

Introduction

Practical Work (=TP)

Purpose:

- 1) Understand the main operating principles of a finite element code.
- 2) Numerically solve mechanical problems of continuous media without analytical solutions: geometry and/or complex boundary conditions.

5 sessions:

December 2: Wings → Introduction, software installation, parameters and representation

December 3: Gravity Dam I → Post-processing and T6 elements

December 9: Gravity Dam II

December 16: Plate with holes I → Boundary conditions, stress concentration

December 17: Plate with holes II

Evaluation on 2 reports:

1) December 15: Gravity Dam

2) January 4: Plate with holes

Reports (introductions/pictures/questions/conclusions) should be in PDF and uploaded before the due date on Moodle.

Documentation and questions in English

Introduction

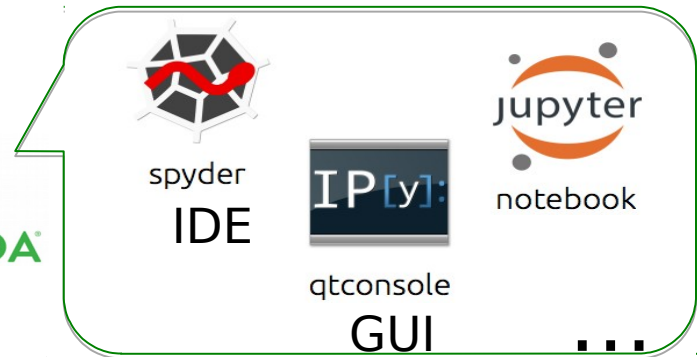
Code :

- Developed in Python (see « [Quick overview of Python language](#) »)
- Low levels (ex: access to local kinematics, to assembly phases,...)
- Written by J. Bleyer (ENPC, IFSTTAR, CNRS UMR 8205)
- Consists mainly of a library called Wombat

Softwares to install :

- **Python + IDE** : programming language and environment.

- (**you can use the Anaconda suite**
- <https://www.anaconda.com/download>)

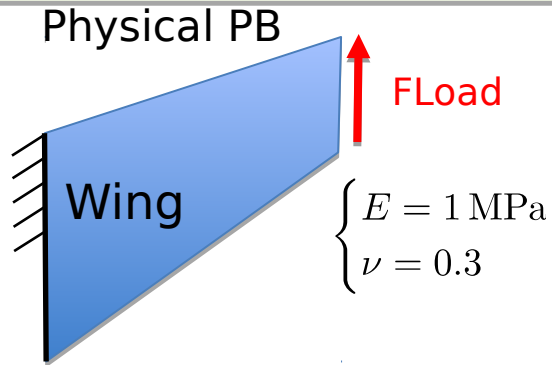


- **The meshio library**: input/output for mesh formats (<https://pypi.org/project/meshio/>)

- **GMSH** = mesher (open source on Geuzaine's website: <http://geuz.org/gmsh/>)



Sequencing .py files



Structure of directories and files:



Geometry and mesh



wing.geo



wing.msh

All functions (in Python):

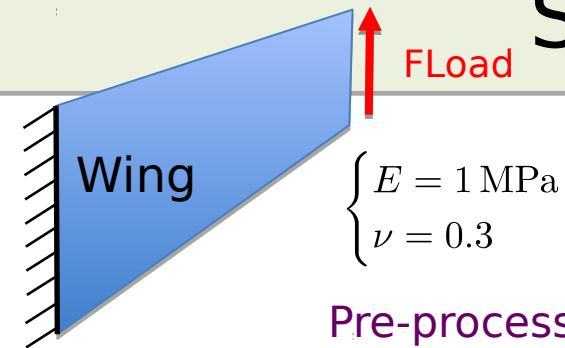
- Boundary conditions
- Elements
- Assembly
- Solving

...

Main (in Python) that:

- Calls the mesh construction
- Uses Wombat functions to solve the FE problem

Sequencing .py files



Pre-processing

wing.geo



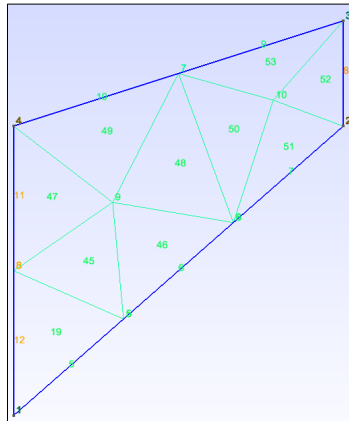
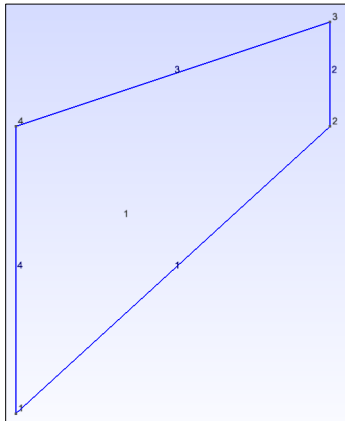
wing.msh

