

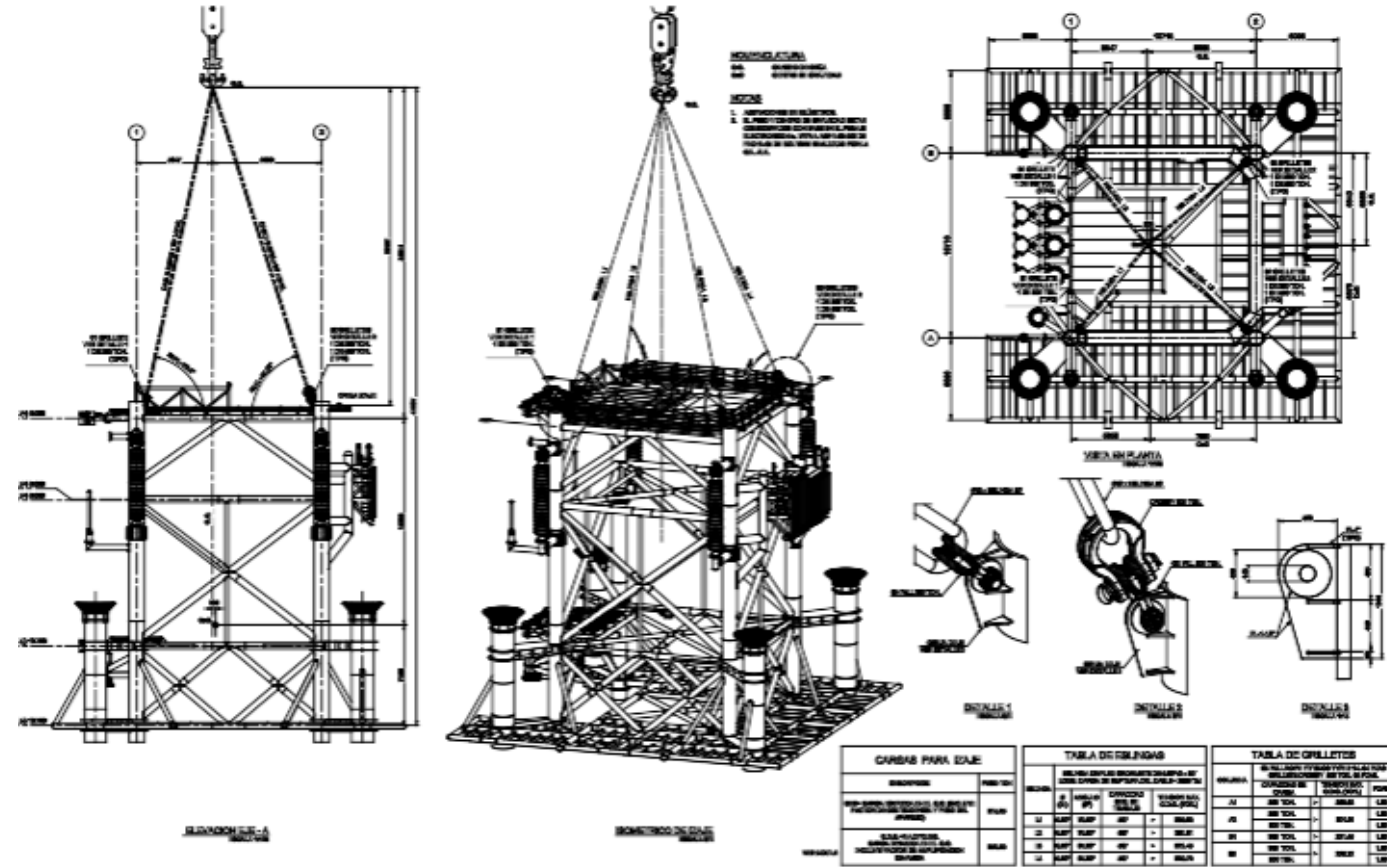
Structural Engineering Course Basic Level



Syllabus Basic Level (12x 2 hours meeting)

1. Material & Element
2. Engineering Mechanic & Structural Behaviour
3. Introduction to Design Code Criteria
4. Environment load 1
5. Environment load 2
6. Environment load 3
7. Environment load 4
8. Preservice – Weighing
9. Preservice - Lifting
10. Preservice - Land Transport
11. Preservice – Sea Transport
12. Final Assignment

