1 Two-Body Problem

To discuss the solution presented in [?], we need to formulate the two-body problem in terms of Delaunay elements. Thus, this section is dedicated to canonical transformations and the Hamilton-Jacobi Equation. Starting from a Hamiltonian formulation in spherical polar coordinates, we will first use Hamilton-Jacobi equations to derive canonical coordinates corresponding to zero Hamiltonian and then use a canonical transformation to derive a Hamiltonian formulation in terms of Delaunay Elements.

The discussion in this section is roughly based on sections 6 and 7 of [?].

1.1 Hamiltonian Formulation

Consider the problem of a point particle with mass m in a gravitational potential V, where the kinetic and potential energy can be expressed via the equations

$$T = \frac{m}{2}(\dot{r}^2 + r^2\dot{\theta}^2 + r^2\cos^2\theta\dot{\phi})$$
 (1)

$$V = -\frac{\mu m}{r} \tag{2}$$

in terms of spherical polar coordinates (r, θ, ϕ) and canonical momentums

$$p_r = \frac{\partial T}{\partial \dot{r}} = m\dot{r} \tag{3}$$

$$p_{\theta} = \frac{\partial T}{\partial \dot{\theta}} = mr^2 \dot{\theta} \tag{4}$$

$$p_{\phi} = \frac{\partial T}{\partial \dot{\phi}} = mr^2 \cos^2 \theta \dot{\phi}. \tag{5}$$

Thus, the Hamiltonian (using the notation in [?]) can be expressed

$$F(r,\theta,\phi,p_r,p_\theta,p_\phi) = T + V = \frac{1}{2m} \left(p_r^2 + \frac{p_\theta^2}{r^2} + \frac{p_\phi^2}{r^2 \cos^2 \theta} \right) - \frac{\mu m}{r}.$$
 (6)

In this section, we will use canonical transformations and the Hamilton-Jacobi equation to derive the Hamiltonian in terms of canonical coordinates and momentums called Delaunay Elements.

1.1.1 Canonical Transformations

Suppose F^* is a Hamiltonian obtained from an another Hamiltonian F with a **generating function** (called **determining function** in [?]) S. Then,

$$F^*(q', p') = F(q, p) + \frac{\partial S}{\partial t}$$
 (7)

In this document, we only use generating functions of the type S = S(q, p'), which satisfy

$$p_i = \frac{\partial S}{\partial q_i}, \quad q_i' = \frac{\partial S}{\partial p_i'}$$
 (8)

1.1.2 Hamilton-Jacobi Equation

With Hamilton-Jacobi equation, we can derive canonical coordinates, which do not appear explicitly in the transformed Hamiltonian. Then, each canonical coordinate and momentum is independent of time. That is, for $F = F(r, \theta, \phi, p_r, p_\theta, p_\phi)$ we wish find a generating function S such that

$$F\left(r,\theta,\phi,\frac{\partial S}{\partial r},\frac{\partial S}{\partial \theta},\frac{\partial S}{\partial \phi}\right) + \frac{\partial S}{\partial t} = 0. \tag{9}$$

Substitution to (6), yields

$$\frac{1}{2m} \left[\left(\frac{\partial S}{\partial r} \right)^2 + \frac{1}{r^2} \left(\frac{\partial S}{\partial \theta} \right)^2 + \frac{1}{r^2 \cos^2 \theta} \left(\frac{\partial S}{\partial \phi} \right)^2 \right] + \frac{\partial S}{\partial t} - \frac{\mu m}{r} = 0.$$
 (10)

We attempt to solve (10) via separation of variables

$$S(r,\theta,\phi,t) = S_r(r) + S_{\theta}(\theta) + S_{\phi}(\phi) + S_t(t). \tag{11}$$

This yields

$$\left(\frac{dS_r}{dr}\right)^2 + \frac{1}{r^2} \left(\frac{dS_\theta}{d\theta}\right)^2 + \frac{1}{r^2 \cos^2 \theta} \left(\frac{dS_\phi}{d\phi}\right)^2 = 2m \left(-\frac{dS_t}{dt} + \frac{\mu m}{r}\right). \tag{12}$$

The dependence w.r.t. variables ϕ and t is limited to $dS_{\phi}/d\phi$ and dS_{t}/dt . Thus, there exists α_{2} and α_{1} such that

$$\frac{dS_{\phi}}{d\phi} = -\alpha_2, \quad \frac{dS_t}{dt} = -\alpha_1. \tag{13}$$