File containing the
geometry of the
structure

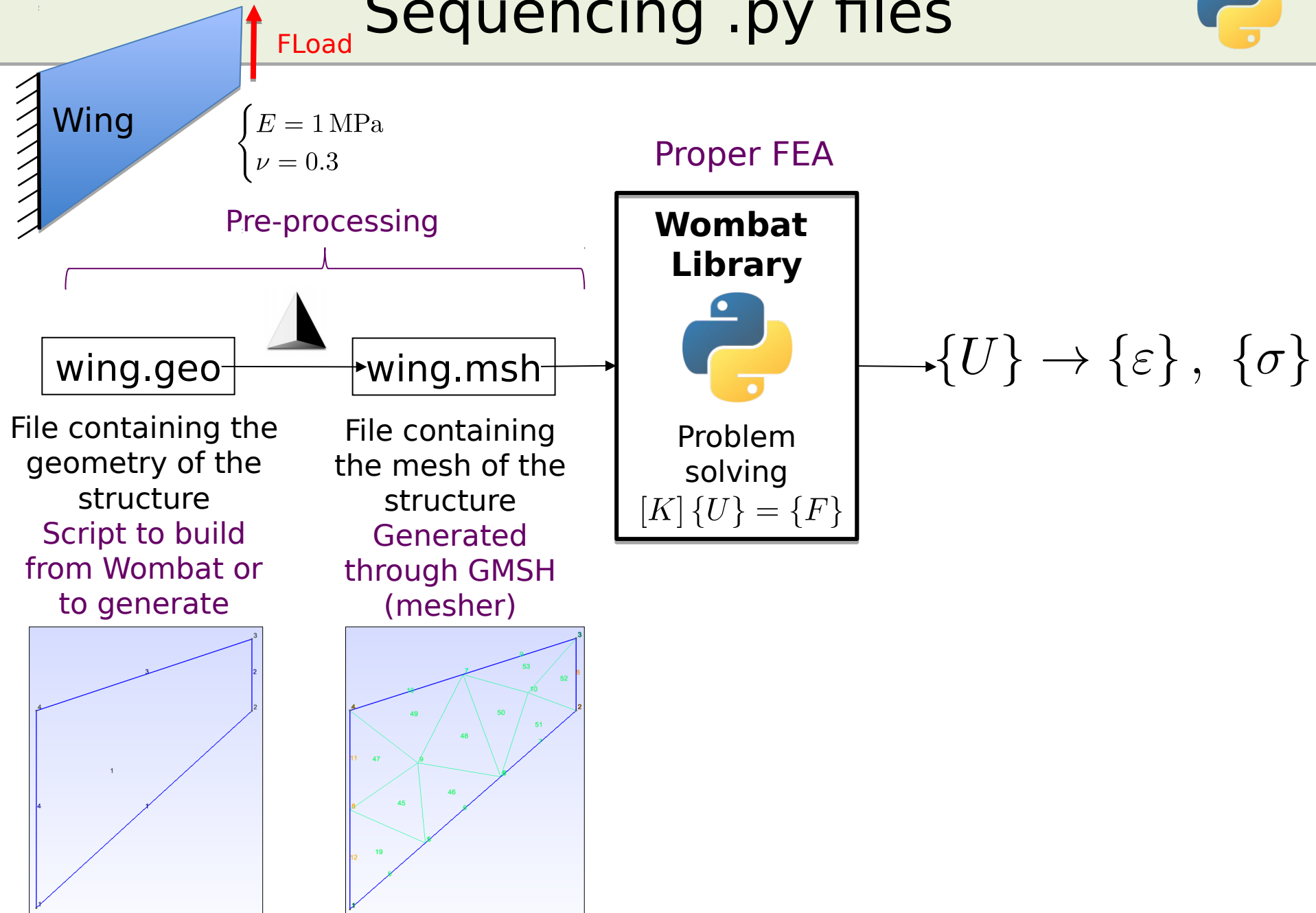
Script to build
from Wombat or
to generate

File containing the
mesh of the
structure

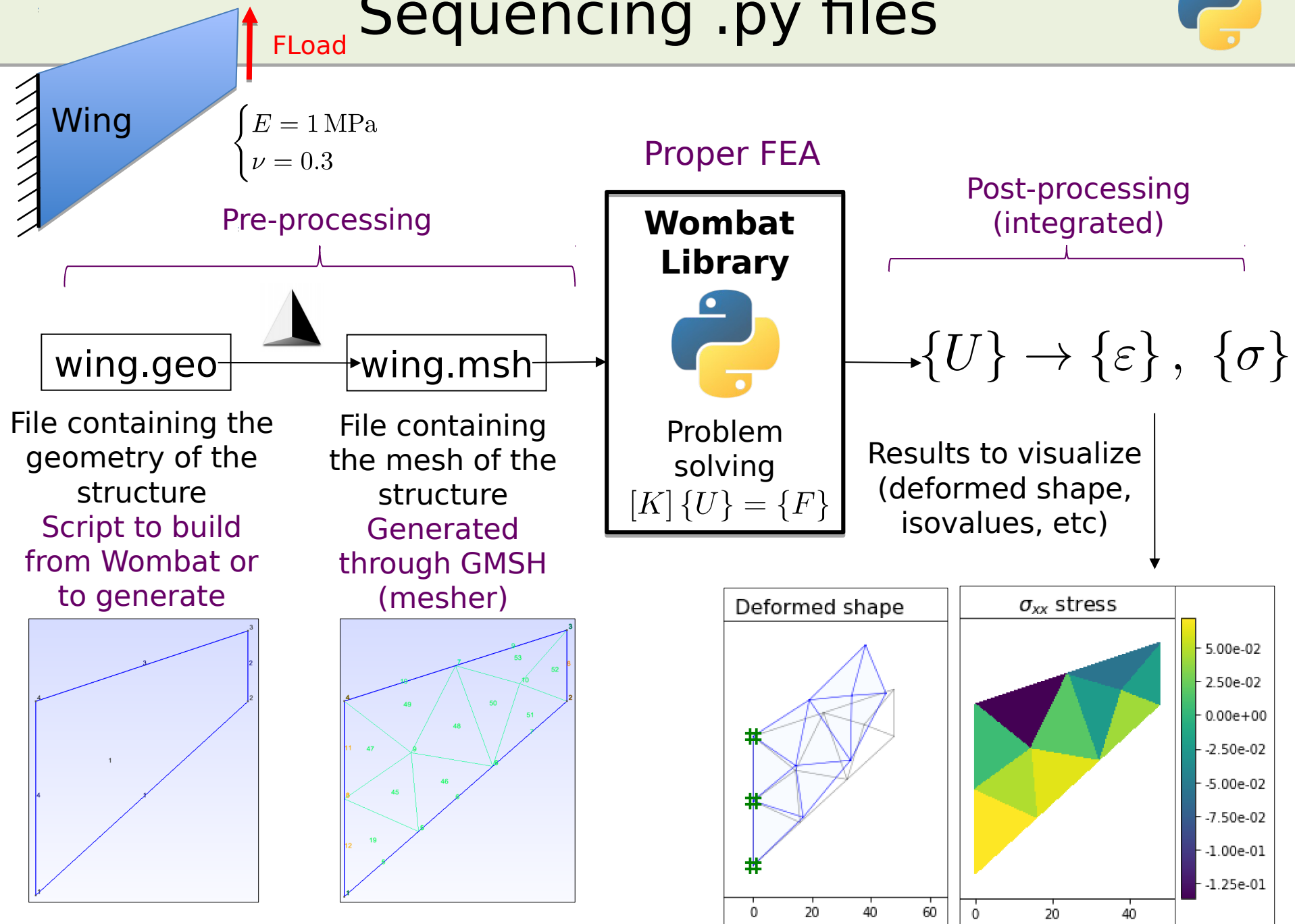
Generated
through GMSH
(mesher)



