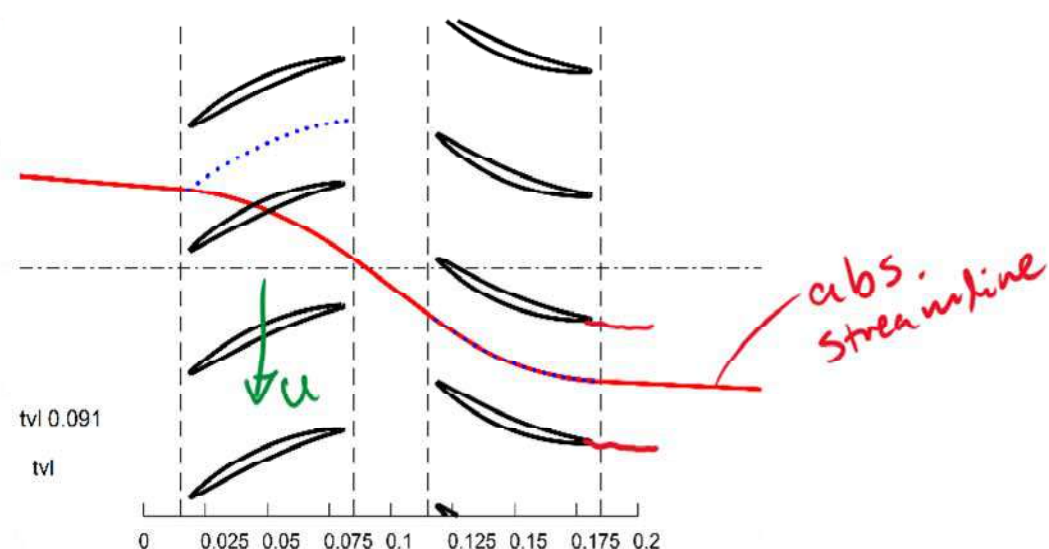


Euler TM eq.

→ general form

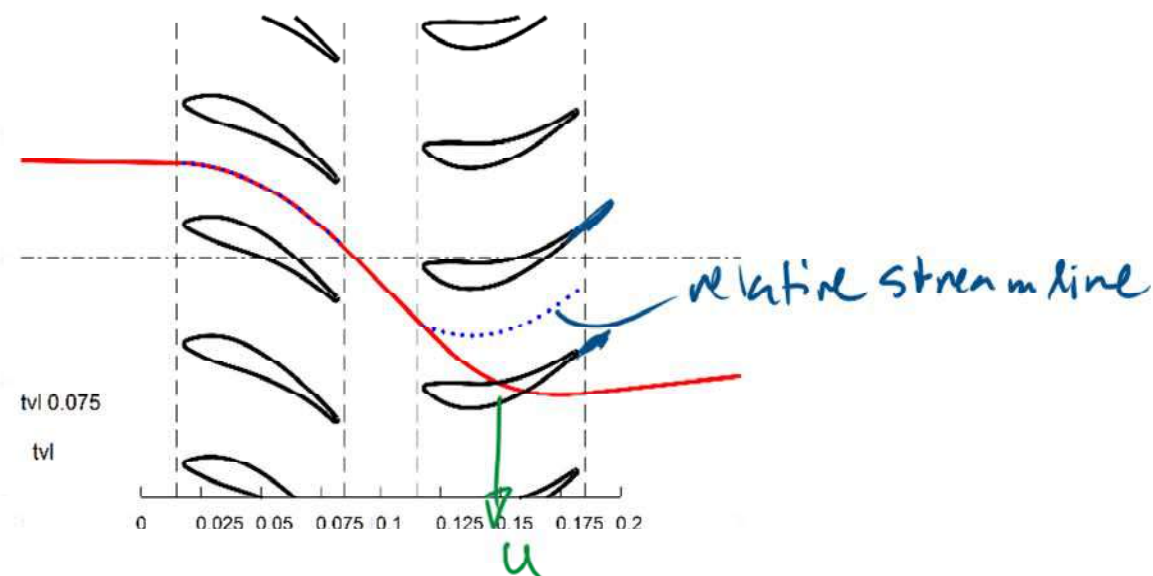
