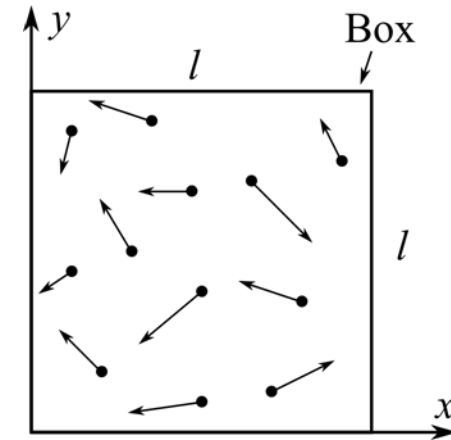
$$\rho(\mathbf{x}) = \frac{1}{V} \sum_i m_i$$

- And defining the mean-square molecular speed

$$\overline{C^2} = \frac{\sum_i m_i C_i^2}{\sum_i m_i}$$

- We obtain:

$$p = \frac{1}{3} \rho \overline{C^2}$$



Temperature

- Temperature is a thermodynamic property related to the average random kinetic energy of the molecules within the gas. It can be defined as

$$E_{tr} = \frac{3}{2} \mathcal{N} \tilde{R} T$$

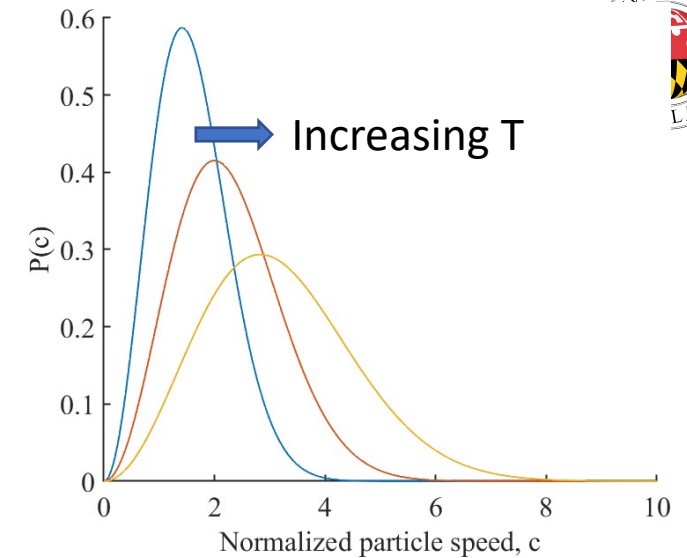
where E_{tr} is the total random molecular kinetic energy, \mathcal{N} is the number of moles of gas, and \tilde{R} is the universal gas constant.

- Noting that $E_{tr} = \frac{1}{2} \sum_i m_i C_i^2$ and comparing for our earlier expression for pressure, we can write