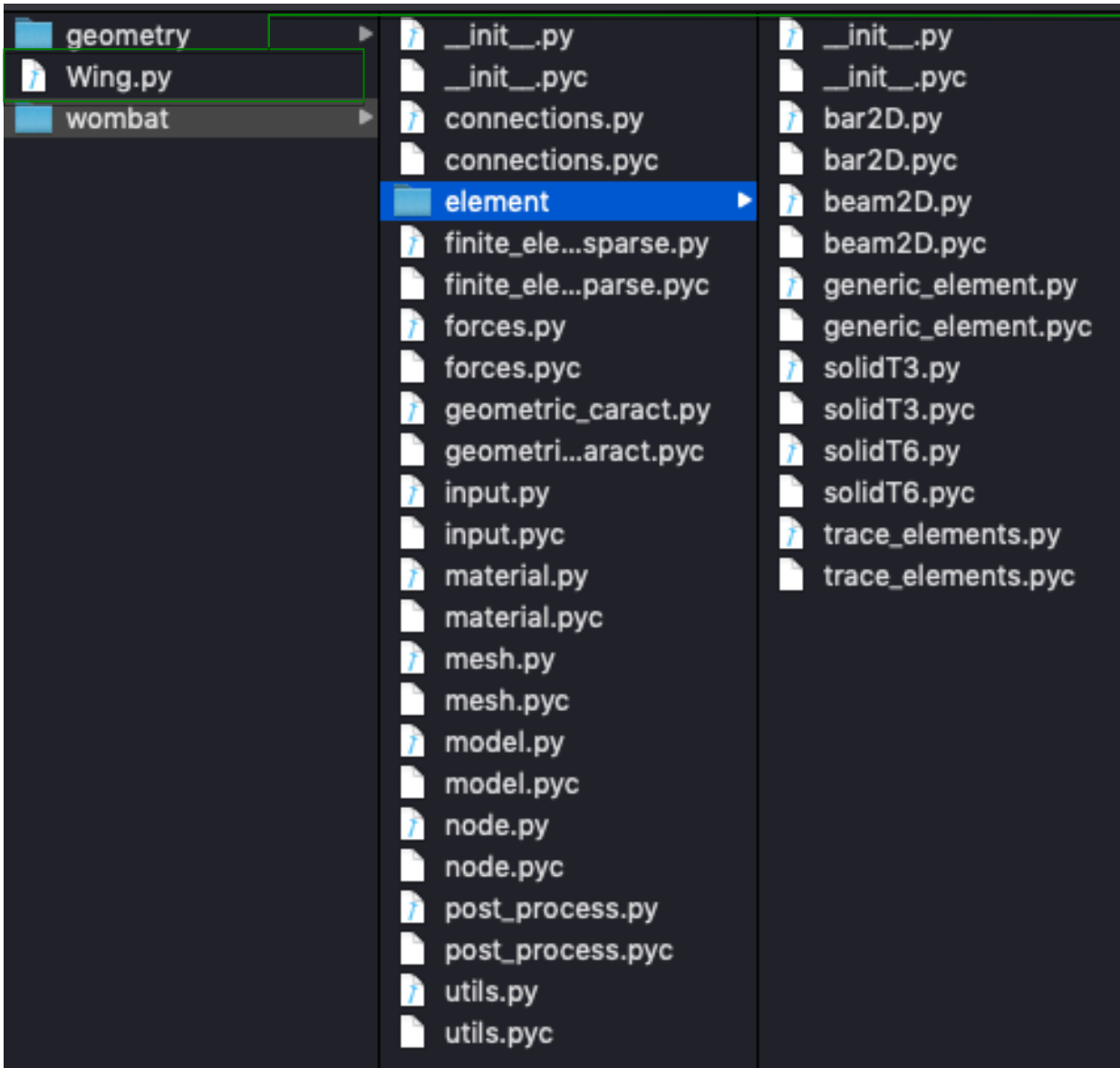
Sequencing .py files



Sequencing .py files



Structure of Wombat



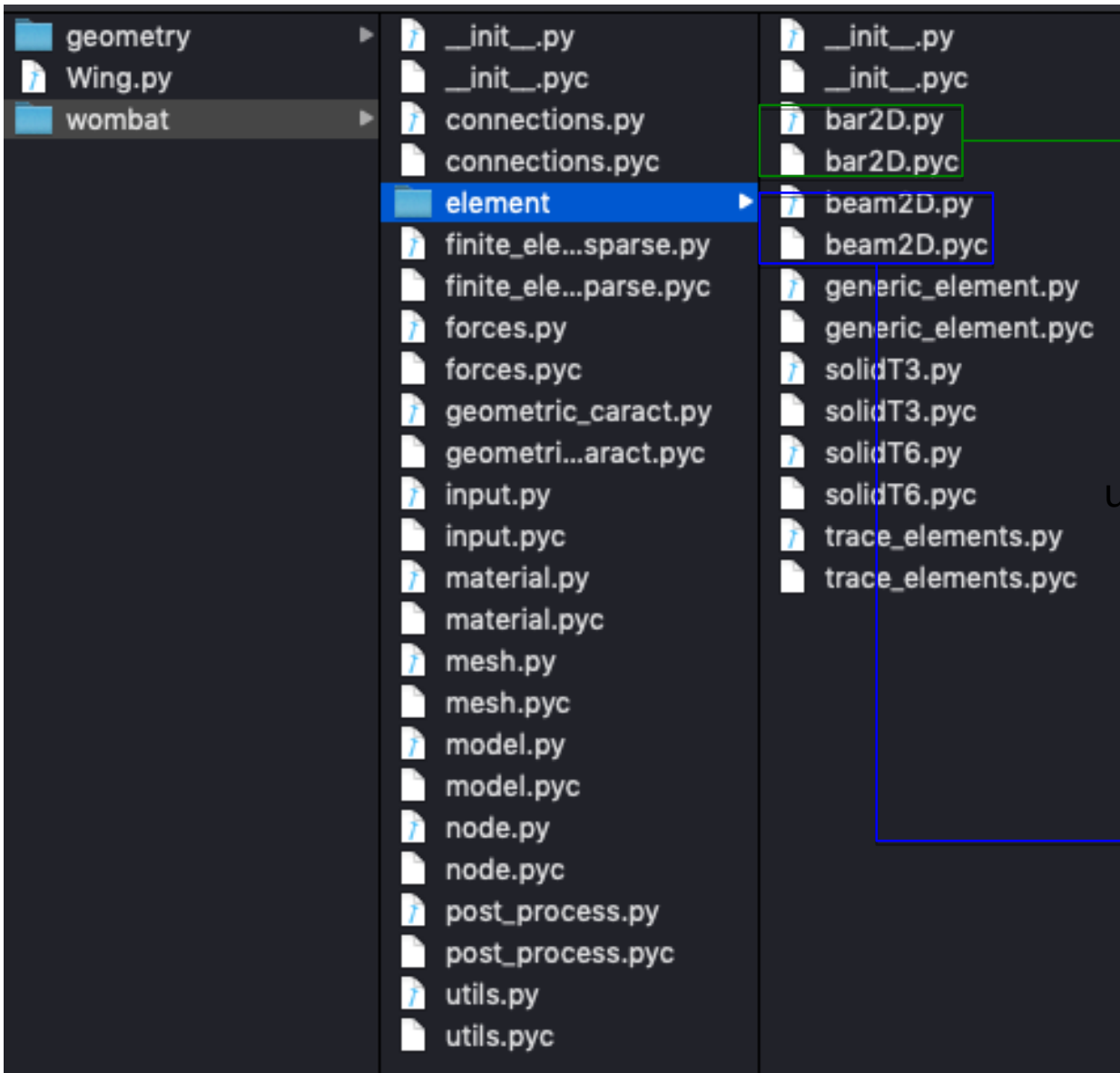
- **Main script**

- Inside **main script** Wing.py:
*call of classes and methods located inside files of the **Wombat folder***

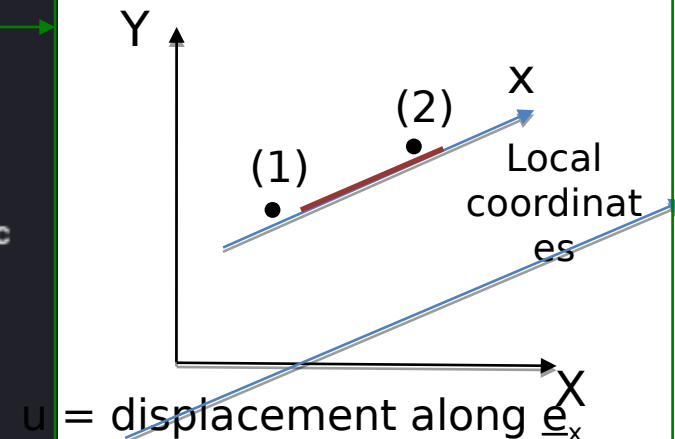
- Inside the **element folder**:
classes and methods concerning the kinematics of elements (shape functions, [Be] & [Ke] matrices, computation of local strains and stresses,...)

Ex of the code:
to be completed!