Lifting Arrangement



Sling Forces

MEMBER FORCES AND MOMENTS REPORT

MEMBER NUMBER	MEMBER END	GROUP ID	LOAD COND	***** kN *****		
				FORCE(X)	FORCE(Y)	FORCE(Z)
P501-PDY1	P501	PDY	4000	9828.1377	1334.2141	-0.6788
			4100	9726.8271	1331.2980	-1.6778
			4200	7808.6665	1075.7202	-1.8225
			4300	9929.4365	1337.1292	0.3201
			4400	11847.5967	1592.7068	0.4649
	PDY1		4000	9828.1377	1334.2141	-0.6788
			4100	9726.8271	1331.2980	-1.6778
			4200	7808.6665	1075.7202	-1.8225
			4300	9929.4365	1337.1292	0.3201
			4400	11847.5967	1592.7068	0.4649

Load Factors

Combination Factor	Factor					Total Factor
	SLF	TEF	YEF	DAF	CF	
a1	1.1	1.05	1.0	1.0	1.0	1.155
a2	1.1	1.05	1.0	1.05	1.0	1.212
a3	1.1	1.05	1.0	1.05	1.15	1.394
a4	1.1	1.05	1.0	1.05	1.30	1.576

Lifting Gears Calculation

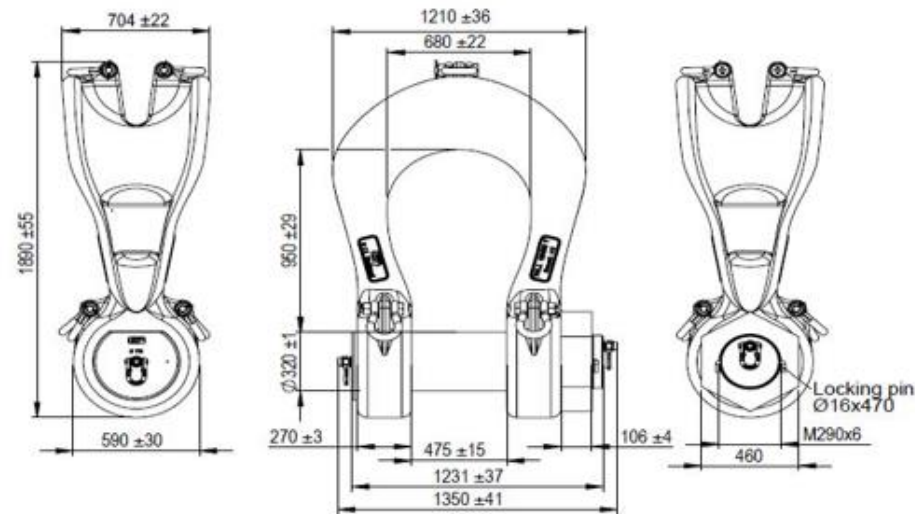
Factored sling load	Fsf =	11000.0	kN, From SACS lifting case output.
Combination factor for padeyes calc	a4 =	1.576	
Basic sling load = Fsf/a4	Fs =	711.5	MT (Metric Ton)
Basic load for padeye, Fpb:	Fpb =	711.5	MT
Weight contingency	γwc =	1.00	(included in SACS model)
COG shift	γcog =	1.00	
DAF	γdaf =	1.05	
SKL	γskl =	1.10	
Tilt Factor	γt =	1.05	
Yaw Factor	γw =	1.00	
Sling design load, Fsd=Fs*γwc*γcog*γdaf*γt*γw*γskl	Fsd =	862.9	MT 1902.6 Kips DNVGL-ST-N001 (2016) sec.16.3.4.1

Dimension of Shackle

Provided Shackle:		Green Pin P-6043, SLPGP1550	with the following data :	
Shackle Safe Working Load (SWL)	SSWL =	1560.0	MT	
Shackle Min Breaking Load MBL	SMBL =	7800.0	MT	
Proof load test result	PLT =	1560.0	MT	(presumed, not available yet)
Pin Diameter	d _p =	320.0	mm	
Jaw Width	Bsh =	460.0	mm	
Inside Length	Hsh =	950.0	mm	
Check Fsd<SSWL*DAF	862.9 MT <	1638.0	MT	DNVGL-ST-N001 (2016) sec.16.5.2.5 ...Accepted
Unity Check	UC =	0.527		
Check Fsd<SMBL/3	862.9 MT <	2600.0	MT	DNVGL-ST-N001 (2016) sec.16.5.2.5 ...Accepted
Unity Check	UC =	0.332		
Check Fsd<PLT	862.9 MT <	1560.0	MT	DNVGL-ST-N001 (2016) sec.16.5.2.5 ...Accepted
Unity Check	UC =	0.553		

Lifting Gears Calculation

Green Pin® Power Sling Shackle
Group : P-6043
Part : SLPGP1550
WLL : 1550t



Dimension of Sling

Nominal safety factor for sling, $\gamma_{sf} = \max(2.3, \gamma_f \cdot \gamma_c \cdot \gamma_r \cdot \gamma_w \cdot \gamma_m)$

