

# High-Fidelity Simulations for Turbulent Flows

Luca Sciacovelli

DynFluid Laboratory  
Arts et Métiers Institute of Technology  
<http://savoir.ensam.eu/moodle>

Master Recherche “Aérodynamique et Aéroacoustique”  
*2021 – 2022*



Sciences et  
Technologies  
**Arts  
et Métiers**



## Part VIII

### High-order schemes for compressible flows

## 1 Introduction

## 2 Methods for smooth flows

- High-order centred derivatives
- Stabilization for smooth flows
- Energy-consistent schemes

## 3 Methods for non smooth flows

## 1 Introduction

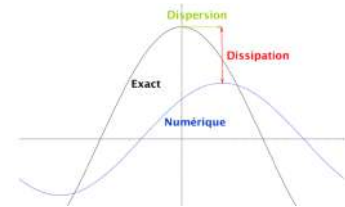
### 2 Methods for smooth flows

- High-order centred derivatives
- Stabilization for smooth flows
- Energy-consistent schemes

### 3 Methods for non smooth flows

# Motivation

- ▶ Numerical schemes introduce **dissipation (phase)** and **dispersion (amplitude)** errors altering the **representation** of a given solution mode
  - The smaller structures are the worst represented
  - Small structures may play a crucial role (turbulence, aeroacoustics, ..)
- ▶ Numerical errors interact with the model used to represent unresolved scales  $\Rightarrow$  Choice of the numerical method is **crucial!**
  - Remember: errors can be **one order of magnitude greater** than the subgrid modelling for low-order spatial discretizations
  - They produce a numerical cutoff well below the theoretical cutoff wavenumber associated with the grid



Two possibilities to palliate to the numerical errors:

1. **Increase grid resolution**: need for massively parallel solver, high memory load and storage, costly, ..
2. **Increase scheme resolution**: *i.e.* increase the scheme order  $p$ , and/or “optimize” the scheme so to lower the error constant  $C$  of the local truncation error  $\varepsilon = C(\Delta x)^p$ 
  - More properly, one should look to the **convergence order**  $q$ , *i.e.*

$$E = ||w_{\text{num}} - w_{\text{ex}}|| = C_{\text{conv}} \Delta x^q$$

- Typically  $q < p$  because of boundary conditions, shocks, time integration, ..

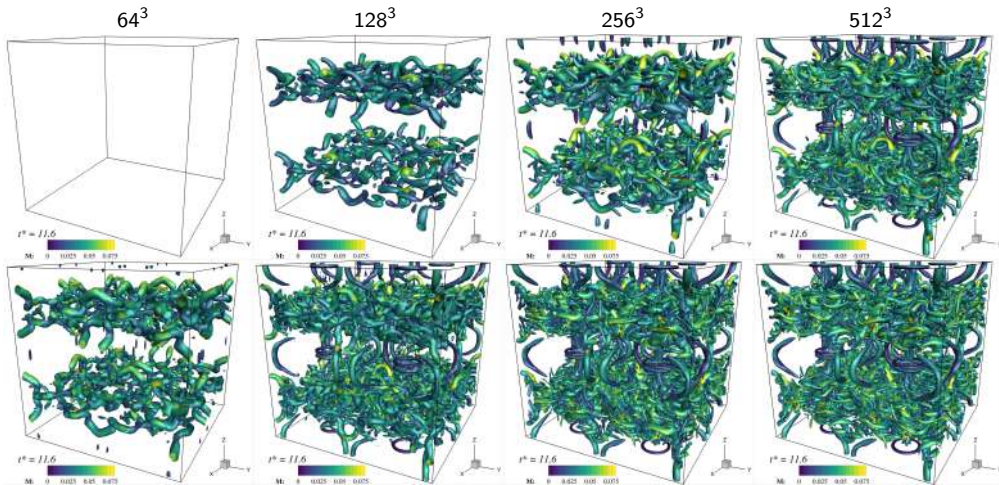
# Example: the Taylor-Green Vortex problem

$$\begin{cases} u_0(x, y, z) = u_\infty \sin(x) \cos(y) \cos(z) \\ v_0(x, y, z) = -u_\infty \cos(x) \sin(y) \cos(z) \\ w_0(x, y, z) = 0 \\ p_0(x, y, z) = p_\infty + \frac{\rho_\infty u_\infty^2}{16} [\cos(2x) + \cos(2y)] [\cos(2z) + 2] \end{cases}$$

$64^3$                        $128^3$                        $256^3$                        $512^3$

- ▶ Classical test case for high-order methods
- ▶ Incompressible limit ( $M < 0.3$ )
- ▶  $\Omega \in [0, 2\pi]^3$
- ▶ Visualization of the  $Q$ -criterion:

3<sup>rd</sup>-order  
scheme



9<sup>th</sup>-order  
scheme

## Problem statement

### Hyperbolic system of conservation laws:

$$\frac{d}{dt} \int_{\Omega} \mathbf{w} \, d\Omega + \sum_{i=1}^3 \int_{\partial\Omega} (\mathbf{F}_i^e - \mathbf{F}_i^v) n_i \, dS = 0 \quad \text{with} \quad \mathbf{w} = \begin{bmatrix} \rho \\ \rho u_j \\ \rho E \end{bmatrix} \quad \mathbf{F}^e = \begin{bmatrix} \rho u_i \\ \rho u_i u_j + p \delta_{ij} \\ \rho u_i H \end{bmatrix} \quad \mathbf{F}^v = \begin{bmatrix} 0 \\ \tau_{ij} \\ \tau_{ik} u_k - q_i \end{bmatrix}$$

- **Smooth flow assumption:** NSE can be recast as  $\frac{\partial \mathbf{w}}{\partial t} + \sum_{i=1}^3 \frac{\partial \mathbf{F}_i^e}{\partial x_i} - \sum_{i=1}^3 \frac{\partial \mathbf{F}_i^v}{\partial x_i} = 0$
- **Euler equations** ( $\mathbf{F}_i^v = 0$ ) have **two important properties** for the development of numerical methods:
  1. **Hyperbolicity:** Recast in characteristic form, the projection in any direction gives rise to a system of coupled wave-like eqs, motivating the study of the model 1D scalar conservation law

$$\frac{\partial w}{\partial t} + \frac{\partial F(w)}{\partial x} = \frac{\partial w}{\partial t} + a(w) \frac{\partial w}{\partial x} = \nu \frac{\partial^2 w}{\partial x^2}$$

2. **Conservation properties:** Apart from conservation of integrals of  $\mathbf{w}$  components, kinetic energy (KE) is also conserved as shown from balance equation

$$\frac{d}{dt} \int_{\Omega} \rho \frac{u_k u_k}{2} \, d\Omega = - \int_{\partial\Omega} \left( \rho \frac{u_k u_k}{2} + p \right) u_i n_i \, dS + \int_{\Omega} \rho \frac{\partial u_i}{\partial x_i} \, d\Omega$$

- Varies only for boundary flux or volumetric pressure work: **convective terms do not cause variations**
- Lead to attempt of building schemes that enforce “KE preservation” **in the discrete sense**
- Similar considerations for thermodynamic **entropy**-related functions (Harten, 1983)

# Common choices for high-resolution numerical methods

- ▶ **Spectral methods:** Fourier for homogeneous dirs, Chebychev/Legendre polyn. for inhomogeneous dirs
  - ✓ Spectral resolution
  - ✗ Simple geometries, incompressible flows
- ▶ **High-order centred schemes:** at least 4th-order; spectral resolution enhanced by minimizing the dispersion error (reducing the formal order), or by using Padé-like fractions (compact schemes)
  - ✓ High-order straightforward to achieve
  - ✗ Non-dissipative, need stabilization
- ▶ **High-order non-centred schemes:** flux decomposition schemes (Roe/AUSM with MUSCL reconstr.), ENO/WENO schemes, OSMP, ...
  - ✓ Stabilization embedded in the scheme
  - ✗ The intrinsic dissipation can be harmful for LES
- ▶ **Energy/Entropy-consistent schemes:** skew-symmetric splitting for the convective term ensuring semidiscrete KE/entropy preservation
  - ✓ Stabilization embedded in the scheme
  - ✗ Not adapted to shocked flows

## Compressible vs Incompressible

Methods for incompressible flows not suited for compressible (especially low- $M$ ) ones

$$\frac{\Delta t_c}{\Delta t_i} = \frac{u}{u + c} = \frac{M}{1 + M}$$

$$\Delta t_c \ll \Delta t_i \text{ for } M \rightarrow 0$$

## Smooth vs shocked compressible flows

- ▶ Standard smooth-flow discretizations can cause strong Gibbs oscillations in presence of shocks
- ▶ Shocked-flow methods exhibit excessive numerical dissipation for smooth flows

⇒ **two distinct classes exist!**

**The perfect scheme for any flow problem does not exist, it depends on what you are looking for!**



## ■ Introduction

## ■ Methods for smooth flows

- High-order centred derivatives
- Stabilization for smooth flows
- Energy-consistent schemes

