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**The utility of ring springs in seismic isolation systems**

*K E Hill, University of Canterbury*

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A thesis submitted for the degree of

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## ABSTRACT

Ring springs are frictional devices consisting of inner and outer ring elements assembled to form a spring stack. External load applied to the spring produces sliding action across mating ring interfaces. Large amounts of energy, as much as 60–70% of the total cycle energy, may be absorbed in overcoming frictional forces.

This thesis details the characteristics and dynamic behaviour of ring spring systems and describes the design and testing of a seismic isolation system that uses ring springs.

Initially the characteristics and fundamental dynamic behaviour of single-degree-of-freedom mass/ring spring systems are studied. This study uses a model based upon the non-linear force/deflection characteristics of the ring spring. A prototype ring spring cartridge that enables dynamic inputs to be applied to a ring spring was then designed and subjected to short duration shock excitation. Experimental results are compared with those given by computer simulation and are seen to be in good agreement.

Ring springs have been identified as suitable devices for use in earthquake-resistant structures. A particularly attractive candidate for use of ring springs is in protecting columnar structures during earthquakes. To enable further study, a pivotal rocking seismic isolation system was developed. So that computational analyses of these systems could be undertaken, the ring spring model has been incorporated within the computer programme RUAUMOKO. Dynamic analyses using RUAUMOKO show that pivotal rocking isolation systems can significantly reduce structural loads during short period type earthquakes.

Subsequently, a pivotal rocking seismic isolation system was designed and manufactured. Shaker table tests were then carried out on the rocking system for a range of earthquake inputs. The experimental results show that for columnar structures with fundamental periods in the range of dominant spectral accelerations, structural loads can be significantly reduced during short period type earthquakes. Experimental results compare well with those given by computer simulation, thus confirming the effectiveness of the isolation system. *& confirming correctness of simulation.*

The work outlined in this thesis has established a basis from which further research can be undertaken. The pivotal rocking seismic isolation system developed has potential application to protecting a wide range of columnar structures during short period type earthquakes.

## PUBLICATIONS

- Hill, K.E. *Aseismic design: A review of methods, developments and applications*, Dept. of Mech. Eng., University of Canterbury, Aug. 1992, 53 pp.
- Hill, K.E. *Fundamental dynamic characteristics of ring springs*, Proc. Vibrations Assoc. of N.Z. Annual Conf., Christchurch, May 1993, pp 182-90.
- Hill, K.E. *Design aspects for incorporating ring springs into practical systems*, Proc. IPENZ Annual Conf., Nelson, Feb. 1994, Vol 1, pp 80-84.  
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- Hill, K.E. *Dynamic properties of ring springs for use as seismic energy dissipators*, Proc. N.Z. Nat. Soc. Earthq. Eng. Annual Conf., Wairakei, Mar. 1994, pp 96-101.
- Hill, K.E. *Dynamic energy absorption utilising ring springs*, Int. Mech. Eng. Congress and Exhibition, Perth, Australia, May 1994, Vol 3, pp 207-212.
- Hill, K.E. *Characteristics and dynamic response of ring spring systems*, IPENZ Trans., Nov. 1994, Vol 21, No. 1/EMCh, pp 6-9.
- Hill, K.E. *A prototype ring spring cartridge for mitigating transient and seismic inputs*, Proc. IPENZ Annual Conf., Palmerston North, Feb. 1995, Vol 2, pp 145-150.  
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- Hill, K.E. *Application of ring springs to seismic isolation systems*, Proc. N.Z. Nat. Soc. Earthq. Eng. Annual Conf., Rotorua, Mar. 1995, pp 21-27.
- Hill, K.E. *Experimental testing and computer simulation of a prototype ring spring cartridge subject to shock excitation*, IPENZ Trans., Nov. 1995, Vol 22, No. 1/EMCh, pp 10-15.
- Hill, K.E. *Seismic isolation of columnar structures utilising ring springs*, Pacific Conf. on Earthq. Eng., Melbourne, Australia, Nov. 1995, Vol 1, pp 101-110.

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## NOTATION

$a$	=	rocking arm radius
$A_i$	=	inner element cross sectional area
$A_o$	=	outer element cross sectional area
$b$	=	height to centre of gravity of column
[C]	=	damping matrix
$E_d$	=	energy absorbed for decreasing load
$E_i$	=	energy absorbed for increasing load
$E$	=	modulus of elasticity
$F$	=	$\mu N$ , frictional force
$F_0$	=	$K_i \Delta$ , force at transition from $K_0$ to $K_i$
$g$	=	gravity
$I_0$	=	$I_G + mb^2 = (4mb^2)/3$ , mass moment of inertia about point 0
$I_G$	=	mass moment of inertia about centre of gravity
$k$	=	spring stiffness
$k^*$	=	either stiffness $K_0$ , $K_i$ , $K_l$ , or $K_d$
$K_0$	=	ring spring stiffness at origin
$K_l$	=	ring spring stiffness between $K_i$ and $K_d$
$K_d$	=	decreasing ring spring stiffness
$K_e$	=	elastic spring stiffness
$K_{eff}$	=	effective spring stiffness (secant stiffness)
$K_i$	=	increasing ring spring stiffness
[K]	=	stiffness matrix
$m$	=	mass
[M]	=	mass matrix
$N$	=	total normal force
$n$	=	total number of interfaces of ring spring
$p\{t\}$	=	time varying load
$P_d$	=	decreasing compressive load
$P_i$	=	increasing compressive load
$r_i$	=	inner ring element mean radius
$r_m$	=	$(r_i+r_o)/2$ , mean radius
$r_o$	=	outer ring element mean radius
{r}	=	influence coefficient vector
$T_1$	=	cycle 1/4 period on stiffness $K_d$
$T_2$	=	cycle 1/4 period on stiffness $K_i$

$T_{lis}$	=	first mode period of isolated system
$T_{lunis}$	=	first mode period of unisolated system
$T_n$	=	period
$t$	=	total time
$u_a$	=	$mg/K_i$ , initial displacement due to load $mg$
$u$	=	displacement of mass
$\ddot{u}$	=	acceleration of mass
$\{v\}$	=	total displacement vector
$\{\dot{v}\}$	=	total velocity vector
$\{\ddot{v}\}$	=	total acceleration vector
$w$	=	relative displacement
$\ddot{w}$	=	relative acceleration
$\{w\}$	=	relative displacement vector
$\{\dot{w}\}$	=	relative velocity vector
$\{\ddot{w}\}$	=	relative acceleration vector
$W$	=	$mg$ , weight
$x_g$	=	ground displacement
$\ddot{x}_g$	=	ground acceleration
$z$	=	input displacement
$\ddot{z}$	=	input acceleration
$\alpha$	=	ring spring taper angle
$\gamma$	=	cycle decay factor
$\delta$	=	deflection of ring spring
$\Delta$	=	ring spring pre-displacement
$\Delta_i$	=	initial spring displacement
$\Delta_i$	=	spring displacement after $i$ cycles
$\Delta_{i+q}$	=	spring displacement after $i+q$ cycles
$\zeta$	=	damping factor
$\theta$	=	angular rotation
$\ddot{\theta}$	=	angular acceleration
$\mu$	=	coefficient of friction
$\omega_{nd}$	=	natural frequency on stiffness $K_d$
$\omega_{ni}$	=	natural frequency on stiffness $K_i$

## INTRODUCTION

### 1.1 GENERAL

Earthquakes plague many regions of the world resulting in significant damage and loss of life each year. Damage caused by several seconds of violent ground shaking can be devastating to communities and countries alike. The destruction earthquakes can cause has been graphically demonstrated in recent years. In 1994 in Northridge, USA, an earthquake of moment magnitude 6.7, killed 57 people and caused damage estimated at US\$18-20 billion (Norton et al. 1994). More recently, in 1995 in Kobe, Japan, approximately 5500 people lost their lives in an earthquake of Richter magnitude 7.2. The direct rebuilding cost as a result of this earthquake has been estimated at approximately NZ\$200 billion (Park et al. 1995).

In today's society of increasingly advanced technology — taller structures, nuclear power generation, increased reliance on lifeline utilities such as water supply and electricity, it is imperative that the nature of earthquakes, and their effect on structures, be understood. This has lead to a vast amount of work being undertaken in the field of earthquake engineering and is evidenced by an enormous amount of published literature. The general aim of such work is to enable a greater understanding of earthquakes and their influence on man-made structures, thus prevent loss of life and minimise damage and economic loss.

### 1.2 OBJECTIVES OF RESEARCH

The initial objective of this study was to examine the various earthquake-resistant (aseismic) design methods used to protect structures and structural systems during earthquakes. This is detailed in Chapter 2.

Based on knowledge of these aseismic design methods, the second objective was to investigate the possibility of utilising a frictional device known as a ring spring in earthquake-resistant structures.

### 1.3 SCOPE AND OUTLINE OF THE THESIS

The following outline describes the scope of the research reported in this thesis. First, the methods and principles utilised in aseismic design are detailed in Chapter 2. Of the methods presented, special discussion is given to that of seismic isolation, a technique that aims to reduce the level of loading experienced by a structure during an earthquake. This is achieved by isolating the structure from the ground motion.

Chapter 3 details the basic characteristics of ring springs. The equations that define the ring spring stiffness characteristics are detailed; these equations form the basis of a model for the ring spring. Design requirements for utilising ring springs in practical applications are then discussed. Subsequently, different types of ring spring configurations are detailed and solutions to the equations of motion for a bi-directional ring spring system are presented.

A prototype ring spring cartridge that allows bi-directional dynamic inputs to be applied to ring springs was designed and manufactured. Chapter 4 details the hysteresis characteristics obtained from experimental tests conducted on the cartridge.

Utilising these hysteresis characteristics, Chapter 5 then details computer simulation analyses that predict the dynamic behaviour of single-degree-of-freedom (SDOF) mass/ring spring systems. The study considers the effect of applying a pre-displacement and a pre-load to the ring spring.

To assess the accuracy of computer simulation results, experimental tests were undertaken. Chapter 6 details experimental testing of a mass/ring spring cartridge system. Testing is carried out on a single-axis shaker table facility and experimental results are determined for various pre-displacements and dynamic inputs. These results are then compared with those given by computer simulation.

The possibility of utilising ring springs in earthquake-resistant applications is examined in Chapter 7. Discussion considers possible situations identified as suitable for use of ring springs. Of these, seismic isolation systems are seen as offering considerable potential. To explore this option, a pivotal rocking seismic isolation system has been developed. Details of this system, as well as results of computer simulation analyses, are presented. To

facilitate this modelling, the ring spring model has been incorporated within the computer program RUAUMOKO.

Chapter 8 details experimental testing of the pivotal rocking seismic isolation system subject to four different short period type earthquakes. Test results are then compared with those given by computer simulation.

Conclusions and recommendations for future research are reported in Chapter 9.

## Chapter 2

# **ASEISMIC DESIGN: METHODS, DEVELOPMENTS AND APPLICATIONS**

### **2.1 INTRODUCTION**

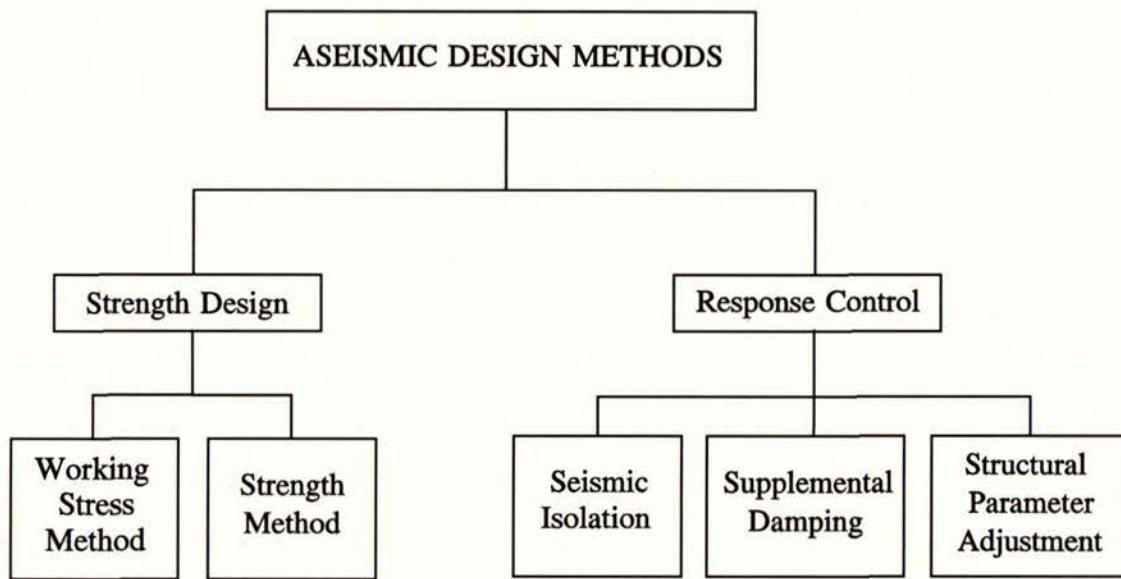
In the design of structures situated in regions of seismic risk, it is necessary that an appropriate earthquake-resistant (aseismic) design technique be employed. This is to minimise the likelihood of loss of life and the collapse of structures during a major earthquake. This chapter examines the various methods used in aseismic design. Discussion covers numerous developments and considers their practical application.

### **2.2 ASEISMIC DESIGN METHODS**

The aseismic design methods utilised in engineering design can be broadly classified into two principal techniques:

- (a) Strength Design: With this technique the structure is designed for strength to ensure it can withstand design level forces. Two procedures are commonly employed in this design approach; these are:
  - Working Stress Method — primarily an elastic design approach.
  - Strength Method — primarily an inelastic design approach.
- (b) Response Control: In this technique the response of a structure is regulated by incorporating specific devices within the system. The following three procedures may be utilised:
  - Seismic Isolation — isolation of the structure from ground disturbances.
  - Supplemental Damping — controlling structural response through increased damping.
  - Structural Parameter Adjustment — incorporating within the structural system actively controlled members.

The interrelationship of these aseismic design methods is presented in Figure 2.1.



**Figure 2.1** Aseismic design methods used in engineering design

Both Strength Design and Response Control techniques can protect structures and components during earthquakes, however, the most appropriate method for a particular application requires careful consideration by the designer. The following sections detail these control techniques.

### 2.3 STRENGTH DESIGN

In aseismic design of structural systems and components, Strength Design techniques are, by far, the most common method employed. As stated earlier, Strength Design involves design of the structure or component for strength to ensure withstand against design level seismic forces. In this section both the Working Stress Method and Strength Method are detailed.

#### 2.3.1 Working stress method

The Working Stress Method is commonly used by mechanical engineers in the design of mechanical systems. Generally, in the design of a specific component, the loading condition is determined, or defined, and detailed design (determination of stresses,

deflections etc.) is based upon this assigned load. The basis of the Working Stress Method is that under working loads, an allowable working stress should not be exceeded. This allowable working stress is usually determined by dividing the yield stress by a design factor.

Generally, structures such as buildings are designed using relatively low seismic factors; their earthquake resistance dependent on post yield ductility. However, for many items of electrical and mechanical equipment, deformation due to yielding of fixings would impair the operation of the equipment, hence, it is generally more appropriate to secure equipment to withstand elastically the earthquake design level forces. Nevertheless, it remains of foremost importance in the design of such plant and equipment that the overall ductility of the component fixings should be considered to avoid sudden failure in the event of the design earthquake load being exceeded. This point is sometimes overlooked by mechanical engineers in the design of equipment fixings required to withstand earthquake induced loads.

### 2.3.2 Strength method

The Strength Method is generally used by civil engineers in structural design. In this method working loads are multiplied by appropriate factors, usually specified in design codes; structural members are then proportioned based upon their ultimate strength — usually concrete at maximum strength and steel at yield. Bridges and buildings are typical structures that are designed using the Strength Method approach.

In structural design, it is generally accepted that it is uneconomic to design a building to remain elastic during the greatest likely earthquake (Park 1987). Thus, the Strength Method employs an inelastic design philosophy. This ensures adequate structural ductility by utilising the principles of capacity design whereby chosen inelastic mechanisms provide both adequate strength and ductility. To ensure yielding occurs within these plastic hinges, other structural elements are provided with increased strength. In conventional buildings these ductile yielding mechanisms are concentrated in beams at beam-column connections. The purpose of incorporating these mechanisms within the system is to allow a considerable amount of the energy associated with the seismic disturbance to be absorbed by the structure itself.

In applying the Strength Method, the designer is able to assess the ductility of the structure in the post-elastic range; the designed collapse mechanisms enable an appreciation of the likely structural response during an earthquake. This fact is not readily appreciated in applying the Working Stress Method.

Engineers, in applying Strength Design techniques in the design of structural systems, are usually required to comply with statutory codes and standards. In New Zealand codes, commonly both the Strength Method and Working Stress Method may be appropriate design methods, except where specifically stated. However, for reasons outlined earlier, clear preference is given to employing the Strength Method. This is emphasised by terming the Working Stress Method the Alternative Method.

The general design philosophy that embodies building codes employing the Strength Method can be summarised as follows: In the event of a major earthquake, significant structural damage may result, however, complete collapse of the structure should be prevented; in the event of a severe earthquake that can be expected during the life of the building, the aim is for minimal or no damage (Park 1987).

In applying the Strength Design approach, the mechanism that provides structural protection is the ability of the structure to resist elastically, or absorb inelastically, the input energy it experiences during a seismic disturbance. A means of enhancing seismic resistance is by controlling structural response, i.e., reducing seismic energy transmitted into the structure. This method, Response Control, is presented in the next section.

## 2.4 RESPONSE CONTROL

Response Control is an aseismic design approach that involves regulating the response of a structure during a seismic disturbance. This approach utilises any of the following methods: Seismic Isolation, Supplemental Damping, or Structural Parameter Adjustment.

In the discussion that follows, the devices these methods employ are described.

### 2.4.1 Passive, active and semi-active control

Response Control techniques may use passive, active or semi-active control.

#### (a) Passive control

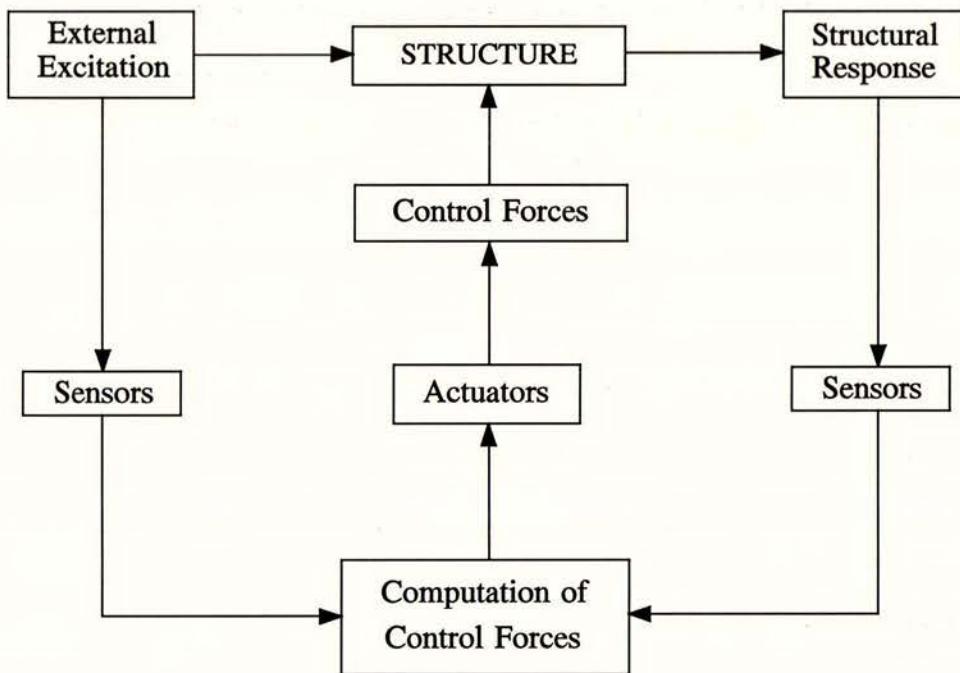
In passive control, passive devices are embodied within the structural system. Passive devices require no power supply; they respond directly to the seismic disturbance and exhibit response characteristics dependent on their physical properties. These devices can either store or dissipate the energy associated with a seismic disturbance. Examples of passive devices for protecting engineering systems from earthquake damage include: lead-rubber bearings, lead extrusion dampers, mild steel energy dissipators, elastic springs, viscous dampers, visco-elastic dampers, and frictional dissipators (Hill 1992, Skinner et al. 1993). Passive devices offer advantages that they are generally inexpensive, reliable and possess well defined characteristics.

#### (b) Active control

In active control, active devices apply control forces at specific locations throughout the structure. These control forces counteract the seismic response of the structure, thereby reducing the forces experienced by structural elements.

Figure 2.2 presents the basic configuration of an active control system (Soong 1988); the components comprise:

- (1) Sensors appropriately positioned within the structure to measure either, the disturbing excitation, structural response variables, or both;
- (2) A control unit to register the sensor inputs and compute the required control forces based on a prescribed control algorithm;
- (3) Actuators that apply the required control forces.



**Figure 2.2** Schematic of an active control system (Soong 1988)

Developed from this configuration are three types of active control system as follows:

- Open-loop control: This system comprises sensors that continually measure only the disturbing excitation; these measurements are used to regulate the actuator control forces.
- Closed-loop control: This system continually measures only the structural response variables; actuator control forces are regulated by these measurements.
- Open-closed loop control: Measurements are continually made of both the disturbing excitation and structural response variables; these are used to regulate the actuator control forces.

Examples of active control devices for protecting engineering systems from earthquake damage include: electro-hydraulic servomechanisms, active mass dampers, active tendons, pulse generators (Hill 1992).

Since active control devices require external energy for their operation, it is imperative that the power supply remains uninterrupted throughout a severe seismic disturbance to maintain structural integrity and performance. This is a major factor limiting widespread

use of active control in protecting engineering systems from earthquake damage. Other disadvantageous features include maintenance requirements as well as the large power requirement. Generally active control systems are more costly and less reliable (due to their complexity) than passive control systems.

Despite these shortcomings, active control has been applied in a number of practical applications to reduce wind-induced vibrations in buildings. For example, a 400-ton auxiliary mass damper is installed on the 59th floor of the Citicorp Building in New York City; the damper controls the first mode bending response of the building (Miller et al. 1988).

(c) Semi-active control

Semi-active control is a control method that combines the characteristics of both passive and active control. In this type of control the structural parameters of the system: mass, stiffness and damping, are modified or adjusted (usually using control actuators) rather than directly applying control forces to counteract the seismic response of the structure (as in active control). Thus, the supply energy requirement of semi-active control systems is generally considerably less than that of active control systems.

As an example of a semi-active control device, Karnopp et al. (1974) discuss a system that incorporates a conventional passive damper whose force across the device is controllable. An example of this type of device is a damper with a variable damping coefficient. It is reported that semi-active control systems can provide many of the performance gains of active systems. These include: improved vibrational control by providing adjustable parameters to suit varying excitation or response characteristics; simpler and less costly hardware than active control systems since semi-active systems require only signal processing and low level power supplies.

Hrovat et al. (1983) compared the performance of a semi-active tuned mass damper (TMD) with both passive and active TMD's; they found that semi-active TMD's offered significant improvements over passive TMD's for reducing wind-induced vibrations in buildings. Furthermore, the performance of a semi-active TMD was shown to be comparable with a fully active TMD.

## 2.5 RESPONSE CONTROL METHODS

This section details the characteristics of the Response Control techniques: Seismic Isolation, Supplemental Damping, and Structural Parameter Adjustment.

### 2.5.1 Seismic isolation

Seismic Isolation is a Response Control technique that aims to protect structures during earthquakes. This is essentially achieved by isolating the structure from the ground disturbance. For over a century, efforts have been made to develop systems that minimise structural damage during earthquakes. Izumi (1988) describes what is possibly the first Seismic Isolation system developed. The design consists of supporting a structure on layers of logs, each layer positioned orthogonally. However, it was not until 1909 that an earthquake-resistant design was patented; this design proposed separating a building from its foundation by a layer of talc (Kelly 1986).

In the past, Seismic Isolation had been resisted by the engineering profession with favour given to firmly attaching a structure to its foundations. However, in recent times, particularly over the last twenty five years, a vast amount of research has been undertaken in the field of Seismic Isolation with some significant advantages over conventional design identified. This has resulted in the development and implementation of numerous systems worldwide. Comprehensive coverage of the history of Seismic Isolation, its development and practical application, is detailed by Lee and Medland (1978), Kelly (1986), Izumi (1988), Tajirian and Kelly (1989), Buckle and Mayes (1990), and Skinner et al (1993).

The following sections discuss aspects of Seismic Isolation systems and detail a number of applications employing various devices.

#### (a) Characteristics of seismic isolation systems

Principally, a Seismic Isolation system incorporates a mechanism that introduces both flexibility and damping into a structural system (characteristically at the base attachment, hence sometimes referred to as Base Isolation). Flexibility is incorporated within the system to lengthen its fundamental period. This action shifts the fundamental period of the

system away from the dominant earthquake energy region, thus significantly reducing the inertia forces induced in the structure. Damping devices are also incorporated within the system; their purpose being to dissipate earthquake energy and resist excessive horizontal displacements at the base of the structure.

Although Seismic Isolation systems may be either passively, actively or semi-actively controlled, few known applications employ active or semi-active control. Passive control on the other hand, has been widely used in practice and the following discussion is confined to this method.

Generally a passive Seismic Isolation system comprises supporting elements and dissipative mechanisms. The former characterises system flexibility, and the latter provides system damping.

#### (b) System flexibility

Several types of supporting elements capable of introducing flexibility into a structural system include: elastomeric bearings, rollers, sliding plates, cable suspensions, sleeved piles, rocking/stepping foundations, air cushions and elastic springs (Hill 1992; Skinner et al. 1993).

An example of a Seismic Isolation system incorporating flexible supports is electrical power transmission equipment mounted on conventional helical springs (Rani and Kumar 1980). In this system, increased flexibility aims to reduce the induced accelerations in both the vertical and the horizontal directions. However, the system does not incorporate additional damping, and so is reliant solely on hysteretic damping within the structure. This condition could lead to excessive relative displacements.

Laminated rubber-steel bearings have also been used to provide system flexibility. Delfosse (1977) developed a system, named GAPEC, that utilises these flexible elements. The system has been applied to the isolation of a 230 kV circuit breaker situated in California (Delfosse 1980). Delfosse reported that for this system the fundamental period was increased from 0.4 sec. (without isolators) to 1.37 sec. (with isolators).

Other types of supporting elements that have been explored as a means of introducing system flexibility include roller bearings and ball bearings. Kelly (1986) describes a seven-storey reinforced concrete building built on spherical bearings. Caspe (1984) proposed supporting an entire building on stainless steel ball bearing clusters and use polytetrafluoroethylene (PTFE) control pads to provide a minimum force to resist service loads.

### (c) Dissipative mechanisms

Practical Seismic Isolation systems also require dissipative elements that absorb the seismic input energy and limit excessive base deflections. These elements can be classified into the following three categories: hysteretic dampers, viscous dampers, and friction dampers.

#### (i) Hysteretic Damping

Hysteretic dampers provide damping by utilising the deformation properties of materials, such as, steel and lead; these dampers generally exhibit displacement-dependent characteristics. Extensive research has been undertaken in New Zealand on the use of steel as a hysteretic damping material. This has resulted in the development of various types of mild steel hysteretic dampers which include: torsional beam devices, cantilever devices (tapered round type and tapered plate), bent round bars, and flexural beam dampers (Skinner et al. 1980; Hill 1992; Skinner et al. 1993).

Research efforts have also concentrated on the property of lead to recrystallize at ambient temperatures. This has resulted in the development of the lead extrusion damper (Robinson and Greenbank 1975, 1976) and the lead-rubber bearing (LRB) (Robinson and Tucker 1977, 1981; Robinson 1982). A typical LRB incorporates a cylinder of lead centrally located within an elastomeric bearing. Recently, Buckle (1989) reported that the LRB was in use in over 50% of all Seismic Isolation structures worldwide, and in more than 80% of all isolated structures in the USA.

## (ii) Viscous damping

Generally, viscous dampers use either a liquid medium or a specific compound possessing visco-elastic characteristics. The damping force of these devices is normally a function of velocity.

As an example of a practical system employing viscous damping for increased seismic protection, a spring-dashpot system has been applied to a machine and foundation at a diesel generator station (Huffmann 1980; Tezcan et al. 1980). This system provides flexibility and damping and can offer seismic protection for vertical, horizontal and rocking motions. It was found that the inclusion of the dashpots significantly reduced horizontal displacements measured at the floor level, and it is suggested that this scheme could be utilised to isolate nuclear power plant reactor buildings during earthquakes.

In Japan, a seismically isolated building supported on laminated bearings and oil dampers has been subjected to numerous earthquakes since its construction in 1987 (Higashino et al. 1988). From observed records it was found that the measured acceleration of the superstructure was less than half that of the ground acceleration.

In another scheme in Japan, two full size buildings, one employing Seismic Isolation using laminated rubber bearings and viscous dampers, the other unisolated, were constructed side by side (Izumi and Yamahara 1988). Site tests have been carried out, as well as observations of thirty earthquakes that occurred throughout a one year period. Seismic Isolation was found to be highly effective in reducing structural accelerations. The acceleration levels of the isolated building were between 1/6 and 1/3 of those experienced by the unisolated building.

Also in Japan, viscous shear dampers have been applied as dissipative devices in over one hundred prestressed concrete railway bridges (Kitta et al. 1973; Ishiguro et al. 1977). These dampers, in providing an increase in energy absorption, seek to reduce the relative displacements at the top of the bridge piers.

Since viscous dampers are velocity dependent devices, they can be effective over both large and small displacements. This has advantages where isolation from ambient ground

noise is a requirement such as in semi-conductor manufacturing factories (Fujita 1985; Fujita et al. 1988; Tyler 1991). However, in some applications viscous dampers have drawbacks. A disadvantage in their use for seismic protection of buildings, is that under service loadings (such as wind loads) the structure will drift; hence, the system requires wind stabilisers. Maintenance requirements may also present problems, especially if the devices are left unused for a long period of time before being called into action during a major earthquake.

### (iii) Frictional damping

Frictional damping is a dissipative mechanism that is reliant on relative motion between contacting surfaces. A system employing frictional elements can be designed such that a designated pre-load must be overcome before relative motion is initiated. This feature enables the system break away force to resist the maximum service loadings anticipated, e.g. wind loading, yet when acted upon by an earthquake, relative motion results and the input energy is dissipated. However, for frictional devices that possess slip-stick characteristics, high frequency accelerations may be generated and these may be damaging to sensitive items of equipment housed in structures (Kelly 1986).

As an example of a system employing frictional elements, bridges often employ PTFE sliding bearings to accommodate relative displacements arising from temperature variations, creep and concrete shrinkage. These motions can be considered quasi-static; however, Tyler (1977a) completed tests on a number of PTFE bearings to determine their frictional characteristics under earthquake conditions for possible use in Seismic Isolation buildings. Test results showed that friction increases as speed increases or as temperature decreases. The paper suggested that systems incorporating PTFE bearings could employ laminated rubber bearings to provide the required self-centring action.

Kelly and Beucke (1983) examined a Seismic Isolation system that employed both rubber bearings and frictional skids. For this system, as the structure is displaced the shear action of the rubber bearing causes a reduction in the bearing height; this transfers a portion of the structure load from the rubber bearings to the skids, thus energy is dissipated by sliding action. An attractive feature of this system is that it offers a fail-safe mechanism should displacements greater than the design level occur; this fail-safe characteristic has been

demonstrated by shaker table testing.

A practical example of a sliding bearing Seismic Isolation system has been implemented in Japan. The system, named TASS, comprises rubber bearings, PTFE sliding bearings and stainless steel bearing plates. Description of the system, as well as details of theoretical modelling and experimental testing is reported by Kawamura et al. (1988), Hisano et al. (1988) and Nagashima et al. (1988). The advantages of this system include:

- for any given dynamic input the structure should not resonate since the frictional sliding mechanism does not have a defined natural period;
- the superstructure is stably supported since the bearings do not deform excessively due to the sliding action;
- the transmitted horizontal input force is limited to the value of the frictional force.

An undesirable property of many frictional damping devices is that they are not self-centring, hence after an earthquake permanent off-set between sliding components may result. Various innovative schemes have been proposed to overcome this problem. One such scheme utilises a device known as a resilient-friction base isolator (R-FBI) (Mostaghel 1986). The R-FBI consists of teflon coated flat rings that slide across each other in conjunction with a central rubber core. The vertical weight is carried entirely by the sliding elements, the rubber elements providing the restoring force for the system. Frictional pre-load enables design for a defined break away force to resist maximum service loadings.

Another scheme, proposed by Zayas et al. (1990), makes use of a friction pendulum system (FPS). The system consists of an articulated friction slider that moves within a concave or convex semi-spherical surface; this motion produces a self centring action. An advantageous feature of this system is that the period of vibration, determined by the radius of curvature of the semi-spherical surface, is independent of the supported mass. Additionally, the centre of rigidity of the connections acting as a group always coincides with the centre of mass of the structure, thus torsional motions are minimised; this has significant benefits in isolating asymmetric structures. Potential applications identified for the FPS include: symmetrical and asymmetrical buildings, low rise buildings, highway bridges, water tanks, museum sculptures, electrical circuit breakers, and computer equipment.

Takai et al. (1988) performed analytical and experimental studies on a friction damper intended for use as a Seismic Isolation device. In this system, a link mechanism transforms translational motion to rotational motion; energy dissipation is obtained by relative motion between friction pads and stainless steel plates. The acceleration level induced in the structure incorporating the friction damper was found to be between 1/3 and 1/2 of that of the system without the damper.

Godden et al. (1980) detailed a proposed Seismic Isolation scheme for protecting a High Voltage Electron Microscope (HVEM). The design requires that the HVEM be isolated from ambient ground noise for normal operational use, and, be capable of resisting earthquake damage. In the proposed system, the HVEM is supported on air bags and incorporates coulomb friction dampers. In a similar situation requiring Seismic Isolation, Kashiwazaki et al. (1988) and Tanaka et al. (1989) describe the design of a three-dimensional Earthquake and Microtremor Isolation Floor system. This system has been installed for use in the semiconductor manufacturing industry.

A number of Seismic Isolation systems have been developed for seismic protection of equipment in industrial facilities (Fujita 1985). These systems utilise roller bearings, coil springs, oil dampers, laminated rubber bearings and friction dampers.

To gain an understanding of the effectiveness of various types of dampers, Fujita et al. (1988) undertook a study of hysteretic, viscous and frictional dampers. The intention was to determine which damping device could provide the most desirable means to protect equipment housed within industrial buildings. Theoretical analyses and experimental tests were performed using these dampers attached to a model two storey steel frame building. Appropriate scaling factors were utilised to represent a 211 tonne two storey steel frame building supported on four 490 kN rubber bearings; earthquake inputs were correspondingly scaled. It was found that dampers possessing roundish hysteresis loops were the most desirable for Seismic Isolation of industrial buildings. The viscous shear damper was found to be the most suitable damper for protecting semiconductor manufacturing factories containing sensitive equipment; however, for nuclear power installations the viscous damper was considered inappropriate and elasto-plastic steel dampers, LRB's and high damping rubber bearings were suggested.

(d) Advantages of seismic isolation

The principal aim of a Seismic Isolation scheme is to isolate a supported structure from seismic disturbances, hence, reduce the level of excitation transmitted into the structure. The advantages of Seismic Isolation can be summarised as follows.

- (1) The benefits of seismically isolating structures include (Buckle 1985):
  - simpler design procedures;
  - use of non-ductile forms or components;
  - construction economies;
  - greater protection against earthquake induced damage (both structural and non-structural);
  - confinement of damage to a readily replaceable element.
- (2) Seismic Isolation can provide economic benefits; numerous applications employing Seismic Isolation have shown it to be cost effective in terms of both initial costs and life cycle costs (Mayes et al. 1984, 1990). As an example, Seismic Isolation of the Wellington Central Police Station cost 10% less than that of a ductile frame design (McKay et al. 1990).
- (3) For seismically isolated structures subjected to earthquakes possessing response spectra whose spectral accelerations reduce at longer periods (similar to that of El Centro 1940 N-S), inertia forces and interstorey drifts of multi-storey structures can be considerably reduced. For earthquakes greater than the design level, Seismic Isolation buildings show fewer plastic hinges and much lower ductility demands when compared with conventional ductile design structures (Andriano and Carr 1991).

(e) Seismic isolation applications in New Zealand

New Zealand has been recognised as a world leader in Seismic Isolation technology. Many innovative Seismic Isolation devices and systems have been developed in New Zealand and adopted both within the country and overseas. The following discussion details New Zealand applications. Coverage of Seismic Isolation systems utilised worldwide is detailed in a recently published book by Skinner et al. (1993).

At present seven buildings in New Zealand employ Seismic Isolation. A summary of these structures and the isolation devices used are:

- (1) William Clayton Building, Wellington: lead-rubber bearings  
(Megget 1978)
- (2) Union House, Auckland: sleeved piles and steel cantilever dissipators  
(Boardman et al. 1983)
- (3) Central Police Station, Wellington: sleeved piles and lead extrusion dampers  
(Charleson et al. 1987)
- (4) Wellington Newspapers Printing Press, Lower Hutt: lead-rubber bearings  
(Dowrick et al. 1991)
- (5) Parliament House, and,
- (6) Parliament Library, Wellington: high damping bearings and lead-rubber bearings  
(Poole and Clendon 1992; Robinson et al. 1995)
- (7) Museum of New Zealand, Wellington: lead rubber bearings  
(Robinson et al. 1995).

In New Zealand, 49 bridges employ Seismic Isolation systems (Skinner et al. 1993); the majority of these applications use LRB's.

Two structures have been constructed that utilise rocking/stepping motion. A chimney, located at the Christchurch airport, rests on lead pads and is designed to rock during strong seismic excitation. In this system two cantilever type hysteretic dampers provide additional energy dissipation (Sharpe and Skinner 1983).

Another structure, the South Rangitikei Rail Bridge, comprises twin concrete legs designed to alternately lift off elastomeric bearings during seismic excitation. To provide damping and control displacements, torsional beam hysteretic dampers are attached to each leg (Cormack 1988).

In New Zealand, several items of high voltage electrical equipment have been protected using Seismic Isolation. Capacitor banks at Haywards substation in Lower Hutt are seismically isolated using rubber bearings and hysteretic steel dampers (Cousins et al. 1991; Skinner et al. 1993). In addition, circuit breakers as well as other items of electrical

equipment, have been fitted with Belleville washer dampers. These are discussed in Chapter 7.

### **2.5.2 Supplemental damping**

It has been realised for many decades that a method that provides additional damping (to augment the damping inherent in a structural system), can be beneficial in limiting the maximum response of a structure during a seismic disturbance. This constitutes the basic concept of Supplemental Damping, which involves controlling structural response by increasing system damping; this is achieved by incorporating suitable damping devices within a structure. In Supplemental Damping systems, two types of damping device are commonly used; these are auxiliary mass dampers and mechanical dissipators. Discussion on these dampers follow.

#### **(a) Auxiliary mass dampers**

Among the different types of the auxiliary mass dampers, the tuned mass damper (TMD) has attracted considerable attention. The use of TMD's to minimise excitations set up by rotating equipment is well established. In the case of narrow bandwidth sinusoidal systems the linear TMD is very effective, however, for broad band response spectra as in earthquakes, the linear TMD is less effective.

Sladek and Klinger (1980), motivated by published literature that reported the use of TMD's could reduce wind induced accelerations of high rise buildings by as much as 40%, investigated the possible use of vibration absorbers in reducing the seismic response of tall buildings. In their study, they considered the response of a 25 storey building employing the following three devices:

- (1) a tuned mass damper (TMD);
- (2) a variable tuned mass damper (VTMD) — the device adjusts its frequency characteristics to suit changes in the system frequency as the building undergoes inelastic deformation;
- (3) an anticipatory vibration absorber (AVA) — an active control unit continually records input accelerations and computes the required forces (applied by an actuator) to counteract the response of the structure.

For this study, computer analyses were undertaken for a building (with a natural period of 1.9 seconds) subjected to the first 30 seconds of the El Centro 1940 N-S earthquake. Both linear and non-linear (degrading stiffness) analyses were performed. It was found that the TMD, VTMD and AVA were ineffective in reducing the maximum seismic response of tall buildings.

Despite the above results, a TMD has been installed in a 125 m high steel tower in Japan (Kitamura et al. 1988). The TMD consists of masses of 10 tonne in the x-direction and 15 tonne in the y-direction. The total structure mass is 1950 tonne. The maximum displacement amplitude of the TMD was set to  $\pm 1$  m. The first and second mode periods of the structure were 2.25 and 0.51 seconds respectively in the x-direction, and were 2.70 and 0.57 seconds respectively, in the y-direction. Results of a computer analysis of the system using the El Centro 1940 N-S earthquake, gave reductions in maximum displacements at the top floor of 40% in the x-direction and 30% in the y-direction as compared with a similar system without a TMD. Storey shear forces were almost unchanged for the upper floors, with reductions between 5% and 30% for the middle and lower floors. Overturning moments were reduced by about 30% in both the x and y directions. The performance of the system has also been observed under actual earthquake loading; it is reported that the tower, comprising of much exterior glazing, sustained no damage during an earthquake of magnitude 6.7.

(b) Mechanical dissipators

Mechanical dissipators, possessing similar properties to the dissipative elements employed in Seismic Isolation systems, are used in Supplemental Damping systems; such devices utilise hysteretic damping, viscous damping, and frictional damping.

Supplemental Damping devices can be incorporated in cross bracings used in structures to resist in-plane shear forces. A typical system comprises diagonal elements attached to a centrally located dissipative device. In New Zealand, testing of energy absorbing elements suitable for braced structures has been undertaken by Tyler (1983, 1985a). These tests were carried out on both circular and square devices attached to a square frame of 4x4 m. Test results showed that the circular devices developed slack in the diagonal members after approximately 20 cycles, whereas, the square devices did not show any slack when

subjected to shear deformations of  $\pm 142$  mm for 200 cycles at 1 Hz. It was reported that use of the square devices was a reliable means of ensuring continued frame action (without developing slack in the diagonal members) under earthquake conditions.

An example of a system employing mechanical dissipators is a cross braced building located in Wanganui, New Zealand, which obtains increased damping from hysteretic dampers (Mathewson and Davey 1980). This six storey precast concrete braced frame building incorporates energy dissipating mild steel devices located at the bottom ends of diagonal braces in the lower three storeys. In the event of a major earthquake, yielding of the dissipators will occur, however the frame remains elastic; hence minimal structural and non-structural damage should result.

Viscous damping is an alternative dissipative mechanism used in Supplemental Damping systems. As an example of application of viscous dissipators, a four storey steel frame building constructed in France, is fitted with visco-elastic butyl blocks which are distributed throughout the building (Zeller 1973). Experimental testing of the building showed a threefold increase in the equivalent viscous damping employing these devices (from 2% to 6%). It was reported that installation of these visco-elastic blocks provides a low cost means of increasing the seismic resistance of structures.

In another study, Bergman and Hanson (1988) report results of extensive dynamic testing of direct shear seismic dampers (DSSD) and steel plate added damping and stiffness (ADAS) devices. For energy dissipation, DSSD devices rely on shearing of a visco-elastic material; whereas, ADAS devices involve yielding of steel plate elements. The results showed that the maximum values of equivalent viscous damping obtained for the DSSD and the ADAS tests were 10.7% and 14.05% respectively; however, for the former, equivalent viscous damping values were strongly dependent upon excitation frequency. The study confirmed that these damping devices are capable of introducing significant amounts of additional damping into structural systems. This suggests that installation of such devices within a structure could reduce the forces experienced by structural members during a seismic disturbance.

Frictional mechanical dissipators have also been considered for use in Supplemental Damping systems. Tyler (1977b) discusses use of PTFE sliding joints to separate secondary

components from the main structure of a building. These joints not only allow temperature variation movement, but also provide additional damping during seismic and wind-induced vibrations. Tyler (1985b) also completed tests on a frictional damper that dissipates seismic energy during the relative displacement of structural elements. The damper comprised of a sliding steel plate clamped against a brake lining. The test results for the friction damper showed significant fluctuation in the shape of the force/deflection diagrams as a result of slip-stick action. These results differ from those obtained by other researchers. Pall (1986), for instance, showed results obtained from dynamic testing of a frictional damper, were reliable and repeatable.

(c) Advantages of supplemental damping

Further research is required before stating any definite advantages in using auxiliary mass dampers for protecting structural systems during earthquakes. However, the advantages of using mechanical dissipators in Supplemental Damping systems have been reported as follows.

(1) Supplemental Damping, by increasing the system damping, primarily reduces the inertia loads induced in the structural system; this effect may provide (Thiel et al. 1986):

- increased protection of the structural system since member loads are reduced;
- a reduction in inelastic deformation sustained by the structure since some of the input energy is dissipated by the mechanical dissipators;
- increased capability of the structure to resist subsequent earthquakes as inelastic deformation of the structure's primary load carrying system is reduced;
- a reduction in non-structural damage since the maximum response of the structure is reduced.

(2) The mechanical dissipators employed in Supplemental damping schemes, can be simple, inexpensive, and exhibit reliable and repeatable characteristics. The devices dissipate a portion of the seismic input energy, hence can reduce the energy that the structure is required to absorb through inelastic deformation (Pall 1986).

### 2.5.3 Structural parameter adjustment

The Structural Parameter Adjustment method is a technique that utilises semi-active control to adjust the parameters of a structural system; these parameters include mass, stiffness, and damping. The system makes use of a control algorithm that optimises these parameters so as to minimise the forces experienced by the structure members during a seismic disturbance.

Experimental testing of this control concept has been reported by Kobori et al. (1988). For this experiment, a model three storey steel frame building was built and tested on a shaker table. The test results showed that the maximum response of the time history record measured at the top floor of the building, was reduced to 1/4 of the response without dynamic control. This system employed a central processing unit for carrying out the optimisation of the structural parameters.

Though this control technique is still in its early stage of development, research in this field is actively being pursued. However, before Structural Parameter Adjustment systems become readily adopted, the system must demonstrate a fail-safe mechanism should the control system power supply fail during a major earthquake; in addition, further work is necessary to establish the economic feasibility of such schemes.

## 2.6 SUMMARY

The aseismic design methods used to protect engineering systems from seismic damage have been detailed. These methods can be classified into two principal techniques: Strength Design and Response Control. Strength Design involves using either the Working Stress Method or the Strength Method. In the latter, structure detail provides plastic hinge mechanisms that absorb inelastically the earthquake input energy.

In Response Control, specific control devices are incorporated within a structure to regulate its response during an earthquake. Response Control utilises either active, passive, or semi-active control and can employ the following methods: Seismic Isolation, Supplemental Damping, or Structural Parameter Adjustment. Details of these systems, the devices they employ, developments and practical applications, have been presented.

Of the Response Control techniques, Seismic Isolation has attracted substantial attention by researchers. This is because it is a simple, yet highly effective means of reducing seismic loads experienced by structures during earthquakes. Extensive research, both developmental and experimental, undertaken over the last 25 years has determined benefits of Seismic Isolation schemes over conventional systems. Further, the performance of several structures during actual earthquakes has demonstrated the effectiveness of this technique. It is anticipated that recognition of Seismic Isolation as an effective method for protecting structures during earthquakes will result in its increased use in seismic risk regions throughout the world.

## RING SPRINGS: CHARACTERISTICS AND DESIGN REQUIREMENTS

### 3.1 INTRODUCTION

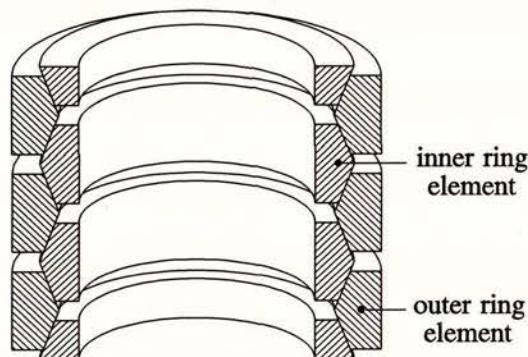
Passive energy dissipation devices are used in earthquake-resistant applications; their purpose is to absorb seismic input energy, thereby reduce structural deformations and hence the level of loading experienced by a structure during an earthquake.

This chapter details the characteristics of a passive dissipation device known as a ring spring. The ring spring stiffness equations and hysteresis diagram are detailed followed by discussion on the design requirements for practical use of these devices. Subsequently, solutions to the equations of motion for a bi-directional ring spring system are determined.

### 3.2 RING SPRINGS

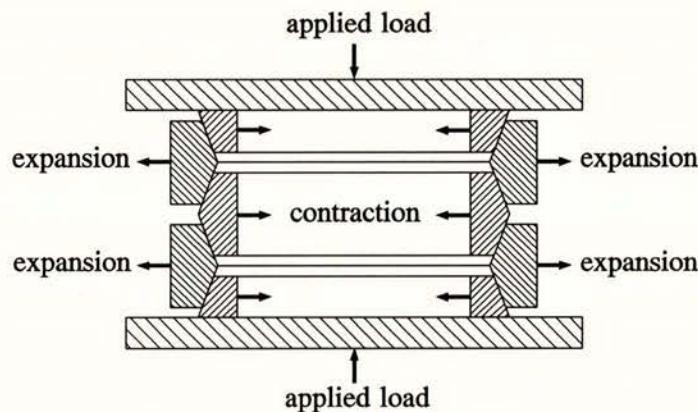
#### 3.2.1 Description and literature review

Ring springs comprise individual inner and outer ring elements assembled in columnar form (Figure 3.1).



**Figure 3.1** Section through ring spring

On application of axial load, wedge action at the interface causes the inner elements to radially contract and the outer elements to radially expand (Figure 3.2). Sliding action between mating elements results in a large amount of energy being absorbed in overcoming frictional forces. This absorbed energy may be as high as 60 to 70% of the total cycle input energy (Simpson 1970).



**Figure 3.2** Loaded ring spring

The ring spring was originally developed, and subsequently patented, in the 1920's by Ringfeder GmbH, Germany, (Ringfeder 1925). A literature review on the subject of ring springs showed surprisingly few publications. The first known paper on ring springs was published in 1924 (Wikander 1924). Ring springs were considered suitable candidates for use in drawbars in the railway industry due to their high energy absorption capability. In a more comprehensive paper published two years later, Wikander outlined formulas for the ring spring as well as presenting a hysteresis diagram obtained by testing a ring spring in a universal testing machine (Wikander 1926). In 1933, Endsley reported on the use of ring springs as draft gear springs for railway use. Details on ring springs can be found in several text books, for example Berry (1961), Wahl (1963), and Gross (1966).

Ring springs are readily available commercially in load-carrying capacities ranging from 5 kN to 1800 kN, with larger load carrying capacities up to 10 000 kN also manufactured (Ringfeder 1984).

Reported applications that utilise ring springs include:

- shock absorbers for cranes, guns, and drawbenches; buffers for cranes; vibration dampeners for power hammers; and arresting hooks for planes (Wahl 1963);
- use with furnace tie rods to allow for expansion and contraction; on guide columns of presses to allow for growth due to heat; on the guide rods of presses to absorb crosshead shock. In collieries - in safety catches, on mine-car arrestors, and in automatic couplings. In marine applications to absorb rudder vibrations; in rolling mills to absorb shocks, and in buffers on conveyors (Simpson 1970).

The majority of these applications are where a high level of damping is desired.

In the late 1960's, the behaviour of six mechanical elements suitable for use in missile shock isolation systems were investigated (Eshleman and Rao 1969; Eshleman 1972). The elements studied were: a helical coil spring; ring spring; friction snubber; liquid spring; pneumatic spring, and a solid rubber elastomer. Analytical and experimental studies were conducted to determine the response of these devices to very high acceleration shock inputs, typically greater than 100 g and of approximately 0.4 seconds duration.

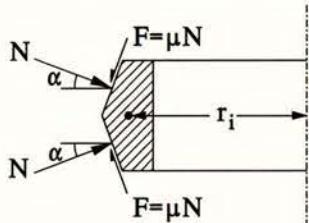
It was reported that the experimental results showed ring springs provide good isolation characteristics, however, analytical solutions were not in good agreement with the experimental results (Eshleman and Rao 1969). For this study of the ring spring, a model employing the wave equation with terms for the spring mass elasticity and damping was used; full details are reported by Eshleman (1972).

More recently, ring springs have been proposed for use in protecting structures during earthquakes. Suggested situations for employing ring springs include their use as seismic energy dissipators in conjunction with holding down bolts on chimneys; as horizontal thrusters on bridges and buildings; and in structural bracing systems (Shepherd and Erasmus 1988; Erasmus 1988). Additionally, ring springs have been proposed for use in seismic isolation systems (Hill 1995b; Hill 1995d). Further details shall be reported in Chapter 7.

The characteristics of ring springs are detailed in the following sections.

### 3.2.2 Ring spring stiffness equations

For the case of an increasing compressive load, the forces acting on an inner ring element are as shown in Figure 3.3.



**Figure 3.3** Inner ring element forces

The total radial force acting on the ring element equals:

$$2(N \cos \alpha - F \sin \alpha) \quad (3.1)$$

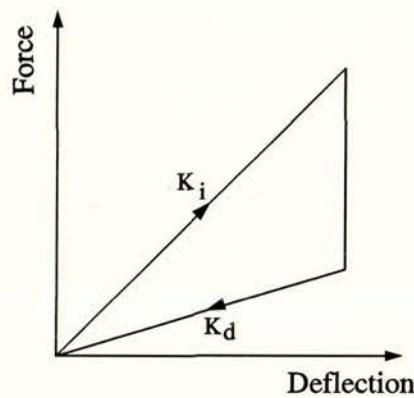
Derivation to find the compressive stress in the inner ring element, and the tensile stress in the outer ring element, is presented by Wahl (1963). By extending these equations it is possible to define the ring spring stiffness equations as follows (Hill 1993; Hill 1994c):

$$K_i = \frac{P_i}{\delta} = \left( \frac{\pi E}{r_m n} \frac{A_i}{\left(1 + \frac{A_i}{A_o}\right)} \frac{\tan \alpha (\tan \alpha + \mu)}{1 - \mu \tan \alpha} \right) \quad (3.2)$$

$$K_d = \frac{P_d}{\delta} = \left( \frac{\pi E}{r_m n} \frac{A_i}{\left(1 + \frac{A_i}{A_o}\right)} \frac{\tan \alpha (\tan \alpha - \mu)}{1 + \mu \tan \alpha} \right) \quad (3.3)$$

### 3.2.3 Ring spring hysteresis diagram

These stiffnesses,  $K_i$  and  $K_d$ , give rise to the general form of the ring spring hysteresis diagram as shown in Figure 3.4. As axial load applied to the ring spring is increased, motion proceeds from the origin up the slope  $K_i$ ; when the axial load is reduced, motion proceeds down the slope  $K_d$ . This effect is the result of reversal of the frictional force ( $\mu N$ ) acting at the element interface.



**Figure 3.4** Ring spring hysteresis diagram

### 3.3 DESIGN REQUIREMENTS FOR PRACTICAL USE OF RING SPRINGS

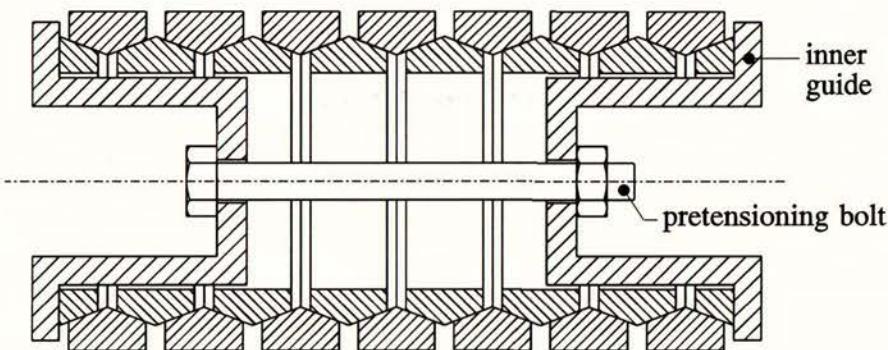
This discussion covers ring spring lubrication, housing and spring guidance, pre-tensioning, and energy absorption and temperature limits.