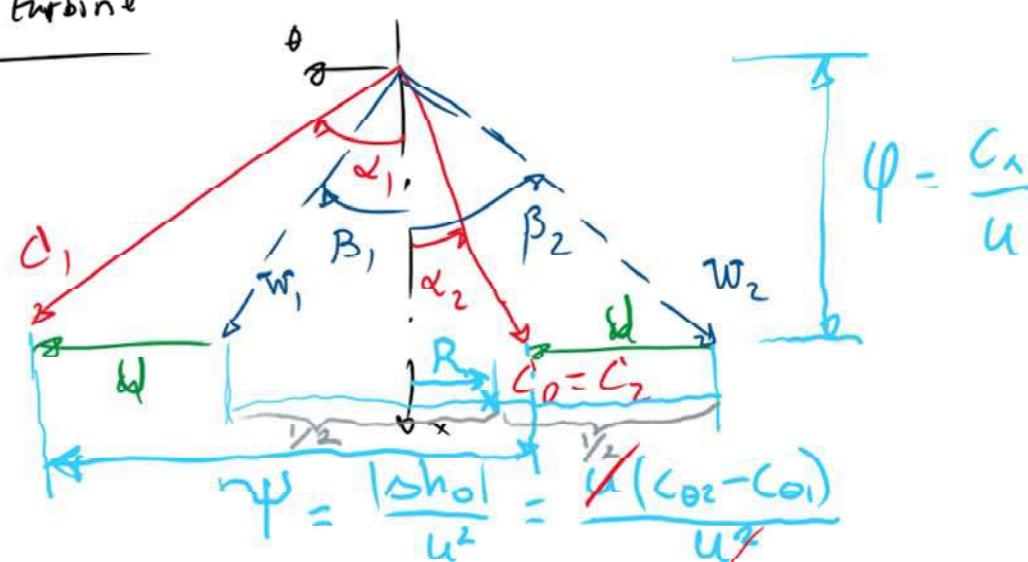
$$h_{02} - h_{01} = u_2 \cdot c_{02} - u_1 \cdot c_{01}$$