The equation (12) can be now written

$$\left(\frac{dS_r}{dr}\right)^2 + \underbrace{\frac{1}{r^2} \left(\frac{dS_\theta}{d\theta}\right)^2 + \frac{\alpha_2^2}{r^2 \cos^2 \theta}}_{\alpha_1/r^2} = 2m\left(\alpha_1 + \frac{\mu m}{r}\right) \tag{14}$$

and denoting the middle two terms with α_1/r^2 we obtain

$$\frac{dS_r}{dr} = \sqrt{2m\left(\alpha_1 + \frac{\mu m}{r}\right) - \frac{\alpha_1}{r^2}}.$$
 (15)

Similarly for S_{θ}

$$\left(\frac{dS_{\theta}}{d\theta}\right)^{2} = 2mr^{2}\left(\alpha_{1} + \frac{\mu m}{r}\right) - r^{2}\left(\frac{dS_{r}}{dr}\right)^{2} - \frac{\alpha_{2}^{2}}{\cos^{2}\theta} = \alpha_{3}^{2} + \frac{\alpha_{2}^{2}}{\cos^{2}\theta}.$$
 (16)

or

$$\frac{dS_{\theta}}{d\theta} = \sqrt{\alpha_3^2 - \frac{\alpha_2^2}{\cos^2 \theta}} \tag{17}$$

The generating function satisfying (9) can be thus written

$$S = -\alpha_1 t + \alpha_2 \phi + \int d\theta \sqrt{\alpha_3^2 - \frac{\alpha_2^2}{\cos^2 \theta}} + \int dr \sqrt{2m \left(\alpha_1 + \frac{\mu m}{r}\right) - \frac{\alpha_1}{r^2}}$$
 (18)

We will select the canonical momentums

$$p_1' := \alpha_1 = mh \tag{19}$$

$$p_2' := \alpha_2 = m\sqrt{a\mu(1 - e^2)}\cos I$$
 (20)

$$p_3' := \alpha_3 = m\sqrt{a\mu(1 - e^2)}$$
 (21)

To derive (19), we can compute

$$\alpha_1 = -\frac{\partial S}{\partial t} = F = T + V = mh,\tag{22}$$

where $h = v^2/2 - \mu/r$ is the **energy integral**.

Derivation of (20) is somewhat more complicated. Note that in spherical polar coordinates

$$x = r\cos\theta\cos\phi$$
$$y = r\cos\theta\sin\phi$$
$$z = r\sin\theta$$

and

$$\dot{x} = \dot{r}\cos\theta\cos\phi - r\dot{\theta}\sin\theta\cos\phi - r\dot{\phi}\cos\theta\sin\phi
\dot{y} = \dot{r}\cos\theta\sin\phi - r\dot{\theta}\sin\theta\sin\phi + r\dot{\phi}\cos\theta\cos\phi
\dot{z} = \dot{r}\sin\theta + r\dot{\theta}\cos\theta$$
(23)

Now, we can note that that the z-component of the angular momentum vector can be expanded in spherical polar coordinates

$$k_{z} = (\mathbf{r} \times \dot{\mathbf{r}}) \cdot \hat{z} = \begin{vmatrix} 0 & 0 & 1 \\ x & y & 0 \\ \dot{x} & \dot{y} & 0 \end{vmatrix}$$

$$= x\dot{y} - y\dot{x}$$

$$= r\cos\theta\cos\phi(\dot{r}\cos\theta\sin\phi - r\dot{\theta}\sin\theta\sin\phi + r\dot{\phi}\cos\theta\cos\phi)$$

$$- r\cos\theta\sin\phi(\dot{r}\cos\theta\cos\phi - r\dot{\theta}\sin\theta\cos\phi - r\dot{\phi}\cos\theta\sin\phi)$$

$$= r\dot{r}\cos^{2}\theta(\cos\phi\sin\phi - \cos\phi\sin\phi)$$

$$+ r^{2}\dot{\theta}\cos\phi\sin\phi(\sin\theta\cos\theta - \sin\theta\cos\theta)$$

$$+ r^{2}\dot{\phi}\cos^{2}\theta(\cos^{2}\phi + \sin^{2}\phi)$$

$$= r^{2}\cos^{2}\theta\dot{\phi}.$$
(24)