γ_{sf}	=	3.26
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DNVGL-ST-N001 (2016) sec.16.4.3

Where

Lifting factor	γ_f	=	1.30
Consequence factor	γ_c	=	1.30
Reduction factor, $\gamma_r = \max(\gamma_s, \gamma_b)$	γ_r	=	1.30
Termination factor	γ_s	=	1.30
Bending factor, $\gamma_b = 1 / (1 - 0.5 / (D_{sh} / d_{sl})^{0.5})$	γ_b	=	1.00
Material factor	γ_m	=	1.35
Wear factor	γ_w	=	1.10

(Ignored as the bending is at the eye sling)

Required Nominal MBL of sling, $MBL_r > \gamma_{sf} \cdot F_{sd}$

MBL _r	=	2815.1 MT
	=	27616.2 kN

DNVGL-ST-N001 (2016) sec.16.4.2.0

Lifting Gears Calculation

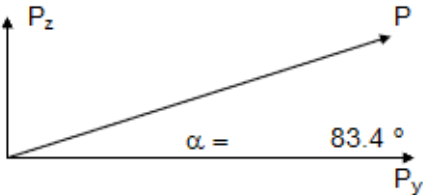
Provide sling rope	=	IWRC 6x36 grade 1770	
Choose sling rope above MBL	=	7750.0 MT	(assumed)
Diameter of Sling	dsl =	162.0 mm	
Unity Check $MBLr/MBL \leq 1$	UC =	0.363	...Accepted

D. Criteria for Padeye Sizing

Diameter of pin hole	$d_h = \text{Max}(dp+2\text{mm}, \text{min}(1.03*dp, dp+6))$	=	326.00 mm	DNVGL-ST-N001 (2016) sec.16.9.5.2
Radius of pin hole	r_h	=	163.00 mm	Pin Hole Radius for the padeye

E. Lift Point Load

Lift Point Design Load	$F_p = F_{pb} * \gamma_{wc} * \gamma_{cog} * \gamma_{daf} * \gamma_t * \gamma_w * \gamma_{skl} * \gamma_c$	=	1121.7 MT
		=	
Load angle	α	=	83.4 degree



P	=	1121.7 MT
P _z	= P sin α =	1114.3 MT
P _y	= P cos α =	128.9 MT

Lifting Gears Calculation

F. Padeye Code Check

Try the following plate thickness :

- Thickness main plate	(t_m)	=	100	mm		
- Max thickness cheek plate 1	(t_{c1})	=	100	mm		
- Max thickness cheek plate 2	(t_{c1})	=	0	mm		
- Total Provide thickness cheek plate		=	100.0	mm		
- Total thickness of padeye (before spacer)	tp	=	$t_m + 2 * t_{c1} + 2 * t_{c2}$			
		=	300.0	mm	Jaw width, Bsh=	460.0
- Actual gap distance between padeye thickness and jaw width, $g1 = Bsh - tp$						
	$g1$	=	160.0	mm		
- Actual gap in each side, $g2 = g1/2$	$g2$	=	80.00	mm		
- Design gap in each side:	$g3$	=	30.00	mm		
- Req. thk. of spacer plate, $tsp = g2 - g3$	tsp	=	50.00	mm	(each side)	
- Provided thk. of spacer plate, tsp	tsp	=	50.00	mm		...Accepted
<hr/>						
- Gap distance Inside Length		=	$H + dp/2 - r_m - ds$	>	$1.3 * d$	
		=	308.00	mm	>	210.6 mm ...Accepted
- Stiffener plate thickness	ts	=	38	mm		
- Total thickness of padeye (with spacer)	tp'	=	$t_m + 2 * (tsp + t_{c1} + t_{c2})$			
		=	400.0	mm	>	Jaw width, Bsh= 460.0 ...Accepted
- Main plate radius	r_m	=	max of (1.25 x Diameter pin hole or Diameter pinhole/2 + 3")			
		=	640	mm		
- Cheek plate radius, $rc = r_m - tc$	r_c	=	main plate radius - thickness cheek plate			
		=	540	mm		
- f_{yield} main plate	f_{ym}	=	355	MPa		
- f_{yield} cheek plate	f_{yc}	=	355	MPa		

Lifting Gears Calculation

Geometric check of Padeye (as per DNVGL ST N001 Sec. P.2.4.2)