## ■ Methods for non smooth flows

# High-order standard centred differences

$$\frac{\partial f}{\partial x}(x_0) = \frac{1}{\Delta x} \sum_{j=-N}^N a_j [f(x_0 + j\Delta x)] = \frac{1}{\Delta x} \sum_{j=1}^N a_j [f(x_0 + j\Delta x) - f(x_0 - j\Delta x)] \quad \text{where} \quad a_j = -a_{-j}$$

All the terms of the Taylor expansion of  $f$  are cancelled until the order  $\Delta x^{2N-1}$  included:

$$f(x_0 + j\Delta x) = \cancel{f(x_0)} + j\Delta x f'(x_0) + \cancel{\frac{(j\Delta x)^2}{2!} f''(x_0)} + \frac{(j\Delta x)^3}{3!} f'''(x_0) + \cancel{\frac{(j\Delta x)^4}{4!} f^{(iv)}(x_0)} + \dots$$

$$f(x_0 - j\Delta x) = \cancel{f(x_0)} - j\Delta x f'(x_0) + \cancel{\frac{(j\Delta x)^2}{2!} f''(x_0)} - \frac{(j\Delta x)^3}{3!} f'''(x_0) + \cancel{\frac{(j\Delta x)^4}{4!} f^{(iv)}(x_0)} + \dots$$

thus

$$\frac{\partial f}{\partial x}(x_0) = \frac{1}{\Delta x} \sum_{j=1}^N a_j \left[ 2j\Delta x f'(x_0) + \frac{2j^3 \Delta x^3}{3!} f'''(x_0) + \frac{2j^5 \Delta x^5}{5!} f^{(v)}(x_0) + \frac{2j^7 \Delta x^7}{7!} f^{(vii)}(x_0) + \dots \right]$$

$$\begin{cases} \sum_{j=1}^N 2ja_j &= 1 \\ \sum_{j=1}^N j^3 a_j &= 0 \\ \vdots & \\ \sum_{j=1}^N j^{2N-1} a_j &= 0 \end{cases} \quad \text{and} \quad a_j = -a_{-j}$$

- Use additional points to increase the formal order of accuracy of the approximation
- Explicit recursive correction of the dispersive error
- Non-dissipative, dispersive at order  $2N+1$ 
  - **Need for numerical stabilization**

**$2N + 1$  points,  $N$  relations, order  $2N$**

# High-order standard centred differences - delta notation

## Recursive correction of the dispersive error: derivation in delta form

3-points, 2<sup>nd</sup>-order centred scheme:  $\frac{\partial f}{\partial x} = f'_j \approx \frac{f_{j+1} - f_{j-1}}{2\Delta x} = \frac{\delta \mu f_j}{\Delta x}$

$$\begin{cases} f_{j+1} = f_j + \Delta x f'_j + \frac{\Delta x^2}{2} f''_j + \frac{\Delta x^3}{6} f'''_j + \dots \\ f_{j-1} = f_j - \Delta x f'_j + \frac{\Delta x^2}{2} f''_j - \frac{\Delta x^3}{6} f'''_j + \dots \end{cases} \Rightarrow (*) \quad f'_j = \frac{\delta \mu f_j}{\Delta x} - \frac{\Delta x^2}{6} f'''_j + \dots$$

Discretization of  $n$ -th deriv. in delta form:  $f_j^n = \frac{\delta^n \mu f_j}{\Delta x^n} \Rightarrow f'''_j \approx \frac{\delta^3 \mu f_j}{\Delta x^3} = \frac{f_{j+2} - 2f_{j+1} + 2f_{j-1} - f_{j-2}}{2\Delta x^3} \quad (\bullet)$

$$\begin{cases} f_{j+2} = f_j + 2\Delta x f'_j + 4\frac{\Delta x^2}{2} f''_j + 8\frac{\Delta x^3}{6} f'''_j + 16\frac{\Delta x^4}{24} f^{iv}_j + 32\frac{\Delta x^5}{120} f^v_j + \dots \\ -2f_{j+1} = -2f_j - 2\Delta x f'_j - 2\frac{\Delta x^2}{2} f''_j - 2\frac{\Delta x^3}{6} f'''_j - 2\frac{\Delta x^4}{24} f^{iv}_j - 2\frac{\Delta x^5}{120} f^v_j + \dots \\ 2f_{j-1} = 2f_j - 2\Delta x f'_j + 2\frac{\Delta x^2}{2} f''_j - 2\frac{\Delta x^3}{6} f'''_j + 2\frac{\Delta x^4}{24} f^{iv}_j - 2\frac{\Delta x^5}{120} f^v_j + \dots \\ -f_{j-2} = -f_j + 2\Delta x f'_j - 4\frac{\Delta x^2}{2} f''_j + 8\frac{\Delta x^3}{6} f'''_j - 16\frac{\Delta x^4}{24} f^{iv}_j + 32\frac{\Delta x^5}{120} f^v_j + \dots \end{cases}$$

$$\Rightarrow f'''_j = \frac{f_{j+2} - 2f_{j+1} + 2f_{j-1} - f_{j-2}}{2\Delta x^3} - \frac{\Delta x^2}{4} f^v_j + \dots$$

# High-order standard centred differences - delta notation (2)

Plugging (●) in (★), one has a 5-point, 4<sup>th</sup>-order centred scheme:

$$f'_j \approx \frac{\delta \mu f_j}{\Delta x} - \frac{\Delta x^2}{6} \frac{\delta^3 \mu f_j}{\Delta x^3} = \left( \mathcal{I} - \frac{1}{6} \delta^2 \right) \frac{\delta \mu f_j}{\Delta x} = \frac{-f_{j+2} + 8f_{j+1} - 8f_{j-1} + f_{j-2}}{12\Delta x}$$

$$\left\{ \begin{array}{l} -f_{j+2} = -f_j - 2\Delta x f'_j - 4\frac{\Delta x^2}{2} f''_j - 8\frac{\Delta x^3}{6} f'''_j - 16\frac{\Delta x^4}{24} f^{iv}_j - 32\frac{\Delta x^5}{120} f^v_j - 64\frac{\Delta x^6}{720} f^{vi}_j - 128\frac{\Delta x^7}{5040} f^{vii}_j + \dots \\ 8f_{j+1} = 8f_j + 8\Delta x f'_j + 8\frac{\Delta x^2}{2} f''_j + 8\frac{\Delta x^3}{6} f'''_j + 8\frac{\Delta x^4}{24} f^{iv}_j + 8\frac{\Delta x^5}{120} f^v_j + 8\frac{\Delta x^6}{720} f^{vi}_j + 8\frac{\Delta x^7}{5040} f^{vii}_j + \dots \\ -8f_{j-1} = -8f_j + 8\Delta x f'_j - 8\frac{\Delta x^2}{2} f''_j + 8\frac{\Delta x^3}{6} f'''_j - 8\frac{\Delta x^4}{24} f^{iv}_j + 8\frac{\Delta x^5}{120} f^v_j - 8\frac{\Delta x^6}{720} f^{vi}_j + 8\frac{\Delta x^7}{5040} f^{vii}_j + \dots \\ f_{j-2} = f_j - 2\Delta x f'_j + 4\frac{\Delta x^2}{2} f''_j - 8\frac{\Delta x^3}{6} f'''_j + 16\frac{\Delta x^4}{24} f^{iv}_j - 32\frac{\Delta x^5}{120} f^v_j + 64\frac{\Delta x^6}{720} f^{vi}_j - 128\frac{\Delta x^7}{5040} f^{vii}_j + \dots \end{array} \right.$$

$$f'_j \approx \frac{\delta \mu f_j}{\Delta x} - \frac{\Delta x^2}{6} \frac{\delta^3 \mu f_j}{\Delta x^3} + \frac{\Delta x^4}{30} f^{iv}_j + \dots$$

**Recursive correction** to obtain higher orders:

$$f^{iv}_j = \frac{\partial^4 f}{\partial x^4} = \frac{\delta^4 \mu f}{\Delta x^4} + \mathcal{O}(\Delta x^2) \quad \Rightarrow \quad f'_j \approx \left( \mathcal{I} - \frac{1}{6} \delta^2 + \frac{1}{30} \delta^4 \right) \frac{\delta \mu f_j}{\Delta x} = 0 \quad \Rightarrow \quad \varepsilon = \frac{\Delta x^6}{140} f^{vii}_j + \mathcal{O}(\Delta x^8)$$

# High-order standard centred differences - delta notation (3)

**General formula** for Directional Non Compact (DNC) centred schemes of any order:

$$\frac{\partial f}{\partial x} \approx \left( \mathcal{I} - \frac{1}{6}\delta^2 + \frac{1}{30}\delta^4 - \frac{1}{140}\delta^6 + \frac{1}{630}\delta^8 + \dots \right) \frac{\delta \mu f_j}{\Delta x} = \left( \mathcal{I} - \sum_{p=0}^P (-1)^p a_p \delta^{2p+2} \right) \frac{\delta \mu f_j}{\Delta x}$$

- ▶ Approximation of order  $2(P+2)$ , dispersive at order  $2(P+2)+1$ , using  $2(P+2)+1$  points in each dir.
- ▶ Non-dissipative schemes  $\implies$  cannot damp spurious oscillations  $\implies$  need for numerical dissipation