On the other hand, for elliptic orbits

$$a = \frac{k^2}{\mu(1 - e^2)},\tag{25}$$

Thus,

$$\alpha_2 = \frac{\partial S}{\partial \phi} = p_\phi = mr^2 \cos^2 \theta \dot{\phi} = mk_z = m\sqrt{a\mu(1 - e^2)} \cos I \tag{26}$$

To derive (21), we can compute

$$\alpha_{3} = \sqrt{\left(\frac{\partial S}{\partial \theta}\right)^{2} + \frac{\alpha_{2}^{2}}{\cos^{2}\theta}}$$

$$= \sqrt{p_{\theta}^{2} + \frac{p\phi^{2}}{\cos^{2}\theta}}$$

$$= mr^{2}\sqrt{\dot{\theta}^{2} + \cos^{2}\theta\dot{\phi}^{2}} = mr^{2}\dot{f} = mk = m\sqrt{a\mu(1 - e^{2})},$$
(27)

where $\dot{f} = \dot{\theta}^2 + \cos^2\theta \dot{\phi}^2$ is the time derivative of the natural anomaly since the sum is the square of total scalar angular speed.

The new canonical coordinates can be computed

$$q_1' = \frac{\partial S}{\partial \alpha_1} = -t + \frac{\partial}{\partial \alpha_1} \int dr \sqrt{2m \left(\alpha_1 + \frac{m\mu}{r}\right) - \frac{\alpha_3^2}{r^2}}$$

$$= -t + \int m \left[2m \left(\alpha_1 + \frac{m\mu}{r}\right) - \frac{\alpha_3^2}{r^2} \right]^{-1/2} dr$$

$$= -t + I_1$$
(28)

$$q_2' = \frac{\partial S}{\partial \alpha_2} = \phi + \frac{\partial}{\partial \alpha_2} \int d\theta \sqrt{\alpha_3^2 - \frac{\alpha_2^2}{\cos^2 \theta}}$$

$$= \phi - \int \frac{\alpha_2}{\cos^2 \theta} \left(\alpha_3^2 - \frac{\alpha_2^2}{\cos^2 \theta}\right)^{-1/2} d\theta$$

$$= \phi - I_2 \alpha_2$$
(29)

$$q_{3}' = \frac{\partial S}{\partial \alpha_{3}} = \frac{\partial}{\partial \alpha_{3}} \int \sqrt{\alpha_{3}^{2} - \frac{\alpha_{2}^{2}}{\cos^{2}\theta}} d\theta + \frac{\partial}{\partial \alpha_{3}} \int \sqrt{2m\left(\alpha_{1} + \frac{\mu m}{r}\right) - \frac{\alpha_{3}^{2}}{r^{2}}} dr$$

$$= \alpha_{3} \int \left(\alpha_{3}^{2} - \frac{\alpha_{2}^{2}}{\cos^{2}\theta}\right)^{-1/2} d\theta - \alpha_{3} \int \frac{1}{r} \left[2m\left(\alpha_{1} + \frac{\mu m}{r}\right) - \frac{\alpha_{3}^{2}}{r^{2}}\right]^{-1/2} dr$$

$$= \alpha_{3} I_{4} - \frac{\alpha_{3}}{m} I_{3}.$$

$$(30)$$

The evaluation of the integrals I_1 , I_2 , I_3 , I_4 is non-trivial and will be performed next.

For I_1 , substituting (19), (21) and $mh = -\mu m/2a$, we obtain

$$I_{1} = m \int \left[2m \left(mh + \frac{m\mu}{r} \right) - \frac{m^{2}}{r^{2}} (a\mu(1 - e^{2})) \right]^{-1/2} dr$$

$$= \int m dr \left[2m \left(-\frac{\mu m}{2a} + \frac{\mu m}{r} - \frac{m^{2}a\mu(1 - e^{2})}{r^{2}} \right) \right]^{-1/2}$$

$$= \frac{1}{\sqrt{\mu}} \int \frac{r dr}{\sqrt{-r^{2}/a + 2r - a(1 - e^{2})}}$$
(31)

Substitute $r = a(1 - e \cos E)$, which leads to $dr = ae \sin E dE$ and

$$I_{1} = \frac{1}{\sqrt{\mu}} \int \frac{a(1 - e\cos E) ae \sin E dE}{\sqrt{-a(1 - e\cos E)^{2} + 2a(1 - e\cos E) - a(1 - e^{2})}}$$

$$= \frac{a^{3/2}}{\sqrt{\mu}} \int (1 - e\cos E) dE$$

$$= \frac{a^{3/2}}{\sqrt{\mu}} (E - e\sin E)$$
(32)

