

$$P_a(\theta) = \int \exp$$

$$\Theta_a(\sigma) \sim \sqrt{P} \gamma \rightarrow$$

Sound velocity  $\rightarrow a = \sqrt{\kappa R T}$

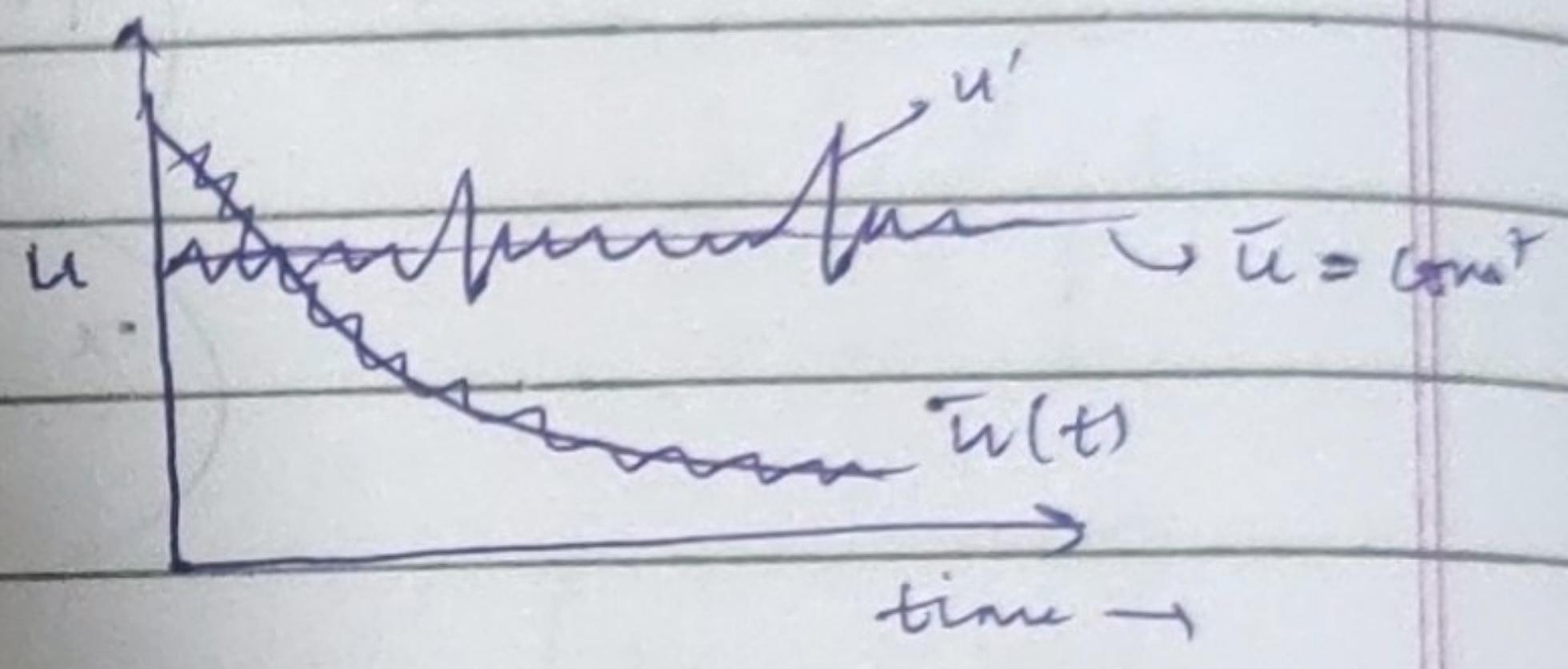
$$t_w = T_1 \left[ 1 + 0.167 Ma^2 \right] \xrightarrow{Ma=0.8} 302.178$$

# TURBULENT BOUNDARY LAYER

chap 6

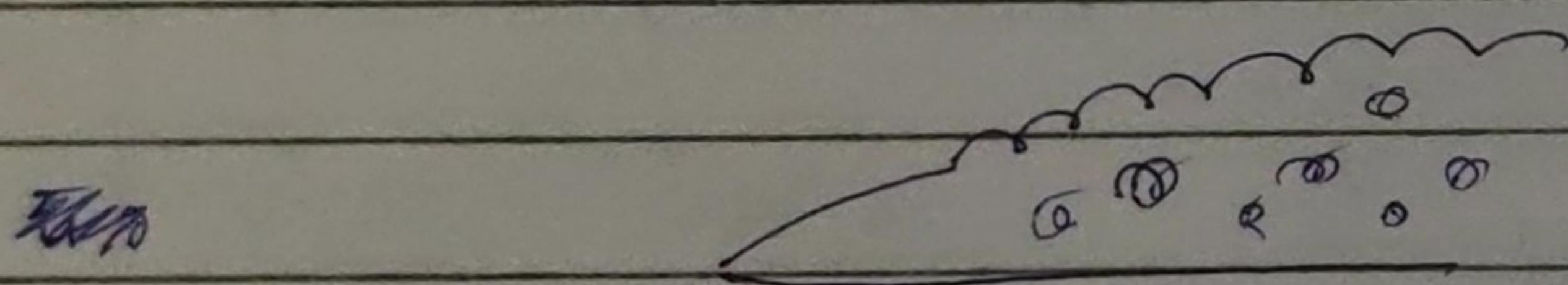
$$u = \bar{u} + u'$$

$$\bar{u} = \frac{1}{t_p} \int u \, dt_p$$



Take time frame

such that avg of fluctuating component is



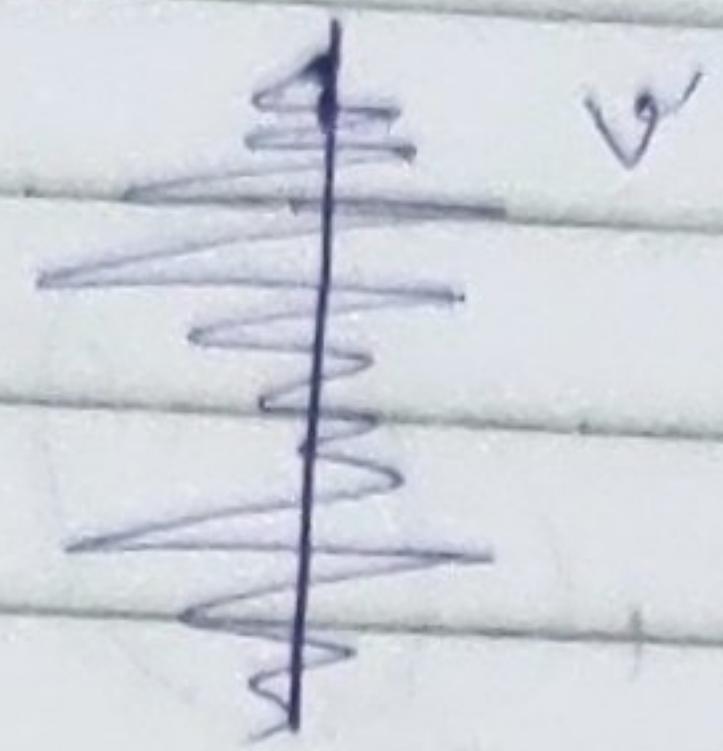
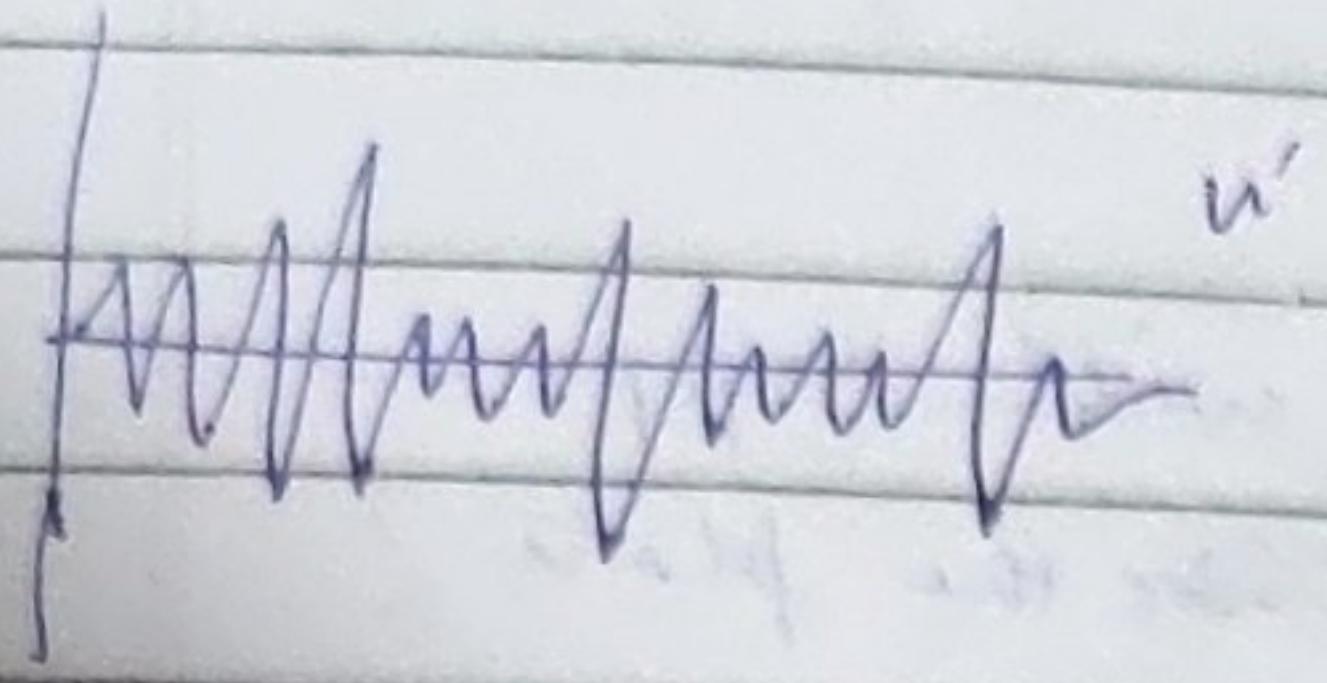
$$\frac{\partial \bar{u}}{\partial x} + \frac{\partial \bar{v}}{\partial y} + \frac{\partial \bar{w}}{\partial z} = 0$$

$$\bar{u} \frac{\partial \bar{u}}{\partial x} + \bar{v} \frac{\partial \bar{u}}{\partial y} + \bar{w} \cancel{\frac{\partial \bar{u}}{\partial z}} = - \frac{1}{\epsilon} \frac{\partial \bar{p}}{\partial x} + \gamma \frac{\partial^2 \bar{v}}{\partial x^2} - \cancel{\frac{\partial \bar{u}^2}{\partial z}} - \cancel{\frac{\partial (\bar{u} \bar{v})}{\partial z}}$$

$$\phi(u') \approx \phi(v')$$

$$\frac{\bar{u} \frac{\partial \bar{u}}{\partial x} + \bar{v} \frac{\partial \bar{v}}{\partial y}}{\partial x} = -\frac{1}{\rho} \frac{\partial \bar{p}}{\partial z} + \gamma \frac{\partial^2 \bar{u}}{\partial y^2} - \frac{\partial}{\partial y} \left( \gamma \bar{v} \right)$$

$$u' \geq 0, \quad g' \leq 0$$



$$\frac{\partial \bar{u}}{\partial x} + \frac{\partial \bar{v}}{\partial y} + \frac{\partial \bar{w}}{\partial z} = 0.$$

$$\bar{u} \frac{\partial \bar{u}}{\partial x} + \bar{v} \frac{\partial \bar{u}}{\partial y} = -\frac{1}{\rho} \frac{d \bar{P}}{dx} + \gamma \frac{\partial^2 \bar{u}}{\partial y^2} - \frac{\partial (\bar{u}' \bar{v}')} {\partial y}$$

$$\bar{u} \frac{\partial \bar{T}}{\partial x} + \bar{v} \frac{\partial \bar{T}}{\partial y} = \alpha \frac{\partial^2 \bar{T}}{\partial y^2} - \frac{\partial (\bar{v}' \bar{T}')} {\partial y} = -ve \cdot \text{Why?}$$

$$\frac{\partial \bar{u}'}{\partial x} + \frac{\partial \bar{v}'}{\partial y} = 0$$

$$\frac{\partial \bar{u}'}{\partial x} = - \frac{\partial \bar{v}'}{\partial y}$$

(if one is +ve, other is -ve)

Reynold's stress.

due to  
eddies

analogous to  $\lambda$ .

$$-\bar{v}' \bar{T}' = \epsilon_n \frac{\partial \bar{T}}{\partial y}$$

eddy thermal  
diffusivity

$\mu \rightarrow$  system property

$\epsilon_n \rightarrow$  not system property -  
depends on the eddies

$$-\bar{u}' \bar{v}' = \epsilon_m \frac{\partial \bar{u}}{\partial y}$$

eddy momentum  
diffusivity

analogous  
to  $\lambda$

$$\text{Momentum eqn} = -\frac{1}{\rho} \frac{d \bar{P}}{dx} + \gamma \frac{\partial^2 \bar{u}}{\partial y^2} + \frac{\partial}{\partial y} \left( \epsilon \frac{\partial \bar{u}}{\partial y} \right)$$

$$+ \frac{\partial}{\partial y} \left( (\alpha + \epsilon_m) \frac{\partial \bar{u}}{\partial y} \right)$$