# Fourier analysis

- ▶ On a grid with spacing  $\Delta x$  we can resolve wavenumbers  $k$  for which  $|k\Delta x| \leq \pi$
- ▶ Order of accuracy tell us what happens when  $k\Delta x \rightarrow 0$ , but how the methods handle wavenumbers that are **not so well resolved**?
- ▶ One should check the **dispersion relation** in  $0 \leq k\Delta x \leq \pi$ . How? **Fourier analysis**! It allows to:
  - Control the **resolvability limits** of the scheme on a mesh
  - Know the cutoff between well- and badly-resolved  $k$

Consider the semi-discrete linear advection  $\frac{dw_j}{dt} = -aDw_j$  and a wave  $w_j(t) = \hat{w}_j(t)e^{ikx_j} = \hat{w}_j(t)e^{ikj\Delta x}$

- ▶ **Exact representation** of first derivative:

$$\frac{\partial \hat{w}}{\partial x} = ik\hat{w}$$

- ▶ **Numerical FD approximation** of first derivative:

$$\frac{\partial w}{\partial x} = \frac{1}{\Delta x} \sum_{j=-N}^N a_j [w(x_0 + j\Delta x)]$$

$$\Rightarrow \frac{\partial \hat{w}}{\partial x} = \left( \frac{1}{\Delta x} \sum_{j=-N}^N a_j e^{ikj\Delta x} \right) \hat{w} = ik^* \hat{w}$$

- ▶ Thus one has

$$k^* = \frac{1}{i\Delta x} \sum_{j=-N}^N a_j e^{ikj\Delta x} = \frac{2}{\Delta x} \sum_{j=1}^N a_j \sin(jk\Delta x)$$

$k^*$  is the **modified wavenumber**

- It measures the accuracy with which derivatives are represented in wavenumber space
- $k^* \Delta x$  is the **reduced wavenumber**

# High-order finite-difference schemes (2): Fourier analysis

$$(\bullet) \quad k^* = \frac{2}{\Delta x} \sum_{j=1}^N a_j \sin(jk\Delta x)$$

► **Analytical solution** of semi-discrete equation:

$$\frac{dw_j}{dt} = -aDw_j \quad \Rightarrow \quad \frac{d\hat{w}}{dt} = -iak^* \hat{w}$$

$$(\dagger) \quad \hat{w}(t) = \hat{w}^0 e^{-iak^* t} = \hat{w}^0 e^{-ia\text{Re}(k^*)t} e^{a\text{Im}(k^*)t}$$

- $\text{Im}(k^*) = 0$ : constant amplitude, damping for  $<0$
- $\text{Im}(k^*) = 0$  for centred schemes (e.g.  $\bullet$ )
- $\text{Im}(k^*) \neq 0$  for upwind-biased schemes  
 $\Rightarrow$  **not ideal candidates for DNS!**

From  $(\dagger)$  with  $a = 1$ , one has

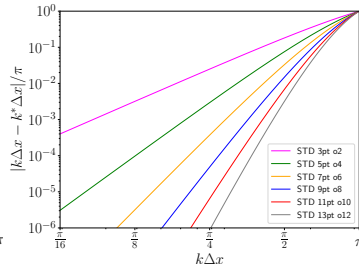
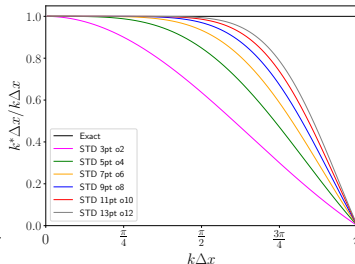
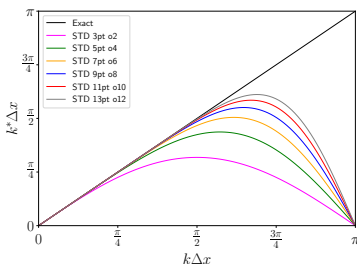
$$w(t) = \hat{w}(t) e^{ikx_j} = \hat{w}^0 e^{ikx_j - ik^* t} = \hat{w}^0 e^{i(kx_j - k^* t)}$$

Phase velocity for the  $k$  mode:

$$c_p(k) = \frac{\omega(k)}{k} = \frac{k^*}{k} = \frac{k^* \Delta x}{k \Delta x}$$

Comparisons usually shown in terms of

1.  $k^* \Delta x$  or  $\frac{k^* \Delta x}{k \Delta x}$  vs  $k \Delta x$  (**phase error**)
2.  $\frac{|k^* \Delta x - k \Delta x|}{\pi}$  vs  $k \Delta x$  (**dispersion error**)



# Optimized schemes: Dispersion Relation Preserving (DRP)

**Idea** (Tam and Webb, 1993): instead of increasing the formal order of accuracy, **minimize the dispersion error**

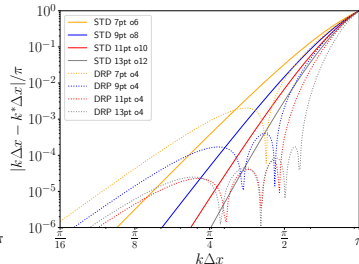
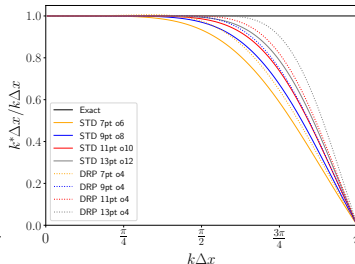
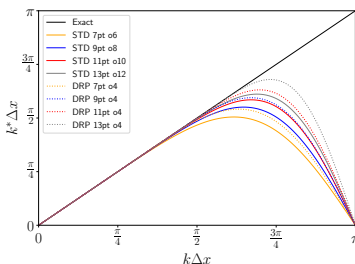
$$E = \int_{\ln(k\Delta x)_{lo}}^{\ln(k\Delta x)_{hi}} |k^* \Delta x - k \Delta x| d[\ln(k\Delta x)] \implies \frac{\partial E}{\partial a_j} = 0$$

- ▶  $(k\Delta x)_{lo}$  and  $(k\Delta x)_{hi}$  to be chosen
- ▶ To obtain an optimized scheme on  $2N + 1$  points of order  $2M$  ( $M < N$ ), one can use:
  - $M$  relations to cancel Taylor terms up to  $\Delta x^{2M-1}$
  - $M - N$  relations of type  $\partial E / \partial a_j = 0$  for  $j = 1, \dots, M - N$
  - Solve a system of  $N$  equations with  $N$  unknowns  $a_j$

**Example** (Bogey and Bailly, 2004)

Optimized scheme on 11 points and order 4:

$$\begin{cases} \sum_{j=1}^N 2ja_j = 1 \\ \sum_{j=1}^N j^3 a_j = 0 \\ \frac{\partial E}{\partial a_1} = 0 \\ \frac{\partial E}{\partial a_2} = 0 \\ \frac{\partial E}{\partial a_3} = 0 \end{cases} \quad \text{with} \quad \begin{cases} (k\Delta x)_{lo} = \frac{\pi}{16} \\ (k\Delta x)_{hi} = \frac{\pi}{2} \end{cases}$$





# Compact directional approximations (I)

- Standard way to obtain high-order accuracy is **adding more points**, but large stencils can be obtained (problems on the boundaries)
- Alternative: **compact schemes**. Starting from Taylor series expansion:

$$f_{j+1} = f_j + \Delta x f'_j + \frac{\Delta x^2}{2} f''_j + \frac{\Delta x^3}{6} f'''_j + \frac{\Delta x^4}{24} f^{iv}_j + \mathcal{O}(\Delta x^5) \quad (1)$$

$$f_{j-1} = f_j - \Delta x f'_j + \frac{\Delta x^2}{2} f''_j - \frac{\Delta x^3}{6} f'''_j + \frac{\Delta x^4}{24} f^{iv}_j + \mathcal{O}(\Delta x^5) \quad (2)$$

## First derivative

1. Adding (1) + (2) and taking first derivative:

$$f'_{j+1} + f'_{j-1} = 2f'_j + \Delta x^2 f'''_j + \frac{\Delta x^4}{12} f^{v}_j + \mathcal{O}(\Delta x^6)$$

2. Subtracting (2) from (1):

$$f_{j+1} - f_{j-1} = 2\Delta x f'_j + 2\frac{\Delta x^3}{6} f'''_j + \mathcal{O}(\Delta x^5)$$

3. Eliminating  $f'''_j$ :

$$f'_{j+1} + 4f'_j + f'_{j-1} = \frac{3}{\Delta x} (f_{j+1} - f_{j-1}) + \mathcal{O}(\Delta x^4)$$

Find  $f'_j$  and  $f''_j$  **solving a tridiagonal system**

## Second derivative

1. Adding (1) + (2):

$$f_{j+1} + f_{j-1} = 2f_j + \Delta x^2 f''_j + \frac{\Delta x^4}{12} f^{iv}_j + \mathcal{O}(\Delta x^6)$$

2. Taking second derivative:

$$f''_{j+1} + f''_{j-1} = 2f''_j + \Delta x^2 f^{iv}_j + \frac{\Delta x^4}{12} f^{vi}_j + \mathcal{O}(\Delta x^6)$$