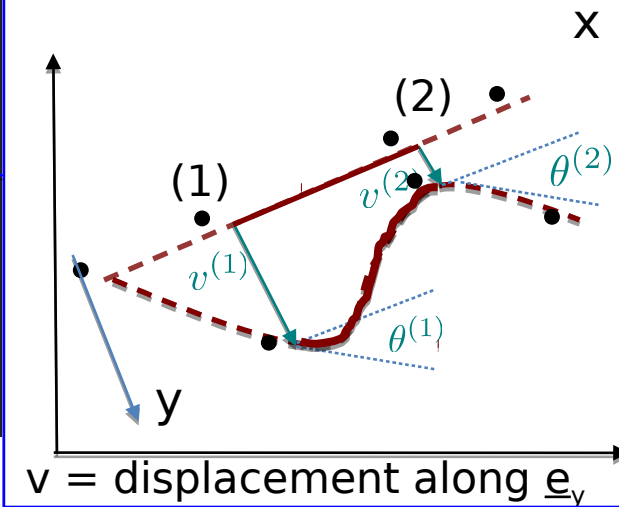
Structure of Wombat



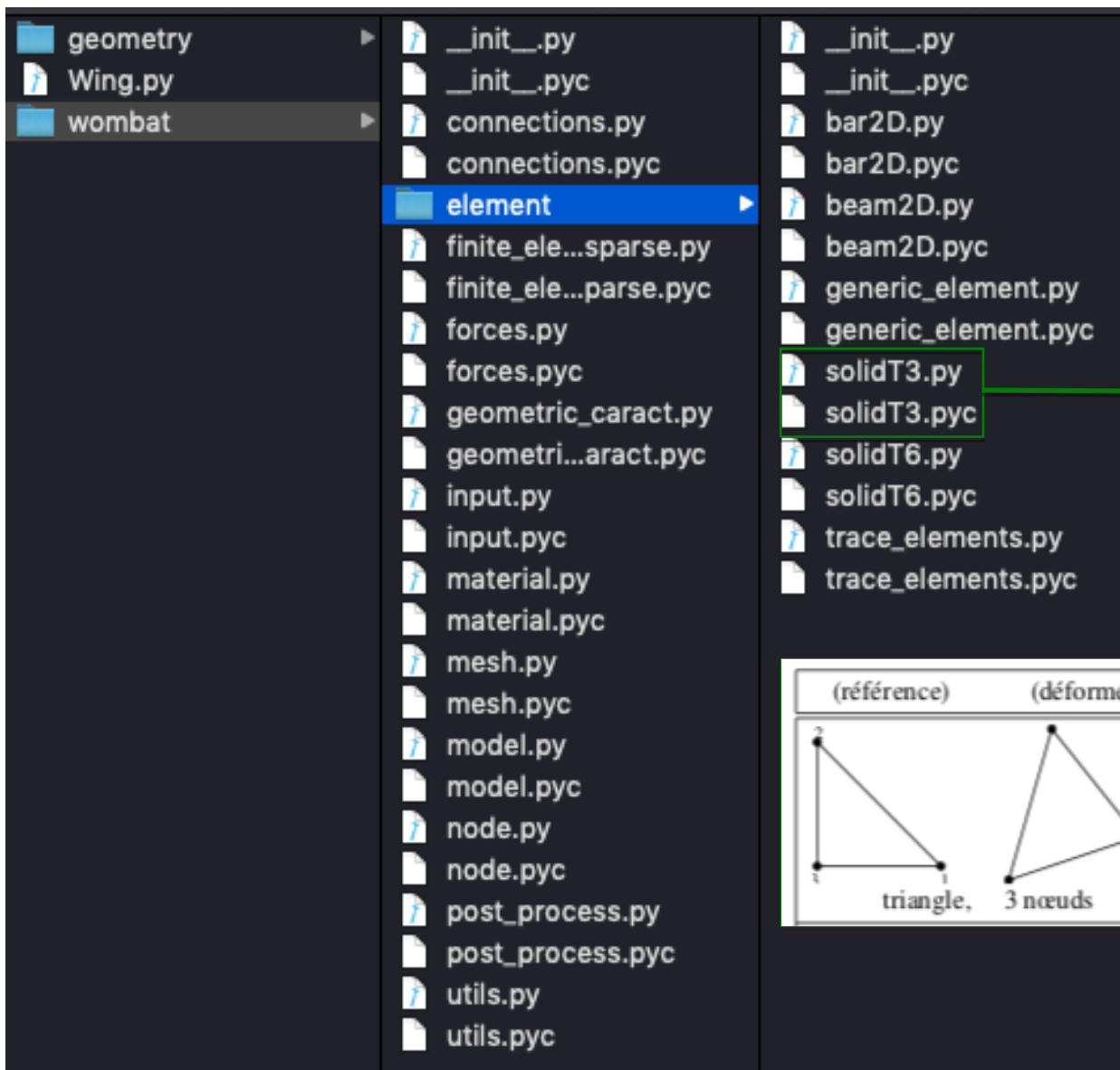
Isoparametric (lagrangian) element

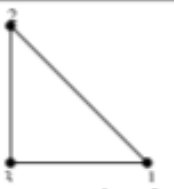



Subparametric (hermitian) element



Structure of Wombat



(référence)	(déformé)	Fonctions de forme
 <p>triangle, 3 nœuds</p>		$N_1(a_1, a_2) = a_1$ $N_2(a_1, a_2) = a_2$ $N_3(a_1, a_2) = 1 - a_1 - a_2$ (degré partiel 1, degré total 1)

With:
$$\begin{cases} a_1 = \xi \text{ (xi)} \\ a_2 = \eta \text{ (eta)} \end{cases}$$

Structure of the main script

- 1. Import wombat library**
- 2. Definition of problem parameters** (size of the structure, values of material properties, of loads...)
- 3. Geometry description and mesh generation** (implies prior definition of .geo)
- 4. Material properties** (computation of the elasticity tensor,...)
- 5. Boundary conditions and load**
- 6. Model construction** (assigning materials properties to the corresponding mesh regions,)
- 7. Assembly phase** (construction of the stiffness matrix $[K]$ and second member $\{F\}$, BC taken into account)
- 8. Solving** (of $[K]\{U\} = \{F\}$)
- 9. Post-processing**
 1. Displacements and deformation
 2. Strains and Stresses fields

**Always the same
steps for any
FEA, whatever
the code used!**

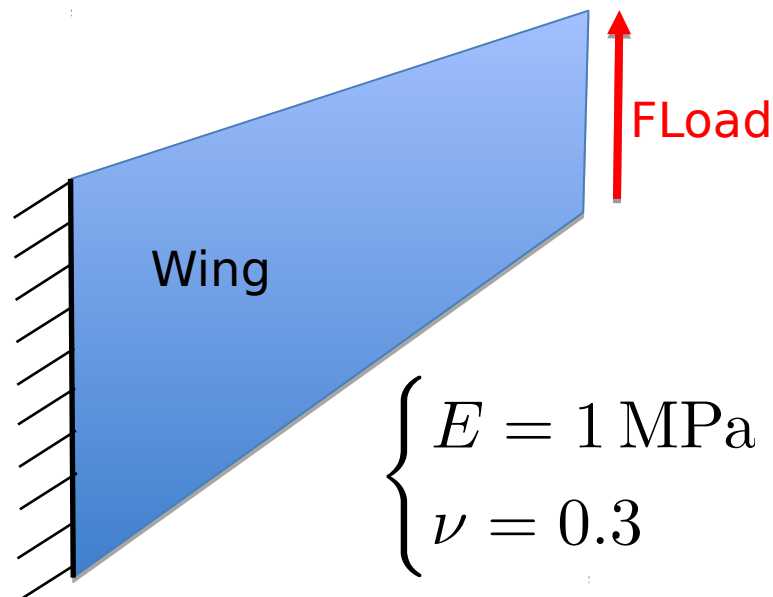
Structure of the main script

1. Import wombat library

```
21 from wombat import *
```