#### 3.3.1 Ring spring lubrication

To ensure that the frictional properties of ring springs remain reasonably constant for repeated load-unload cycling, it is essential that the ring elements are lubricated. Hence, grease lubrication is provided during initial spring assembly. Lubricant selection directly affects the coefficient of friction,  $\mu$ , thus determining the ring spring stiffness characteristics given by Equations 3.2 and 3.3. The best compromise between energy absorption capability and requirement for spring recoil gives rise to practical ring springs with a taper angle,  $\alpha=14^\circ\text{--}15^\circ$ , and a coefficient of friction,  $\mu=0.09\text{--}0.12$ . Commercially manufactured ring springs utilise a taper angle of one in four, i.e.  $\alpha=14.04^\circ$  (Ringfeder 1984). It has been reported that well lubricated ring spring assemblies do not require further lubrication for 100 000 load-unload cycles provided ingress of contaminants is avoided (Simpson 1970).

### 3.3.2 Housing and guidance of ring springs

For ring springs to be used in practical applications a cartridge assembly is required. Since ring springs accommodate axial displacements only, either internal or external guides are necessary to prevent transverse buckling of the spring stack. These guides must provide sufficient radial clearance to accommodate contraction and expansion of the inner and outer elements during operation. Internal guides are preferable since axial alignment can be assured during cartridge assembly. A typical ring spring assembly utilising internal guides is shown in Figure 3.5.



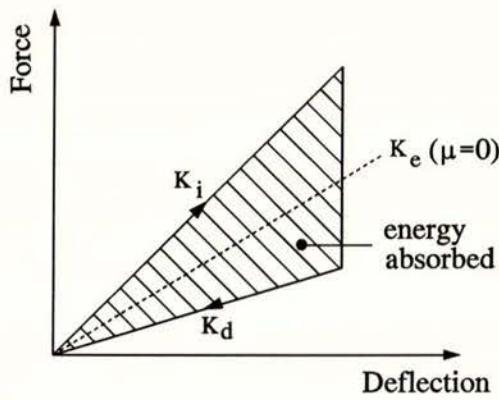
**Figure 3.5** Ring spring assembly

### 3.3.3 Spring pre-displacement

Ring springs must be provided with a minimum pre-tension force to ensure the ring elements both remain aligned after repeated load-unload cycles and do not separate. Pre-tension of 5-10% of the maximum load capacity ensures that this condition is met. So that the initial lubricant film will remain adequate, the pre-tension force should not exceed 50% of the maximum load capacity of the ring spring (Ringfeder 1984).

### 3.3.4 Energy absorption and temperature limits

The energy absorbed by a ring spring during loading/unloading can be obtained directly from the ring spring hysteresis diagram. For a full load-unload cycle, the area enclosed between stiffnesses  $K_i$  and  $K_d$  on the hysteresis diagram gives the total energy absorbed; this is seen by the hatched area in Figure 3.6.

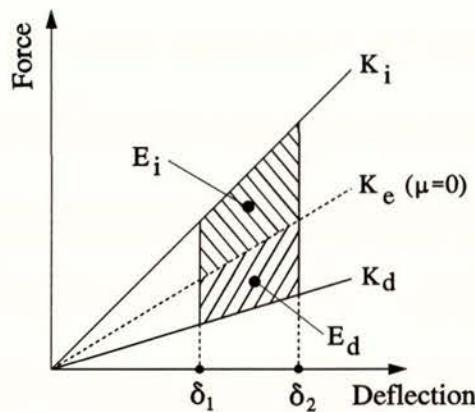


**Figure 3.6** Ring spring hysteresis diagram

In addition to calculating the frictional energy absorbed for the complete load-unload cycle, it is desirable to determine the energy absorbed during the load and unload portions separately. This is determined by firstly considering the elastic stiffness of the ring spring,  $K_e$ , obtained by setting  $\mu=0$  in either Equation 3.2 or 3.3. The elastic stiffness for the ring spring is given by Equation 3.4.

$$K_e = \left( \frac{\pi E}{r_m n} \frac{A_i}{\left( 1 + \frac{A_i}{A_o} \right)} \frac{\tan^2 \alpha}{1} \right) \quad (3.4)$$

As shown in Figure 3.7, the area below  $K_i$  and above  $K_e$  represents the energy absorbed for spring loading,  $E_i$ , while the area below  $K_e$  and above  $K_d$  represents the energy absorbed for spring unloading,  $E_d$  (Hill 1994b).



**Figure 3.7** Energy absorption for loading and unloading

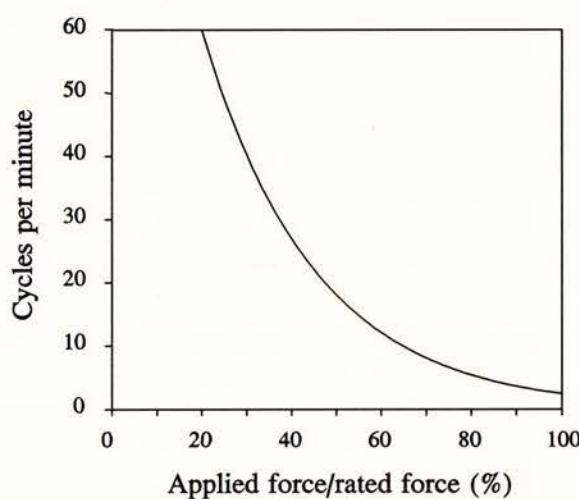
Expressions for the energy absorbed by the enclosed areas  $E_i$  and  $E_d$  are:

$$E_i = \frac{(K_i - K_e) (\delta_2^2 - \delta_1^2)}{2} \quad (3.5)$$

$$E_d = \frac{(K_e - K_d) (\delta_2^2 - \delta_1^2)}{2} \quad (3.6)$$

Heat is generated at the interfaces of mating ring elements as energy is absorbed during spring loading/unloading. The corresponding temperature rise is dependent on the following factors: the magnitude and duration of the input; the physical properties of the spring; and the surrounding temperature conditions. The maximum permissible temperature of the ring spring is limited by the melting point of the lubricant. Typically, commercial ring springs permit operating temperatures of 100°C, but by employing special grease lubricants, operating temperatures as high as 200°C may be accommodated (Ringfeder 1984).

To maintain the operating temperature of the ring spring below its maximum, the loading frequency at full ring spring stroke should not exceed an upper limit. This frequency limit is increased as the proportion of the full stroke is reduced. Figure 3.8 shows a typical graph of operating frequency versus % of full stroke for a ring spring assembly having a permissible 20 second cycle at maximum stroke (Bowen 1981).



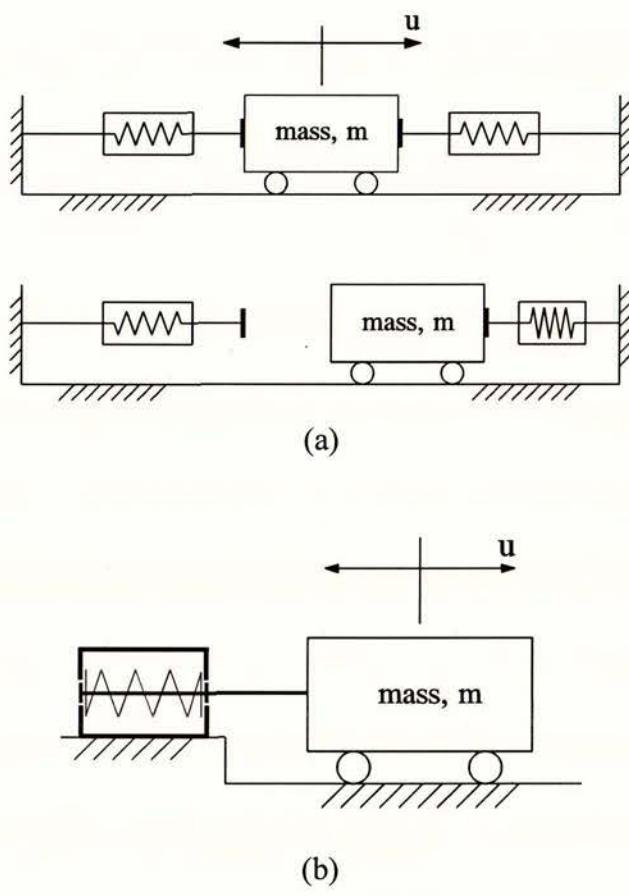
**Figure 3.8** Permissible operating frequency versus % of full stroke

These temperature limits apply to situations where a continuous excitation is applied to the ring spring. However, for short duration dynamic inputs, such as earthquakes, large displacements are expected to occur over several seconds only, hence ring spring temperatures should not approach their limits.

### 3.4 RING SPRING CONFIGURATIONS

#### 3.4.1 Bi-directional ring spring systems

For ring springs to accommodate bi-directional inputs, a configuration that utilises either dual springs, or a single spring, can be used. Two such systems are considered here – a dual spring bi-directional system and a single spring bi-directional system. These systems are shown in Figure 3.9 (a) and (b) respectively.

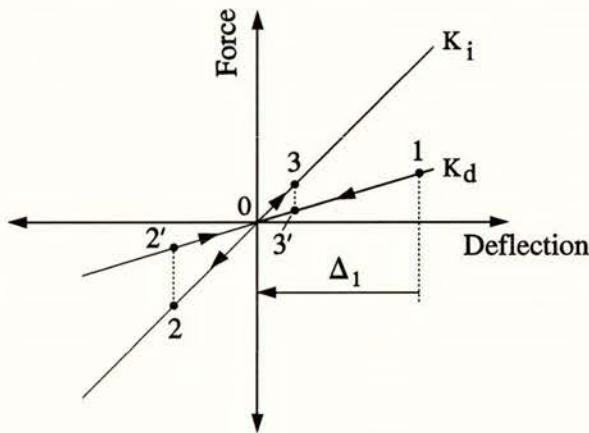


**Figure 3.9** Bi-directional ring spring systems:  
(a) dual spring; (b) single spring

The system in Figure 3.9 (a) employs two uni-directional ring spring cartridges, whereas, in Figure 3.9 (b) a single bi-directional ring spring cartridge is used.

### 3.4.2 Force/deflection characteristics

During operation, only one spring is active at any instant in time in the dual spring system shown in Figure 3.9 (a), while in the system shown in Figure 3.9 (b) the single spring is always active. If each system comprises similar ring springs, the force/deflection characteristics will be the same for each case as shown in Figure 3.10.



**Figure 3.10** Force/deflection diagram for bi-directional ring spring systems

## 3.5 FREE VIBRATION RESPONSE OF BI-DIRECTIONAL MASS/RING SPRING SYSTEMS

### 3.5.1 System model

The dynamic behaviour of bi-directional ring spring systems is now determined. Consider the system as shown in either Figure 3.9 (a) or Figure 3.9 (b) released from an initial displacement  $\Delta_1$ ; motion proceeds around the force/deflection diagram from points 1 to 0, 0 to 2, 2' to 0, then 0 to 3 for one full cycle of motion (Figure 3.10).

### 3.5.2 Governing equation of motion

To determine the dynamic response of this system, the equation of motion can be expressed as:

$$m\ddot{u} + k^*u = 0 \quad (3.7)$$

where:

- $k^*$  is the particular stiffness on the force/deflection diagram at the instant of motion considered (either  $K_i$  or  $K_d$ );
- $u(t)$  is the displacement of the mass.

The system has two separate natural frequencies, a natural frequency on the lower slope,

$$\omega_{nd} = \left( \frac{K_d}{m} \right)^{\frac{1}{2}} \quad (3.8)$$

a natural frequency on the upper slope,

$$\omega_{ni} = \left( \frac{K_i}{m} \right)^{\frac{1}{2}} \quad (3.9)$$

and four cycle 1/4 periods, comprising both  $T_1$  and  $T_2$ , where:

$$T_1 = \frac{\pi}{2\omega_{nd}} \quad (3.10)$$

$$T_2 = \frac{\pi}{2\omega_{ni}} \quad (3.11)$$

### 3.5.3 Dynamic response

#### (a) Displacement solution

Using Equation 3.7, four equations, one for each phase of motion, can be solved piece-wise to give (Hill 1994a):

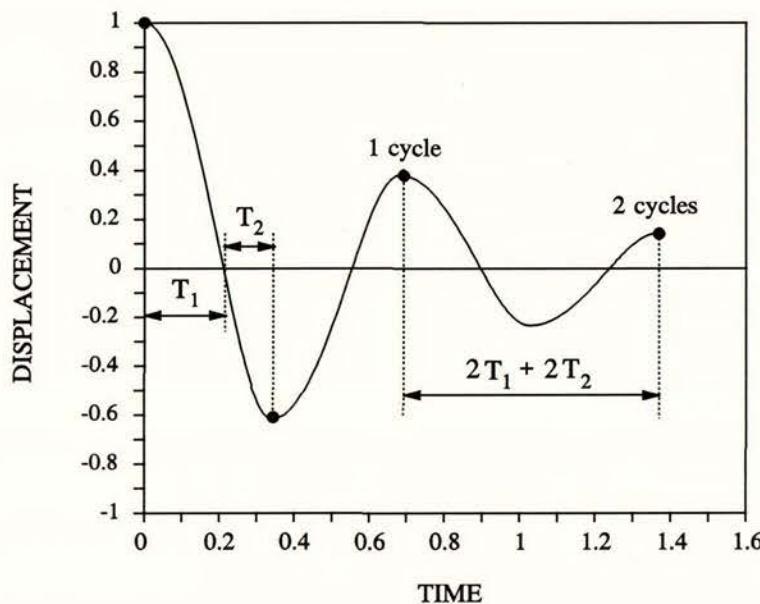
$$\text{pt 1 to 0: } u = \Delta_1 \cos(\omega_{nd} t) \quad 0 \leq t \leq T_1$$

$$\text{pt 0 to 2: } u = -\left(\frac{\omega_{nd}}{\omega_{ni}}\right) \Delta_1 \sin(\omega_{ni}(t-T_1)) \quad T_1 \leq t \leq T_1 + T_2$$

$$\text{pt 2' to 0: } u = -\left(\frac{\omega_{nd}}{\omega_{ni}}\right) \Delta_1 \cos(\omega_{nd}(t-(T_1+T_2))) \quad T_1 + T_2 \leq t \leq 2T_1 + T_2$$

$$\text{pt 0 to 3: } u = \left(\frac{\omega_{nd}}{\omega_{ni}}\right)^2 \Delta_1 \sin(\omega_{ni}(t-(2T_1+T_2))) \quad 2T_1 + T_2 \leq t \leq 2T_1 + 2T_2$$

A typical free vibration displacement curve, given by these equations, is presented in Figure 3.11. The displacement response is shown to be periodic over each full cycle of motion, the period being  $2T_1 + 2T_2$ .



**Figure 3.11** Free vibration displacement for two full cycles

## (b) Acceleration solution

The acceleration solution can also be solved in a piece-wise manner. This gives:

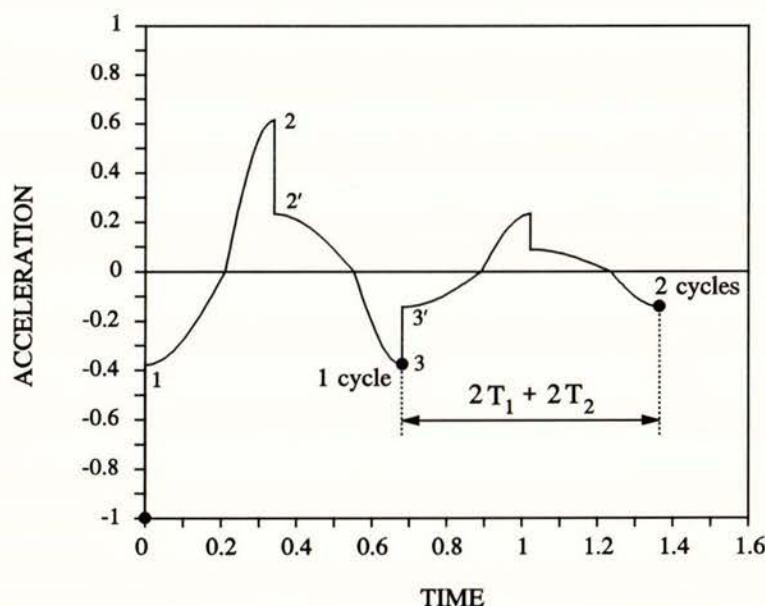
$$\text{pt 1 to 0: } \ddot{u} = -(\omega_{nd})^2 \Delta_1 \cos(\omega_{nd} t) \quad 0 \leq t \leq T_1$$

$$\text{pt 0 to 2: } \ddot{u} = (\omega_{nd} \cdot \omega_{ni}) \Delta_1 \sin(\omega_{ni}(t-T_1)) \quad T_1 \leq t \leq T_1 + T_2$$

$$\text{pt } 2' \text{ to 0: } \ddot{u} = \left( \frac{\omega_{nd}^3}{\omega_{ni}} \right) \Delta_1 \cos(\omega_{nd}(t-(T_1+T_2))) \quad T_1 + T_2 \leq t \leq 2T_1 + T_2$$

$$\text{pt 0 to 3: } \ddot{u} = -(\omega_{nd})^2 \Delta_1 \sin(\omega_{ni}(t-(2T_1+T_2))) \quad 2T_1 + T_2 \leq t \leq 2T_1 + 2T_2$$

A typical free vibration acceleration curve, given by these equations, is presented in Figure 3.12. The steep lines shown on the response between points 2 and 2', and 3 and 3', correspond to a stiffness change, from  $K_i$  to  $K_d$ , as motion reverses direction (Figure 3.10).



**Figure 3.12** Free vibration acceleration for two full cycles

### 3.5.4 Ring spring damping factor

From the displacement solution, the initial displacement  $\Delta_i$  is shown to reduce over one full cycle by a factor given by:

$$\gamma = \left( \frac{\omega_{nd}}{\omega_{ni}} \right)^2 \quad (3.12)$$

Utilising Equations 3.2 and 3.3 and Equations 3.8 and 3.9, this gives:

$$\gamma = \frac{(\tan\alpha - \mu)(1 - \mu\tan\alpha)}{(1 + \mu\tan\alpha)(\tan\alpha + \mu)} \quad (3.13)$$

After  $q$  cycles, the initial displacement reduces by a factor of  $\gamma^q$ . For a ring spring with a taper angle of  $\alpha=15^\circ$  and coefficient of friction  $\mu=0.12$ , the displacement reduces after one, two and three cycles to 0.358, 0.128 and 0.046 of the initial displacement respectively. The free vibration decay curve shown in Figure 3.11 is for  $\alpha=15^\circ$  and  $\mu=0.12$ . These results show that free vibration oscillations reduce to small values within a few cycles of motion.

The above equations can be used to define the ring spring damping factor. For free vibration decay the damping factor can be approximated from the ratio of displacement amplitudes measured after  $q$  cycles:

$$\zeta = \frac{1}{2\pi q} \ln \left( \frac{\Delta_i}{\Delta_{i+q}} \right) \quad (3.14)$$

Utilising Equation 3.13 the ring spring damping factor is given by (Hill 1994b):

$$\zeta = \frac{1}{2\pi} \ln \left( \frac{1}{\gamma} \right) \quad (3.15)$$

Thus, the damping factor for the ring spring is a function of both the spring taper angle,  $\alpha$ , and the coefficient of friction,  $\mu$ .

For ring springs with a taper angle,  $\alpha=14^\circ\text{--}15^\circ$ , and a coefficient of friction,  $\mu=0.09\text{--}0.12$ , the above equations give the damping factor as  $0.119 \leq \zeta \leq 0.176$ ; an equivalent viscous damping value of between 11.9% and 17.6%.

### 3.6 SUMMARY

Ring springs are passive devices capable of absorbing as much as 60–70% of the total cycle input energy. Their stiffness equations and hysteresis characteristics have been detailed. The requirements for utilising ring springs in practical systems have been presented. Consideration has been given to ring spring lubrication, guidance, pre-tensioning, and energy absorption and temperature limits.

Solutions to the equations of motion for a bi-directional ring spring system have been determined. These show that free vibration oscillations diminish within a few cycles of motion. Ring springs have damping factors of between 11.9% and 17.6%.

## Chapter 4

# HYSERESIS CHARACTERISTICS OF A BI-DIRECTIONAL RING SPRING CARTRIDGE ASSEMBLY

### 4.1 INTRODUCTION

So that the dynamic response of mass/ring spring systems can be studied the nature of the hysteresis characteristics of pre-displaced ring spring cartridge assemblies needs to be investigated.

This chapter details the development of a prototype ring spring cartridge that allows bi-directional inputs to be applied to ring springs. The theoretical hysteresis characteristics of bi-directional ring spring systems are discussed, followed by the quasi-static hysteresis characteristics obtained from experimental tests conducted on the ring spring cartridge.

### 4.2 RING SPRING SELECTION AND QUASI-STATIC TEST RESULTS

For this series of tests a commercially available ring spring (type 03800) with a maximum load capacity of 20 kN, provided by Ringfeder GmbH, Germany, was chosen.

The ring spring, comprising inner, outer and half inner elements, is shown in Figure 4.1. The half inner elements are fitted at each end of the spring stack.



**Figure 4.1** Ring spring components (inner, outer and half inner elements)

#### 4.2.1 Physical properties of the selected ring spring

Details of the chosen ring spring are presented in Figure 4.2 and in the specifications below.

ring spring external diameter = 38.1 mm

ring spring internal diameter = 31.7 mm

free length of ring spring = 266.0 mm

$n = 56$  interfaces

$\tan \alpha = 0.25$  ( $\alpha = 14.04^\circ$ )

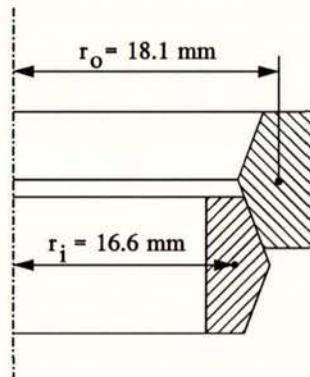
$\mu = 0.10$

$E = 200$  GPa

$A_o = 14.12$  mm $^2$

$A_i = 11.09$  mm $^2$

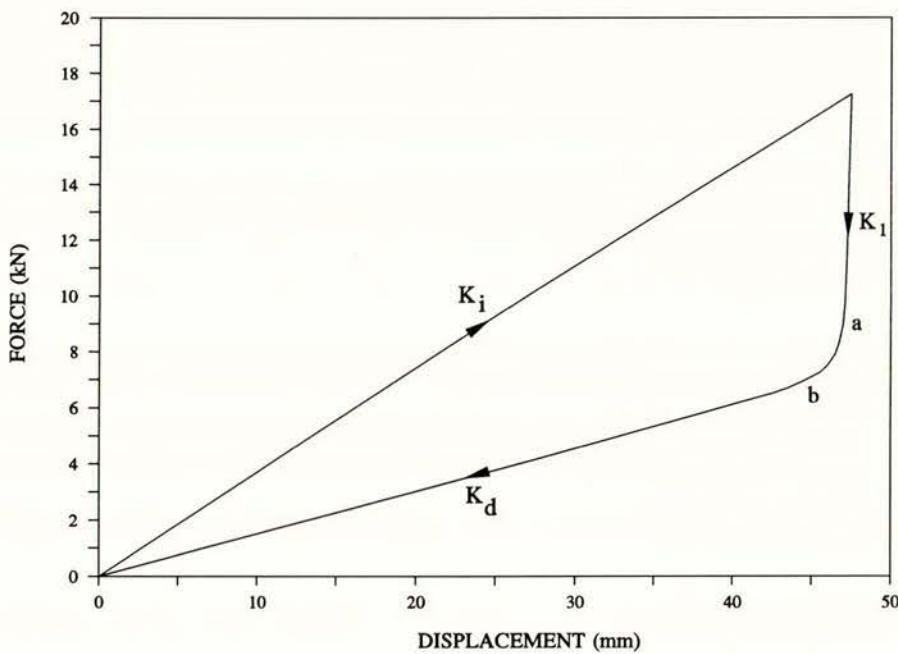
$r_m = 17.35$  mm



**Figure 4.2** Ring spring dimensions

#### 4.2.2 Quasi-static hysteresis diagram

To experimentally determine the hysteresis characteristics of the ring spring, a universal testing machine at the University of Canterbury was used. Tests were performed for a spring loading rate of 15 mm/min, and an unloading rate of 5 mm/min. The resultant quasi-static hysteresis diagram is presented in Figure 4.3 (Hill 1995a).



**Figure 4.3** Ring spring hysteresis diagram

#### 4.2.3 Force/deflection stiffnesses

Two principal ring spring stiffnesses,  $K_i$  and  $K_d$  (as described in Chapter 3), are shown in Figure 4.3. The measured value of these stiffnesses are:

$$K_i = 363 \times 10^3 \text{ N/m}$$

$$K_d = 153 \times 10^3 \text{ N/m}$$

To assess how these stiffnesses compare with those given by the ring spring stiffness equations, the physical properties of the spring may be substituted into Equations 3.2 and 3.3. Utilising these equations gives calculated values for the ring spring stiffnesses as:

$$K_i = 360 \times 10^3 \text{ N/m}$$

$$K_d = 147 \times 10^3 \text{ N/m}$$

These values correlate closely with those given by experimental tests.

In addition to these two principal stiffnesses, a third stiffness acts during the transition from stiffness  $K_i$  to  $K_d$ . This phase of motion is now discussed. The ring spring stiffness,

$K_i$ , given by Equation 3.2, is determined by the forces set up at the element interfaces for an increasing applied load. After maximum load is reached and applied load begins to reduce, the frictional force acting across the interface of individual ring elements also reduces. During this period of unloading, the coefficient of friction undergoes a change from  $+\mu$  to  $-\mu$ . Throughout this phase of motion, sliding of mating ring elements is prevented due to the frictional forces set up at the element interfaces. However, strain energy stored within the ring spring is released giving rise to a new stiffness  $K_t$  as shown in Figure 4.3. In conjunction with this release of strain energy, sliding across some ring elements initiates at point a, and at point b sliding of all ring elements proceeds (Figure 4.3). This effect is primarily due to machining tolerances of the ring spring elements as well as differences in the inner and outer element radii and cross sectional areas.

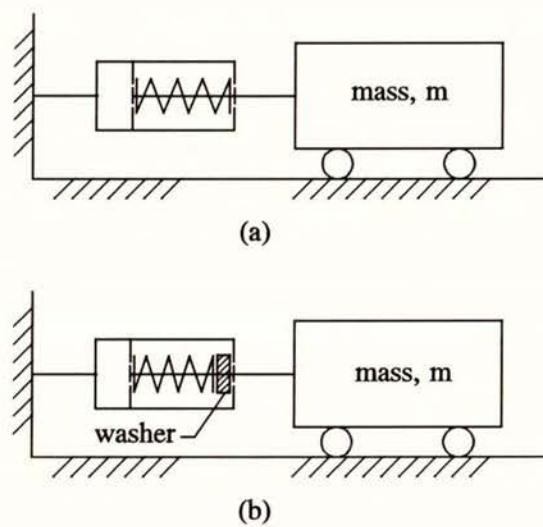
The measured value of this transition stiffness is  $K_t = 17 \times 10^6$  N/m. This value is almost 50 times larger than the value of stiffness  $K_i$ .

### **4.3 THEORETICAL HYSTERESIS CHARACTERISTICS OF BI-DIRECTIONAL RING SPRING SYSTEMS**

The ring spring hysteresis characteristics detailed to date, have considered the characteristics of the ring spring only. However, as detailed in Chapter 3, practical use of ring springs requires an appropriate cartridge assembly. This section discusses the theoretical force/deflection characteristics of two bi-directional ring spring cartridge assemblies. Initially, an assembly with zero pre-displacement is detailed. Subsequently, a system with applied pre-displacement is discussed.

#### **4.3.1 Force/deflection stiffnesses**

A schematic of two mass/ring spring systems that each embody a bi-directional ring spring cartridge is shown in Figures 4.4 (a) and (b).

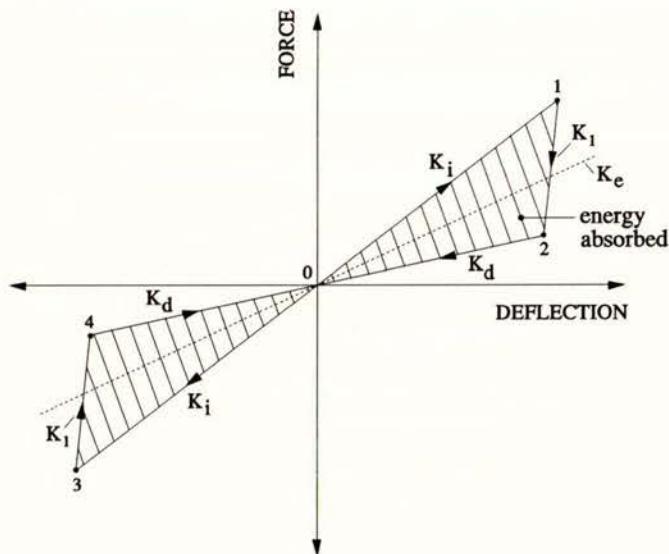


**Figure 4.4** Ring spring cartridges: (a) zero spring pre-displacement  
(b) spring pre-displacement

The ring spring cartridge shown in Figure 4.4 (a) is subject to zero pre-displacement, whereas in Figure 4.4 (b) a washer fitted within the cartridge applies an initial displacement (pre-displacement) to the ring spring. Each of these systems shall be discussed.

(a) Zero pre-displacement systems

An idealised force/deflection diagram for a ring spring assembly with zero pre-displacement is presented in Figure 4.5.



**Figure 4.5** Ring spring force/deflection diagram (zero pre-displacement)

Initially, consider the system shown in Figure 4.4 (a) undergoing displacement. As increasing load is applied to the ring spring cartridge, motion proceeds along stiffness  $K_i$  (points 0-1) as shown in Figure 4.5. Once a maximum displacement is reached, point 1, spring unload proceeds down stiffness  $K_l$  (points 1-2) then onto stiffness  $K_d$  and back to the origin (point 0). This motion is repeated when the direction of the cartridge loading/unloading is reversed. Thus, for a complete load/unload cycle, motion proceeds around points 0-1, 1-2, 2-3, 3-4, and 4-0. The area enclosed (hatched) represents the total energy absorbed for one full load-unload cycle (Figure 4.5).

The characteristics of the two principal ring spring stiffnesses,  $K_i$  and  $K_d$ , as well as the transition stiffness  $K_l$ , active between points 1 and 2 (Figure 4.5), have been outlined earlier.

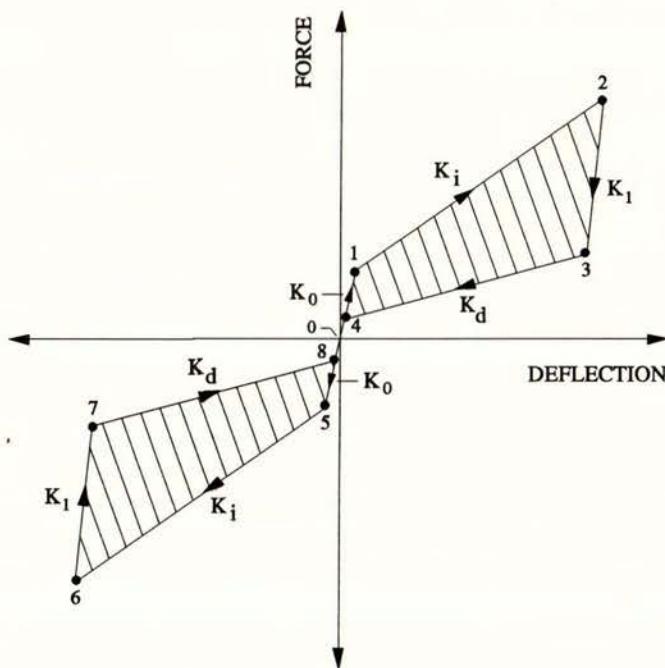
#### (b) Pre-displaced systems

The system shown in Figure 4.4 (b), fitted with a pre-displacement washer of thickness  $\Delta$ , is termed a pre-displaced mass/ring spring system. To quantify the amount of pre-displacement for any particular system, the term % pre-displacement has been developed.

This is defined as follows:

$$\% \text{ pre-displacement} = \frac{\text{initial ring spring displacement}}{\text{maximum ring spring displacement}} \cdot \frac{100}{1} \quad (4.1)$$

The idealised force/deflection diagram for a pre-displaced ring spring system may be represented as shown in Figure 4.6.



**Figure 4.6** Force/deflection diagram for pre-displaced ring springs

In applying load to a pre-displaced system, sufficient force must be applied to overcome frictional resistance before sliding action within the ring spring commences. Thus, a force greater than  $F_0 = (K_i \Delta)$  must be applied to the ring spring for motion to proceed along stiffness  $K_i$ , point 1 in Figure 4.6. This gives rise to an elastic stiffness,  $K_0$ , active between points 0-1. As load is increased motion continues between points 1-2. During the next phase of motion unloading proceeds between points 2-3, and then continues along stiffness  $K_d$  to point 4, returning to the origin along stiffness  $K_0$ . This motion is repeated when the direction of the cartridge loading/unloading is reversed. Thus, motion proceeds around points 0-1, 1-2, 2-3, 3-4, 4-5, 5-6, 6-7, 7-8, and 8-0 for a complete load/unload cycle. The hatched area represents the total energy absorbed for one full cycle (Figure 4.6).

The hysteresis diagrams shown in Figures 4.5 and 4.6 show that as displacement increases, the energy absorption per unit increment in displacement, also increases. Hence, greater energy absorption is possible at larger displacements.

#### 4.4 A PROTOTYPE BI-DIRECTIONAL RING SPRING CARTRIDGE

This section discusses the development of a prototype bi-directional ring spring cartridge that is used to verify the theoretical hysteresis diagrams outlined above.

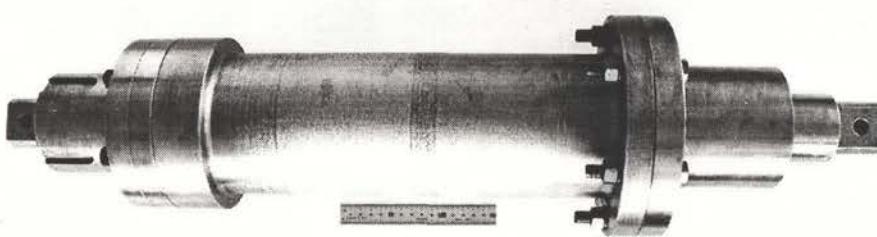
#### 4.4.1 Design detail

The ring spring cartridge primarily comprises of an outer housing and an inner shaft. The outer housing enables attachment at its end and the inner shaft locates and provides guidance for the ring spring. Pistons are used at each end of the spring to accommodate bi-directional displacements. Each piston is lined with 3 mm thick rubber to cushion contacting as the cartridge undergoes displacements.

Ring spring pre-displacement is provided by a spacer installed within the cartridge. Two steel spacers, 8 mm and 14 mm in thickness, were designed so that the effect of varying the cartridge pre-displacement could be investigated. Use of these spacers determines the maximum stroke capacity of the bi-directional assembly. Since the ring spring may accommodate displacements up to 50 mm, use of the 8 mm and 14 mm spacers reduces the maximum operating displacement of the cartridge to  $\pm 42$  mm and  $\pm 36$  mm respectively. These pre-displacements may be expressed as % pre-displacements by using Equation 4.1. With the maximum possible ring spring displacement of 50 mm, an 8 mm pre-displacement equates to  $(8/50)*100 = 16\%$  pre-displacement. The 14 mm pre-displacement equates to  $(14/50)*100 = 28\%$  pre-displacement.

#### 4.4.2 Manufactured ring spring cartridge

A scaled general assembly drawing of the bi-directional ring spring cartridge is presented in Appendix B. The cartridge is shown in Figure 4.7.



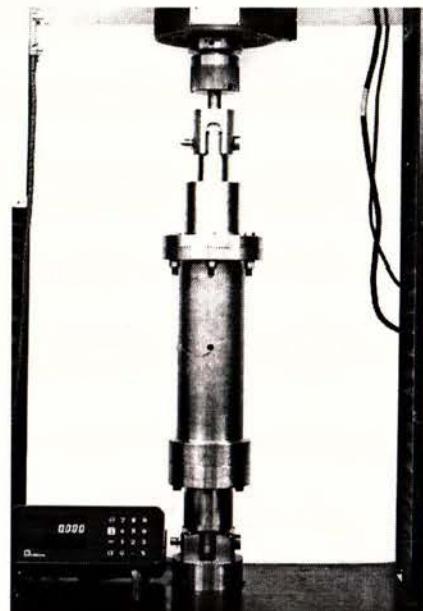
**Figure 4.7** Prototype ring spring cartridge (with 150 mm ruler)

## 4.5 EXPERIMENTAL TESTING OF THE RING SPRING CARTRIDGE

Subsequently, tests were undertaken to determine the hysteresis characteristics of the ring spring cartridge for 16% and 28% pre-displacements.

### 4.5.1 Test facility

Experimental testing was undertaken to determine the quasi-static force/deflection characteristics of the pre-displaced ring spring cartridge. The cartridge, tested in a universal testing machine, was displaced through  $\pm 35.0$  mm and  $\pm 30.0$  mm for the 16% and 28% cartridge pre-displacements respectively. Testing was carried out for loading and unloading rates of 15 mm/min and 5 mm/min. The experimental system is shown in Figure 4.8.

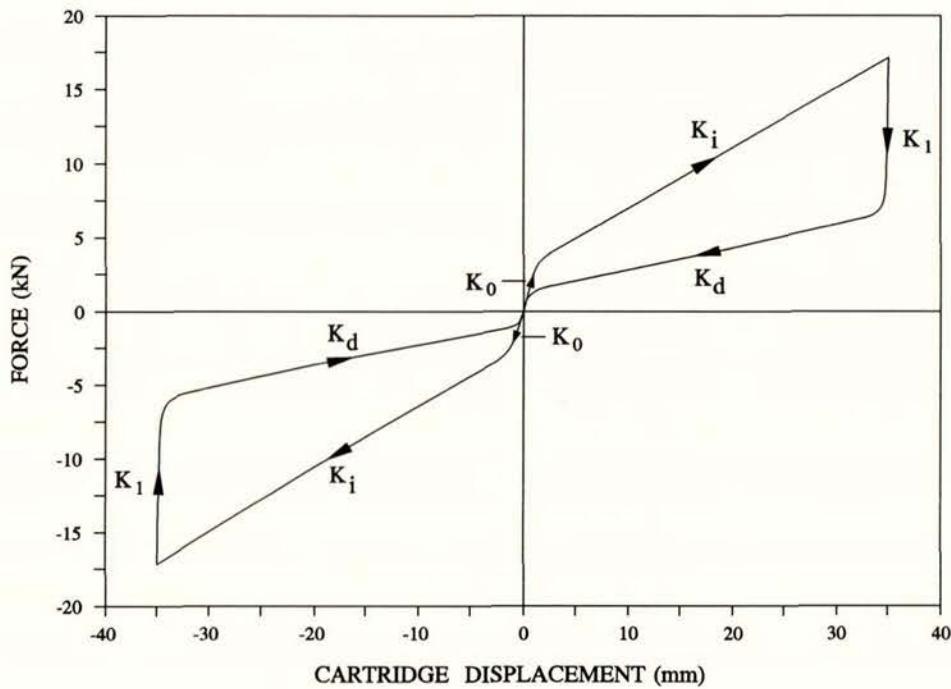


**Figure 4.8** Ring spring cartridge in universal testing machine

#### 4.5.2 Quasi-static hysteresis diagrams

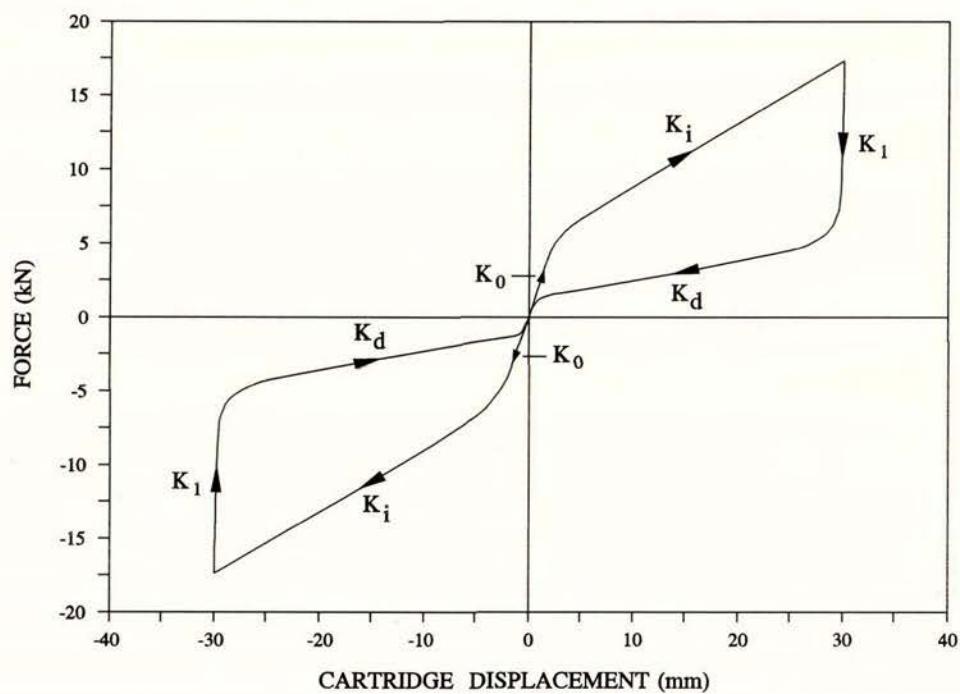
The hysteresis characteristics obtained from these tests are presented in Figures 4.9 and 4.10.

(a) 16% pre-displacement



**Figure 4.9** Ring spring cartridge hysteresis diagram  
(16% pre-displacement)

(b) 28% pre-displacement



**Figure 4.10** Ring spring cartridge hysteresis diagram  
(28% pre-displacement)

(c) Force/deflection stiffnesses

The experimental results shown in Figures 4.9 and 4.10 confirm the general nature of the theoretical force/deflection characteristics for pre-displaced ring springs (Figure 4.6). For both 16% and 28% pre-displacements four separate stiffnesses,  $K_0$ ,  $K_i$ ,  $K_1$ , and  $K_d$  are seen. These stiffnesses, based on mean values determined from both quadrants of the hysteresis diagram, are given below.

For the 16% pre-displacement:

$$K_0 = 2.5 \times 10^6 \text{ N/m}$$

$$K_i = 385 \times 10^3 \text{ N/m}$$

$$K_1 = 33 \times 10^6 \text{ N/m}$$

$$K_d = 142 \times 10^3 \text{ N/m}$$

For the 28% pre-displacement:

$$K_0 = 2 \times 10^6 \text{ N/m}$$

$$K_i = 410 \times 10^3 \text{ N/m}$$

$$K_l = 38 \times 10^6 \text{ N/m}$$

$$K_d = 135 \times 10^3 \text{ N/m}$$

In addition, the force at the transition from stiffness  $K_0$  to  $K_i$  was measured as:

$$F_0 = 2900 \text{ N} \text{ (for the 16% pre-displacement)}$$

$$F_0 = 5500 \text{ N} \text{ (for the 28% pre-displacement)}$$

These forces compare with calculated values of  $F_0 = 3080 \text{ N}$  and  $5740 \text{ N}$  (using  $F_0 = K_i \Delta$ ) for the 16% and 28% pre-displacements respectively.

Determination of the above hysteresis characteristics enables dynamic modelling of pre-displaced mass/ring spring systems to be undertaken. For this, the ring spring model is able to be based upon four non-linear force/deflection stiffnesses -  $K_0$ ,  $K_i$ ,  $K_l$ , and  $K_d$ .

#### 4.6 SUMMARY

The development and testing of a prototype ring spring cartridge suitable for applying bi-directional inputs to ring springs has been presented. Quasi-static testing of the cartridge for two different pre-displacements has established that the hysteresis characteristics of pre-displaced ring spring systems comprise of four stiffness,  $K_0$ ,  $K_i$ ,  $K_l$ , and  $K_d$ . The dynamic behaviour of ring spring systems may now be determined based upon these hysteresis characteristics.

## FREE VIBRATION BEHAVIOUR OF PRE-DISPLACED AND PRE-LOADED MASS/RING SPRING SYSTEMS

### 5.1 INTRODUCTION

Understanding of the dynamic behaviour of ring spring systems is a prerequisite to use of these devices in applications subject to dynamic excitation. This chapter investigates the dynamic behaviour of pre-displaced, pre-loaded, and pre-displaced pre-loaded single-degree-of-freedom (SDOF) mass/ring spring systems. Each of these systems is defined, then their free vibration behaviour is detailed.

### 5.2 COMPUTER MODELLING OF SDOF MASS/RING SPRING SYSTEMS

In Chapter 3 the free vibration response of a SDOF mass/ring spring system was determined by solving piece-wise the equations of motion. It was assumed that the ring spring hysteresis characteristics comprised of only two stiffnesses,  $K_i$  and  $K_d$ , thus, making it relatively easy to determine the free vibration response.

However, for practical ring spring systems, the hysteresis diagram consists of four stiffnesses:  $K_0$ ,  $K_i$ ,  $K_l$  and  $K_d$ . To determine the free vibration response of systems with these stiffnesses, it is difficult and cumbersome to solve the equations of motion for the system in a piece-wise manner. A more convenient method for determining the dynamic response is to employ a numerical integration technique. This allows the dynamic response of a non-linear system to be determined.

#### 5.2.1 Numerical integration method

A numerical technique, the Newmark ( $\beta = 0.25$ ) constant average acceleration method (Newmark 1962) was chosen to determine the dynamic response of the SDOF mass/ring spring systems to be analysed. This method was selected as it is unconditionally stable.

The equations formulated using the Newmark method are detailed in several text books, eg. Craig (1981), and are not restated here.

### **5.2.2 Ring spring model computer algorithm**

To enable mass/ring spring systems to be analysed, a ring spring model has been developed. The model is based upon the non-linear force/deflection characteristics of the ring spring and uses five variables that fully define the ring spring hysteresis characteristics. These are the four stiffnesses:  $K_0$ ,  $K_i$ ,  $K_l$ , and  $K_d$  and the ring spring pre-displacement.

In addition, a computer algorithm was written that embodies the ring spring model and tracks the current stiffness on the hysteresis diagram at any given instant in time. The ring spring model computer algorithm is presented in Appendix A. Further discussion of this algorithm is reported in Chapter 7.

### **5.2.3 Types of ring spring systems considered in the study**

The following sections detail the free vibration behaviour of:

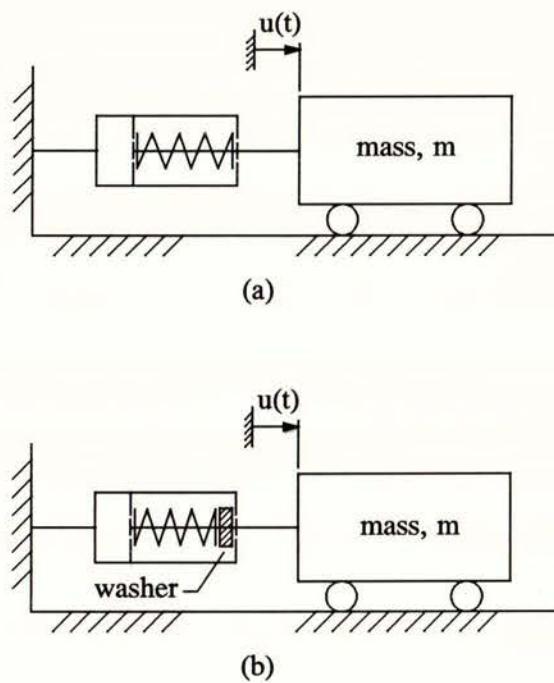
- pre-displaced ring spring systems,
- pre-loaded ring spring systems,
- pre-displaced pre-loaded ring spring systems.

## **5.3 FREE VIBRATION BEHAVIOUR OF PRE-DISPLACED RING SPRING SYSTEMS**

This section details the dynamic response of a SDOF mass/ring spring system for both zero and 16% pre-displacement.

### **5.3.1 System model**

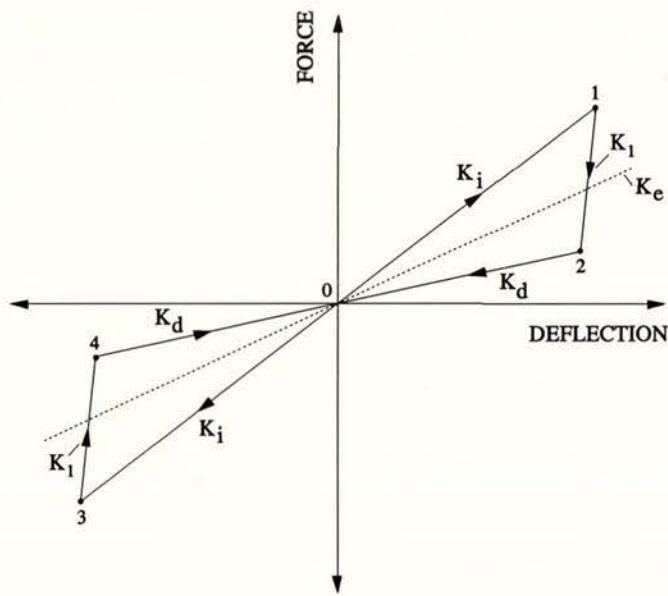
The model for a SDOF pre-displaced mass/ring spring system consists of a spring cartridge attached to a mass as shown in Figure 5.1. In (a) the system has zero pre-displacement, whereas in (b) the cartridge, fitted with a washer, is pre-displaced. These systems have been detailed in Chapter 4.



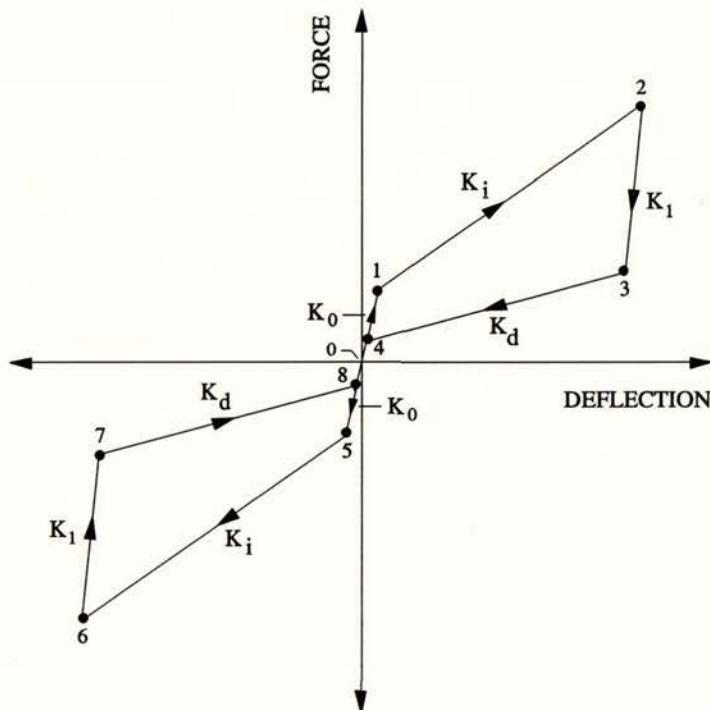
**Figure 5.1** Mass/ring spring systems: (a) zero spring pre-displacement  
(b) spring pre-displacement

### 5.3.2 Force/deflection characteristics

The force/deflection characteristics of the systems shown in Figure 5.1 have also been outlined in Chapter 4. In summary, the idealised force/deflection diagrams for a system with zero pre-displacement comprises three stiffness:  $K_i$ ,  $K_l$ , and  $K_d$  and for a complete load/unload cycle motion proceeds around points 0-1, 1-2, 2-3, 3-4, and 4-0 (Figure 5.2). For a system with pre-displacement, the idealised force/deflection diagram comprises four stiffnesses:  $K_0$ ,  $K_i$ ,  $K_l$ , and  $K_d$ , and motion proceeds around points 0-1, 1-2, 2-3, 3-4, 4-5, 5-6, 6-7, 7-8, and 8-0 for a complete load/unload cycle (Figure 5.3).



**Figure 5.2** Force/deflection diagram for ring springs with zero pre-displacement



**Figure 5.3** Force/deflection diagram for pre-displaced ring springs

### 5.3.3 Equation of motion

The generalised equation of motion for a pre-displaced mass/ring spring system is:

$$m\ddot{u} + k^*u = 0 \quad (5.1)$$

where:

- $k^*$  is the particular stiffness on the force/deflection diagram at the instant of motion considered (either  $K_0$ ,  $K_i$ ,  $K_1$ , or  $K_d$ ); and
- $u(t)$  is the displacement of the mass.

The following parameters have been used in analysing the dynamic response of the mass/ring spring systems reported in this chapter:

$$K_0 = 2.5 \times 10^6 \text{ N/m}$$

$$K_i = 385 \times 10^3 \text{ N/m}$$

$$K_1 = 33 \times 10^6 \text{ N/m}$$

$$K_d = 142 \times 10^3 \text{ N/m}$$

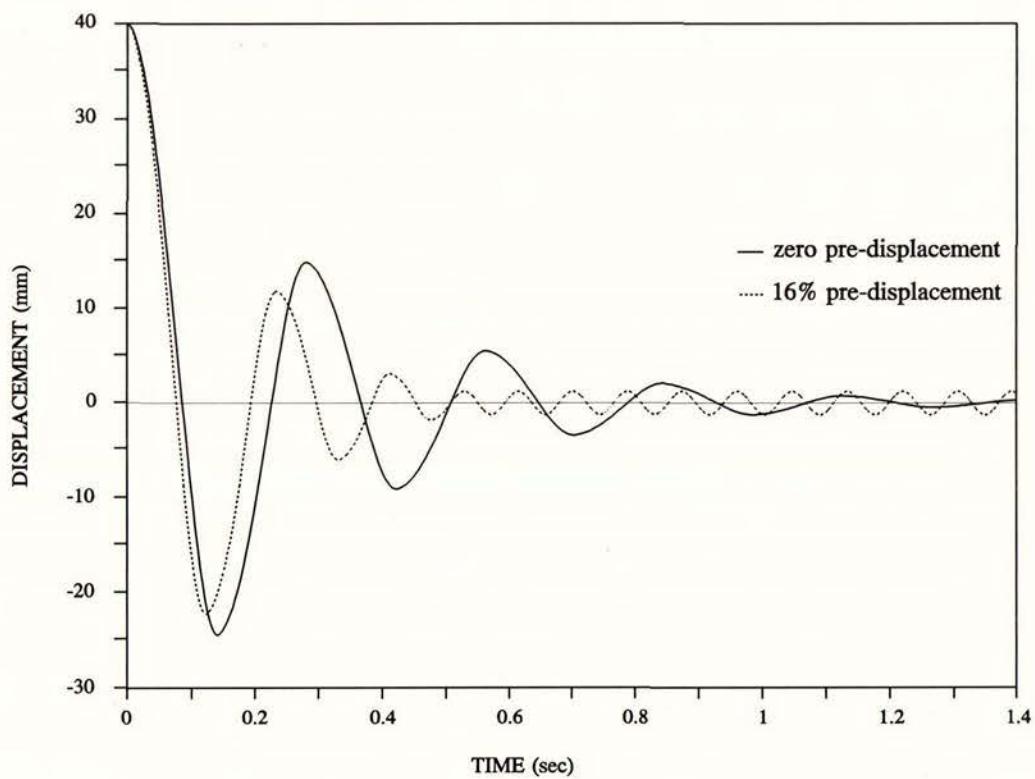
$$m = 472 \text{ kg}$$

These stiffnesses were obtained from tests carried out on the prototype ring spring cartridge reported in Chapter 4. The selected mass was chosen to correspond with that of an item of equipment available for use in subsequent experimental testing.