3. Eliminating  $f^{iv}_j$ :

$$f''_{j+1} + 10f''_j + f''_{j-1} = \frac{12}{\Delta x^2} (f_{j+1} - 2f_j + f_{j-1}) + \mathcal{O}(\Delta x^4)$$

# Compact directional approximations (II)

From Lele (1992):  $\beta f'_{i-2} + \alpha f'_{i-1} + f'_i + \alpha f'_{i+1} + \beta f'_{i+2} = a \frac{f_{i+1} - f_{i-1}}{2\Delta x} + b \frac{f_{i+2} - f_{i-2}}{4\Delta x} + c \frac{f_{i+3} - f_{i-3}}{6\Delta x}$

The terms of the Taylor expansion are canceled:

$$a + b + c = 1 + 2\alpha + 2\beta \quad (\text{order 2})$$

$$a + 2^2 b + 3^2 c = 2 \frac{3!}{2!} (\alpha + 2^2 \beta) \quad (\text{order 4})$$

$$a + 2^4 b + 3^4 c = 2 \frac{5!}{4!} (\alpha + 2^4 \beta) \quad (\text{order 6})$$

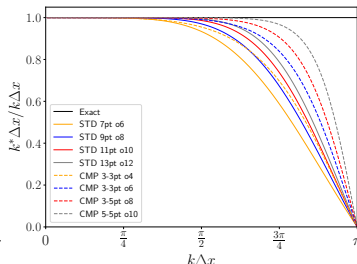
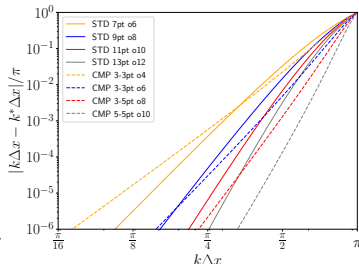
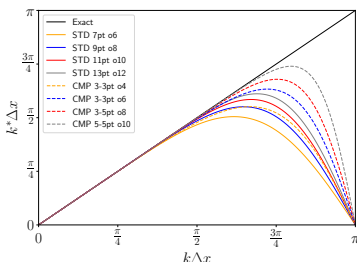
$$a + 2^6 b + 3^6 c = 2 \frac{7!}{6!} (\alpha + 2^6 \beta) \quad (\text{order 8})$$

$$a + 2^8 b + 3^8 c = 2 \frac{9!}{8!} (\alpha + 2^8 \beta) \quad (\text{order 10})$$

**Modified wavenumber:**

$$k^* \Delta x = \frac{a \sin(k\Delta x) + (b/2) \sin(2k\Delta x) + (c/3) \sin(3k\Delta x)}{1 + 2\alpha \cos(k\Delta x) + 2\beta \cos(2k\Delta x)}$$

- ▶ Standard schemes retrieved for  $\alpha = \beta = 0$
- ▶ Discretization stencil more compact
- ▶ Pentadiagonal ( $\beta \neq 0$ ) or tridiagonal ( $\beta = 0$ ) system has to be solved



# Spectral analysis for second derivative

The same analysis may be performed for **second derivatives**. Following the same study, one has

► **Exact** and **numerical** representation of second derivative:

$$\frac{\widehat{\partial^2 w}}{\partial x^2} = -k^2 \widehat{w} \quad \text{and} \quad \frac{\widehat{\partial^2 w}}{\partial x^2} = \left[ \frac{1}{(\Delta x)^2} \sum_{j=-N}^N a_j e^{ikj\Delta x} \right] \widehat{w} = -k^{*2} \widehat{w} \quad \Rightarrow \quad k^{*2} = -\frac{1}{(\Delta x)^2} \sum_{j=-N}^N a_j e^{ikj\Delta x}$$

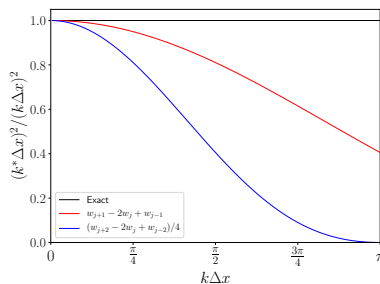
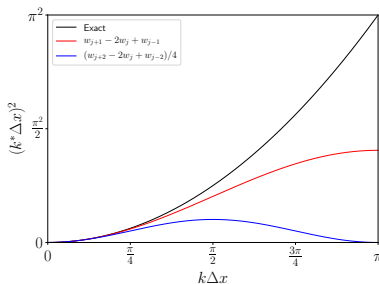
Consider 2 approximations, obtained by Taylor series and successive applications of 1<sup>st</sup> der., respectively:

$$\frac{\partial^2 w_j}{\partial x^2} = \frac{\delta^2 w_j}{\Delta x^2} = \frac{w_{j+1} - 2w_j + w_{j-1}}{\Delta x^2}$$

$$\frac{\partial^2 w_j}{\partial x^2} = \frac{\partial}{\partial x} \left[ \frac{\partial w_j}{\partial x} \right] = \frac{(\delta \mu)^2 w_j}{\Delta x^2} = \frac{w_{j+2} - 2w_j + w_{j-2}}{4\Delta x^2}$$

$$\frac{\widehat{\partial^2 w}}{\partial x^2} = 2[1 - \cos(k\Delta x)]$$

$$\frac{\widehat{\partial^2 w}}{\partial x^2} = \sin^2(k\Delta x)$$



- Both 2<sup>nd</sup>-order accurate, but different spectral responses!
  - For the latter, effective diff. coeff.  $\rightarrow 0$  for  $k\Delta x \rightarrow \pi$ 
    - Does not stabilize the solution
    - Effective dissipation  $\rightarrow 0$  whatever the  $\nu_{sgs}$  model is
- Not acceptable for LES!

# High-order dissipation term

► Appropriate values for  $a_m$  and  $b_l$  allow to:

- **Maximize formal accuracy** (the minimum truncation error being  $\mathcal{O}(\Delta x^{2(L+M)})$ )
- **Shape the spectral response** of the scheme and improve the representation of the Fourier modes with the highest wavenumbers supported by the grid

$$\sum_{m=-M}^M a_m Df_{j+m} = \frac{1}{\Delta x} \sum_{l=-L}^L b_l f_{j+l}$$

► Centred FD for  $b_{-l}=b_l$  and  $a_{-m}=a_m \implies$  null dissipation error in linear setting

- Grid-to-grid oscillations or **wiggles** (every 2 points, i.e.  $k\Delta x = \pi$ ) are not resolved by centred FD  
 $\hookrightarrow$  Can appear near stiff velocity gradients or discontinuities (such as BCs) and contaminate the solution

## How to control them?

1. **Selective filtering**: use of a centred (thus non dispersive) filter to dissipate only high-frequencies (Gaitonde and Visbal, 2000; Bogey and Bailly, 2004):

$$f^{\text{filtered}}(x_0) = f(x_0) - \sigma_d D_f(x_0) \quad \text{with} \quad 0 \leq \sigma_d \leq 1 \quad \text{and} \quad D_f(x_0) = \sum_{j=-N}^N d_j f(x_0 + j\Delta x)$$

$\underbrace{\quad}_{\text{centred approx.}} \quad - \quad \underbrace{\quad}_{\text{dissipative term}}$

2. **Artificial dissipation** (Jameson et al., 1981; Kim and Lee, 2001):  $\frac{\partial w}{\partial t} + \frac{\delta F}{\Delta x} = 0$  with  $F = \underbrace{\quad}_{\text{centred approx.}} - \underbrace{\quad}_{\text{dissipative term}}$
3. **Skew-symmetric formulations**: Employ **energy-consistent schemes** using a skew-symmetric splitting for the convective term (Blaisdell et al., 1996; Ducros et al., 2000; Honein and Moin, 2004; Pirozzoli, 2011)

$$\frac{\partial(fu_j)}{\partial x_j} = \frac{1}{2} \frac{\partial(fu_j)}{\partial x_j} + \frac{1}{2} u_j \frac{\partial f}{\partial x_j} + \frac{1}{2} f \frac{\partial u_j}{\partial x_j}$$

# High-order selective standard filters

The terms of the Taylor expansion are canceled until the order  $\Delta x^{2N-1}$ :