Repetition

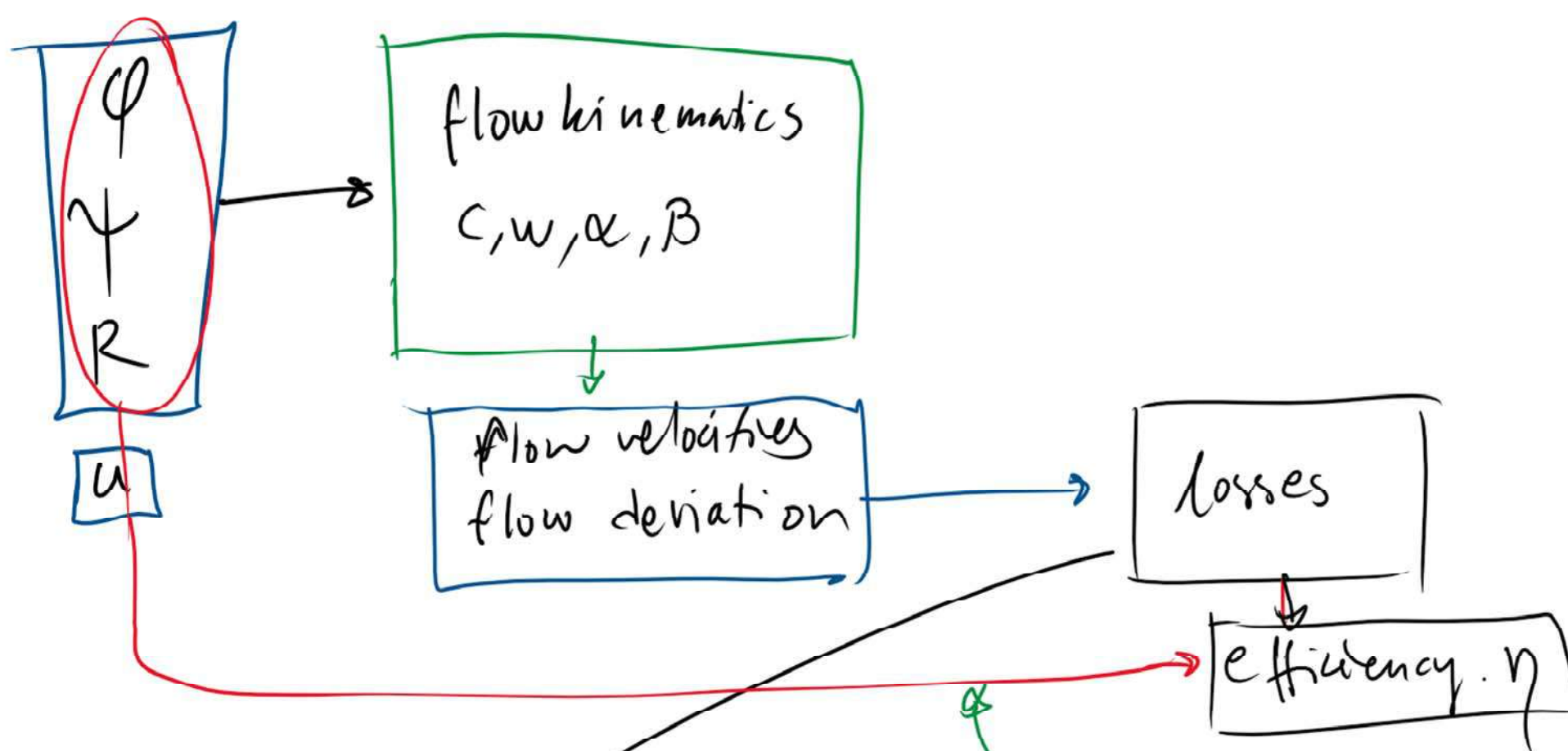


compressor

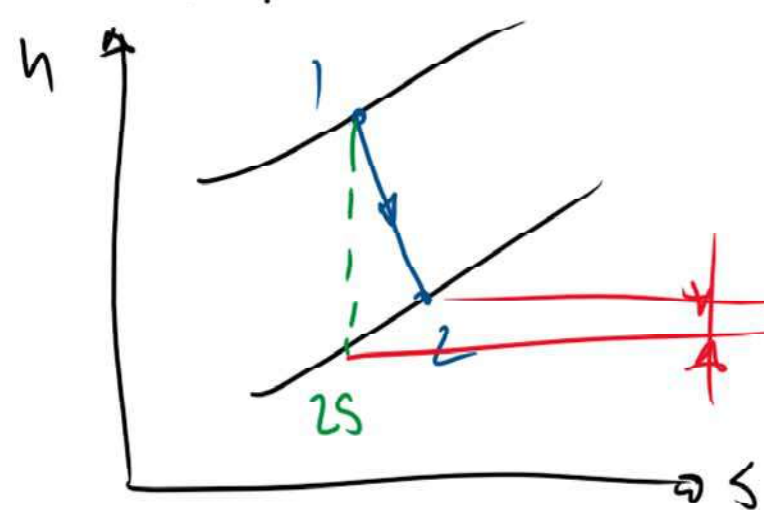
e.g. turbine



turbine



→ simplified loss models



correlation for loss coefficient

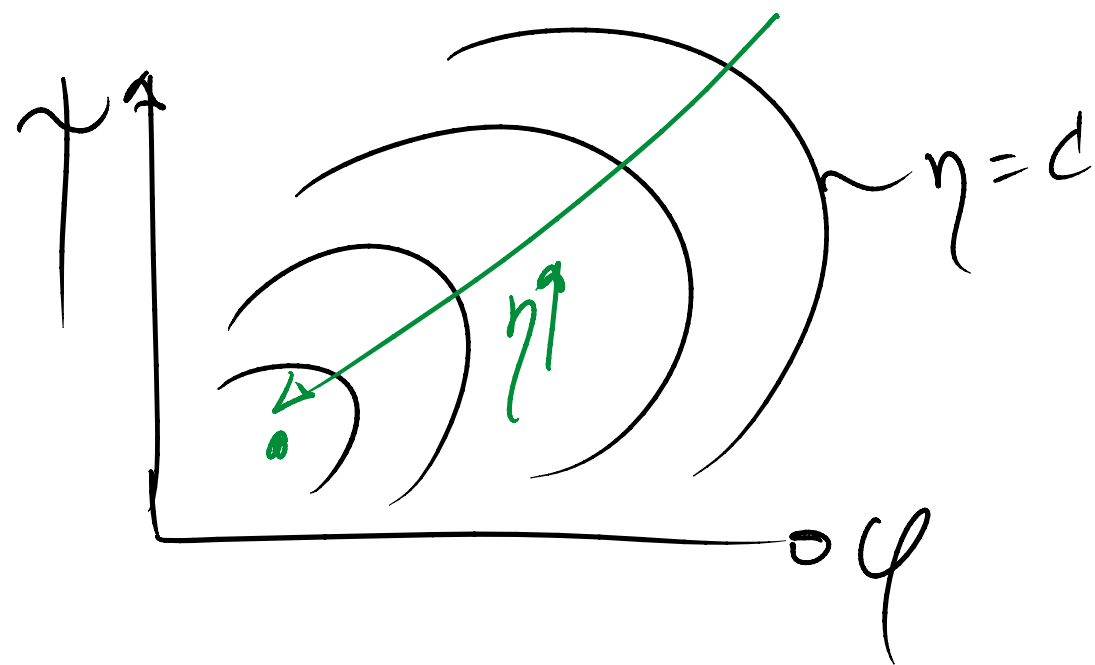
$$\epsilon = 0.04 + 0.06 \cdot \left(\frac{\epsilon}{100} \right)^2$$

Soderberg's loss correlation
where

ϵ = flow deviation in a blade row

$$\begin{cases} \epsilon_u = \alpha_0 - \alpha_1 \\ \epsilon_r = \beta_1 - \beta_2 \end{cases}$$