2. Definition of problem parameters (size of the structure, values of material properties, of loads...)

```
23 # Problem parameters  
24 Yg_mod = 1  
25 Nu_poi = 0.3  
26 FLoad = 0.0625
```



Structure of the main script

1. Geometry description and mesh generation (implies prior definition of .geo)

```
28 # Geometry description and mesh generation
29 mesh, boundary = call_gmsh('geometry/wing.geo', SolidT3)
30 right = boundary.get_elem_from_tag("right").get_nodes()
31 left = boundary.get_elem_from_tag("left").get_nodes()
```

Type of
element
required

```
20 def call_gmsh(geofile, elem_type):
21     """ Calls `Gmsh` to generate a mesh
22
23     Parameters
24     -----
25     geofile : str
26         path to GEO file, must end in ".geo"
27     elem_type : {class: `SolidT3`, class: `SolidT6`}
28         type of generated elements
29
30     Returns
31     -----
32     regions
33     a :class: `Mesh <mesh.Mesh>` object
34     """
35     if issubclass(elem_type, Triangle6):
36         order = 2
37     else:
38         order = 1
39     command = 'gmsh -order '+str(order)+' '+geofile+' -2'
40     os.system(command)
41     print "Finished meshing!\n"
42     mshfile = geofile[:-3]+'msh'
43     return read_msh(mshfile, elem_type)
```

Input.py

Located in the **geometry** folder



Modify path in call_gmsh method.
(here for instance for a MacOS)

```
39 command = '/Applications/Gmsh.app/Contents/MacOS/gmsh -order '+str(order)+' '+geofile+' -2'
```