Outside radius of the padeye main plate shall be no less than the diameter of the pin hole

$$r_m \geq p+6))$$

$$640.0 \text{ mm} \geq 326.0 \text{ mm}$$

...Accepted

The pad eye thickness at the hole shall not be less than 75% the inside width of the shackle

$$tp' \geq 75\% \cdot B_{sh}$$

$$400.0 \text{ mm} \geq 345.0 \text{ mm}$$

...Accepted

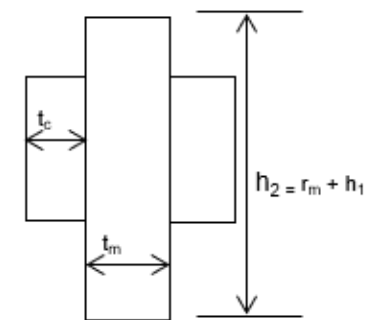
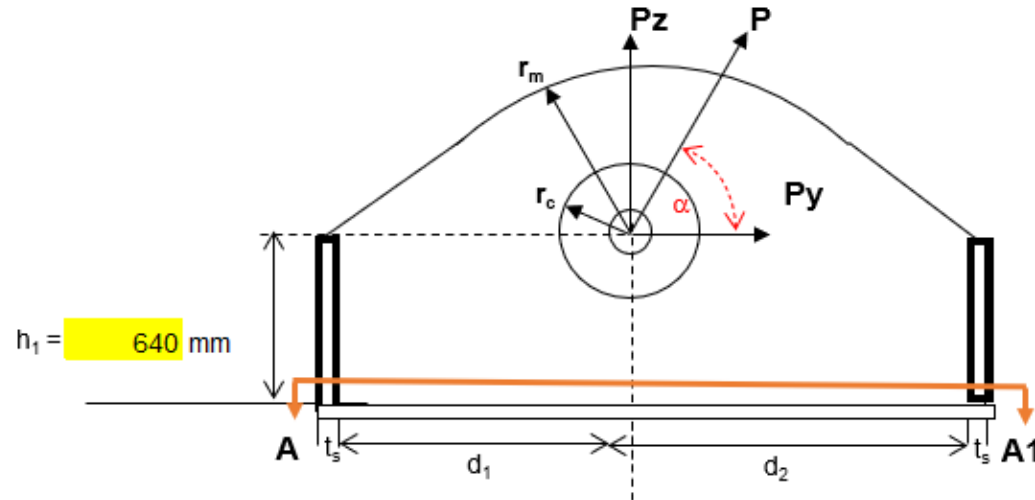
Shackle pin max. diameter including tolerance should be considered in order to ensure that the pin will enter the hole

$$r_{m, \min}(1.03 \geq dp$$

$$356.0 \text{ mm} \geq 320.0 \text{ mm}$$

...Accepted

Sketch of Padeye



$$t_m = 100.0 \text{ mm}$$

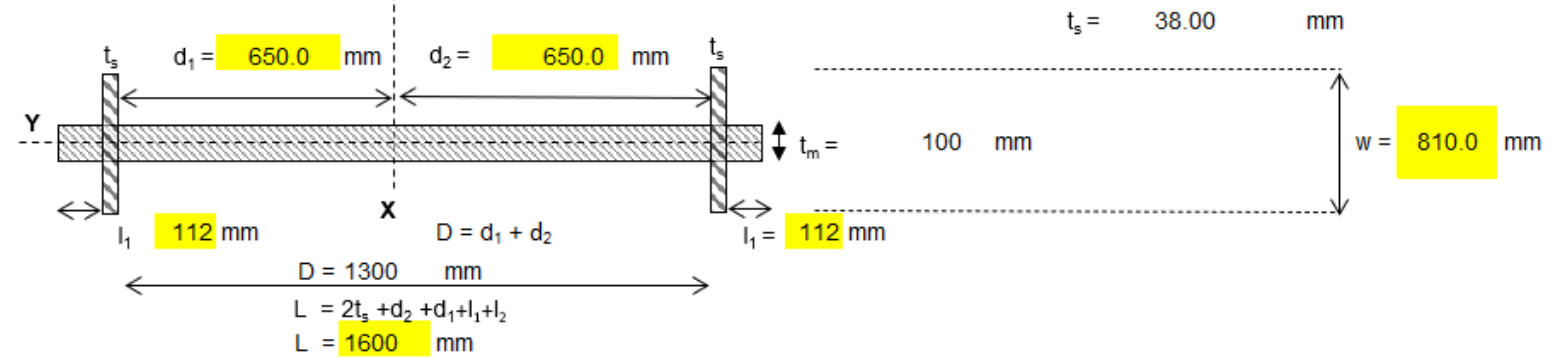
$$t_c = 100.0 \text{ mm}$$

$$h_2 = 1280.0 \text{ mm}$$

Lifting Gears Calculation

section A-A1 :