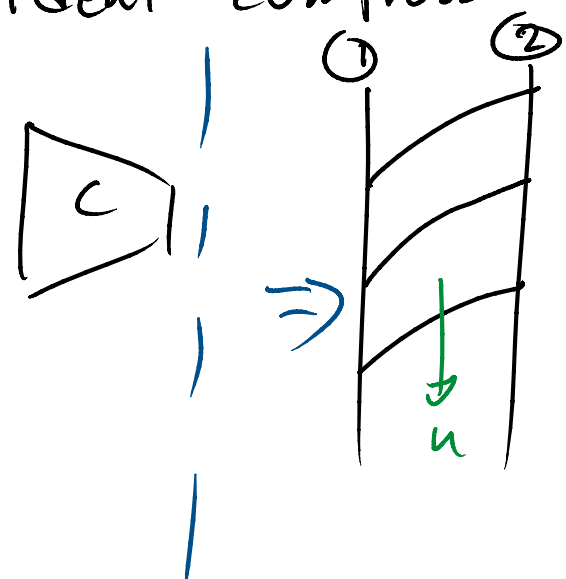
→ Smith charts



for a Mach number of 0.5

Example 4) from "stage thermodynamics" hand-out

ideal compressor rotor $\Pi = \frac{P_{02}}{P_{01}} = 1.7$



$$T_{01} = 291 \text{ K}$$

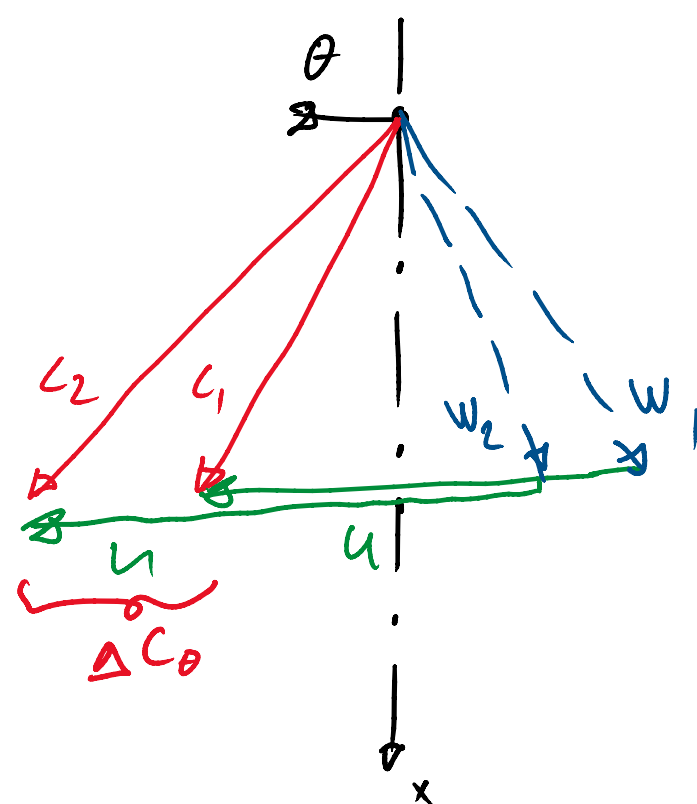
$$p_{01} = 101.3 \text{ kPa}$$

$$\text{ideal gas (air)} \quad \gamma = 1.4$$

$$c_p = 1004.5 \text{ J/kgK}$$

$$u = 300 \text{ m/s}$$

What change in tang. velocity component is required for $\Pi = 1.7$?
 ΔC_θ



$$\text{Euler:} \quad \Delta h_0 = u \cdot \Delta C_\theta \quad (2)$$

$$\text{ideal gas:} \quad \Delta h_0 = c_p \Delta T_0 \quad (1)$$

Since it's an ideal compressor (reversible - no losses)
 \rightarrow isentropic relation between total conditions

$$(1) \Rightarrow \Delta h_0 = c_p T_0 = 1004.5 (T_{02} - T_{01})$$

$$\text{isentropic relationship} \Rightarrow \frac{T_{02}}{T_{01}} = \left(\frac{P_{02}}{P_{01}} \right)^{\frac{\gamma-1}{\gamma}} \Rightarrow T_{02} = 291 \cdot (1.7)^{\frac{1.4-1}{1.4}} = 338.6 \text{ K}$$

$$(1) \Rightarrow \Delta h_0 = c_p T_0 = 1004.5 (338.6 - 291) = 47.8 \text{ kJ/kg}$$

$$(2) \Rightarrow \underline{\Delta c_0} = \frac{\Delta h_0}{u} = \frac{47800}{300} = \underline{159.5 \text{ m/s}}$$

problem 5) ... in the same hand out

Axial turbine stage

$$r_m = r \rightarrow u = r \cdot \omega \text{ because } u = r_m \cdot \omega \quad \frac{2\pi \cdot N}{60} \leftarrow \text{rotational speed}$$

