
Application of Kane's Method to a General Tree-Topology Spacecraft with Gimbaled or Spherical Joints

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Purpose

We seek equations of motion suitable for solution by a computer, using the numerical integrator of your choice. We will obtain a set of coupled equations which are linear in the derivatives of the dynamic state variables:

$$(\Omega^T I \Omega + V^T m V) \dot{u} = \Omega^T (T - \omega \times H - I \alpha_r) + V^T (F - m a_r)$$

This package presents notation, derivations, and an example to help fix the underlying concepts and guide the assembly of these equations for a general tree-topology N-body spacecraft model.

" $F = ma$. The rest is just accounting." -Unknown

Kane's Equation

A set of scalar equations, one per generalized speed (u_r).

$$F_r + F_r^* = 0 \quad r = 1, \dots, N_u$$

where

$$F_r = \sum_k \vec{\omega}_r^k \cdot \vec{T}_k + \sum_k \vec{v}_r^k \cdot \vec{F}_k \quad \text{(Generalized active force)}$$

$$F_r^* = \sum_k \vec{\omega}_r^k \cdot (-I_k \vec{\alpha}_k - \vec{\omega}^k \times \vec{H}_k) + \sum_k \vec{v}_r^k \cdot (-m_k \vec{a}_k) \quad \text{(Generalized inertia force)}$$

Rewriting,

$$\sum_k \left[\vec{\omega}_r^k \cdot (\vec{T}_k - I_k \vec{\alpha}_k - \vec{\omega}^k \times \vec{H}_k) \right] + \sum_k \left[\vec{v}_r^k \cdot (\vec{F}_k - m_k \vec{a}_k) \right] = 0$$

$\vec{\omega}_r^k$ is called the r th partial angular velocity of the k th body. It is a vector that projects Euler's equation into the state-space dimension of u_r .

\vec{v}_r^k is called the r th partial velocity of the k th body. It is a vector that projects Newton's equation into the state-space dimension of u_r .

How do we find $\vec{\omega}_r$ and \vec{v}_r , and how do we assemble a system of equations?

Vectors and Components

- A vector is independent of any reference frame.
- For numerical computation, we need a numerical representation.
- So we often represent a vector as a product of some scalar *components* and a set of *basis vectors*

$$\vec{v} = v_1 \hat{a}_1 + v_2 \hat{a}_2 + v_3 \hat{a}_3$$

$$\vec{v} = \begin{bmatrix} v_1 & v_2 & v_3 \end{bmatrix} \begin{bmatrix} \hat{a}_1 \\ \hat{a}_2 \\ \hat{a}_3 \end{bmatrix}$$

$$\vec{v} = \begin{bmatrix} \hat{a}_1 & \hat{a}_2 & \hat{a}_3 \end{bmatrix} \begin{bmatrix} v_1 \\ v_2 \\ v_3 \end{bmatrix}$$

$$\begin{aligned} \vec{v} &= \mathbf{A} \mathbf{v}, \\ \mathbf{v} &= \mathbf{A}^T \vec{v} \end{aligned}$$

Basis Dyads

- \vec{v} is a *vector*
- v is an *array of components*
- $\hat{a}_1, \hat{a}_2, \hat{a}_3$ are (orthonormal) *basis vectors*
- Hughes calls \mathbf{A} a *vectrix*. I'll use the term *basis dyad*.
- Like a vector, a dyad is independent of any reference frame, but unsuited to computerized manipulation. So we'll eventually seek a matrix representation.

Reference Frames

- A *reference frame* has an origin and three orthonormal basis vectors.
- We will make use of N , a Newtonian (i.e. non-accelerating, or inertial) frame, and several body-fixed frames, $B_1, B_2, \dots B_{Nb}$.
- Note that the direction cosine matrix relating two frames is the product of the frames' basis dyads:
- $${}^B C^A = \mathbf{B}^T \mathbf{A}$$
- And the components of a vector, expressed in two different reference frames, are related by:

$$\vec{v} = \mathbf{A}^A \mathbf{v} = \mathbf{B}^B \mathbf{v} \quad \rightarrow \quad {}^B \mathbf{v} = {}^B C^A \mathbf{v}$$

Angular and Linear Velocity

- Angular velocity of B in N :

$${}^N\vec{\omega}^B = \omega_1 \hat{b}_1 + \omega_2 \hat{b}_2 + \omega_3 \hat{b}_3 = \mathbf{B} \omega_B$$

- Linear velocity of B 's mass center in N :

$${}^N\vec{v}^B = v_1 \hat{n}_1 + v_2 \hat{n}_2 + v_3 \hat{n}_3 = \mathbf{N} v_B$$

Joint Partial

Consider two bodies connected by a gimballed joint. The relative angular velocity of the outer body, B , with respect to the inner body, A , may be written as a function of the gimbal angles, θ , and the gimbal rates, $\sigma \equiv \dot{\theta}$. For example, a Body-3, 2-1-3 Euler rotation through $(\theta_1, \theta_2, \theta_3)$ gives:

$${}^A\vec{\omega}^B = \mathbf{B} \underbrace{\begin{bmatrix} c_2 s_3 & c_3 & 0 \\ c_2 c_3 & -s_3 & 0 \\ -s_2 & 0 & 1 \end{bmatrix}}_I \underbrace{\begin{pmatrix} \sigma_1 \\ \sigma_2 \\ \sigma_3 \end{pmatrix}}_{\sigma}$$

Where $s_2 = \sin(\theta_2)$, $c_2 = \cos(\theta_2)$, etc, and the basis dyad \mathbf{B} appears because $\vec{\omega}$ is a vector, and I and σ are a matrix and an array.