Zapp  
apparent

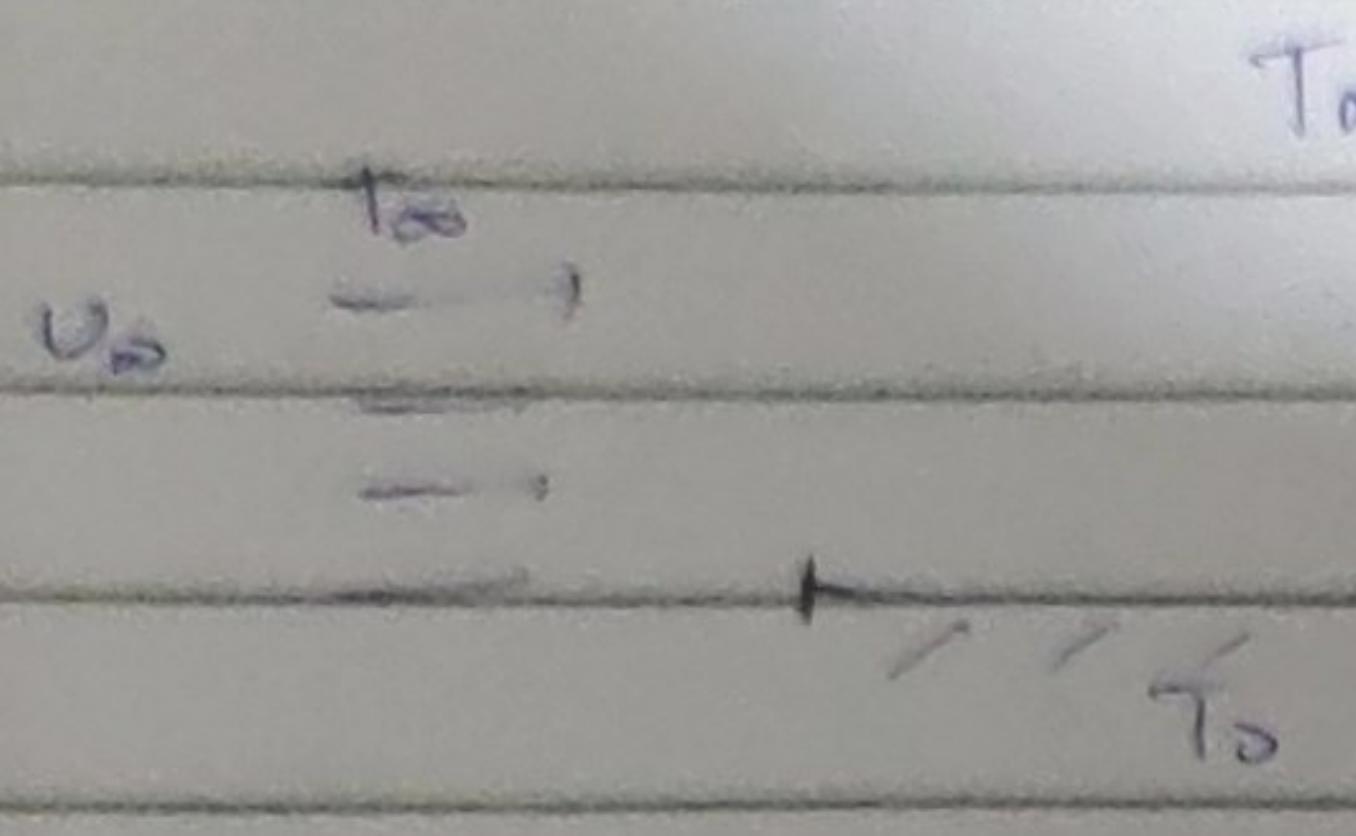
Energy eqn :-

$$\frac{\partial}{\partial y} \left( (\alpha + \epsilon_n) \frac{\partial \bar{T}}{\partial y} \right)$$

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$\epsilon_m, \epsilon_n, \bar{u}, \bar{v}, \bar{P}, \bar{T} \rightarrow$  unknown.

Take  $\bar{P} = \text{const}$



what will happen away from  
the plate?

$$\begin{aligned}
 \frac{\partial \bar{u}}{\partial x} + \frac{\partial \bar{v}}{\partial y} &= -\frac{1}{\nu} \frac{\partial \bar{p}_\infty}{\partial x} + \left\{ \frac{\partial \bar{u}}{\partial y^2} - \frac{\partial(\bar{u}\bar{v})}{\partial y} \right\} + \frac{1}{\nu} \left[ \frac{\partial}{\partial y} \left\{ \mu \frac{\partial \bar{u}}{\partial x} - (\bar{u}\bar{v}) \right\} \right] \\
 &+ \left[ \frac{\partial}{\partial y} \left\{ (\gamma + \epsilon_M) \frac{\partial \bar{u}}{\partial y} \right\} \right] \\
 &= \nu \epsilon_M \frac{\partial^2 \bar{u}}{\partial y^2}
 \end{aligned}$$

zapp  
regime then

$$\frac{\partial \bar{u}}{\partial x} + \frac{\partial \bar{v}}{\partial y} = -\frac{1}{\nu} \frac{d \bar{p}_\infty}{dx} + \left[ \frac{\partial}{\partial y} \left\{ (\mu + \epsilon_M) \frac{\partial \bar{u}}{\partial y} \right\} \right]$$

$\approx 0$

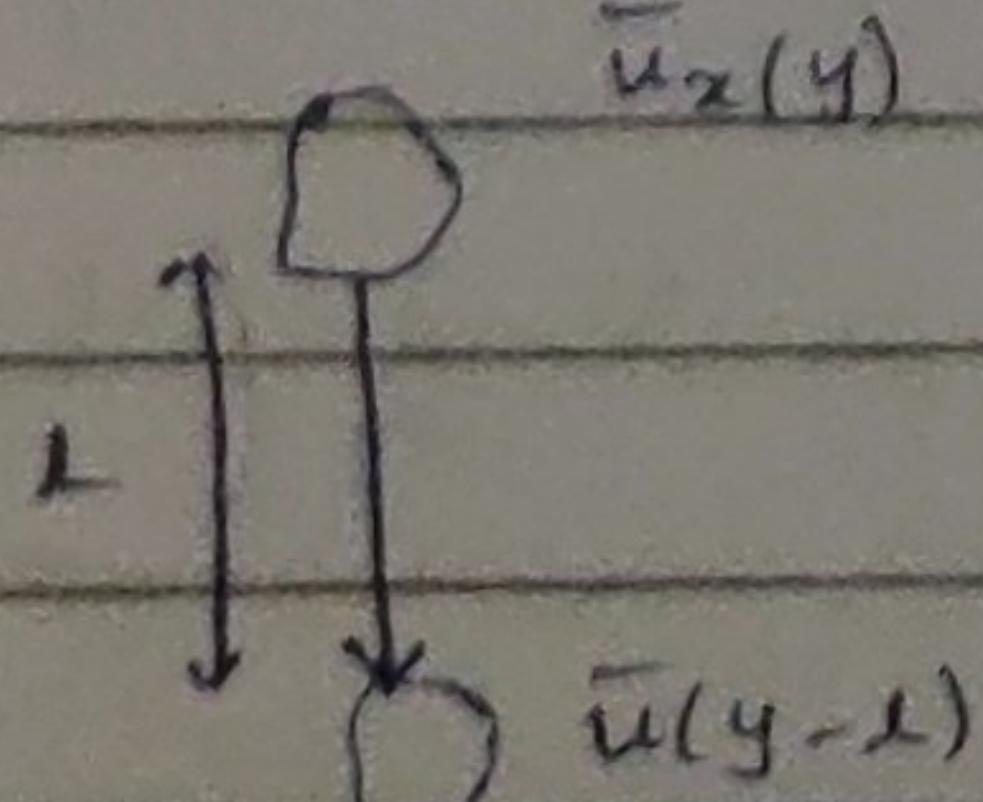
$$\frac{\partial \bar{u}}{\partial x} + \frac{\partial \bar{v}}{\partial y} = \left[ \frac{\partial}{\partial y} \left\{ (\alpha + \epsilon_H) \frac{\partial \bar{T}}{\partial y} \right\} \right]$$

$$\frac{\partial \bar{u}}{\partial x} + \frac{\partial \bar{v}}{\partial y} = 0$$

unknowns:  $\bar{u}, \bar{v}, \bar{T}, \epsilon_M, \epsilon_H$  (5)

3 equations

Mixing length model



$$\sim (\bar{u}_x(y) - \bar{u}_x(y-L)) \sim \sigma(u') \sim L \frac{\partial \bar{u}}{\partial y}$$

$$\sigma(u') \sim \sigma(v')$$

$$-\bar{u}'v' \sim L^2 \left( \frac{\partial \bar{u}}{\partial y} \right)^2$$

$$-\bar{u}'\bar{v}' = \rho \epsilon_M \frac{\partial \bar{u}}{\partial y}$$

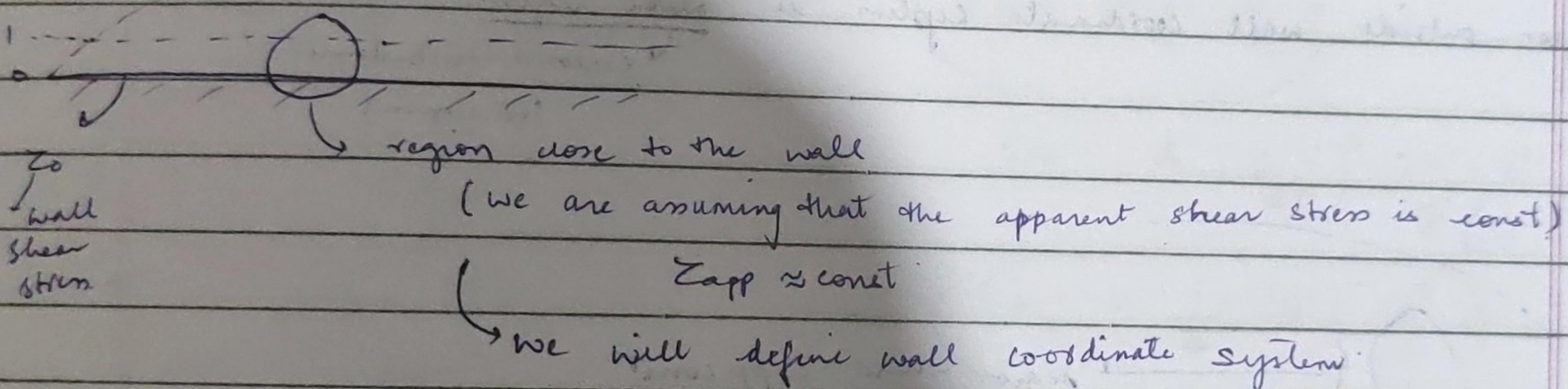
$$\epsilon_M = l^2 \left| \frac{\partial \bar{u}}{\partial y} \right|$$

$$\rightarrow l \propto y \sim K_y$$

kappa  
const

→ We need the velocity profile to find the wall shear stress.

→



$$z_{app} = \nu \frac{\partial \bar{u}}{\partial y} - \bar{u}'\bar{v}' = \nu \frac{\partial \bar{u}}{\partial y} + \epsilon_M \frac{\partial \bar{u}}{\partial y}$$

$$z_{app} = \nu (1 + \epsilon_M) \frac{\partial \bar{u}}{\partial y}$$

$$\epsilon_M = l^2 \left| \frac{\partial \bar{u}}{\partial y} \right|$$

$$z_{app} = \nu (1 + \epsilon_M) \frac{\partial \bar{u}}{\partial y} = z_0$$

⇒ Why do we use wall coordinate system?

$$u^* = \frac{\bar{u}}{u^*}$$

$$u^+ = \frac{\bar{u}}{u^*}$$

~~$$v^+ = \frac{\bar{v}}{u^*}$$~~

$$v^+ = \frac{\bar{v}}{u^*}$$

$$y^+ = \frac{y \bar{u}^*}{u^*}$$

$$x^+ = \frac{x \bar{u}^*}{u^*}$$

$$u^* = \left( \frac{z_0}{\nu} \right)^{1/2}$$

$$(r + \epsilon_M) \frac{du}{y} = \frac{z_0}{\rho}$$

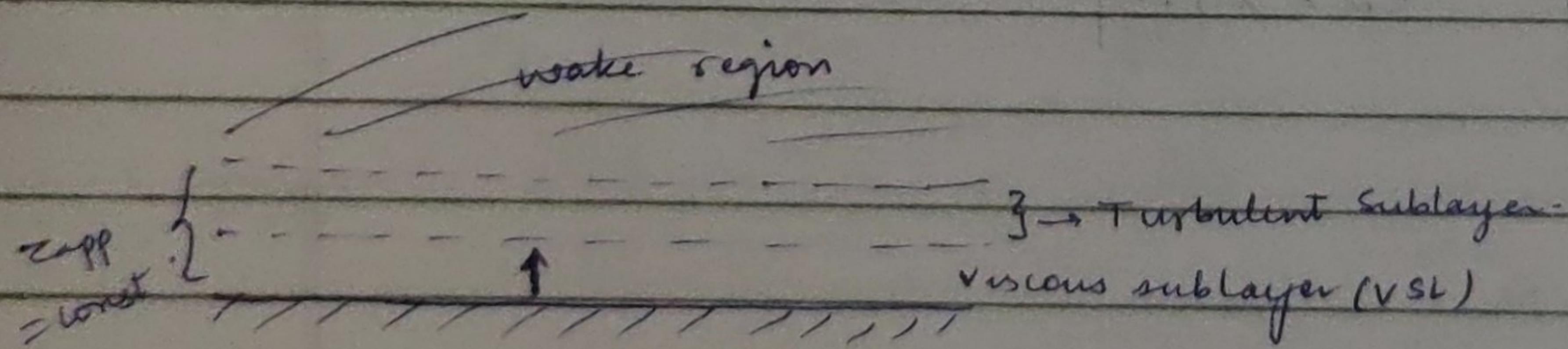
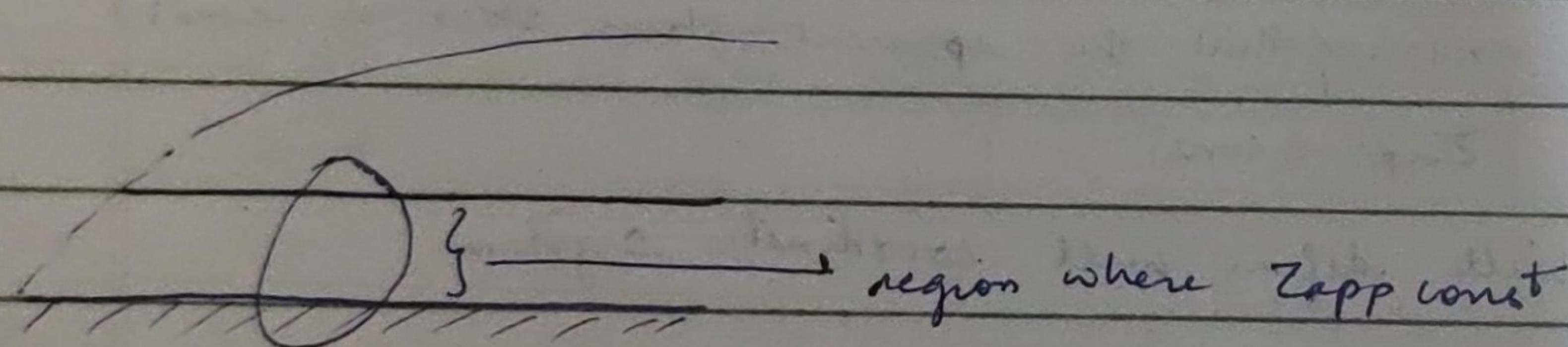
$$(r + \epsilon_M) \frac{(u^+)^2}{y} \frac{du^+}{dy} = (u^+)^2$$

$$\frac{du^+}{y^+} = \frac{1}{(1 + \frac{\epsilon_M}{r})}$$

$$\left(1 + \frac{\epsilon_M}{r}\right) \frac{du^+}{y^+} = 1$$

valid only where  
 $\epsilon_M \ll \text{const}$

region outside wall coordinate system is wake region  
excessive turbulence  
(BT region)