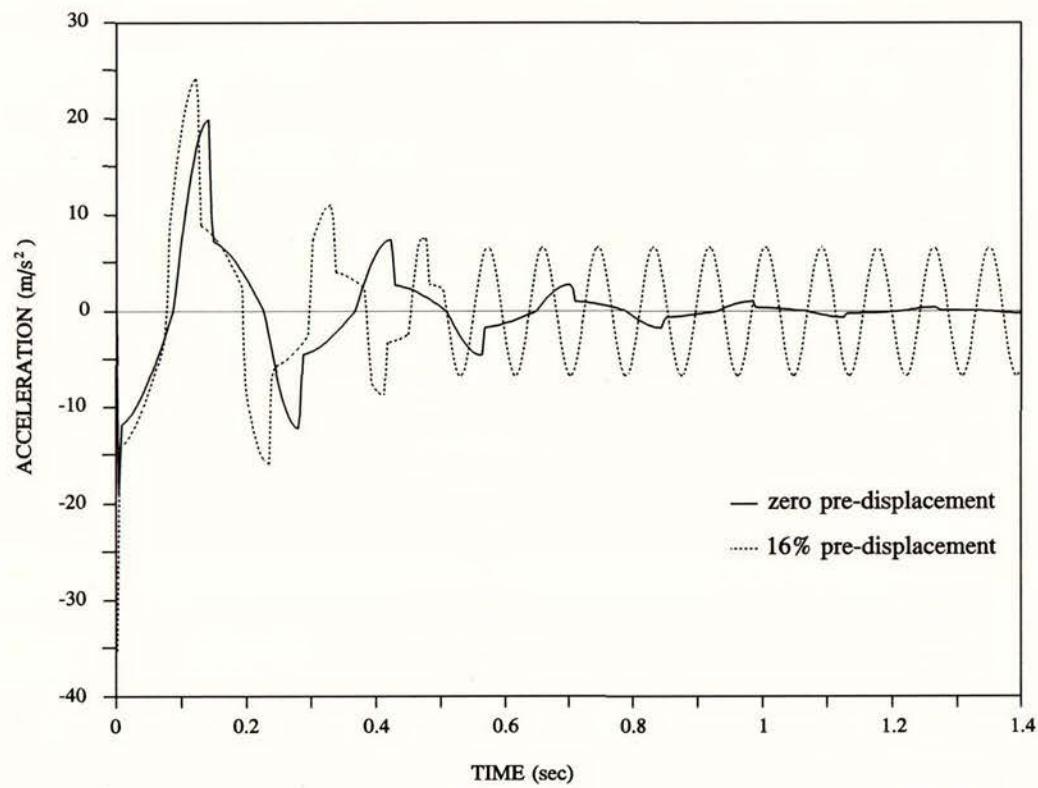
### 5.3.4 Dynamic response

Utilising these values in the computer algorithm, the free vibration response of a mass/ring spring system with zero, and with 16% pre-displacement, was determined. The systems, as shown in Figure 5.1 (a) and (b), were each released from an initial displacement of 40 mm. Their displacement and acceleration responses are presented in Figures 5.4 and 5.5. Further, force/deflection response curves for these systems are given in Figures 5.6 (a) and (b). These diagrams give plots of the ring spring hysteresis characteristics developed during the response.

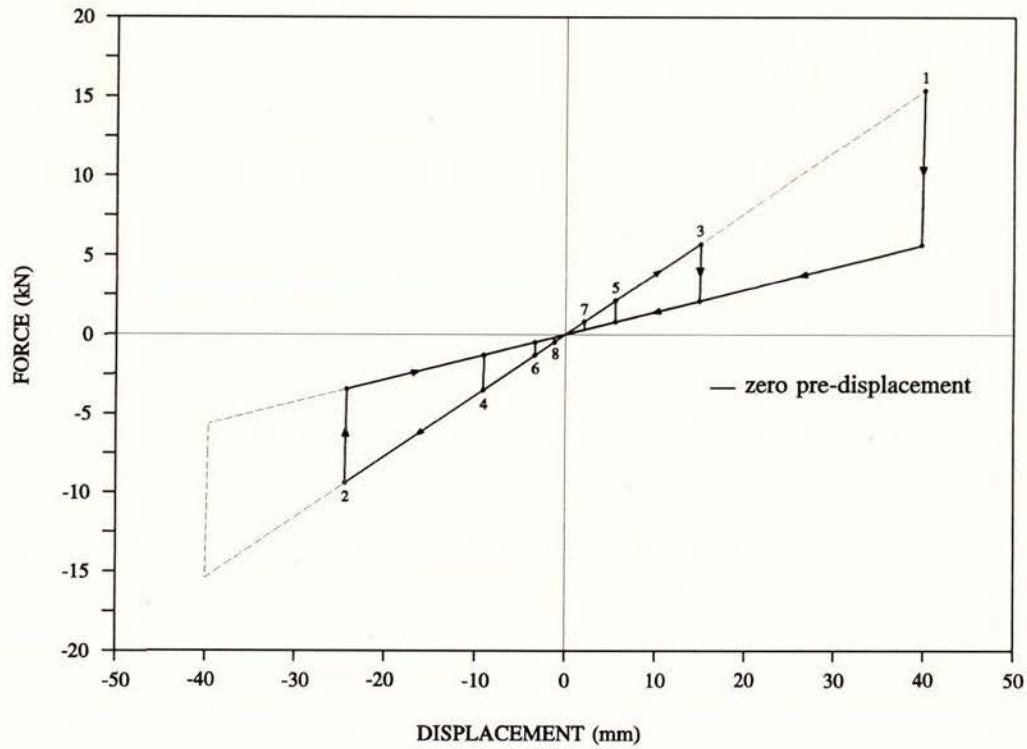
Discussion of the results obtained for the pre-displaced systems follow.



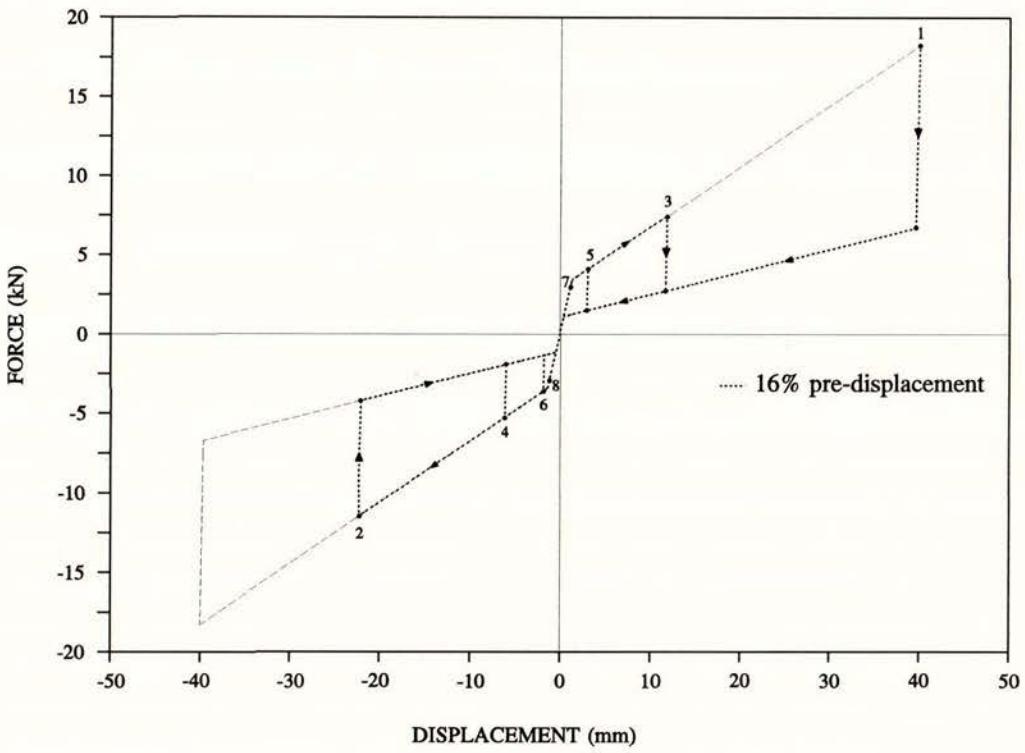
**Figure 5.4** Displacement response of pre-displaced mass/ring spring systems



**Figure 5.5** Acceleration response of pre-displaced mass/ring spring systems



(a)



(b)

**Figure 5.6** Pre-displaced ring spring force/deflection response curves:  
 (a) zero pre-displacement; (b) 16% pre-displacement

Initially, consider the system with zero pre-displacement. For this case, the dynamic response reduces to almost zero after four full cycles of motion. Additionally, the period between each full cycle of motion is seen to remain constant (Figure 5.4). This is because the secant stiffness,  $K_i$ , (measured between points 1 and 3 in Figure 5.2) remains constant as the response reduces. The dynamic response, as shown by the force/deflection response curve (Figure 5.6 (a)), shows motion proceeding between points 1-2-3-4-5-6-7-8 etc. Theoretically this decay in displacement will continue, but not reach, zero.

Results for the 16% pre-displacement system show that over the first few cycles of motion, the magnitude of the dynamic response is less than that for the system with zero pre-displacement (Figure 5.4). This is because energy absorption is greater for a pre-displaced system over the same displacement range.

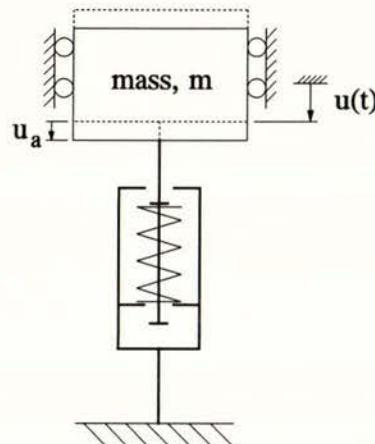
Also, the period for the pre-displaced system is seen to reduce over the first three full cycles (Figure 5.4). This is because the secant stiffness increases as the dynamic response reduces, thus, for larger displacements the system is more flexible. In addition, the displacement and acceleration responses of the pre-displaced system show a constant amplitude oscillation from 0.5 seconds onward (Figures 5.4 and 5.5). This results from energy absorption decreasing the magnitude of the response until motion decays onto stiffness  $K_0$ , i.e. oscillations occurring between points 7 and 8 on the force/deflection response curve (Figure 5.6 (b)). However, as all practical systems contain some degree of damping, these oscillations would rapidly diminish.

## **5.4 FREE VIBRATION BEHAVIOUR OF PRE-LOADED RING SPRING SYSTEMS**

This section considers the dynamic response of a SDOF mass/ring spring system subject to two levels of spring pre-load both with zero pre-displacement.

### **5.4.1 System model**

The model of a SDOF pre-loaded mass/ring spring system is shown in Figure 5.7.



**Figure 5.7** Pre-loaded mass/ring spring system

In this configuration, the ring spring cartridge supports load  $mg$ . Application of this load results in the system undergoing a displacement  $u_a$  (Figure 5.7); its magnitude given by  $u_a = mg/K_i$ . For a mass of 472 kg and a stiffness  $K_i = 385 \times 10^3$  N/m, this gives a spring compression of 12.0 mm.

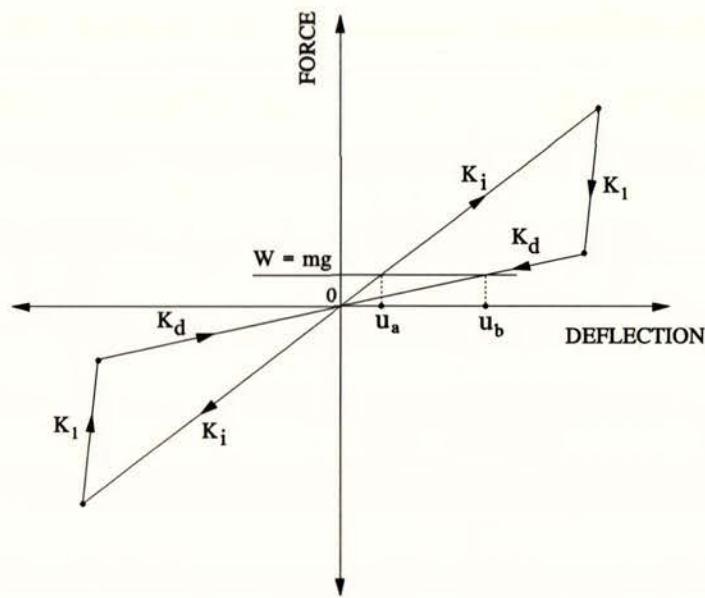
To quantify the amount of pre-load for a particular system, the term % pre-load has been developed. This is defined as follows:

$$\% \text{ pre-load} = \frac{\text{initial ring spring load}}{\text{maximum ring spring load}} \cdot \frac{100}{1} \quad (5.2)$$

For the above system with a maximum ring spring load capacity of 20 kN, the % pre-load =  $(472 \times 9.81 / 20000) \times 100 = 23.2\%$ .

#### 5.4.2 Force/deflection characteristics

The force/deflection diagram for the pre-loaded ring spring is the same as that for a pre-displaced spring with the addition of a new equilibrium line given by the pre-load force. The force/deflection diagram for the system with a pre-load force of  $W = mg$  is shown in Figure 5.8.



**Figure 5.8** Force/deflection diagram for pre-loaded ring springs with zero pre-displacement

#### 5.4.3 Equation of motion

The generalised equation of motion for the free vibration of pre-loaded mass/ring spring systems is:

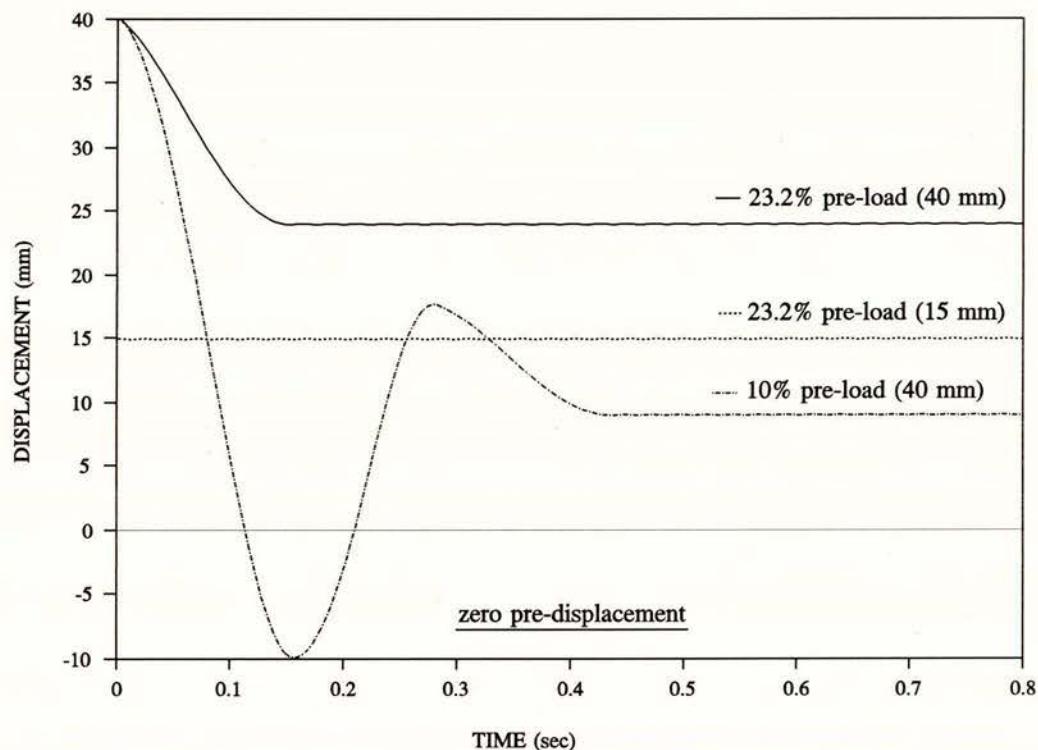
$$m\ddot{u} + k^*u = mg \quad (5.3)$$

where  $k^*$  and  $u(t)$  are as defined previously and  $mg$  is a constant force acting on the system. This equation can be solved using the ring spring model computer algorithm.

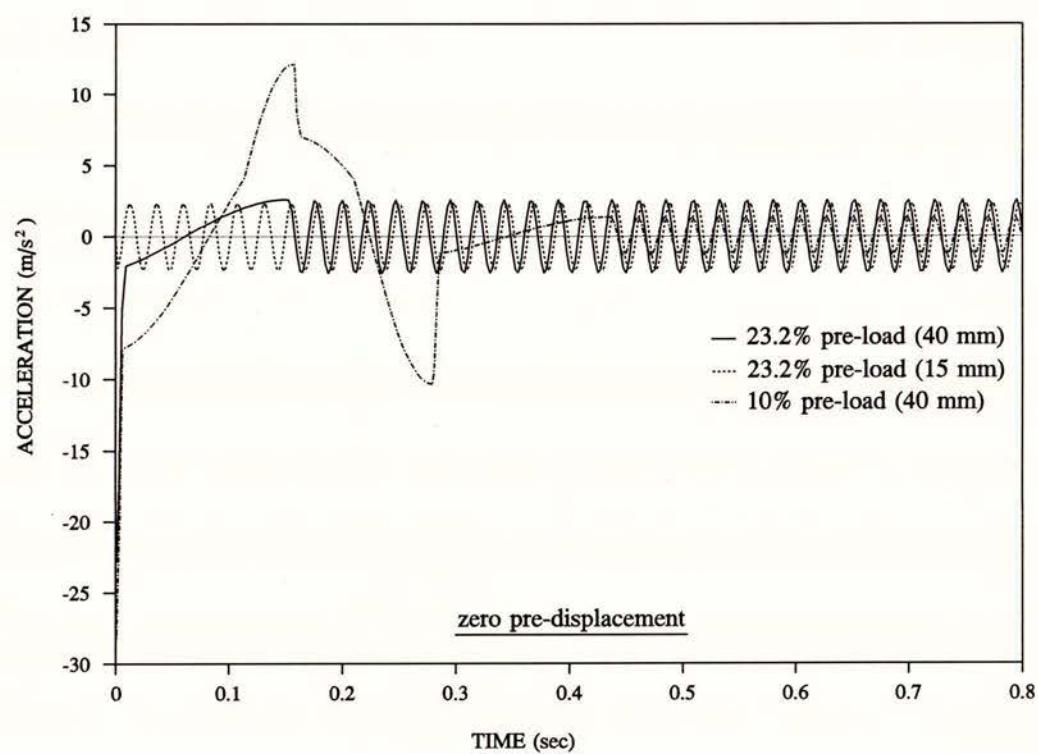
#### 5.4.4 Dynamic response

The free vibration response of a mass/ring spring system supporting weight  $mg$  (i.e. for 23.2% pre-load) and released from initial displacements of 40 mm and 15 mm was determined. In addition, the response of a system with a 10% pre-load, released from an initial displacement of 40 mm, was obtained. The displacement and acceleration responses for these systems are presented in Figures 5.9 and 5.10, and their force/deflection response curves are given in Figures 5.11 (a) and (b).

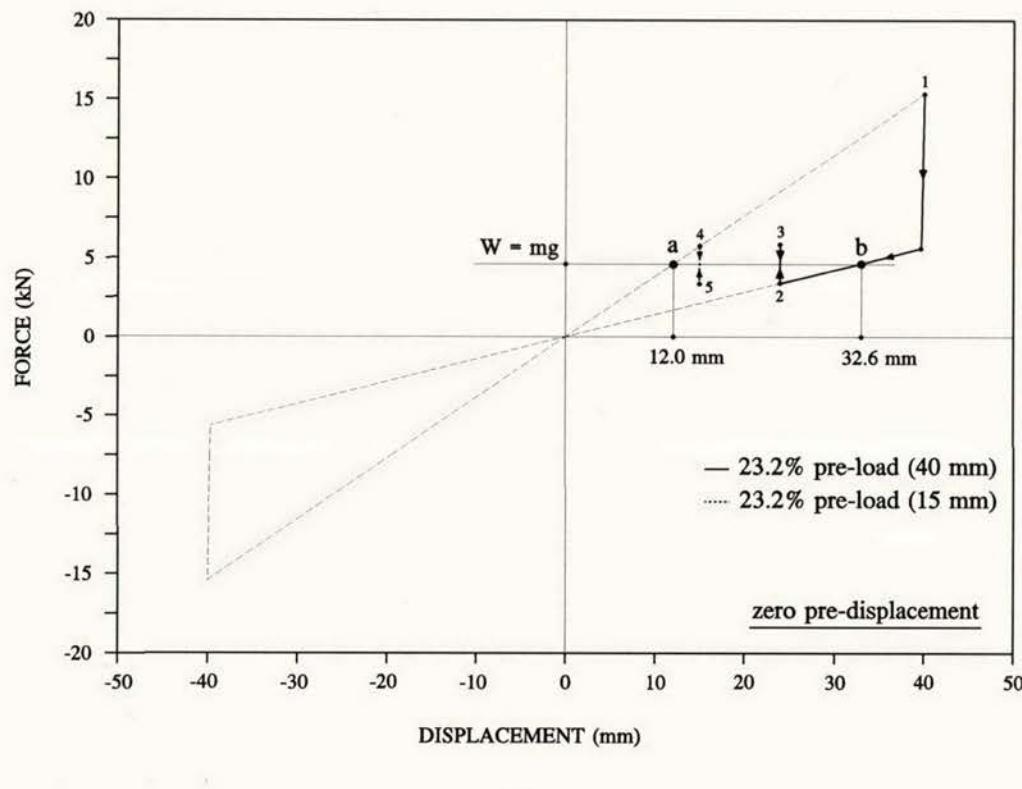
Discussion of the results for the pre-loaded systems follow.



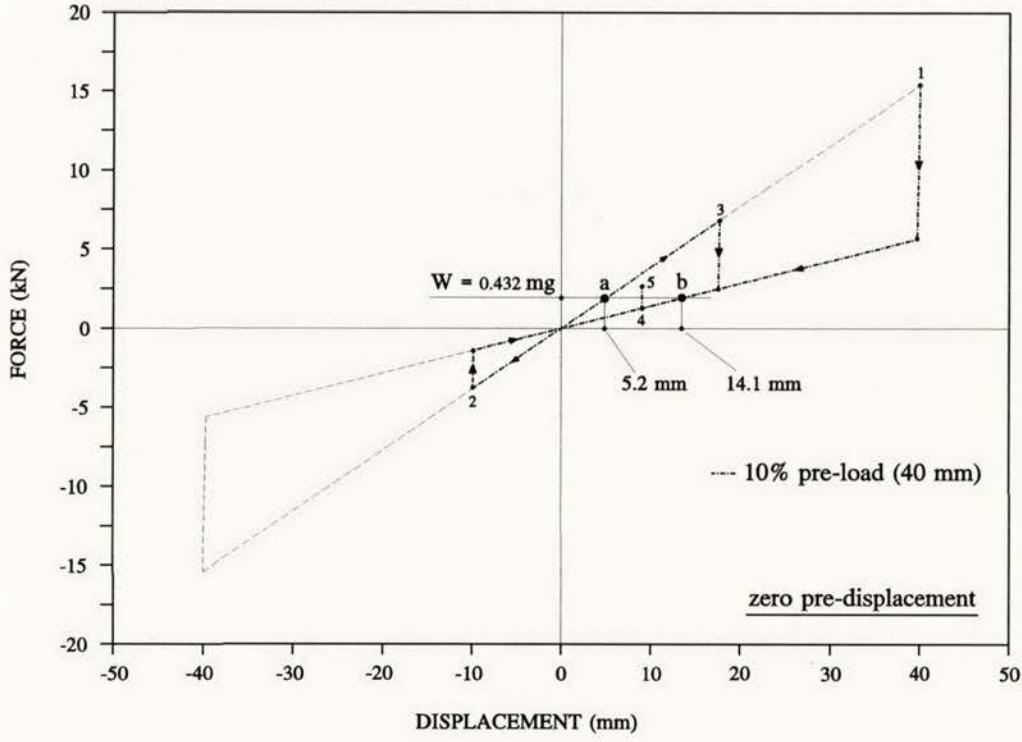
**Figure 5.9** Displacement response of pre-loaded ring springs



**Figure 5.10** Acceleration response of pre-loaded ring springs



(a)



(b)

**Figure 5.11** Pre-loaded ring spring force/deflection response curves:  
 (a) 23.2% pre-load; (b) 10% pre-load

For the system with 23.2% pre-load released from an initial displacement of 40 mm, the dynamic response reduces to approximately 24 mm after  $\frac{1}{2}$  cycle. Thereafter this system oscillates at small amplitude displacements (Figure 5.9). The same system, when released from a displacement of 15 mm, vibrates at small amplitude oscillations only (Figure 5.9); the acceleration of these oscillations are seen in Figure 5.10.

These effects are more clearly shown in the force/deflection response curve shown in Figure 5.11 (a). When the system is released from a displacement of 40 mm (point 1), motion proceeds about the equilibrium line  $W = mg$  to point 2. Subsequently, small amplitude oscillations occur between points 2-3. When the system is released from 15 mm, motion proceeds about the equilibrium line between points 4-5.

It is seen from these results that once motion settles into zone a-b, small amplitude oscillations proceed about the equilibrium line  $W = mg$ . The larger the pre-load force, the larger the zone a-b. The larger this zone, the more likely the response will be solely within this region. It should be noted that for all situations incorporating ring springs, energy is absorbed only when motion proceeds on stiffnesses  $K_i$  or  $K_d$ ; stiffnesses  $K_0$  and  $K_1$  being elastic. Thus, for small amplitude oscillations within zone a-b, energy is not dissipated.

It is beneficial at this point to investigate the effect of reducing system pre-load. In the above case, the load of 100%  $mg$  compressed the ring spring 12.0 mm. If the ring spring carries a load of only 43.2%  $mg$ , the spring will compress 5.2 mm. This pre-load equates to 10% of the maximum spring load, i.e. a 10% pre-load. For this system the remainder of the load (56.8%  $mg$ ) is supported by other means.

The dynamic response for the case of 10% pre-load, released from an initial displacement of 40 mm, is presented in Figure 5.9. The response undergoes  $1\frac{1}{2}$  cycles of motion, then continues to oscillate at small amplitude; this is observed in the acceleration solution (Figure 5.10). This response is shown in the force/deflection response curve (Figure 5.11 (b)). Motion proceeds between points 1-2, 2-3, and 3-4, with energy being absorbed along stiffnesses  $K_d$  and  $K_i$ . Motion then continues between points 4-5 on stiffness  $K_1$ . For the 10% pre-load system, energy absorption continues for  $1\frac{1}{2}$  cycles; this compares with  $\frac{1}{2}$  cycle for the 23.2% pre-load case. This increase in the number of cycles of energy absorption is due to a reduced pre-load force, i.e. a smaller zone a-b. For the 10% pre-load

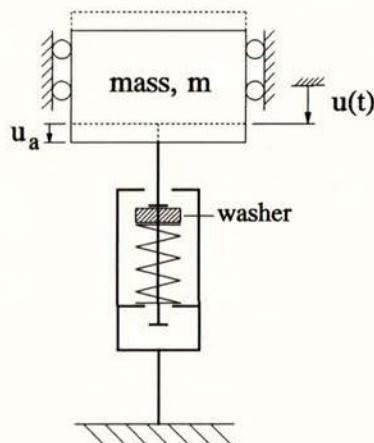
case, zone a-b falls between 5.2 and 14.1 mm (Figure 5.11 (b)), this being much smaller than the zone between 12.0 and 32.6 mm for 23.2% pre-load case (Figure 5.11(a)).

## 5.5 FREE VIBRATION BEHAVIOUR OF PRE-DISPLACED PRE-LOADED RING SPRING SYSTEMS

This section details the dynamic response of a SDOF mass/ring spring system that is both pre-displaced and pre-loaded. The system analysed is subjected to two levels of spring pre-load and has a pre-displacement of 16%.

### 5.5.1 System model

The model of a SDOF pre-displaced pre-loaded mass/ring spring system is shown in Figure 5.12.

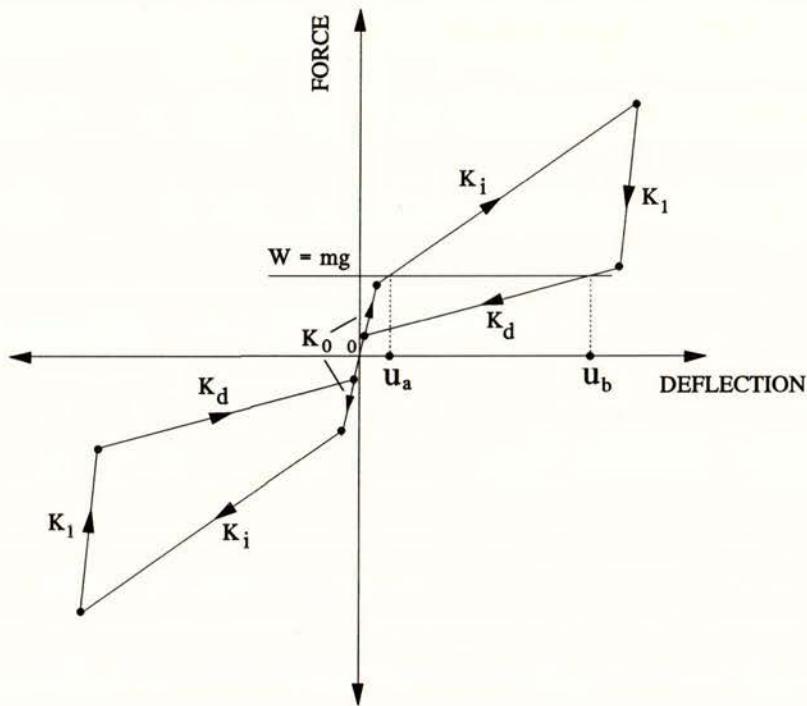


**Figure 5.12** Pre-displaced pre-loaded mass/ring spring system

In this system a pre-displaced ring spring cartridge supports load  $mg$ . For the case where the total weight (100%  $mg$ ) is carried by the cartridge, this equates to a pre-load of 23.2% (as calculated in Section 5.4.1).

### 5.5.2 Force/deflection characteristics

The force/deflection diagram for a pre-displaced pre-loaded ring spring is the same as that for a pre-displaced system with an equilibrium line given by the pre-load force. The force/deflection diagram for the system with a pre-load of  $W = mg$  is shown in Figure 5.13.



**Figure 5.13** Force/deflection diagram for pre-displaced pre-loaded ring springs

### 5.5.3 Equation of motion

The equation of motion for a pre-displaced pre-loaded system is given by:

$$m\ddot{u} + k^*u = mg \quad (5.4)$$

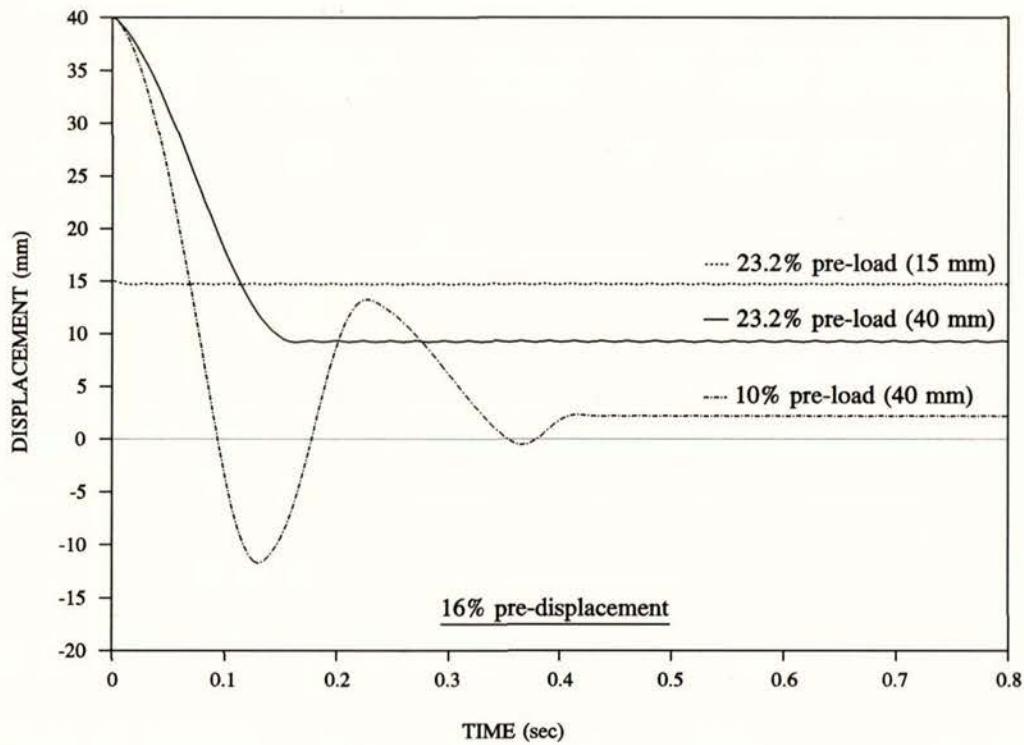
This equation is the same as that for a pre-loaded ring spring system with zero pre-displacement (Equation 5.3).

### 5.5.4 Dynamic response

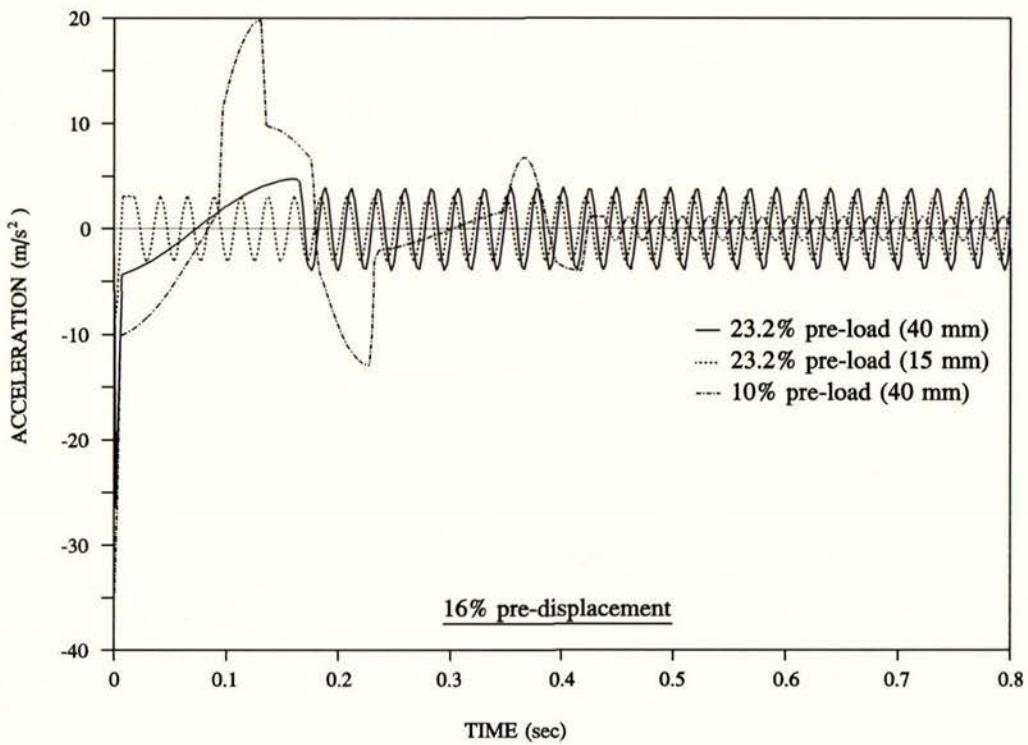
The free vibration response of a mass/ring spring system of 16% pre-displacement and 23.2% pre-load, released from initial displacements of 40 mm and 15 mm, was computed. Also analysed was a system with 10% pre-load released from a displacement of 40 mm.

The displacement and acceleration response for these systems are presented in Figures 5.14 and 5.15. Their force/deflection response curves are given in Figures 5.16 (a) and (b).

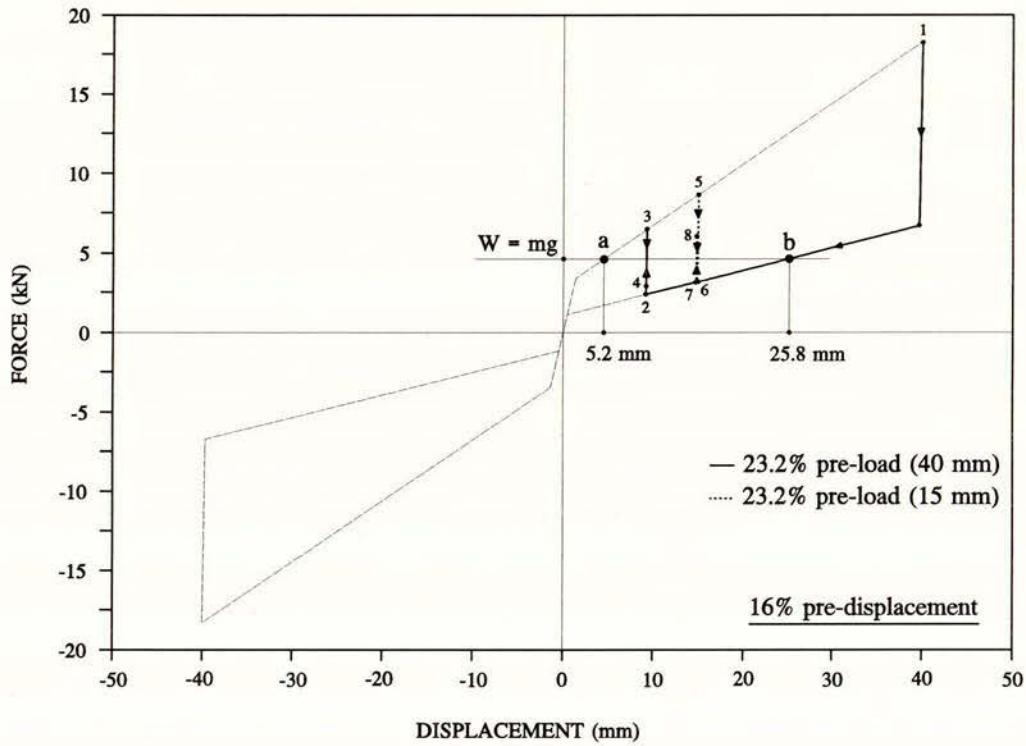
Results obtained for these pre-displaced pre-loaded systems are now discussed.



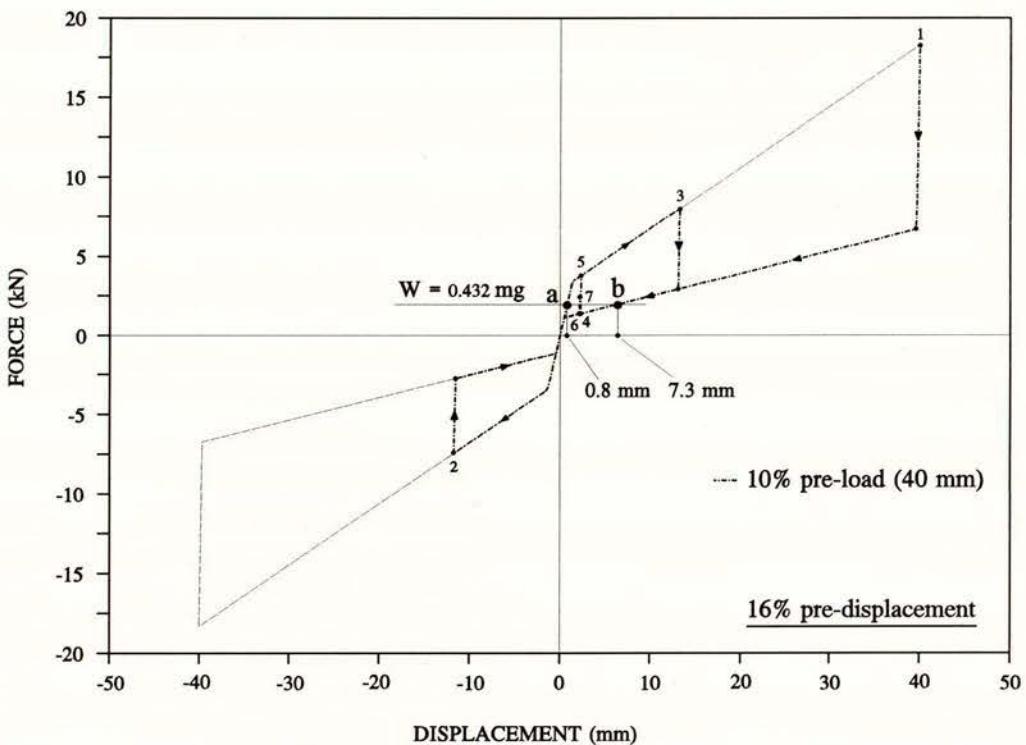
**Figure 5.14** Displacement response of pre-displaced pre-loaded ring springs



**Figure 5.15** Acceleration response of pre-displaced pre-loaded ring springs



(a)



(b)

**Figure 5.16** Pre-displaced pre-loaded ring spring force/deflection response curves:  
 (a) 23.2% pre-load; (b) 10% pre-load

For the 23.2% pre-load system released from displacements of 40 mm and 15 mm, motion reduces to small amplitude oscillations after  $\frac{1}{2}$  cycle (Figure 5.14). In the latter, this  $\frac{1}{2}$  cycle of motion occurs over a very small displacement. The oscillations beyond the first  $\frac{1}{2}$  cycle of motion are shown on the force/deflection response curve between points 3 and 4 for the 40 mm displacement, and between points 7 and 8 for the 15 mm displacement (Figure 5.16 (a)).

For the system with a reduced pre-load of 10%, the dynamic response undergoes 2 full cycles of motion, then proceeds to oscillate at small amplitude (Figure 5.14). This response is shown in Figure 5.16 (b) where motion proceeds around points 1-2, 2-3, 3-4, 4-5, 5-6, and then oscillates between points 6-7 on stiffness  $K_1$ .

Again, greater energy is absorbed for the case of reduced pre-load due to its smaller zone a-b. This zone spans between 0.8 mm and 7.3 mm for 10% pre-load (Figure 5.16 (b)), and between 5.2 mm and 25.8 mm for the 23.2% pre-load (Figure 5.16 (a)).

## 5.6 SUMMARY

The free vibration response of pre-displaced, pre-loaded, and pre-displaced pre-loaded mass/ring spring systems have been analysed.

The response of a mass/ring spring system with zero pre-displacement is shown to reduce to almost zero after four cycles of motion. This is due to the large energy absorption capability of the ring spring. For a pre-displaced mass/ring spring system, increased energy is absorbed over a given displacement range than for a system with zero pre-displacement. Hence, for the same initial displacement the free vibration response of a pre-displaced system reduces more rapidly.

In pre-loaded mass/ring spring systems, the dynamic response proceeds about an equilibrium line equal to the pre-load force. A zone a-b is set up within which motion eventually settles and oscillates at small amplitude displacements. For increased pre-load, zone a-b increases, this in turn reduces the energy absorbed during free vibration of a particular system.

## Chapter 6

# DYNAMIC TESTING OF A PRE-DISPLACED MASS/RING SPRING CARTRIDGE SYSTEM: COMPARISON WITH COMPUTER SIMULATION RESULTS

### 6.1 INTRODUCTION

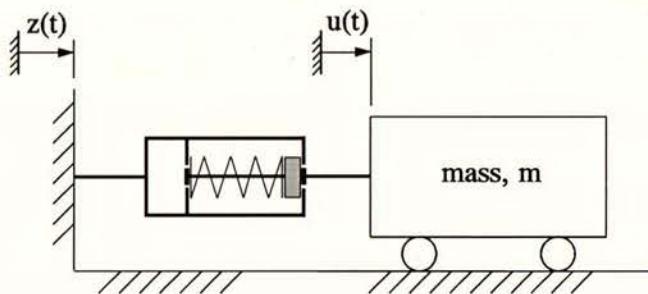
Computer analyses provide a powerful tool for determining the response of structural systems subject to dynamic excitation. However, it is important that their accuracy be determined by comparing results with those given by experimental testing. This chapter details results of experimental testing of a mass/ring spring system subject to short duration excitation. The system comprises of a ring spring cartridge (detailed in Chapter 4) attached to a rigid mass and mounted on a shaker table. Shaker table test results for spring pre-displacements of 16% and 28% are presented. These results are then compared with those given by computer simulation.

### 6.2 EXPERIMENTAL SYSTEM

To enable the dynamic behaviour of mass/ring spring systems to be analysed experimentally, a test assembly was built and subsequently tested (Hill 1995c). This system is now detailed.

#### 6.2.1 Mass/ring spring cartridge system

The experimental system comprises of a bi-directional ring spring cartridge attached to a mass,  $m$ , and mounted on a shaker as shown in Figure 6.1.



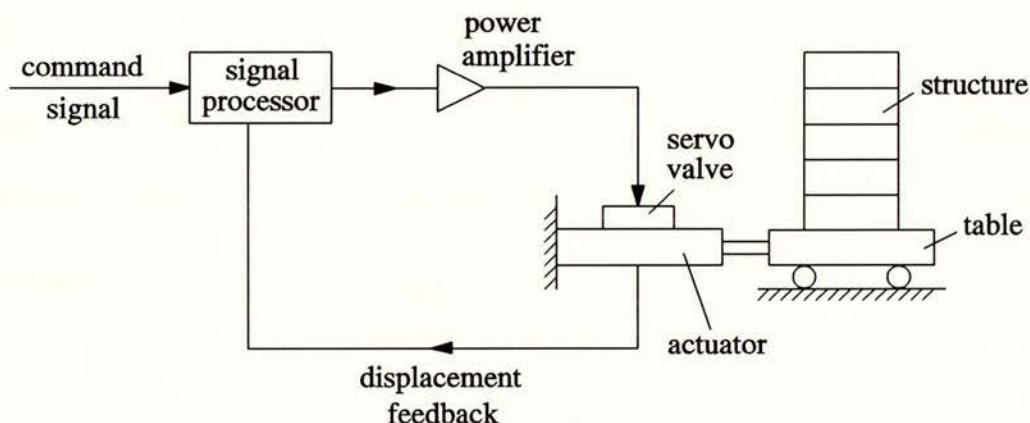
**Figure 6.1** Schematic of mass/ring spring cartridge system

### 6.2.2 Experimental test facility

A shaker table facility, located at Industrial Research Ltd, Christchurch, was selected for the experimental tests. Specifications for the shaker are as follows:

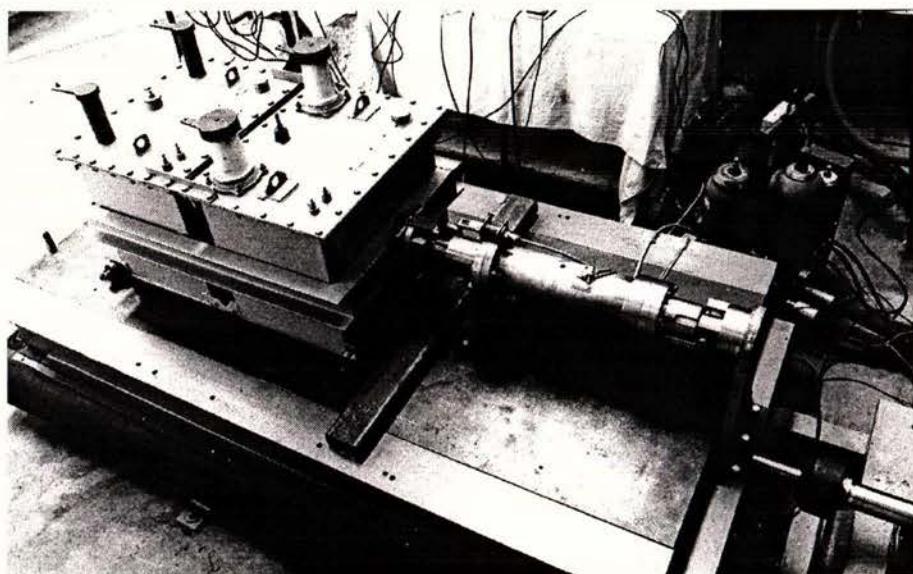
- shaker table size: 1 m x 2 m (plan area)
- maximum operating capacity:  $\pm 125$  mm @ 1 Hz
- maximum load: 5000 kg at table level

The shaker table is controlled using a closed loop system. For this, a command signal is sent to the shaker servo-valve; the table then displaces and an actuator transducer returns the measured displacement to a signal processor for comparison and compensation if necessary, thus tracking the desired input. A schematic of the shaker table control system is shown in Figure 6.2.



**Figure 6.2** Schematic of shaker table control system

The experimental test assembly comprises of the ring spring cartridge (detailed in Chapter 4) attached to two small transformers rigidly coupled together and of total mass 472 kg; the ring spring cartridge uses either 16% or 28% pre-displacement. The system mounted on the shaker table is shown in Figure 6.3.



**Figure 6.3** Configuration of test rig on shaker table

### 6.2.3 Instrumentation

For the experimental tests, instrumentation was required to measure:

- the shaker table input acceleration;
- the acceleration of the rigid mass, and
- the ring spring cartridge displacement.

#### (a) Accelerometers

Accelerations of the shaker table and the rigid mass were measured using two Brüel and Kjær (B&K) accelerometers. These were attached to the shaker table actuator and directly to the rigid mass. The accelerometers were calibrated using a B&K accelerometer calibrator so that a 1.0 g acceleration produced a charge amplifier output of 1.0 volt.

(b) Displacement transducer

The ring spring cartridge displacement was measured using a Pulsonic displacement sensor rigidly attached to the ring spring cartridge. The transducer was calibrated so that a 25.4 mm displacement gives an output of 1 volt.

#### 6.2.4 Data acquisition system

A PC fitted with a 16 channel data acquisition board was used to record the results. For all the tests undertaken, a sampling frequency of 300 Hz was used.

### 6.3 COMPUTATIONAL MODELLING

To enable computer simulation results to be compared with those given experimentally a computer model was developed to determine the response of mass/ring spring systems subject to dynamic inputs.

#### 6.3.1 Computer model for mass/ring spring system

The system comprises of a mass,  $m$ , and a pre-displaced ring spring cartridge (as shown in Figure 6.1). The dynamic response of this system is evaluated by expressing the generalised equation of motion as:

$$m\ddot{u} + k^*(u-z) = 0 \quad (6.1)$$

where:

- $k^*$  is the particular stiffness on the force/deflection diagram at the instant of motion considered (either  $K_0$ ,  $K_i$ ,  $K_l$ , or  $K_d$ );
- $z(t)$  is the input displacement; and
- $u(t)$  is the displacement of the mass.

This expression may be rewritten in terms of a relative displacement formulation by using  $w = (u-z)$ , where  $w(t)$  defines the ring spring displacement. Equation 6.1 now becomes:

$$m\ddot{w} + k^*w = -m\ddot{z} \quad (6.2)$$

where  $\ddot{z}$  is the input acceleration.

This equation can be solved using the computer algorithm detailed in Chapter 5, where  $-m\ddot{z}$  may be considered as the effective load.

### 6.3.2 Ring spring cartridge stiffnesses

Equation 6.2 requires values for the ring spring cartridge stiffnesses. The values used in the computer algorithm were those determined from experimental testing of the ring spring cartridge (as reported in Chapter 4). These stiffnesses are as follows.

For the ring spring cartridge subject to 16% pre-displacement:

$$K_0 = 2.5 \times 10^6 \text{ N/m}$$

$$K_i = 385 \times 10^3 \text{ N/m}$$

$$K_l = 33 \times 10^6 \text{ N/m}$$

$$K_d = 142 \times 10^3 \text{ N/m}$$

For the ring spring cartridge subject to 28% pre-displacement:

$$K_0 = 2 \times 10^6 \text{ N/m}$$

$$K_i = 410 \times 10^3 \text{ N/m}$$

$$K_l = 38 \times 10^6 \text{ N/m}$$

$$K_d = 135 \times 10^3 \text{ N/m}$$

## 6.4 EXPERIMENTAL TESTS AND COMPUTER SIMULATION RESULTS

This section details the results of experimental tests and computer simulation analyses for the mass/ring spring system. Testing was conducted for two ring spring cartridge pre-displacements: 16% and 28%, and each system was subjected to two dynamic inputs.

### 6.4.1 Dynamic inputs

The two dynamic inputs used in the study are:

- 1 second shock input,
- 4 second dynamic input.

The 1 second shock input was generated by manually controlling the shaker table displacement using a hand-held potentiometer. Different signals were obtained for the 16% and 28% pre-displaced systems.

The 4 second dynamic input was obtained using a signal derived from the first 12 seconds of the Pacoima 1971 S16E earthquake (this earthquake is detailed in Chapter 7). The earthquake acceleration was trebled and its time base reduced by a factor of three to give the desired short duration excitation, i.e., from a 12 second record to a 4 second record. This signal was then used as the shaker table input for the 16% and 28% pre-displaced systems, thus enabling these experimental results to be compared for the same input.

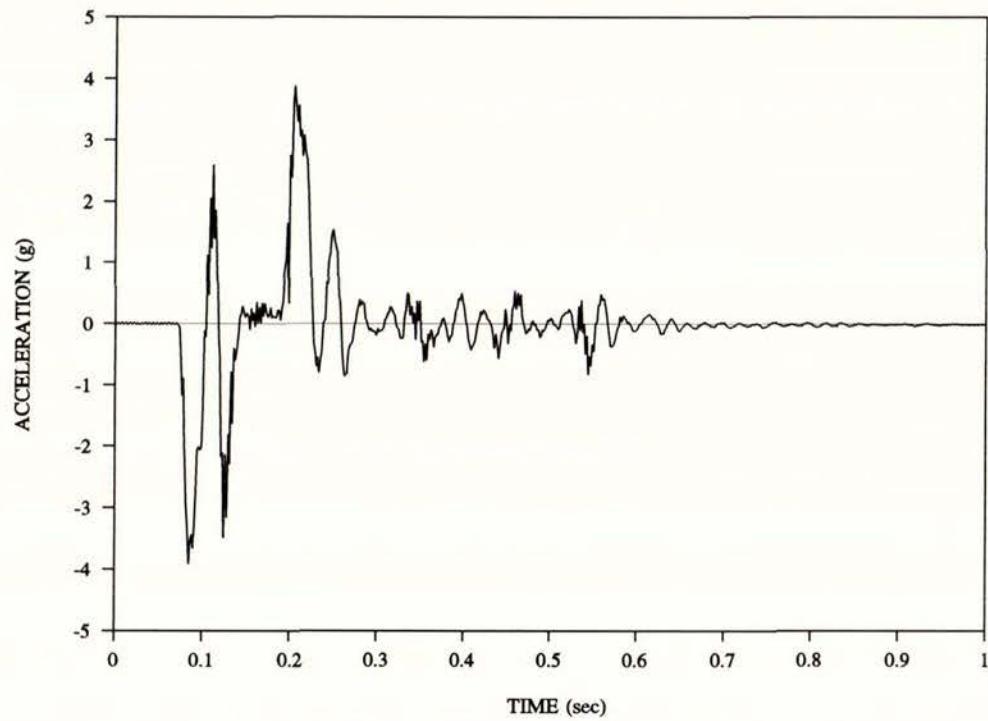
The measured shaker table accelerations were used as the computer algorithm inputs to enable experimental and computational results to be compared for the same excitation.

