

Doppler's Shift :

A LEO satellite in circular polar orbit with altitude $h = 1000$ km. A transmitter of a satellite on a frequency (f_t) of 2.68 GHz.

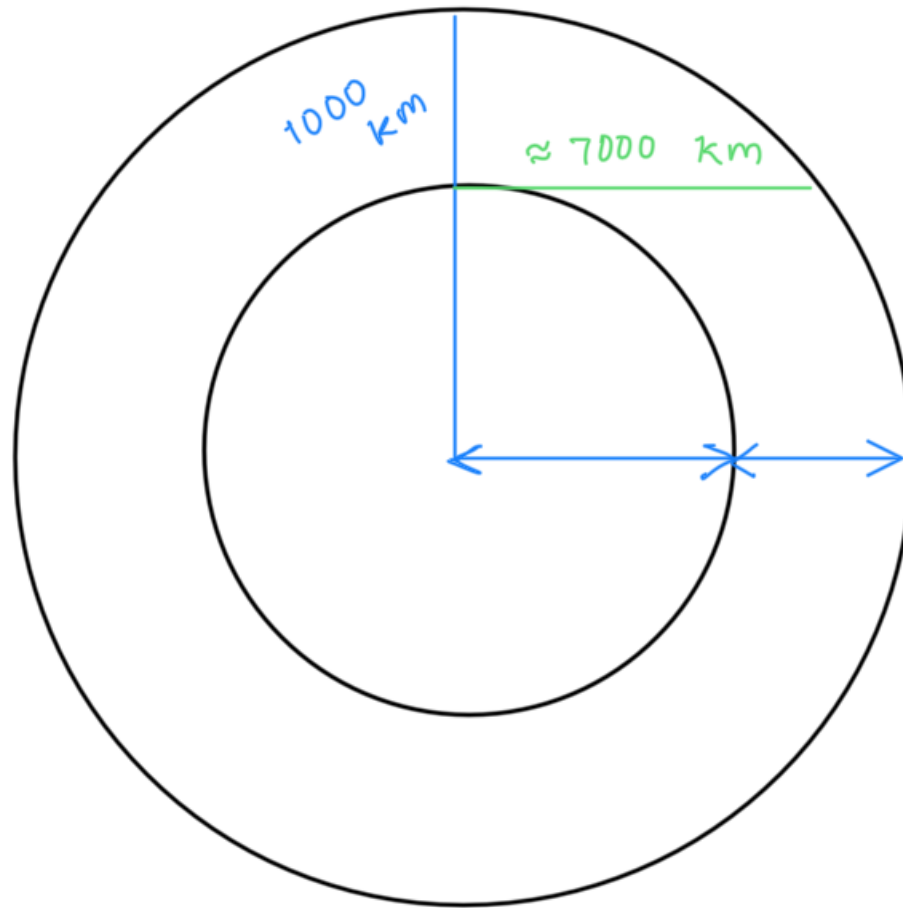
Find out:

1. Velocity of the satellite in orbit.
2. Components of the velocity towards observer on earth.
3. Find the doppler's shift of the received signal assume $R_e = 6378$ km. (Given: $f_t = 20$ GHz)



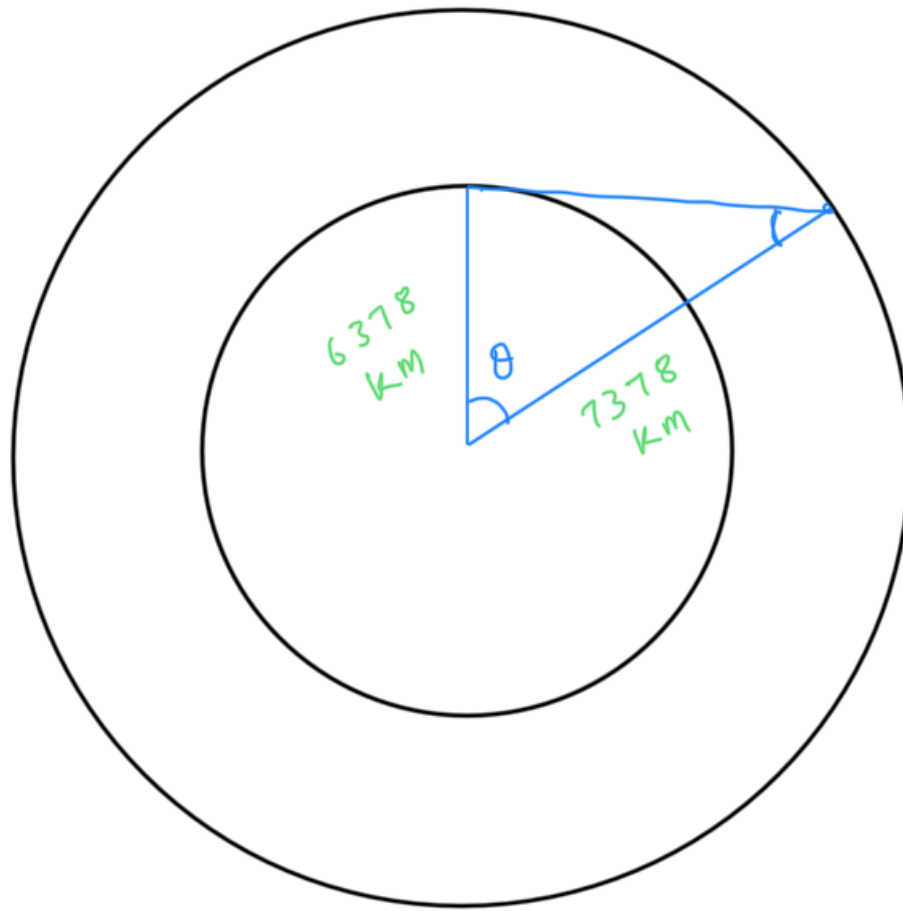
$$\frac{f_R - f_T}{f_T} = \frac{\Delta f}{f_T} = \frac{V_T}{V_P} \Rightarrow \Delta f = \frac{V_T}{V_P / f_T}$$
$$= \frac{V_T \cos \theta}{\lambda}$$

V_{approach}
is used
for
doppler
shift
calculation



$$T^2 = \frac{4\pi^2}{\mu} a^3$$

$$V_T = \frac{\cancel{2\pi} a}{\frac{\cancel{2\pi} a \sqrt{a}}{\sqrt{\mu}}} = \sqrt{\frac{\mu}{a}}$$



$$V_{\text{approach}} = V_T \cos \theta$$

$$= 6.354 \text{ km/s}$$

$$V_{\text{approach}} = 6354 \text{ m/s}$$

$$\Delta f = \frac{6354 \times 2.65 \times 10^9}{3 \times 10^8} = 56130 \text{ Hz} = 56.13 \text{ kHz}$$

$$f_T = 20 \text{ GHz (Given)}$$

$$\Delta f = 423.3 \text{ KHz}$$

There are 2 doppler frequencies, we use based on different approaches.