$\frac{\epsilon_M}{r} \rightarrow$  will determine which layer  
will dominate (TSL or VSL)

$$\eta = \frac{\mu}{\epsilon}$$

molecular diffusivity

$\gamma \gg \epsilon_M \rightarrow VSL$

$$\frac{\partial u^+}{\partial y^+} = 1$$
$$\int \frac{\partial u^+}{\partial y^+} dy^+ = \int dy^+$$
$$u^+ = y^+ \rightarrow 0 \leq y^+ \leq y_{VSL}^+$$

$\rightarrow \epsilon_M \gg \gamma$

$$\epsilon_M \frac{\partial u^+}{\partial y^+} = 1$$

$$\epsilon_M = \nu \frac{\partial \bar{u}}{\partial y}$$

$$k^2 (y^+)^2 \frac{\partial u^+}{\partial y^+} \frac{\partial u^+}{\partial y^+} = 1$$

$$\epsilon_M = k^2 y^2 \frac{\partial \bar{u}}{\partial y}$$

$$\frac{\partial u^+}{\partial y^+} = \frac{1}{k y^+}$$

$$\epsilon_M = k^2 \left( \frac{y^+ \gamma}{u^+} \right)^2 \frac{\partial u^+ \times u^+}{\partial y^+ \frac{\gamma}{u^+}}$$

$$\int u^+ dy^+ = \int \frac{dy^+}{k y^+}$$
$$u^+ \Big|_{y_{VSL}^+} = y^+ \Big|_{y_{VSL}^+}$$

$$\epsilon_M = k^2 \frac{\sqrt{u^+} (y^+)^2}{u^+} \frac{\partial u^+}{\partial y^+}$$

$$\epsilon_M = k^2 \frac{u^+}{u^+} (y^+)^2 \frac{\partial u^+}{\partial y^+}$$

$$u^+ - u^+_{VSL} = \frac{1}{k} \ln \left( \frac{y^+}{y_{VSL}^+} \right)$$

$$y_{VSL}^+ = 11 \cdot 1$$
$$k = 0.1$$

$$u^+ = y_{VSL}^+ + \frac{1}{k} \ln \left( \frac{y^+}{y_{VSL}^+} \right)$$

$$u^+ = y_{VSL}^+ - \underbrace{\frac{1}{k} \ln(y_{VSL}^+)}_{\text{const.}} + \underbrace{\left( \frac{1}{k} \right) \ln y^+}_{\text{const.}}$$

$$u^+ = A + B \ln y^+$$
$$A \sim 5.5 \quad B \sim 2.5$$

## Turbulent BL

$$u^+ = y^+ \quad 0 < y^+ \leq y_{VSL}^+$$

$$u^+ = A + B \ln y^+$$

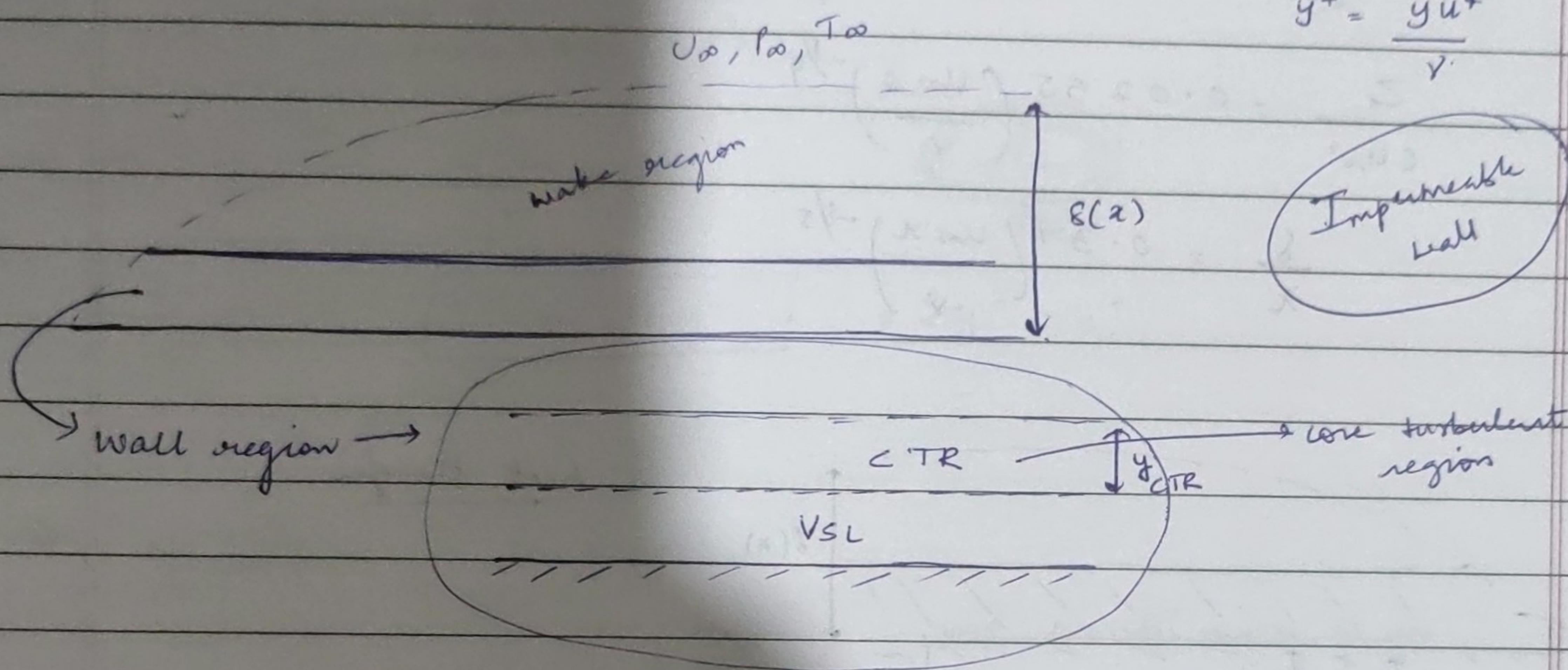
$$y_{VSL}^+ \leq y^+ \leq y_{CTR}^+$$

friction velocity  
( $z_0$  is unknown)

$$u^* = \left( \frac{z_0}{\epsilon} \right)^{1/2}$$

$$u^+ = \frac{\bar{u}}{u^*}$$

$$y^+ = \frac{y \bar{u}^*}{\epsilon}$$



$C_{app} = \text{constant}$

How to check whether the function is correct within the domain?

- ① experimental values    ② check the BC.

assume  $u^+ = f(y^+)$ ,  $\frac{\bar{u}}{u^*} = f\left(\frac{y \bar{u}^*}{\epsilon}\right) \rightarrow \text{inside the boundary layer}$

$$\text{at the edge of BL} \rightarrow \frac{u_\infty}{(z_0/\epsilon)^{1/2}} = f\left(\frac{\delta(z_0/\epsilon)^{1/2}}{\epsilon}\right) \rightarrow ①$$

$$\text{Prandtl's } 1/7 \text{ law} \rightarrow u^+ = 8 \cdot 7 (y^+)^{1/7} \quad z_0, \delta \text{ are unknowns.}$$

$$\text{Use integral approach} \rightarrow \bar{u} \frac{\partial \bar{u}}{\partial y} + \bar{v} \frac{\partial \bar{u}}{\partial x} = -\frac{1}{\rho} \frac{d P_\infty}{d x} + \sqrt{\frac{\partial^2 \bar{u}}{\partial y^2}}$$

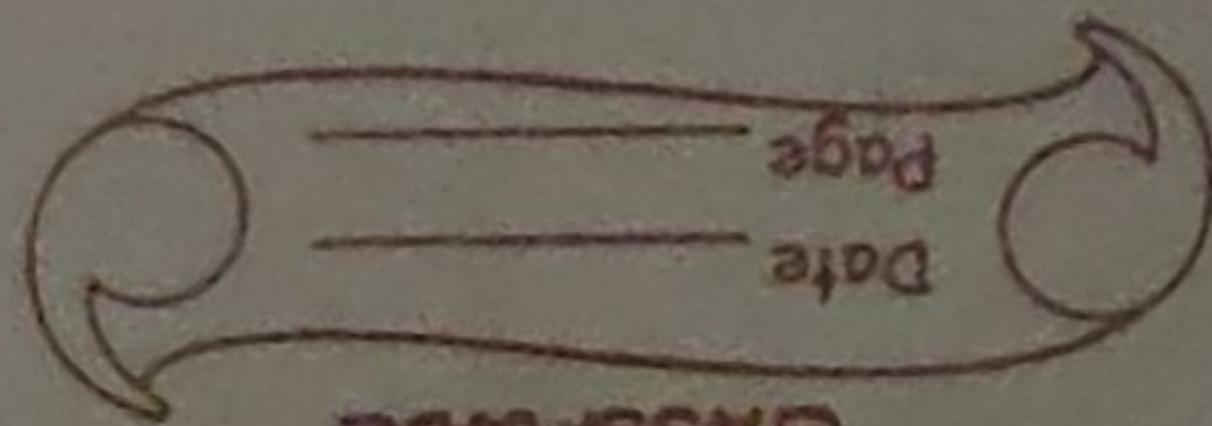
$P_r \sim 1$  for turbulent BL  
assumption

$$\int_0^s \bar{u} \frac{\partial \bar{u}}{\partial y} dy + \int_0^s \bar{v} \frac{\partial \bar{u}}{\partial x} dy = \int_0^s \sqrt{\frac{\partial^2 \bar{u}}{\partial y^2}} dy$$

$$\frac{d}{dx} \int_0^s \bar{u}^2 dy + [\bar{v} \bar{u}]_0^s = \left[ \frac{\partial \bar{u}}{\partial y} \right]_0^s$$

$$\frac{d}{dx} \int_0^s \bar{u}^2 dy + u_\infty \bar{v} \Big|_0^s = -\frac{1}{\rho} (z_0)$$

$$\frac{d}{dx} \int_0^s \bar{u} dy + u_\infty \left[ -\frac{d}{dx} \int_0^s \bar{u} dy \right] = -\frac{z_0}{\epsilon}$$



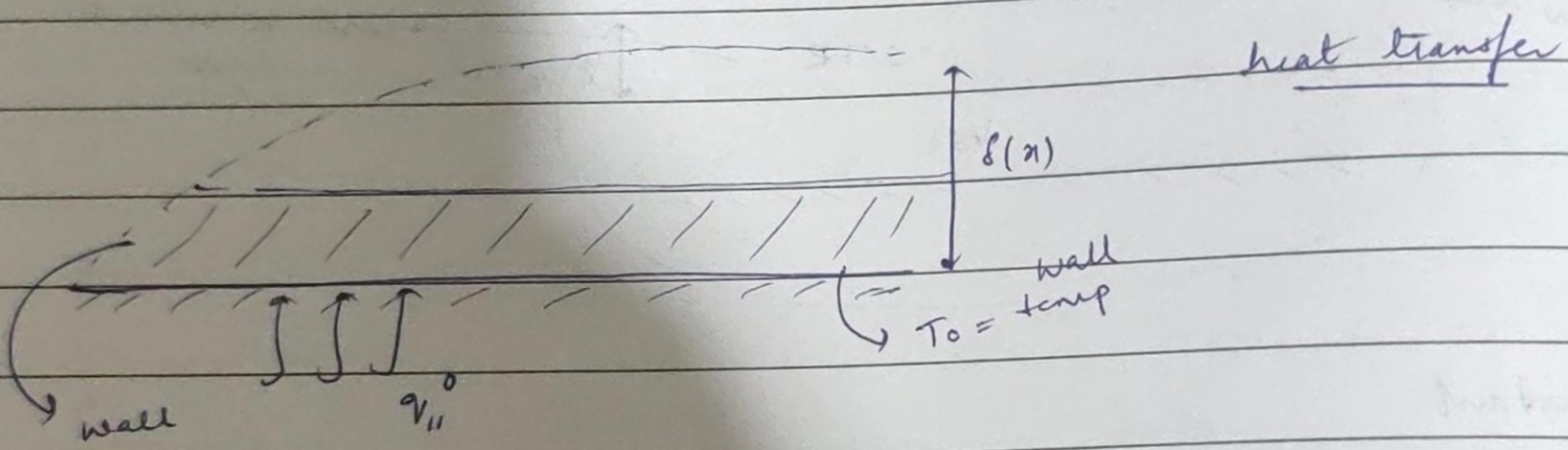
$$\frac{d}{dx} \int_0^{\delta} \bar{u}(u_{\infty} - \bar{u}) dy = \frac{z_0}{\rho} \rightarrow \textcircled{2} \quad \text{use } u^+ = 8 \cdot 2(y^+)^{1/4}$$

After integrating    $v_{\infty} = f(\delta, z_0) \rightarrow \textcircled{3}$

Compare eqn (3) & ①

$$\frac{z_0}{\rho u_{\infty}^2} = 0.0255 \left( \frac{u_{\infty} \delta}{\nu} \right)^{-1/4}$$

$$\frac{\delta}{x} = 0.37 \left( \frac{u_{\infty} x}{\nu} \right)^{-1/5}$$



region (apparent heat flux is const)

$$(\alpha + \epsilon_H) \frac{\partial \bar{T}}{\partial y} = - \bar{q}_o'' \text{, app.}$$

