

Module 04 Heat Transfer Problem

(You may screen shot or paint this page and insert in in your exam package to save time)

1. **Name:** _____

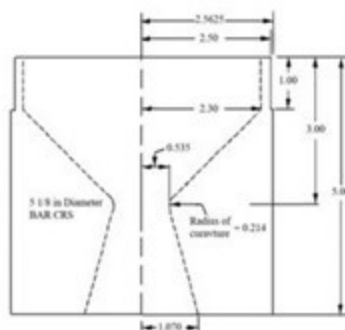
2. **Given:**

A small test rocket uses heat sink cooling of the nozzle throat by making this part out of a thick metal. The baseline nozzle material is **steel** and the alternative material is **copper**. Assume 1-D transient conduction in the metal and that the thickness of the nozzle material is much greater than the heat-affected zone.

For the $(T_{\infty} - T_i)$ term use:

- Temperature of Throat = $2667.2K = T_{\infty} = T_{\text{ex}}$
- Initial Temperature = $298K = T_i$

4. Schematic



Material Properties

Steel	Copper
$h = 10,000 \text{ W/(m}^2 \cdot \text{K)}$	$h = 15,100 \text{ W/(m}^2 \cdot \text{K)}$
$k = 60.5 \text{ W/(m} \cdot \text{K)}$	$k = 380 \text{ W/(m} \cdot \text{K)}$
$\alpha = 17.7 \times 10^{-6} \text{ m}^2/\text{s}$	$\alpha = 7.75 \times 10^{-6} \text{ m}^2/\text{s}$
$T_{\text{melt}} = 1643K$	$T_{\text{melt}} = 1185 K$

5. Find:

- A plot of the steel thermal profiles (x from 0.0 to 3.0) at 2, 5, 10 and 15 seconds.
- Determine the time when steel reaches melting point.
- A plot of the copper thermal profiles (x from 0.0 to 3.0) at 2, 5, 10 and 20 seconds.
- Determine the time when copper reaches melting point.
- Compare the thermal properties of each material and describe the parameters help extend the possible test and why they effect the answer.

6. Assume:

7. Analysis

- Show equations used in symbolic form.
- You do not have to show sample calculations with units.
- You can use the software developed on your homework assignment to make the calculations.
- Put a screen shot of tabular data and graphs in Exam Upload
- Upload any software used to make your calculations

Write your answer to question (b) in the box below before the end of the exam period.