#### **6.4.2 Dynamic response of system subjected to 1 second input**

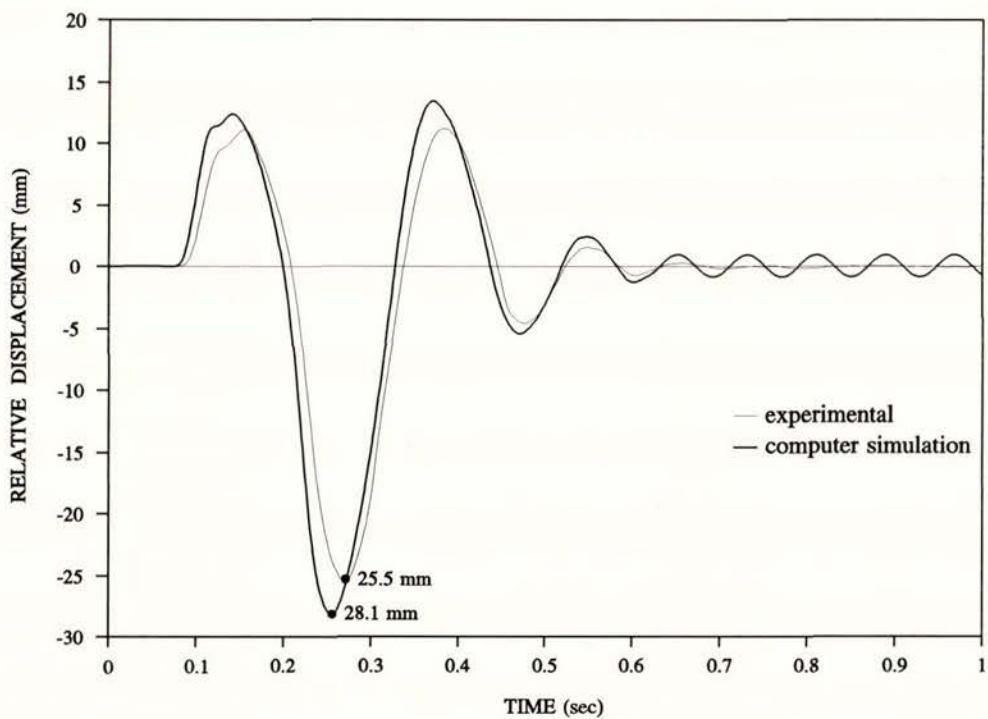
##### **(a) 16% pre-displacement**

The graphical results presented for the 16% pre-displaced system subjected to the 1 second shock input are:

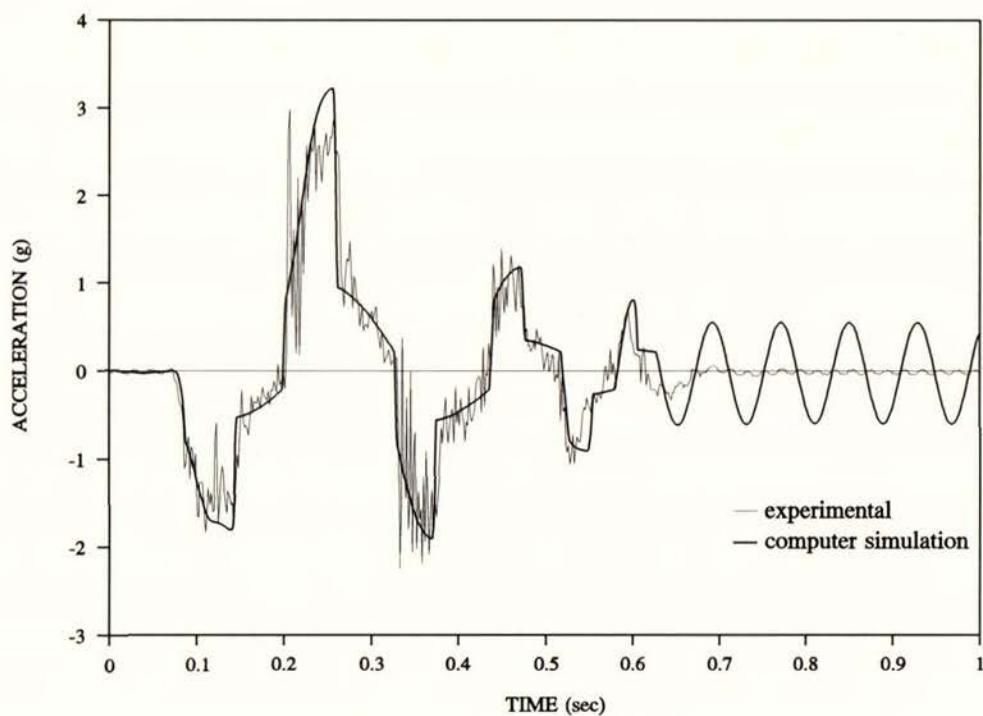
- recorded shaker table input acceleration (Figure 6.4)
- ring spring cartridge displacement - experimental and computational (Figure 6.5)
- acceleration of the mass - experimental and computational (Figure 6.6)



**Figure 6.4** Recorded shock input (1 second)



**Figure 6.5** Displacement response: experimental and computer simulation  
(16% pre-displacement)

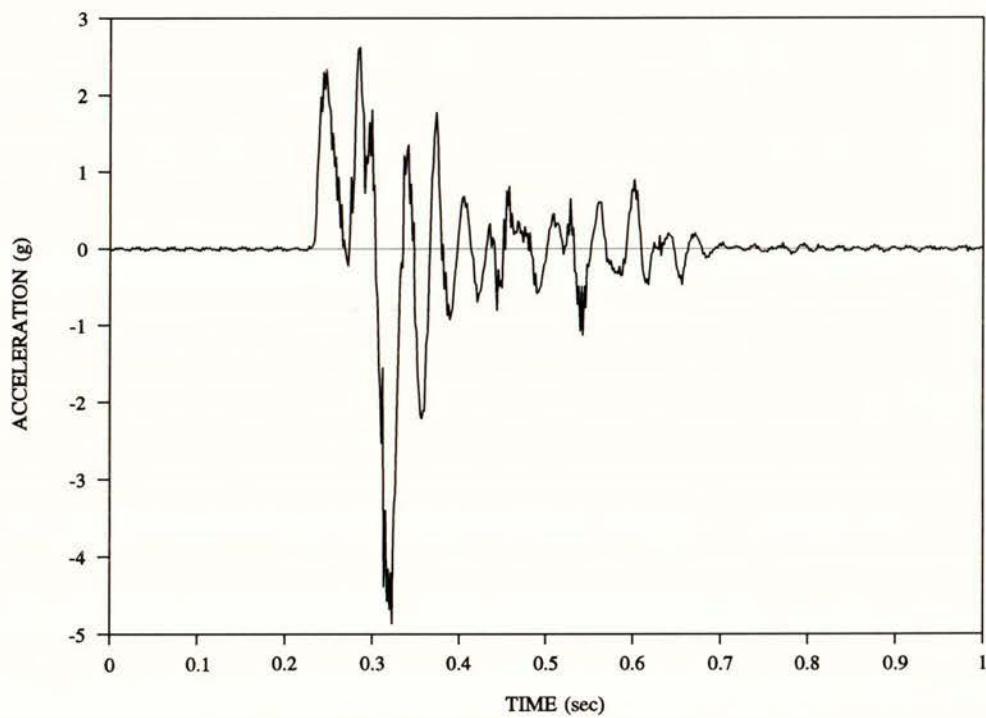


**Figure 6.6** Acceleration response: experimental and computer simulation  
(16% pre-displacement)

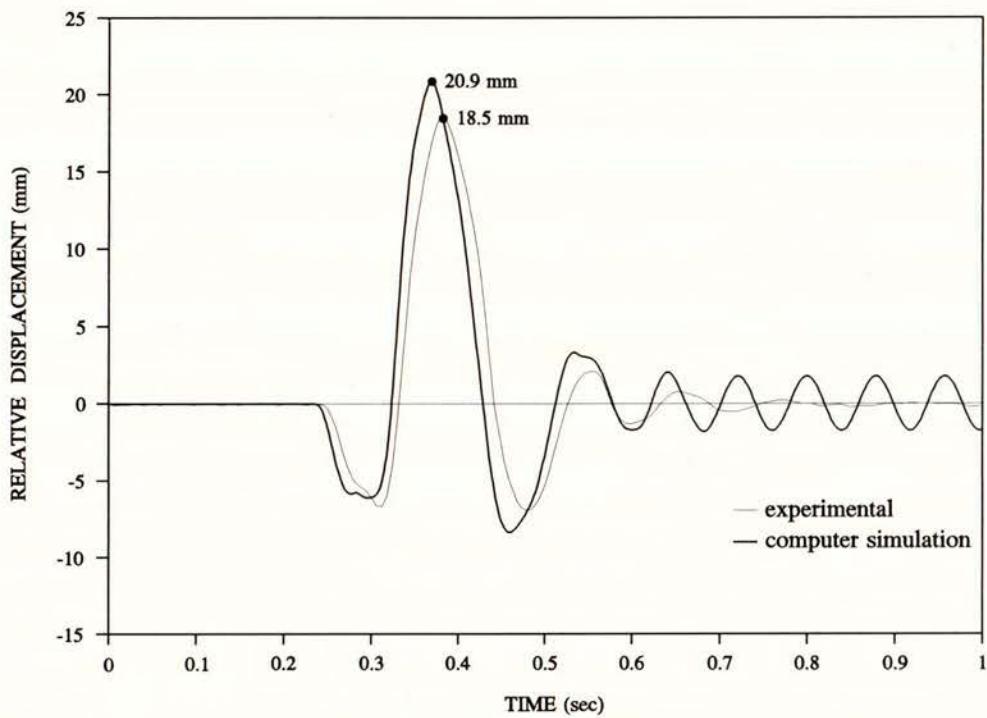
(b) 28% pre-displacement

The graphical results presented for the 28% pre-displaced system subjected to the 1 second shock input are:

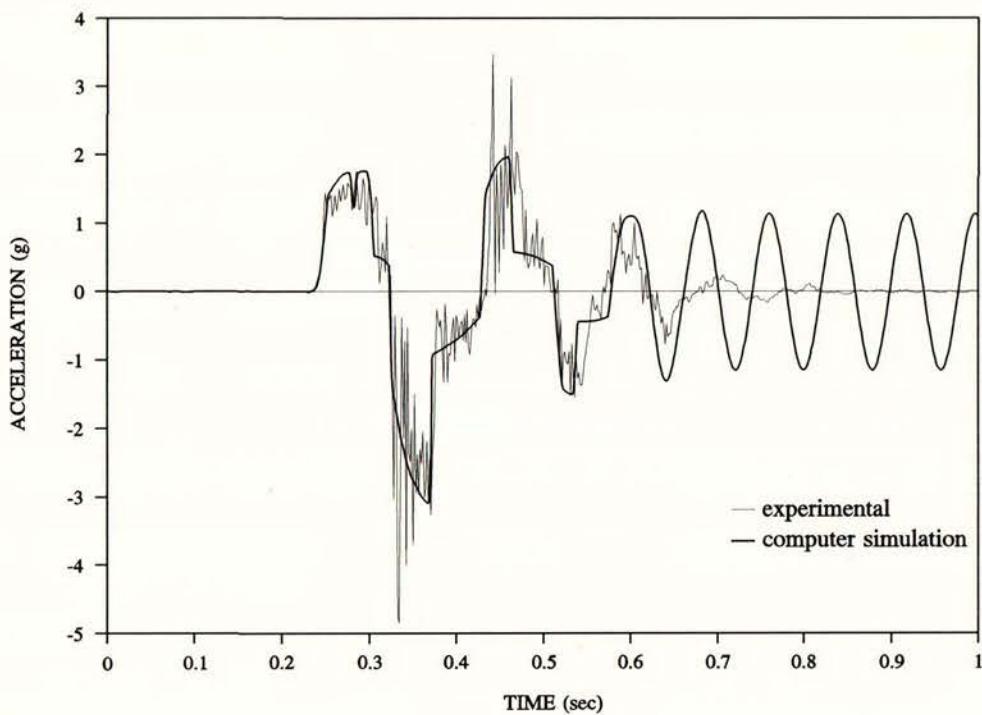
- recorded shaker table input acceleration (Figure 6.7)
- ring spring cartridge displacement - experimental and computational (Figure 6.8)
- acceleration of the mass - experimental and computational (Figure 6.9)



**Figure 6.7** Recorded shock input (1 second)



**Figure 6.8** Displacement response: experimental and computer simulation  
(28% pre-displacement)



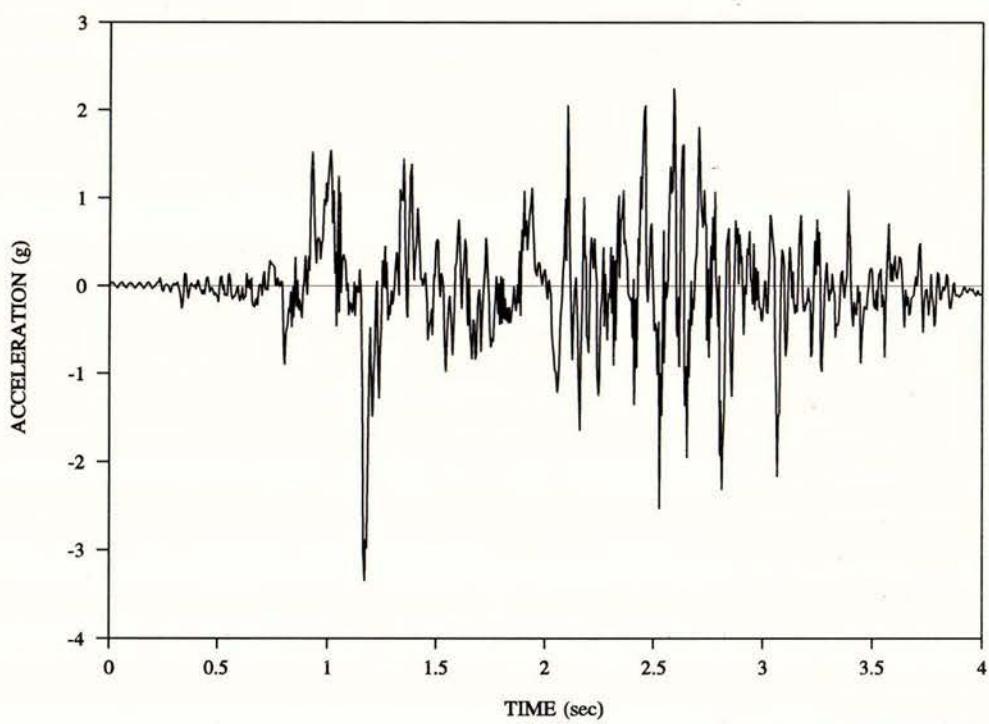
**Figure 6.9** Acceleration response: experimental and computer simulation  
(28% pre-displacement)

#### 6.4.3 Dynamic response of system subjected to 4 second input

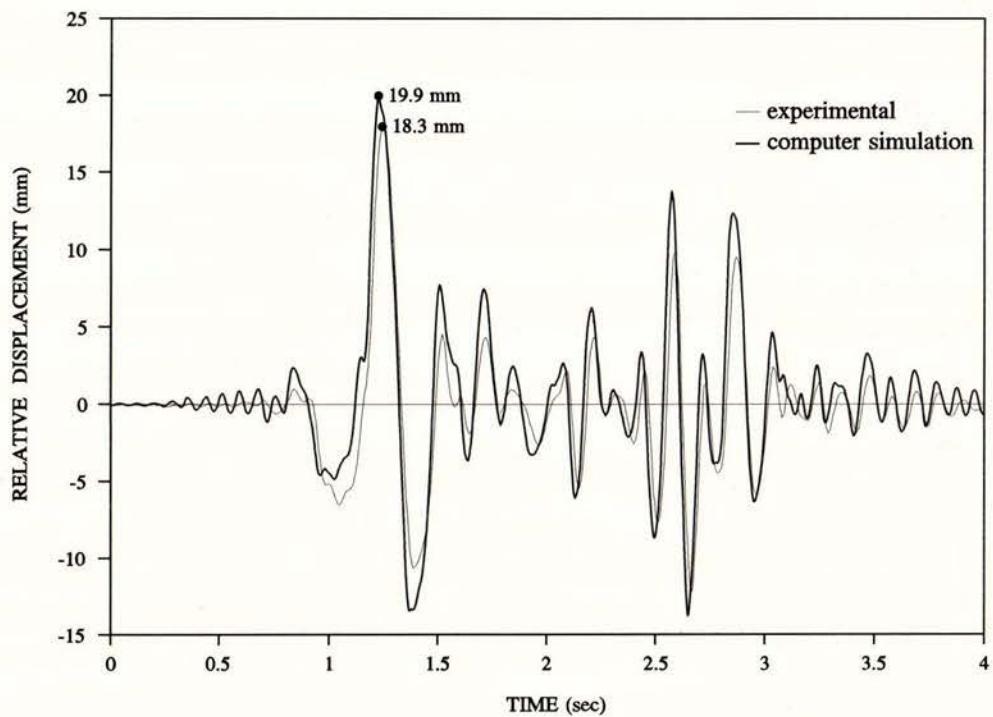
(a) 16% pre-displacement

The graphical results presented for the 16% pre-displaced system subjected to the 4 second dynamic input are:

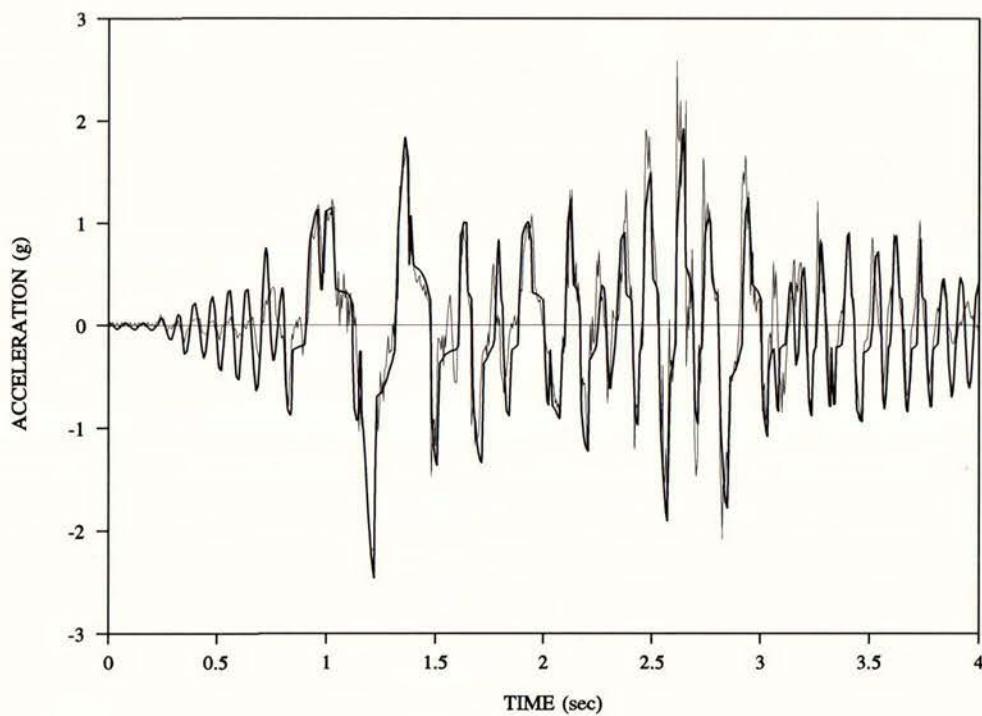
- recorded shaker table input acceleration (Figure 6.10)
- ring spring cartridge displacement - experimental and computational (Figure 6.11)
- acceleration of the mass - experimental and computational (Figure 6.12)



**Figure 6.10** Recorded shock input (4 second)



**Figure 6.11** Displacement response: experimental and computer simulation  
(16% pre-displacement)

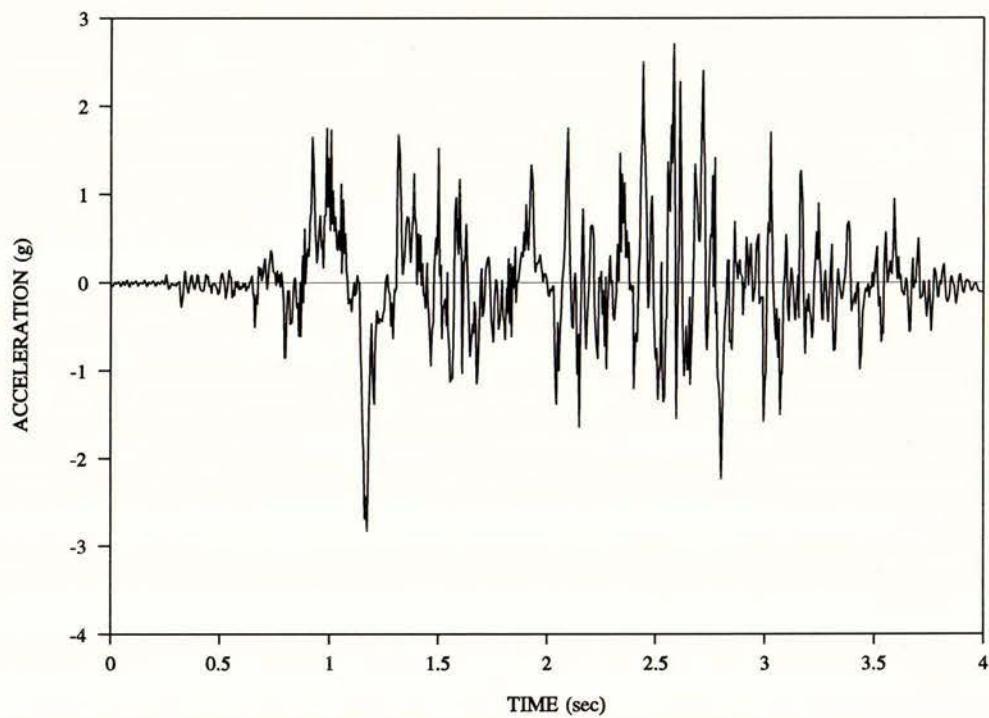


**Figure 6.12** Acceleration response: experimental and computer simulation  
(16% pre-displacement)

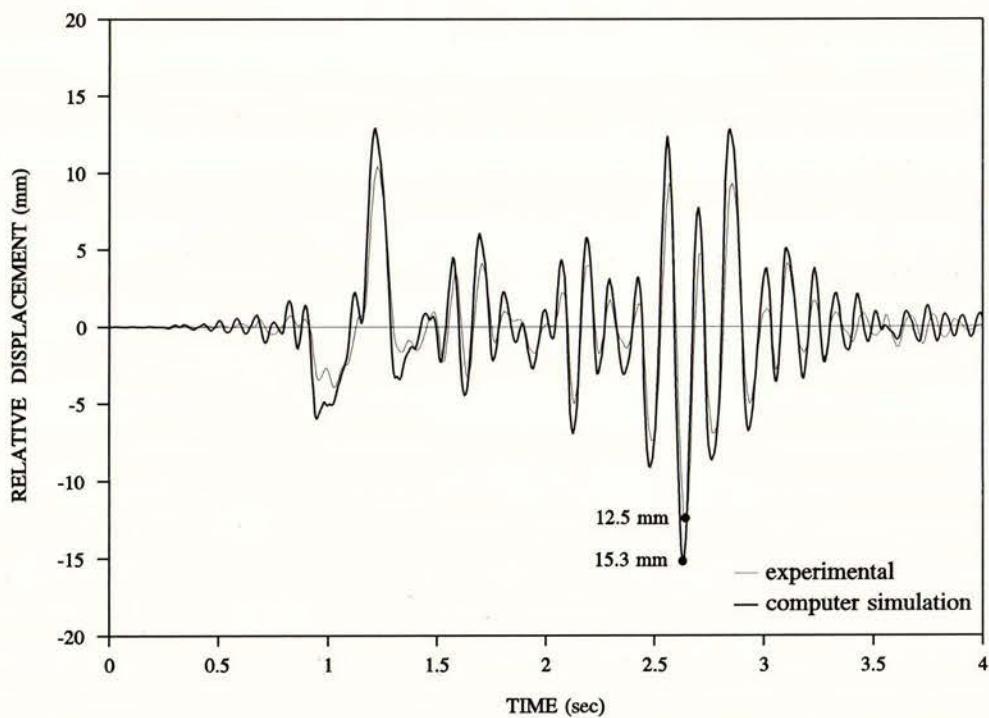
(b) 28% pre-displacement

The graphical results presented for the 28% pre-displaced system subjected to the 4 second dynamic input are:

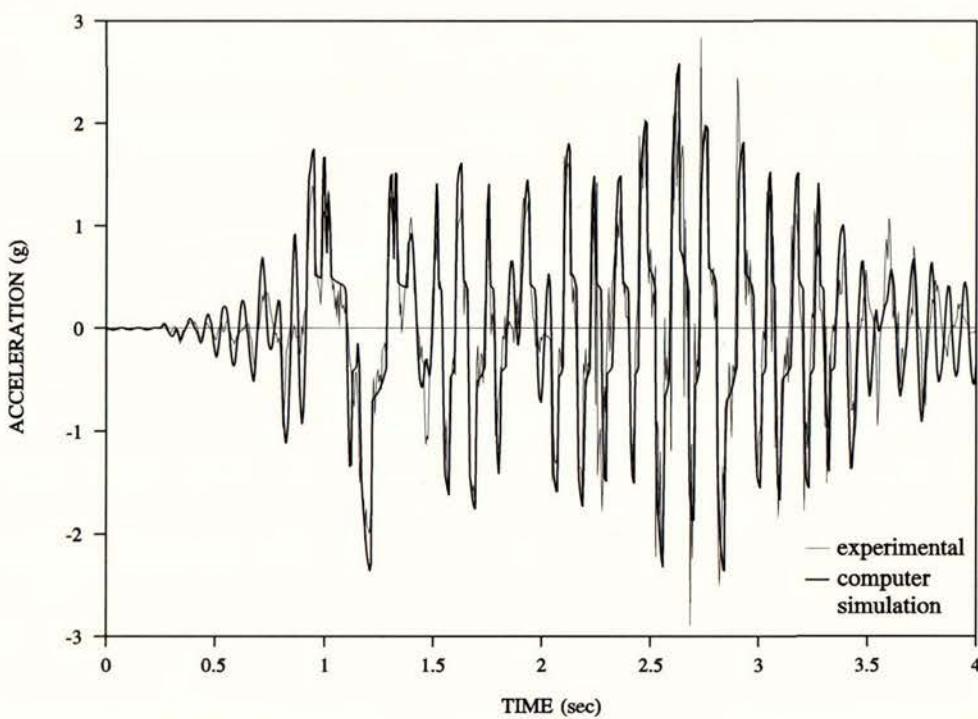
- recorded shaker table input acceleration (Figure 6.13)
- ring spring cartridge displacement - experimental and computational (Figure 6.14)
- acceleration of the mass - experimental and computational (Figure 6.15)



**Figure 6.13** Recorded shock input (4 second)



**Figure 6.14** Displacement response: experimental and computer simulation  
(28% pre-displacement)



**Figure 6.15** Acceleration response: experimental and computer simulation (28% pre-displacement)

#### 6.4.4 Discussion of results

The ring spring cartridge displacements obtained by experimental tests and computer simulation, are seen to be in good agreement for all four cases investigated. This shows that the force/deflection characteristics behaved in accordance with those obtained by quasi-static testing. A summary of the displacement results is presented in Table 6.1.

**Table 6.1** Summary of displacement results

Test Condition	Displacement (mm)	
	Experimental	Computational
1 sec input (16% pre-displacement)	25.5 mm	28.1 mm
1 sec input (28% pre-displacement)	18.5 mm	20.9 mm
4 sec input (16% pre-displacement)	18.3 mm	19.9 mm
4 sec input (28% pre-displacement)	12.5 mm	15.3 mm

The maximum displacements for the computer simulation results are greater than those obtained experimentally. However, as the experimental system contains additional damping in the form of roller bearing friction, etc., this would account for an overall reduction in the experimental response.

For the 16% and 28% pre-displacement systems subjected to the 1 second shock inputs, the experimental displacements decay to almost zero shortly after the impulsive input is applied (Figures 6.5 and 6.8). However, the computer simulation displacements continue to oscillate at constant amplitude. This is because the computer simulation response has reduced onto the undamped stiffness  $K_0$  on the ring spring hysteresis diagram. As all real systems contain damping, these oscillations would soon diminish (as shown by the experimental results).

For the 1 second shock input, short duration pulses are seen in the experimental acceleration results (Figures 6.6 and 6.9). These can be attributed to contacting between the piston and the stop faces within the ring spring cartridge during loading/unloading. This effect is less noticeable for the 4 second dynamic inputs because the input signal is less impulsive and of reduced magnitude.

## 6.5 SUMMARY

The results presented show that experimental tests and computer simulation results agree well and that the ring spring hysteresis characteristics behaved in accordance with those obtained by quasi-static testing. For the systems subjected to the 1 second shock input, experimental results show that displacements reduce to almost zero after three full cycles of motion. This is due to the high energy absorption capability of the ring spring.

## Chapter 7

# EARTHQUAKE-RESISTANT DESIGN UTILISING RING SPRINGS

### 7.1 INTRODUCTION

The investigation to date has detailed the characteristics and dynamic behaviour of ring spring systems. Based on these characteristics, ring springs would appear suitable candidates for use in situations requiring additional flexibility as well as damping capability.

This chapter considers practical applications in which ring springs may be employed. One application identified as being particularly promising, is in utilising ring springs in seismic isolation systems. The benefits of utilising ring springs in these applications are discussed.

Further, an innovative isolation system that aims to reduce the level of dynamic loading experienced by a structure during an earthquake, has been developed. This system is detailed, followed by computer analyses undertaken to establish the efficacy of the design. A summary of the results of these analyses is presented.

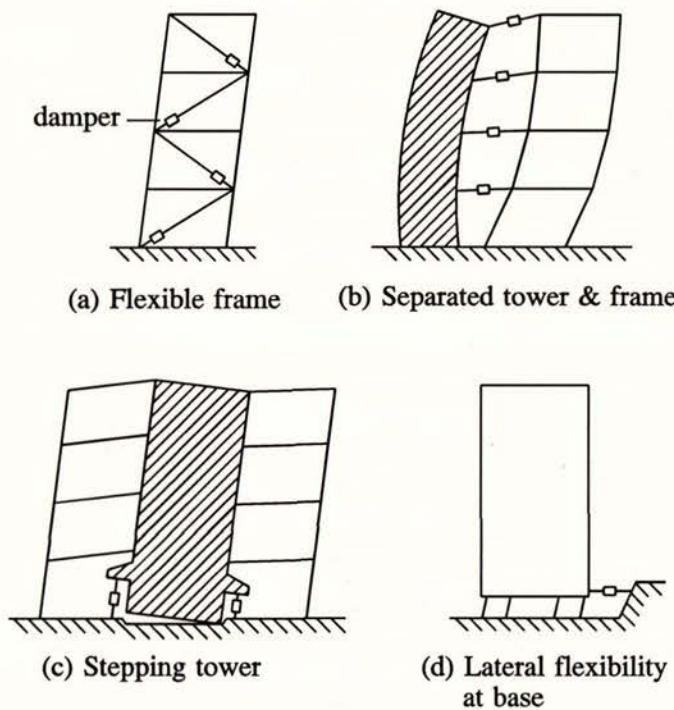
### 7.2 EARTHQUAKE-RESISTANT DESIGN

In the design of structures located in regions of seismic risk, it is imperative that the influence of earthquake generated forces be considered. In Chapter 2, the earthquake-resistant design techniques, along with numerous devices developed for use within earthquake-resistant structures, were reported. The following sections investigate the possibility of utilising ring springs in earthquake-resistant systems.

#### 7.2.1 Seismic isolation systems utilising ring springs

As reported in Chapter 2, ring springs have been suggested for use as seismic energy dissipators (Shepherd and Erasmus, 1988; Erasmus 1988). In addition, Hill (1995d)

suggested use of ring springs in the earthquake-resistant structures outlined by Skinner et al. (1975). These situations are shown in Figure 7.1.



**Figure 7.1** Structural situations in which hysteretic dampers may be used  
 (Skinner et al. 1975)

Of these situations, ring springs would appear to be a particularly attractive candidate for use in rocking/stepping seismic isolation systems as shown in Figure 7.1 (c). The function of the ring spring in these applications would be to act as a stiff member during moderate earthquakes (governed by the ring spring pre-displacement), while act as an energy absorber limiting the quasi-resonant build-up of structural deformations and forces during severe earthquakes.

For these rocking/stepping type applications, the ring spring cartridges fitted at the base of the structure may be pre-displaced, or both pre-displaced and pre-loaded. For the latter cases, the ring spring cartridges are pre-displaced and when fitted to the system may support the weight, or a portion of the weight, of the structure. It is in these applications that ring springs would appear ideally suited, since with increased height to the centre of gravity of the structure (with respect to the distance between the spring cartridges) the input seen at the ring springs is increased; this results in a more flexible system.

### **7.2.2 Potential benefits of ring springs**

The potential benefits of utilising ring springs in seismic isolation systems are (Hill 1995d):

- both the dissipative mechanism (friction) and system flexibility (spring) are incorporated within a single unit;
- ring springs remain fully elastic throughout the seismic disturbance - replacement after a major earthquake is not necessary thus providing protection against aftershocks;
- the elastic restoring force relocates the structure near to its original position;
- spring pre-tension may be adjusted such that sufficient rigidity exists to resist service loadings (such as wind and small earthquakes);
- minimal or no maintenance is required;
- a wide range of loading situations is possible as ring springs are commercially available in load carrying capacities ranging from 5 kN to 10 000 kN.

### **7.2.3 Practical applications for use of ring springs**

Columnar type structures are those primarily suited to rocking/stepping systems. Possible candidates include multi-storey buildings, chimneys, towers, as well as other similarly proportioned structures.

An application for which ring springs would appear to be suitable is in protecting high voltage electrical equipment during earthquakes. A number of features specific to high voltage electrical equipment make it necessary to provide these items with adequate protection from seismic damage. First, it is generally considered that electricity supply is essential after a major earthquake, therefore, electrical equipment should be designed to withstand a maximum design level earthquake. In addition, many items of electrical equipment possess fundamental periods in the range 0.125–0.5 seconds and thereby experience acceleration amplification during earthquakes with dominant periods in this range. Also, high voltage electrical equipment of necessity requires adequate electrical insulation from the ground and surrounding structures, thus, many items incorporate brittle insulating members which are highly susceptible to fracture and possess little inherent damping. Typical damping factors of 2% are common for high voltage electrical equipment

(Pham and Hoby 1991), however, structural damping as low as 0.8% of critical has been measured for some items (Rutledge et al. 1984).

The susceptibility of high voltage electrical equipment has been demonstrated by damage sustained during only moderately sized earthquakes. The San Fernando earthquake on 9 February 1971 of magnitude 6.6, caused damage totalling \$30 million to a new \$110 million electrical converter station; circuit breakers were reported as being especially vulnerable (Housner and Jennings 1972). In New Zealand, an earthquake 9 km north of the Tokaanu Power Station on 5 March 1984 of magnitude 5.6, caused damage to 5 out of 33 current transformers, and 10 out of 12 capacitor voltage transformers broke at their base (Rutledge et al. 1984). An earthquake near Edgecumbe in the North Island of New Zealand, on 2 March 1987 of magnitude 6.3, caused extensive damage to buswork, transformers and circuit breakers at the Edgecumbe and Kawerau Substations (Pender and Robertson 1987; Rutledge 1988).

Work undertaken in New Zealand since the late 1960's has considered methods of protecting items of high voltage electrical equipment during earthquakes (Hitchcock 1969, Hitchcock 1973). This has lead to the development and installation of several isolation systems (Pham 1988; Safi et al. 1989; Cousins et al. 1991; Coad and Pham 1993).

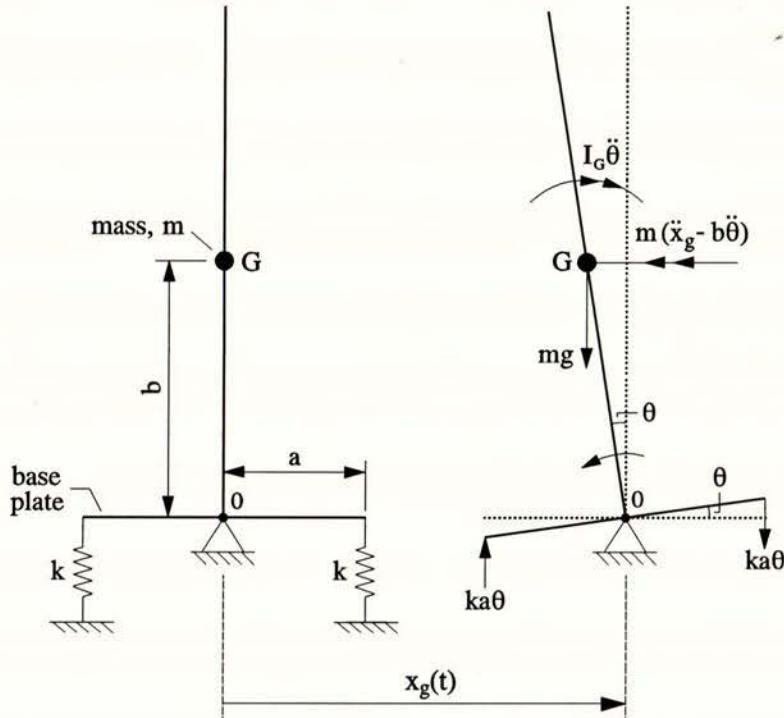
Recently, Pham reported on the use of Belleville washers in protecting such equipment (Pham and Hoby 1991). The system developed utilises Belleville washers fitted between the item of equipment and its support stand, thus increasing system flexibility and damping. Tests carried out on a Belleville washer damper gave a damping factor of 4.4%, with site tests conducted on three items of electrical equipment fitted with dampers giving an average increase in damping of 4.1% (Pham and Hoby 1991). Ring springs, with damping factors as high as 17.6% (Hill 1994b), could provide these items with increased seismic protection.

### 7.3 A PIVOTAL ROCKING SEISMIC ISOLATION SYSTEM INCORPORATING RING SPRINGS

This section details the development of a pivotal rocking seismic isolation system (PRSIS) incorporating ring springs for protecting columnar structures during earthquakes.

#### 7.3.1 Detail of system

The system comprises of the structure attached to a stiff base plate and supported on a pivot, with spring elements fitted around the periphery of the base and is shown in Figure 7.2.



**Figure 7.2** Pivotal rocking seismic isolation system subject to base excitation

The function of the pivot is to support the weight (or a portion of the weight) of the structure; carry the shear loading set up by the earthquake, and also to permit rotation of the base. However, for situations where it may not be practical to support the weight as shown in Figure 7.2, ground support is possible (Figure 7.1 (c)).

The function of the PRSIS is to introduce additional flexibility into the system. This lengthens the fundamental period of the system, thereby reducing spectral accelerations and

hence the inertia forces induced in the structure. Further, the addition of damping enables dissipation of the earthquake input energy and limits displacements.

### 7.3.2 Fundamental period of system

A simplified model of the PRSIS comprises of a structure of mass,  $m$ , with its centre of gravity located at height,  $b$ , and with a rocking arm of radius  $a$ , measured from the pivot to the position of the spring (Figure 7.2).

To determine the dynamic behaviour of this system, it is convenient to consider the springs as being of stiffness  $k$ . For non-linear devices stiffness  $k$  may be taken as the secant stiffness,  $k_{\text{eff}}$ . The forces acting on the isolation system are obtained using d'Alembert's principle (Craig 1981); this gives a reversed effective torque and reversed effective force as shown in Figure 7.2. The equation of motion for the rocking system for small  $\theta$ , i.e.  $\sin \theta \approx \theta$ , is:

$$I_0 \ddot{\theta} + (2ka^2 - mgb)\theta = m\ddot{x}_g b \quad (7.1)$$

$$\text{where } I_0 = I_G + mb^2 = \frac{4mb^2}{3}.$$

From Equation 7.1 the period of the system is defined as:

$$T_n = 2\pi \sqrt{\frac{I_0}{2ka^2 - mgb}} \quad (7.2)$$

For systems where  $mgb \ll 2ka^2$ , substitution of  $I_0$  into Equation 7.2 gives:

$$T_n = 2\pi \frac{b}{a} \sqrt{\frac{2m}{3k}} \quad (7.3)$$

Thus, for a particular system of mass,  $m$ , and springs of stiffness,  $k$ , the period of the system is defined by a linear relationship between the height to the centre of gravity and the rocking arm radius. This expression gives a lower bound for  $T_n$  since in any real

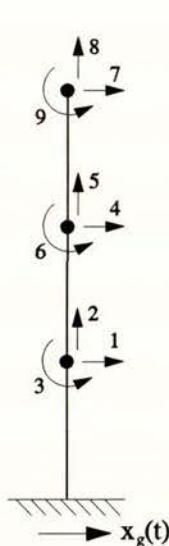
system the inclusion of column flexibility will increase the period of the system. It must be noted that the  $mgb$  term may not always be negligible; for systems with large  $b/a$  ratios, or very flexible springs, this term becomes increasingly important.

## 7.4 COMPUTER MODELLING OF PRSIS's

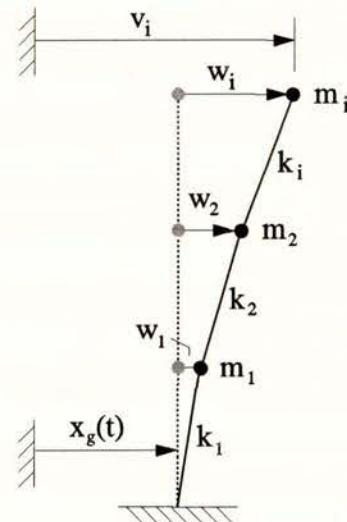
This section details a computer model for analysing multi-degree-of-freedom (MDOF) systems incorporating ring springs. A computer program for modelling these systems, as well as the earthquakes chosen for computer analyses, are detailed.

### 7.4.1 Generalised equation of motion

In studying the behaviour of structural systems subject to dynamic excitation, it is useful to employ a distributed parameter model whereby the system may be discretised into lumped masses assigned to specified nodes within the structure (Figure 7.3).



**Figure 7.3** Lumped mass model



**Figure 7.4** Structure displacements

The generalised equation of motion to determine the dynamic response of multi-degree-of-freedom systems may be expressed in matrix form as:

$$[M]\{\ddot{v}\} + [C]\{\dot{v}\} + [K]\{v\} = \{p(t)\} \quad (7.4)$$

where  $[M]$ ,  $[C]$  and  $[K]$  are the mass, damping and stiffness matrices,  $\{\ddot{v}\}$ ,  $\{\dot{v}\}$ , and  $\{v\}$  are the total accelerations, velocities, and displacements of the structure respectively, and

$\{p(t)\}$  is the time varying load acting on the structure. Expressing the total displacement  $\{v\}$  in relation to the relative displacement  $\{w\}$  and ground displacement  $x_g(t)$  (Figure 7.4), then:

$$\{v\} = \{w\} + \{r\} x_g(t) \quad (7.5)$$

where  $\{r\}$  contains the influence coefficients due to unit support displacement.

Substituting Equation 7.5 into Equation 7.4 and then simplifying the expression by considering ground motion to be uniform over the site (i.e. travelling waves are not considered)  $[K]\{r\} = \{0\}$ , and if the damping forces are considered to be due to differential velocities only,  $[C]\{r\} = \{0\}$ , the expression becomes:

$$[M]\{\ddot{w}\} + [C]\{\dot{w}\} + [K]\{w\} = \{p(t)\} - [M]\{r\}\ddot{x}_g(t) \quad (7.6)$$

where  $-[M]\{r\}\ddot{x}_g(t)$  may be considered as the effective earthquake load.

Equation 7.6 can be solved for linear elastic systems using the principle of mode superposition, however, for non-linear systems, step-by-step numerical integration techniques are required.

#### 7.4.2 PRSIS model

To determine the effectiveness of utilising ring springs in PRSIS's, computer simulation analyses were undertaken for a system consisting of a flexible column supported on ring springs isolators. This system is shown in Figure 7.5.

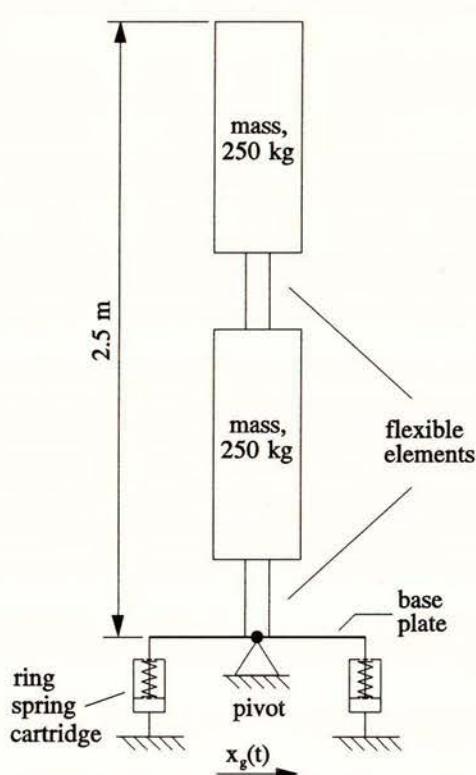


Figure 7.5 PRSIS

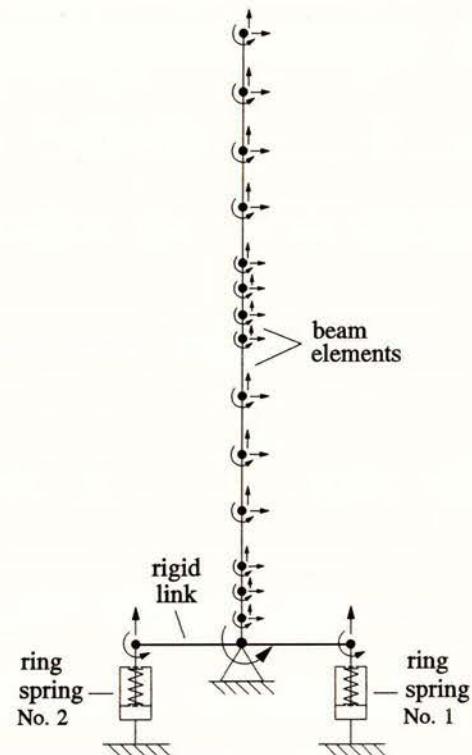


Figure 7.6 Finite element model

The columnar component in the system comprises two solid sections of mass 250 kg, connected by two flexible elements each of mass 5 kg, and a base plate of mass 20 kg. The height of each flexible element is 0.25 m and that of each solid mass is 1.0 m, giving a total column height of 2.5 m. The height and mass of the column were chosen to correspond with a typical small item of high voltage electrical equipment. The study was undertaken for three columns with fundamental periods of  $T_{\text{lunis}} = 0.125, 0.2$ , and  $0.5$  sec. (i.e. 8, 5 and 2 Hz fundamental frequencies). The variation in  $T_{\text{lunis}}$  was obtained by adjusting the diameter of the flexible elements, thus:

$\varnothing 93.5$  mm gives  $T_{\text{lunis}} = 0.125$  sec.,

$\varnothing 73.1$  mm gives  $T_{\text{lunis}} = 0.2$  sec., and

$\varnothing 45.9$  mm gives  $T_{\text{lunis}} = 0.5$  sec.

For columnar structural systems, it is common that the structure would comprise of a number of sections with flanged connections. The use of flexible elements within the model, as described above, is designed to give the structure additional flexibility in these regions, however, it is not a representation of the rotational stiffness set up by flanged configurations. Analysis of such connections requires specific consideration.

The computer model of the isolation system is presented in Figure 7.6. The model comprises beam elements with lumped nodal masses and rigid link elements modelling the base plate. The ring spring elements represent bi-directional ring spring cartridges. These elements are defined by the ring spring force/deflection characteristics. The stiffness values chosen for the ring spring elements are as follows:

$$K_o = 2.5 \times 10^6 \text{ N/m}$$

$$K_i = 385 \times 10^3 \text{ N/m}$$

$$K_l = 33 \times 10^6 \text{ N/m}$$

$$K_d = 142 \times 10^3 \text{ N/m}$$

These values have been obtained from tests carried out on a prototype ring spring cartridge reported in Section 4.5.2. Though these stiffness values apply to the ring spring chosen, the same stiffnesses may be obtained for ring springs of different sizes simply by increasing or decreasing the number of ring elements in a particular stack. The fundamental period of the rocking system,  $T_{lis}$ , based upon stiffness  $K_i$  for each of the three columns is as follows: for the column of  $T_{lunis} = 0.125$  sec., the fundamental period of the isolated system is  $T_{lis} = 1.006$  sec; for the column of  $T_{lunis} = 0.2$  sec., the fundamental isolated period is  $T_{lis} = 1.018$  sec; and for the column of period  $T_{lunis} = 0.5$  sec., the fundamental isolated period is  $T_{lis} = 1.116$  sec. These fundamental periods have been determined using a computer program capable of modelling MDOF systems incorporating ring springs; this program is now discussed.

#### 7.4.3 RUAUMOKO computer program

The ring spring model, as reported in section 5.2.2, was incorporated within a commercially available time history analysis software package - RUAUMOKO. This program was initially written by Sharpe (1974), and has been significantly extended by Carr (1995). RUAUMOKO uses the unconditionally stable Newmark constant average acceleration (Newmark  $\beta = 0.25$ ) method to compute the time history response of non-linear two-dimensional general frame structures subject to dynamic excitation. The ring spring algorithm incorporated within RUAUMOKO is presented in Appendix A.

#### 7.4.4 Earthquake excitations used in analyses

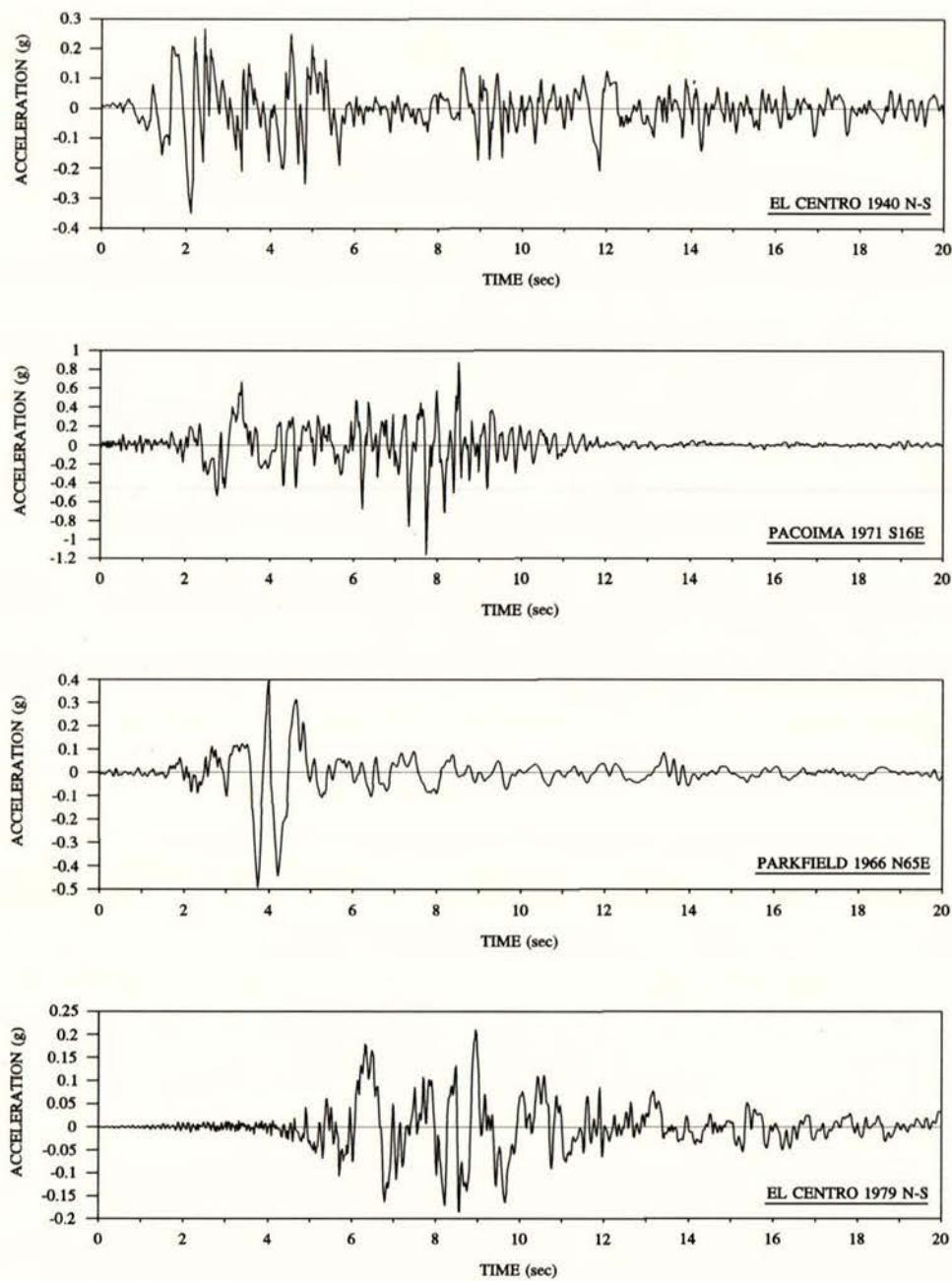
The PRSIS was subjected to six earthquakes: four short period and two long period. These are detailed below.

##### (a) Short period type earthquakes

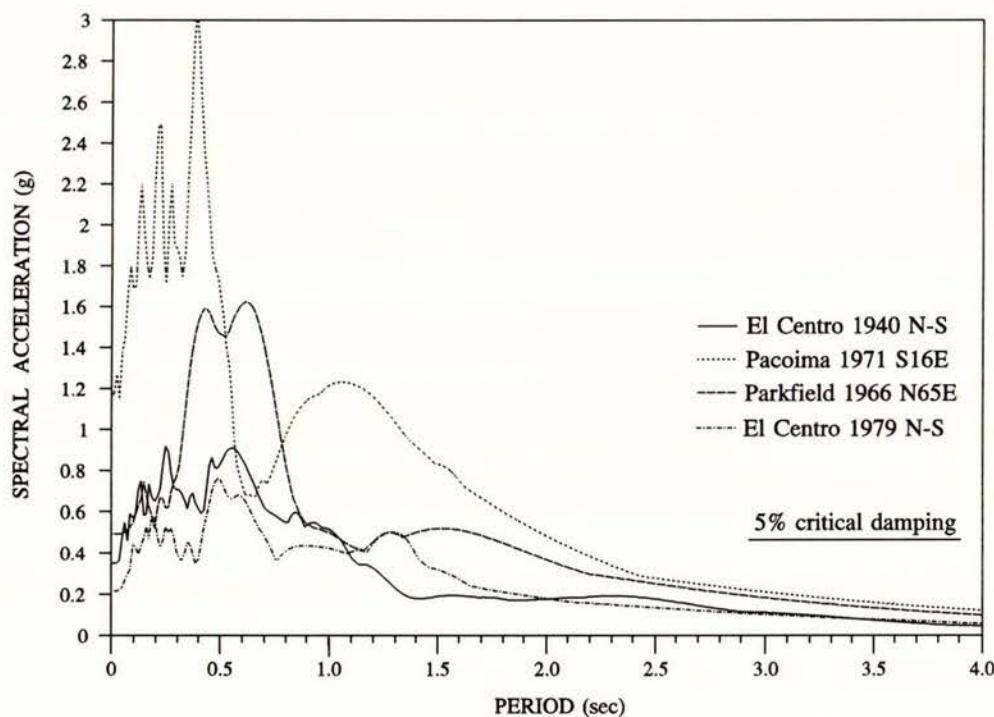
Short period type earthquakes are classed as those earthquakes whose dominant spectral accelerations occur at periods less than about 0.8–1.0 second. Four such earthquakes were chosen for the study. These are:

- El Centro 1940 N-S (Trifunac and Brune 1970)
- Pacoima Dam 1971 S16E (Oakeshott 1975)
- Parkfield 1966 N65E (Housner and Trifunac 1967)
- El Centro 1979 N-S (Brandow and Leeds 1980)

Detailed discussion on the specific nature of these earthquakes may be found in the above references; accelerograms of the earthquakes, as well as their response spectra, are presented in Figures 7.7 and 7.8.