$q_{\text{app}} = \text{const}$   
 $\rightarrow \neq f(y)$

$$\bar{q}_o'' \text{, app} = - \bar{q}_o'' = - \rho C_p (\alpha + \epsilon_H) \frac{\partial \bar{T}}{\partial y}$$

$$y^+ = \frac{y u^+}{\nu} \quad \frac{\nu}{u^+} \frac{\partial y^+}{\partial y} = \frac{\partial y}{\partial y}$$

$$\frac{\partial \bar{T}}{\partial y^+} \left( \frac{\rho C_p u^+}{\bar{q}_o''} \right) = \frac{1}{\left( \frac{\alpha}{\nu} + \frac{\epsilon_H}{\nu} \right)}$$

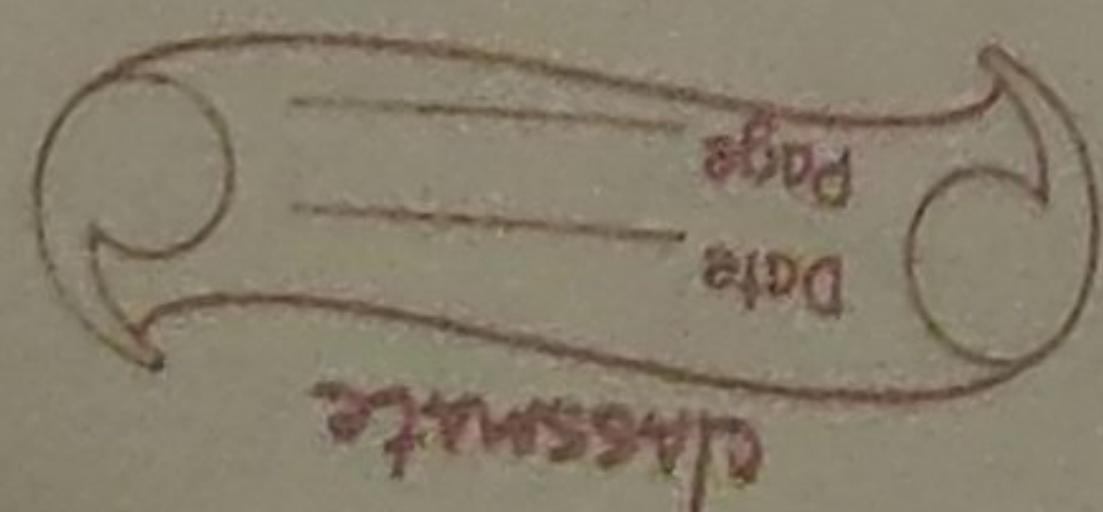
$$\frac{\partial (T_0 - \bar{T})}{\partial y^+} \times \frac{\rho C_p u^+}{\bar{q}_o''} =$$

$$\bar{T}^+ (y^+, x^+)$$

$$\frac{\partial \bar{T}^+}{\partial y^+} = \frac{1}{\left( \frac{1}{P_f} + \frac{\epsilon_M}{\nu} \times \frac{\epsilon_H}{\epsilon_M} \right)}$$

eddy thermal diffusivity  
 $\epsilon_H = \frac{1}{P_f} \rightarrow \text{not a constant}$

eddies are carrying the energy & momentum  
In the absence of eddies  $\epsilon_H, \epsilon_M$  &  $P_f$  are constant.



$P_f$  due to eddies

$\epsilon_M$  due to molecular

eddy momentum diffusivity

$$\frac{\partial T^+}{\partial y^+} = \left( \frac{1}{\frac{1}{Pr} + \frac{Em}{Y} \frac{1}{Pr_t}} \right)$$

close to the wall  $\rightarrow$  VSL  $\rightarrow$  due to molecular movement we have momentum transfer.

At conduction sub layer  $\rightarrow$   $\frac{Em}{Y} \frac{1}{Pr_t} \ll \frac{1}{Pr}$

↓ similar to viscous sub layer

ratio is very small

(?)  
some assumption?  
(check)

At CSL,  $\frac{\partial T^+}{\partial y^+} = Pr$

$$T^+ = Pr y^+ \quad 0 < y^+ < y_{CSL}^+$$

→ at CTR

$$\frac{\partial T^+}{\partial y^+} = \frac{1}{\frac{Em}{Y} \times \frac{1}{Pr_t}} \quad \text{use prandtl mixing length.}$$

$$l = Ky$$

$$\frac{Em}{Y} = K^2 y_+^2 \frac{du}{dy^+} = Ky_+$$

$$u^+ = A + B \ln y^+$$

$$\frac{du^+}{dy^+} = \frac{B}{y^+}$$

$$B = \frac{1}{K}$$

(at a particular  $y^+ \rightarrow Pr_t$  is taken to be const)

$$\frac{\partial T^+}{\partial y^+} = \frac{Pr_t}{Ky^+}$$

$$\int_{T_{CSL}}^{T^+} dT^+ = Pr_t \int_{y_{CSL}^+}^{y^+} \frac{dy^+}{y^+}$$

amt of energy & momentum transferred by eddies is same wrt  $y$

$$T^+ = Pr y^+ + \frac{Pr_t}{K} \ln \left( \frac{y^+}{y_{CSL}^+} \right)$$

## TURBULENT BL

$$\tau^+ = \rho_r y^+ \quad 0 < y^+ < y_{csl}^+$$

$$\tau^+ = \rho_r y_{csl}^+ + \frac{\rho_r k}{\kappa} \ln\left(\frac{y^+}{y_{csl}^+}\right) \quad y_{csl}^+ < y^+ < y^+$$

$$v^+ = \gamma y^+ \quad 0 < y^+ < y_{csl}^+$$

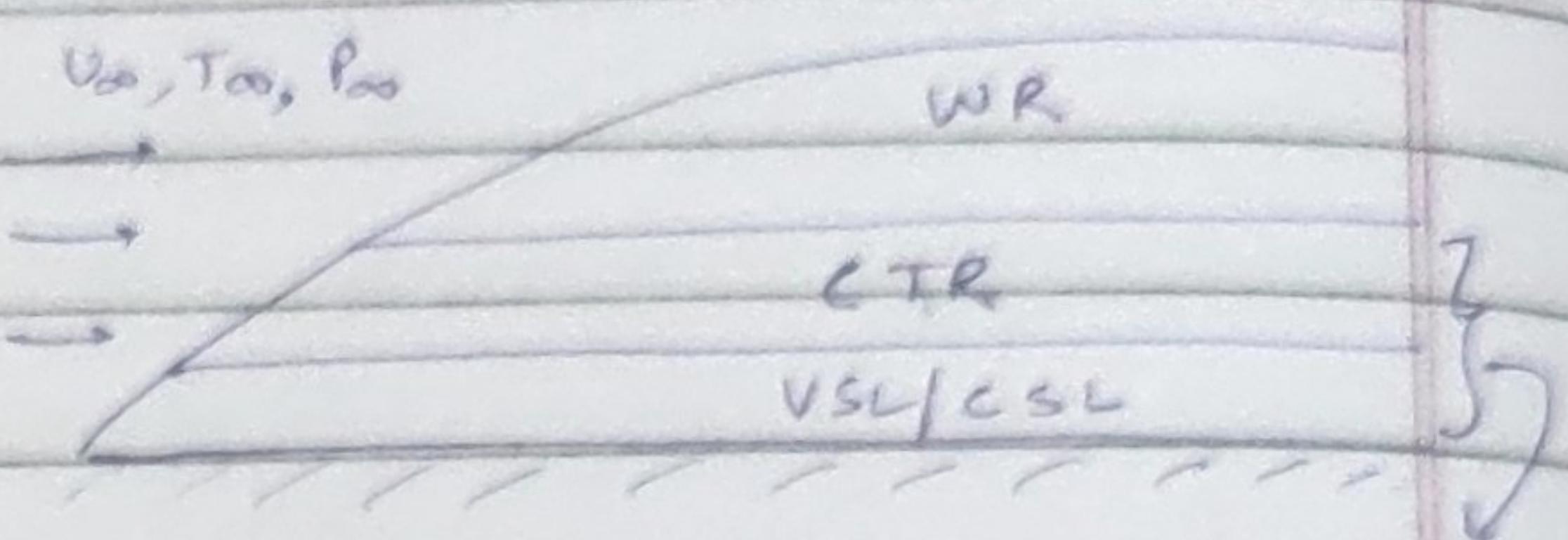
$$v^+ = \lambda + B \ln y^+ \quad y_{csl}^+ < y^+$$

$$u^* = (z_0/c)^{1/2}$$

friction velocity  
velocity  
length  
or the shear layer

$$\tau^+ = (T_0 - \bar{T}) \frac{u^* e_{cp}}{\eta''}$$

wall  
coordinate  
system



→ apply this at the edge of BL becoz we know the boundary condition at the edge of the BL

$$\tau^+ = \rho_r y_{csl}^+ + \frac{\rho_r k}{\kappa} \left[ \ln y^+ - \ln y_{csl}^+ \right]$$

$$\frac{(T_0 - \bar{T}) u^* e_{cp}}{\eta''} = \rho_r y_{csl}^+ + \frac{\rho_r k}{\kappa} \left[ \ln \frac{u^*}{\eta''} - \ln y_{csl}^+ \right]$$

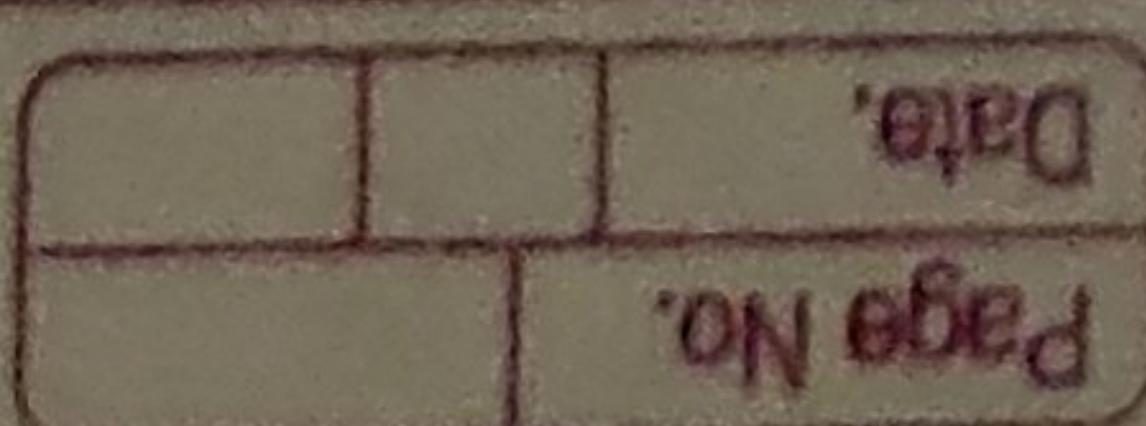
at BL  $\bar{T} = T_0$

extend this at the edge of BL

$$\frac{(T_0 - T_a) u^* e_{cp}}{\eta''} = \rho_r y_{csl}^+ + \frac{\rho_r k}{\kappa} \left[ \ln \frac{u^*}{\eta''} - \ln y_{csl}^+ \right] \quad \text{--- (1)}$$

$$\frac{\delta}{z} = 0.37 \left( \frac{u_{\infty} z}{\eta''} \right)^{1/5}$$

cannot use this ??  
we have assumed a velocity &  
got this  $\delta$  (now we won't  
assume any velocity profile)



$$u^+ = A + B \ln y^+$$

$$u^+ = y_{VSL}^+ + \frac{1}{K} \ln \left( \frac{y^+}{y_{VSL}^+} \right)$$

Let's assume

$$u^+ = y_{VSL}^+ - \frac{1}{K} \ln y_{VSL}^+ + \frac{1}{K} \ln y^+$$

$$u^+ = B + \frac{1}{K} \ln y^+$$

$$\frac{u}{u^+} = \frac{1}{K} \ln \frac{y u^+}{y} + B$$

Extend this at the edge of BL

$$\left[ \frac{u_\infty}{u^+} = \frac{1}{K} \ln \left( \frac{8 u^+}{\nu} \right) + B \right] \rightarrow (2)$$

$$\left( \frac{u_\infty}{u^+} \right)^2 = \frac{u_\infty^2}{2 \nu} = \frac{2}{(2 \nu / c_{f,2})} = \frac{2}{c_{f,2} \nu}$$

$$\left[ \left( \frac{u_\infty}{u^+} \right)^2 = \left( \frac{2}{c_{f,2} \nu} \right)^{1/2} \right]$$

$$\ln \left( \frac{8 u^+}{\nu} \right) = K \left[ \frac{u_\infty}{u^+} - B \right]$$

$$\frac{h}{\rho C_p u_\infty} = St_x = \frac{Nu}{Pr} = \frac{Nu}{Re \times Pr} \rightarrow \checkmark$$

