# Torque Estimation for a 3-Axis Mobile Gimbal with 3-6 kg Payload

### 1. Torque Components and Estimation

The peak torque requirement for each gimbal axis can be estimated as the sum of inertial, frictional, and aerodynamic (wind) torques with an appropriate design margin:

$$T_{\text{peak}} \approx J \left( \alpha_{\text{cmd}} + \alpha_d \right) + b \omega + T_{\text{Coulomb}} + T_{\text{wind}}$$
 (1)

where

- J: payload + frame moment of inertia about the axis (kg·m<sup>2</sup>)
- $\alpha_{\rm cmd}$ : commanded angular acceleration (rad/s<sup>2</sup>)
- $\alpha_d$ : disturbance angular acceleration from base motion (rad/s<sup>2</sup>)
- b: viscous damping coefficient (N·m·s/rad)
- $T_{\text{Coulomb}}$  : static friction/Coulomb torque (N·m)
- $T_{\text{wind}}$ : wind-induced torque (N·m)

The wind-induced torque can be estimated using a flat-plate approximation:

$$T_{\text{wind}} \approx \frac{1}{2} \rho \, C_d \, A \, V^2 \, r \tag{2}$$

where

- $\rho$ : air density  $\approx 1.2 \,\mathrm{kg/m^3}$
- $C_d$ : drag coefficient (1.0–1.2 for bluff bodies)
- A: projected area normal to wind  $(m^2)$
- V: wind/airflow speed (m/s)
- $\bullet \ r$  : lever arm from axis to center of pressure (m)

1

# 2. Example Calculations

#### Example 1: Moderate Mapping/Inspection Case

- Payload mass:  $m = 5 \,\mathrm{kg}$
- Approx. box:  $0.22 \times 0.18 \times 0.15$  m
- Inertia about roll/pitch axis:

$$J \approx \frac{1}{12} m(b^2 + c^2) \approx 0.023 \,\mathrm{kg \cdot m^2}$$

- Commanded accel:  $\alpha_{\rm cmd} = 600^{\circ}/{\rm s}^2 \approx 10.47\,{\rm rad/s}^2$
- Disturbance accel:  $\alpha_d = 800^{\circ}/\text{s}^2 \approx 13.96 \, \text{rad/s}^2$
- Wind:  $A = 0.05 \,\mathrm{m}^2$ ,  $V = 12 \,\mathrm{m/s}$ ,  $r = 0.1 \,\mathrm{m}$ ,  $C_d = 1.1 \,\mathrm{m}$

$$T_{\text{inertia}} = J(\alpha_{\text{cmd}} + \alpha_d) = 0.023 \times (10.47 + 13.96) \approx 0.57 \,\text{N·m}$$
  
 $T_{\text{wind}} \approx 0.5 \times 1.2 \times 1.1 \times 0.05 \times 12^2 \times 0.10 \approx 0.48 \,\text{N·m}$ 

 $T_{\rm total, peak} \approx 0.57 + 0.48 + 0.1 \approx 1.15 \, \text{N} \cdot \text{m}$ 

In kg·cm:  $1.15 \,\mathrm{N\cdot m} \times 10.197 \approx 11.7 \,\mathrm{kg\cdot cm}$ 

Add  $2 \times \text{margin} \Rightarrow 23-30 \text{ kg} \cdot \text{cm}$  peak.

# Example 2: Rougher Outdoor Case (Landing in 30-60 kg·cm)

- Payload mass:  $m = 6 \,\mathrm{kg}$
- Approx. box:  $0.26 \times 0.22 \times 0.16 \,\mathrm{m}$
- Inertia:  $J \approx 0.050 \,\mathrm{kg \cdot m^2}$
- Commanded accel:  $\alpha_{\rm cmd} = 800^{\circ}/{\rm s}^2 \approx 14.0\,{\rm rad/s}^2$
- Disturbance accel:  $\alpha_d = 1200^{\circ}/\text{s}^2 \approx 21.0 \, \text{rad/s}^2$
- Wind:  $A = 0.08 \,\mathrm{m}^2$ ,  $V = 15 \,\mathrm{m/s}$ ,  $r = 0.12 \,\mathrm{m}$

$$T_{\rm inertia} \approx 0.050 \times (14.0 + 21.0) = 1.75 \,\mathrm{N\cdot m}$$
  
 $T_{\rm wind} \approx 0.5 \times 1.2 \times 1.1 \times 0.08 \times 15^2 \times 0.12 \approx 1.43 \,\mathrm{N\cdot m}$   
 $T_{\rm total, \ peak} \approx 1.75 + 1.43 + 0.2 \approx 3.38 \,\mathrm{N\cdot m}$ 

In kg·cm:  $3.38 \times 10.197 \approx 34.4 \,\mathrm{kg\cdot cm}$ 

With 1.5–2× margin for gusts and shocks:

Peak torque range  $\approx 52-70 \,\mathrm{kg\cdot cm}$ 

## 3. Key Takeaways for BLDC Motor Selection

- Continuous torque is much lower (10–20 kg·cm for 3–6 kg payloads).
- Peak torque of 30–60 kg·cm is realistic for off-road or windy outdoor conditions with safety margin.
- Use **direct-drive frameless BLDC** for micro-jitter; 2–3:1 timing belt if space or cost constrained.
- Place encoder on **load side** if belts are used to avoid compliance/backlash in the loop.

4. Quick Design Checklist

- 1. Compute  $J_r, J_p, J_y$  from CAD for payload + gimbal plate.
- 2. Decide  $\alpha_{\rm cmd}$  (slew profile) and estimate  $\alpha_d$  from base IMU logs.
- 3. Estimate A, r, and wind speed V for outdoor/off-road use.
- 4. Compute peak torque using Eq. (1) and convert to kg·cm.
- 5. Apply  $1.5-2 \times$  margin for shocks and uncertainty.
- 6. Select BLDC motors such that
  - $T_{\rm continuous} \ge T_{\rm rms}$
  - $T_{\text{peak}} \ge \text{calculated peak torque}$