$$pV = \mathcal{N} \tilde{R} T \quad \text{or} \quad p = \rho \tilde{R} T$$

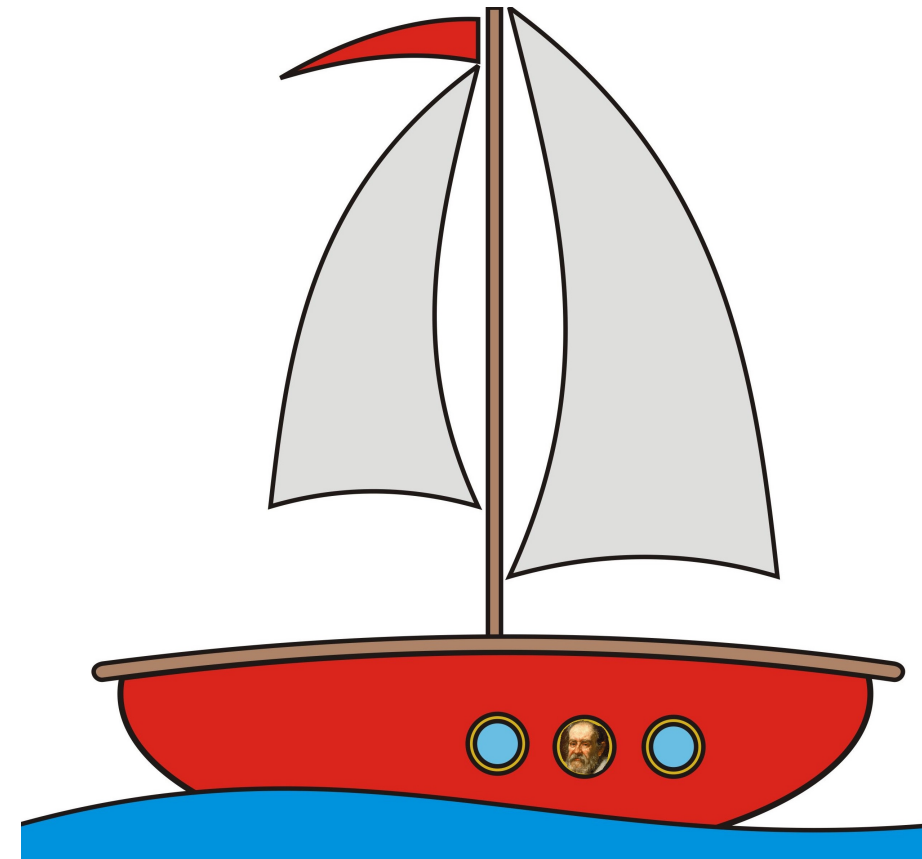
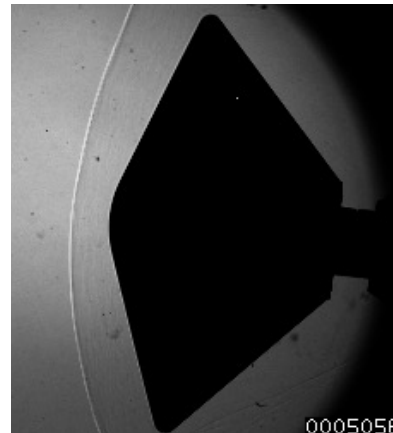
These are alternative expressions of the ideal gas law.

- At low speeds, temperature is essentially a passive scalar quantity (only way it affects the flow is through the viscosity) but for compressible flows it plays a much more important role.



Velocity

- Fluid velocity, $\mathbf{v}(\mathbf{x}) = (u, v, w)$ or (v_x, v_y, v_z) , is the mean velocity of a fluid element at \mathbf{x} .
- We denote the velocity magnitude V .
- An important concept in fluid mechanics is Galilean invariance:
 - “The laws of fluid motion are unchanged in any inertial reference frame.” (an inertial frame is one moving at a constant velocity)
 - Note that all purely thermodynamic properties (e.g., p, ρ, T, s) are also unchanged by a shift in inertial reference frame
 - Galilean invariance makes possible the testing of flight vehicles in wind tunnels:



Friction/shear stress

- Consider two adjacent elements of fluid, moving parallel but at slightly different speeds; we can imagine them exerting a tangential “rubbing” force on one another (though the physical origin of this force is the diffusional transport of momentum across their common boundary).
- If this tangential force is dF_f and the area over which it is applied is dA , the local shear stress, τ , is

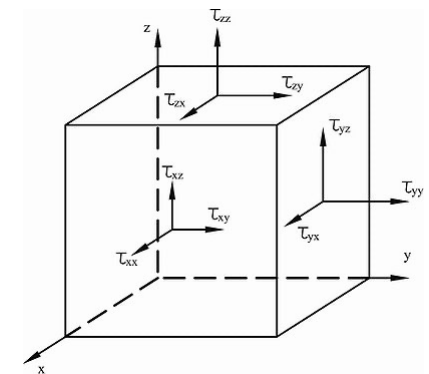
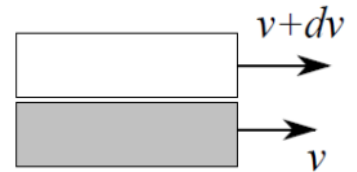
$$\tau = \lim_{dA \rightarrow 0} \frac{dF_f}{dA}$$

- For flow past a solid surface, the shear stress is similarly the tangential force per unit area.
- A *Newtonian* fluid (as are most common fluids) is one for which the shear stress is proportional to the velocity derivative normal to the surface element, e.g., for flow in the x direction that changes only in the y direction,

$$\tau = \mu \frac{du}{dy}$$

where μ , the coefficient of viscosity, depends only on temperature.

- Note that, in general, the velocity has three components, each of which may change in every direction, so the shear stress is actually a 2nd-order tensor (9-element matrix).



Units