Pr ~ 1

$$St_x \propto \frac{1}{2} c_{f,2} \nu$$

Pr should be close to 1

$$\underline{\underline{c_{f,2} = St_x (Pr)^{1/2}}}$$

Von Karman assumed 3 layers - Buffer layer

$$\rightarrow \bar{Nu}_2 = \frac{1}{2} C_f \rho_\infty^{1/2} \quad 0 < \gamma < 0.5$$

$$\approx 0.0296 \text{ Re}_x^{4/5} \rho_\infty^{1/2}$$

$$\bar{Nu} = 0.033 \text{ Re}_x^{4/5} \rho_\infty^{1/2} \quad \text{Pr} \geq 0.5$$

$$\frac{d\bar{u}}{dy} (\gamma + \epsilon_M) = - \frac{2}{\rho} \frac{dP}{dx}$$

$$\frac{d\bar{T}}{dy} (\gamma + \epsilon_M) = - \frac{\alpha''}{\rho C_p}$$

assume:-  $\rho_\infty = 1 \Rightarrow \epsilon_M = \epsilon_M \quad \gamma \approx 1 \quad \alpha = \sqrt{}$

$$\frac{d\bar{T}}{d\bar{u}} = - \frac{\alpha''}{\rho C_p}$$

$$\int_{T_0}^{T_\infty} d\bar{T} = - \frac{1}{\rho} \frac{\alpha''}{C_p} \int_0^{\bar{u}_\infty} d\bar{u}$$

$$\frac{T_\infty - T_0}{\alpha''} = +1 \quad \frac{U_\infty}{C_p \bar{u}_0}$$

$$C_p \left( \frac{T_0 - T_\infty}{\alpha''} \right) = \frac{U_\infty}{\bar{u}_0}$$

$$\frac{C_p U_\infty}{\alpha''} = \frac{U_\infty^2}{\bar{u}_0}$$

$$\frac{C_p U_\infty}{h} = \frac{U_\infty^2}{\frac{1}{2} \bar{u}_0}$$

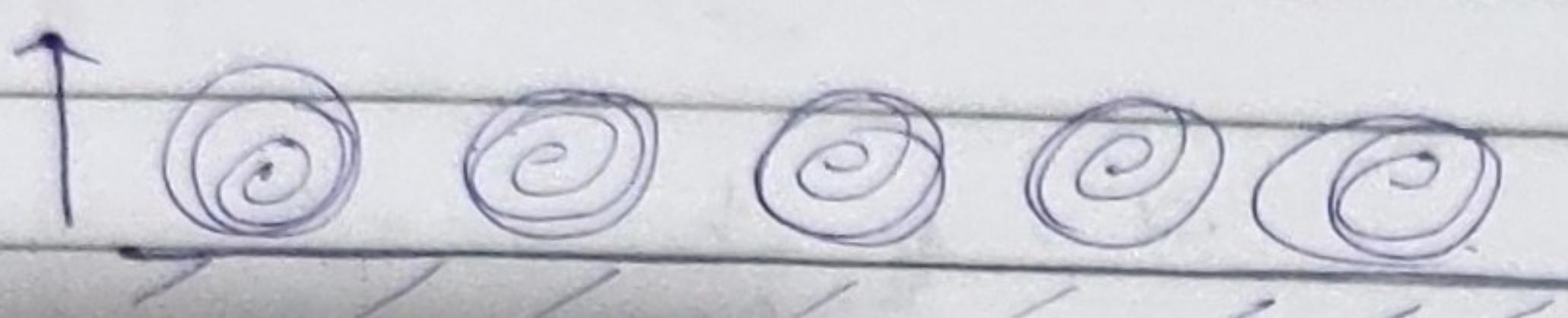
$$\frac{h}{C_p U_\infty} = \frac{C_f \bar{u}_0}{2}$$

$$St_x = \frac{1}{2} C_f x$$

reynolds  
analogy  
assumption  
 $(\gamma = 1)$

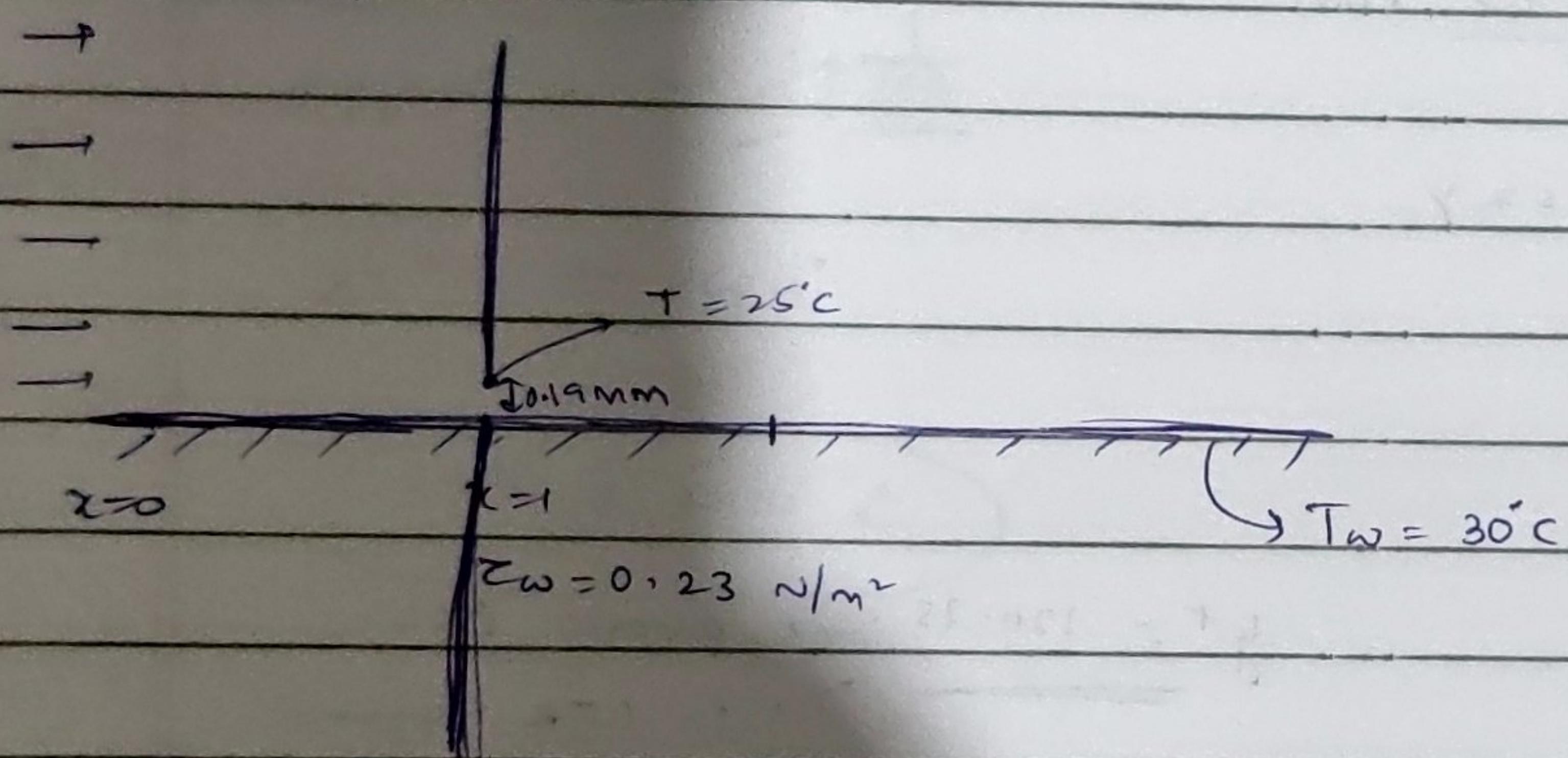
assumption  
for many  
conditions

$$P_{x,t} = 1$$



amount of momentum transferred by eddies = amt of energy transferred by eddies  
 $\underline{e_M \times e_H}$

Q)  $U_{\infty} = 10 \text{ m/s}$ ,  $T_{\infty} = 10^\circ\text{C}$



$$q_{h,v}'' = h \Delta T$$

$$y_{CSL}^+ = 13.2$$

18 mm  $\rightarrow$  CTR region

$$y^+ = y \frac{u^*}{\sqrt{}}$$

$$u^* = \left( \frac{z_0}{\epsilon} \right)^{1/2}$$

$$y^+ = 5.1$$

at  $y_{min} \rightarrow CSL$

$$u^* = \left( \frac{0.23}{1.13} \right)^{1/2}$$

$$\gamma = 16.7 \times 10^4$$

$$q_{h,v}'' = -k \left( \frac{dT}{dy} \right)_{y=0} = y = 0.19 \text{ mm}$$

$$q_{h,v}'' = -k \left[ \frac{5}{0.19 \times 10^{-3}} \right] \checkmark = 15.67 \%$$

$$(b) St = Pr^{-2/3} \frac{C_f x}{2}$$

$$C_{f,x} = Z_0 = \frac{0.23}{\frac{1}{2} C_p u^2} = \frac{0.23}{\frac{1}{2} \times 1.13 \times 16^2}$$

$$St = 2.53 \times 10^{-3}$$

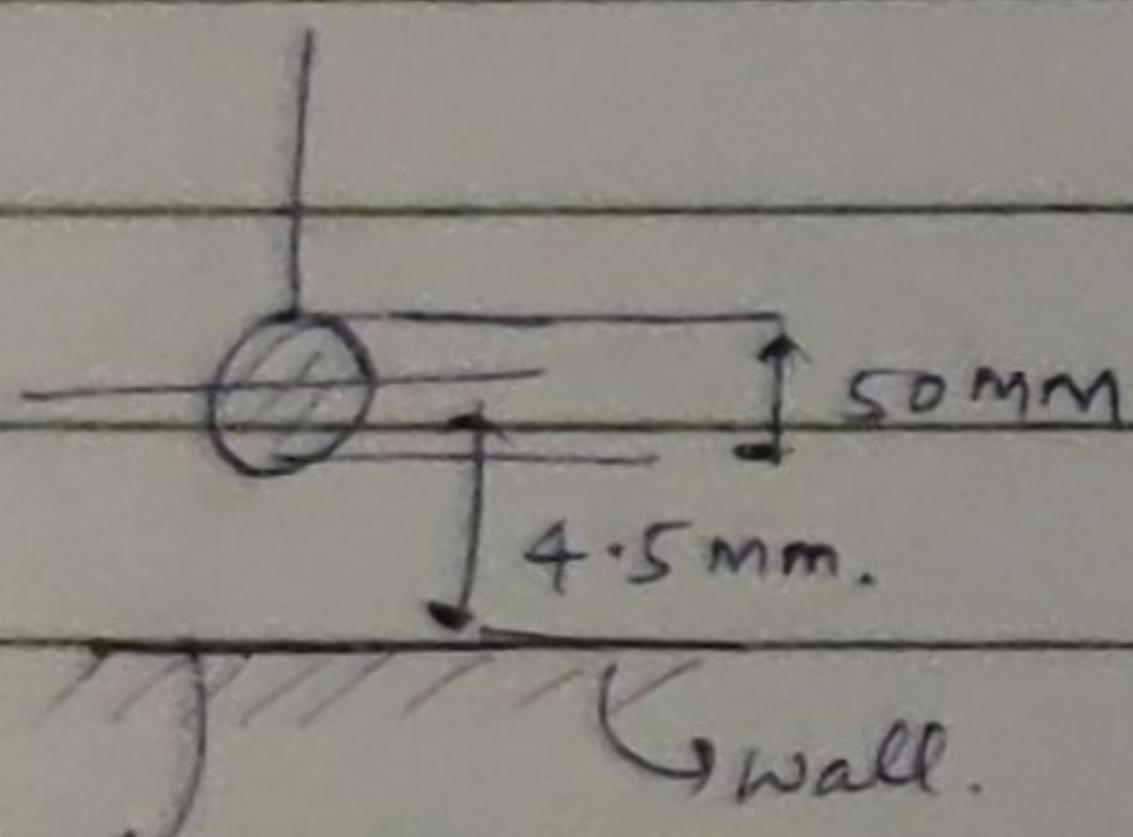
$$St = \frac{h}{\rho C_p u_\infty} = \frac{q_w''}{\rho C_p u_\infty (T_w - T_\infty)}$$

$$h (T_w - T_\infty) = q_w''$$

$$q_w'' = 3 \times 10^{-3}$$

$$\text{error} = \frac{3 - 2.53}{3} \times 100$$

$$= 15.67\%$$



$$T_w = 30^\circ\text{C}$$

$$y^+ = 120.78 \rightarrow \text{above CSL} \\ \rightarrow \text{in CTR regime}$$

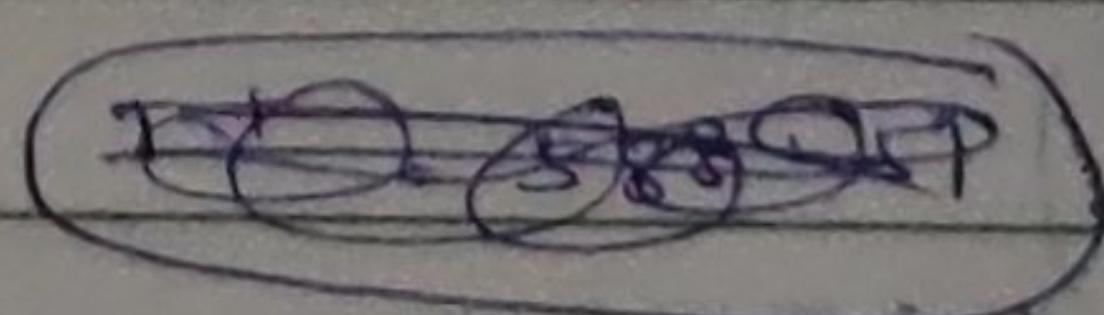
$$T^+ = Pr \frac{y^+}{y_{CSL}^+} + \frac{Pr_t}{K} \ln \left( \frac{y^+}{y_{CSL}^+} \right)$$