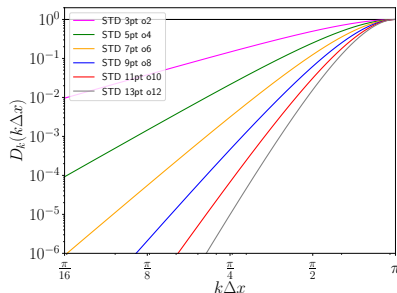
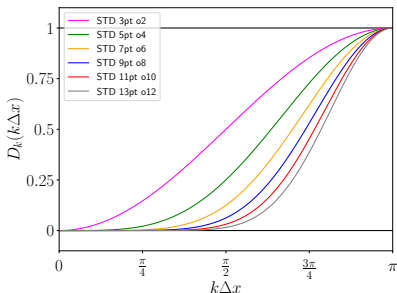
$$\begin{aligned}
 D_f(x_0) &= d_0 f(x_0) + \sum_{j=1}^N d_j [f(x_0 + j\Delta x) + f(x_0 - j\Delta x)] \\
 &= d_0 f(x_0) + \sum_{j=1}^N d_j \left[ 2f(x_0) + j^2 \Delta x^2 f''(x_0) + \frac{2j^4 \Delta x^4}{4!} f^{(4)}(x_0) + \frac{2j^6 \Delta x^6}{6!} f^{(6)}(x_0) + \dots \right]
 \end{aligned}$$

$$\left\{ \begin{array}{l} d_0 + 2\sum_{j=1}^N d_j = 0 \\ \sum_{j=1}^N j^2 d_j = 0 \\ \vdots \\ \sum_{j=1}^N j^{2N-2} d_j = 0 \end{array} \right. \quad \begin{array}{l} N \text{ relations + in Fourier's space:} \\ \blacktriangleright \left( D_k(0) = 0 \implies d_0 + 2\sum_{j=1}^N d_j = 0 \right) \\ \text{(Redundant, condition already used)} \\ \blacktriangleright D_k(\pi) = 1 \implies d_0 + 2\sum_{j=1}^N (-1)^j d_j = 1 \end{array}$$

**Damping function** of centred exp. flt.:

$$D_k(k\Delta x) = d_0 + \sum_{j=1}^N 2d_j \cos(jk\Delta x)$$

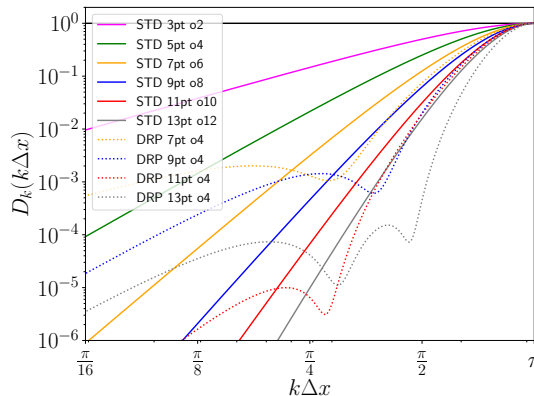
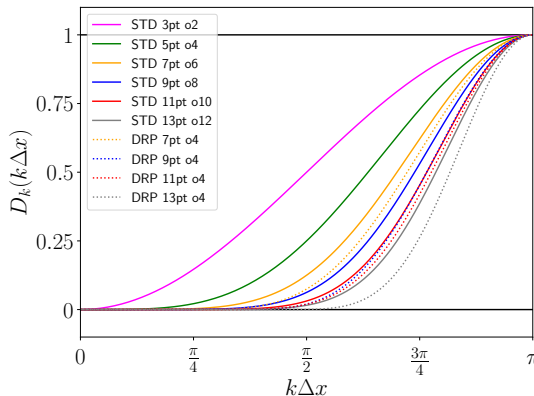
**$2N + 1$  points,  $N + 1$  relations  
order  $2N$**



# High-order selective filters (2): optimized filters

Optimized filters: minimization of the dissipation (Bogey and Bailly, 2004)

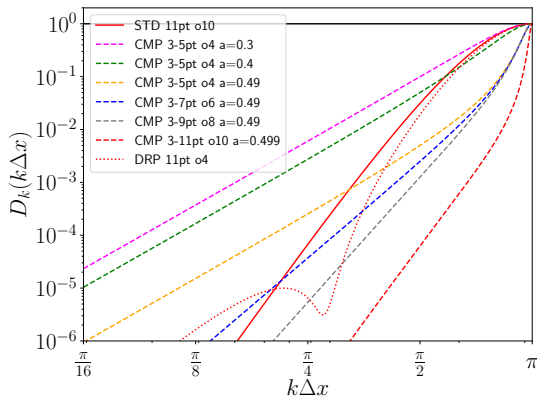
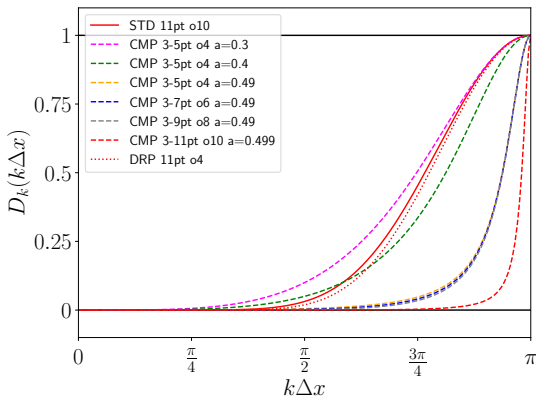
$$E = \int_{\ln(\pi/16)}^{\ln(\pi/2)} D_k(k\Delta x) d[\ln(k\Delta x)]$$



# High-order selective filters (3): compact filters

$$\beta f_{i-2}^f + \alpha f_{i-1}^f + f_i^f + \alpha f_{i+1}^f + \beta f_{i+2}^f = a f_i + \frac{b}{2} (f_{i+1} + f_{i-1}) + \frac{c}{2} (f_{i+2} + f_{i-2}) + \frac{d}{2} (f_{i+3} + f_{i-3})$$

**Damping function:** 
$$D_k(k\Delta x) = \frac{a + b \cos(k\Delta x) + c \cos(2k\Delta x) + d \cos(3k\Delta x)}{1 + 2\alpha \cos(k\Delta x) + 2\beta \cos(2k\Delta x)}$$

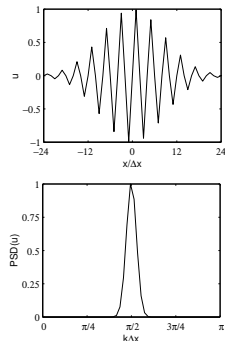


- Tridiagonal filters on  $2N + 1$  points of order  $2N$  (Gaitonde and Visbal, 2000):  $\beta = 0$ ,  $0.3 \leq \alpha_f < 0.5$
- Optimized pentadiagonal filter (Lele, 1992), 6th-order + 2 constraints:  $\frac{d^2 D_k(\pi)}{d(k\Delta x)^2} = 0$  and  $\frac{d^4 D_k(\pi)}{d(k\Delta x)^4} = 0$

# Test case: Solution of the advection equation

From Bogey and Bailly (2004):

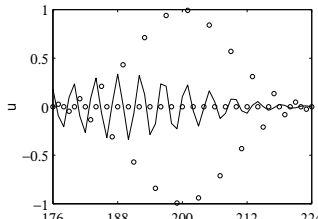
- ▶ Advection equation:  $\frac{\partial u}{\partial t} + a \frac{\partial u}{\partial x} = 0$  with  $a = 1$  and  $\Delta t = \text{CFL} \frac{\Delta x}{a}$
- ▶ Initial perturbation:  $u(x) = \sin\left(\frac{2\pi x}{4\Delta x}\right) \exp\left(-\ln 2 \left(\frac{x}{9\Delta x}\right)^2\right)$ 
  - principal wavenumber for  $k\Delta x = \frac{\pi}{2}$  ( $\lambda_0 = 4\Delta x$ )
- ▶ Propagation over a large distance:  $200\Delta x = 50\lambda_0$
- ▶ Numerical error:  $e_{\text{num}} = \left(\sum (u_{\text{calc}} - u_{\text{exact}})^2 / \sum u_{\text{exact}}^2\right)^{1/2}$



Solutions obtained with **optimized Runge-Kutta scheme with 6 substeps** and:

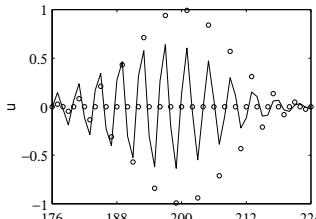
**opt. 9 pts FD + opt. 9 pts SF**

$$\Rightarrow e_{\text{num}} = 0.905$$



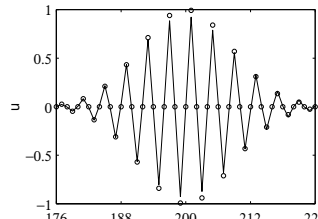
**opt. 11 pts FD + opt. 11 pts SF**

$$\Rightarrow e_{\text{num}} = 0.488$$



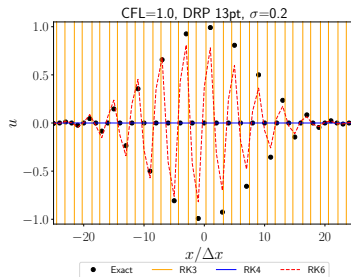
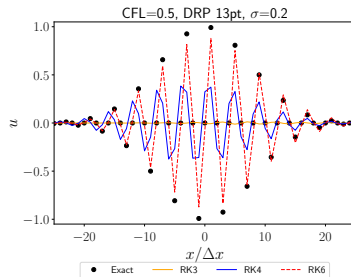
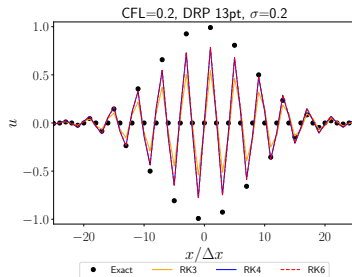
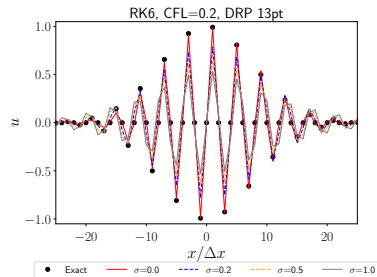
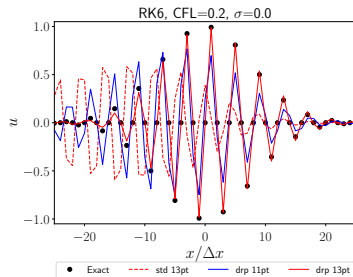
**opt. 13 pts FD + opt. 13 pts SF**