Structure of the main script

1. Geometry description and mesh generation (implies prior definition of .geo)

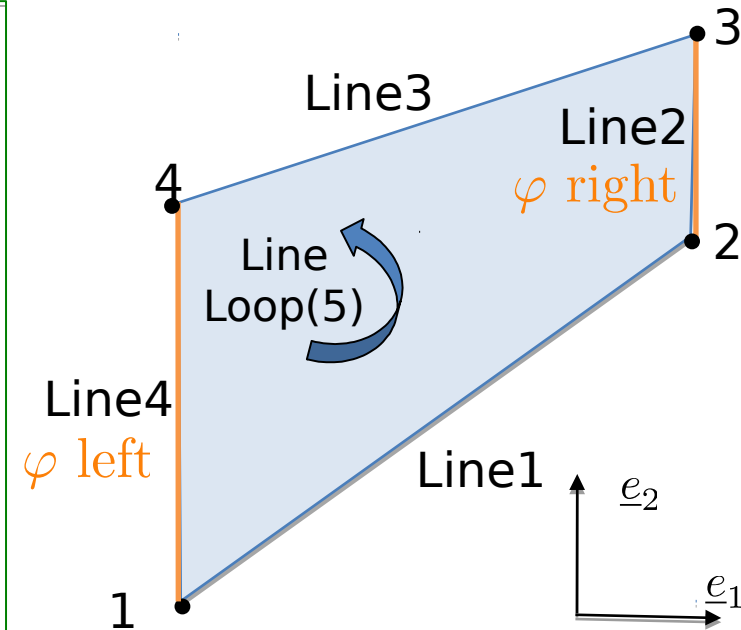
```
// global dimensions      // surface
L1 = 48.0;                Plane Surface(1) = {5};
L2 = 44.0;
L3 = 60.0;

// mesh density
lc1 = 30.;

// points coordinates
Point(1) = {0.0,0.0,0.0,lc1};
Point(2) = {L1,L2,0.0,lc1};
Point(3) = {L1,L3,0.0,lc1};
Point(4) = {0.0,L2,0.0,lc1};

// lines and line loops
Line(1) = {1,2};
Line(2) = {2,3};
Line(3) = {3,4};
Line(4) = {4,1};
Line Loop(5) = {1,2,3,4};
```

Physical Line ("right") = {2};
Physical Line ("left") = {4};
Physical Surface ("wing") = {1};



Physical Line = where BC are applied

Physical Surface = where material properties are assigned

wing.geo

Structure of the main script

```
$MeshFormat
```

```
2.2 0 8
```

```
$EndMeshFormat
```

```
$PhysicalNames
```

```
3
```

```
1 1 "left"
```

```
1 2 "right"      (Dim, n° Phys. Entity, name)
```

```
2 3 "wing"
```

```
$EndPhysicalNames
```

```
$Nodes
```

```
10 ← Total number of nodes
```

```
1 0 0 0
```

```
2 48 44 0    ~ Coordinates matrix
```

```
3 48 60 0    (n° node, X, Y, Z)
```

```
4 0 44 0
```

```
5 15.999999999994201 14.6666666666661351 0
```

```
6 31.999999999994015 29.333333333327847 0
```

```
7 24.000000000005996 52.00000000001999 0
```

```
8 0 22.000000000004024 0
```

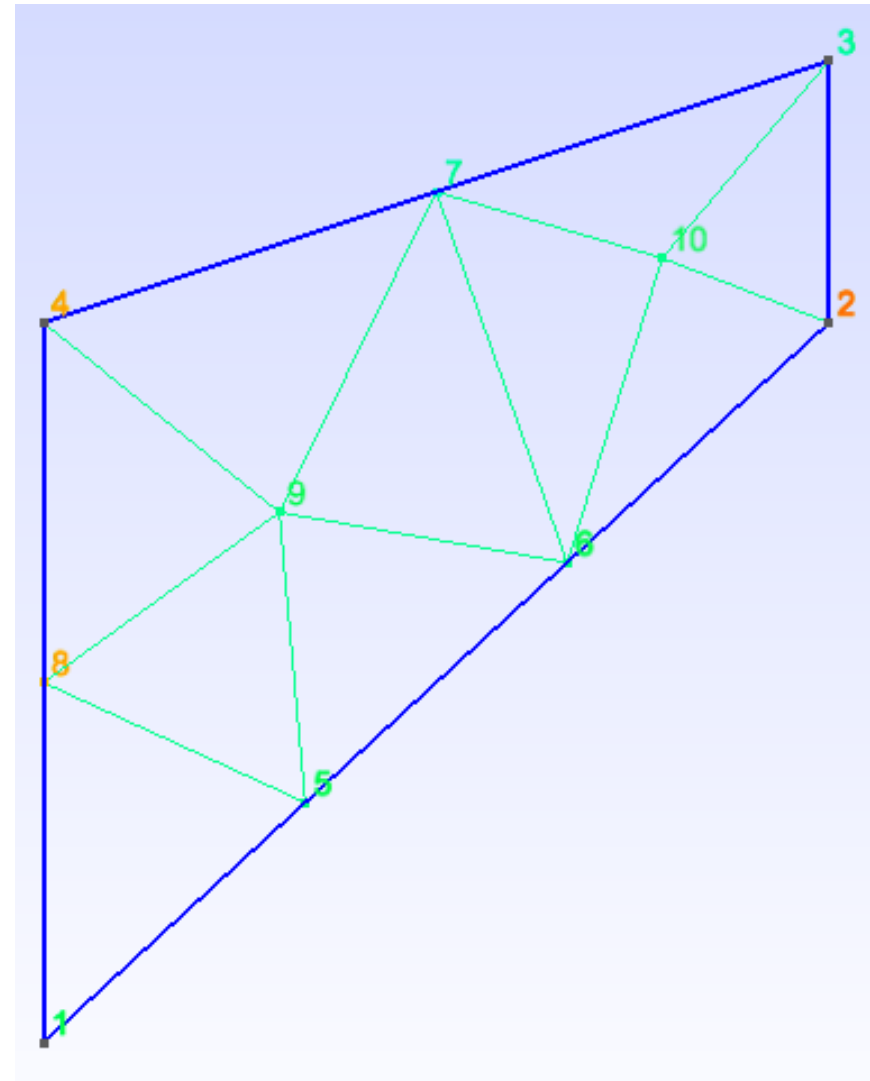
```
9 14.41534391534047 32.40404040399452 0
```

```
10 37.80952380953335 47.95238095237472 0
```

```
$EndNodes
```

```
...
```

wing.msh



Structure of the main script

wing.msh

```
...
$Elements
13 ← Total number of elements
```

```
1 1 2 1 2 2 3
2 1 2 2 4 4 8
3 1 2 2 4 8 1
```

~ **Connectivity matrix**
(n° node, Type, ...,
Phys. Set, Orig. Geom. Set,
Nodes connectivity)