$$Pr_f = 0.9$$

$$K = 0.41$$

$$T^+ = 2.195 y^+ + 358$$

$$y_{CSL}^+ = 13.2$$



$$T^+ = 14.10$$

$$T^+ = (T_w - \bar{T}) \frac{\rho C_p u^+}{\nu} \approx 14.10$$

$$\bar{T} = 11^\circ\text{C}$$

$$\frac{d\tau^+}{dy^+} = 2.195 \frac{1}{y^+}$$

$$\left. \frac{d\tau^+}{dy^+} \right|_{y^+ = 121} = \frac{2.195}{121} = 0.018$$

distance is small  $\rightarrow$  assume linear variation

$$\frac{\Delta \tau^+}{d^+} = 0.018$$

$$d^+ = \frac{d \times u^+}{\sqrt{}} = 1.35$$

$$\Delta \tau^+ = 0.024$$

$\rightarrow$  difference  $\rightarrow \underline{\Delta \tau^+}$

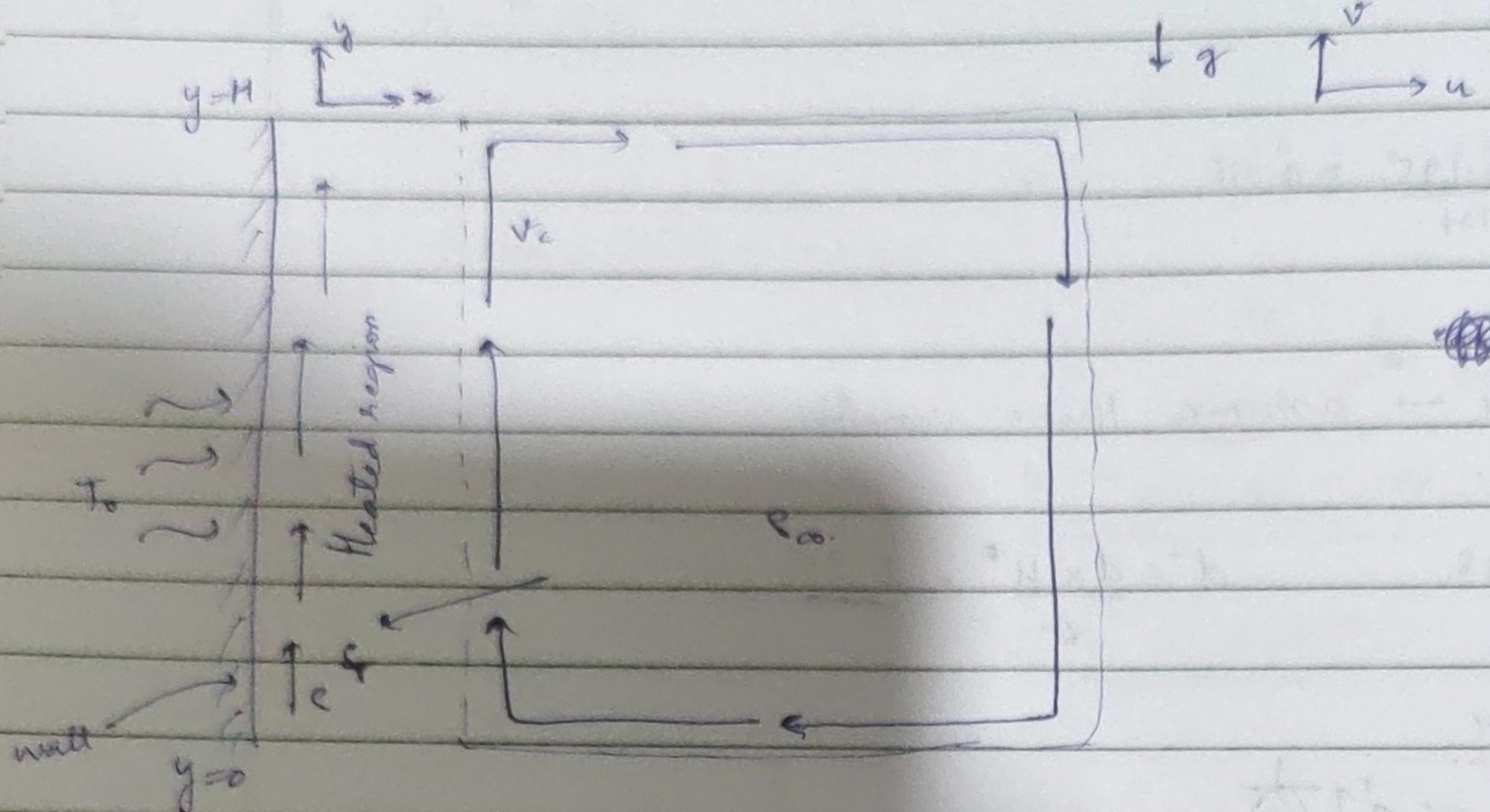
$$\Delta \tau^+ = \underline{\Delta \tau} \text{ eqn } u^*$$

$\tau''$

$$\Delta \tau = 3.2 \times 10^{-2} \text{ s}$$

(e) Yes or No  $\rightarrow$   $\Delta \tau$   $\rightarrow$   $\Pr_t \rightarrow$  will change

## NATURAL CONVECTION



temperature difference  
is the main cause  
of velocity

$$\delta_T \ll H$$

$$\checkmark Q = h (T_a - T_\infty)$$

from solid to liquid

average heat transfer coeff.

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

not required

$$\rho \left( u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} \right) = - \frac{\partial p}{\partial x} + \mu \left( \frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) \Rightarrow \frac{\partial p}{\partial y} = \frac{dp_\infty}{\partial y}$$

$$u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \alpha \left( \frac{\partial T}{\partial x^2} + \frac{\partial T}{\partial y^2} \right) + \text{Additional terms}$$

$$\rho \left( u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} \right) = - \frac{\partial p}{\partial y} + \mu \left( \frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) * \text{eg}$$

$$\frac{dp_\infty}{dy} = - \rho_\infty g$$

because  
buoyancy  
approx

$$\Rightarrow \rho \left( u \frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} \right) = \mu \frac{\partial^2 v}{\partial x^2} + g (\rho_\infty - \rho)$$

Assumption! Effect of temp on density in buoyancy force  
is greater than effect of temp on any other physical property  
( $\rho, \alpha, \mu$ )

$$\frac{1}{\rho} \frac{D\rho}{Dt} \approx 0 \rightarrow \text{incompressible fluid} \Rightarrow \frac{D\rho}{Dt} = 0$$

✓ rate of change of volume of fluid with respect to single particle

$$\beta = \frac{1}{V} \frac{\partial V}{\partial T} = -\frac{1}{\rho_\infty} \frac{d\rho_\infty}{dT} \Big|_\infty$$

$$\rho = \rho_\infty + \frac{\partial \rho}{\partial T} \Big|_\infty (T - T_\infty) + \text{higher order term}$$

$$\rho = \rho_\infty - \rho_\infty \beta_\infty (T - T_\infty)$$

$$(\rho_\infty - \rho) = \rho_\infty \beta_\infty (T - T_\infty)$$

$$\rightarrow \rho \left( \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \right) = \frac{\partial^2 u}{\partial x^2} + \rho_\infty \beta_\infty (T - T_\infty) g$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = \underbrace{\frac{\partial^2 u}{\partial x^2}}_{\text{Inertia}} + \underbrace{\beta_\infty (T - T_\infty) g}_{\text{buoyancy}}$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = \underbrace{\frac{\partial^2 T}{\partial x^2}}_{\text{Conduction}}$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

$$u^* = \frac{u}{V_c}$$

$$v^* = \frac{v}{V_c}$$

$$x^* = \frac{x}{H}$$

$$y^* = \frac{y}{H}$$

$$\theta = \frac{T - T_\infty}{T_0 - T_\infty}$$

$$\left[ \frac{\partial^2 u^*}{\partial x^2} + \frac{\partial^2 v^*}{\partial y^2} \right] = \left[ \frac{V_c}{H^2} \frac{\partial^2 v^*}{\partial x^2} + \beta_\infty \theta g (T_0 - T_\infty) \right] \frac{H}{V_c^2}$$

$$\frac{\partial^2 u^*}{\partial x^2} + \frac{\partial^2 v^*}{\partial y^2} = \frac{H}{V_c^2} \frac{\partial^2 v^*}{\partial x^2} + \left[ \frac{\beta_\infty g (T_0 - T_\infty) H}{V_c^2} \right] \theta$$

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natural vs forced

will tell you whether natural convection will dominate or not

$$\frac{Bo \cdot g \cdot \Delta T \cdot H}{\sqrt{2}} \times \frac{\sqrt{2}}{V_c \cdot H^2} \cdot \frac{1}{Re^2}$$

$$Gr_H$$

$$\frac{Gr_H}{Re^2} = (Ri) \text{ (Richardson's number)}$$

will tell you  
whether natural or forced convection  
dominates

### Scaling analysis

$$x \sim \delta_T$$

$$y \sim H$$

$$v \sim V_c \text{ (characteristic velocity)}$$

$$u \sim u$$

$$\Delta T \sim T_0 - T_\infty$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \sim 0$$

$$\sim \frac{u}{\delta_T} \sim \frac{V_c}{H}$$

$$u \sim \frac{V_c \Delta T}{H}$$

$$u \frac{\partial T}{\partial x} + v \frac{\partial T}{\partial y} = \alpha \left( \frac{\partial^2 T}{\partial x^2} \right)$$

$$\sim \frac{u \Delta T}{\delta_T} \sim \frac{V_c \Delta T}{H} \sim \alpha \frac{\Delta T}{\delta_T^2}$$

$$\frac{V_c \Delta T}{H} \sim \frac{V_c \Delta T}{\delta_T}$$

$$\frac{V_c \Delta T}{H} \sim \alpha \frac{\Delta T}{\delta_T^2}$$

$$\delta_T \sim \sqrt{\frac{H}{V_c}}$$

$$V_c \sim \frac{\alpha H}{\delta_T^2}$$

$\delta_T, V_c$  are unknown

$$\frac{\partial v}{\partial x} + v \frac{\partial v}{\partial y} = \sqrt{\frac{\partial^2 v}{\partial x^2}} + g\beta(T - T_{\infty})$$

$\downarrow$        $\downarrow$   
 $\sim \frac{v_c \delta_T}{H} \sim \frac{v_c v_c}{H}$        $\sim \sqrt{\frac{v_c}{\delta_T^2}} \sim g\beta \Delta T$   
 $\sim \frac{v_c^2}{H} \sim \frac{v_c^2}{H}$        $\sim \sqrt{\frac{v_c}{\delta_T^2}} \sim g\beta \Delta T$   
*Inertia*      *Viscous / friction*      *Bouyancy*      *omnipresent*

$$\begin{aligned}
 \text{Inertia} &= \frac{v_c^2}{H} = \frac{\alpha^2 H^2}{\delta_T^4 \times \frac{1}{H}} = \frac{\alpha^2 H^3}{\delta_T^4} = \frac{\alpha^2 H^3}{\delta_T^4 (g\beta \Delta T)} = \frac{\alpha^2 n}{\delta_T^4} \times \frac{n^3 \gamma}{H^3} \\
 \text{Bouyancy} & g\beta \Delta T = \frac{1}{\frac{g\beta \Delta T H^3}{\alpha^2}} = \left(\frac{H}{\delta_T}\right)^4 = \frac{1}{R_{\text{an}}^4}
 \end{aligned}$$

$$\begin{aligned}
 \text{Friction} &= \frac{v_c}{\delta_T^2} = \frac{\alpha n}{\delta_T^4} \times \frac{H^3}{n^3} \\
 \text{Bouyancy} & \frac{1}{g\beta \Delta T}
 \end{aligned}$$

$$R_{\text{an}} = \frac{1}{\frac{g\beta \Delta T H^3}{\alpha^2}} \times \left(\frac{H}{\delta_T}\right)^4 = R_{\text{an}}^{-1} \left(\frac{H}{\delta_T}\right)^4$$

$$\Rightarrow \sim R_{\text{an}}^{-1} \text{Pr}^{-1} \left(\frac{H}{\delta_T}\right)^4 \sim R_{\text{an}}^{-1} \left(\frac{H}{\delta_T}\right)^4 \sim 1$$

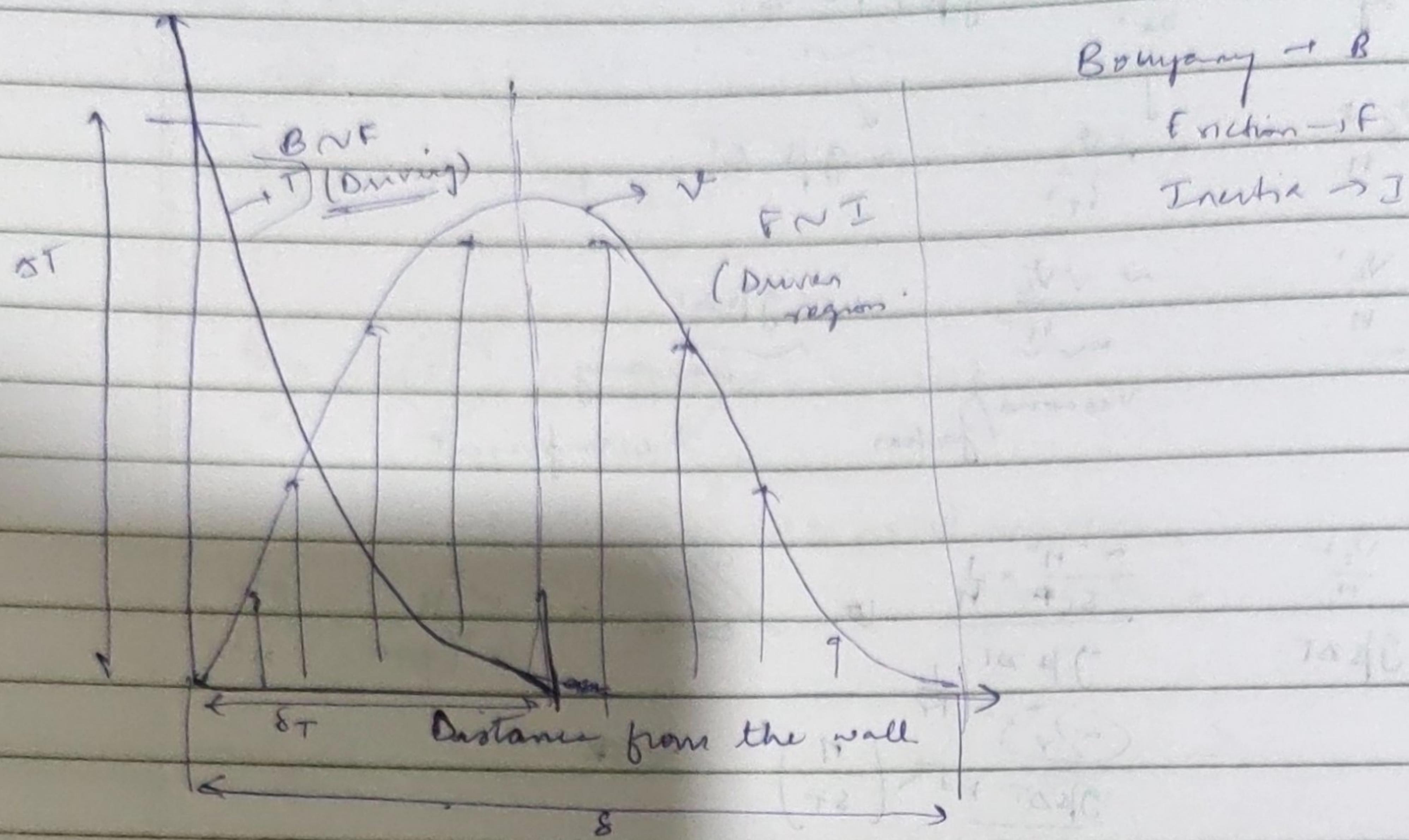
high  $\text{Pr} \rightarrow \text{Pr} \gg 1 \rightarrow$  Inertia has very low magnitude.

