

Application of Class Note #5:

Solution of the single DOF arbitrary gust problem

Jean-Camille Chassaing, Sorbonne Université (Oct. 2021)

<http://www.dalembert.upmc.fr/home/chassaing>

Question 1

The operateur of the structural inertial forces is simply given by : $\mathcal{I}_m = m\ddot{z}(t)$ where m denotes the lineic mass and $z(t)$ is the vertical displacement of the aircraft. We remark that the sign of z is the opposite to those considered in Classnote #5.

Defining the non-dimensional time $s = U_\infty t/b$, we apply the derivation chain rule :

$$\begin{aligned} z' &= \frac{dz}{dt} \frac{dt}{ds} = \frac{U_\infty}{b} \dot{z}(s) \\ z''(s) &= \frac{d\dot{z}}{dt} \frac{dt}{ds} = \left(\frac{U_\infty}{b} \right)^2 \ddot{z}(s) \end{aligned}$$

which gives

$$\mathcal{I}_m(s) = m(U_\infty/b)^2 z''(s)$$

Question 2

According to Eq. 35 in ClassNote #5, we know that the lift induced by an gust with arbitrary amplitude is given by

$$L_G(s) = 2\pi\rho_\infty U_\infty b \left[w_G(0)\psi^{kus}(s) + \int_0^s \frac{dw_G(\sigma)}{d\sigma}(\sigma)\psi^{kus}(s-\sigma)d\sigma \right] \quad (1)$$

Integrating by parts the integro-differential terms in Eq. 1 equations gives

$$\begin{aligned} \int_0^s \frac{dw_G(\sigma)}{d\sigma}(\sigma)\psi^{kus}(s-\sigma)d\sigma &= \left[w_G(\sigma)\psi^{kus}(s-\sigma) \right]_{\sigma=0}^{\sigma=s} - \int_0^s w_G(\sigma) \left[-\frac{d\psi^{kus}}{d\sigma}(s-\sigma) \right] d\sigma \\ &= 0 - w_G(0)\psi^{kus}(s) + \int_0^s w_G(\sigma) \frac{d\psi^{kus}}{d\sigma}(s-\sigma)d\sigma \end{aligned} \quad (2)$$

Inserting the previous relation in Eq. 2 gives the expected results

$$L_G(s) = 2\pi\rho U_\infty b \int_0^s w_G(\tau) \frac{d\psi^{kus}}{d\tau}(s-\tau)d\tau \quad (3)$$

Question 3

The lift corresponding to an arbitrary motion is described by (see Eq. 36 in ClassNote #5)

$$L_M(s) = L_{\Gamma=0} - 2\pi\rho_\infty U_\infty b \left[w_{3/4}(0)\psi^{wag}(s) + \int_0^s \frac{dw_{3/4}(\sigma)}{d\sigma}(\sigma)\psi^{wag}(s-\sigma)d\sigma \right] \quad (4)$$

Since we consider the vertical motion of a rigid aircraft, we straightfully have the analogy: $z(t) \sim h(t)$ and $w_{3/4}(t) = \dot{h}(t) \sim \dot{z}(t)$.

Moreover, the non-circulatory part of the lift due to the aircraft motion can be deduced from:

$$L_{\Gamma=0}(t) = \pi\rho_\infty b^2 (\ddot{h} + U_\infty \dot{\alpha} - ab\ddot{\alpha}) = -\pi\rho_\infty b^2 \ddot{z}(t) \quad (5)$$

or equivalently

$$L_{\Gamma=0}(s) = -\pi\rho_\infty U_\infty^2 z''(s) \quad (6)$$

Consequently, we obtain

$$L_M(s) = -\pi\rho_\infty U_\infty^2 z''(s) - 2\pi\rho_\infty U_\infty b \left[z(0)\psi^{wag}(s) + \int_0^s \frac{dz(\sigma)}{d\sigma}(\sigma)\psi^{wag}(s-\sigma)d\sigma \right] \quad (7)$$

Question 4

Assembling Eqs ???, 3 and 7, the equation of the aeroelastic motion given by $I_m = L_G + L_M$ reads

$$\left(\frac{U}{b} \right)^2 m z''(s) = 2\pi\rho U_\infty b \int_0^s w_G(\tau) \frac{d\psi^{kus}}{d\tau}(s-\tau)d\tau - \pi\rho_\infty U_\infty^2 \left[z''(s) + 2 \int_0^s z''(\tau)\psi^{wag}(s-\tau)d\tau \right] \quad (8)$$

Question 5

Let $\bar{f}(p) = \mathcal{L}[f(s)] = \int_0^\infty f(s)e^{-pt}dt$ with $z(0) = z'(0) = 0$ and $\psi^{kus}(0) = 0$ (see Fig. 6 in Classnote #5).