**Figure 7.7** Accelerograms of four short period type earthquakes



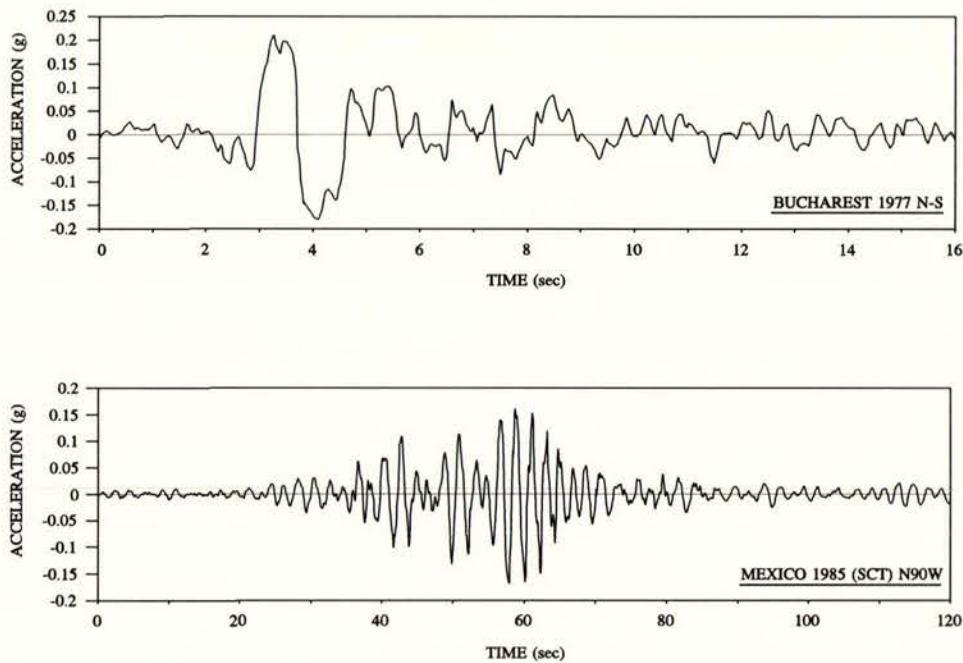
**Figure 7.8** Response spectra of four short period type earthquakes

(b) Long period type earthquakes

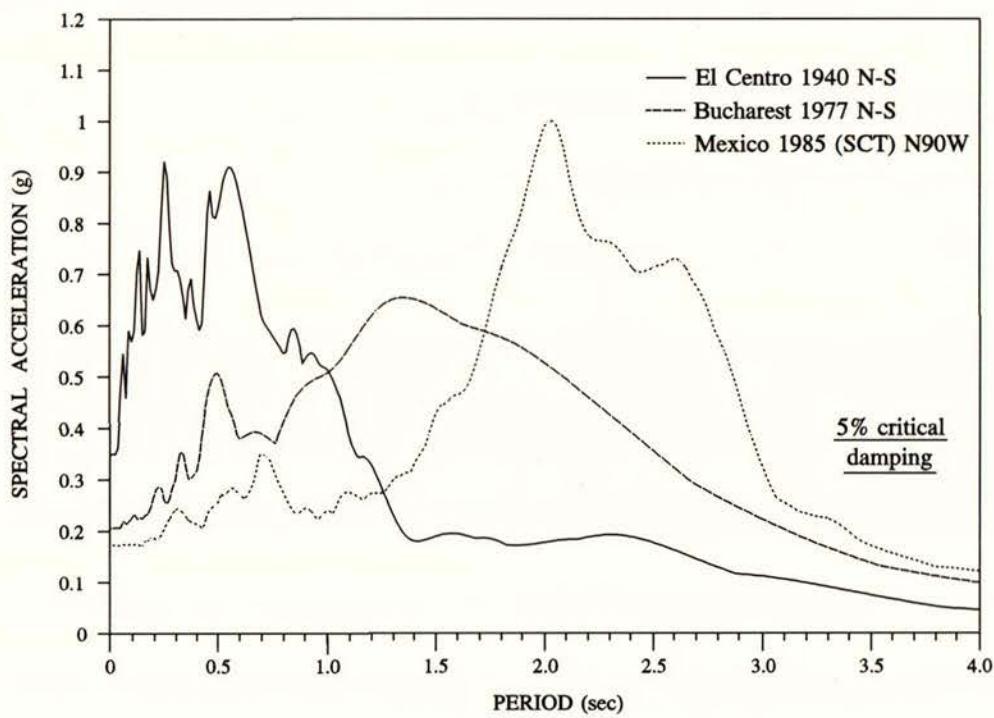
Long period type earthquakes are classed as those earthquakes whose dominant spectral accelerations occur at periods greater than about 1.0 second. Two such earthquakes chosen for this study are:

- Bucharest 1977 N-S (Hartzell 1979)
- Mexico 1985 (SCT) N90W (Seed et al. 1988)

Accelerograms of these earthquakes, as well as their response spectra, are presented in Figures 7.9 and 7.10.



**Figure 7.9** Accelerograms of two long period type earthquakes



**Figure 7.10** Response spectra of two long period type earthquakes

These earthquakes were chosen to cover a representative range of both short and long period type earthquakes. However, the use of seismic isolation systems is generally more

suited to structures subjected to short period type earthquakes. This is because the addition of flexibility moves the period of the system away from the vicinity of dominant spectral accelerations. However, for structures exposed to long period type earthquakes, inclusion of additional flexibility, could lead to a period shift into a region of increased spectral accelerations. For these systems, an earthquake-resistant design that incorporates devices that do not lengthen the period of the structure, but provide increased amounts of damping, would be desirable. Such devices would include viscous dampers.

The likely characteristics of an earthquake at a particular site are important since this determines which aseismic design method is appropriate. Several earthquakes recorded in New Zealand recently, have shown response spectra characteristics similar to those of short period type earthquakes (Sritharan and Dowrick 1994). These sites may be appropriate for use of seismic isolation systems.

However, for completeness of the computer simulation analyses, both short and long period type earthquakes are considered.

## **7.5 COMPUTER SIMULATION RESULTS FOR ISOLATED (PRSIS) AND UNISOLATED SYSTEMS**

This section details results of computer simulation analyses (using RUAUMOKO) for isolated (PRSIS) and unisolated systems consisting of a column of  $T_{\text{unis}} = 0.2$  seconds and subjected to the El Centro 1940 N-S earthquake. The following cases are analysed:

- unisolated (fixed base)
- PRSIS with 0% pre-displacement and 0% pre-load
- PRSIS with 16% pre-displacement and 0% pre-load
- PRSIS with 16% pre-displacement and 10% pre-load
- PRSIS with 16% pre-displacement and 20% pre-load

To compare the magnitude of loading experienced by each of these systems, the structure bending moment for each case was determined. Bending moments were used as the structure shear forces contribute only a few percent toward the total structural load.

The graphical results presented in the following sub-sections give the bending moment time histories measured at the base of the structure; for all cases studied the maximum bending

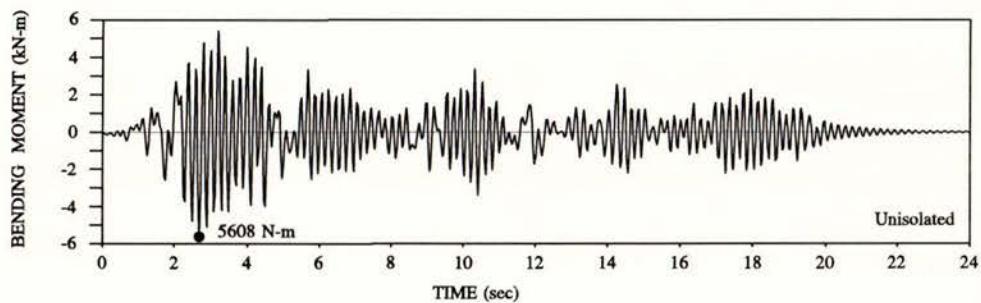
moment occurred at this point.

The RUAUMOKO results for all cases considered were obtained using a consistent mass matrix, constant damping of 2% applied to all modes, and a time step of 0.0001 sec. This time step was shown to give a converged solution for the analyses considered. Also, 2% damping was chosen to be representative of lightly damped systems. A small displacement analysis was used and to avoid the high frequencies of the structure in the y direction, the y degrees of freedom of the column were constrained.

The results of the computational analyses are presented graphically in Sections 7.5.1, 7.5.2, and 7.5.3. These are then discussed in Section 7.5.4.

### 7.5.1 Dynamic response of unisolated systems

The unisolated system was modelled by constraining the x, y and  $\theta$  degrees-of-freedom at the base of the column (Figure 7.6). The bending moment time history for the unisolated system subjected to the El Centro 1940 N-S earthquake is presented in Figure 7.11.



**Figure 7.11** Dynamic response of unisolated system: bending moment time history

### 7.5.2 Dynamic response of pre-displaced PRSIS's

#### (a) 0% pre-displacement

The graphical results presented for the 0% pre-displaced 0% pre-loaded PRSIS subjected to the El Centro 1940 N-S earthquake are:

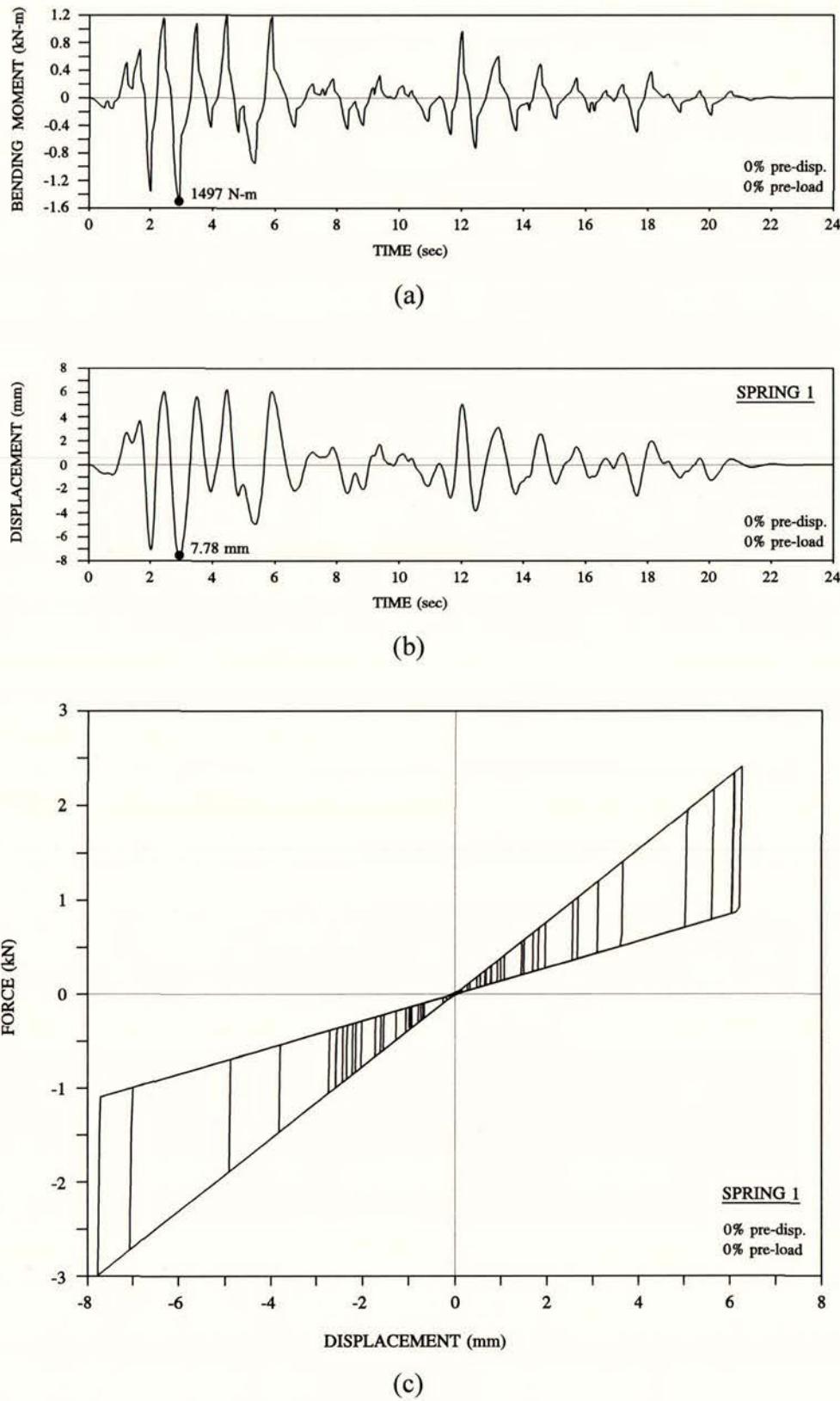
- bending moment time history - measured at the base node (Figure 7.12 (a))
- ring spring cartridge displacement for spring No. 1 (Figure 7.12 (b))
- force/deflection response curve for spring No. 1 (Figure 7.12 (c))

Ring spring cartridge No. 1 is shown in Figure 7.6.

#### (b) 16% pre-displacement

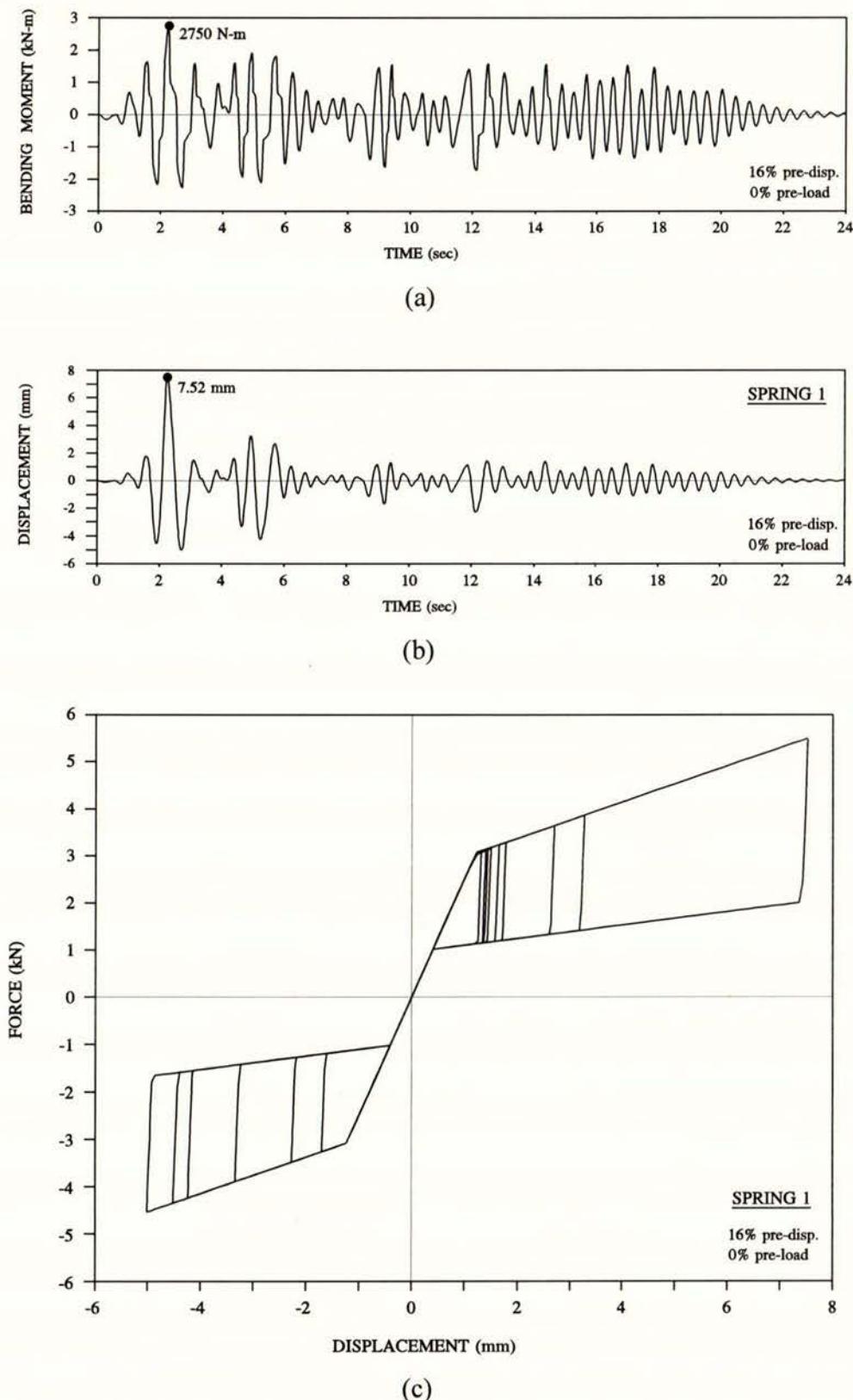
The graphical results presented for the 16% pre-displaced 0% pre-loaded PRSIS subjected to the El Centro 1940 N-S earthquake are:

- bending moment time history - measured at the base node (Figure 7.13 (a))
- ring spring cartridge displacement for spring No. 1 (Figure 7.13 (b))
- force/deflection response curve for spring No. 1 (Figure 7.13 (c))



**Figure 7.12** Dynamic response of PRSIS (0% pre-displacement, 0% pre-load):

- (a) bending moment time history
- (b) ring spring displacement time history
- (c) ring spring force/deflection response curve



**Figure 7.13** Dynamic response of PRSIS (16% pre-displacement, 0% pre-load):

- (a) bending moment time history
- (b) ring spring displacement time history
- (c) ring spring force/deflection response curve

### 7.5.3 Dynamic response of pre-displaced pre-loaded PRSIS's

#### (a) 16% pre-displacement 10% pre-load

The graphical results presented for the 16% pre-displaced 10% pre-loaded PRSIS subjected to the El Centro 1940 N-S earthquake are:

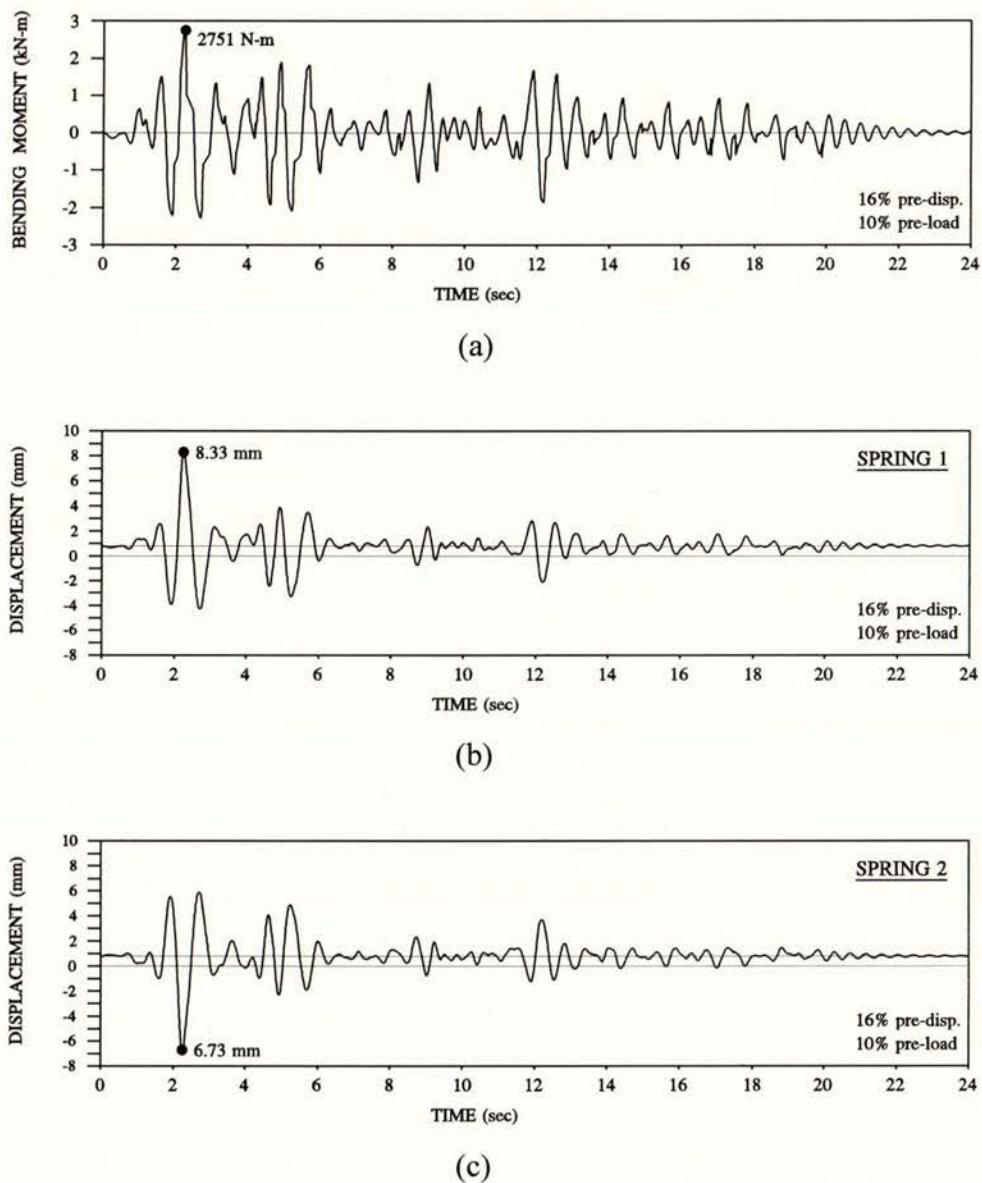
- bending moment time history - measured at the base node (Figure 7.14 (a))
- ring spring cartridge displacement for spring No. 1 (Figure 7.14 (b))
- ring spring cartridge displacement for spring No. 2 (Figure 7.14 (c))
- force/deflection response curve for spring No. 1 (Figure 7.15 (a))
- force/deflection response curve for spring No. 2 (Figure 7.15 (b))

Ring spring cartridges No. 1 and 2 are shown in Figure 7.6.

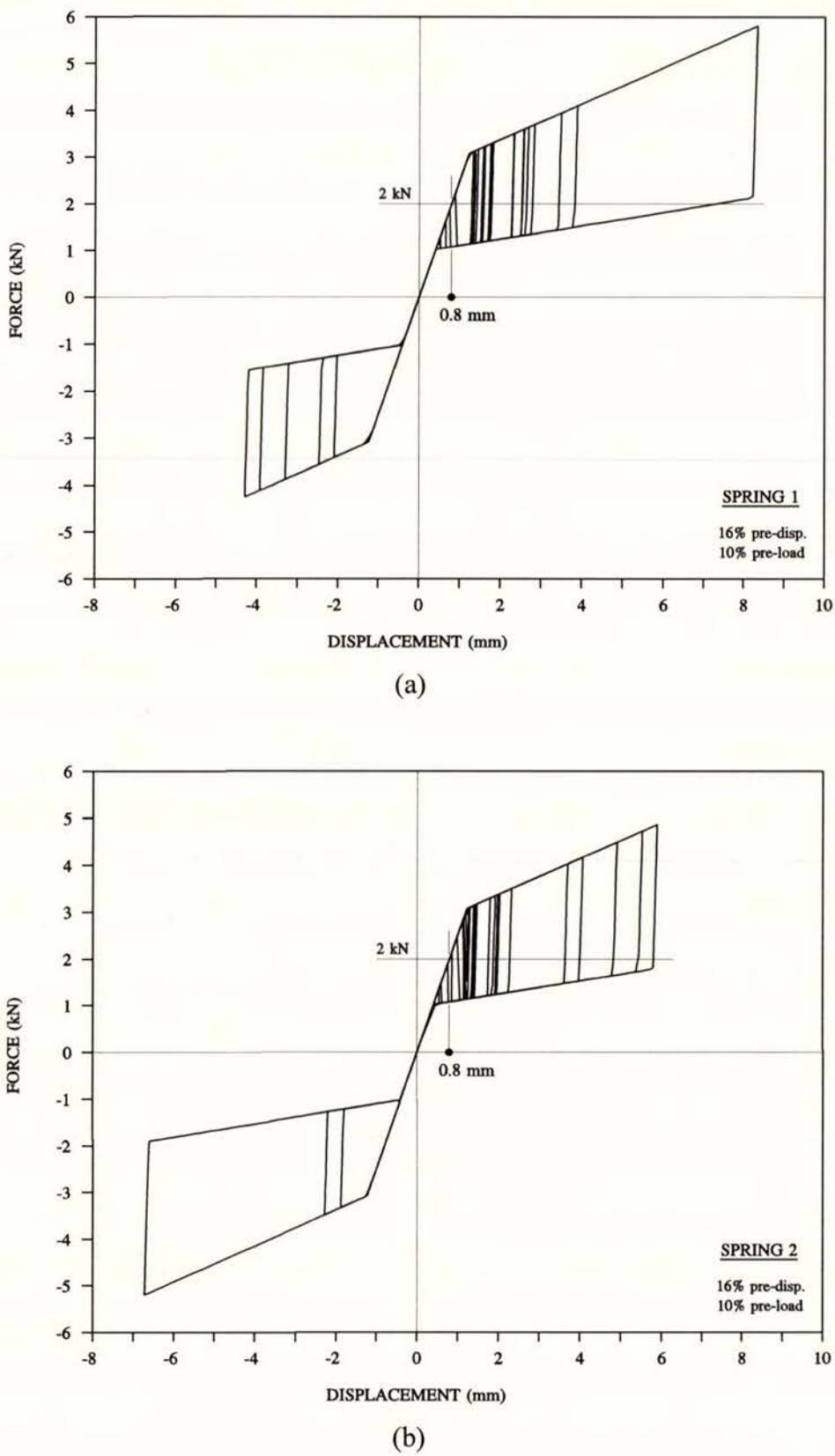
#### (b) 16% pre-displacement 20% pre-load

The graphical results presented for the 16% pre-displaced 20% pre-loaded PRSIS subjected to the El Centro 1940 N-S earthquake are:

- bending moment time history - measured at the base node (Figure 7.16 (a))
- ring spring cartridge displacement for spring No. 1 (Figure 7.16 (b))
- ring spring cartridge displacement for spring No. 2 (Figure 7.16 (c))
- force/deflection response curve for spring No. 1 (Figure 7.17 (a))
- force/deflection response curve for spring No. 2 (Figure 7.17 (b))



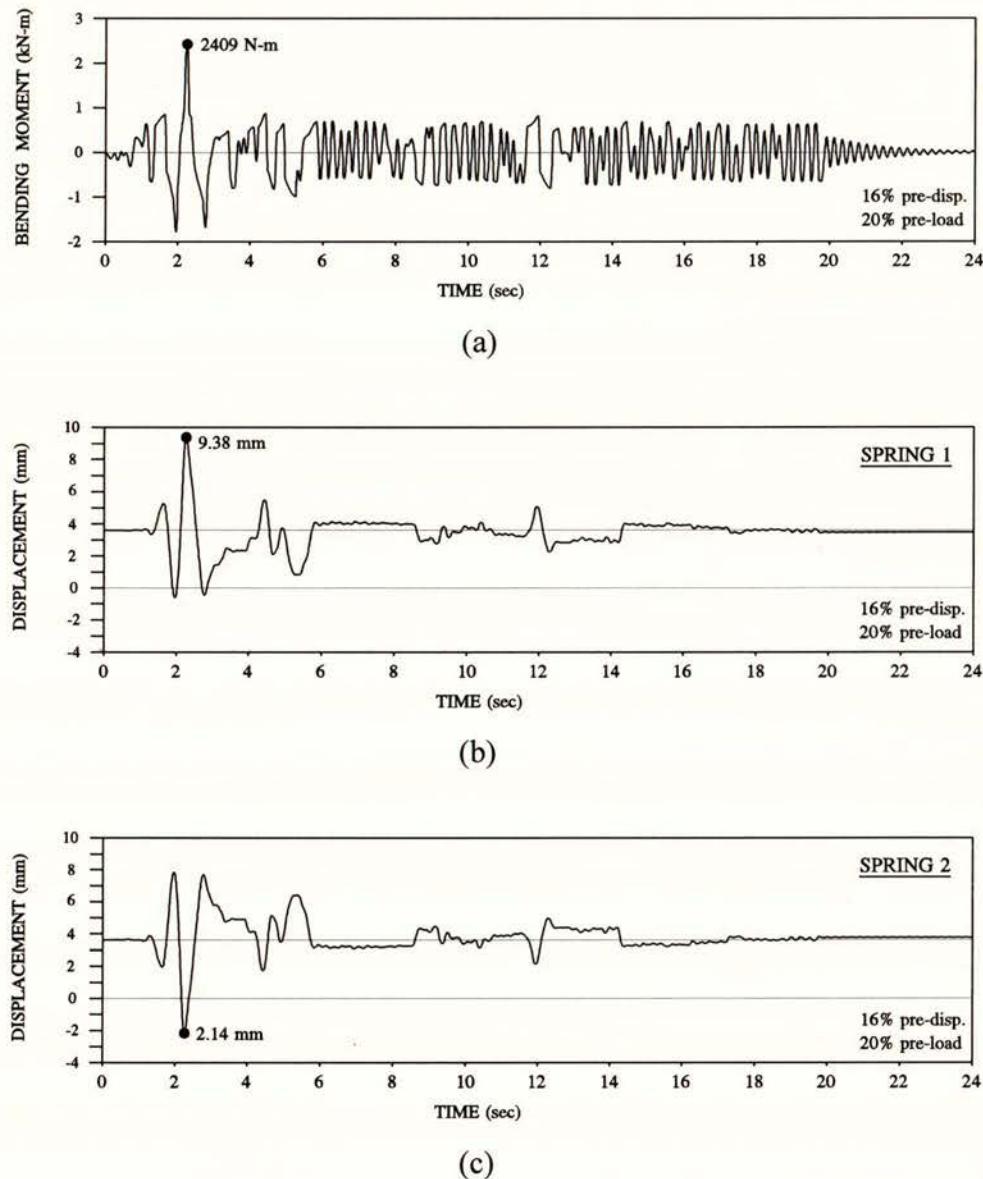
**Figure 7.14** Dynamic response of PRSIS (16% pre-displacement, 10% pre-load):  
 (a) bending moment time history  
 (b) displacement time history: ring spring No. 1  
 (c) displacement time history: ring spring No. 2



**Figure 7.15** Dynamic response of PRSIS (16% pre-displacement, 10% pre-load):

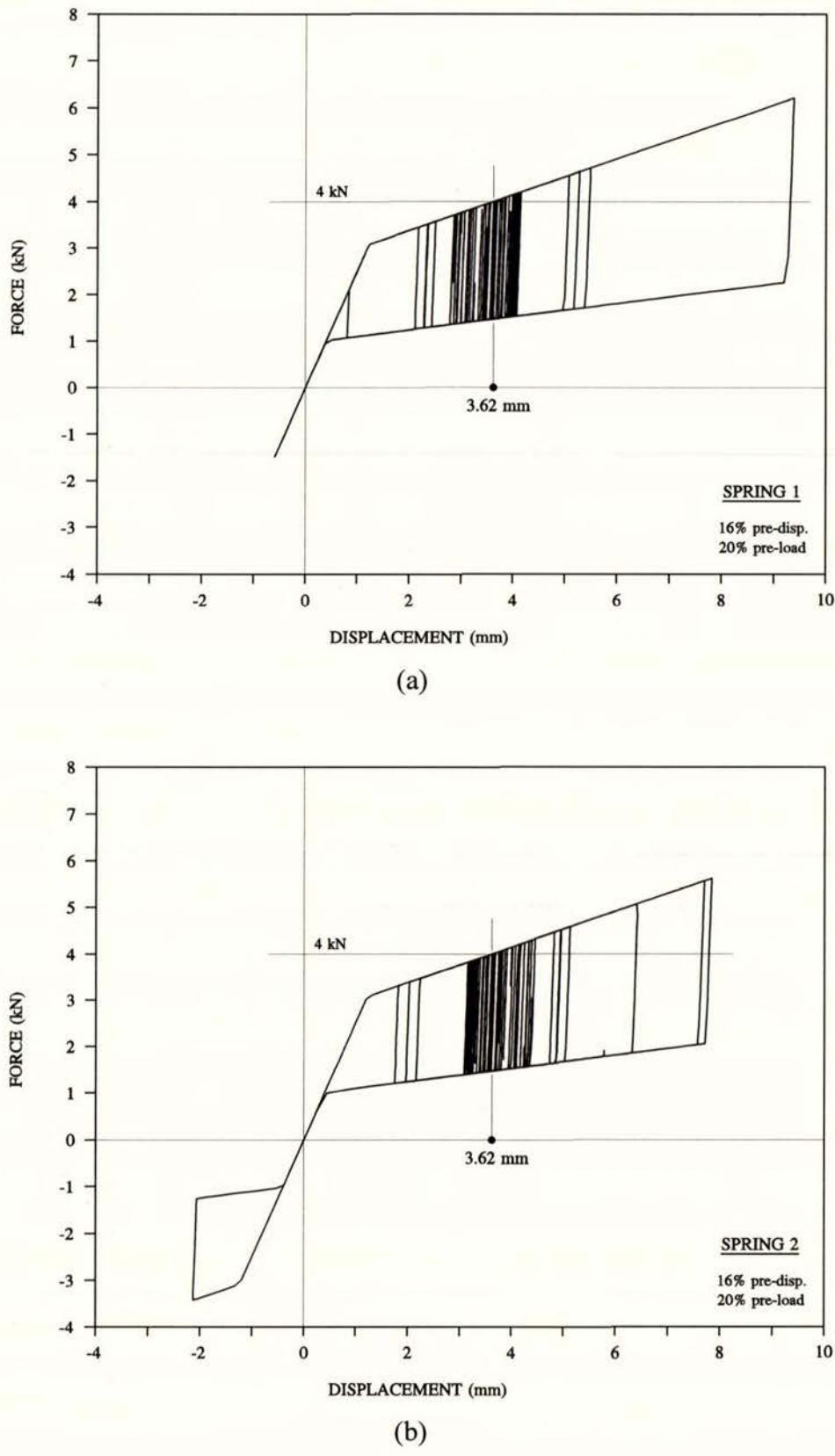
(a) force/deflection response curve: ring spring No. 1

(b) force/deflection response curve: ring spring No. 2



**Figure 7.16** Dynamic response of PRSIS (16% pre-displacement, 20% pre-load):

- (a) bending moment time history
- (b) displacement time history: ring spring No. 1
- (c) displacement time history: ring spring No. 2



**Figure 7.17** Dynamic response of PRSIS (16% pre-displacement, 20% pre-load):

- force/deflection response curve: ring spring No. 1
- force/deflection response curve: ring spring No. 2

#### 7.5.4 Discussion of results

The results presented show that the PRSIS is effective in reducing maximum structure loads for the El Centro 1940 N-S earthquake. For the unisolated system a maximum bending moment of 5608 N-m is given. For the PRSIS's the following maximum bending moments are obtained: 1497 N-m, 2750 N-m, 2751 N-m, and 2409 N-m for the 0% pre-displacement 0% pre-load, 16% pre-displacement 0% pre-load, 16% pre-displacement 10% pre-load, and 16% pre-displacement 20% pre-load cases respectively. Thus, bending moments for the PRSIS are at least half of those given for the unisolated system.

The ring spring displacement time histories and force/deflection response curves for the cases of 0% pre-displacement 0% pre-load and 16% pre-displacement 0% pre-load (Figures 7.12 (b) and (c), and 7.13 (b) and (c)), are given for spring No. 1 only. However, as the PRSIS is symmetric (Figure 7.6), the results for spring No. 2 will be equal and opposite.

For the case of 16% pre-displacement 10% pre-load, the initial pre-load of 2 kN equates to a compression of the spring of 0.8 mm. This results in the displacement time histories oscillating about a line at 0.8 mm (Figures 7.14 (b) and (c)). The displacements about 0.8 mm for springs No. 1 and 2 are equal and opposite. However, since the equilibrium line on the force/deflection response curve is shifted to 2 kN, the forces in the springs during the time history are not equal and opposite (Figures 7.15 (a) and (b)).

For the case of 16% pre-displacement 20% pre-load, the initial pre-load of 4 kN equates to a compression of the spring of 3.62 mm. Figures 7.16 (b) and (c) show equal and opposite displacements oscillating about the 3.62 mm displacement. It is seen that the response reduces to a value close to the original displacement of 3.62 mm; this offset results from the structure settling near to its initial position and oscillating on stiffness  $K_1$ . Also, the bending moment time history (Figure 7.16 (a)) shows periods of reasonably constant amplitude oscillations, for example between 6 and 8 seconds. During this period, displacements are seen to be close to 3.62 mm and remain relatively constant (Figures 7.16 (b) and (c)). This can be seen on the force/deflection response curves by the darker patches either side of the 3.62 mm displacement (Figures 7.17 (a) and (b)). During this phase of motion the response is oscillating on stiffness  $K_1$ .

## 7.6 MAXIMUM STRUCTURE BENDING MOMENTS FOR ISOLATED AND UNISOLATED SYSTEMS

The above analyses have been repeated to consider a comprehensive range of isolated (PRSIS) and unisolated systems subjected to the six different earthquake excitations.

The systems analysed were for the flexible column with unisolated fundamental periods of:

$$T_{1\text{unis}} = 0.125 \text{ seconds}$$

$$T_{1\text{unis}} = 0.2 \text{ seconds}$$

$$T_{1\text{unis}} = 0.5 \text{ seconds}$$

and used in the following system configurations:

- unisolated (fixed base)
- PRSIS with 0% pre-displacement and 0% pre-load
- PRSIS with 16% pre-displacement and 0% pre-load
- PRSIS with 16% pre-displacement and 10% pre-load
- PRSIS with 16% pre-displacement and 20% pre-load

Tabular results for all cases studied are presented in Appendix C. These tables give the maximum structure bending moment for the isolated and unisolated systems for each earthquake input. The case for  $T_{1\text{unis}} = 0.2 \text{ seconds}$ , and for 0% pre-displacement 0% pre-load, is presented in Table 7.1.

**Table 7.1** Maximum structure bending moments for isolated and unisolated systems for  $T_{1\text{unis}} = 0.2$  sec (isolation system comprises 0% pre-displacement and 0% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1\text{unis}} = 0.200$ s $T_{2\text{unis}} = 0.031$ s	Isolated Structure $T_{1\text{is}} = 1.018$ s $T_{2\text{is}} = 0.034$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	5608 N-m @ 2.69 s	1497 N-m @ 2.94 s	<b>26.7 %</b>
Pacoima 1971 S16E	19280 N-m @ 9.31 s	5607 N-m @ 3.55 s	<b>29.1 %</b>
Parkfield 1966 N65E	3794 N-m @ 3.80 s	3020 N-m @ 4.97 s	<b>79.6 %</b>
El Centro 1979 N-S	5587 N-m @ 8.93 s	1812 N-m @ 6.63 s	<b>32.4 %</b>
Bucharest 1977 N-S	2044 N-m @ 3.99 s	3026 N-m @ 4.28 s	<b>148.0 %</b>
Mexico 1985 (SCT) N90W	1328 N-m @ 58.83 s	2359 N-m @ 59.06 s	<b>177.6 %</b>

The right hand column of Table 7.1 gives the percentage ratio of the maximum bending moment for the isolated system to that of the unisolated system. This enables the effectiveness of the PRSIS to be assessed. These results show that for the El Centro 1940 N-S, Pacoima Dam 1971 S16E, and the El Centro 1979 N-S earthquakes, structure loads are significantly reduced; for these cases, by a factor greater than three.

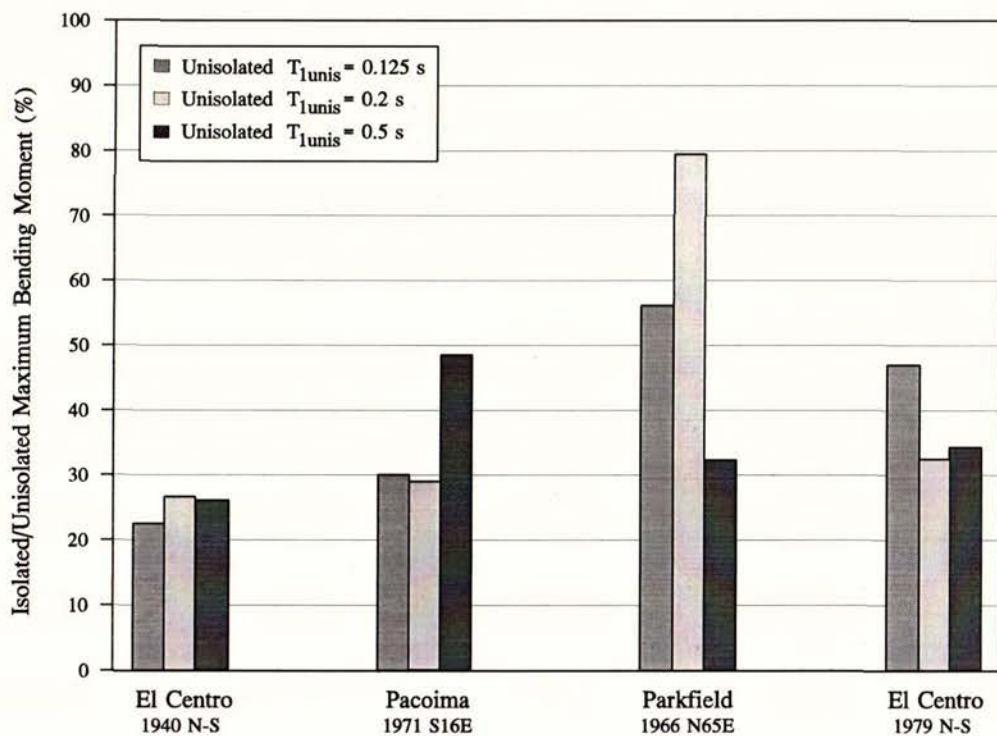
Results also show that for the Parkfield 1966 N65E input, a reduced, but not insignificant, level of isolation is obtained. This can be explained by considering the response spectra characteristics for this particular earthquake. As is shown in Figure 7.8, dominant spectral accelerations occur over a narrow range of periods between 0.35 and 0.75 seconds. Outside this range, spectral accelerations are significantly reduced. Thus, for the column of period  $T_{1\text{unis}} = 0.2$  seconds, use of the PRSIS shows reduced effectiveness.

Results given for the Bucharest 1977 N-S and Mexico 1985 (SCT) N90W earthquakes show increased column bending moments for the isolated system. This is due to a shift in the period of unisolated structure into a region of increased spectral accelerations (Figure 7.10). For earthquakes with peak spectral accelerations at longer periods, such as the Bucharest 1977 N-S and Mexico 1985 (SCT) N90W records (where peak spectral accelerations occur at about 1.3 sec. and 2 sec. respectively), an earthquake-resistant design

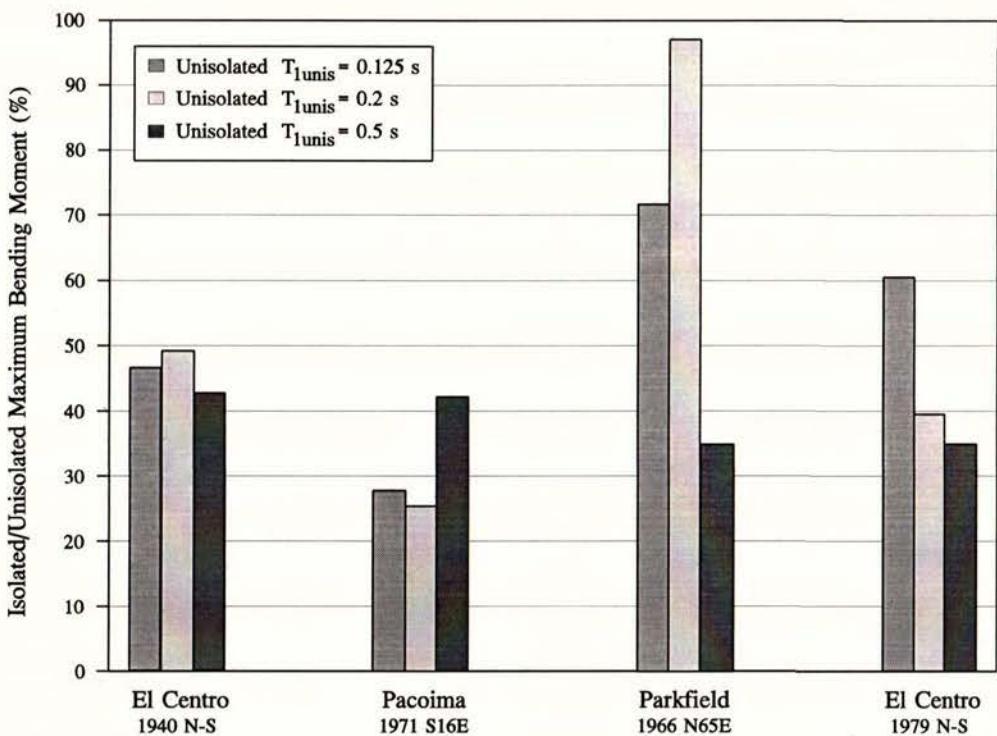
should incorporate devices such as viscous dampers to provide increased amounts of damping.

The tabular results presented in Appendix C have been summarised in charts for the four short period type earthquakes. These are presented in Figures 7.18, 7.19, 7.20, and 7.21. In these figures the y axis gives the percentage ratio of the maximum bending moment for the isolated system to that of the unisolated system; the x axis gives the three unisolated cases of  $T_{\text{unis}} = 0.125$  seconds,  $T_{\text{unis}} = 0.2$  seconds, and  $T_{\text{unis}} = 0.5$  seconds for each of the four earthquakes. Four charts are given for:

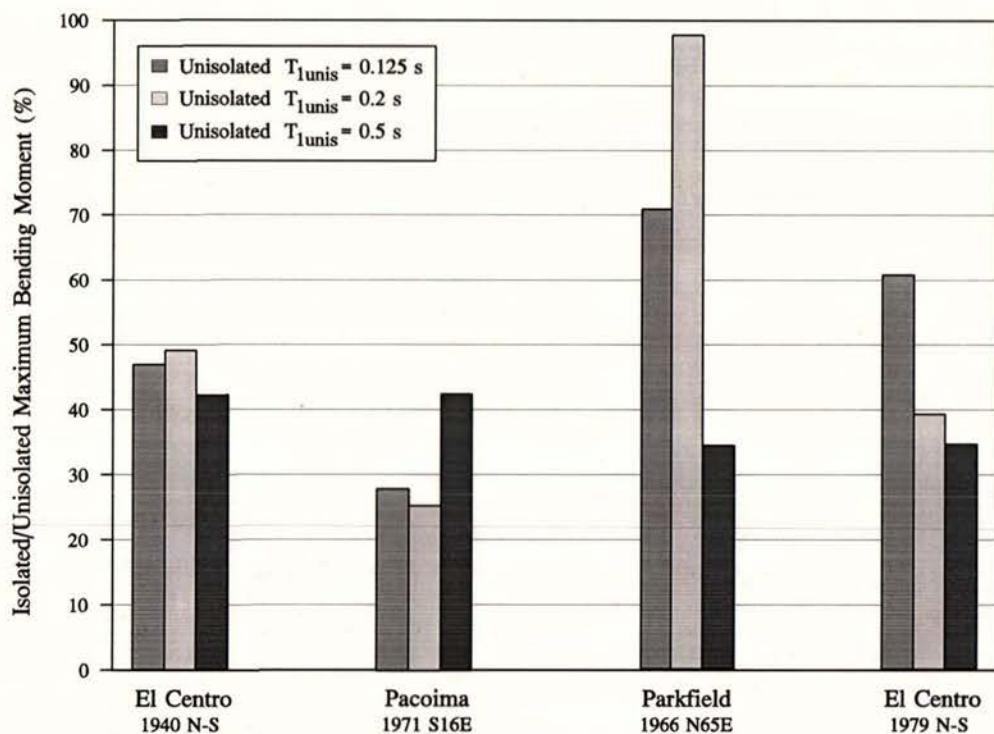
- PRSIS with 0% pre-displacement and 0% pre-load (Figure 7.18)
- PRSIS with 16% pre-displacement and 0% pre-load (Figure 7.19)
- PRSIS with 16% pre-displacement and 10% pre-load (Figure 7.20)
- PRSIS with 16% pre-displacement and 20% pre-load (Figure 7.21)



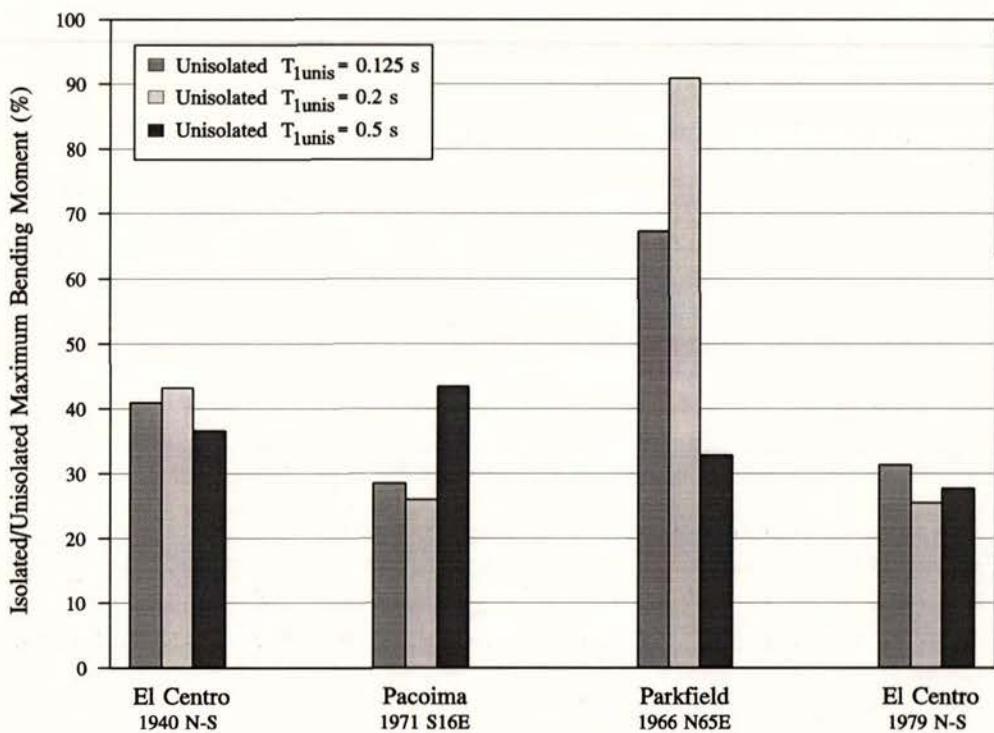
**Figure 7.18** Percentage relationship of maximum structure bending moment for isolated and unisolated systems (0% pre-displacement and 0% pre-load)



**Figure 7.19** Percentage relationship of maximum structure bending moment for isolated and unisolated systems (16% pre-displacement and 0% pre-load)



**Figure 7.20** Percentage relationship of maximum structure bending moment for isolated and unisolated systems (16% pre-displacement and 10% pre-load)



**Figure 7.21** Percentage relationship of maximum structure bending moment for isolated and unisolated systems (16% pre-displacement and 20% pre-load)

### 7.6.1 Discussion of summarised results

The results show that structural loads experienced during short period type earthquakes, can be significantly reduced utilising pivotal rocking seismic isolation systems. Structure bending moments were reduced by factors of approximately 2–3 or greater in some cases. The exceptions were the isolation systems incorporating flexible columns of period  $T_{1\text{unis}} = 0.125$  and 0.2 seconds, subjected to the Parkfield 1966 N65E earthquake.

Results were obtained for an isolation system fundamental period ( $T_{1\text{is}}$ ) of approximately 1.0–1.1 seconds, suggesting that these periods are sufficient to ensure effective isolation. The maximum rotation of the isolation system base plate occurred for the system of period  $T_{1\text{is}} = 1.116$  sec (0% pre-displacement and 0% pre-load) when subjected to the Pacoima Dam 1971 S16E earthquake. A maximum base rotation of 7.8° was obtained.

Generally, ring spring pre-displacement and pre-load did not have a significant effect on the results. However, in practical systems it would be desirable to keep both the pre-displacement and pre-load small, thus allow a larger displacement range for a given ring spring cartridge.

In addition to the computer simulation results presented, the effect of using a large displacement analysis was studied. This enabled changes in the geometry of the system, as well as gravity effects of the column self weight, to be considered. Two pre-displaced pre-loaded systems subjected to the El Centro 1940 N-S earthquake were analysed. Results for these systems are as follows:

- 16% pre-displacement 0% pre-load ( $T_{1\text{unis}} = 0.2$  sec):
  - bending moment (max.) = 2708 N-m @ 2.23 sec (large displacement analysis)
  - bending moment (max.) = 2750 N-m @ 2.24 sec (small displacement analysis)
- 16% pre-displacement 20% pre-load ( $T_{1\text{unis}} = 0.2$  sec):
  - bending moment (max.) = 2419 N-m @ 2.24 sec (large displacement analysis)
  - bending moment (max.) = 2409 N-m @ 2.25 sec (small displacement analysis)

These results show differences of less than 2% for large and small displacement analyses.

Further, the influence of simultaneously applying both the horizontal and vertical components of an earthquake to a PRSIS was studied. For PRSIS's, vertical inputs are fed directly through the pivot into the column. Hence to determine the effect on structure bending moment, two pre-displaced pre-loaded systems, subjected to the El Centro 1940 N-S and El Centro 1940 U-D (vertical) earthquakes simultaneously, were analysed. Results for these systems are as follows:

- 16% pre-displacement 0% pre-load ( $T_{1\text{unis}} = 0.2 \text{ sec}$ ):  
bending moment (max.) = 2708 N-m @ 2.23 sec (large displacement analysis)
- 16% pre-displacement 20% pre-load ( $T_{1\text{unis}} = 0.2 \text{ sec}$ ):  
bending moment (max.) = 2420 N-m @ 2.24 sec (large displacement analysis)

These results are almost identical to those above for the El Centro 1940 N-S earthquake only. However, the effect on column bending moment would be more noticeable in cases where a structure is deformed near to its maximum and is then subject to a large vertical input.

## 7.7 SUMMARY

Ring springs have been identified as possible candidates for use in rocking/stepping seismic isolation systems. The potential benefits of utilising ring springs in these applications have been outlined.

Subsequently, a pivotal rocking seismic isolation system (PRSIS) incorporating ring springs was developed. To determine the effectiveness of utilising ring springs in these systems, computer simulation analyses were undertaken. The study investigated the behaviour of isolated and unisolated systems subject to six different earthquake inputs.

Results show that for columnar structures with fundamental periods between 0.125–0.5 seconds, PRSIS's can significantly reduce structural loads during short period type earthquakes. To confirm the performance of these isolation systems, experimental testing of a PRSIS subjected to a range of earthquake excitations, is required.

## **DEVELOPMENT OF A PROTOTYPE PIVOTAL ROCKING SEISMIC ISOLATION SYSTEM: EXPERIMENTAL TESTING AND COMPUTER SIMULATION**

### **8.1 INTRODUCTION**

Results of computer simulation analyses, as reported in Chapter 7, show that pivotal rocking seismic isolation systems can significantly reduce structural loads during short period type earthquakes. This chapter details the design and testing of a manufactured PRSIS. Initially, design details of the PRSIS are presented, followed by results of shaker table tests. These test results, for both isolated and unisolated systems, are then compared with results given by computer simulation.

### **8.2 DESIGN OF A PROTOTYPE PIVOTAL ROCKING SEISMIC ISOLATION SYSTEM**

This section details the design of a prototype PRSIS consisting of a pivot mechanism and a series of ring spring cartridge assemblies.

#### **8.2.1 Pivot mechanism**

A Hooke's joint was chosen for the PRSIS pivot mechanism. This selection was made primarily because of the simplicity and ease of manufacture of a Hooke's joint.