Application of Kepler's equation and Kepler's third law $n = \sqrt{\mu/a^3}$, we obtain the additive inverse for the time of perihelion

$$q_1' = -t + \frac{M}{n} = -\tau. (33)$$

For I_2 , it follows from (20), (21) that $\alpha_2/\alpha_3 = \cos I$

$$\alpha_2 I_2 = \int \frac{\alpha_2}{\cos^2 \theta \sqrt{\alpha_3^2 - \frac{\alpha_2^2}{\cos^2 \theta}}} = \frac{\alpha_2}{\alpha_3} \int \frac{d\theta}{\cos^2 \theta \sqrt{1 - \frac{(\alpha_2/\alpha_3)^2}{\cos^2 \theta}}} = \int \frac{\cos I d\theta}{\cos^2 \theta \sqrt{1 - \frac{\cos^2 I}{\cos^2 \theta}}}$$
(34)

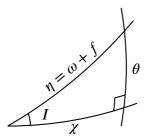


Figure 1: The relationship between the angles η , θ and χ .

Via application of Napier's rule to the angles in Figure 1, we obtain

$$\sin \chi = \cot(\pi/2 - \theta) \cot I = \tan \theta \cot I \tag{35}$$

and

$$\frac{d\theta}{\cos^2\theta} = \tan i \cos \chi d\chi. \tag{36}$$

Thus,

$$\alpha_2 I_2 = \int \frac{\cos I \tan I \cos \chi d\chi}{\sqrt{1 - \cos^2 I (1 + \tan^2 I \sin^2 \chi)}} = \int \frac{\sin I \cos \chi d\chi}{\sqrt{\sin^2 I (1 - \sin^2 \chi)}} = \int d\chi = \chi. \tag{37}$$

and

$$q_2' = \phi - \alpha_2 I_2 = \phi - \chi = \Omega. \tag{38}$$

For I_3 ,

$$I_{3} = m \int \frac{1}{r^{2}} \left[2m \left(-\frac{\mu m}{2a} + \frac{\mu m}{r} \right) - m^{2} \frac{a\mu (1 - e^{2})}{r^{2}} \right]^{-1/2} dr$$

$$= \frac{1}{\sqrt{\mu}} \int \frac{dr}{r\sqrt{-r^{2}/a + 2r - a(1 - e^{2})}}$$
(39)

Substitute $r = a(1 - e \cos E), dr = ae \sin E dE$, to obtain

$$I_{3} = \frac{1}{\sqrt{\mu}} \int \frac{ae \sin E dE}{a(1 - e \cos E)\sqrt{-a(1 - 2e \cos E + e^{2} \cos^{2} E) + 2a(1 - e \cos E) - a(1 - e^{2})}}$$

$$= \frac{1}{\sqrt{\mu}} \int \frac{e \sin E}{(1 - e \cos E)\sqrt{a(e^{2} \sin^{2} E)}}$$

$$= \frac{1}{\sqrt{a\mu}} \int \frac{\sin E dE}{\sin E(1 - e \cos E)}$$

$$= \frac{1}{\sqrt{a\mu(1 - e^{2})}} \int \frac{\sqrt{1 - e^{2}} dE}{1 - e \cos E}$$
(40)

Substituting $\sin f = \sqrt{1 - e^2} \sin E / (1 - e \cos E)$ leads to

$$\cos f df = \sqrt{1 - e^2} \frac{\cos f}{1 - e \cos E} dE \tag{41}$$

Dividing by $\cos f$ and substituting back, we obtain

$$I_3 = \frac{f}{\sqrt{a\mu(1-e^2)}}. (42)$$

For I_4 , substituting (20), (21)

$$I_4 = \int \left(\alpha_3^2 - \frac{\alpha_2^2}{\cos^2 \theta}\right)^{-1/2} = \frac{1}{m\sqrt{a\mu(1 - e^2)}} \int \left(1 - \frac{\cos^2 I}{\cos^2 \theta}\right)^{-1/2} d\theta \tag{43}$$

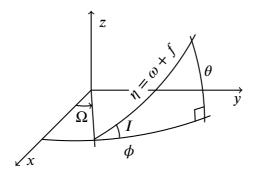


Figure 2: The relationship between the spherical polar coordinates and the Keplerian elements.

