Line	Satellite number	Class	International designator			Epoch (UTC)		$\dot{M}/2$ [rev/day]	$\ddot{M}/6$ [rev/day^2]	B^* (drag term)	Model	TLE number	Check
			Year	Launch	Piece	Year	Day of the year (as fraction)						
1	5 5 1 2 4	U	9 8	0 6 7	U R	2 3	0 8 6 . 1 6 7 7 8 9 9 4	S . 0 1 5 4 0 4 0 7	S 1 4 1 0 2 - 2	S 2 6 6 3 4 - 2	0	9 9 9	1

Line	Satellite number	Inclination i [deg]	Right ascension Ω [deg]	Eccentricity e	Argument perigee ω [deg]	Mean anomaly M [deg]	Mean motion n [rev/deg]	Number of revolutions	Check
2	5 5 1 2 4	5 1 . 6 2 3 7	5 . 0 0 1 7	0 0 1 0 7 5 7	1 8 8 . 3 8 6 9	1 7 1 . 6 9 6 3	1 6 . 0 1 8 7 8 1 9 9	1 3 7 9 2	