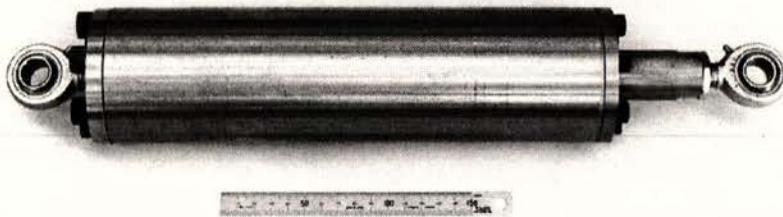
#### **8.2.2 Uni-directional ring spring cartridges**

In conjunction with the pivot mechanism, a system of four ring spring assemblies fitted at 90° intervals around the periphery of a stiff base, was developed. Symmetry of this design enables the system to be analysed as an equivalent two-dimensional system for inputs applied in line, as well as at an angle of 45°, with the axes of the ring spring cartridges.

Uni-directional ring spring cartridges were selected for this design because they provide half the stiffness of that of a bi-directional cartridge system of similar size. This enables reduced structure mass for a similar fundamental period for the system.

Each uni-directional cartridge comprises of an outer housing that provides external guidance for the ring spring and a piston that permits spring compression. The piston is fitted with an insertion rubber liner that cushions contacting as the cartridge is loaded and unloaded. The cartridges each incorporate a ring spring of maximum load capacity of 20 kN. These springs, type 03800, were provided by Ringfeder GmbH, Germany.

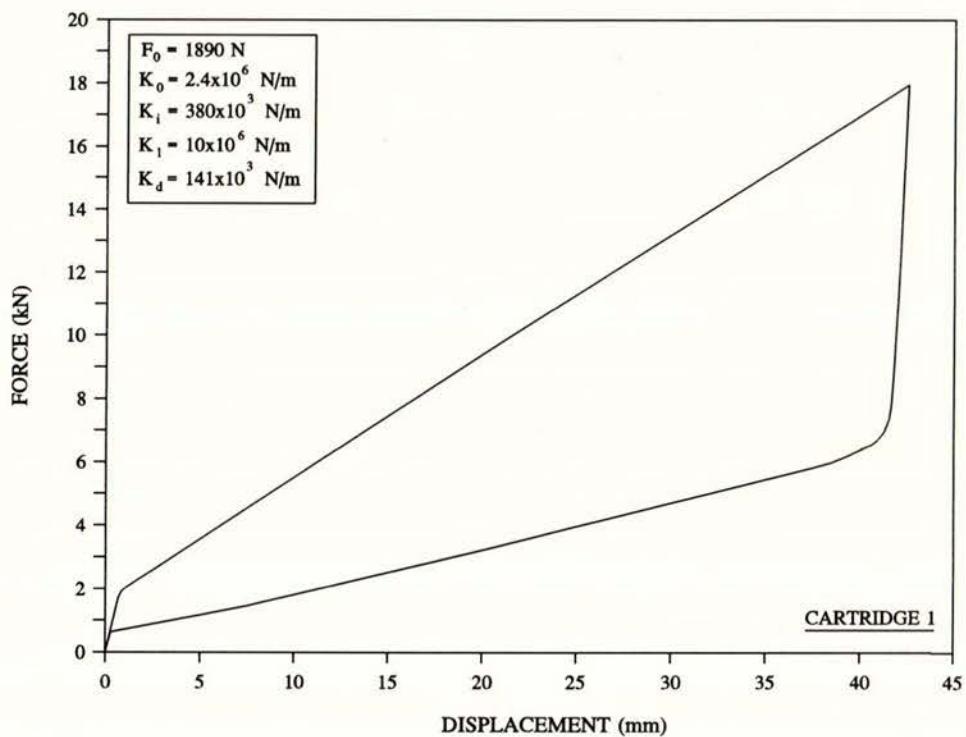
Rod ends, attached to both ends of the ring spring cartridge, ensure only axial loads are applied to the ring spring cartridge. A scaled general assembly drawing of the uni-directional ring spring cartridge is presented in Appendix B. The manufactured cartridge is shown in Figure 8.1.



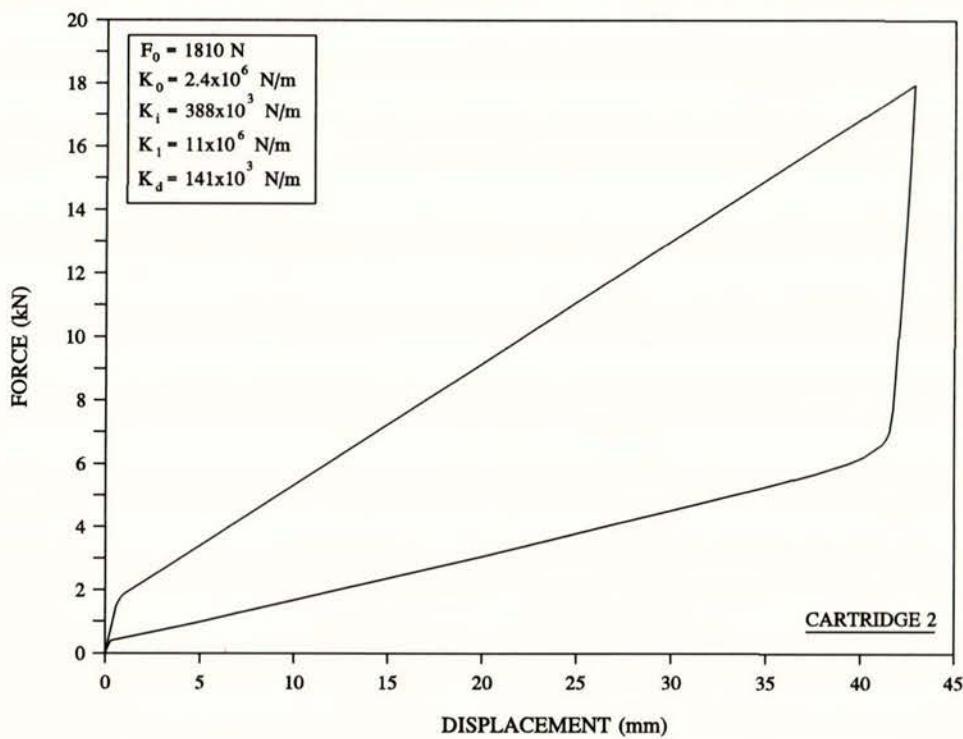
**Figure 8.1** Uni-directional ring spring cartridge  
(with 150 mm ruler)

(a) Cartridge hysteresis diagrams

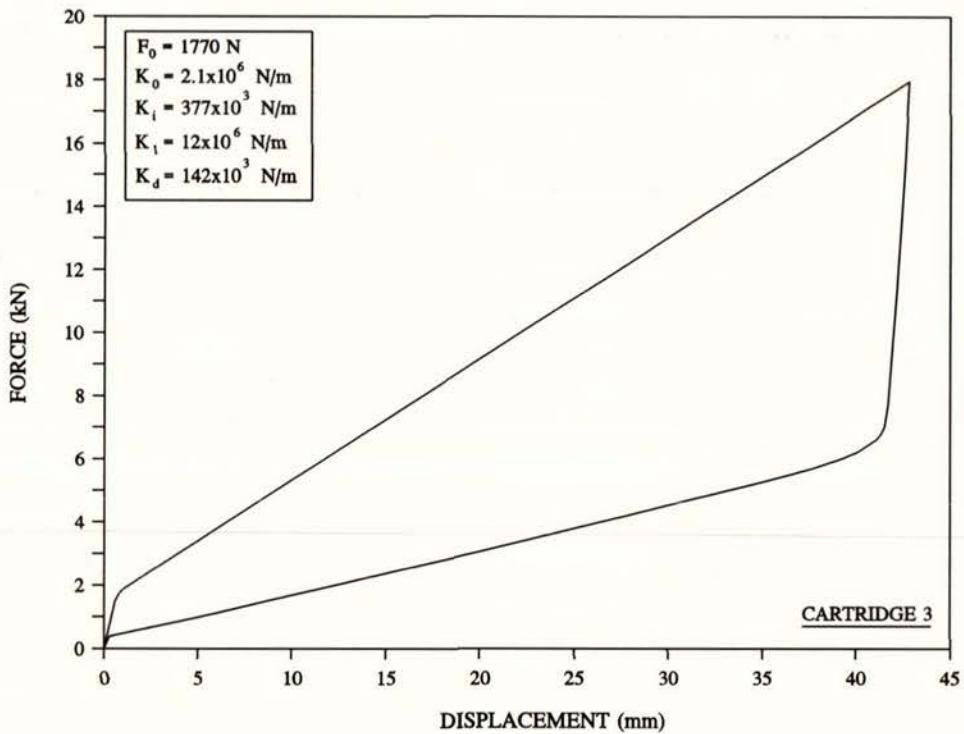
Each cartridge was tested in a universal testing machine to determine its force/deflection characteristics. The cartridges were each subjected to a 5 mm pre-displacement (10% pre-displacement) and testing was carried out for loading and unloading rates of 15 mm/min and 5 mm/min respectively. The hysteresis diagrams for each of these cartridges is presented in Figures 8.2, 8.3, 8.4, and 8.5.



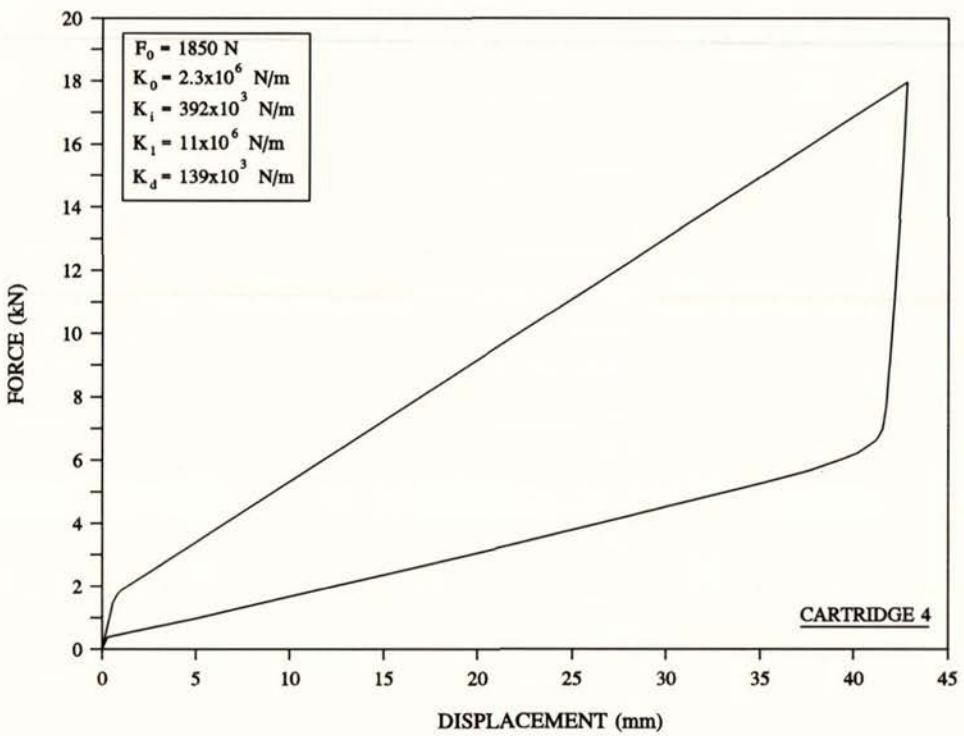
**Figure 8.2** Hysteresis diagram: ring spring cartridge No. 1



**Figure 8.3** Hysteresis diagram: ring spring cartridge No. 2



**Figure 8.4** Hysteresis diagram: ring spring cartridge No. 3



**Figure 8.5** Hysteresis diagram: ring spring cartridge No. 4

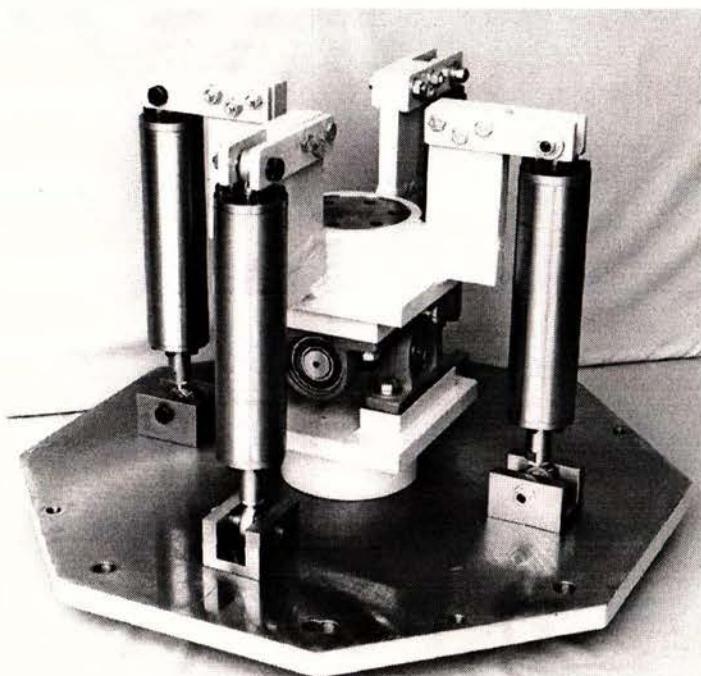
A summary of the stiffnesses for the four ring spring cartridges is presented in Table 8.1. These results show good uniformity of the stiffness properties.

**Table 8.1** Ring spring cartridge stiffnesses

	Force $F_0$	Stiffness $K_0$	Stiffness $K_i$	Stiffness $K_l$	Stiffness $K_d$
Cartridge 1	1890 N	$2.4 \times 10^6$ N/m	$380 \times 10^3$ N/m	$10 \times 10^6$ N/m	$141 \times 10^3$ N/m
Cartridge 2	1810 N	$2.4 \times 10^6$ N/m	$388 \times 10^3$ N/m	$11 \times 10^6$ N/m	$141 \times 10^3$ N/m
Cartridge 3	1770 N	$2.1 \times 10^6$ N/m	$377 \times 10^3$ N/m	$12 \times 10^6$ N/m	$142 \times 10^3$ N/m
Cartridge 4	1850 N	$2.3 \times 10^6$ N/m	$392 \times 10^3$ N/m	$11 \times 10^6$ N/m	$139 \times 10^3$ N/m
<hr/>					
<b>Average</b>	1830 N	$2.3 \times 10^6$ N/m	$384.25 \times 10^3$ N/m	$11 \times 10^6$ N/m	$140.75 \times 10^3$ N/m

### 8.2.3 PRSIS assembly

The PRSIS assembly, utilises fabricated offset arms for attachment of the ring spring cartridges; this enables a compact arrangement. A scaled assembly drawing of the PRSIS assembly is presented in Appendix B, with the manufactured system shown in Figure 8.6.



**Figure 8.6** PRSIS assembly

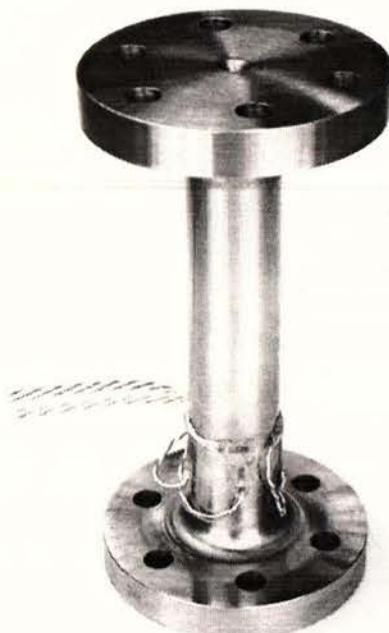
#### 8.2.4 Design details of columnar structure

It was desired that the fundamental period of a columnar structure to be tested fall within the range of dominant spectral accelerations for short period type earthquakes. A columnar structure with an overall height of 1.8 m and with a computed fundamental period of 0.292 seconds, was designed. This period was selected to be typical of systems reported in Section 7.2.3.

The columnar structure comprises of flexible elements and solid members.

##### (a) Flexible elements

Two flexible elements were manufactured using high strength steel (AISI 4340); the material properties, as supplied by the manufacturer, are as follows:  $\sigma_{yield} = 756$  MPa and  $\sigma_{UTS} = 912$  MPa. High strength steel was chosen for the design to accommodate the high flexural loads induced in the element during test conditions. One of the manufactured flexible elements is shown in Figure 8.7. The element has an overall height of 280 mm, a shaft diameter of  $\varnothing 46.50$  mm and is of mass = 8.3 kg.



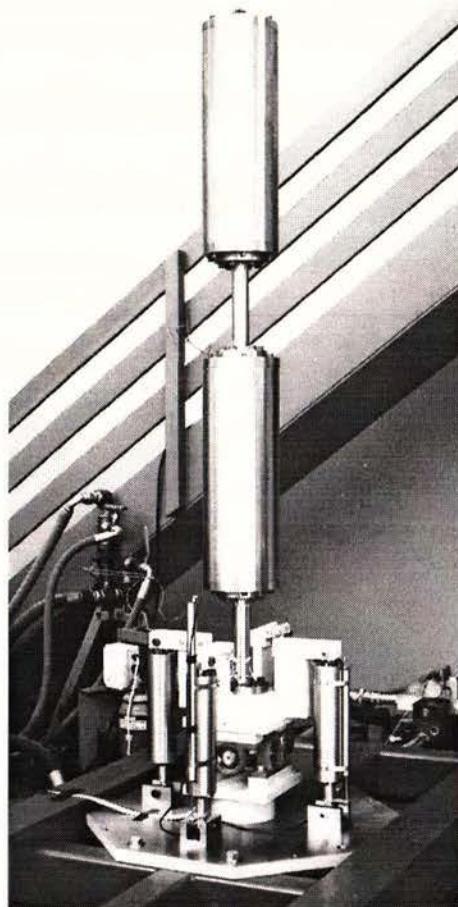
**Figure 8.7** Column flexible element

(b) Solid members

The column design also embodies two solid steel members. These members, each of height 620 mm, and of mass 152 kg, were chosen to give the desired column fundamental period.

### 8.2.5 Complete test rig assembly

The complete test rig assembly comprises of: flexible elements, solid members as well as the PRSIS discussed above. A scaled assembly drawing of the complete assembly is presented in Appendix B. This system is shown in Figure 8.8.



**Figure 8.8** Complete test rig assembly mounted on shaker table

### 8.3 EXPERIMENTAL SYSTEM

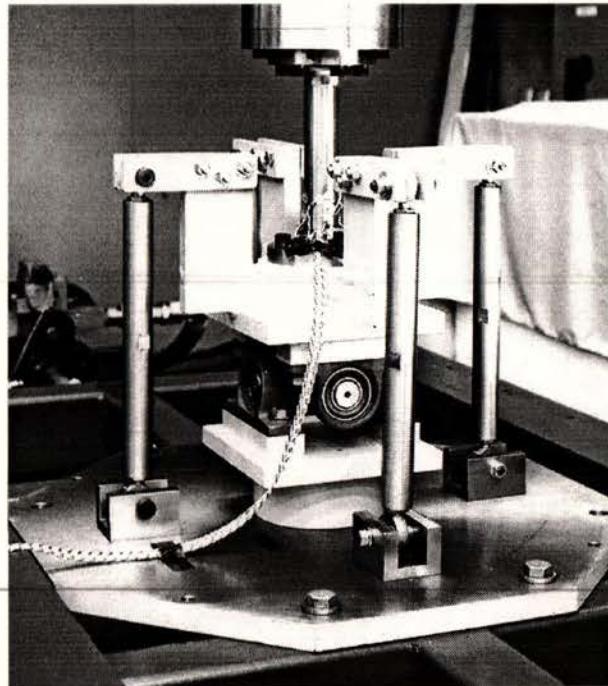
This section reports on the detail of the experimental system. Discussion covers the test configurations used for dynamic testing and the instrumentation used to measure the dynamic response of the system. The free vibration characteristics of the unisolated system are also detailed.

#### 8.3.1 System configurations

To enable the effectiveness of the PRSIS to be determined, an isolated and unisolated system were tested.

##### (a) Unisolated system

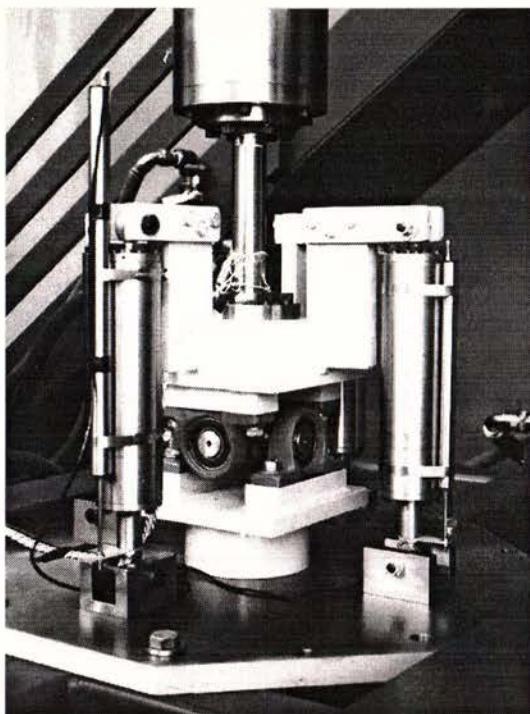
The unisolated system was rigidly attached to the shaker table by replacing the ring spring cartridges of the PRSIS with solid steel bars of Ø39 mm. The unisolated system is shown in Figure 8.9.



**Figure 8.9** Unisolated system mounted on shaker table

(b) Isolated system (PRSIS)

The PRSIS has been detailed above. To enable computer modelling of the effects of both pre-displacement and pre-load, the assembly was subjected to a 10% pre-displacement and a 11.8% pre-load. This pre-load was obtained by compressing each ring spring 2.0 mm. The PRSIS, attached to the shaker table, is presented in Figure 8.10



**Figure 8.10** PRSIS mounted on shaker table

### 8.3.2 Test facility

Experimental testing was conducted using the shaker table test facility at Industrial Research Ltd in Christchurch. Detail of this facility is reported in Section 6.2.2.

### 8.3.3 Instrumentation

For the experimental tests undertaken, the following parameters were recorded: shaker table input acceleration; ring spring cartridge displacement; and column bending moment. The instrumentation used to measure these parameters is now discussed.

(a) Accelerometer

The shaker table input acceleration was measured using a Brüel and Kjær (B&K) accelerometer, attached directly to the shaker table actuator. The accelerometer was calibrated using a B&K accelerometer calibrator. An acceleration of 1.0 g gives a charge amplifier output of 1.0 volt.

(b) Linear variable differential transducers

Two Schlumberger linear variable differential transducers (LVDT's) were used to measure ring spring cartridge displacements. The LVDT's were attached to two orthogonal ring spring cartridges, thus enabling displacements to be measured at 90° positions. Each LVDT was calibrated to give a 1.0 volt output for a 10.0 mm ring spring displacement. Attachment of the LVDT's to the ring spring cartridges is seen in Figure 8.10.

(c) Strain gauges

Bending moments induced in the column during dynamic loading were measured by attaching a series of  $120\Omega$  strain gauges at 90° intervals around the periphery of the shaft section of the flexible element. This is seen in Figure 8.7; only the vertically mounted gauges were used in this study. To minimise the influence of localised stress concentration effects, the strain gauges were attached 30 mm away from the fillet radius at the junction of the shaft section and the flange (Figure 8.7).

Calibration of the strain gauges was undertaken by loading the flexible element in a universal testing machine and measuring quasi-static strains using a six channel Kyowa dynamic strain amplifier. Pairs of strain gauges, each fitted at 180° intervals on the flexible element, were connected in a half bridge configuration. The gauges were connected to give the summation of the strains for the axial load conditions experienced during testing in the universal testing machine.

For applied loads up to 200 kN, the mean strain amplifier output voltage was 1.202 volts/1000  $\mu\text{m}$  strain. Using the sectional properties of the flexible element and  $E_{\text{steel}} = 210$  GPa, the equivalent bending moment/volt output is 862.3 N-m/volt.

For the shaker table tests the strain gauges were connected to give the difference in strains; this was because the loading experienced by the flexible element during these tests was in bending.

#### **8.3.4 Data acquisition system**

The recording of the experimental data was undertaken using a PC fitted with a 16 channel data acquisition board. For all the tests undertaken, a sampling frequency of 100 Hz was used. Either three or five channels were used to record the data for the experimental tests. For testing of the unisolated system three data acquisition channels were used to record:

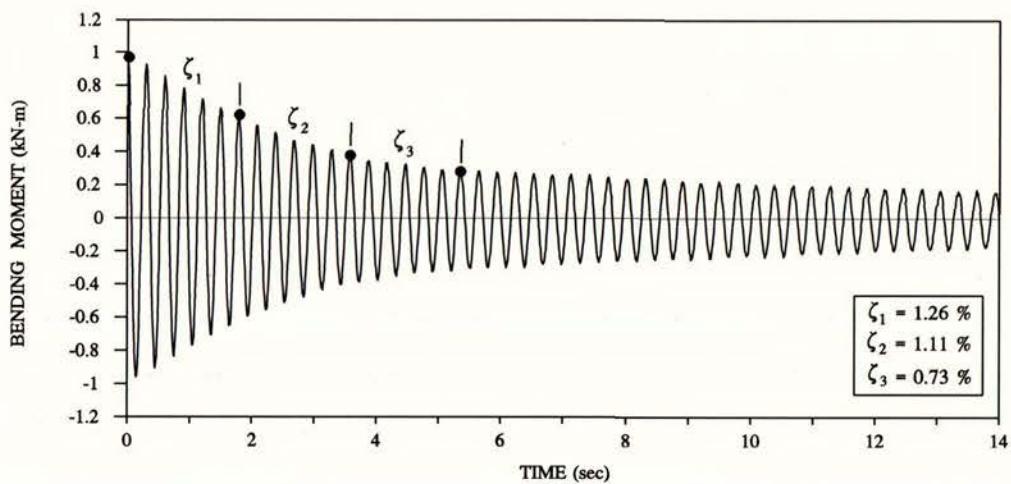
- shaker table input acceleration;
- column bending strains in line with the axis of the shaker table;
- column bending strains at 90° with the axis of the shaker table

For testing of the PRSIS, two further data acquisition channels were used. These recorded:

- ring spring cartridge displacement in line with the axis of the shaker table;
- ring spring cartridge displacement at 90° with the axis of the shaker table;

#### **8.3.5 Unisolated system fundamental period and damping factor**

Prior to commencing dynamic testing of the unisolated system and the PRSIS, the fundamental period and damping of the column were determined. Testing was carried out by subjecting the unisolated system to an impulsive input and measuring the free vibration decay of the column. The dynamic response measured by the strain gauges fitted in line with the axis of the shaker table is presented in Figure 8.11.



**Figure 8.11** Free vibration response of unisolated system

The fundamental period of the unisolated system was measured as 0.280 seconds. This was determined by applying a Fast Fourier Transform (FFT) to the free vibration response signal (Figure 8.11). This experimental value compares well with the computed fundamental period of 0.292 seconds.

An attempt was made to determine the second mode period of the unisolated system by striking the column at the lower solid member and then recording the free vibration response. However, an FFT analysis of the response did not show a sufficiently measurable value to determine the second mode period. The computed second mode period for the system was 0.042 seconds.

The damping of the unisolated system was obtained from the free vibration decay taken over intervals of six successive cycles. Using Equation 3.14, damping factors of  $\zeta_1 = 1.26\%$ ;  $\zeta_2 = 1.11\%$ , and  $\zeta_3 = 0.73\%$  were obtained over the first eighteen cycles of the response (Figure 8.11). This reduction in damping as the response diminishes is typical of systems where the level of damping is dependent on the magnitude of displacement.

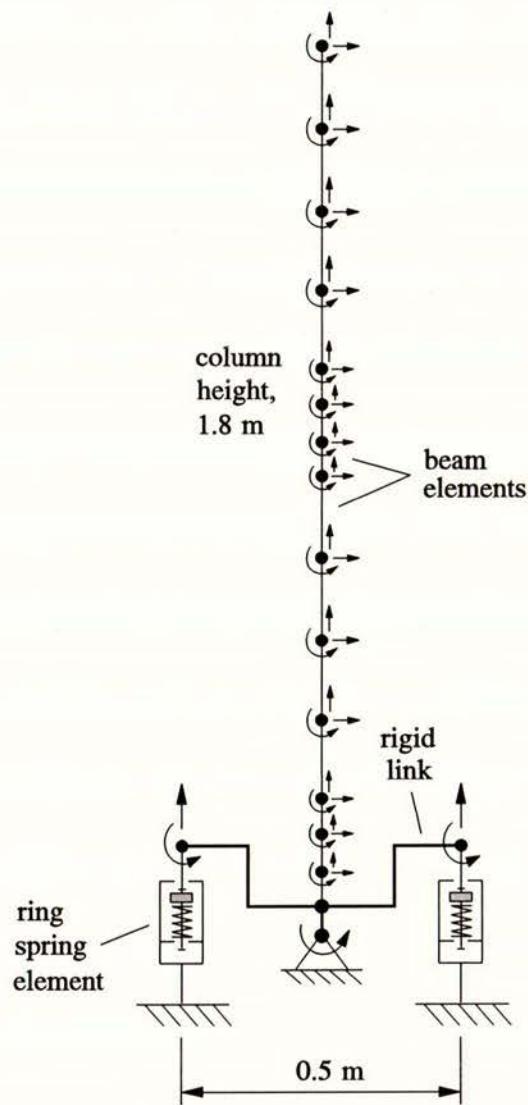
#### 8.4 COMPUTATIONAL MODELLING

Computer simulation analyses were undertaken to enable a comparison between experimental tests and computer simulation results. These analyses were conducted using the computer program RUAUMOKO (Carr 1995). To facilitate these analyses, a model of

the PRSIS was developed for use in RUAUMOKO.

#### 8.4.1 PRSIS computer model

The computer model of the PRSIS comprises of a series of beam elements with lumped nodal masses representing the flexible elements and solid masses. Rigid links have been selected to represent the base arms attaching the ring spring cartridges. Further, a rigid link was incorporated at the base of the column to model a stiff connection between the centre of the Hooke's joint and the column. Dimensional coordinates and physical properties of the elements of the model were obtained directly from the experimental system. The PRSIS computer model is shown in Figure 8.12.



**Figure 8.12** Schematic of PRSIS model

The node, directly above the base of the column, was assigned the same coordinates as that of the position of the strain gauges in the experimental system. This was to enable a direct comparison of the experimental and computer simulation results. The stiffnesses chosen for the ring springs elements,  $K_0$ ,  $K_i$ ,  $K_l$ ,  $K_d$ , were obtained from the average of the experimental values for the four cartridges; these are given in Table 8.1.

The computer model employs a consistent mass matrix and linear damping of 1% for all modes. 1% damping was chosen as being a representative level of damping based upon the free vibration response. The time step used in all analyses was 0.0001 seconds; this gave a converged solution for the cases considered.

Large displacement analyses were undertaken to enable the effect due to changes in system geometry to be included. The gravity effects of the column self weight, as well as the rotational inertia of the PRSIS, were also considered. The calculated PRSIS mass moment of inertia  $I_0 = 3.197 \text{ kg-m}^2$ .

## **8.5 EXPERIMENTAL TESTS AND COMPUTER SIMULATION RESULTS FOR UNISOLATED AND PRSIS SYSTEMS**

This section details the earthquake excitations used for dynamic testing, followed by the dynamic response of experimental tests and computer simulation analyses for the unisolated and PRSIS systems.

### **8.5.1 Earthquake inputs selected for dynamic testing**

The four earthquakes chosen for this study are: El Centro 1940 N-S; Pacoima 1971 S16E; Parkfield 1966 N65E; and El Centro 1979 N-S. Due to shaker table displacement limitations these earthquakes were not all able to be applied full size. Hence, the following scaled earthquake inputs were used for the experimental tests:

- El Centro 1940 N-S (100%)
- Pacoima 1971 S16E (33.3%)
- Parkfield 1966 N65E (70%)
- El Centro 1979 N-S (80%)

Acceleration inputs, recorded by an accelerometer attached to the shaker table, were used as the RUAUMOKO input file. RUAUMOKO results for the PRSIS were output for:

- (1) the bending moment at the node positioned at the strain gauges;
- (2) the ring spring displacement.

For the unisolated system, column bending moment only was output.

Results of analyses for the El Centro 1940 N-S earthquake are detailed in the following sections. The study considers the dynamic behaviour of the following four cases:

- (i) the unisolated system built in at its base;
- (ii) the unisolated system including the effects of subsystem flexibility;
- (iii) the PRSIS mounted in line with the shaker table axis;
- (iv) the PRSIS mounted at 45° with the shaker table axis.

### **8.5.2 Dynamic response of unisolated system**

In this section the dynamic behaviour of the unisolated system, excluding, as well as including, sub-system flexibility is detailed.

#### **(a) Excluding sub-system flexibility**

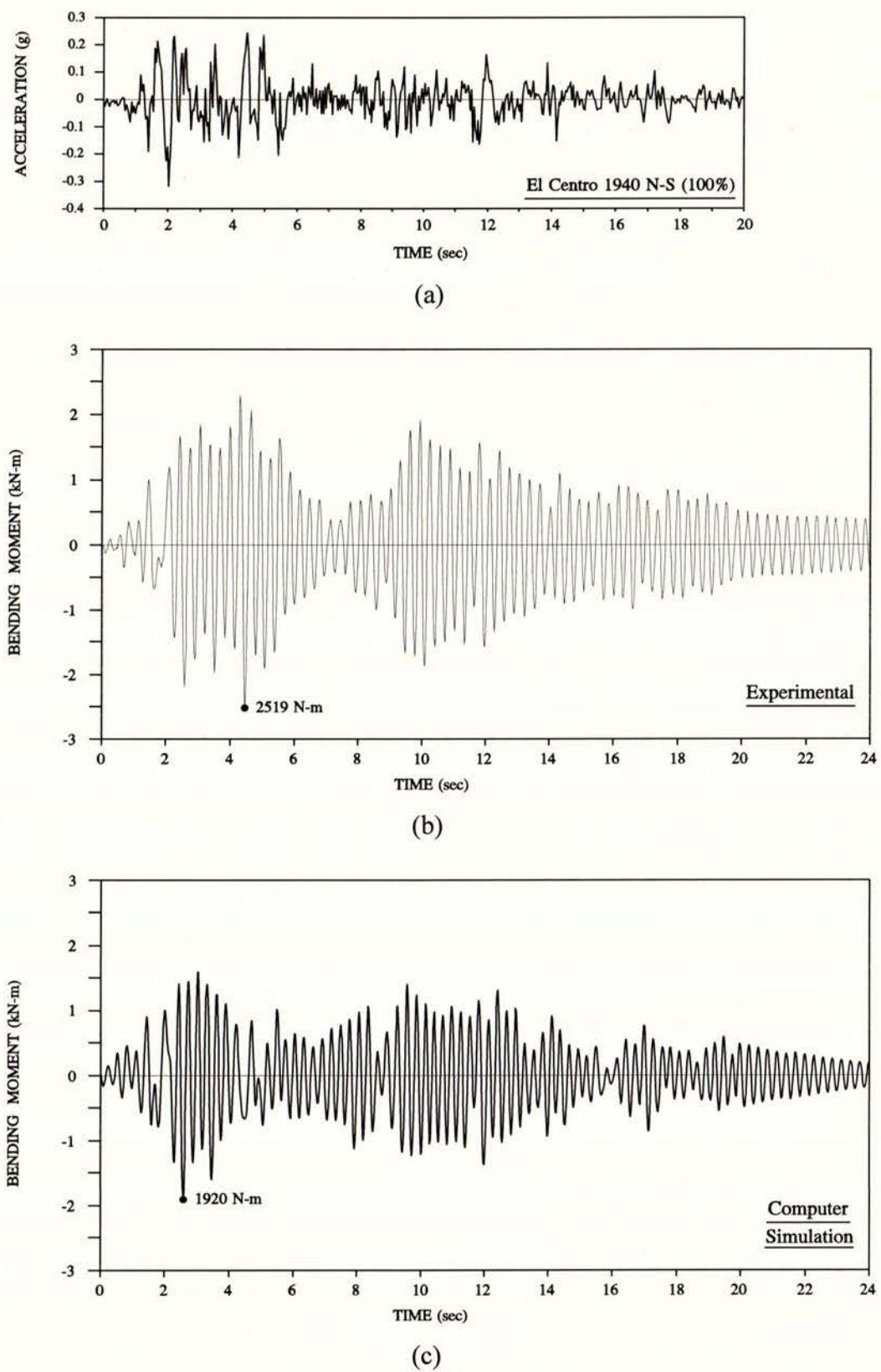
The system, mounted on the shaker table, was subjected to the El Centro 1940 N-S earthquake. The shaker table acceleration, measured by the accelerometer attached to the shaker table, is presented in Figure 8.13 (a). The bending moment time history experienced by the flexible element during this excitation is shown in Figure 8.13 (b). These results are plotted using the same time base.

For the computer simulation analysis, the computer model considered the column to be built in at its base, that is, the base node of the column was constrained in the x, y and θ directions. The fundamental period of this system, given by the computer model, is 0.292 seconds. This compares with a value of 0.280 seconds obtained experimentally.

The computer simulation bending moment time history using the measured shaker table acceleration is presented in Figure 8.13 (c).

The bending moment time history results show a generally larger level of bending moment given experimentally than by computer simulation. A maximum bending moment of 2519 N-m @ 4.46 seconds was obtained experimentally; this compares with 1920 N-m @ 2.58 seconds given by computer simulation. It is seen that the values occurring for both the experimental and computer simulation cases are similar at 2.58 seconds, however an amplification in the level of bending moment for the experimental case is seen at 4.46 seconds. This maximum bending moment occurs during the high level of impulsive loading in the 4-5 second interval for the input.

The response of a structure during this interval of excitation will be influenced by the phasing of the input with respect to the structural deformation at that time. For lightly damped structures, small differences in their properties can lead to sizable differences in the loading experienced. Such an effect is observed in a typical earthquake response spectrum for light damping.



**Figure 8.13** Dynamic response of unisolated system -

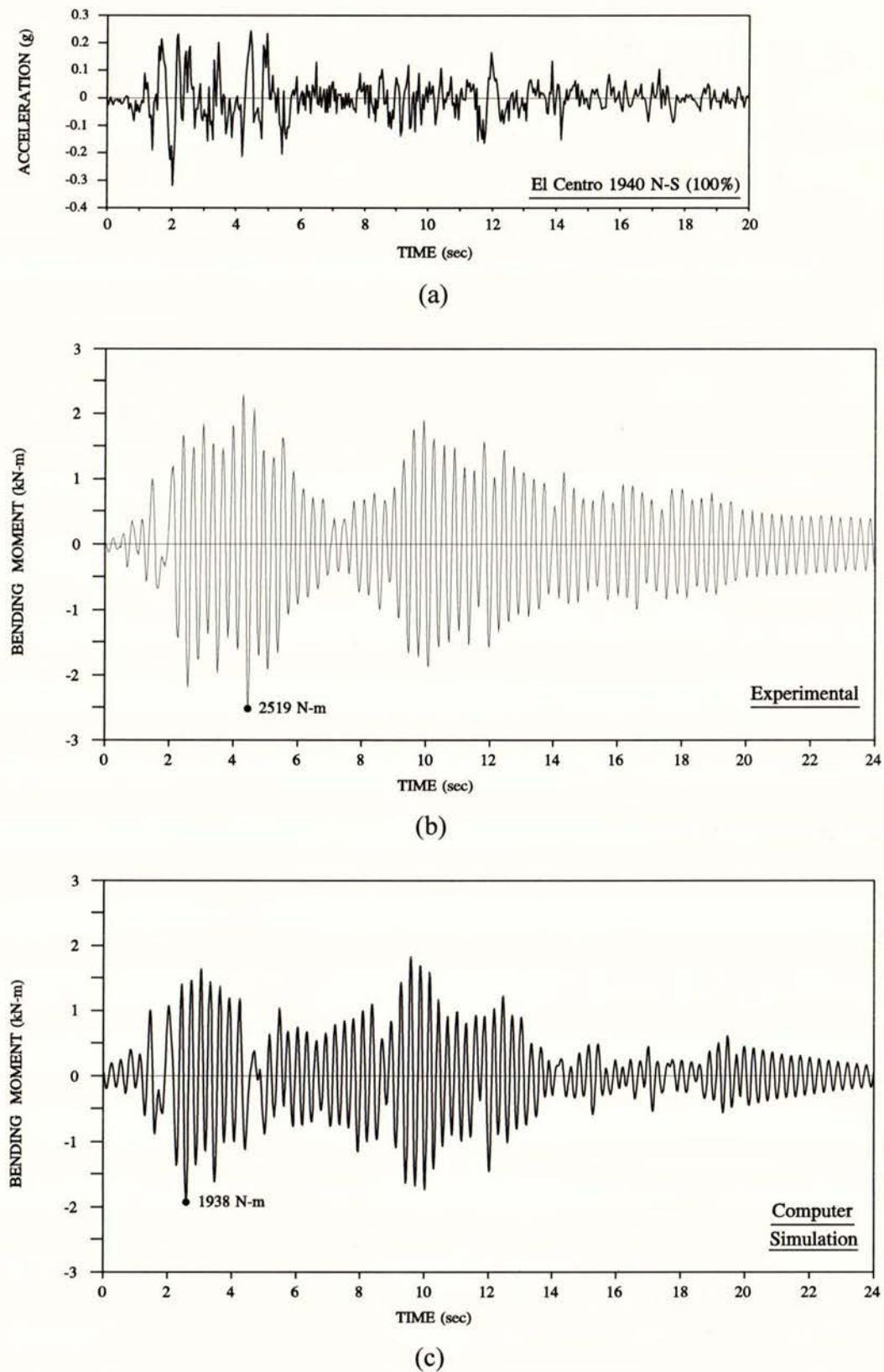
- Earthquake: El Centro 1940 N-S (100%)
- Bending moment time history (experimental)
- Bending moment time history (computer simulation)

## (b) Including sub-system flexibility

The computer model was modified to include the flexibility of the sub-system. The flexural stiffness of the base arms and the axial stiffness of the steel bars was calculated, thus enabling the properties of an equivalent spring member to be determined. The computer model for this system is similar to that shown in Figure 8.12, except that the sub-system now incorporates the equivalent elastic spring members. A fundamental period of 0.297 seconds was given by RUAUMOKO for this system; this compares with 0.292 seconds for the system excluding sub-system flexibility.

The same El Centro 1940 N-S earthquake as was used in the system excluding sub-system flexibility, was used as the input excitation in RUAUMOKO. Computer simulation results for the system including sub-system flexibility are presented in Figure 8.14 (c). A small increase is observed in the maximum bending moment: up from 1920 N-m to 1938 N-m. It is also observed that the amplitude of the bending moment is increased over the 2–4.5 and 9–10.5 second interval for the system including sub-system flexibility. Though this increase is small, it more closely represents the experimental results.

The next section considers the dynamic behaviour of the PRSIS.



**Figure 8.14** Dynamic response of unisolated system including sub-system flexibility -

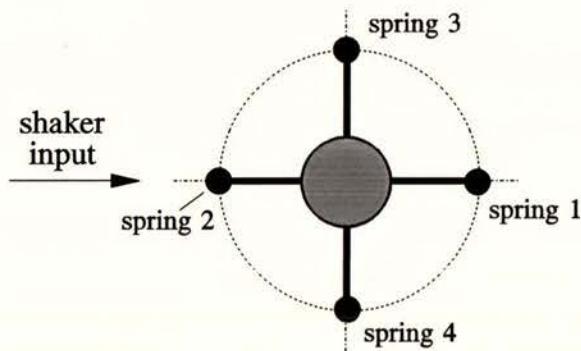
- (a) Earthquake: El Centro 1940 N-S (100%)
- (b) Bending moment time history (experimental)
- (c) Bending moment time history (computer simulation)

### 8.5.3 Dynamic response of pre-displaced pre-loaded PRSIS

The dynamic behaviour of the PRSIS (as detailed in Section 8.3.1 (b)) is now considered for the system with the dynamic input along the axis of the PRSIS, and also at  $45^\circ$  to the axis of the PRSIS.

#### (a) Dynamic input along axis of PRSIS

A plan view of the PRSIS with one of its axes in line with that of the shaker table is shown in Figure 8.15.



**Figure 8.15** Plan view of PRSIS in line with axis of shaker table

This system was subjected to the El Centro 1940 N-S earthquake. The recorded shaker table acceleration is presented in Figure 8.16 (a) and the bending moment time history experienced by the flexible element during this excitation is shown in Figure 8.16 (b). The bending moment time history obtained using RUAUMOKO is presented in Figure 8.16 (c).

For both the experimental and computer simulation cases, similar trends are seen over the full duration of the bending moment time histories. The maximum bending moment obtained experimentally was 1597 N-m @ 2.24 seconds. This compares with 1295 N-m @ 2.20 seconds given by computer simulation.

The ring spring displacement time histories obtained experimentally and by computer simulation are presented in Figures 8.17 (b) and (c). Displacements oscillate about a value of 2 mm; this being the initial ring spring displacement due to a spring pre-load of 11.8%.

Good correlation of the results is seen, with a maximum ring spring displacement of 12.55 mm obtained experimentally and 11.23 mm given by computer simulation.

It is seen from the experimental results that between 6–8 seconds the structure has not fully returned to its zero position as for the computer simulation results. During this time, for both the experimental and computer simulation cases, small amplitude ring spring displacements occur while on stiffness  $K_1$  and  $K_0$  respectively, hence the structure experiences bending moments as shown in Figures 8.16 (b) and (c).

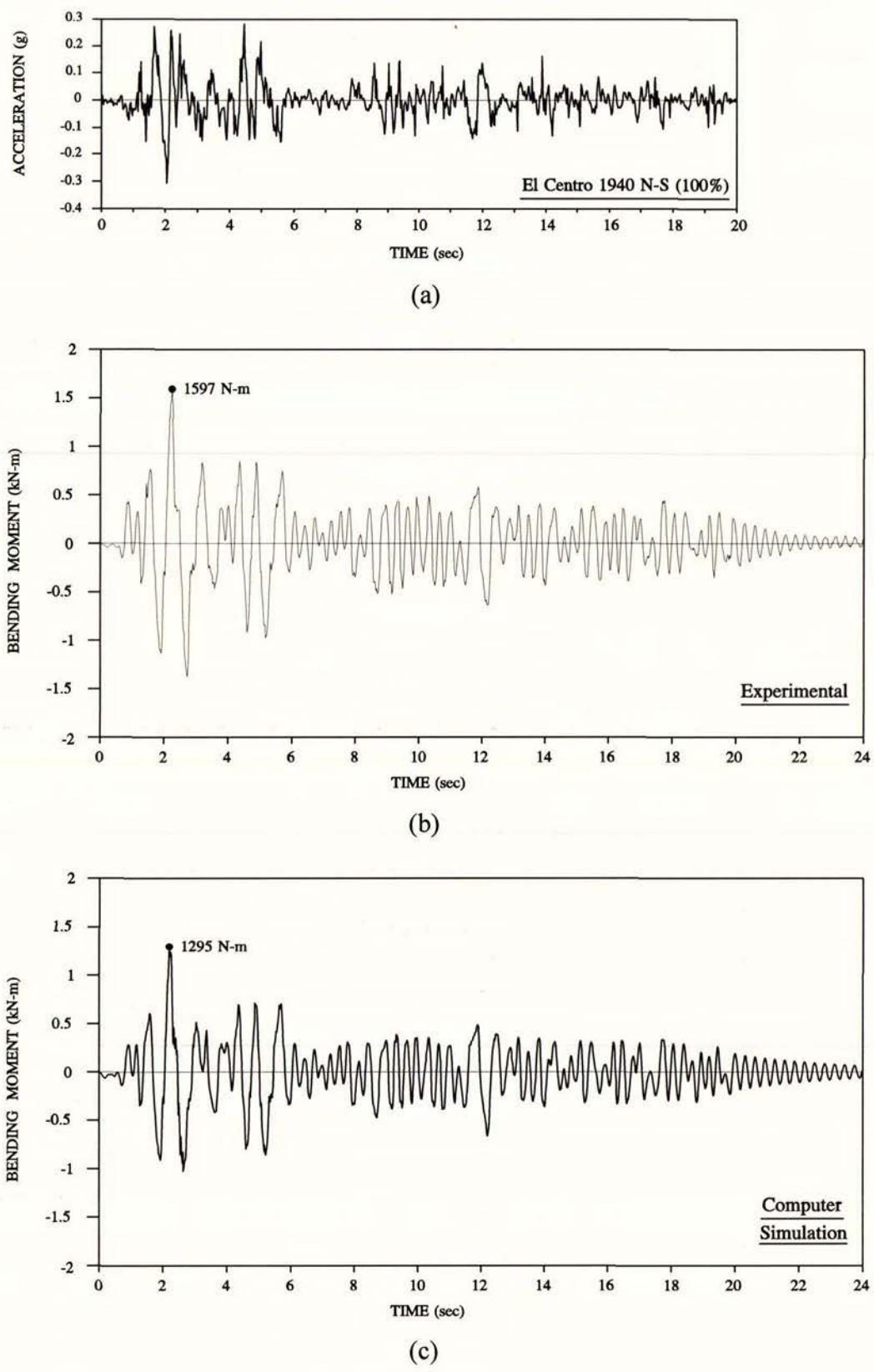
To determine the effectiveness of the PRSIS, their results are compared with those given for the unisolated system. The percentage ratios of the isolated/unisolated bending moments are as follows:

Experimental:  $(1597/2519) = 63.4\%$

Computer simulation:  $(1295/1920) = 67.4\%$

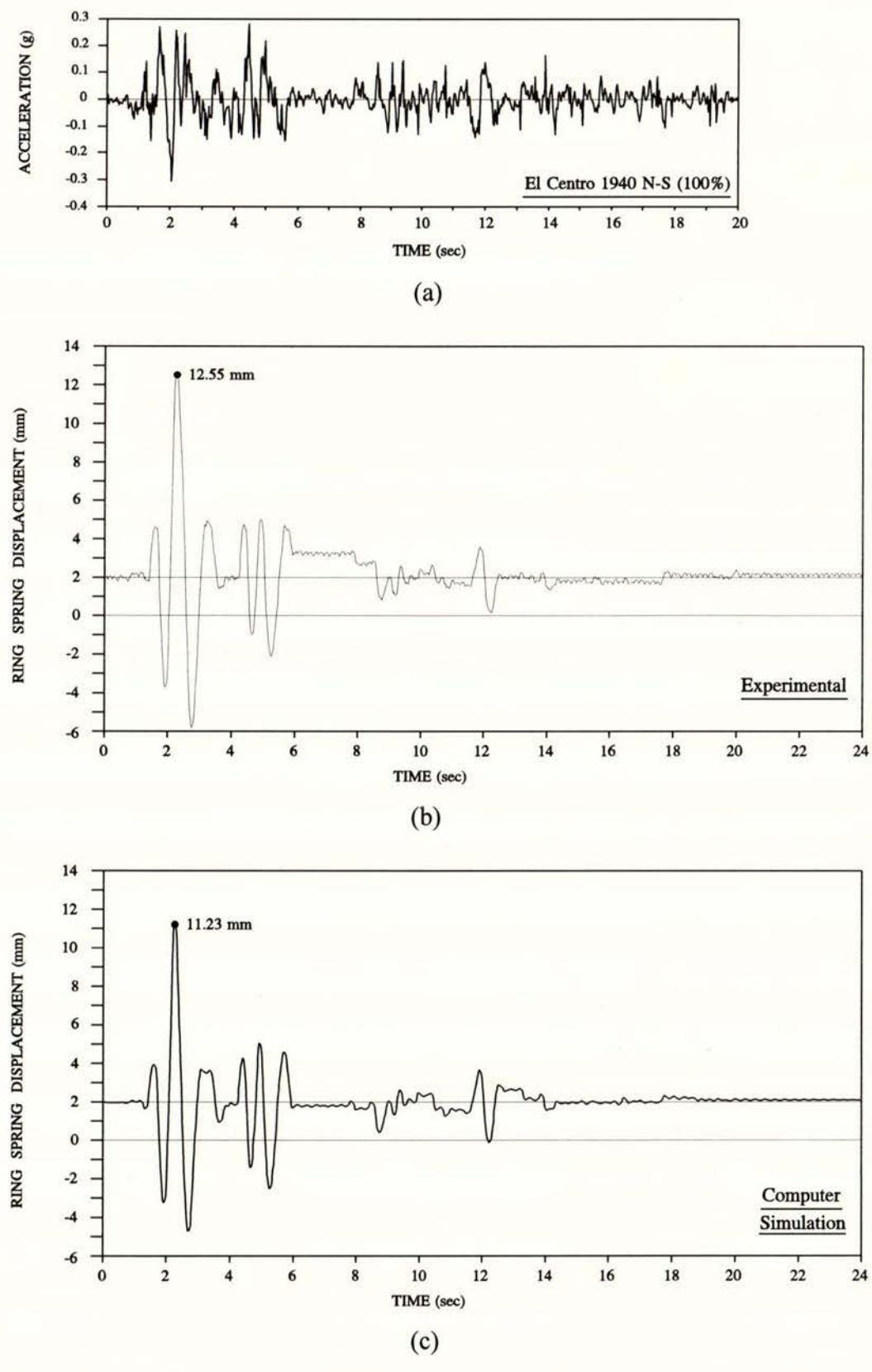
This reduction in the structure bending moment confirms that the PRSIS is effective in providing isolation for the system.

In practical systems, the direction of the earthquake input need not be in line with either axis of the PRSIS. To assess the influence of inputs not in line with the PRSIS axes, the case with an earthquake input at an angle of  $45^\circ$  was studied.



**Figure 8.16** Dynamic response of PRSIS at  $0^\circ$  -

- (a) Earthquake: El Centro 1940 N-S (100%)
- (b) Bending moment time history (experimental)
- (c) Bending moment time history (computer simulation)

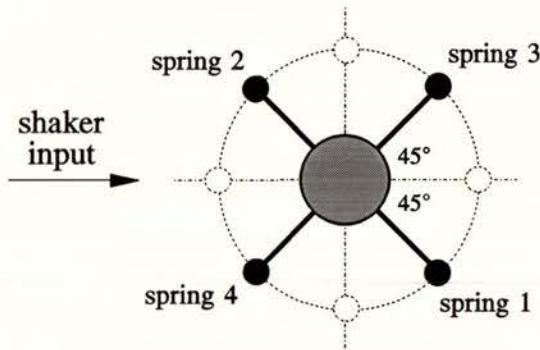


**Figure 8.17** Dynamic response of PRSIS at 0° -

- (a) Earthquake: El Centro 1940 N-S (100%)
- (b) Ring spring displacement time history (experimental)
- (c) Ring spring displacement time history (computer simulation)

(b) Dynamic input at  $45^\circ$  to axis of PRSIS

A plan view of the PRSIS with its axes at  $45^\circ$  to that of the shaker table is shown in Figure 8.15.



**Figure 8.18** Plan view of PRSIS rotated  $45^\circ$

Shaker table tests were carried out for the system subjected to the El Centro 1940 N-S earthquake. The measured shaker table acceleration and the bending moment time history experienced by the flexible element, are presented in Figures 8.19 (a) and (b).

The computer model used in RUAUMOKO consists of a modification to that shown in Figure 8.12. Due to symmetry of the system under consideration (Figure 8.18), it is possible to re-position the ring spring members onto the axis of the input. Thus, the system can be considered as having two ring spring cartridges, either side of the pivot, at a reduced radius of  $250/\sqrt{2} = 176.78$  mm. This enables computational modelling of the system as a two-dimensional case.

Bending moment time history results, obtained using RUAUMOKO, are presented in Figure 8.19 (c). For both the experimental and computer simulation results, similar trends are seen over the full duration of the bending moment time histories (Figures 8.19 (b) and (c)). The maximum bending moment obtained experimentally was 1511 N-m @ 2.16 seconds; this compares with 1337 N-m @ 2.20 seconds given by computer simulation.