Multiplying with α_3 leads

$$\alpha_3 I_4 = \int \frac{\cos \theta d\theta}{\sqrt{\cos^2 \theta - \cos^2 I}} \tag{44}$$

From Figure 2, we obtain

$$\frac{\sin \theta}{\sin I} = \frac{\sin \eta}{\sin \pi/2} \tag{45}$$

which leads to

$$\cos^2 \theta = 1 - \sin^2 I \sin^2 \eta \tag{46}$$

$$\cos\theta d\theta = \sin i \cos \eta d\eta. \tag{47}$$

$$\alpha_3 I_4 = \int \frac{\sin I \cos \eta d\eta}{\sqrt{1 - \sin^2 I \sin^2 \eta - \cos^2 I}} = \int \frac{\sin I \cos \eta d\eta}{\sqrt{\sin^2 I \cos^2 \eta}} = \int d\eta = \eta. \tag{48}$$

Thus,

$$q_3' = \alpha_3 I_4 - \frac{\alpha_3}{m} I_3 = \eta - \frac{m\sqrt{a\mu(1-e^2)}}{m} \frac{f}{\sqrt{a\mu(1-e^2)}} = \eta - f = \omega.$$
 (49)

We now have the new canonical coordinates and momentums

$$q'_{1} = -\tau,$$
 (50)
 $q'_{2} = \Omega,$ (51)
 $q'_{3} = \omega,$ (52)
 $p'_{1} = mh,$ (53)

$$q_2' = \Omega, \tag{51}$$

$$q_3' = \omega, \tag{52}$$

$$p_1' = mh, (53)$$

$$p_2' = m\sqrt{a\mu(1-e^2)}\cos I,$$
 (54)

$$p_3' = m\sqrt{a\mu(1-e^2)}, (55)$$

as well as the new Hamiltonian

$$H' = 0. (56)$$

1.1.3 Delaunay Elements

2 Spherical Harmonics

2.1 Spheroid Potential

The gravitational potential of an arbitrary distribution of mass can be expanded in terms of spherical harmonics

$$V(\mathbf{r}) = -G \int \frac{\rho(\mathbf{r}')}{|\mathbf{r} - \mathbf{r}'|} d^3 \mathbf{r}^3$$

$$= -\frac{G}{r} \left(1 - \sum_{n=1}^{\infty} \left(\frac{R}{r} \right)^n J_n P_n(\cos \theta) - \sum_{n=1}^{\infty} \sum_{m=1}^{\infty} \left(\frac{R}{r} \right)^n P_n^m(\cos \theta) \left(c_{nm} \cos m\phi + s_{nm} \sin n\phi \right) \right)$$
(57)

where θ is the polar angle, G is the gravitational constant, and R is an arbitrary scaling factor. The Zonal and Tesseral harmonics can be written

$$J_n = -\frac{1}{MR^n} \int P_n(\cos\theta') r'^n \rho(\mathbf{r}') d^3 \mathbf{r}', \qquad (58)$$

$$c_{nm} = -2\frac{(n-m)!}{(n+m)!} \frac{1}{MR^n} \int P_n^m(\cos\theta') \cos m\phi' r'^n \rho(r') d^3r'$$
 (59)

$$s_{nm} = -2\frac{(n-m)!}{(n+m)!} \frac{1}{MR^n} \int P_n^m(\cos\theta') \sin m\phi' r'^n \rho(r') d^3r, \tag{60}$$

where P_n is the Legendre polynomial of degree n and P_n^m is the Associated Legendre Polynomial of degree n and order m. If the body is axially symmetric w.r.t. z axis, the tesseral harmonics c_{nm} and s_{nm} disappear. If the body is symmetric w.r.t. the equatorial z = 0 plane $P_n(-x) = (-1)^n P_n(x)$ and J_n will disappear for odd-valued n.

Since the body of interest is a spheroid, the tesseral harmonics disappear, and we have an expansion in terms of only zonal harmonics

$$V(r,\theta) = -\frac{\mu}{r} \left(1 - \sum_{n=2}^{\infty} \left(\frac{R}{r} \right)^n J_n P_n(\cos \theta) \right). \tag{61}$$