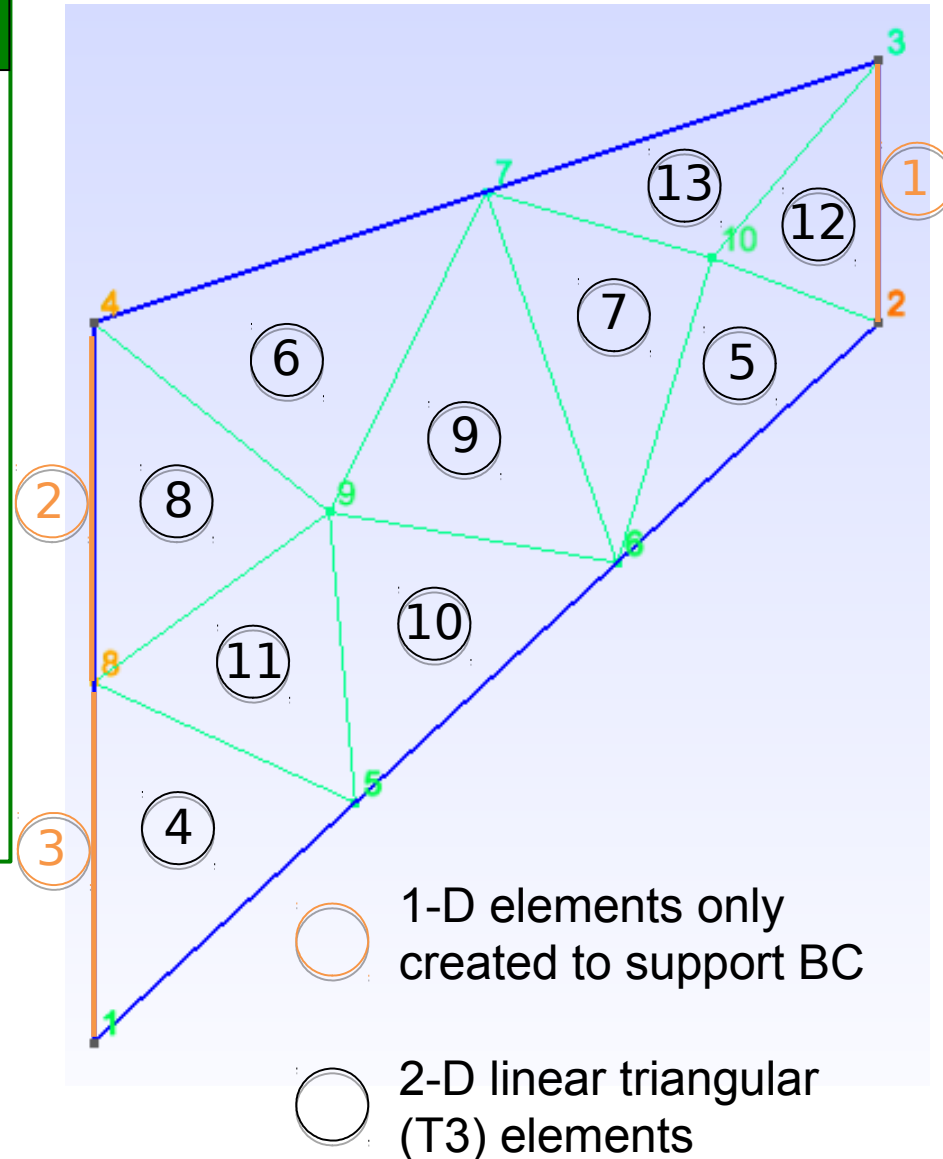
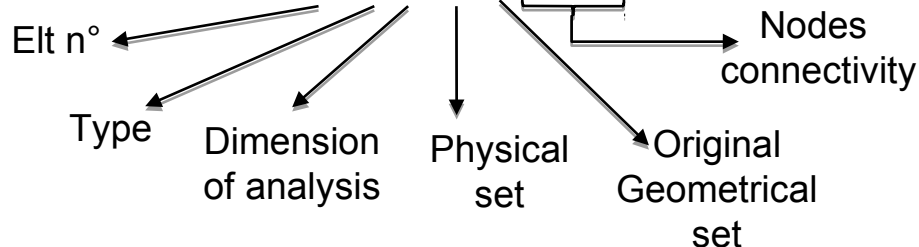
Type:

- 1=2-node line
- 2=3-node triangle
- etc

```
4 2 2 3 1 1 5 8
5 2 2 3 1 2 10 6
6 2 2 3 1 4 9 7
7 2 2 3 1 6 10 7
8 2 2 3 1 4 8 9
9 2 2 3 1 6 7 9
10 2 2 3 1 5 6 9
11 2 2 3 1 5 9 8
12 2 2 3 1 2 3 10
13 2 2 3 1 3 7 10
```

```
$EndElements
```

Ex : 6 2 2 3 1 4 9 7



Structure of the main script

1. **Material properties** (computation of the elasticity tensor,...)

```
33 # Material properties
34 mat_concr = LinearElastic(E=Yg_mod, nu=Nu_poi, model="plane_stress")
35
```

```
31 class LinearElastic(Material):
32     """ Linear elastic material
33
34     Attributes
35     -----
36     Young_modulus : float
37         material Young modulus :math:`E`
38     Poisson_coeff : float
39         material Poisson coefficient :math:`\nu` (with :math:`-1 < \nu < 1/2`),
40         ignored for :class:`Bar2D <bar2D.Bar2D>` and :class:`Beam2D <beam2D.Beam2D>` elements
41     rho : float
42         material volumetric mass density :math:`\rho`
43     model : {'plane_strain', 'plane_stress'}
44         type of 2D model
45     C : ndarray
46         elasticity matrix :math:`[C]` shape=(3,3)
47     """
```

material.py

Available Plane elasticity models (no axisymmetry so far...)