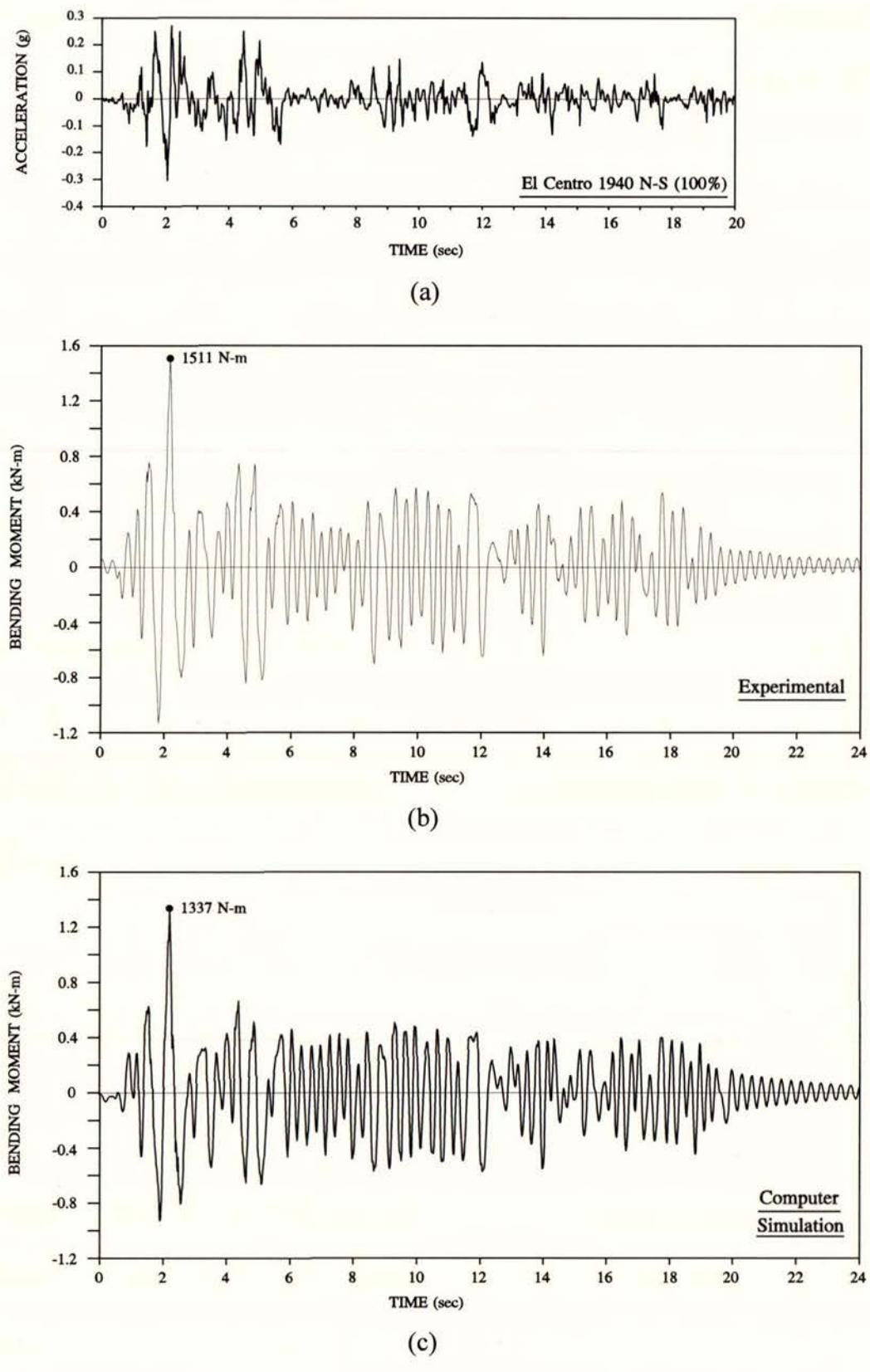
These results are similar to those obtained for the PRSIS mounted in line with the shaker table axis, c.f., 1597 N-m @ 2.24 seconds obtained experimentally and 1295 N-m @ 2.20

seconds given by computer simulation.

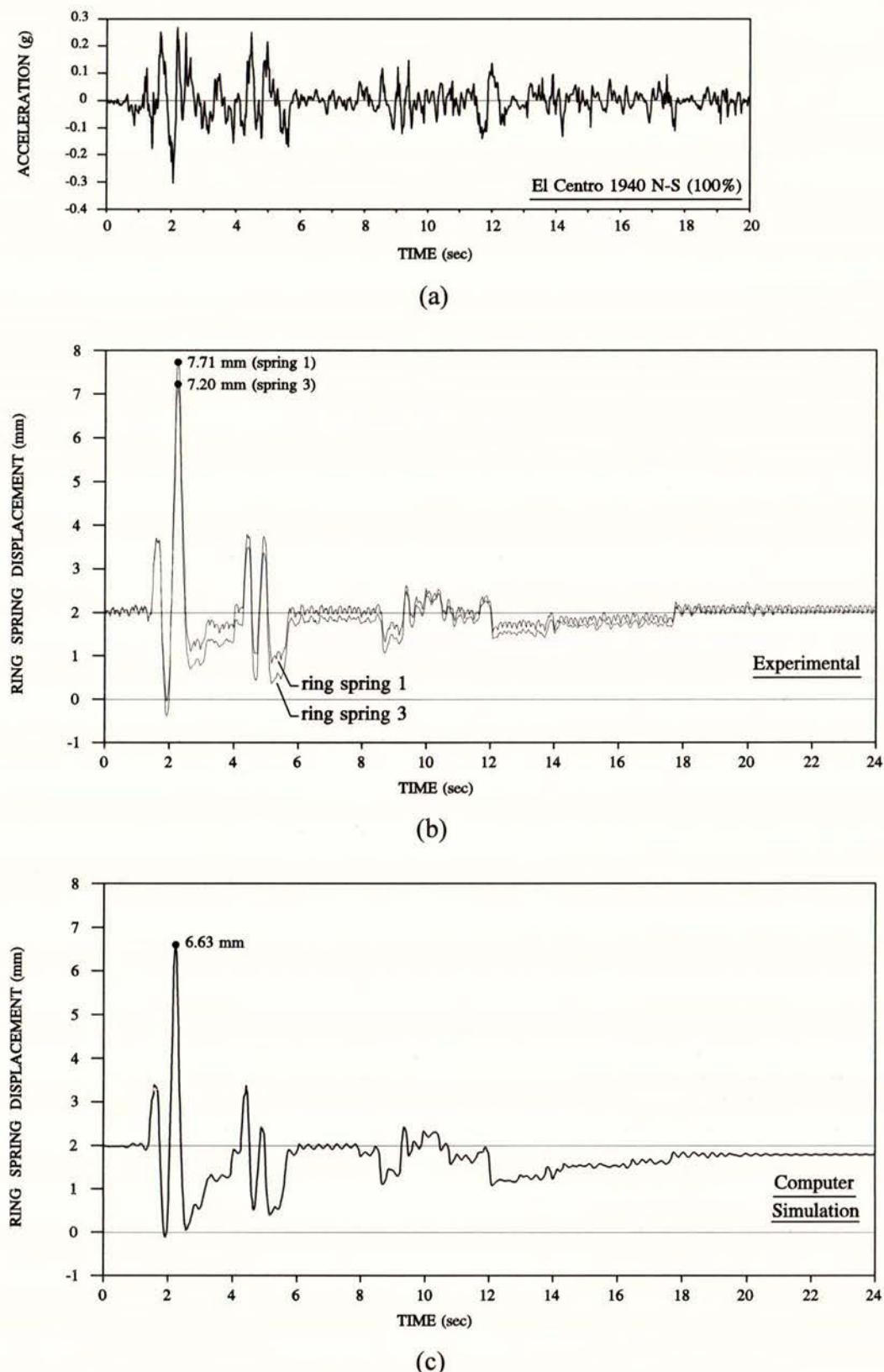
The ring spring displacement time histories, obtained experimentally and by computer simulation for the PRSIS rotated 45°, are presented in Figures 8.20 (b) and (c).

Displacement results for ring spring cartridges 1 and 3, both fitted with LVDT's, are seen to be in good agreement (Figure 8.20 (b)). This suggests that rocking of the system occurred about an axis orthogonal to the shaker input. Good correlation is seen for the experimental and computer simulation results, with maximum ring spring displacements of 7.20 mm and 7.71 mm obtained experimentally and 6.63 mm given by computer simulation.

It is seen from the computer simulation results (Figure 8.20 (c)) that the structure returns near to its initial position of 2 mm; its final displacement being 1.79 mm. This effect of a small offset from the original displacement is common to pre-displaced pre-loaded systems.



**Figure 8.19** Dynamic response of PRSIS rotated 45° -  
 (a) Earthquake: El Centro 1940 N-S (100%)  
 (b) Bending moment time history (experimental)  
 (c) Bending moment time history (computer simulation)



**Figure 8.20** Dynamic response of PRSIS rotated 45° -

- (a) Earthquake: El Centro 1940 N-S (100%)
- (b) Ring spring displacement time history (experimental)
- (c) Ring spring displacement time history (computer simulation)

#### **8.5.4 Summary results of maximum structure bending moments for unisolated and PRSIS systems**

The above study has considered the dynamic behaviour of an unisolated system and a PRSIS subjected to the El Centro 1940 N-S earthquake. Testing of these systems subjected to the Pacoima 1971 S16E, Parkfield 1966 N65E, and El Centro 1979 N-S earthquakes was undertaken and the results are presented in Appendix D.

These results are summarised below.

(a) Experimental test results

A summary of the maximum bending moments obtained by experimental testing of the unisolated and PRSIS systems is presented in Table 8.2.

**Table 8.2** Summary of experimental test results

Earthquake Input	Experimental Test Results Structure Bending Moment (N-m)		
	Unisolated Structure	Isolated Structure	Percentage ratio: isolated/unisolated bending moment
El Centro 1940 N-S (100%)	2519 N-m	1597 N-m	<b>63.4 %</b>
Pacoima 1971 S16E (33.3%)	2491 N-m	779 N-m	<b>31.3 %</b>
Parkfield 1966 N65E (70%)	1807 N-m	1836 N-m	<b>101.6 %</b>
El Centro 1979 N-S (80%)	1282 N-m	819 N-m	<b>63.9 %</b>

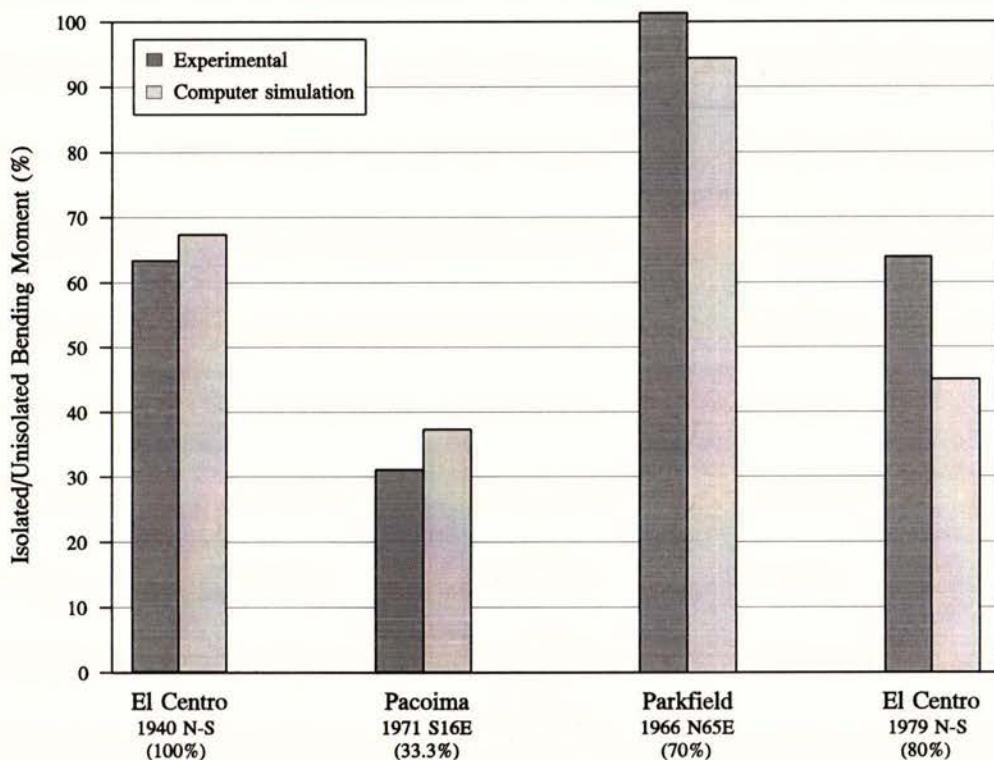
(b) Computer simulation results

A summary of the maximum bending moments given by computer simulation of the unisolated and PRSIS systems is presented in Table 8.3.

**Table 8.3** Summary of computer simulation results

Earthquake Input	Computer Simulation Results Structure Bending Moment (N-m)		
	Unisolated Structure	Isolated Structure	Percentage ratio: isolated/unisolated bending moment
El Centro 1940 N-S (100%)	1920 N-m	1295 N-m	67.4 %
Pacoima 1971 S16E (33.3%)	1889 N-m	708 N-m	37.5 %
Parkfield 1966 N65E (70%)	1549 N-m	1461 N-m	94.3 %
El Centro 1979 N-S (80%)	1597 N-m	718 N-m	45.0 %

The results given in Tables 8.2 and 8.3 are presented in Figure 8.21. This gives an overall summary of the experimental and computer simulation results.

**Figure 8.21** Comparison of isolated/unisolated structure bending moment: experimental and computer simulation results

These results confirm the effectiveness of the PRSIS for structures that have fundamental periods in the range of dominant earthquake spectral accelerations. For the El Centro 1940 N-S, Pacoima Dam 1971 S16E, and the El Centro 1979 N-S earthquakes, structure bending moments are significantly reduced.

Results for the Parkfield 1966 N65E earthquake show that there is virtually no difference in the bending moments for the isolated and unisolated systems. This is because the fundamental period of the unisolated system is outside the region of dominant spectral accelerations for this earthquake (c.f. the response spectra shown in Figure 7.8).

## **8.6 SUMMARY**

An innovative pivotal rocking seismic isolation systems for protecting columnar structures during earthquakes has been built and tested. Experimental tests have been conducted on a system with an unisolated fundamental period of 0.280 seconds.

Test results show that the PRSIS is effective in reducing structural loads and is able to provide significantly improved protection for structures that have fundamental periods in the range of dominant earthquake spectral accelerations.

## CONCLUSIONS AND RECOMMENDATIONS

### 9.1 CONCLUSIONS

This research project has examined the characteristics, design requirements, and dynamic response of ring spring systems. Based on this and a comprehensive study of the aseismic design methods used in engineering design, an innovative seismic isolation system incorporating ring springs was developed and subsequently tested. The key findings of this research are presented below.

- Aseismic design methods can be broadly classified into two principal techniques: strength design and response control. Of the response control techniques, seismic isolation is a simple yet effective method of reducing structural loads during earthquakes.
- Ring springs are friction devices capable of absorbing large amounts of input energy; these devices have application where a high level of damping is required. Practical use of ring springs requires: lubrication of the ring elements; either internal or external guidance of the spring stack, and a minimum pre-tension force of at least 5–10% of the maximum load capacity of the spring. Based upon their free vibration response, practical ring springs have damping factors of between 11.9% and 17.6%.
- Experimental testing of a prototype bi-directional ring spring cartridge showed its hysteresis characteristics comprise of four stiffnesses:  $K_0$ ,  $K_i$ ,  $K_l$ , and  $K_d$ .
- Mass/ring spring systems may be pre-displaced, pre-loaded, or pre-displaced pre-loaded. The theoretical free vibration response of pre-displaced mass/ring spring systems showed that motion proceeds around the hysteresis diagram until settling onto the origin stiffness ( $K_0$ ); for pre-loaded systems, motion proceeds about an equilibrium line equal to the pre-load force. A zone is set up within which motion eventually settles.

- The theoretical behaviour of pre-displaced mass/ring spring systems was verified by experimental tests. These test results showed that the ring spring hysteresis characteristics behaved in accordance with those given by quasi-static testing.
- Ring springs are possible candidates for use in earthquake-resistant applications. They are deemed especially suitable for use in rocking/stepping seismic isolation systems. To investigate this option, a pivotal rocking seismic isolation system (PRSIS) was developed.
- Computer analyses for a range of isolated and unisolated systems subjected to six different earthquake inputs, showed that PRSIS's can significantly reduce structural loads during short period type earthquakes - in some cases by a factor of between three and four.
- Shaker table tests conducted on a PRSIS showed that the design is effective in reducing structural loads during short period type earthquakes.
- An application seen as ideal for use of PRSIS's is in protecting high voltage electrical equipment during earthquakes. Though this is one specific situation, the isolation system has potential application to protecting a wide range of columnar type structures during earthquakes; for larger systems ground support of the structure weight may be required. Possible candidates include multi-storey buildings, chimneys, towers, as well as other similarly proportioned structures.

## 9.2 RECOMMENDATIONS FOR FURTHER RESEARCH

The work outlined in this thesis has established a basis from which further research can be undertaken. Suggested areas for additional work are as follows.

- Design guidelines outlining steps in the selection of suitable ring springs for specific applications are required. Design charts may be useful for presenting this information.
- RUAUMOKO is at present being expanded to enable dynamic modelling of non-

linear three-dimensional systems. When fully developed, a study should be undertaken to investigate the effect of simultaneously applying two horizontal earthquake components to a structure. Further, inputs not in line with the axes of the PRSIS should be considered.

- A study is required that compares the performance of PRSIS's incorporating ring springs with other types of isolation systems and devices, for example, Belleville washers. RUAUMOKO may be a suitable software package for undertaking computer simulation analyses.
- The option of utilising a PRSIS consisting of three cartridges attached to the base should be investigated; this design could provide economic advantages over a system that uses four cartridges. Also, rather than a single support for the pivot mechanism, multi-support configurations should be explored. These systems may be suitable for support of larger structures.

## REFERENCES

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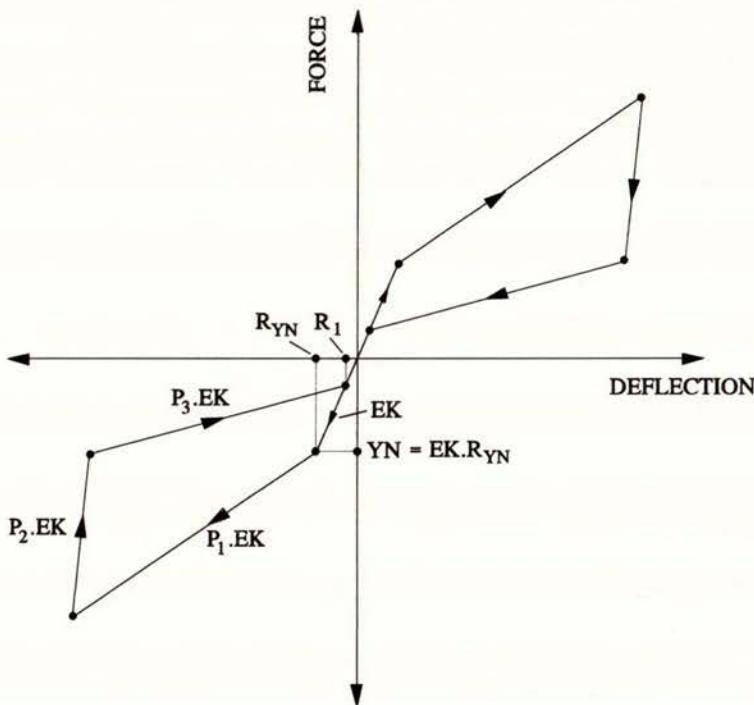
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## Appendix A

**RING SPRING MODEL COMPUTER ALGORITHM**

The computer algorithm for the ring spring model, written in FORTRAN and used in RUAUMOKO (Carr 1995), is listed below. Figure A.1 shows the variables used in the algorithm.



**Figure A.1** Ring spring force/deflection diagram

```

SUBROUTINE RINGSP(S,R,DR,XS,F,EK,P1,P2,P3,RINIT,RBP,RBN,KTYPE)
C ****
C RING SPRING MODEL
C S = Spring force
C R = Spring displacement
C DR = Current displacement increment
C XS = Excess force
C F = Current stiffness factor, current stiffness = EK*F
C EK = Initial elastic stiffness
C P1 = Stiffness factor, loading beyond "yield" displacement
C P2 = Stiffness factor, unloading "steep"
C P3 = Stiffness factor, unloading "lower"
C RINIT = Initial pre-displacement
C RBP,RBN = Maximum displacements positive & negative
C KTYPE = 0: Uni-directional; =1: Bi-directional
C YP,YN = "Yield" forces positive & negative
C ****

```

```

INTEGER KTYPE
CS  REAL S,R,DR,XS,F,EK,P1,P2,P3,RINIT,RBP,RBN
CS  REAL R0,R1,RYP,RYN,YP,YN,RNEW,TOL
CD  DOUBLE PRECISION S,R,DR,XS,F,EK,P1,P2,P3,RINIT,RBP,RBN
CD  DOUBLE PRECISION R0,R1,RYP,RYN,YP,YN,RNEW,TOL
DATA          TOL/0.0001/
C
XS = 0.0
IF(ABS(DR).LE.0.0) GO TO 100
RINIT = ABS(RINIT)
R0 = ((P3-P1)/(1.0-P3)+P1)*RINIT
R1 = -R0
RYP = P1*RINIT
RYN = -RYP
YP = EK*RYP
YN = -YP
XS = S+F*EK*DR
RNEW = R+DR
IF(RNEW.GT.R0) THEN
  IF(RBP.GT.RYP) THEN
    IF(DR.GT.0.0) THEN
      IF(XS.GE.YP+P1*EK*(RNEW-RYP)-TOL) THEN
        F = P1
        S = YP+P1*EK*(RNEW-RYP)
      ELSEIF(RNEW.GE.R0.AND.RNEW.LE.RYP) THEN
        F = 1.0
        S = EK*RNEW
      ELSE
        F = P2
        S = S+DR*P2*EK
      ENDIF
    ELSE
      IF(XS.LE.EK*R0+P3*EK*(RNEW-R0)+TOL) THEN
        F = P3
        S = EK*R0+P3*EK*(RNEW-R0)
      ELSEIF(RNEW.GE.R0.AND.RNEW.LE.RYP) THEN
        F = 1.0
        S = EK*RNEW
      ELSE
        F = P2
        S = S+P2*EK*DR
      ENDIF
    ENDIF
  ELSE
    IF(RNEW.GE.RYP) THEN
      F = P1
      S = YP+P1*EK*(RNEW-RYP)
    ELSE
      F = 1.0
      S = EK*RNEW
    ENDIF
  ENDIF
ELSEIF(RNEW.LT.R1.AND.KTYPE.GT.0) THEN
  IF(RBN.LT.RYN) THEN
    IF(DR.LT.0.0) THEN
      IF(XS.LE.YN+P1*EK*(RNEW-RYN)+TOL) THEN
        F = P1
        S = YN+P1*EK*(RNEW-RYN)
      ELSE
        F = 1.0
        S = EK*RNEW
      ENDIF
    ENDIF
  ENDIF
ENDIF

```

```
ELSEIF(RNEW.LE.R1.AND.RNEW.GE.RYN) THEN
    F = 1.0
    S = EK*RNEW
ELSE
    F = P2
    S = S+DR*EK*P2
ENDIF
ELSE
IF(XS.GE.EK*R1+P3*EK*(RNEW-R1)-TOL) THEN
    F = P3
    S = EK*R1+P3*EK*(RNEW-R1)
ELSEIF(RNEW.LE.R1.AND.RNEW.GE.RYN) THEN
    F = 1.0
    S = EK*RNEW
ELSE
    F = P2
    S = S+DR*P2*EK
ENDIF
ENDIF
ELSE
IF(RNEW.LE.RYN) THEN
    F = P1
    S = YN+P1*EK*(RNEW-RYN)
ELSE
    F = 1.0
    S = EK*RNEW
ENDIF
ENDIF
ELSE
IF(RNEW.GE.0.0.OR.(RNEW.LT.0.0.AND.KTYPE.GT.0)) THEN
    F = 1.0
    S = EK*RNEW
ELSEIF(RNEW.LT.0.0) THEN
    F = 0.0
    S = 0.0
ENDIF
ENDIF
R = RNEW
XS = XS-S
RBP = MAX(R,RBP)
RBN = MIN(R,RBN)
C
100 RETURN
END
```

## Appendix B

### **SCALED GENERAL ASSEMBLY DRAWINGS**

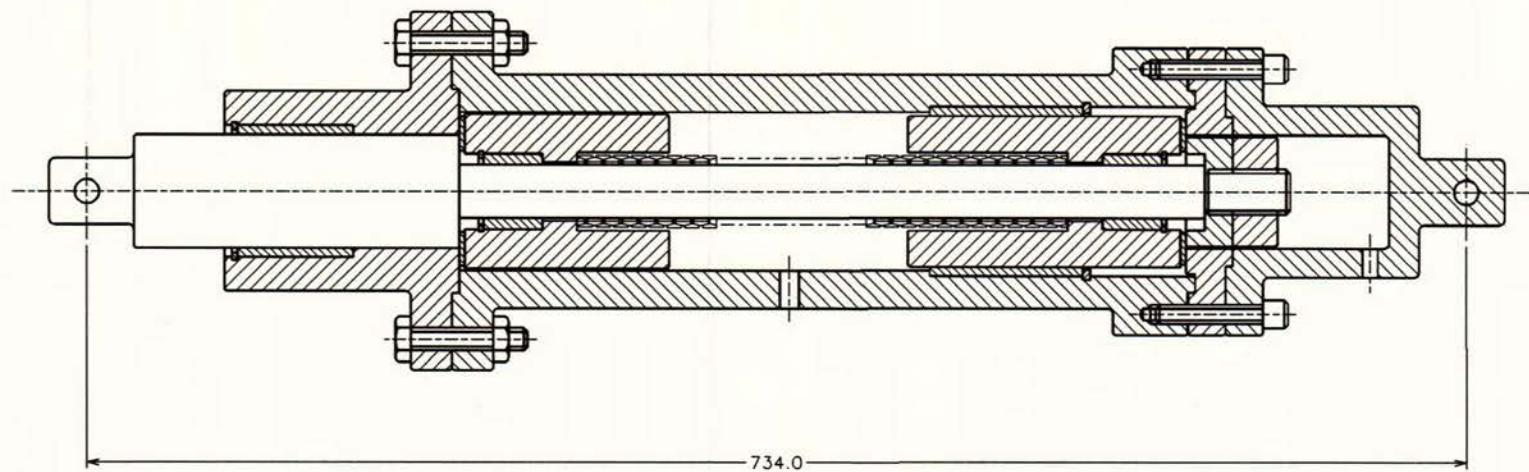
This appendix presents four scaled general assembly drawings. These are:

Drawing No. 1: Bi-directional ring spring cartridge

Drawing No. 2: Uni-directional ring spring cartridge

Drawing No. 3: Pivotal rocking seismic isolation system (PRSIS)

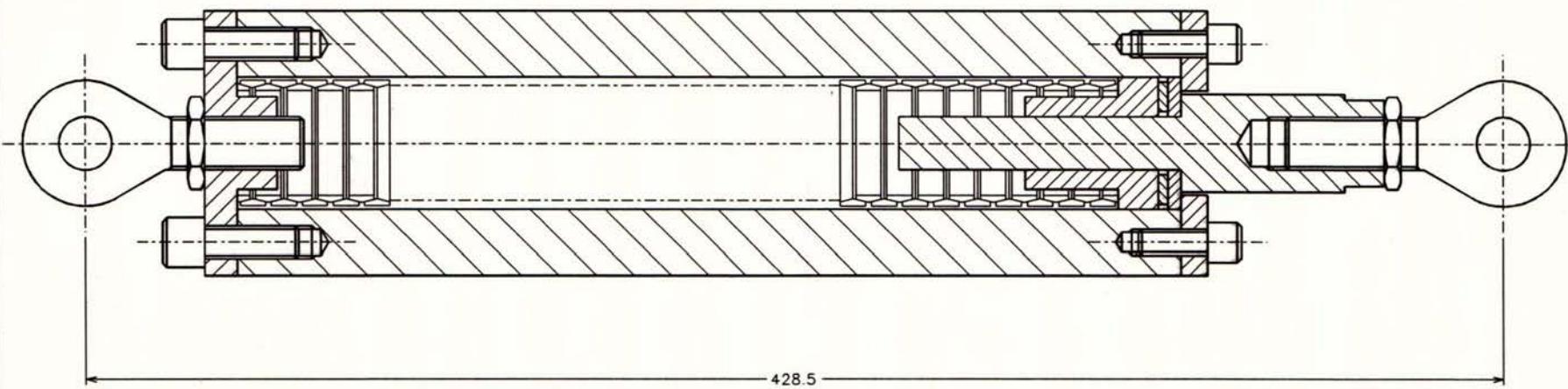
Drawing No. 4: Complete test rig assembly



BI-DIRECTIONAL RING SPRING CARTRIDGE

MECHANICAL ENGINEERING DEPARTMENT  
UNIVERSITY OF CANTERBURY

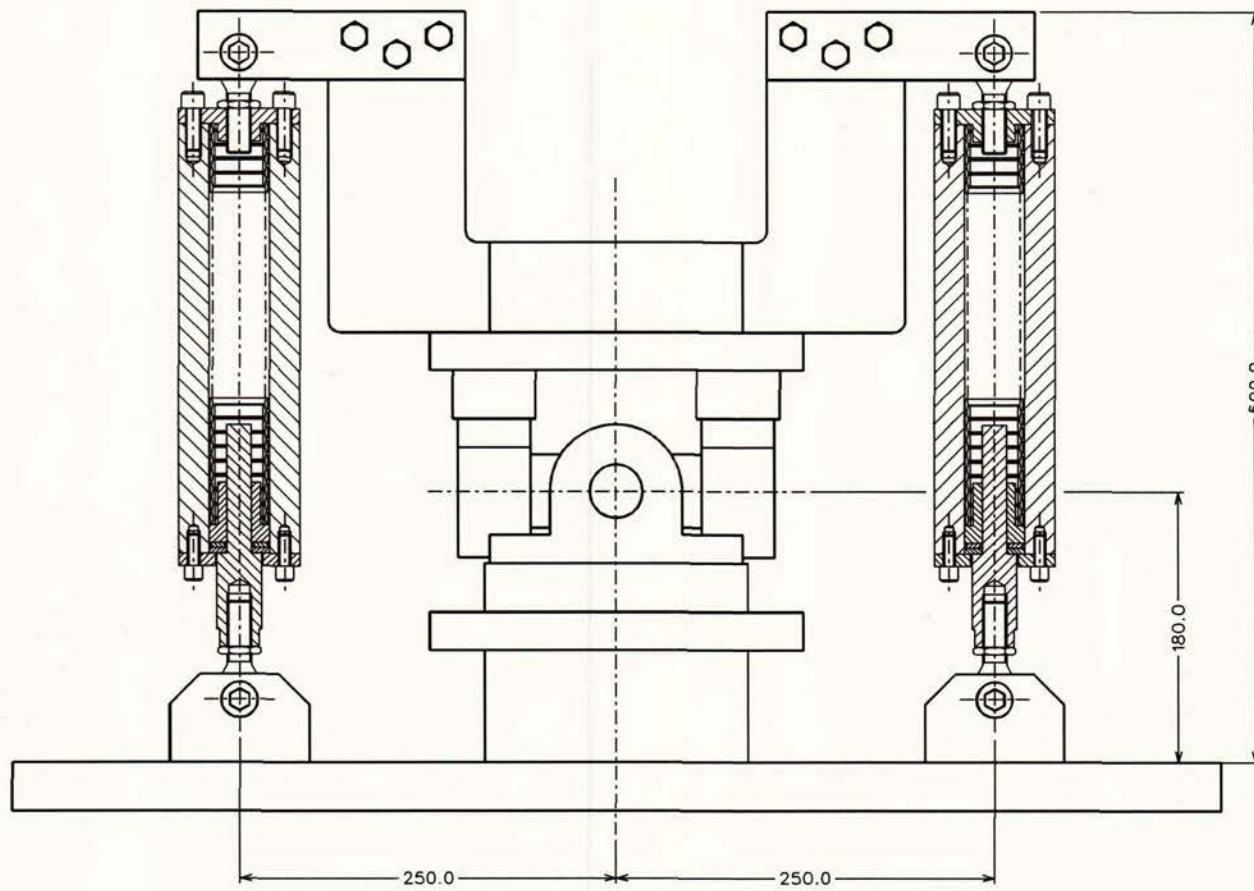
DESIGN: K E HILL	SCALE: 1:4	DRG No: 1
DRAWN: K E HILL	DIMENSIONS IN MM	



UNI-DIRECTIONAL RING SPRING CARTRIDGE

MECHANICAL ENGINEERING DEPARTMENT  
UNIVERSITY OF CANTERBURY

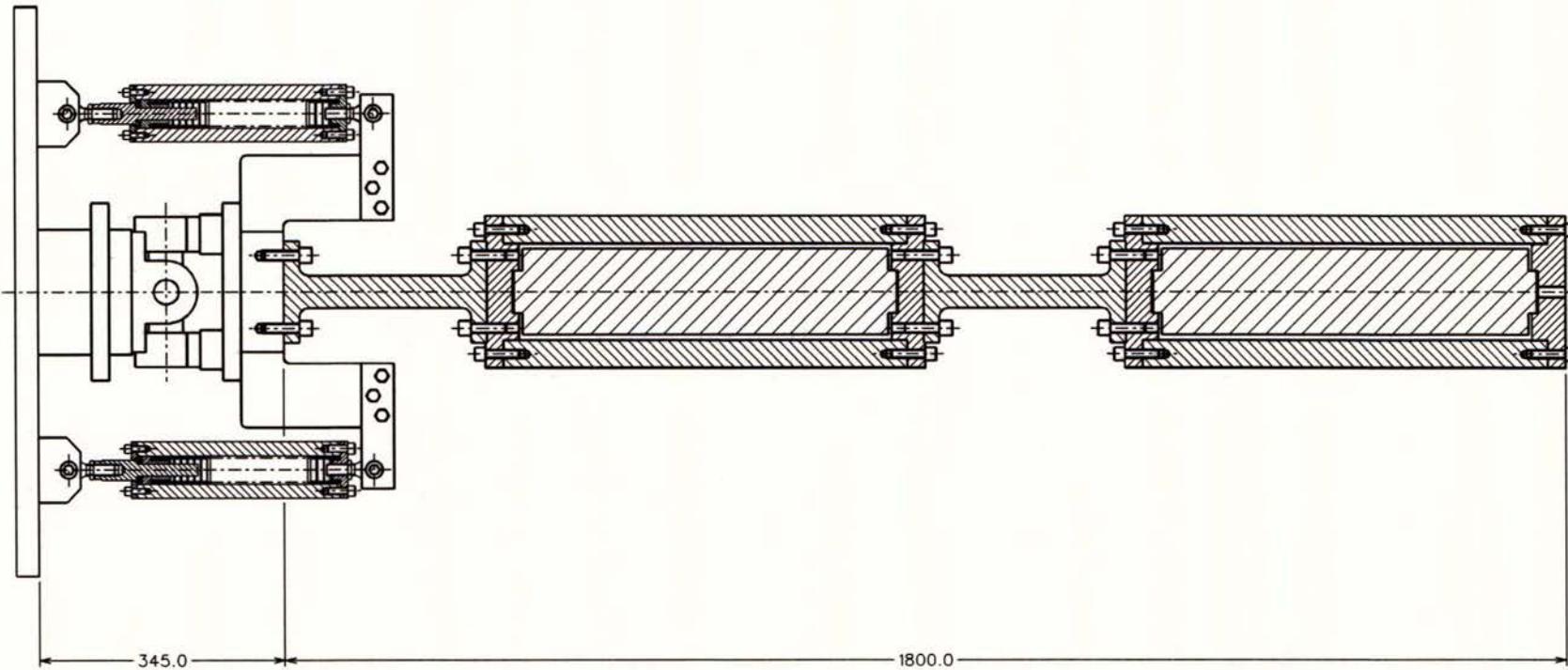
DESIGN: K E HILL	SCALE: 1:2	DRG No: 2
DRAWN: K E HILL	DIMENSIONS IN MM	



PIVOTAL ROCKING SEISMIC ISOLATION SYSTEM

MECHANICAL ENGINEERING DEPARTMENT  
UNIVERSITY OF CANTERBURY

DESIGN: K E HILL	SCALE: 1:5	DRG No: 3
DRAWN: K E HILL	DIMENSIONS IN MM	



COMPLETE TEST RIG ASSEMBLY

MECHANICAL ENGINEERING DEPARTMENT  
UNIVERSITY OF CANTERBURY

DESIGN: K E HILL	SCALE: 1:10	DRG No: 4
DRAWN: K E HILL	DIMENSIONS IN MM	

## Appendix C

**TABULAR DATA FOR COMPUTER SIMULATION ANALYSES**

**Table C.1** Maximum structure bending moments for isolated and unisolated systems for  
 $T_{1\text{unis}} = 0.125$  sec (PRYSIS: 0% pre-displacement and 0% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1\text{unis}} = 0.125$ s $T_{2\text{unis}} = 0.019$ s	Isolated Structure $T_{1\text{is}} = 1.006$ s $T_{2\text{is}} = 0.022$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	5406 N-m @ 2.61 s	1214 N-m @ 2.91 s	<b>22.5 %</b>
Pacoima 1971 S16E	15530 N-m @ 7.75 s	4626 N-m @ 3.55 s	<b>29.8 %</b>
Parkfield 1966 N65E	4612 N-m @ 4.21 s	2583 N-m @ 4.97 s	<b>56.0 %</b>
El Centro 1979 N-S	3366 N-m @ 6.82 s	1590 N-m @ 6.62 s	<b>47.2 %</b>
Bucharest 1977 N-S	1692 N-m @ 3.99 s	2520 N-m @ 4.30 s	<b>148.9 %</b>
Mexico 1985 (SCT) N90W	1255 N-m @ 58.11 s	2012 N-m @ 59.08 s	<b>160.3 %</b>

**Table C.2** Maximum structure bending moments for isolated and unisolated systems for  
 $T_{1\text{unis}} = 0.2$  sec (PRYSIS: 0% pre-displacement and 0% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1\text{unis}} = 0.200$ s $T_{2\text{unis}} = 0.031$ s	Isolated Structure $T_{1\text{is}} = 1.018$ s $T_{2\text{is}} = 0.034$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	5608 N-m @ 2.69 s	1497 N-m @ 2.94 s	<b>26.7 %</b>
Pacoima 1971 S16E	19280 N-m @ 9.31 s	5607 N-m @ 3.55 s	<b>29.1 %</b>
Parkfield 1966 N65E	3794 N-m @ 3.80 s	3020 N-m @ 4.97 s	<b>79.6 %</b>
El Centro 1979 N-S	5587 N-m @ 8.93 s	1812 N-m @ 6.63 s	<b>32.4 %</b>
Bucharest 1977 N-S	2044 N-m @ 3.99 s	3026 N-m @ 4.28 s	<b>148.0 %</b>
Mexico 1985 (SCT) N90W	1328 N-m @ 58.83 s	2359 N-m @ 59.06 s	<b>177.6 %</b>

**Table C.3** Maximum structure bending moments for isolated and unisolated systems for  $T_{1unis} = 0.5$  sec (PRYSIS: 0% pre-displacement and 0% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1unis} = 0.500$ s $T_{2unis} = 0.076$ s	Isolated Structure $T_{1is} = 1.116$ s $T_{2is} = 0.083$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	6369 N-m @ 2.38 s	1665 N-m @ 3.00 s	<b>26.1 %</b>
Pacoima 1971 S16E	13550 N-m @ 8.86 s	6573 N-m @ 3.58 s	<b>48.5 %</b>
Parkfield 1966 N65E	10510 N-m @ 4.34 s	3394 N-m @ 4.99 s	<b>32.3 %</b>
El Centro 1979 N-S	6136 N-m @ 9.04 s	2090 N-m @ 9.92 s	<b>34.1 %</b>
Bucharest 1977 N-S	3946 N-m @ 3.97 s	3657 N-m @ 4.28 s	<b>92.7 %</b>
Mexico 1985 (SCT) N90W	1929 N-m @ 64.88 s	2349 N-m @ 58.02 s	<b>121.8 %</b>

**Table C.4** Maximum structure bending moments for isolated and unisolated systems for  $T_{1unis} = 0.125$  sec (PRYSIS: 16% pre-displacement and 0% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1unis} = 0.125$ s $T_{2unis} = 0.019$ s	Isolated Structure $T_{1is} = 1.006$ s $T_{2is} = 0.022$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	5406 N-m @ 2.61 s	2526 N-m @ 2.22 s	<b>46.7 %</b>
Pacoima 1971 S16E	15530 N-m @ 7.75 s	4293 N-m @ 3.47 s	<b>27.6 %</b>
Parkfield 1966 N65E	4612 N-m @ 4.21 s	3312 N-m @ 4.89 s	<b>71.8 %</b>
El Centro 1979 N-S	3366 N-m @ 6.82 s	2037 N-m @ 9.10 s	<b>60.5 %</b>
Bucharest 1977 N-S	1692 N-m @ 3.99 s	1920 N-m @ 4.02 s	<b>113.5 %</b>
Mexico 1985 (SCT) N90W	1255 N-m @ 58.11 s	1354 N-m @ 58.98 s	<b>107.9 %</b>

**Table C.5** Maximum structure bending moments for isolated and unisolated systems for  $T_{1unis} = 0.2$  sec (PRYSIS: 16% pre-displacement and 0% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1unis} = 0.200$ s $T_{2unis} = 0.031$ s	Isolated Structure $T_{1is} = 1.018$ s $T_{2is} = 0.034$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	5608 N-m @ 2.69 s	2750 N-m @ 2.24 s	<b>49.0 %</b>
Pacoima 1971 S16E	19280 N-m @ 9.31 s	4859 N-m @ 3.48 s	<b>25.2 %</b>
Parkfield 1966 N65E	3794 N-m @ 3.80 s	3683 N-m @ 4.91 s	<b>97.1 %</b>
El Centro 1979 N-S	5587 N-m @ 8.93 s	2214 N-m @ 9.13 s	<b>39.6 %</b>
Bucharest 1977 N-S	2044 N-m @ 3.99 s	2203 N-m @ 4.07 s	<b>107.8 %</b>
Mexico 1985 (SCT) N90W	1328 N-m @ 58.83 s	1421 N-m @ 61.20 s	<b>107.0 %</b>

**Table C.6** Maximum structure bending moments for isolated and unisolated systems for  $T_{1unis} = 0.5$  sec (PRYSIS: 16% pre-displacement and 0% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1unis} = 0.500$ s $T_{2unis} = 0.076$ s	Isolated Structure $T_{1is} = 1.116$ s $T_{2is} = 0.083$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	6369 N-m @ 2.38 s	2711 N-m @ 2.79 s	<b>42.6 %</b>
Pacoima 1971 S16E	13550 N-m @ 8.86 s	5725 N-m @ 3.52 s	<b>42.3 %</b>
Parkfield 1966 N65E	10510 N-m @ 4.34 s	3669 N-m @ 3.93 s	<b>34.9 %</b>
El Centro 1979 N-S	6136 N-m @ 9.04 s	2139 N-m @ 9.70 s	<b>34.9 %</b>
Bucharest 1977 N-S	3946 N-m @ 3.97 s	2545 N-m @ 4.14 s	<b>64.5 %</b>
Mexico 1985 (SCT) N90W	1929 N-m @ 64.88 s	1648 N-m @ 62.53 s	<b>85.4 %</b>

**Table C.7** Maximum structure bending moments for isolated and unisolated systems for  $T_{1unis} = 0.125$  sec (PRYSIS: 16% pre-displacement and 10% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1unis} = 0.125$ s $T_{2unis} = 0.019$ s	Isolated Structure $T_{1is} = 1.006$ s $T_{2is} = 0.022$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	5406 N-m @ 2.61 s	2538 N-m @ 2.23 s	<b>46.9 %</b>
Pacoima 1971 S16E	15530 N-m @ 7.75 s	4302 N-m @ 3.47 s	<b>27.7 %</b>
Parkfield 1966 N65E	4612 N-m @ 4.21 s	3263 N-m @ 4.88 s	<b>70.8 %</b>
El Centro 1979 N-S	3366 N-m @ 6.82 s	2046 N-m @ 9.10 s	<b>60.8 %</b>
Bucharest 1977 N-S	1692 N-m @ 3.99 s	2051 N-m @ 4.05 s	<b>121.2 %</b>
Mexico 1985 (SCT) N90W	1255 N-m @ 58.11 s	1232 N-m @ 58.94 s	<b>98.2 %</b>

**Table C.8** Maximum structure bending moments for isolated and unisolated systems for  $T_{1unis} = 0.2$  sec (PRYSIS: 16% pre-displacement and 10% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1unis} = 0.200$ s $T_{2unis} = 0.031$ s	Isolated Structure $T_{1is} = 1.018$ s $T_{2is} = 0.034$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	5608 N-m @ 2.69 s	2751 N-m @ 2.24 s	<b>49.1 %</b>
Pacoima 1971 S16E	19280 N-m @ 9.31 s	4843 N-m @ 3.48 s	<b>25.1 %</b>
Parkfield 1966 N65E	3794 N-m @ 3.80 s	3710 N-m @ 4.91 s	<b>97.8 %</b>
El Centro 1979 N-S	5587 N-m @ 8.93 s	2196 N-m @ 9.13 s	<b>39.3 %</b>
Bucharest 1977 N-S	2044 N-m @ 3.99 s	2196 N-m @ 4.08 s	<b>107.4 %</b>
Mexico 1985 (SCT) N90W	1328 N-m @ 58.83 s	1267 N-m @ 58.97 s	<b>95.4 %</b>

**Table C.9** Maximum structure bending moments for isolated and unisolated systems for  $T_{1\text{unis}} = 0.5$  sec (PRYSIS: 16% pre-displacement and 10% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1\text{unis}} = 0.500$ s $T_{2\text{unis}} = 0.076$ s	Isolated Structure $T_{1\text{is}} = 1.116$ s $T_{2\text{is}} = 0.083$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	6369 N-m @ 2.38 s	2682 N-m @ 2.78 s	<b>42.1 %</b>
Pacoima 1971 S16E	13550 N-m @ 8.86 s	5764 N-m @ 3.52 s	<b>42.5 %</b>
Parkfield 1966 N65E	10510 N-m @ 4.34 s	3613 N-m @ 3.94 s	<b>34.4 %</b>
El Centro 1979 N-S	6136 N-m @ 9.04 s	2120 N-m @ 9.69 s	<b>34.6 %</b>
Bucharest 1977 N-S	3946 N-m @ 3.97 s	2540 N-m @ 4.15 s	<b>64.4 %</b>
Mexico 1985 (SCT) N90W	1929 N-m @ 64.88 s	1573 N-m @ 57.77 s	<b>81.5 %</b>

**Table C.10** Maximum structure bending moments for isolated and unisolated systems for  $T_{1\text{unis}} = 0.125$  sec (PRYSIS: 16% pre-displacement and 20% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1\text{unis}} = 0.125$ s $T_{2\text{unis}} = 0.019$ s	Isolated Structure $T_{1\text{is}} = 1.006$ s $T_{2\text{is}} = 0.022$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	5406 N-m @ 2.61 s	2211 N-m @ 2.25 s	<b>40.9 %</b>
Pacoima 1971 S16E	15530 N-m @ 7.75 s	4446 N-m @ 3.48 s	<b>28.6 %</b>
Parkfield 1966 N65E	4612 N-m @ 4.21 s	3098 N-m @ 4.88 s	<b>67.2 %</b>
El Centro 1979 N-S	3366 N-m @ 6.82 s	1050 N-m @ 6.60 s	<b>31.2 %</b>
Bucharest 1977 N-S	1692 N-m @ 3.99 s	2369 N-m @ 4.15 s	<b>140.0 %</b>
Mexico 1985 (SCT) N90W	1255 N-m @ 58.11 s	1892 N-m @ 58.03 s	<b>150.8 %</b>

**Table C.11** Maximum structure bending moments for isolated and unisolated systems for  $T_{1\text{unis}} = 0.2$  sec (PRYSIS: 16% pre-displacement and 20% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1\text{unis}} = 0.200$ s $T_{2\text{unis}} = 0.031$ s	Isolated Structure $T_{1\text{is}} = 1.018$ s $T_{2\text{is}} = 0.034$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	5608 N-m @ 2.69 s	2409 N-m @ 2.25 s	<b>43.0 %</b>
Pacoima 1971 S16E	19280 N-m @ 9.31 s	5019 N-m @ 3.48 s	<b>26.0 %</b>
Parkfield 1966 N65E	3794 N-m @ 3.80 s	3447 N-m @ 4.90 s	<b>90.9 %</b>
El Centro 1979 N-S	5587 N-m @ 8.93 s	1427 N-m @ 6.58 s	<b>25.5 %</b>
Bucharest 1977 N-S	2044 N-m @ 3.99 s	2484 N-m @ 4.16 s	<b>121.5 %</b>
Mexico 1985 (SCT) N90W	1328 N-m @ 58.83 s	2064 N-m @ 58.01 s	<b>155.4 %</b>

**Table C.12** Maximum structure bending moments for isolated and unisolated systems for  $T_{1\text{unis}} = 0.5$  sec (PRYSIS: 16% pre-displacement and 20% pre-load)

Earthquake Input	Maximum Structure Bending Moment (N-m)		
	Unisolated Structure $T_{1\text{unis}} = 0.500$ s $T_{2\text{unis}} = 0.076$ s	Isolated Structure $T_{1\text{is}} = 1.116$ s $T_{2\text{is}} = 0.083$ s	Percentage ratio: isolated/unisolated max. bending moment
El Centro 1940 N-S	6369 N-m @ 2.38 s	2329 N-m @ 2.81 s	<b>36.6 %</b>
Pacoima 1971 S16E	13550 N-m @ 8.86 s	5895 N-m @ 3.50 s	<b>43.5 %</b>
Parkfield 1966 N65E	10510 N-m @ 4.34 s	3455 N-m @ 3.95 s	<b>32.9 %</b>
El Centro 1979 N-S	6136 N-m @ 9.04 s	1700 N-m @ 7.08 s	<b>27.7 %</b>
Bucharest 1977 N-S	3946 N-m @ 3.97 s	2668 N-m @ 4.16 s	<b>67.6 %</b>
Mexico 1985 (SCT) N90W	1929 N-m @ 64.88 s	1919 N-m @ 58.04 s	<b>99.5 %</b>

## Appendix D

**ANALYSIS OF UNISOLATED AND PRSIS SYSTEMS -  
EXPERIMENTAL AND COMPUTER SIMULATION RESULTS**

Listed below is a summary of the figures given in this appendix. These graphs present experimental and computer simulation results for unisolated and PRSIS systems subjected to four short period type earthquakes.