For SGP4, the zonal harmonics are obtained from WGS72[TODO] with the values

$$J_2 = 1082.616 \cdot 10^{-6}$$

 $J_3 = -2.53881 \cdot 10^{-6}$
 $J_4 = -1.65597 \cdot 10^{-6}$

The first four Legendre polynomials can be written

$$P_1(\cos\theta) = \cos\theta \tag{62}$$

$$P_2(\cos\theta) = \frac{1}{2} \left(3\cos^2\theta - 1 \right) \tag{63}$$

$$P_3(\cos\theta) = \frac{1}{2} \left(5\cos^3\theta - 3\cos\theta \right) \tag{64}$$

$$P_4(\cos\theta) = \frac{1}{8} \left(35\cos^4\theta - 30\cos^2\theta + 3 \right)$$
 (65)

$$P_5(\cos\theta) = \frac{1}{8} \left(63\cos^5\theta - 70\cos^3\theta + 15\cos\theta \right) \tag{66}$$

Substituting (63) - (65) back to (61), we obtain the potential

$$V(r,\theta) = -\frac{\mu}{r}$$

$$+ \frac{\mu J_2 R^2}{2r^3} \left(3\cos^2 \theta - 1 \right)$$

$$+ \frac{\mu J_3 R^3}{2r^4} \left(5\cos^3 \theta - 3\cos \theta \right)$$

$$+ \frac{\mu J_4 R^4}{8r^5} \left(35\cos^4 \theta - 30\cos^2 \theta + 3 \right)$$

$$+ \frac{\mu J_5 R^5}{8r^6} \left(63\cos^5 \theta - 70\cos^3 \theta + 15\cos \theta \right)$$
(67)

We use $R = a_E$ to denote the equatorial radius of Earth and define

$$k_2 = \frac{J_2 a_E^2}{2}, \quad k_4 = -\frac{3J_4 a_E^4}{8}, \quad A_{3,0} = -J_3 a_E^3, \quad A_{5,0} = -J_5 a_E^5.$$
 (68)

Thereafter, using β to denote the geocentric latitude, we obtain $\cos \theta = \sin \beta$ and

$$V(r,\theta) = -\frac{\mu}{r}$$

$$-\frac{\mu k_2}{r^3} \left(1 - 3\sin^2 \beta \right)$$

$$-\frac{\mu k_4}{r^5} \left(1 - 10\sin^2 \beta + \frac{35}{3}\sin^4 \beta \right)$$

$$-\frac{\mu A_{3,0}}{r^4} \left(-\frac{3}{2}\sin \beta + \frac{5}{2}\sin^3 \beta \right)$$

$$-\frac{\mu A_{5,0}}{r^5} \left(\frac{15}{8}\sin \beta - \frac{35}{4}\sin^3 \beta + \frac{63}{8}\sin^5 \beta \right).$$
(69)

Note that Brouwer calls the gravitational potential the force function and uses different sign [?]. That is, U = -V.

2.2 Disturbing Function

The first term in (69) corresponds to the unperturbed potential of a two-body problem, where the Keplerian elements are constants. The additional terms constitute a perturbing potential that leads to perturbations in the Keplerian elements. That is, instead of the Hamiltonian (??), we have

$$F = \frac{\mu}{2L^2} + R,\tag{70}$$

where $R = R_2 + R_3 + R_4 + R_5$ is the **disturbing function** and R_k denote the part of R corresponding to k:th degree in (69). In this section, we will derive R in terms of Delaunay variables.

In perifocal frame, where the non-disturbed orbit is on the z=0 plane and positive x axis corresponds to to the rising node, we can write the position vector as

$$r\left[\cos(g+f), \sin(g+f), 0\right]^{T}. \tag{71}$$

The position vector in an equatorial inertial frame with x axis corresponding to the rising node can be obtained via application of an anti-clockwise rotation w.r.t. x coordinate

$$\begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos I & -\sin I \\ 0 & \sin I & \cos I \end{bmatrix} \begin{bmatrix} \cos(g+f) \\ \sin(g+f) \\ 0 \end{bmatrix} = \begin{bmatrix} \cos(g+f) \\ \cos I \sin(g+f) \\ \sin I \sin(g+f) \end{bmatrix}$$
(72)