$$\Rightarrow e_{\text{num}} = 0.077$$





# Test case: Solution of the advection equation (2)



# Artificial Dissipation: DNC schemes

**Dissipative DNC schemes** may be constructed via an **upwind recursive correction** of the truncation error using the flux of a first-order (dissipative) scheme:

$$\frac{dw_j}{dt} + \frac{\delta F}{\Delta x} = 0 \quad \text{with} \quad F = \underbrace{H}_{\text{central approx}} - \underbrace{D}_{\text{dissipative term}} = \mu f - \frac{1}{2} Q \delta w \quad Q \text{ dissipation matrix}$$

By doing that, one loses one order but **introduce dissipation**. General formula for dissipative DNC schemes:

$$\frac{dw_j}{dt} + \left( \mathcal{I} - \sum_{p=0}^P (-1)^p a_p \delta^{2p+2} \right) \frac{\delta \mu f_j}{\Delta x} = \frac{\delta D}{\Delta x} = \left( (-1)^{P+1} \frac{a_P}{2} |Q| \delta^{2P+3} \right) \frac{\delta w}{\Delta x}$$

► Scheme of order  $2P + 3$ , using  $2(P + 2) + 1$  points in each direction. Examples for  $P = 0, 1, 2, 3$ :

$$F_{j+\frac{1}{2}} = \left[ \left( \mathcal{I} - \frac{1}{6} \delta^2 \right) \mu f + \frac{1}{12} |Q| \delta^3 w \right]_{j+\frac{1}{2}} \quad (\text{order } 3, P=0)$$

$$F_{j+\frac{1}{2}} = \left[ \left( \mathcal{I} - \frac{1}{6} \delta^2 + \frac{1}{30} \delta^4 \right) \mu f - \frac{1}{60} |Q| \delta^5 w \right]_{j+\frac{1}{2}} \quad (\text{order } 5, P=1)$$

$$F_{j+\frac{1}{2}} = \left[ \left( \mathcal{I} - \frac{1}{6} \delta^2 + \frac{1}{30} \delta^4 - \frac{1}{140} \delta^6 \right) \mu f + \frac{1}{280} |Q| \delta^7 w \right]_{j+\frac{1}{2}} \quad (\text{order } 7, P=2)$$

$$F_{j+\frac{1}{2}} = \left[ \left( \mathcal{I} - \frac{1}{6} \delta^2 + \frac{1}{30} \delta^4 - \frac{1}{140} \delta^6 + \frac{1}{630} \delta^8 \right) \mu f - \frac{1}{1260} |Q| \delta^9 w \right]_{j+\frac{1}{2}} \quad (\text{order } 9, P=3)$$

# Energy-consistent schemes: conservation properties

$$(\star) \quad \frac{\partial w}{\partial t} + \frac{\partial F(w)}{\partial x} = \frac{\partial w}{\partial t} + a(w) \frac{\partial w}{\partial x} = \nu \frac{\partial^2 w}{\partial x^2} \quad \text{with } F \text{ vanishing at } x \rightarrow \pm\infty$$

**Invariants** in the inviscid limit:

## 1. Primary conservation property:

$$\int_{-\infty}^{+\infty} (\star) dx \implies \frac{d \int w dx}{dt} = 0 \implies w \text{ conserved}$$

- **Not affected** by discontinuous  $w$  or finite viscosity

## 2. Secondary conservation property:

$$\int_{-\infty}^{+\infty} w \times (\star) dx \implies \frac{d \int \frac{w^2}{2} dx}{dt} = 0 \implies \frac{w^2}{2} \text{ conserved}$$

- **Destroyed** by discontinuous  $w$  or finite viscosity

- If the numerical discretization can be recast in the locally conservative form:

$$\frac{dw_j}{dt} = -\frac{1}{\Delta x} \left( \hat{f}_{j+1/2} - \hat{f}_{j-1/2} \right)$$

- The primary conservation property holds
- The numerical flux respects the **telescopic property**, i.e. at the semi-discrete level:

$$\frac{d}{dt} \sum_j w_j = 0$$

- Similar considerations for the second invariant:

$$\frac{d \frac{w_j^2}{2}}{dt} = -\frac{1}{\Delta x} \left( \hat{g}_{j+1/2} - \hat{g}_{j-1/2} \right)$$

# Conservation of Kinetic energy (I)

Consider the semi-discrete inviscid Burgers equation  $\frac{\partial w}{\partial t} + w \frac{\partial w}{\partial x} = 0$  in advective and conservative form:

$$\frac{\partial w}{\partial t} = -w \frac{\partial w}{\partial x} \approx -w_j \frac{(w_{j+1} - w_{j-1}))}{2\Delta x}$$

$$\frac{\partial w}{\partial t} = -\frac{1}{2} \frac{\partial w^2}{\partial x} \approx -\frac{1}{2} \frac{(w_{j+1}^2 - w_{j-1}^2)}{2\Delta x}$$

► Both can be cast in locally conservative form, by defining:  $\hat{f}_{j+1/2} = \frac{w_j w_{j+1}}{2}$  and  $\hat{f}_{j+1/2} = \frac{w_j^2 + w_{j+1}^2}{4}$

⇒ **Both** satisfy the primary conservation property ( $w$  conserved)

► But **neither of them** satisfy the second conservation property ( $w^2$  not conserved)

- It is not possible to identify a locally conservative flux form. One has indeed that:

$$\begin{aligned} \frac{d}{dt} \int \frac{w_j^2}{2} dx &= \sum w_j \left[ -w \frac{\partial w}{\partial x} \right]_j \Delta x \\ &= \sum \left[ w_j^2 (w_{j-1} - w_{j+1}) \right] \\ &= .. + w_{j-1}^2 (w_{j-2} - w_j) \\ &\quad + w_j^2 (w_{j-1} - w_{j+1}) + .. \neq 0 \end{aligned}$$

$$\begin{aligned} \frac{d}{dt} \int \frac{w_j^2}{2} dx &= \sum w_j \left[ -\frac{1}{2} \frac{\partial w^2}{\partial x} \right]_j \Delta x \\ &= \sum \left[ \frac{w_j}{2} (w_{j-1}^2 - w_{j+1}^2) \right] \\ &= .. + \frac{w_{j-1}}{2} (w_{j-2}^2 - w_j^2) \\ &\quad + \frac{w_j}{2} (w_{j-1}^2 - w_{j+1}^2) + .. \neq 0 \end{aligned}$$

► And if we consider a **linear combination** of the two approximations?

## Conservation of Kinetic energy (II)

Combine both schemes and find the value of  $\alpha$  for which  $w^2$  is preserved:

$$\begin{aligned} \frac{dw_j}{dt} &= -\frac{\alpha}{2\Delta x} \alpha w_j (w_{j+1} - w_{j-1}) - \frac{1-\alpha}{4\Delta x} (w_{j+1}^2 - w_{j-1}^2) \\ \frac{d}{dt} \int \frac{w_j^2}{2} dx &= -\frac{1}{2} \sum \left[ \alpha w_j^2 (w_{j+1} - w_{j-1}) + \frac{1-\alpha}{2} w_j (w_{j+1}^2 - w_{j-1}^2) \right] \\ &= .. + \alpha w_{j-1}^2 (w_j - w_{j-2}) + \frac{1-\alpha}{2} w_{j-1} (w_j^2 - w_{j-2}^2) \\ &\quad + \alpha w_j^2 (w_{j+1} - w_{j-1}) + \frac{1-\alpha}{2} w_j (w_{j+1}^2 - w_{j-1}^2) + .. = 0 \\ &= .. + \alpha (\textcolor{red}{w}_{j-1}^2 w_j - w_{j-1}^2 w_{j-2}) + \frac{1-\alpha}{2} (w_{j-1} \textcolor{blue}{w}_j^2 - w_{j-1} w_{j-2}^2) \\ &\quad + \alpha (w_j^2 w_{j+1} - \textcolor{blue}{w}_j^2 w_{j-1}) + \frac{1-\alpha}{2} (w_j w_{j+1}^2 - \textcolor{red}{w}_j w_{j-1}^2) + .. = 0 \implies \alpha + \frac{1-\alpha}{2} = 0 \implies \boxed{\alpha = \frac{1}{3}} \\ \frac{dw_j}{dt} &= -\frac{1}{6\Delta x} \left[ w_j (w_{j+1} - w_{j-1}) - (w_{j+1}^2 - w_{j-1}^2) \right] = \left[ -\frac{1}{2\Delta x} \left[ \frac{w_{j+1} + w_j + w_{j-1}}{3} (w_{j+1} - w_{j-1}) \right] \right] \end{aligned}$$

This conserves **both**  $w$  and  $w^2$ !