$c_x = c = 100 \text{ m/s}$
normal repetition stage, $\vec{c}_0 = \vec{c}_2$

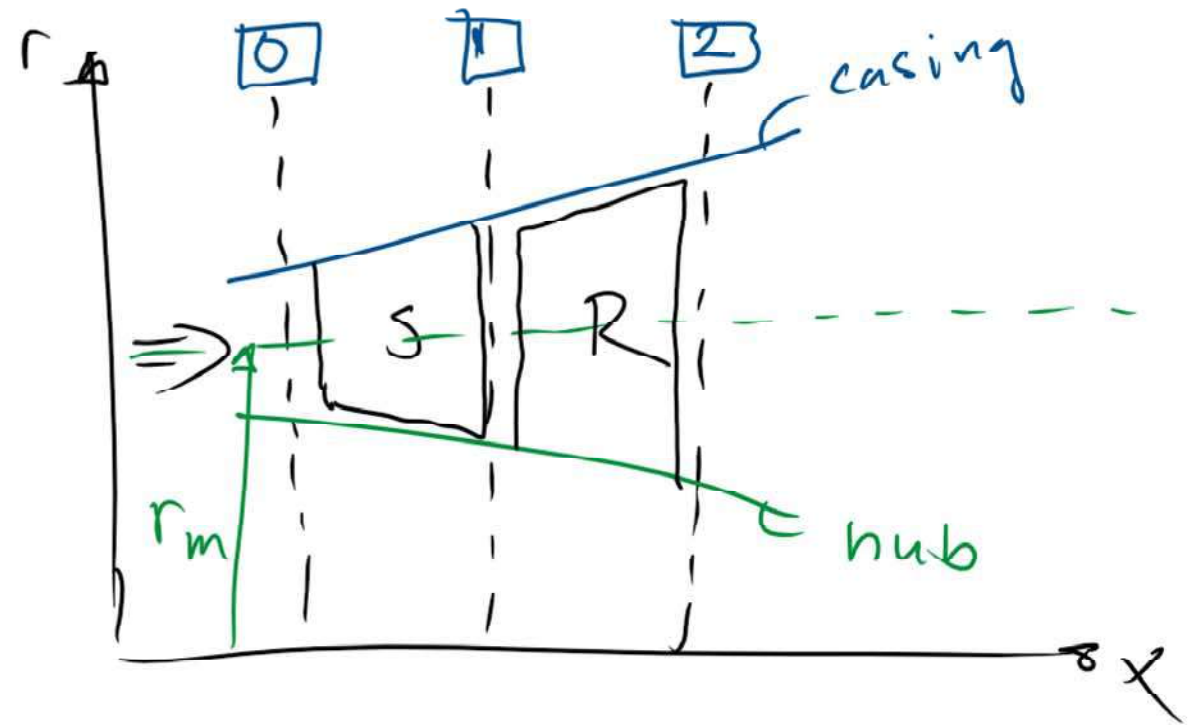
$$c_1 = 300 \text{ m/s}$$

$$w_2 = 200 \text{ m/s}$$

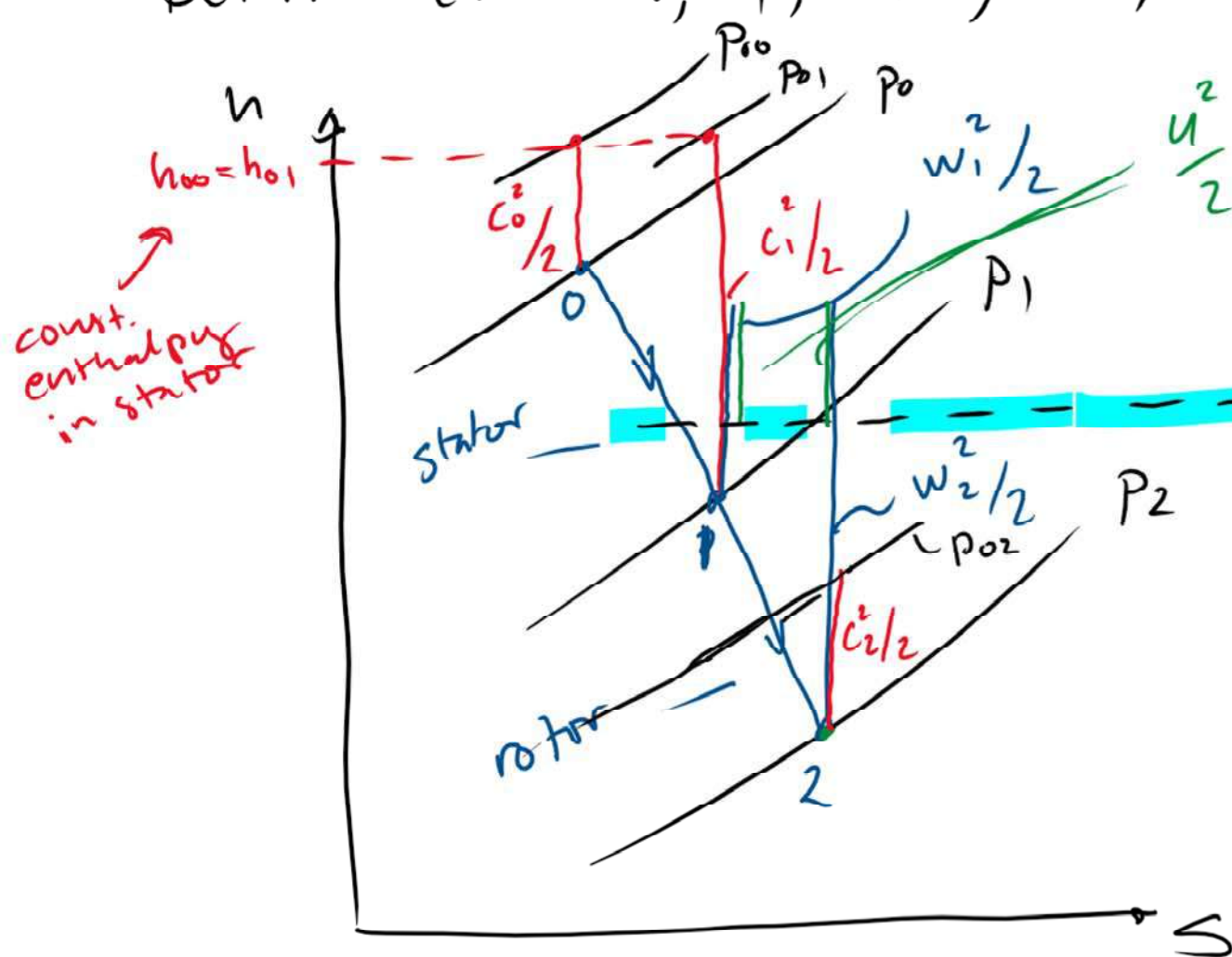