$$R_{\text{an}}^{-1} \left(\frac{H}{\delta_T}\right)^4 \sim 1$$

$$\boxed{\delta_T \sim H \times R_{\text{an}}^{-1/4}}$$

$$h \sim \frac{k}{\delta_T}$$

$$Nu \sim \frac{h H}{k} \sim R_{\text{an}}^{1/4}$$



Friction  $\sim$  Inertia

$$\frac{\delta \nu_c}{\delta^2} \sim \frac{\nu_c^2}{H}$$

$$\delta^2 \sim \frac{\nu H}{\frac{\alpha H}{\delta^2}}$$

$$\delta^2 \sim \left(\frac{\nu}{\alpha}\right) \times \delta T^2$$

$$\delta^2 \sim \rho \nu R_{in}^{-1/2} H^2$$

$$\delta \sim \rho \nu^{1/2} R_{in}^{-1/4} H$$

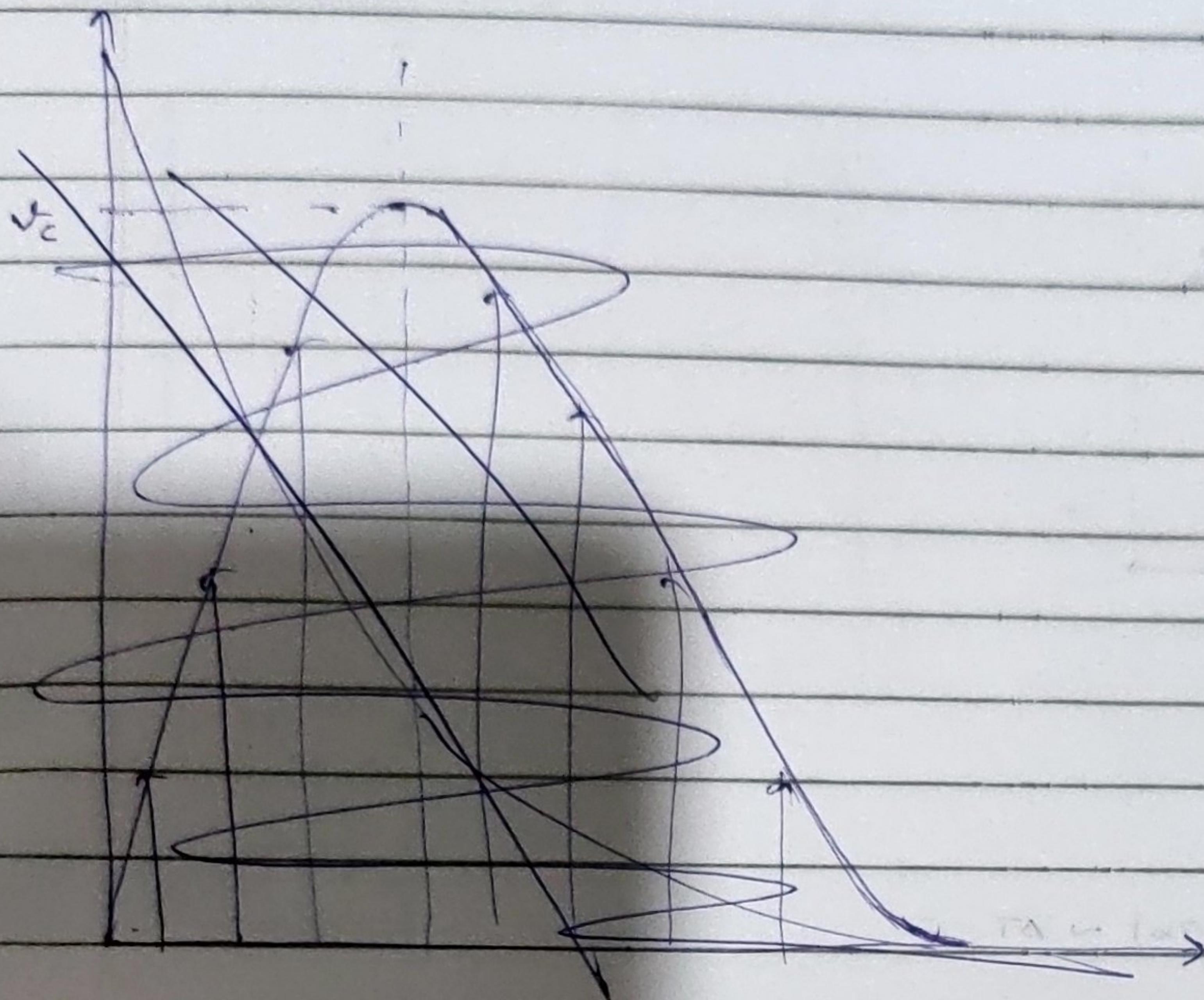
$$\frac{\delta}{\delta T} \sim \frac{\rho \nu^{1/2} H R_{in}^{-1/4}}{H R_{in}^{-1/4}} \sim \rho \nu^{1/2}$$

low Pr  $\quad \text{Pr} \ll 1$

$$R_{in}^{-1} \Pr^{-1} \left(\frac{H}{\delta T}\right)^4 \sim 1$$

$$\delta T \sim H (Bo)^{-1/4} \quad Bo \rightarrow \text{Boussinesq Number.}$$

$$Nu \sim Bo^{1/4}$$



$Gr \rightarrow$  idea about scale of viscous sub layer.

Reynolds

$$Gr \approx B \sim 10^8 - 10^{10}$$

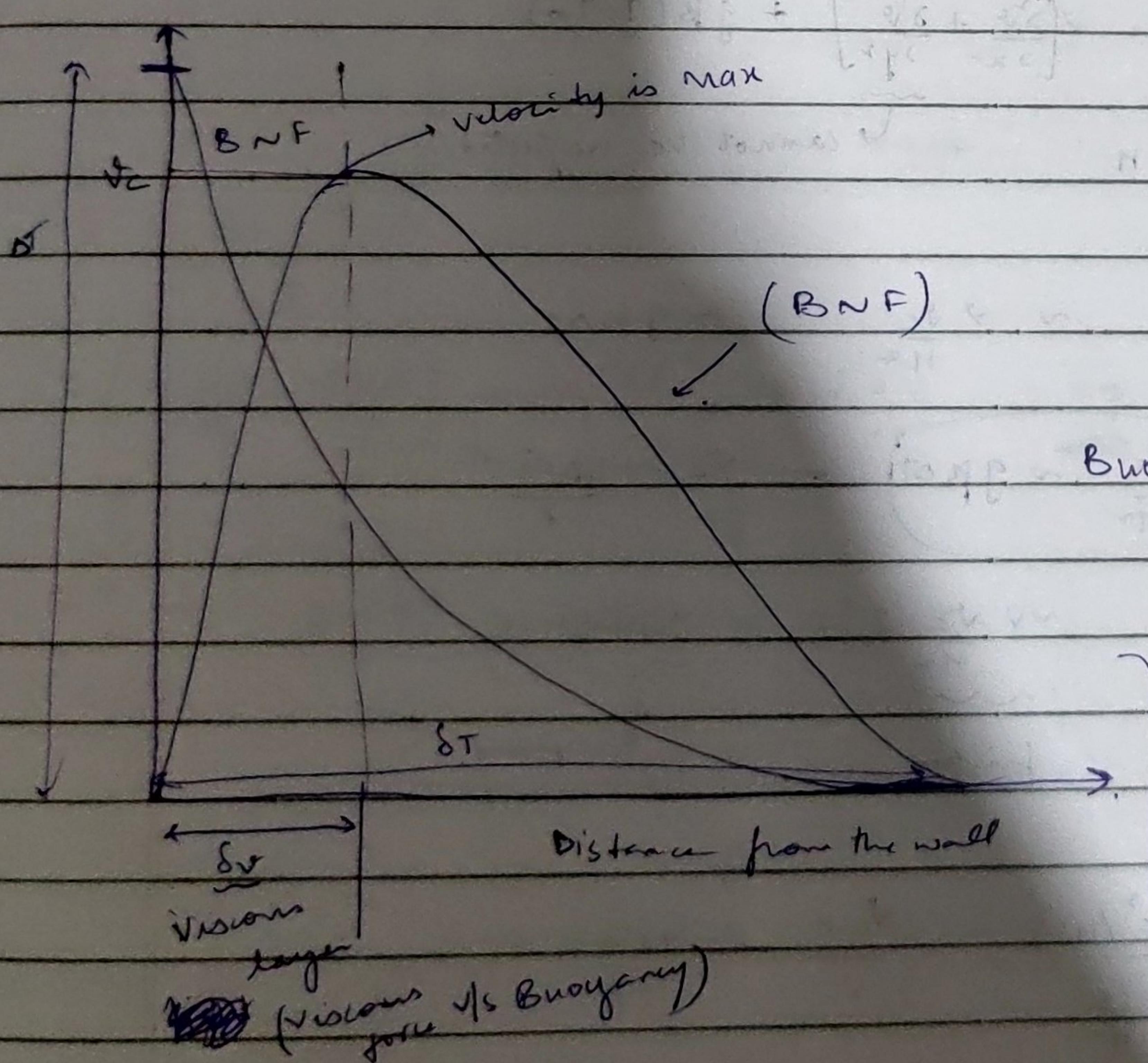
$$F \sim \sim \sim$$

$$Gr^{1/4} = \frac{H}{\delta_T}$$

$$Ra_n^{1/4} = \frac{H}{\delta_T}$$

$$Pr > 1$$

$$Bo^{1/4} = \frac{H}{\delta_T} \quad Pr < 1$$



Buoyancy  $\sim$  friction

$$g/k\delta T \sim \nu \frac{V_c}{\delta_T^2}$$

$$\delta_V^2 \sim \nu \frac{\alpha H}{\delta_T^2}$$

$$\delta_V^2 \sim H^4$$

$$g/k\delta T n^3 \times \delta_T^2$$

$$V_c \sim \frac{H}{\delta_T^2}$$

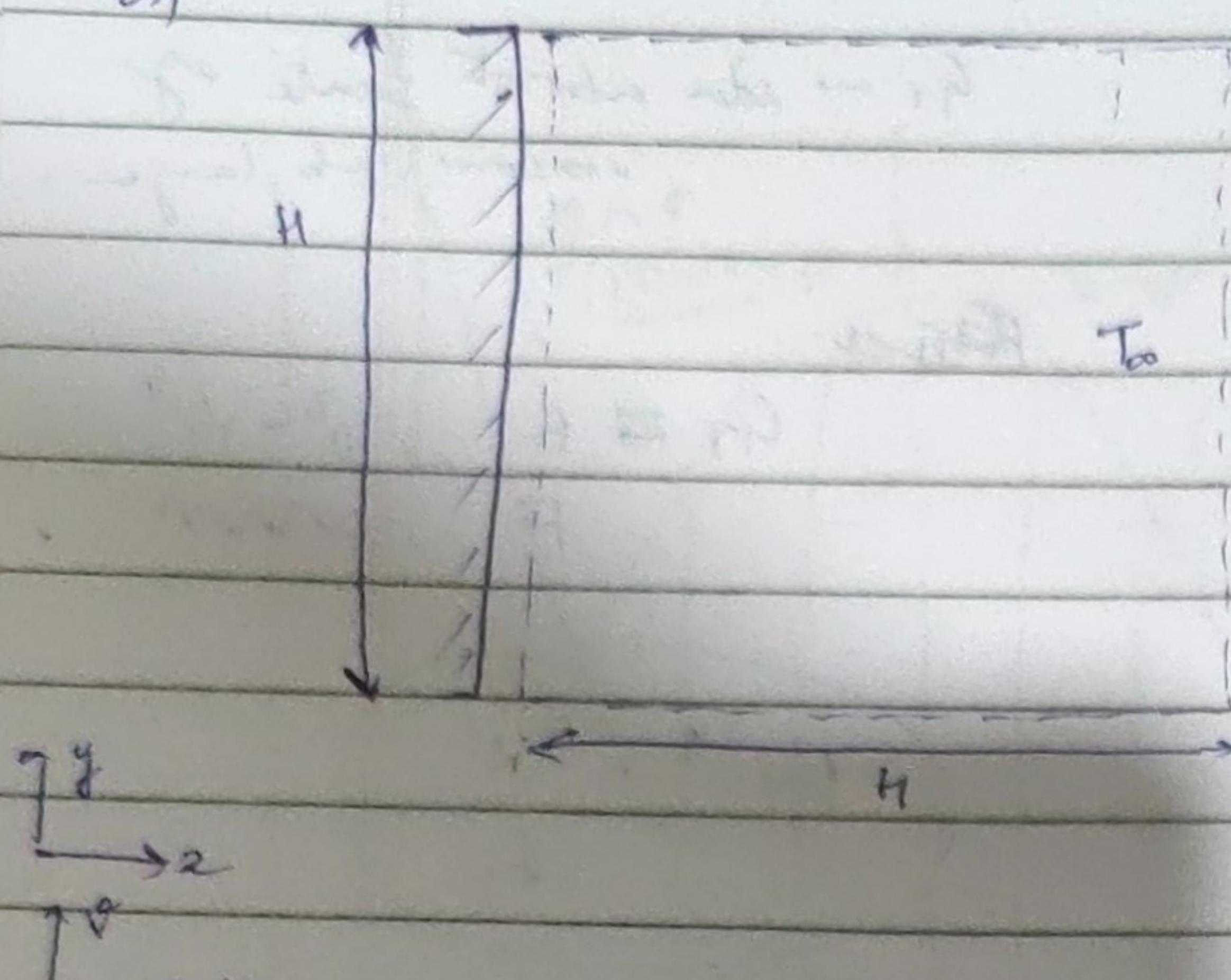
~~$$\delta_V^2 \sim H^2 \rho_{\text{air}}^{1/2} Ra_n^{1/2}$$~~

$$\delta_V^2 \sim H^2 \left( \frac{Pr}{Ra_n} \right)^{1/2}$$

$$\delta_V^2 \sim H^2 (Gr)^{1/2}$$

$$\delta_V \sim H (Gr)^{1/4}$$

8)



$$(T - T_0) \sim (T_0 - T_\infty) \sim \Delta T$$

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = \sqrt{\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 v}{\partial y^2}} + g \beta (T - T_\infty)$$

$$x \sim H, y \sim H$$

cannot be neglected

$$\sim u \frac{v_c}{H} \sim \frac{v_c^2}{H} \sim \frac{v v_c}{H^2} \sim g \beta \Delta T$$