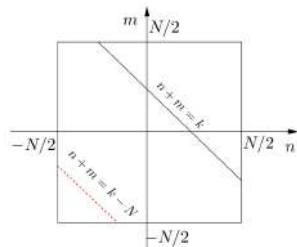
- A numerical flux can be written:  $\hat{g}_{j+1/2} = \frac{w_j^2 + w_j w_{j+1} + w_{j+1}^2}{6}$  such that the telescopic property holds
- **KE preservation**  $\implies$  the numerical solution cannot diverge in finite time
- This is valid also for a **generic central difference operator**  $\frac{dw_j}{dt} = -\alpha w_j Dw_j - \frac{1-\alpha}{2} (Dw^2)_j$

# Aliasing and nonlinear stability

**Aliasing** arises whenever two functions are multiplied on a discrete grid in physical space. Consider the Fourier expansions of  $u$  and  $v$

$$u_j = \sum_{n=-N/2}^{N/2-1} \hat{u}_n e^{i(2\pi j/N)n} \quad v_j = \sum_{m=-N/2}^{N/2-1} \hat{v}_m e^{i(2\pi j/N)m}$$

Their pointwise product is  $\hat{w}_k = \sum_{n+m=k} \hat{u}_n \hat{v}_m + \sum_{n+m=k \pm N} \hat{u}_n \hat{v}_m$



- ▶ Generation of high-freq modes that cannot be resolved on the mesh  $\Rightarrow$  **aliased** to lower-frequency modes
  - The error is large only when the modes near the highest resolvable wavenumber carry significant energy (case for LES and partially DNS)
- ▶ **Aliasing errors modified by truncation errors** (i.e., Fourier modes multiplied by modified wavenumbers)
  - Spectral methods: de-aliasing techniques mandatory
  - FD:  $k^*$  decrease at high  $k \Rightarrow$  problem mitigated

- ▶ Consider two approximations:

$$N_1 = \frac{duv}{dx} \quad N_2 = u \frac{dv}{dx} + v \frac{du}{dx}$$

- It can be shown that aliasing errors of  $N_1$  and  $N_2$  are of opposite sign!
- $\Rightarrow$  **Skew-symmetric form:**

$$N_3 = \frac{N_1 + N_2}{2} = \frac{1}{2} \frac{duv}{dx} + \frac{1}{2} u \frac{dv}{dx} + \frac{1}{2} v \frac{du}{dx}$$

well-behaved even without de-aliasing!

- Aliasing errors and KE preservation intimately related!

# Skew-symmetric form for compressible flows

$$(I) \quad \frac{\partial \rho u_i \varphi}{\partial x_i} = \frac{1}{2} \frac{\partial \rho u_i \varphi}{\partial x_i} + \frac{1}{2} \varphi \frac{\partial \rho u_i}{\partial x_i} + \frac{1}{2} \rho u_i \frac{\partial \varphi}{\partial x_i}$$

$$(II) \quad \frac{\partial \rho u_i \varphi}{\partial x_i} = \frac{1}{2} \frac{\partial \rho u_i \varphi}{\partial x_i} + \frac{1}{2} u_i \frac{\partial \rho \varphi}{\partial x_i} + \frac{1}{2} \rho \varphi \frac{\partial u_i}{\partial x_i}$$

$$(III) \quad \frac{\partial \rho u_i \varphi}{\partial x_i} = \alpha \frac{\partial \rho u_i \varphi}{\partial x_i} + \beta \left[ u_i \frac{\partial \rho \varphi}{\partial x_i} + \rho \frac{\partial u_i \varphi}{\partial x_i} + \varphi \frac{\partial \rho u_i}{\partial x_i} \right] \\ + (1 - 2\alpha - 2\beta) \left[ \rho u_i \frac{\partial \varphi}{\partial x_i} + \rho \varphi \frac{\partial u_i}{\partial x_i} + u_i \varphi \frac{\partial \rho}{\partial x_i} \right]$$

(I) semidiscrete KE preservation (Honein and Moin, 2004)

(II) minimization of the aliasing error (Blaisdell et al., 1996)

(III) semidiscrete KE preservation for  $\alpha = \beta = \frac{1}{4}$  (Kennedy and Gruber, 2008) and additional robustness

✓ All formulation yield locally conservative schemes with explicit central formulas (Pirozzoli, 2010)

✗ Recall that discrete energy conservation only applies for **smooth solutions**

- Energy is dissipated in the presence of shocks

⇒ Applications to shocked flows may yield **stable but unphysical** numerical solutions

## 1 Introduction

## 2 Methods for smooth flows

- High-order centred derivatives
- Stabilization for smooth flows
- Energy-consistent schemes

## 3 Methods for non smooth flows



# Shock-capturing schemes

- ▶ Spurious Gibbs oscillations near shock jumps may lead to nonlinear instabilities. Two strategies available:
  1. **Shock-fitting approaches:** shock are genuine discontinuities, governed by their own set of algebraic equation and uses RH relation as BCs on the two sides
    - ✓ Very accurate
    - ✗ Only feasible for steady shocks and simple geometries
  2. **Shock-capturing approaches:** use the same discretization everywhere and achieve regularization by addition of numerical dissipation.
    - 2.1 **Hybrid Schemes** (WENO + centred differences): switch based on a smoothness sensor
    - 2.2 **Nonlinear Selective Filtering:** add a low-order filter
    - 2.3 **Artificial viscosity:** add a low-order artificial flux or artificial transport properties
    - 2.4 **Flux limiters:** well-suited for strong shocks, but costly and convergence issues

## Properties of shock-capturing schemes:

- ▶ Analysis difficult because of their **inherent nonlinearity**
  - Classical Fourier analysis cannot be applied generally
  - Comparison possible only on a case-by-case basis
- ▶ Major flaw: **reduction of accuracy near shocks!**  
Even high-order schemes yield first-order accurate solutions downstream of moving shocks
- ▶ Paramount importance of the **shock-detector function**

# Weighted Essentially Non-Oscillatory (WENO) Methods

**Idea:** determine the numerical flux from a high-order reconstruction over an adaptive stencil that is selected to avoid as much interpolation across discontinuities as possible.

- Construct the flux by convex linear combination of lower-order polynomial reconstructions, with weights selected to achieve maximum formal order of accuracy in smooth regions.

**Example:** WENO3 ( $L = 3$ )

$L$  points  $\rightarrow L + 1$  substencils:  $\hat{f}_{j+\frac{1}{2}} = \sum_{l=0}^L \omega_l \hat{f}_{j+\frac{1}{2}}^l$

with  $\hat{f}_{j+\frac{1}{2}}^l$  the numerical flux resulting from polynomial

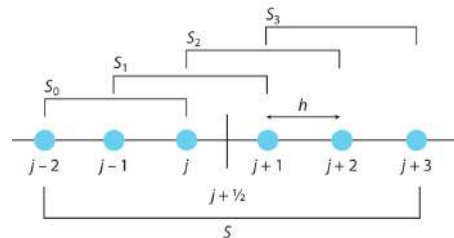
reconstruction over the stencil  $S_l$ :  $\hat{f}_{j+\frac{1}{2}}^l = \sum_{m=0}^{L-1} c_{lm} f_{j-L+1+l+m}$   
and the weights defined as

$$\omega_l = \frac{\alpha_l}{\sum_{m=0}^L \alpha_m} \quad \alpha_l = \frac{d_l}{(\varepsilon + \beta_l)^2}$$

are functions of the smoothness measurements associated with the substencils

$$\beta_l = \sum_{m=1}^{L-1} \left[ \sum_{n=1}^{L-1} \gamma_{lmn} f_{j-L+1+l+n} \right]^2$$

- Choose  $d_l$  s.t. maximum order of accuracy ( $2L$ ) is obtained
- Choose  $\omega_l$  s.t. it nullifies if  $S_l$  contains a jump



- ✓ Well established, very robust method
- ✓ No tuning parameters
- ✗ Highly dissipative  
⇒ hybridization with central schemes
- ✗ Characteristic variable transform needed  
⇒ high computational cost

# Artificial viscosity: Jameson's approach

From Jameson et al. (1981); Kim and Lee (2001):

- **Explicit addition** of a dissipative low-order term
- Starting from the dissipative flux of DNC schemes:

$$D_{j+\frac{1}{2}} = \rho(A)_{j+\frac{1}{2}} \left( \varepsilon_2 \delta w - \varepsilon_4 \delta^3 w \right)_{j+\frac{1}{2}}$$

$$\varepsilon_{2,j+\frac{1}{2}} = k_2 \max(\varphi_j, \varphi_{j+1}) \quad \varepsilon_{4,j+\frac{1}{2}} = \max(0, k_4 - \varepsilon_{2,j+\frac{1}{2}})$$