$$u = 150 \text{ m/s}$$

$$T_0 = 800 \text{ K} \quad \text{static temp. at inlet}$$

$$c_p = 1148 \text{ J/kgK}$$

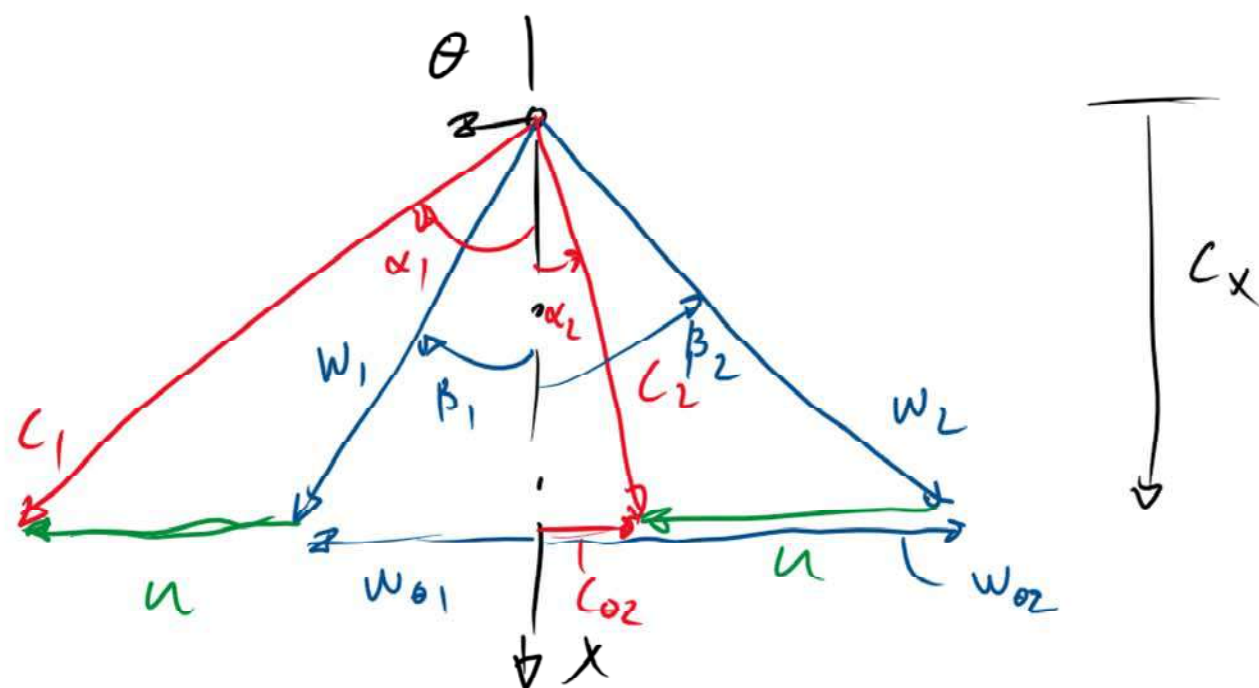


Determine: $h_0, h_1, h_2, h_{00}, h_{01}, h_{02}$!



