## Lecture 5- Orbital Elements

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## Orbital Elements:

て, す

Cibilal elevants are an alternate set of coordinates that are used instead of F,V (Cartesian Coordinates)

B/C 6 Cartesian Coordinates =) 6 orbital elevants

We have already been using 3 of the abital elements: a, e, v

We have so for just thought of orbits in 2D - is in the arbital place

The other 3 orbital elements describe the orientation of the orbital place in 3D space \$ the orientation of the orbital place.

i=inclination: angle between Th \$ = (I=TXV)

I = long; trude of the ascending node: anyle from it to the ascending node

ascending nade is the point on the orbit where the orbital glare crosses
the countries of the six is ascending (also "mains towards 2"

the equational plane to the 5/c is ascending (aska "morry towards 2)

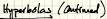
W= argument of periapsis: argle from the ascending node to periapsis (in the arbital plane)

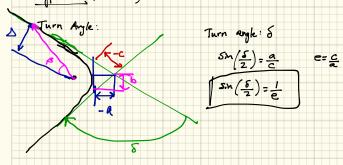
in the direction of spacecraft motion.

a,e,i, sz,w, V

D= miss distance

All the abotal elevants are constant except v, valess we execute a maneuver.





h=rvcos 8

At 00, V is parallel to the asymptotes
h=rvcos 8= VSSiny3 = VOO D

Canonical Units: useful when dony calculations by hand

wether of madinensimalisation

1 AU = Astronomical Unit = distance from the Sun to the Earth

= 149,597,871 km

- interplanty trajectories

DU = distance units = offen, radius of the Earth

M=1 DV3/TU2

We pick a DU ( some # of km) and we know that u=1 003/TV2. So we an solve for TV.

1 DU = 6378 km

M= 1 DO3/102 = 3.986 × 105 km2/52

 $(\frac{Du^3}{TU^2}, \frac{(6348 \text{ Lym})^3}{(1 DU})^3, \frac{(1 TU)^2}{\chi \text{ sec}} = 3.986 \times 10^5 \frac{\text{km}^3}{5^2}$ 

=) solve for X. =) ITV = X sec

Then, express F, v, etc (a) in terms of DU & DU/TV.