We can compute

$$\mathcal{L} \left[\int_0^s w_G(\tau) \frac{d\psi^{kus}}{d\tau}(s-\tau)d\tau \right] = \mathcal{L}(w_G(s))L(d\psi^{kus}/ds)(s) = \bar{w}_G(p) \left[p\psi^{kus}(p) - \underbrace{\psi^{kus}(0)}_{=0} \right] \quad (9)$$

and

$$\mathcal{L} \left[\int_0^s z''(\tau)\psi^{wag}(s-\tau)d\tau \right] = p^2 \bar{z}(p) \bar{\psi}^{wag}(p) \quad (10)$$

Then, the dynamic equation of motion written in the Laplace domain are

$$\underbrace{\left(\frac{U}{b} \right)^2 m p^2 \bar{z}(p)}_{inertial\ structural\ effect} = \underbrace{2\pi\rho_\infty U_\infty b \bar{w}_G(p) \bar{\psi}^{kus}}_{gust\ effect} - \underbrace{\pi\rho_\infty U_\infty^2 p^2 \bar{z}}_{added\ mass\ effect} - \underbrace{2\pi\rho_\infty U_\infty^2 p^2 \bar{z}(p) \bar{\psi}^{wag}}_{arbitrary\ motion} \quad (11)$$

Dividing each member of Eq. (ref{EqLaplace1} by \$ 2 \pi \rho_\infty U_\infty^2\$, we obtain

$$\frac{m}{2\pi\rho_\infty b_\infty^2} p^2 \bar{z}(p) = b p \frac{\bar{w}_G}{U_\infty} \bar{\psi}^{kus} - \frac{p^2}{2} \bar{z} - p^2 \bar{z} \bar{\psi}^{wag} \quad (12)$$

Introducing the mass ratio $\lambda = m/(2\pi\rho_\infty b_\infty^2)$, the solution in the Laplace domain can be formely derived as

$$\bar{z}(p) = b \frac{(\bar{w}_G/U_\infty) \bar{\psi}^{kus}}{p \left(\lambda + \frac{1}{2} + \bar{\psi}^{wag} \right)} \quad (13)$$

where the expressions of $\bar{w}_G(s)$, $\bar{\psi}^{kus}(s)$ and $\bar{\psi}^{wag}(s)$, which are the Laplace transforms of $w_G(s)$, $\psi^{kus}(s)$ and $\psi^{wag}(s)$ must be prescribed to complete the computation of $\bar{z}(s)$.

Question 6

Because taking the Laplace transform of exact expressions of $\bar{\psi}^{wag}(s)$ and $\bar{\psi}^{kus}(s)$ is impractical, we consider the following approximations

$$\begin{aligned} \psi^{kus}(s) &= 1 - 0.5e^{-\alpha_1 s} - 0.5e^{-\alpha_2 s} \\ \psi^{wag}(s) &= 1 - b_1 e^{-\beta_1 s} - b_2 e^{-\beta_2 s} \end{aligned} \quad (14)$$

where $\alpha_{1,2}$, $\beta_{1,2}$ et $b_{1,2}$ are positive constants.

Then we compute easily

$$\begin{aligned} \bar{\psi}^{kus}(p) &= \mathcal{L}[\psi^{kus}(s)] = \frac{1}{p} - \frac{0.5}{p + \alpha_1} - \frac{0.5}{p + \alpha_2} \\ \bar{\psi}^{wag}(p) &= \mathcal{L}[\psi^{wag}(s)] = \frac{1}{p} - \frac{b_1}{p + \beta_1} - \frac{b_2}{p + \beta_2} \end{aligned} \quad (15)$$

Consequently, Eq. (13) becomes

$$\bar{z}(p) = \frac{b}{U} \bar{w}_G \frac{1}{p} \underbrace{\left(\frac{1}{p} - \frac{1/2}{p + \alpha_1} - \frac{1/2}{p + \alpha_2} \right)}_{f_1(p)} \underbrace{\left(\lambda + \frac{1}{2} + \frac{1}{p} - \frac{b_1}{p + \beta_1} - \frac{b_2}{p + \beta_2} \right)^{-1}}_{f_2(p)} \quad (16)$$

with $f_1(p) = M_1(p)/N_1(p)$ where $M_1(p)$ and $N_1(p)$ represent polynomials of degree 2 and 4 respectively. Similary, $f_2(p) = M_2(p)/N_2(p)$ where $M_2(p)$ and $N_2(p)$ represent polynomials of degree 3.

A compact form of Eq. 16 may be written as

$$\bar{z}(p) = \frac{b}{U} \bar{w}_G(p) \frac{M(p)}{N(p)} \quad (17)$$

where $M(p)$ et $N(p)$ are polynomials of degree 5 and 7 respectively.

Question 7

Finally, the solution can be obtained in the physical domain as

$$\bar{z}(t) = \int_0^s w_G(u) \left(\sum_{k=1}^7 \frac{\mathcal{L}^{-1}(M(p_k))}{\mathcal{L}^{-1}(N(p_k)')} e^{p_k(s-u)} \right) du \quad (18)$$

where p_k denote the roots of the polynomial $N(p)$.

End of this notebook