The z-coordinate can be related to the geocentric latitude β in (69) with the equation

$$\sin \beta = \sin I \sin(g+f). \tag{73}$$

Then, by using the relations

$$\sin^{2}(g+f) = (1-\cos(2g+2f))/2,$$

$$\sin(g+f)\cos(2g+2f) = \frac{1}{2}\sin(3g+3f) - \frac{1}{2}\sin(g+f),$$

$$\cos 2\theta = 2\cos^{2}\theta - 1,$$

$$\sin(g+f)\cos(4g+4f) = \frac{1}{2}\sin(5g+5f) - \frac{1}{2}\sin(3g+3f),$$

we obtain

$$\sin^{2}\beta = \frac{1}{2}\sin^{2}I\left(1 - \cos(2g + 2f)\right), \tag{74}$$

$$\sin^{3}\beta = \sin^{3}I\left(\frac{3}{4}\sin(g + f) - \frac{1}{4}\sin(3g + 3f)\right), \tag{74}$$

$$\sin^{4}\beta = \frac{1}{4}\sin^{4}I\left(1 - 2\cos(2g + 2f) + \cos^{2}(2g + 2f)\right)$$

$$= \frac{1}{8}\sin^{4}I\left(3 - 4\cos(2g + 2f) + \cos(4g + 4f)\right)$$

$$\sin^{5}\beta = \frac{1}{16}\left[6\sin(g + f) - 4\sin(3g + 3f) + 4\sin(g + f) + \sin(5g + 5f) - \sin(3g + 3f)\right]$$

$$= \sin^{5}I\left[\frac{5}{8}\sin(g + f) - \frac{5}{16}\sin(3g + 3f) + \frac{1}{16}\sin(5g + 5f)\right].$$

2.2.1 The J_2 Harmonic

We can expand

$$1 - 3\sin^2\beta = 1 - \frac{3}{2}\sin^2 I \left(1 - \cos(2g + 2f)\right) \tag{75}$$

$$= 1 - \frac{3}{2} (1 - \cos(2g + 2f)) + \frac{3}{2} \cos^2 I (1 - \cos(2g + 2f))$$
 (76)

$$= \left(-\frac{1}{2} + \frac{3}{2}\cos^2 I\right) + \left(\frac{3}{2} - \frac{3}{2}\cos^2 I\right)\cos(2g + 2f) \tag{77}$$

to obtain the disturbing function R_2 corresponding to the J_2 harmonic

$$R_{2} := \frac{\mu k_{2}}{r^{3}} \left(1 - 3\sin^{2}\beta \right)$$

$$= \frac{\mu k_{2}}{a^{3}} \left[\left(-\frac{1}{2} + \frac{3}{2}\cos^{2}I \right) \frac{a^{3}}{r^{3}} + \left(\frac{3}{2} - \frac{3}{2}\cos^{2}I \right) \cos(2g + 2f) \frac{a^{3}}{r^{3}} \right]$$

$$= \frac{\mu k_{2}}{a^{3}} \left[\left(-\frac{1}{2} + \frac{3}{2} \frac{H^{2}}{G^{2}} \right) \frac{a^{3}}{r^{3}} + \left(\frac{3}{2} - \frac{3}{2} \frac{H^{2}}{G^{2}} \right) \cos(2g + 2f) \frac{a^{3}}{r^{3}} \right].$$
(78)

2.2.2 The J_4 Harmonic

We can expand

$$1 - 10\sin^{2}\beta + \frac{35}{3}\sin^{4}\beta$$

$$= 1 - 5\sin^{2}I(1 - \cos(2g + 2f)) + \frac{35}{24}\sin^{4}I(3 - 4\cos(2g + 2f) + \cos(4g + 4f))$$

$$= \left[1 - 5(1 - \cos^{2}I) + \frac{35}{8}(1 - \cos^{2}I)^{2}\right] + \left[5(1 - \cos^{2}I) - \frac{35}{6}(1 - \cos^{2}I)^{2}\right]\cos(2g + 2f)$$

$$+ \frac{35}{24}(1 - \cos^{2}I)^{2}\cos(4g + 4f)$$

$$= \left(\frac{3}{8} - \frac{15}{4}\cos^{2}I + \frac{35}{8}\cos^{4}I\right) + \left(-\frac{5}{6} + \frac{20}{3}\cos^{2}I - \frac{35}{6}\cos^{4}I\right)\cos(2g + 2f)$$

$$+ \left(\frac{35}{24} - \frac{35}{12}\cos^{2}I + \frac{35}{24}\cos^{4}I\right)\cos(4g + 4f).$$

$$(79)$$

to obtain the disturbing function R_4 corresponding to the J_4 harmonic

$$R_{4} := \frac{\mu k_{4}}{r^{5}} \left[\left(\frac{3}{8} - \frac{15}{4} \cos^{2} I + \frac{35}{8} \cos^{4} I \right) + \left(-\frac{5}{6} + \frac{20}{3} \cos^{2} I - \frac{35}{6} \cos^{4} I \right) \cos(2g + 2f) \right]$$