**Figure D.1** Dynamic response of unisolated system: El Centro 1940 N-S (100%)

**Figure D.2** Dynamic response of unisolated system: Pacoima 1971 S16E (33.3%)

**Figure D.3** Dynamic response of unisolated system: Parkfield 1966 N65E (70%)

**Figure D.4** Dynamic response of unisolated system: El Centro 1979 N-S (80%)

**Figure D.5** Dynamic response of unisolated system including sub-system flexibility:

El Centro 1940 N-S (100%)

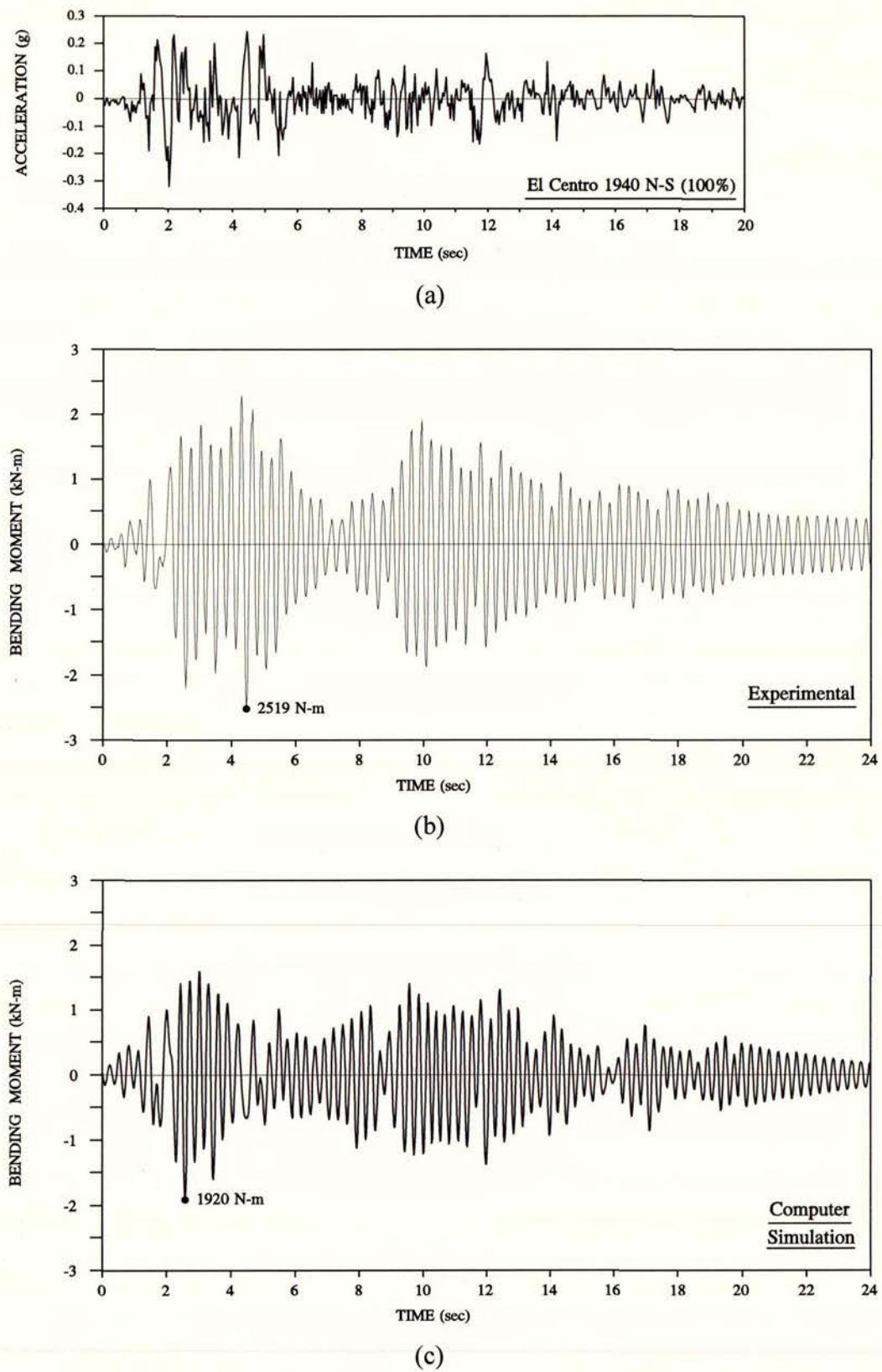
**Figures D.6 & D.7** Dynamic response of PRSIS at 0°: El Centro 1940 N-S (100%)

**Figures D.8 & D.9** Dynamic response of PRSIS at 0°: Pacoima 1971 S16E (33.3%)

**Figures D.10 & D.11** Dynamic response of PRSIS at 0°: Parkfield 1966 N65E (70%)

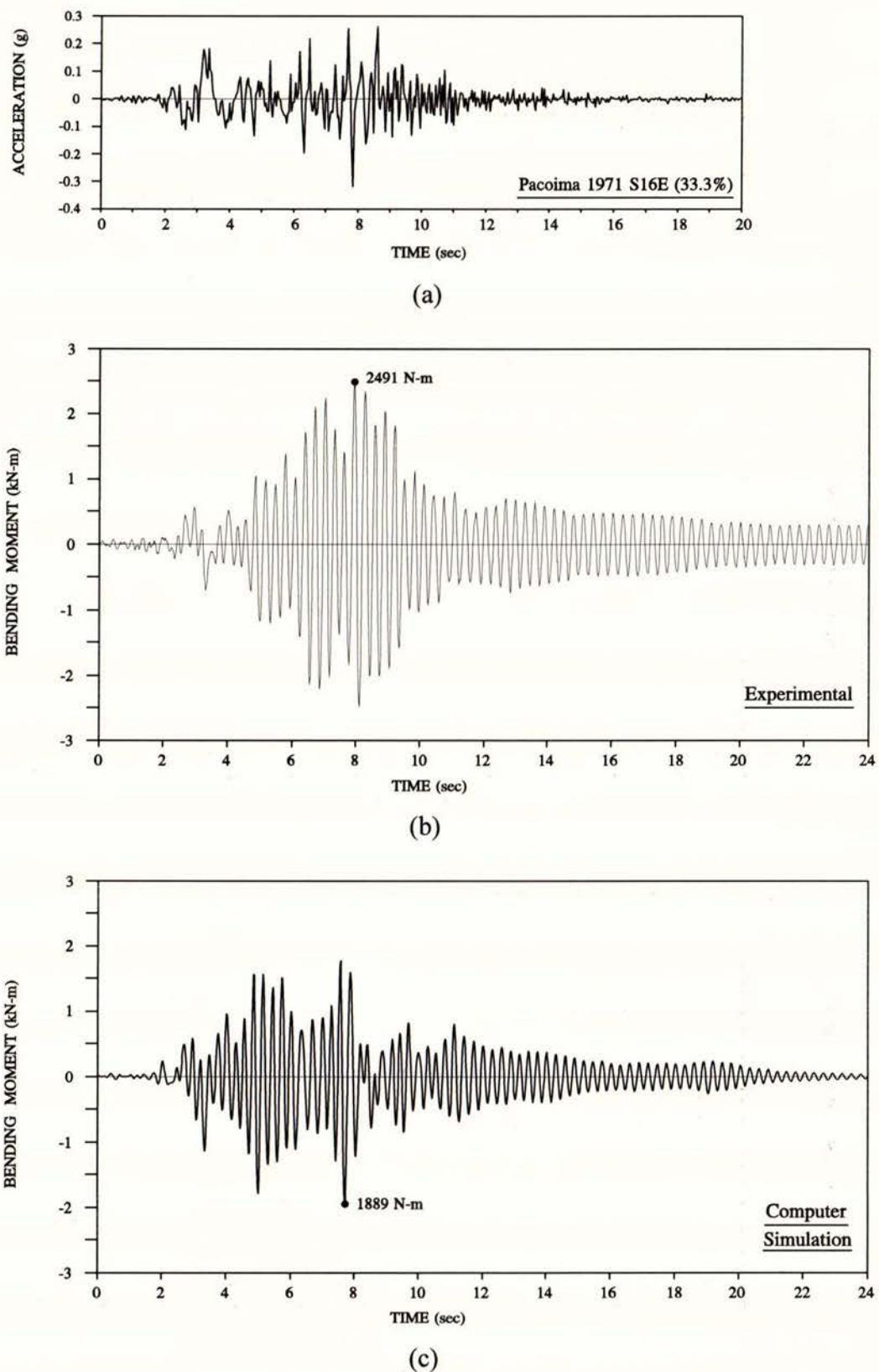
**Figures D.12 & D.13** Dynamic response of PRSIS at 0°: El Centro 1979 N-S (80%)

**Figures D.14 & D.15** Dynamic response of PRSIS at 45°: El Centro 1940 N-S (100%)



**Figure D.1** Dynamic response of unisolated system -

- Earthquake: El Centro 1940 N-S (100%)
- Bending moment time history (experimental)
- Bending moment time history (computer simulation)

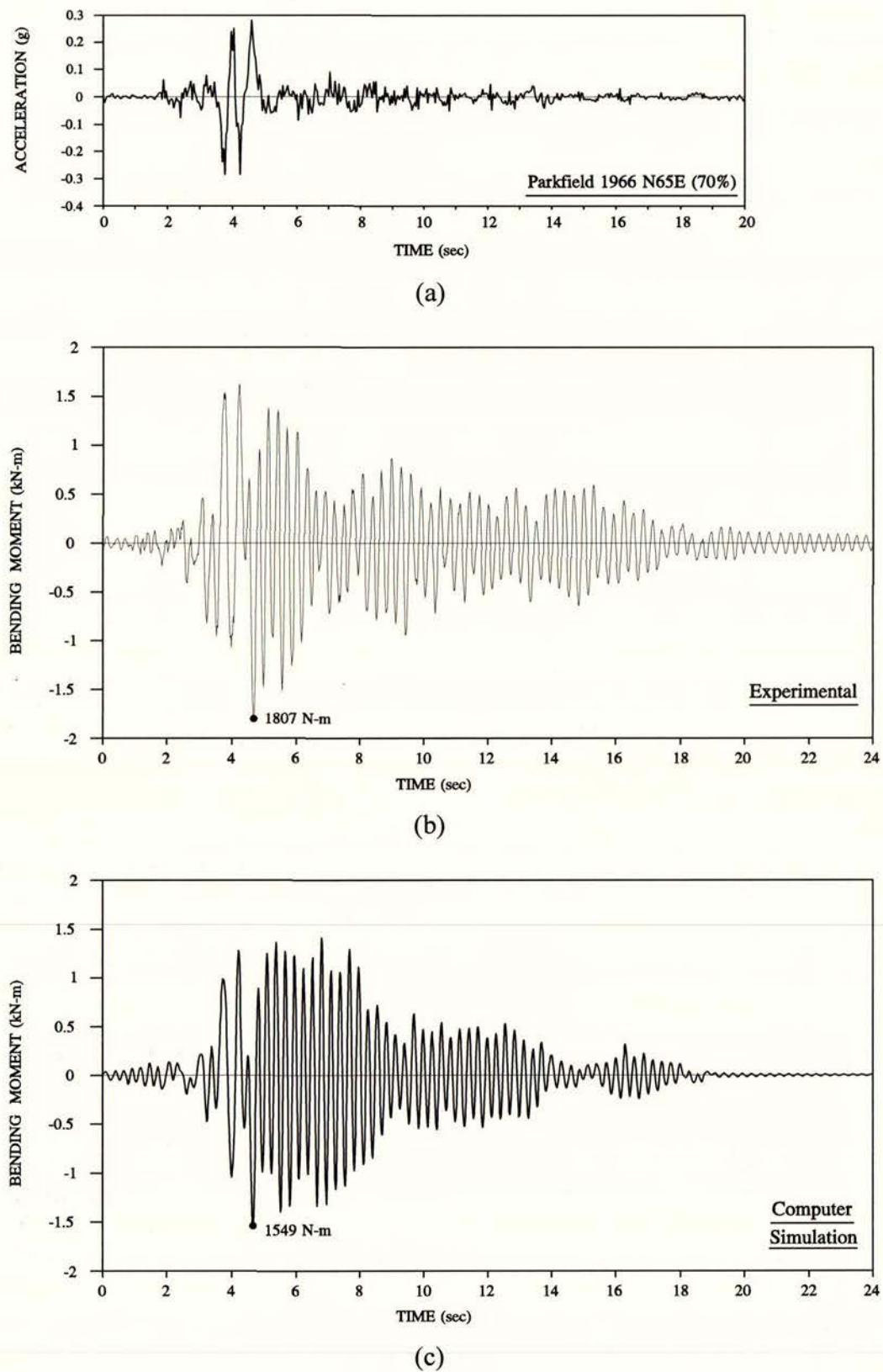


**Figure D.2** Dynamic response of unisolated system -

(a) Earthquake: Pacoima 1971 S16E (33.3%)

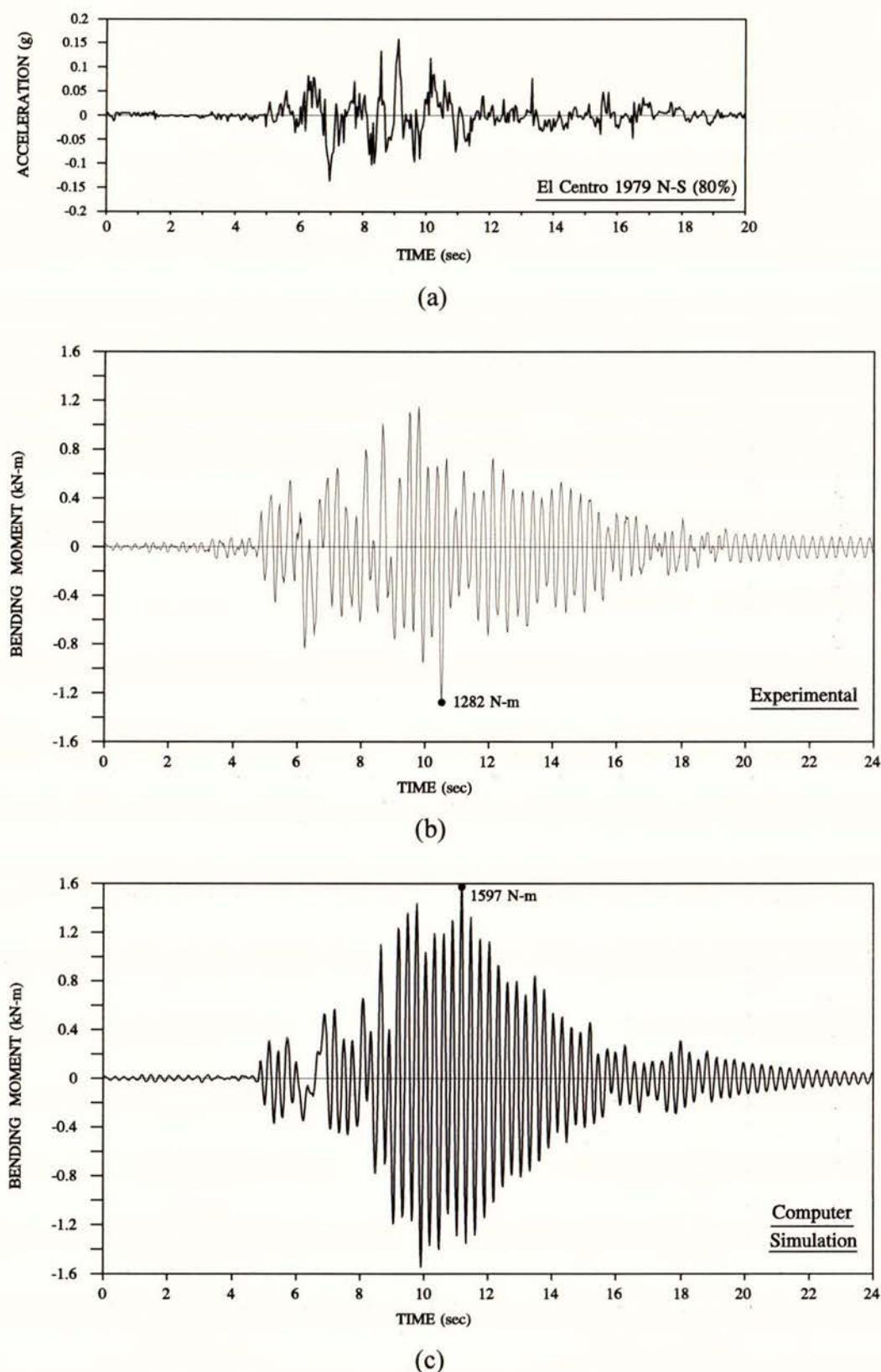
(b) Bending moment time history (experimental)

(c) Bending moment time history (computer simulation)



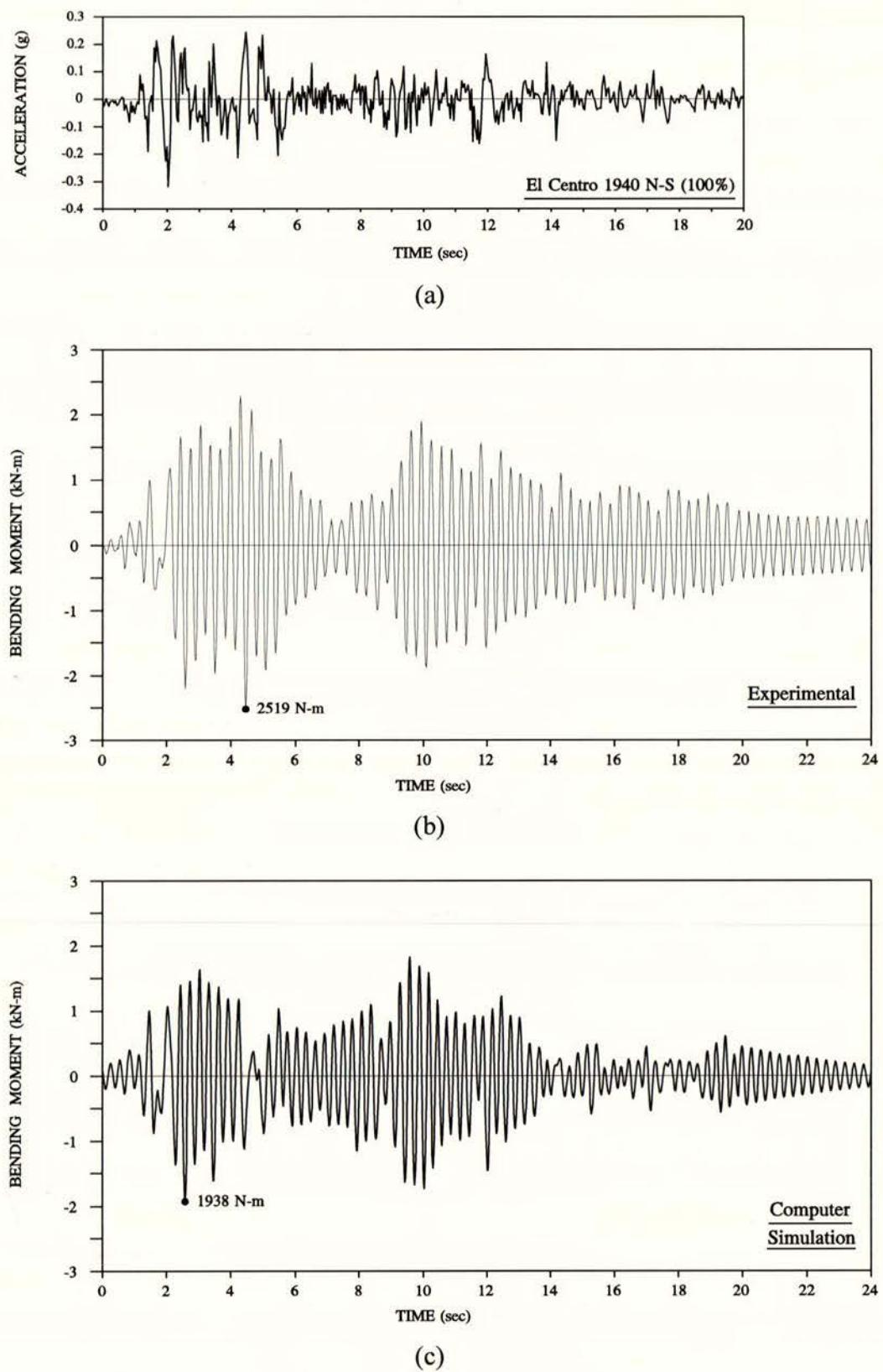
**Figure D.3** Dynamic response of unisolated system -

- (a) Earthquake: Parkfield 1966 N65E (70%)
- (b) Bending moment time history (experimental)
- (c) Bending moment time history (computer simulation)

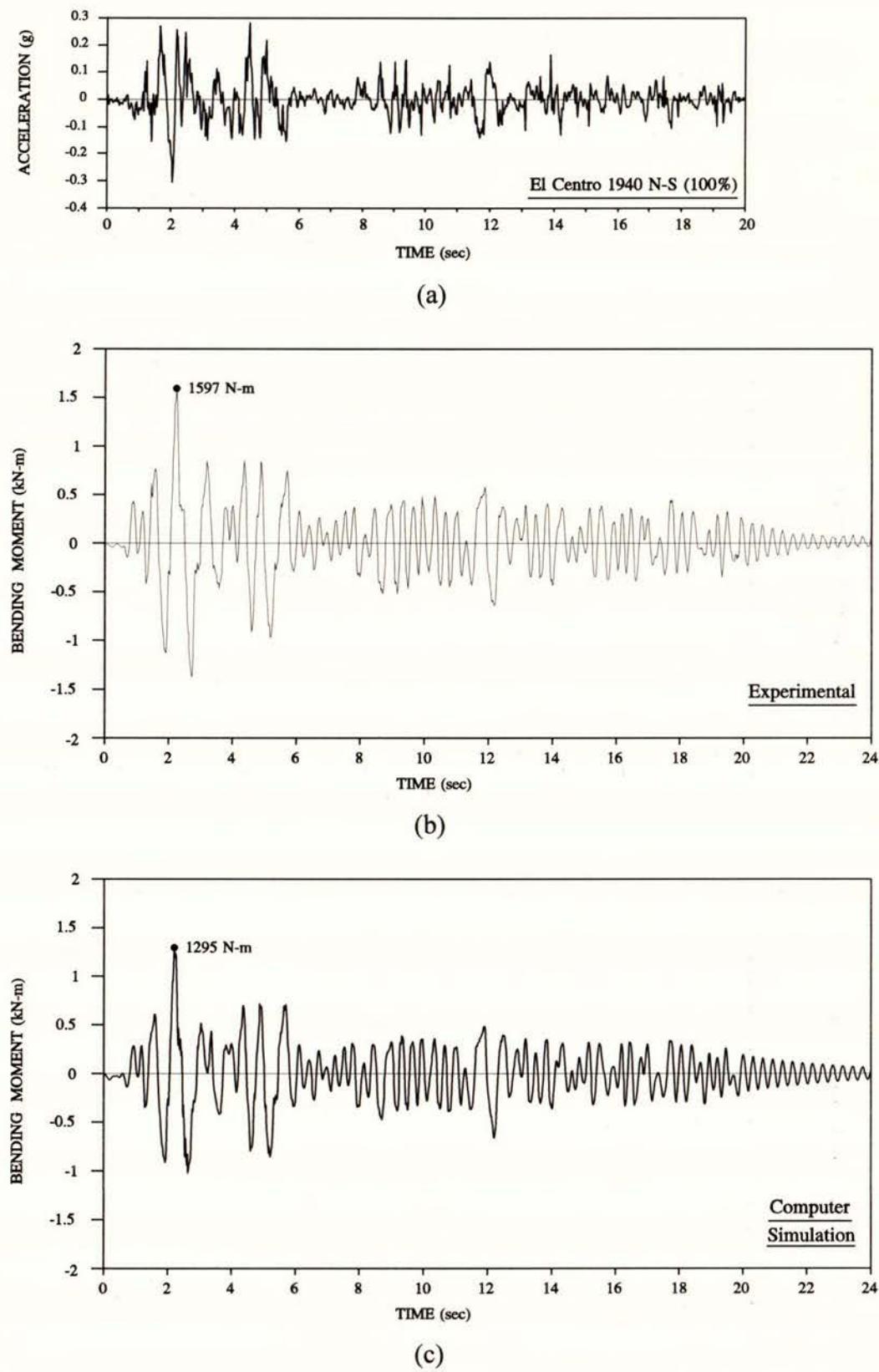


**Figure D.4** Dynamic response of unisolated system -

- (a) Earthquake: El Centro 1979 N-S (80%)
- (b) Bending moment time history (experimental)
- (c) Bending moment time history (computer simulation)



**Figure D.5** Dynamic response of unisolated system including sub-system flexibility -  
 (a) Earthquake: El Centro 1940 N-S (100%)  
 (b) Bending moment time history (experimental)  
 (c) Bending moment time history (computer simulation)

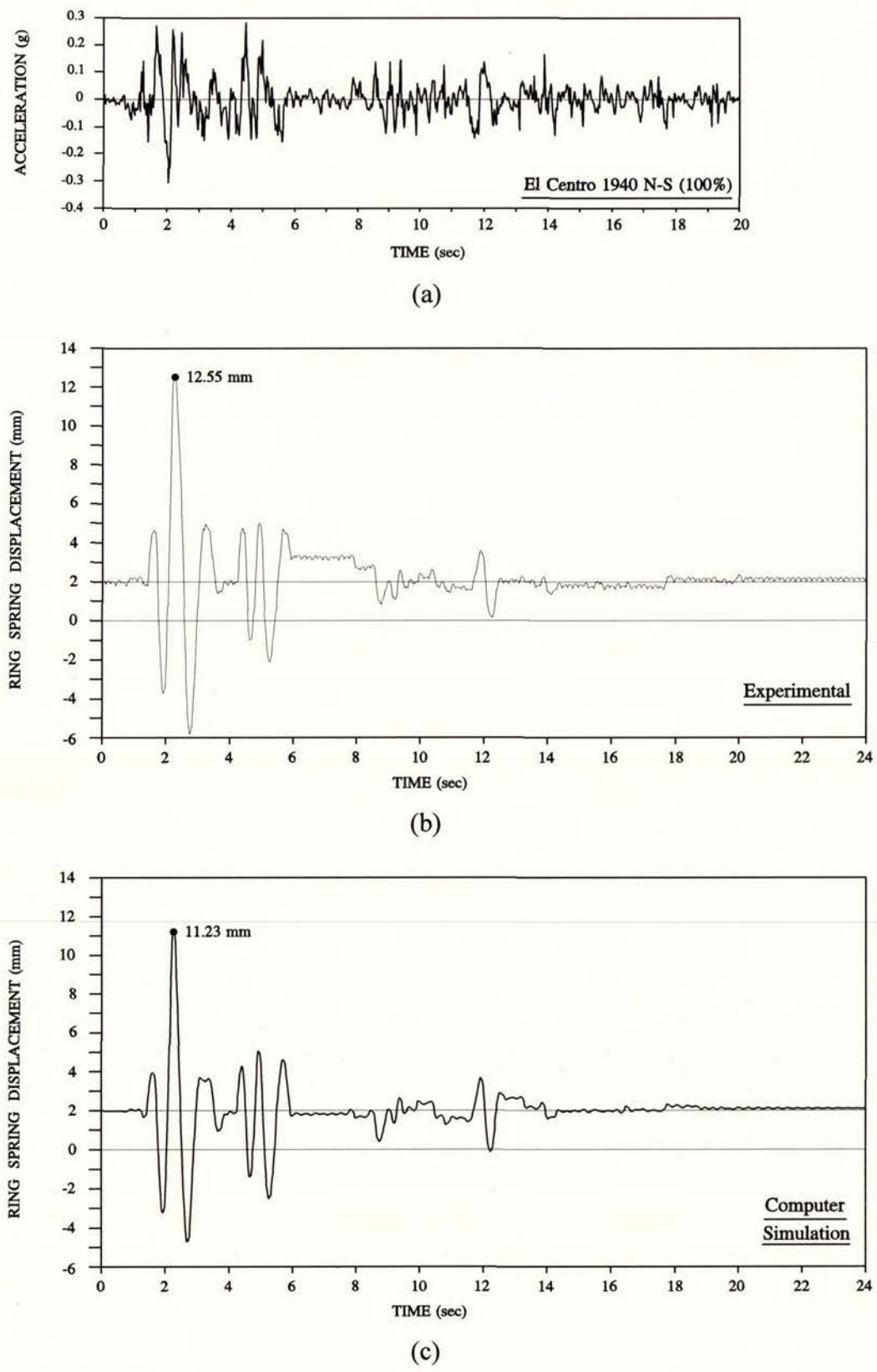


**Figure D.6** Dynamic response of PRSIS at  $0^\circ$  -

(a) Earthquake: El Centro 1940 N-S (100%)

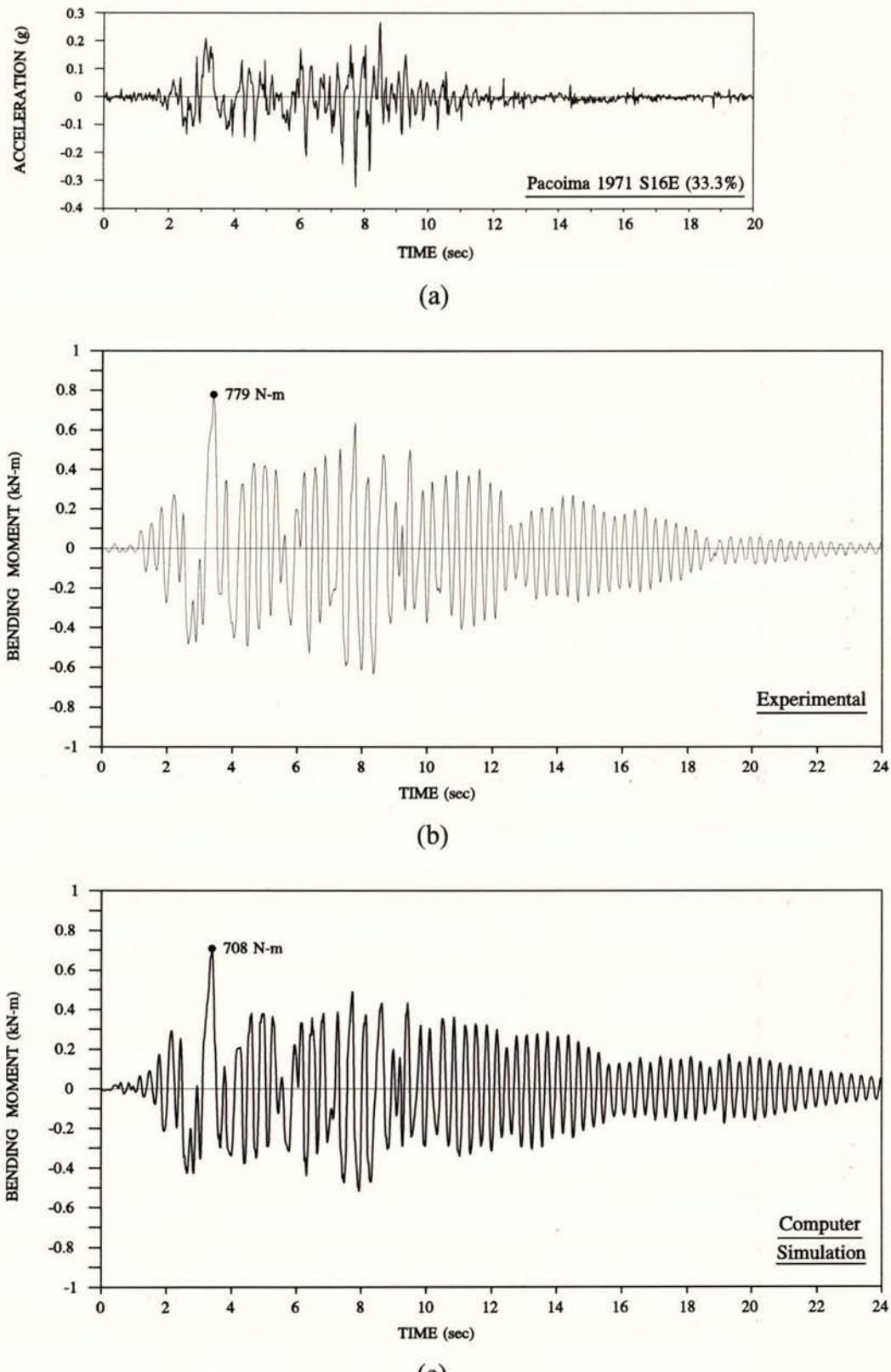
(b) Bending moment time history (experimental)

(c) Bending moment time history (computer simulation)



**Figure D.7** Dynamic response of PRSIS at  $0^\circ$  -

- Earthquake: El Centro 1940 N-S (100%)
- Ring spring displacement time history (experimental)
- Ring spring displacement time history (computer simulation)

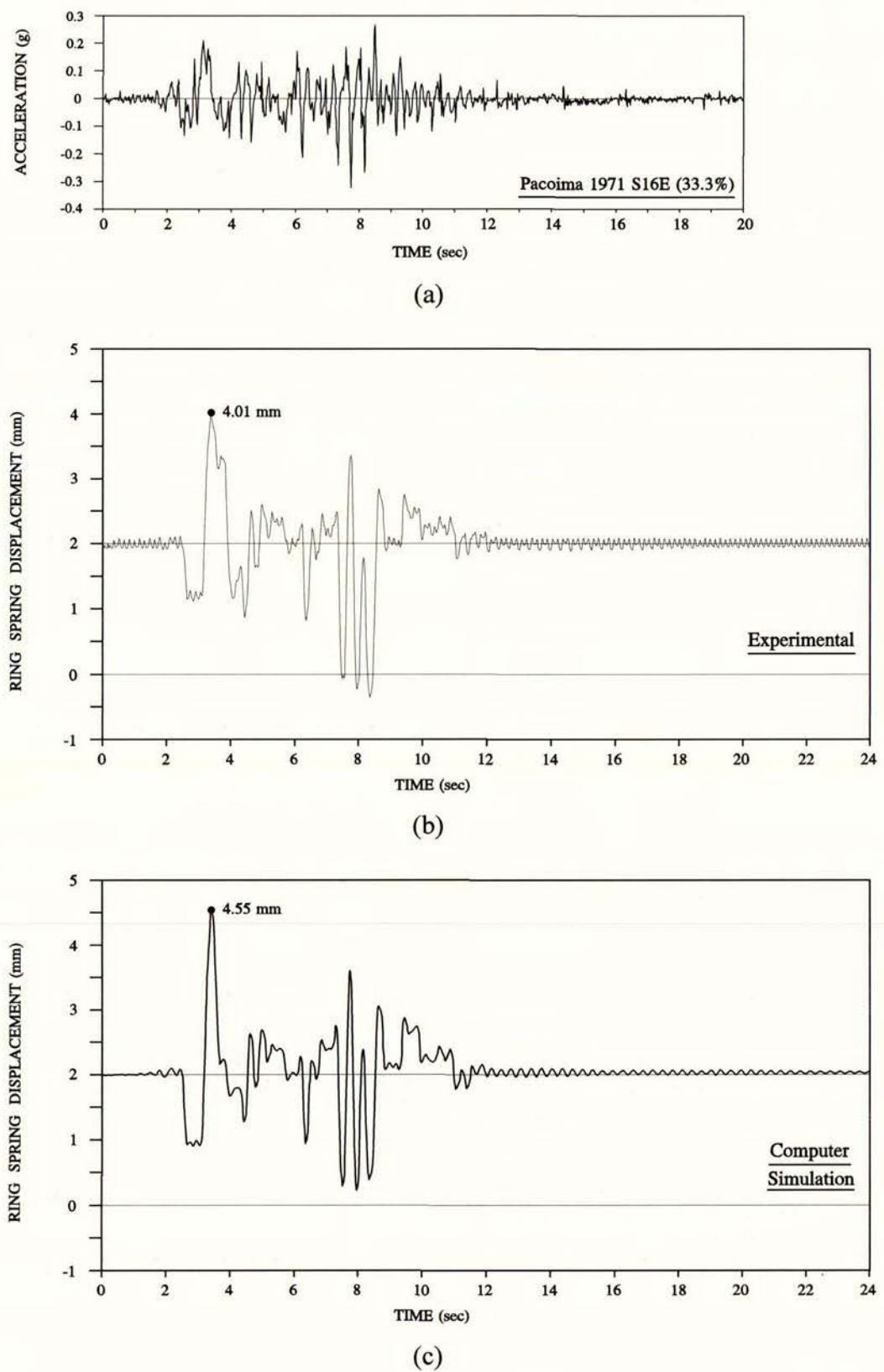


**Figure D.8** Dynamic response of PRSIS at  $0^\circ$  -

(a) Earthquake: Pacoima 1971 S16E (33.3%)

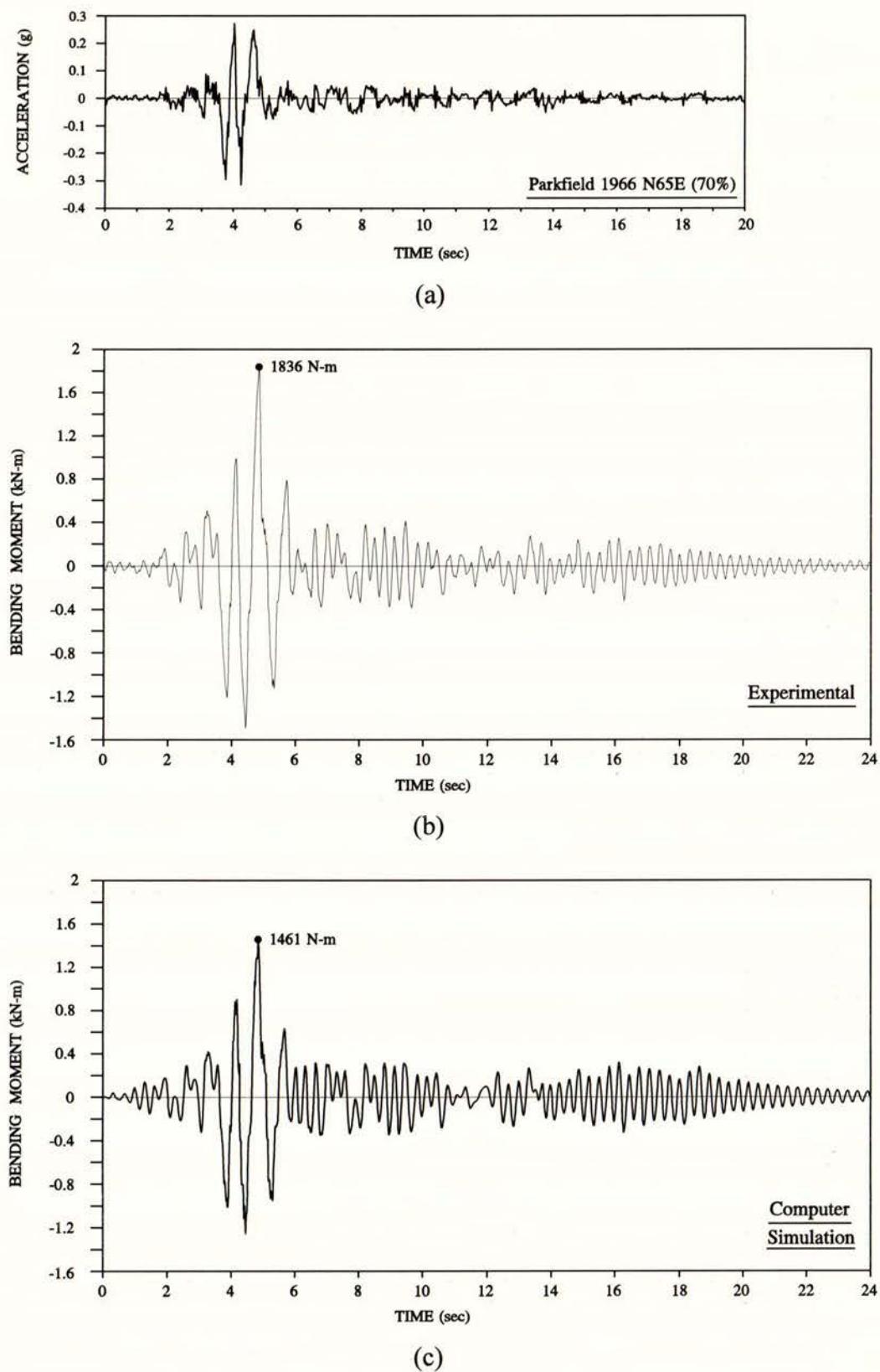
(b) Bending moment time history (experimental)

(c) Bending moment time history (computer simulation)



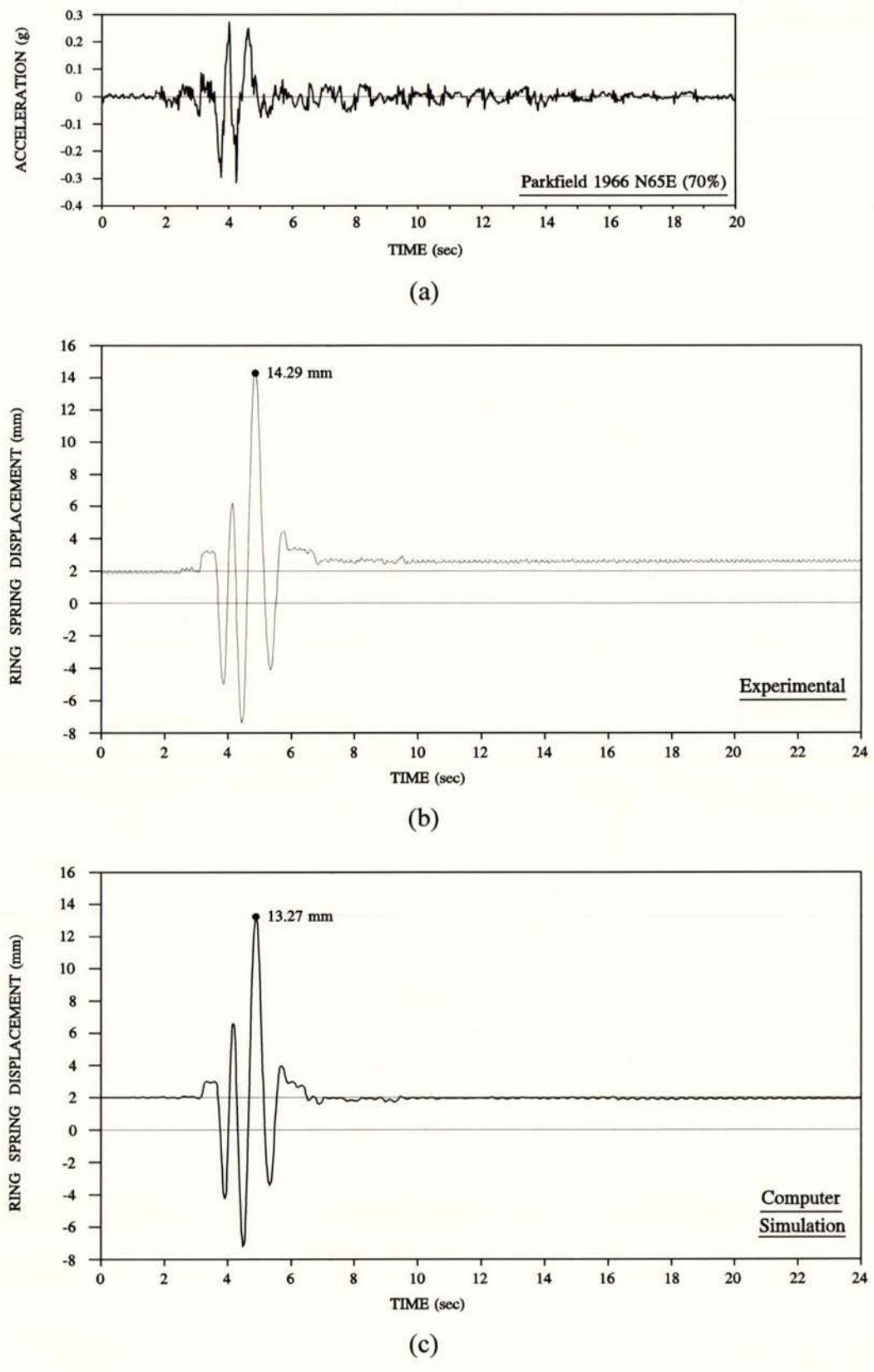
**Figure D.9** Dynamic response of PRSIS at  $0^\circ$  -

- Earthquake: Pacoima 1971 S16E (33.3%)
- Ring spring displacement time history (experimental)
- Ring spring displacement time history (computer simulation)



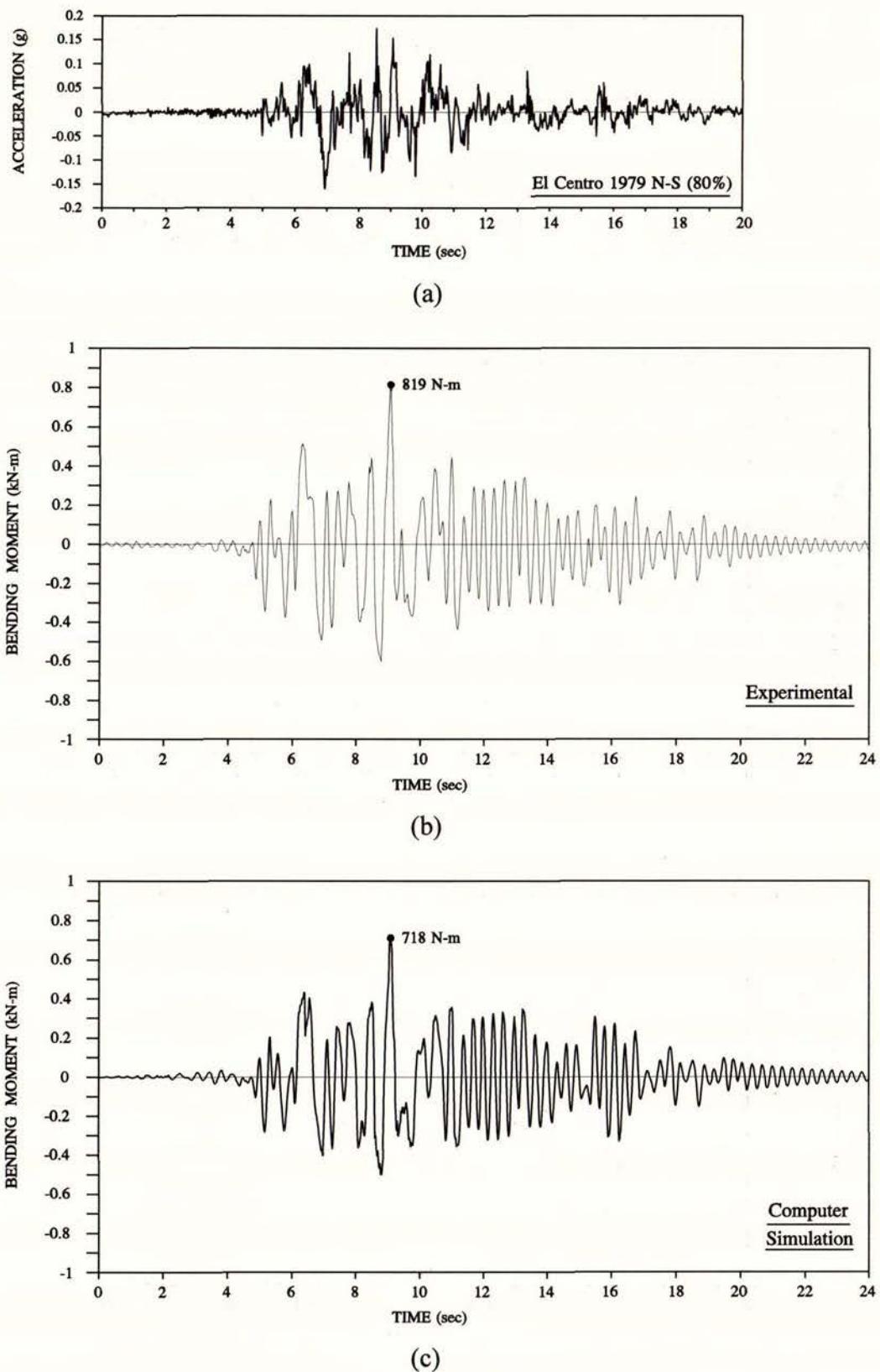
**Figure D.10** Dynamic response of PRSIS at  $0^\circ$  -

- Earthquake: Parkfield 1966 N65E (70%)
- Bending moment time history (experimental)
- Bending moment time history (computer simulation)



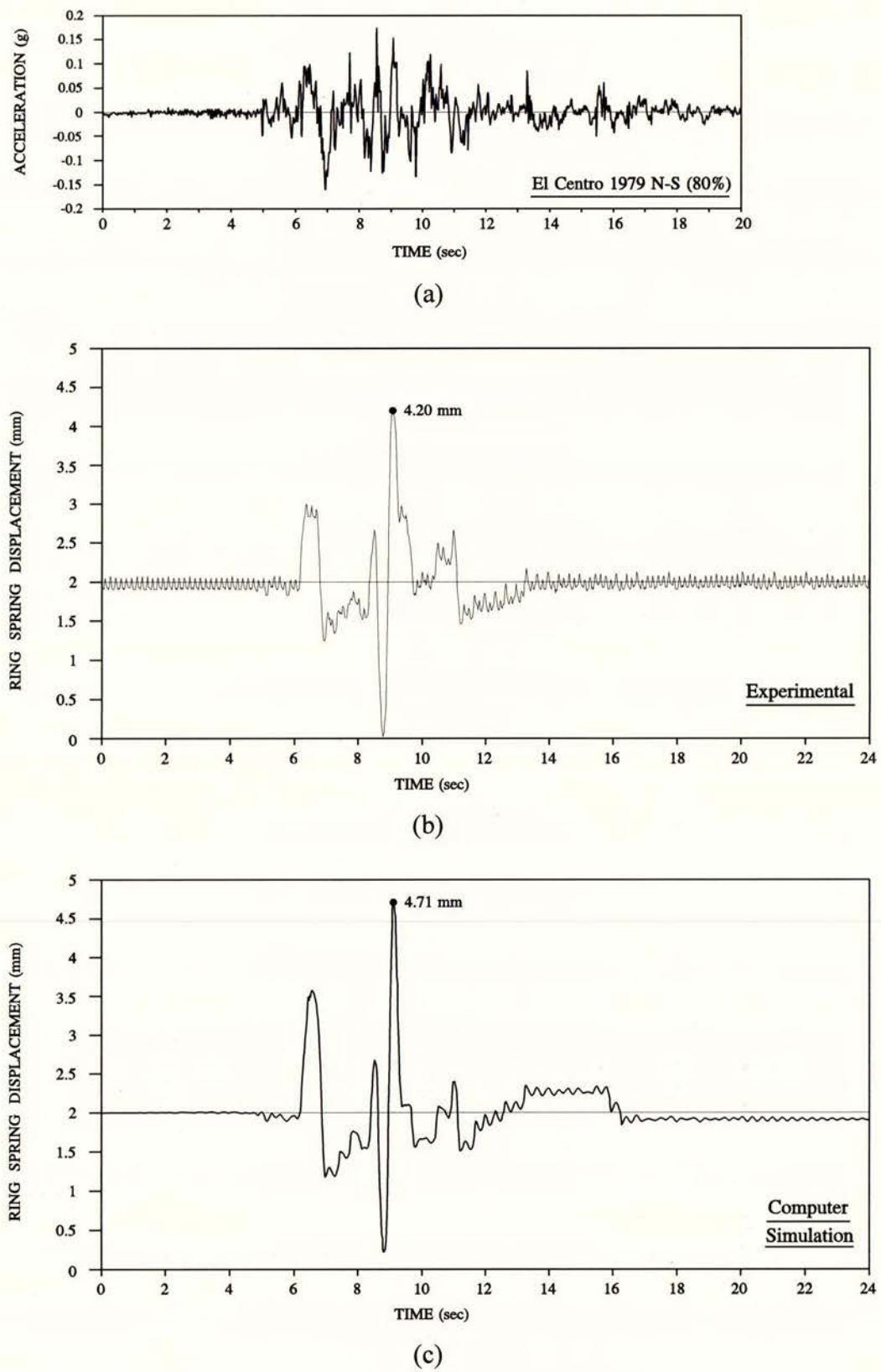
**Figure D.11** Dynamic response of PRSIS at  $0^\circ$  -

- Earthquake: Parkfield 1966 N65E (70%)
- Ring spring displacement time history (experimental)
- Ring spring displacement time history (computer simulation)



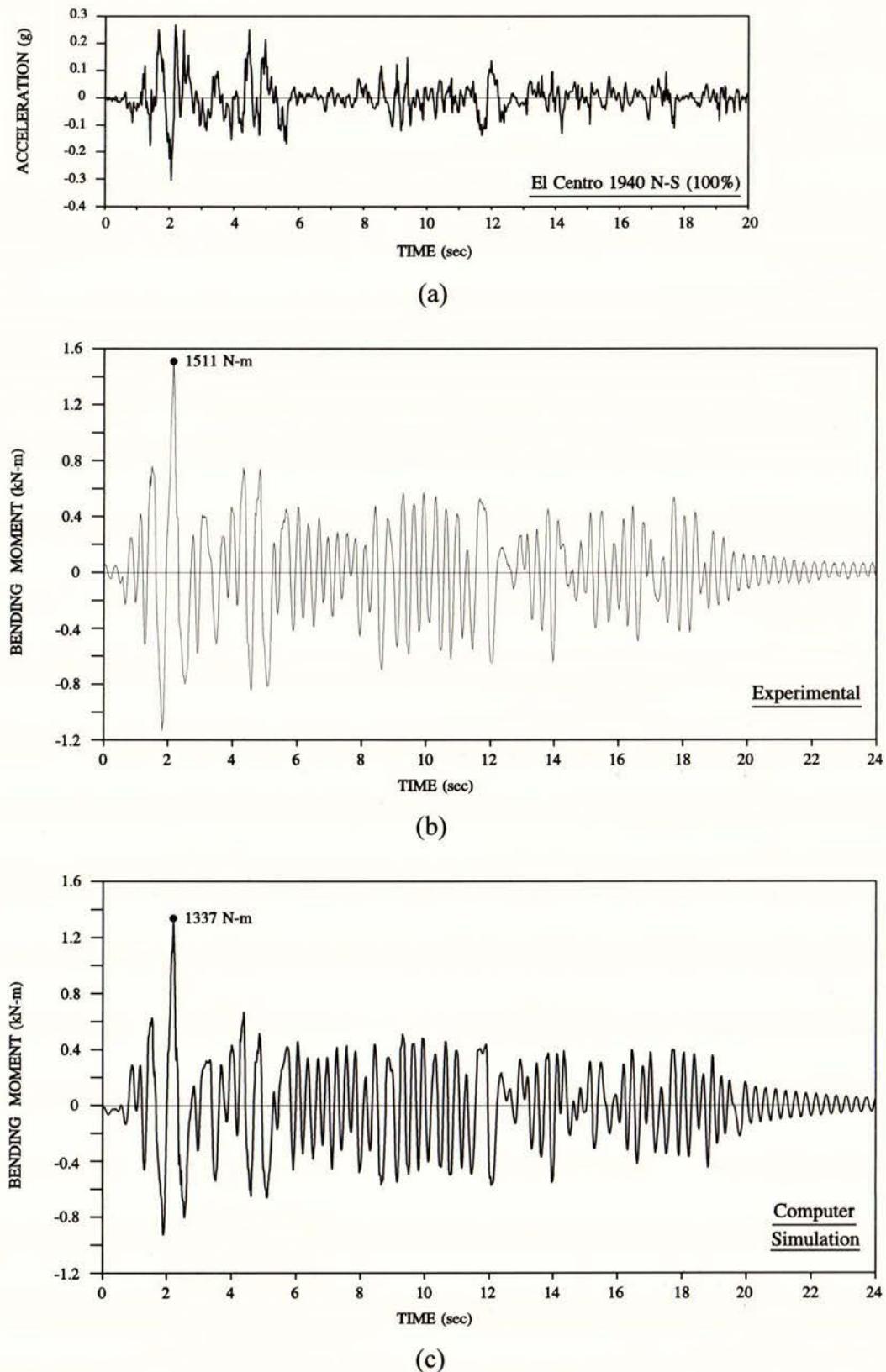
**Figure D.12** Dynamic response of PRSIS at  $0^\circ$  -

- (a) Earthquake: El Centro 1979 N-S (80%)
- (b) Bending moment time history (experimental)
- (c) Bending moment time history (computer simulation)



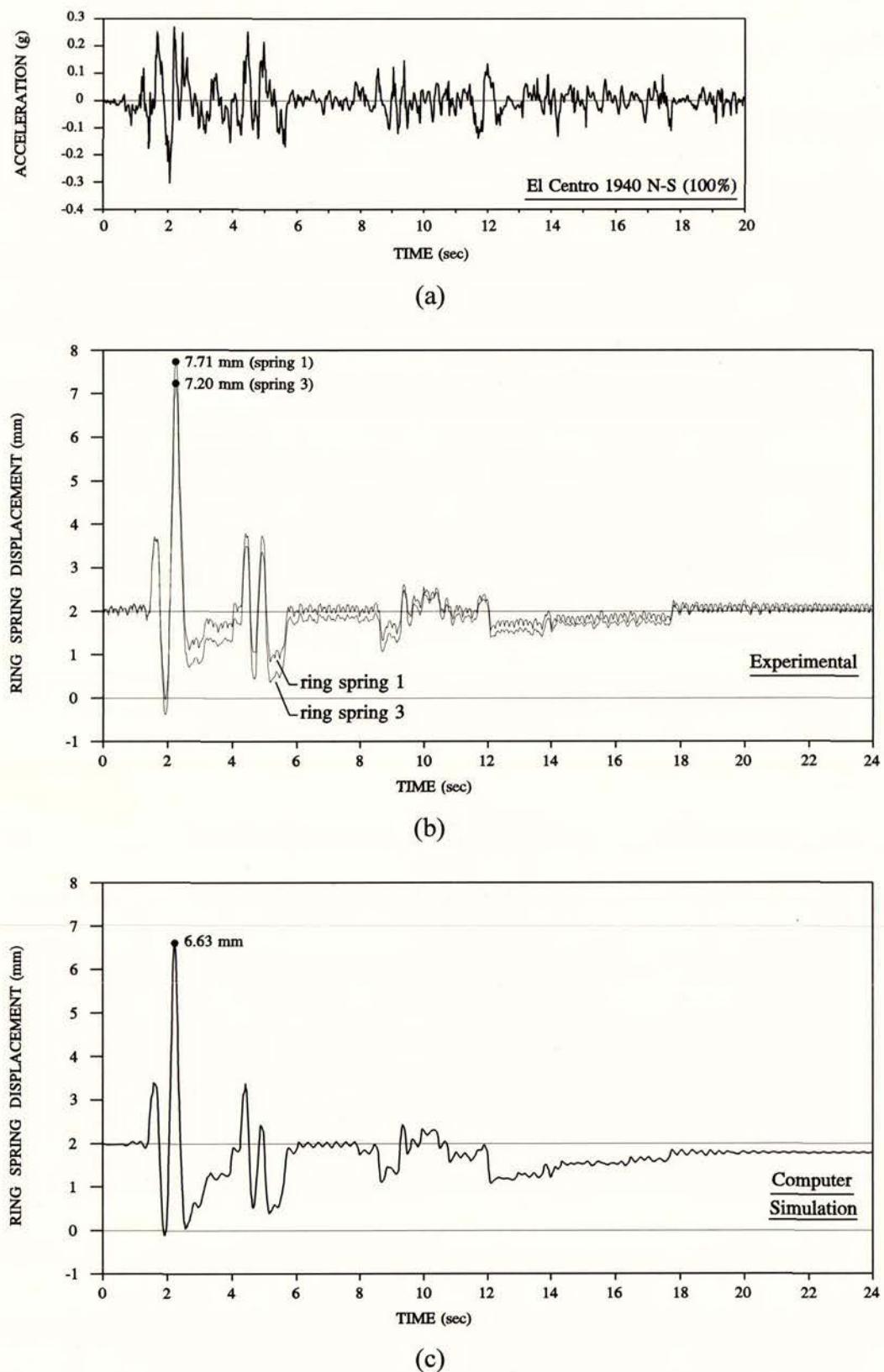
**Figure D.13** Dynamic response of PRSIS at  $0^\circ$  -

- (a) Earthquake: El Centro 1979 N-S (80%)
- (b) Ring spring displacement time history (experimental)
- (c) Ring spring displacement time history (computer simulation)



**Figure D.14** Dynamic response of PRSIS rotated 45° -

- (a) Earthquake: El Centro 1940 N-S (100%)
- (b) Bending moment time history (experimental)
- (c) Bending moment time history (computer simulation)



**Figure D.15** Dynamic response of PRSIS rotated 45° -

- Earthquake: El Centro 1940 N-S (100%)
- Ring spring displacement time history (experimental)
- Ring spring displacement time history (computer simulation)