(Assumed as Beam Section)



F.1 Pin Bearing Stress

Check Bearing Capacity

Allowable Bearing Stress

$$= 0.9 \times f_{yield}$$

$$= 319.5 \text{ MPa}$$

Bearing Capacity :

$$A = 0.9 \times f_{yield} \times A$$

$$= d_p \times t / \sqrt{2}$$

$$= 67882.3 \text{ mm}^2$$

Bearing capacity

$$= 0.9 \times f_{yield} \times A$$

$$= 2210.8 \text{ MT}$$

Bearing Load

$$F_b = 1121.7 \text{ MT}$$

Unity Check

$$= 0.510$$

< Bearing Capacity

< 1

...Accepted

Bearing stress due to radial pressure $f_b = F_b / (t \times d_p)$

$$= 114.6 \text{ MPa}$$

Unity Check

$$= 0.360$$

< $0.9 \times f_y$

< 1

...Accepted

Max stress due to pressure between elastic bodies

Max stress

$$f_{he} = 0.591 \times \sqrt{F \times E / (T \times K \times B)} \leq 2.5 F_y$$

$$= 383.9 \text{ MPa} < 888 \text{ MPa}$$

Unity Check

$$= 0.430$$

< 1

...Accepted

Lifting Gears Calculation

Where,

Bearing Load

$$F_p = 11004013.3 \text{ N}$$

Bearing load per unit length of contact surface, $p = F_p/t_p$

$$p = 36680.0 \text{ N/mm}$$

Young's Modulus

$$E = 200,000.0 \text{ MPa}$$

Hertz pressure :

For 1560MT shackle Green Pin P-6043, SLPGP1550

Pin hole dia.

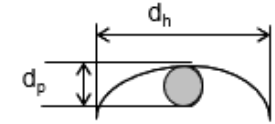
$$d_h = 326.0 \text{ mm}$$

Pin dia.

$$d_p = 320.0 \text{ mm}$$

$$K_d = (d_h \times d_p) / (d_h - d_p)$$

$$= 17386.7$$



F.2 Pin Pull-Out Shear

Allowable Shear Stress

$$= 0.4 \times F_y$$

Shear force, P_y

$$P_y = 1121.7 \text{ MT}$$

$$= 11004013.3 \text{ N}$$

Effective shear section area

$$A_e = (2 \times (r_m - d_h/2) \times t_m) + (4 \times (r_c - d_h/2) \times t_c)$$