$$\left( \frac{v v_c}{H^2} \sim g \beta \Delta T \right) \rightarrow v_c \sim \frac{g \beta \Delta T H^2}{v}$$

$$\sim \frac{v_c^2}{H} \sim \frac{v v_c}{H^2} \sim g \beta \Delta T$$

inertia      friction      Buoyancy

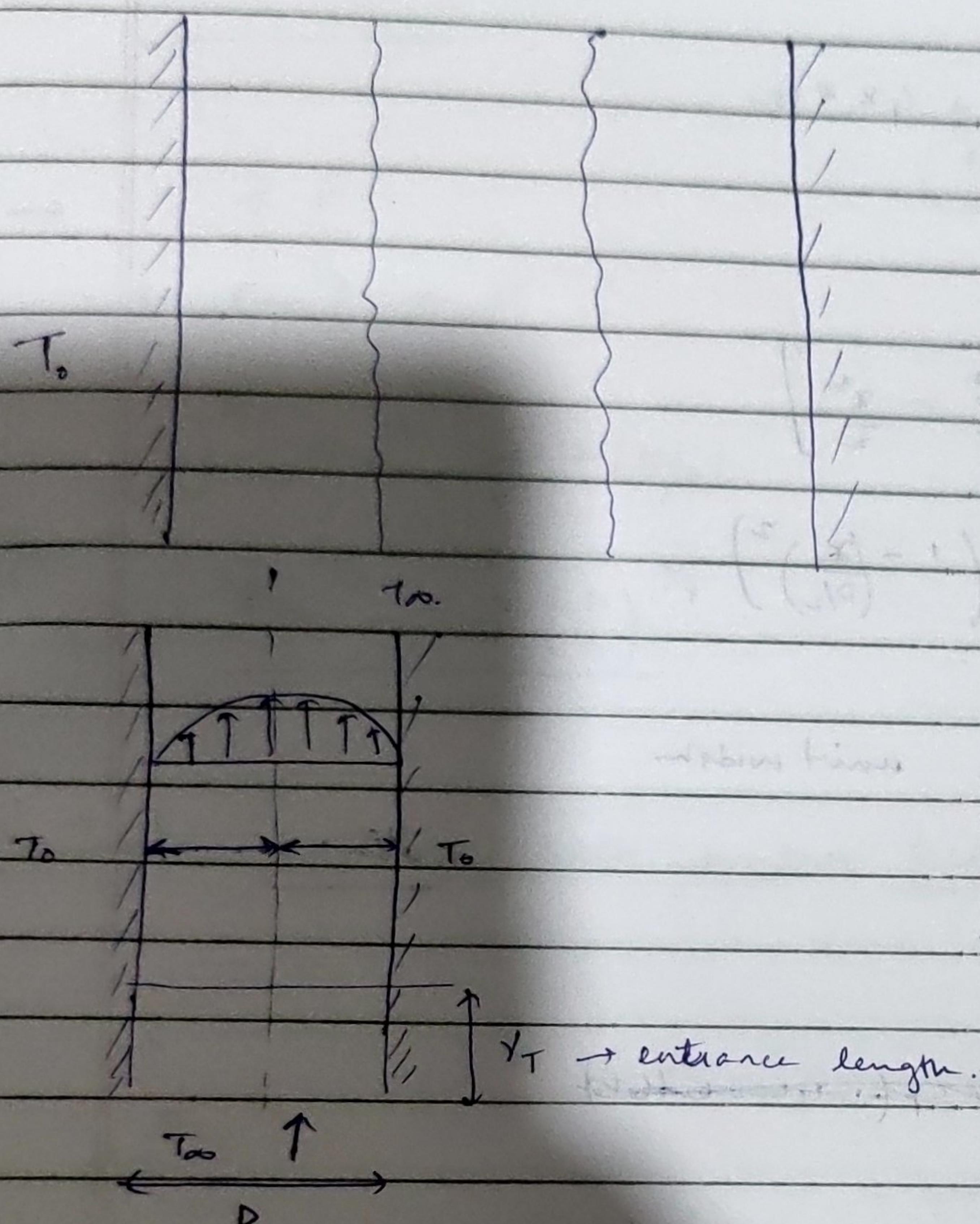
$$\frac{v_c^2/H}{\gamma \frac{v_c}{H^2}} = \frac{v_c H}{\gamma} = \text{Gr.}$$

$$\Rightarrow \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \sim \alpha \frac{\partial^2 T}{\partial x^2}$$

$$\sim \frac{\partial u}{\partial x} \sim v_c \frac{\Delta T}{H} \sim \alpha \frac{\Delta T}{H^2}$$

$$\sim v_c \frac{\Delta T}{H} \sim H + \frac{\Delta T}{H}$$

Q) Solar chimney.



$$\rho \left( u \frac{\partial v}{\partial x} + v \frac{\partial u}{\partial y} \right) = - \left( \frac{\partial p_\infty}{\partial y} \right) + \frac{\rho g v^2}{2x^2} - \rho g$$

fully developed.

$$u \approx v \quad \frac{\partial v}{\partial x} \approx 0 \quad \frac{\partial u}{\partial y} \approx 0$$

$$= \frac{\rho g v^2}{2x^2} - g(\rho - \rho_\infty)$$

$$= \frac{\rho g v^2}{2x^2} + g \beta_\infty (T - T_\infty)$$

$$0 = \frac{\partial v}{\partial x^2} + g \beta_\infty (T - T_\infty)$$

$$\frac{\partial v}{\partial x^2} = - g \beta_\infty (T - T_\infty) \quad \begin{matrix} \text{we need energy eqn} \\ \text{to solve this eqn.} \end{matrix}$$

$$\text{BC} \rightarrow x = \frac{D}{2} \quad v = 0$$

$$x = 0, \quad \frac{\partial v}{\partial x} = 0$$

$$(T - T_\infty) \ll (T_0 - T_\infty) \rightarrow \text{assumption}$$

$$\frac{d^2V}{dx^2} = -g \beta_{\infty} (T_0 - T_{\infty})$$

$$V(x) = -\frac{g \beta_{\infty} (T_0 - T_{\infty})}{24} x^2 + C_2$$

(C<sub>2</sub>)

$$V(x) = -\frac{g \beta_{\infty} (T_0 - T_{\infty})}{24} \left[ \frac{D^2}{8} - \frac{x^2}{2} \right]$$

$$V(x) = \frac{g \beta_{\infty} (T_0 - T_{\infty}) D^2}{8} \left( 1 - \left( \frac{x}{D/2} \right)^2 \right)$$

mass flowrate =  $\rho V_{avg} A$

$$\dot{m} = \rho \int_0^{D/2} V(x) dx$$

unit width.

width = w

$$\dot{q} = \dot{m} c_p \Delta T$$

~~if  $\dot{m} \ll \dot{q}$  (constant width)~~

(out-in.)

$$\text{average heat flux} \rightarrow \frac{d}{dx} \dot{q}''_{0-H} = \frac{\dot{q}}{2 \times (w \times H)}$$

w=1

$$Nu_{0-H} = \frac{h_{0-H} \times H}{k} = \frac{c_{\infty} g \beta_{\infty} (\Delta T)^2 D^3 c_p}{24 \times H \times k \Delta T} = \frac{Ra}{24}$$

( $Y_T < H$ )

$$\rightarrow Pr > 1, \quad \delta \sim H R_{\text{an}}^{-1/4}$$

$$\frac{D}{2} \sim Y_T \cdot H \cdot R_{\text{an}}^{-1/4} \quad \left( Y_T \sim \frac{D}{2} R_{\text{an}}^{1/4} \right)$$

$$\Rightarrow Y_T < H \rightarrow \frac{D}{2} R_{\text{an}}^{1/4} < H$$

$$\frac{Ra_n}{Ra} = \left(\frac{H}{D}\right)^3$$

$$\rightarrow \gamma_T < H$$

$$\frac{D}{2} Ra_n^{1/4} < H$$

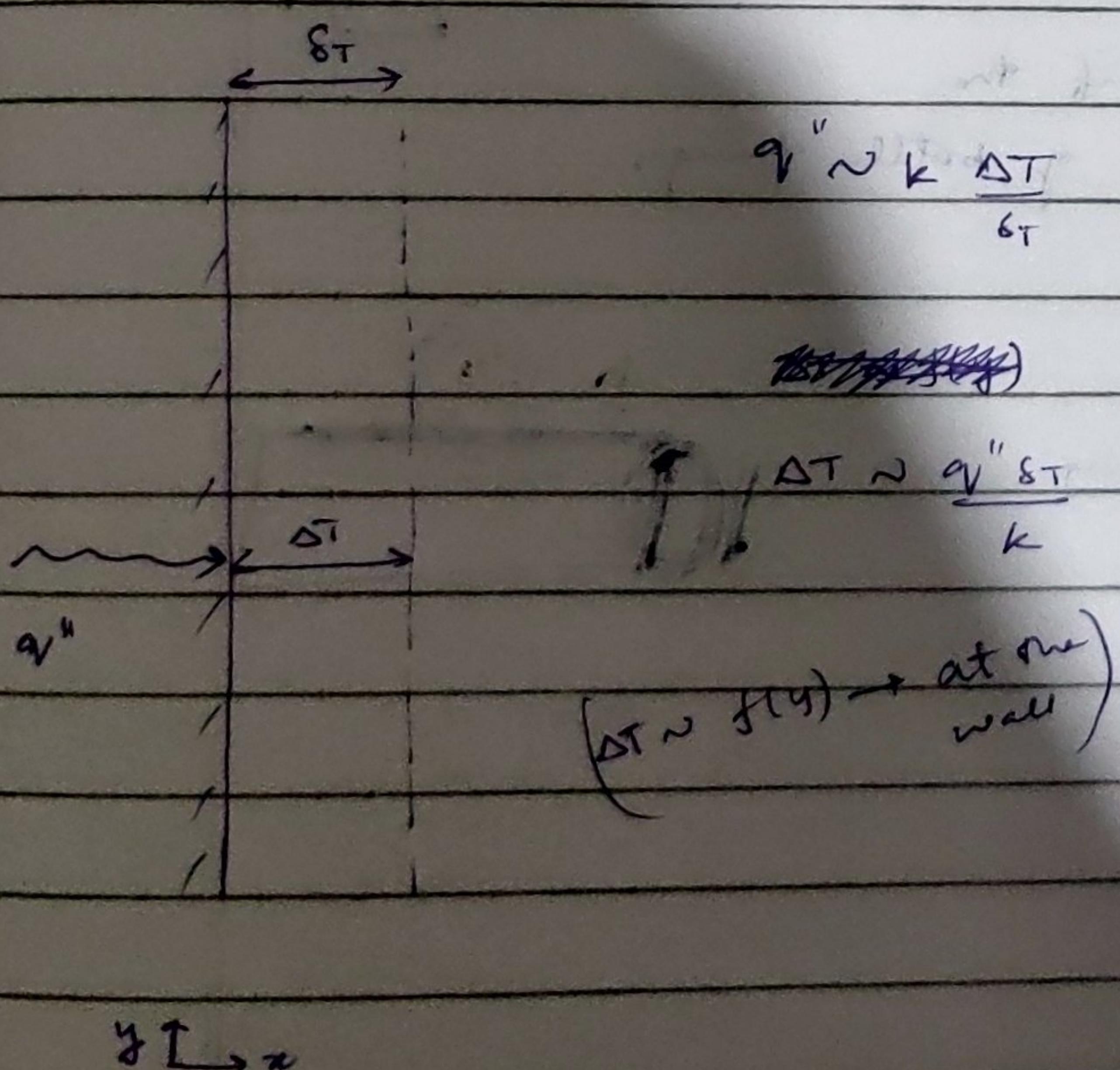
$$\left(\frac{H}{D}\right)^{3/4} Ra_n^{1/4} < 2(H/D)$$

$$Ra_n^{1/4} < 2 \left(\frac{H}{D}\right)^{1/4}$$

Problem 4.13

FIN analysis

Natural convection with uniform heat flux



$$q'' \sim k \frac{\Delta T}{\delta_T} \rightarrow \text{scaling}$$

$$Pr > 1$$

$$\Delta T \sim q'' \delta_T / k$$

$$\delta_T \sim H (Ra_n)^{-1/4}$$

$$\sim H \left( \frac{g \beta \Delta T H^3}{\nu \alpha} \right)^{-1/4}$$

$$\delta_T^4 \sim H^4 \left[ \frac{g \beta q'' \delta_T H^3}{k} \right]^{-1}$$

$$\delta_T^5 \sim H^5 \left[ \frac{\nu \alpha k}{g \beta \Delta T q'' H^4} \right]$$

$$\boxed{\delta_T \sim H (Ra_n^*)^{-1/5}}$$

$Ra_n^* \rightarrow$  Rayleigh number based on heat flux

$P_r < 1$

$$St \sim H (Pr Ra_H)^{-1/5}$$

Problem 3

Assumption

- ① resistance of bottle is negligible
- ② Natural convection only
- ③ lump capacitance model ( $T = f(t) \rightarrow$  only)

$$\dot{Q} = \frac{m_C \rho_{co} \Delta T}{dt} \Big|_{co} \quad (co \rightarrow \text{cold drink})$$

rate at  
which cold  
drink is losing  
heat

$$\dot{Q} = h_A (T_{co} - T_{air})$$

area of  
bottle

$$m_C \rho_{co} \frac{d\Delta T}{dt} \Big|_{co} = h_A (T_{co} - T_{air}) \rightarrow \text{at } ss$$

will change if the

$$\frac{1}{h} = \frac{1}{h_{air}} + \frac{1}{h_{co}}$$

orientation of bottle changes.

$$\frac{m_C \rho \Delta T}{t} \sim h A \Delta T \rightarrow \text{scaling}$$

$$t \sim \frac{m_C \rho}{h A}$$

$t_1 \rightarrow$  time taken to cool  $\rightarrow$  vertical

$t_2 \rightarrow$  horizontal

Nu<sub>air</sub>

Nu<sub>co</sub>

$$\frac{t_1}{t_2} = \frac{h_2}{h_1}$$

$$\frac{h_{co}}{h_{air}} = \frac{k (Ra_H)^{1/4}}{k (Ra_H)^{1/4}} \sim 58.1 \checkmark$$

$$\frac{1}{h} = \frac{1}{h_{air}} + \frac{1}{h_{co}}$$

$$h \approx h_{air}$$

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