Structure of the main script

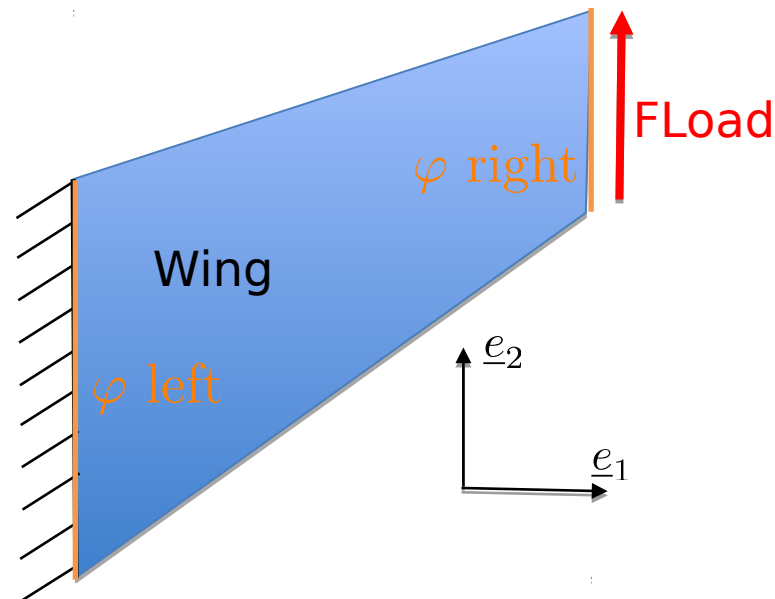
1. Boundary conditions and load

```
36 # Boundary conditions and load
37 appuis = Connections()
38 appuis.add_imposed_displ(left, ux=0, uy=0)
39
40 forces = ExtForce()
41 forces.add_distributed_forces(right, fy=FLoad)
```

See : [connections.py](#) & [forces.py](#)

Possibility to define :

- **Imposed displacements or imposed relations** between nodes displacements (master/slaves relationships)
- **concentrated and distributed forces** (and **couples** for beam elements only)



Structure of the main script

1. **Model construction** (assigning materials properties to the corresponding mesh regions,)

```
43 # Model construction
44 model = Model(mesh, mat)
```

model.py

2. **Assembly phase** (construction of the stiffness matrix [K] and second member {F}, BC taken into account)

```
46 # Assembly phase
47 K = assembl_stiffness_matrix(model)
48 L, Ud = assembl_connections(appuis, model)
49 F = assembl_external_forces(forces, model)
```

finite_elements_sparse.py

Structure of the main script

3. Solving (of $[K]\{U\}=\{F\}$)

```
51 # Solving
52 U, lamb = solve(K, F, L, Ud)
```

finite_elements_sparse.py

Augmented Lagrangian solving method

U = displacements at nodes

lamb = lambda = Lagrange multipliers = Reaction forces components

4. Post-processing

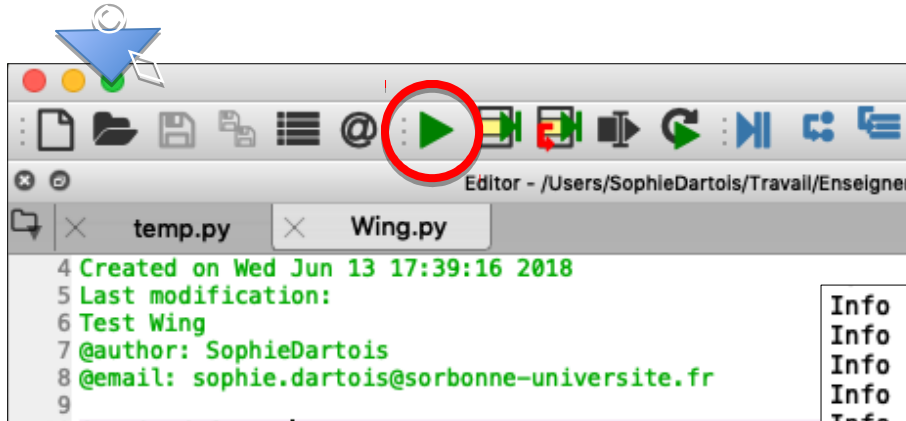
```
# Post-processing
coeff = 100
list_res = post_proc_res()
list_res.compute_res(U, model)
list_res.trace_res(mesh, coeff, U, appuis, mat.model)
```

res_treat.py


Displacement computation, stresses (compute_res),
and display as isovalues (trace_res)

Running the code

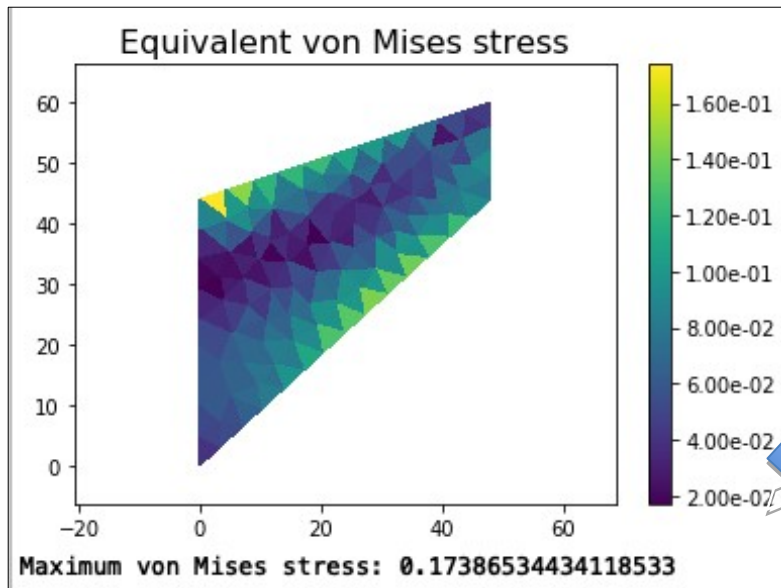
1- In Spyder, run the script Wing.py by clicking on the « run » icon for instance :



2- You should see infos from gmsh indicating that meshing was performed successfully



```
Info : Started on Wed Jul 24 16:30:08 2019
Info : Reading 'geometry/wing.geo'...
Info : Done reading 'geometry/wing.geo'
Info : Finalized high order topology of periodic connections
Info : Meshing 1D...
Info : Meshing curve 1 (Line)
Info : Meshing curve 2 (Line)
Info : Meshing curve 3 (Line)
Info : Meshing curve 4 (Line)
Info : Done meshing 1D (0.000975 s)
Info : Meshing 2D...
Info : Meshing surface 1 (Plane, Delaunay)
Info : Done meshing 2D (0.00553918 s)
Info : 103 vertices 208 elements
Info : Writing 'geometry/wing.msh'...
Info : Done writing 'geometry/wing.msh'
Info : Stopped on Wed Jul 24 16:30:08 2019
Finished meshing!
```



3- In the end you should see in the console isovalue graphs asked for as well as any information needing to be calculated and displayed (such as max VM stress here)