$$+ \left(\frac{35}{24} - \frac{35}{12} \cos^{2} I + \frac{35}{24} \cos^{4} I \right) \cos(4g + 4f)$$

$$= \frac{\mu^{6} k_{4}}{L^{10}} \left[\left(\frac{3}{8} - \frac{15}{4} \frac{H^{2}}{G^{2}} + \frac{35}{8} \frac{H^{4}}{G^{4}} \right) \frac{a^{5}}{r^{5}} \right]$$

$$+ \left(-\frac{5}{6} + \frac{20}{3} \frac{H^{2}}{G^{2}} - \frac{35}{6} \frac{H^{4}}{G^{4}} \right) \frac{a^{5}}{r^{5}} \cos(2g + 2f)$$

$$+ \left(\frac{35}{24} - \frac{35}{12} \frac{H^{2}}{G^{2}} + \frac{35}{24} \frac{H^{4}}{G^{4}} \right) \frac{a^{5}}{r^{5}} \cos(4g + 4f)$$

$$+ \left(\frac{35}{24} - \frac{35}{12} \frac{H^{2}}{G^{2}} + \frac{35}{24} \frac{H^{4}}{G^{4}} \right) \frac{a^{5}}{r^{5}} \cos(4g + 4f)$$

2.2.3 The J_3 Harmonic

We can expand

$$-\frac{3}{2}\sin\beta + \frac{5}{2}\sin^3\beta = -\frac{3}{2}\sin I\sin(g+f) + \frac{5}{2}\sin^3I\left(\frac{3}{4}\sin(g+f) - \frac{1}{4}\sin(3g+3f)\right)$$
(81)
$$= \left(-\frac{3}{2}\sin I + \frac{15}{8}\sin^3I\right)\sin(g+f) - \frac{5}{8}\sin^3I\sin(3g+3f).$$

to obtain the disturbing function R_3 corresponding to the J_3 harmonic

$$R_3 := \frac{\mu A_{3,0}}{r^4} \left[\left(-\frac{3}{2} \sin I + \frac{15}{8} \sin^3 I \right) \sin(g+f) - \frac{5}{8} \sin^3 I \sin(3g+3f) \right]$$

$$= \frac{\mu A_{3,0}}{r^4} \sin I \left[\left(-\frac{3}{2} + \frac{15}{8} \frac{H^2}{G^2} \right) \sin(g+f) - \frac{5}{8} \frac{H^2}{G^2} \sin(3g+3f) \right].$$
(82)

2.2.4 The J_5 Harmonic

We can expand

$$\frac{15}{8}\sin\beta = \sin I \left[\frac{15}{8}\sin(g+f) \right]
-\frac{35}{4}\sin^{3}\beta = \sin^{3}I \left[-\frac{105}{16}\sin(g+f) + \frac{35}{16}\sin(3g+3f) \right]
\frac{63}{8}\sin^{5}\beta = \sin^{5}I \left[\frac{315}{64}\sin(g+f) - \frac{315}{128}\sin(3g+3f) + \frac{63}{128}\sin(5g+5f) \right]$$
(83)

to obtain the disturbing function R_5 corresponding to the J_5 harmonic

$$R_{5} := \frac{\mu A_{5,0}}{r^{6}} \left[\left(\frac{15}{8} \sin I - \frac{105}{16} \sin^{3} I + \frac{315}{64} \sin^{5} I \right) \sin(g+f) + \left(\frac{35}{15} \sin^{3} I - \frac{315}{128} \sin^{5} I \right) \sin(3g+3f) + \left(\frac{63}{128} \sin^{5} I \right) \sin(5g+5f) \right]$$
(84)

2.3 Hamiltonian Formulation

3 Solution of the Problem

- 3.1 Von Zeipel Method and Canonical Transformations
- **3.2** Explicit Solution for J_2 Harmonic
- 3.2.1 Short-Period Terms
- 3.2.2 Long-Period Terms
- 3.2.3 Secular Terms
- 3.3 Elements and Orbit State Vectors

4 SGP4 Solution

- 4.1 Approximation
- 4.2 Coordinate Systems and Frames
- 4.2.1 Mean-of-Date (MoD)
- 4.2.2 True-of-Date (ToD)
- 4.2.3 True-Equator, Mean-Equinox (TEME)
- 4.2.4 Pseudo Earth-Fixed (PEF)
- 4.3 Atmospheric Drag
- **4.4** Two-Line Elements
- 4.5 Kozai Mean Elements

References