$$= 246200.0 \text{ mm}^2$$

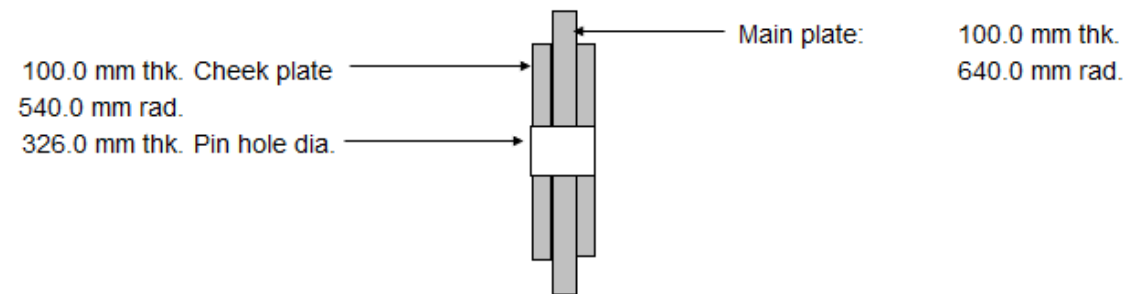
Shear Stress, $f_s = P_y/A_e$

$$f_s = 44.7 \text{ MPa} < 0.4 \times F_y = 142 \text{ MPa}$$

Unity Check

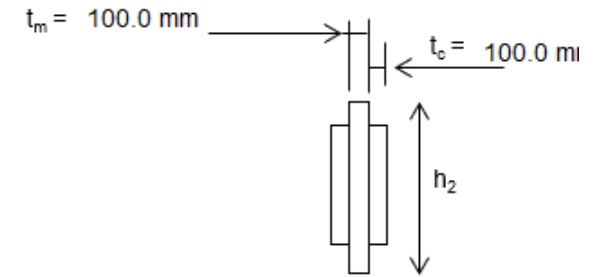
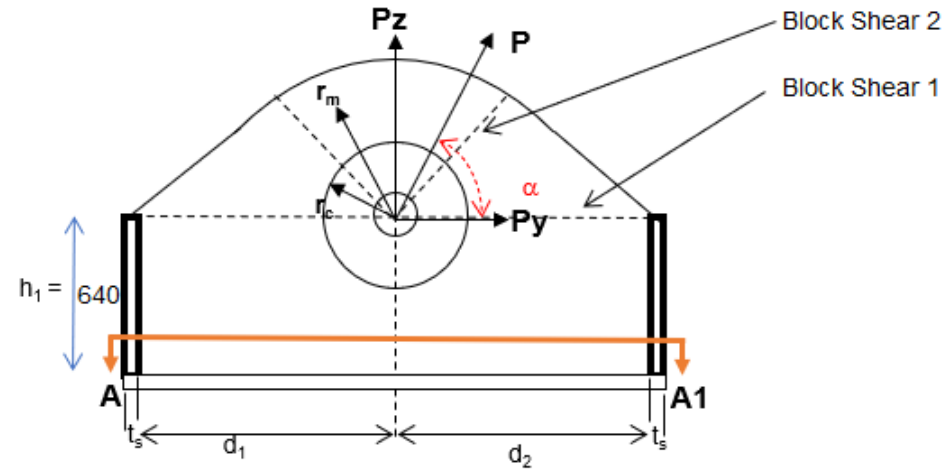
$$= 0.310 < 1$$

...Accepted



Lifting Gears Calculation

F.3 Tension Failure



Refer to AISC D1 - Block Shear 1

Allowable Tensile Stress without pin hole

Maximum Tension

Effective tension section area

Tensile Stress, $f_t = F_t / A_e$

Unity Check

Refer to AISC D1 - Block Shear 2

Allowable Tensile Stress without pin hole

Tensile Capacity

Effective tension section area

$$\begin{aligned}
 &= 0.60 \times F_y \\
 F_t = P_z &= 1114.3 \text{ MT} \\
 &= 10931086.8 \text{ N} \\
 A_e &= \text{Block Shear 1 length} \times t_m \times 0.60 F_y \\
 &= (D - d_h) \times t_m \\
 &= 974 \times 100 \\
 &= 97400.0 \text{ mm}^2 \\
 f_t &= 112.2 \text{ MPa} < 0.60 \times F_y = 213 \text{ MPa} \\
 &= \mathbf{0.527} < 1
 \end{aligned}$$

...Accepted

$$\begin{aligned}
 &= 0.60 \times F_y \\
 &= 213 \text{ MPa} \\
 A_e &= \text{Block Shear 2 length} \times t_m \times 0.60 F_y \\
 &= 2 \times (r_m - d_h / 2) \times t_m
 \end{aligned}$$

Lifting Gears Calculation

Tensile Stress, $f_t = F_t / A_e$
Unity Check

$$= \frac{954 \times 100}{95400.0} \text{ mm}^2$$
$$= 114.6 \text{ MPa}$$
$$= 0.538$$

$$< 0.60 \times F_y = 213 \text{ MPa}$$
$$< 1$$

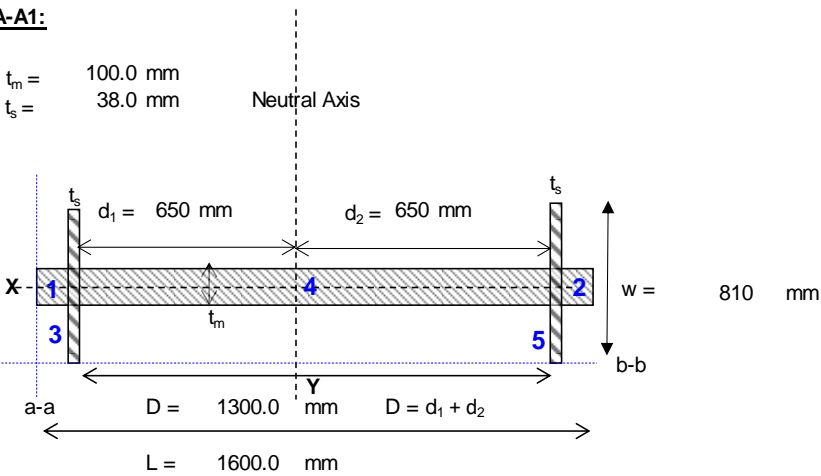
...Accepted

F.4 Combined Stress

Lift Point Design Load

$$P = 1121.7 \text{ MT}$$
$$P_z = P \sin \alpha$$
$$= 1114.3 \text{ MT}$$
$$P_y = P \cos \alpha$$
$$= 128.9 \text{ MT}$$

section A-A1:



No.	b mm	h mm	Area (mm ²)	Ordnat to a-b		A*X' mm ³	A*Y' mm ³	Centre of Gravity	
				x' mm	y' mm			X _{cog} mm	Y _{cog} mm
1	112.0	100.0	11200	56.0	405.0	627200.000	4536000.000	800.00	405.00
2	112.0	100.0	11200	1544.0	405.0	17292800	4536000		
3	38.0	810.0	30780	131.0	405.0	4032180	12465900		
4	1300.0	100.0	130000	800.0	405.0	104000000	52650000		
5	38.0	810.0	30780	1469.0	405.0	45215820	12465900		
Total			213960			171168000	86653800		

Lifting Gears Calculation

No.	Ordinat Relative to COG		A*X^2	A*Y^2	A*X*Y	I _{x0}	I _{y0}	I _x	I _y	I _{xy}
	X mm	Y mm								
			mm4	mm4	mm3	mm4	mm4	mm4	mm4	mm4
1	-744.00	0.00	6199603200	0	0	9333333.333	11707733.33	9333333.333	6.211E+09	0
2	744.00	0.00	6199603200	0	0	9333333.333	11707733.33	9333333.333	6.211E+09	0
3	-669.00	0.00	13775927580	0	0	1682896500	3703860	1682896500	1.378E+10	0
4	0.00	0.00	0	0	0	108333333.3	18308333333	108333333.3	1.831E+10	0
5	669.00	0.00	13775927580	0	0	1682896500	3703860	1682896500	1.378E+10	0
								3502126333	8.5E+10	0

F.5 In-plane Bending

	I _y	=	Σ I _g + Σ (A x x' ²)	
		=	8.500E+10 mm ⁴	
	S _y	=	I _y / (X _{cog})	
		=	1.063E+08 mm ³	
Inplane Moment M		=	P _y * h ₁	
		=	809455276.8 N.mm	
Allowable In-plane Bending Stress	F _b	=	0.66 x f _y yield	
		=	234.3 MPa	
In-plane Bending Stress	f _{ip}	=	M / S _y	
		=	7.618 MPa	
Unity Check		=	0.033	< 1 ...Accepted

F.6 Out-of-Plane Bending

	S _x	=	I _x / (Y _{cog})	
		=	4377657.9 mm ³	= 71.7370 m ³
5% of static sling force will be applied, as per API 20th edition section 2.4.2.				
5% of sling force	F _{op}	=	0.05*P	

Lifting Gears Calculation

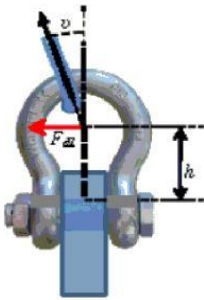
Moment arm length
Out-of-plane Moment M

Allowable out-of-plane Bending Stress

Out-of-plane bending stress

Unity Check

$$\begin{aligned} h_2 &= H_{sh}/2 = 475 \text{ mm} \\ M_{op} &= F_{op} \cdot h_2 = 2.613E+08 \text{ N.mm} \\ f_{op} &= 0.75 \times f_{yld} = 266.25 \text{ MPa} \\ &= M / S_x = 59.7 \text{ MPa} \\ &= \mathbf{0.224} < 1 \end{aligned}$$



...Accepted

Axial Tensile Stress

Allowable tensile stress

Axial Tensile Force (P_z)

Tensile Area (A_s)

Axial Tensile Stress

Unity Check

$$\begin{aligned} F_t &= 0.6 \times F_y = 213 \text{ MPa} \\ &= 1.093E+07 \text{ N} \\ &= 2 \cdot (((r_m - d_h)/2) \cdot t_m) + 2 \cdot ((R_c - d_h)/2) \cdot t_c \\ &= 2.140E+05 \text{ mm}^2 \\ &= P_z / A_s \\ f_a &= 51.1 \text{ MPa} \\ &= \mathbf{0.24} < 1 \end{aligned}$$

...Accepted

Combined Axial Tension and Bending

Unity Check

$$\begin{aligned} UC &= f_a / F_t + f_{ip} / F_b + f_{op} / F_b \\ &= 0.24 + 0.0 + 0.224 \\ &= \mathbf{0.497} < 1 \end{aligned}$$

...Accepted

Horizontal Direction (Main Plate)

Allowable shear stress

Shear force, max(P_y , F_{op})

Shear Area (A_s)

Shear stress

Unity Check

$$\begin{aligned} &= 0.4 \times f_{yld} = 142.0 \text{ MPa} \\ &= 1.265E+06 \text{ N} \\ &= 2.140E+05 \text{ mm}^2 \\ &= P_y / A_s = 5.9 \text{ MPa} \\ &= \text{Shear Stress} / \text{Allowable Shear Stress} \\ &= \mathbf{0.042} < 1 \end{aligned}$$