$$I = \phi$$

$$\text{rothalpy, } I = h + \frac{w^2}{2} - \frac{u^2}{2}$$



enthalpies throughout the stage

$$\underline{h_0} = c_p \cdot T_0 = 1148 \frac{\text{J}}{\text{kgK}} \cdot 800 \text{ K} \approx \underline{918.4 \frac{\text{kJ}}{\text{kg}}}$$

$$\underline{h_{01}} = \underline{h_{00}} = h_0 + \frac{c_0^2}{2} = 918.4 \frac{\text{kJ}}{\text{kg}} + \frac{102.7^2}{2} = \underline{923.7 \frac{\text{kJ}}{\text{kg}}}$$

$$C_0: \text{ we have that } C_x = W_x \rightarrow W_{\theta 2} = -\sqrt{W_2^2 + W_x^2} = -173 \text{ m/s}$$

\uparrow 200
 \uparrow 100 = C_x

$$C_{\theta 2} = W_{\theta 2} + u = -173 + 150 = -23.2 \text{ m/s}$$

$$\Rightarrow C_2 = \sqrt{C_x^2 + C_{\theta 2}^2} = \sqrt{100^2 + (-23.2)^2} = 102.7 \text{ m/s}$$

$C_2 = C_0$

$$h_1 = h_{01} - \frac{C_1^2}{2} = 923.7 - \frac{300^2}{2} = 878.7 \text{ kJ/kg}$$

$$h_2: \text{ go through rotating } \left(I = h + \frac{w^2}{2} - \frac{u^2}{2} \right)$$

$$W_1: \text{ we have } C_x = W_x = 100 \text{ m/s}$$

$$\Rightarrow C_{\theta 1} = \sqrt{C_1^2 - C_x^2} = 282.8 \text{ m/s}$$

$$W_{\theta 1} = C_{\theta 1} - u = 282.8 - 150 = 132.8 \text{ m/s}$$

$$W_1 = \sqrt{W_{\theta 1}^2 + W_x^2} = \sqrt{132.8^2 + 100^2} = 166.3 \text{ m/s}$$

$$\Rightarrow h_1 + \frac{W_1^2}{2} - \frac{u_1^2}{2} = h_2 + \frac{W_2^2}{2} - \frac{u_2^2}{2} \quad \text{since } u_1 = u_2$$

$$\Rightarrow \underline{h_2} = h_1 + \frac{W_1^2}{2} - \frac{W_2^2}{2} = 878.7 + \frac{166.3^2}{2} - \frac{200^2}{2} = 872.5 \text{ kJ/kg}$$

$$\underline{h_{02}} = h_2 + \frac{C_2^2}{2} = 872.5 + \frac{102.7^2}{2} = 877.8 \text{ kJ/kg}$$

$$\text{Stage specific power } \underline{\Delta h_0} = h_{02} - h_{01} = 877.8 - 923.7 = \underline{-45.9 \text{ kJ/kg}}$$

From Euler TM eq.

$$\begin{cases} C_{\theta 1} = 282.8 \text{ m/s} \\ C_{\theta 2} = -23.2 \text{ m/s} \\ u = 150 \text{ m/s} \end{cases}$$

$$\underline{\Delta h_0} = u \cdot (C_{\theta 2} - C_{\theta 1}) = 150(-23.2 - 282.8) = \underline{-45.9 \text{ kJ/kg}}$$

Power output (\dot{W}) or P if $\dot{m} = 20 \text{ kg/s}$

$$\Rightarrow \underline{\dot{W}} = P = \dot{m} \cdot |\Delta h_0| = 20 \cdot |-45.9 \cdot 1000| = \underline{918 \text{ kW}}$$