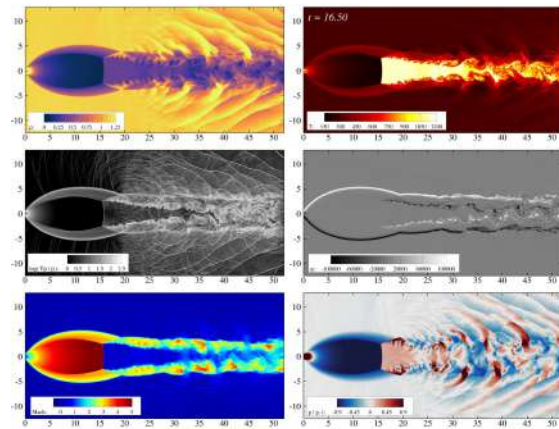
with  $\varphi_j \propto \left| \frac{\partial^2 \phi_j}{\partial x^2} \right|$  a scalar discontinuity sensor; e.g.,

$$\text{Jameson's pressure sensor: } \varphi_j = \left| \frac{p_{j+1} - 2p_j + p_{j-1}}{p_{j+1} + 2p_j + p_{j-1}} \right|$$

- The nonlinear term is  $\mathcal{O}(\Delta x^3)$  in smooth regions and becomes  $\mathcal{O}(\Delta x)$  close to discontinuities
- Straightforward extensions to higher orders by suitable modification of the term  $\varepsilon_4 \delta^3 w$
- Other variables or combinations of variables can be used for the discontinuity sensor

## Example: DNC-Jameson 9

- 2D underexpanded  $N_2 - O_2$  jet
- Nozzle Pressure Ratio  $NPR = 15$
- $L_x \times L_y = 50D \times 25D$ ,  $\Delta x = 0.25 \text{ mm}$ ,  $16 \cdot 10^6$  pts



# Sensor Example: DNC9

$$\varphi_j = \underbrace{\frac{1}{2} \left[ 1 - \tanh \left( 2.5 + 10 \frac{\Delta x}{a} \nabla \cdot \mathbf{u} \right) \right]}_{\text{(I) Bhagatwala \& Lele}} \times \underbrace{\frac{(\nabla \cdot \mathbf{u})^2}{(\nabla \cdot \mathbf{u})^2 + |\nabla \times \mathbf{u}|^2 + \epsilon}}_{\text{(II) Ducros}} \times \underbrace{\left| \frac{p_{j+1} - 2p_j + p_{j-1}}{p_{j+1} + 2p_j + p_{j-1}} \right|}_{\text{(III) Jameson}}$$

(I):

Excessive damping of acoustic waves

(II):

Always active in solenoidal regions

(III):

Damping of turbulent motions

(I)+(II)+(III):

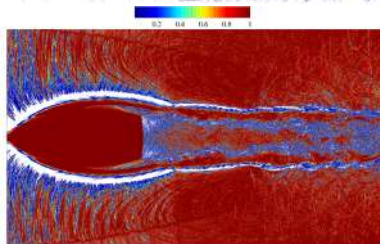
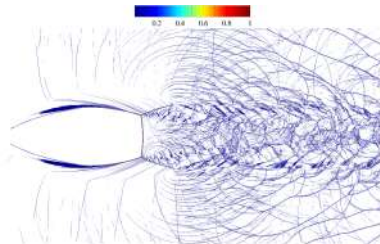
**Highly-selective combination**

✓ Low computational cost

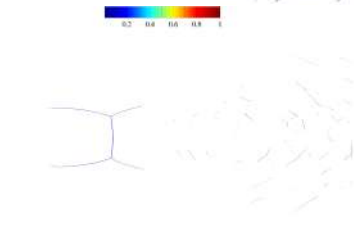
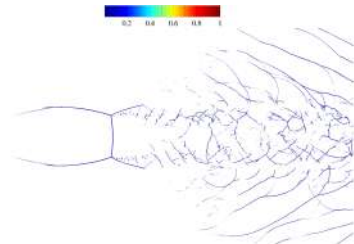
✓ Few tuning parameters

✗ Large stencils needed for high orders

✗ Not very robust



Top: I, Bottom: II



Top: III, Bottom: I+II+III

# Localized Artificial Diffusivity (LAD)

► **Idea:** Add artificial transport properties to regularize equations (Kawai et al., 2010)

- $\mu^*$  for unresolved sgs eddies
- $\beta^*$  for shock waves
- $\kappa^*$  for contact discontinuities

$$\mu^* = C_\mu \rho \left| \frac{\partial^r F_\mu}{\partial x^r} (\Delta x)^2 \right| D_\mu^2$$

$$\beta^* = C_\beta \rho f_{sw} \left| \frac{\partial^r F_\beta}{\partial x^r} \left[ \Delta x \frac{\nabla \rho}{|\nabla \rho|} \right]^2 \right| D_\beta^2$$

$$\kappa^* = C_\kappa \frac{\rho a}{T} \left| \frac{\partial^r F_\kappa}{\partial x^r} \Delta x \frac{\nabla e}{|\nabla e|} \right|$$

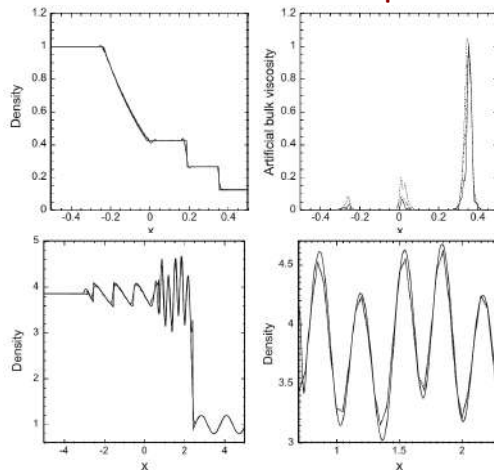
with

$$F_\mu = |S_{ij}|, \quad F_\kappa = e = \frac{p}{(\gamma - 1)\rho}, \quad F_\beta = \nabla \cdot \mathbf{u}$$

$$f_{sw} = H(-\nabla \cdot \mathbf{u}) \cdot \frac{(\nabla \cdot \mathbf{u})^2}{(\nabla \cdot \mathbf{u})^2 + |\nabla \times \mathbf{u}|^2 + \epsilon}$$

►  $D_\mu, D_\beta$  wall-damping functions

## Sod shock tube and Shu-Osher problem



- ✓ Low computational cost (not as cheap as Central)
- ✓ Spectral resolution achieved with compact FD
- ✗ Several tuning parameters
- ✗ High diffusivity reduce  $(\Delta t)_{stab}$  on stretched grids

# References I

- Blaisdell, G., Spyropoulos, E., and Qin, J. (1996). The effect of the formulation of nonlinear terms on aliasing errors in spectral methods. *Applied Numerical Mathematics*, 21(3):207–219.
- Bogey, C. and Bailly, C. (2004). A family of low dispersive and low dissipative explicit schemes for flow and noise computations. *Journal of Computational Physics*, 194(1):194–214.
- Ducros, F., Laporte, F., Soulères, T., Guinot, V., Moinat, P., and Caruelle, B. (2000). High-order fluxes for conservative skew-symmetric-like schemes in structured meshes: application to compressible flows. *Journal of Computational Physics*, 161(1):114–139.
- Gaitonde, D. V. and Visbal, M. R. (2000). Padé-type higher-order boundary filters for the Navier-Stokes equations. *AIAA journal*, 38(11):2103–2112.
- Harten, A. (1983). On the symmetric form of systems of conservation laws with entropy.
- Honein, A. E. and Moin, P. (2004). Higher entropy conservation and numerical stability of compressible turbulence simulations. *Journal of Computational Physics*, 201(2):531–545.
- Jameson, A., Schmidt, W., and Turkel, E. (1981). Numerical solutions of the Euler equations by finite volume methods using Runge-Kutta time stepping schemes. *AIAA Journal*, 81(1259).
- Kawai, S., Santhosh, K., and Lele, S. (2010). Assessment of localized artificial diffusivity scheme for large-eddy simulation of compressible turbulent flows. *Journal of Computational Physics*, 229(5):1739–1762.
- Kennedy, C. A. and Gruber, A. (2008). Reduced aliasing formulations of the convective terms within the navier–stokes equations for a compressible fluid. *Journal of Computational Physics*, 227(3):1676–1700.
- Kim, J. and Lee, D. (2001). Adaptive nonlinear artificial dissipation model for Computational Aeroacoustics. *AIAA Journal*, 39(5):810–818.

## References II

- Lele, S. (1992). Compact finite difference schemes with spectral-like resolution. *Journal of Computational Physics*, 103(1):16–42.
- Pirozzoli, S. (2010). Generalized conservative approximations of split convective derivative operators. *Journal of Computational Physics*, 229(19):7180–7190.
- Pirozzoli, S. (2011). Stabilized non-dissipative approximations of euler equations in generalized curvilinear coordinates. *Journal of Computational Physics*, 230(8):2997–3014.
- Tam, C. and Webb, J. (1993). Dispersion-relation-preserving finite difference schemes for computational acoustics. *Journal of Computational Physics*, 107(2):262–281.