...Accepted

Lifting Gears Calculation

F.7 Von Misses Stress

Von Misses Stress

where:

total stress in x-dir.

total stress in y-dir.

shear stress

Allowable stress

Unity Check

$$\sigma = (f_x^2 + f_y^2 + 3f_s^2)^{0.5}$$

$$f_x = 58.71 \text{ MPa (due to in-plane moment and tension)}$$

$$f_y = 59.70 \text{ MPa (due to out of plane moment)}$$

$$f_s = 5.91 \text{ MPa (due to } P_v)$$

$$\sigma = 84.353 \text{ MPa}$$

$$\sigma_a = 0.75 \times F_y$$

$$= 266.25 \text{ MPa}$$

$$\sigma/\sigma_a = 0.317 < 1$$

...Accepted

G. Welding Sizing

Weld of cheek plate to main plate

Weld type : E70XX electrode with Fv

Allowable Shear Stress for Fillet Weld is the lesser of :

Allowable Shear Stress

Load at Check Plate

Effective weld length

Minimum required fillet Size, $t_w = P_{cp}/(F_{sw} \times l_w)$

Size of fillet welds: Min 0.7 x Thk Cheek Plate

Min 0.7 x Thk Cheek Plate

Max Thk Cheek Plate-2mm

Used fillet weld size

$$F_{exx} = 311.4 \text{ MPa}$$

$$F_{sw} = 0.6 \times (1/\sqrt{2}) \times F_{exx}$$
$$= 66.1 \text{ MPa}$$

$$= 66.1 \text{ MPa}$$

$$P_{cp} = P_s \times (t_c / (2 \times T_c + t_m))$$
$$= 3668004.4 \text{ N}$$

$$l_w = \text{half perimeter of cheek plate}$$
$$= 1696.5 \text{ mm}$$

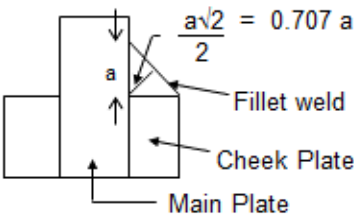
$$t_{wr} = a = 32.73 \text{ mm}$$

$$= 70.0 \text{ mm}$$

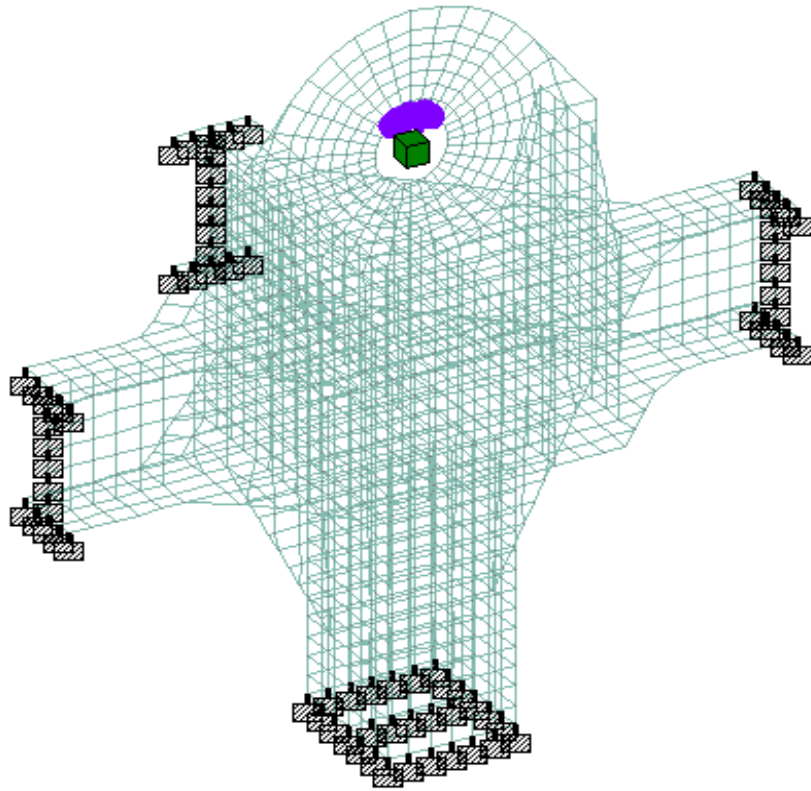
$$= 98.0 \text{ mm}$$

$$t_{ws} = 70.0 \text{ mm} < t_{wr} \quad 32.73 \text{ mm}$$

...Accepted



Lifting Gears Calculation



Max Von Mis
MPa

<= 1.67
16.4
31.1
45.9
60.6
75.4
90.1
105
120
134
149
164
179
193
208
223
>= 237