The matrix I is called the *matrix of joint partials*. Each Euler sequence, including one- and two-DOF gimbals, has its own. The dimensions of I are 3 x NDOF.

For a spherical joint, $\sigma = \omega$, and $I = U$, the identity matrix.

From 3-space to State Space

- Consider a system of N_b bodies, connected in a tree topology by $(N_g = N_b - 1)$ revolute joints (gimballed or spherical)
- We choose a set of independent generalized speeds:
 - Components of root body's angular velocity
 - Joint angular rates (1-, 2-, or 3-DOF gimbal) or joint angular velocity components (spherical)
 - Components of root body's linear velocity
- We write the generalized speeds and generalized coordinates as

$$u \equiv \begin{pmatrix} \omega_1 \\ \sigma_1 \\ \vdots \\ \sigma_{N_g} \\ v_1 \end{pmatrix} \quad x \equiv \begin{pmatrix} q_1 \\ \theta_1 \\ \vdots \\ \theta_{N_g} \\ p_1 \end{pmatrix}$$

From State Space to 3-space

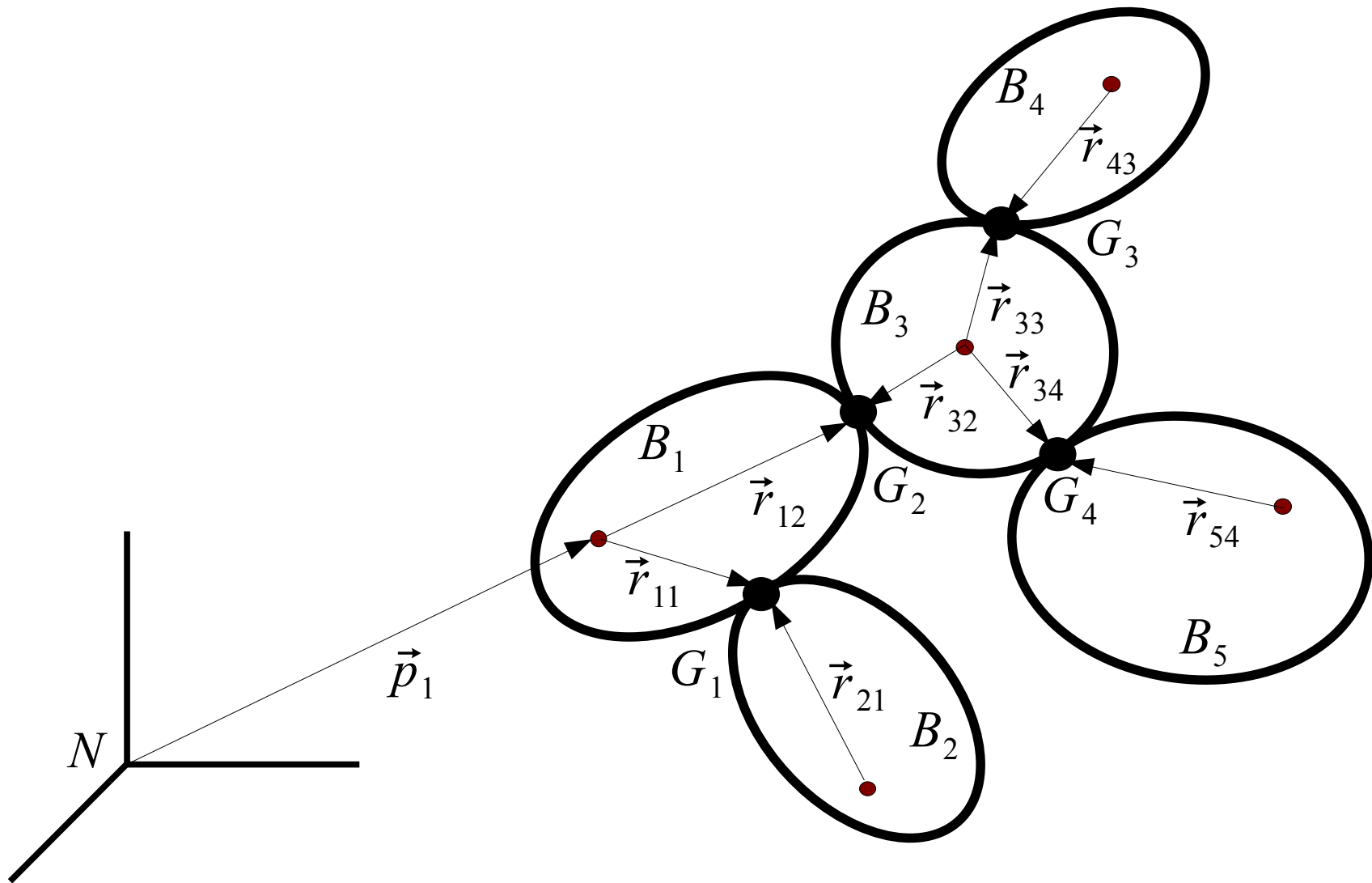
- For Kane's equations, we need $\vec{\alpha}$ and \vec{a} for each rigid body, which we obtain by differentiating $\vec{\omega}$ and \vec{v} of each rigid body.
- Using joint partials, we may construct expressions for these as functions of the generalized speeds.
- We'll look at the structure of these functions by example next.
- For now, simply note that we may write

$$\begin{pmatrix} \vec{\omega}_1 \\ \vec{\omega}_2 \\ \vdots \\ \vec{\omega}_{Nb} \end{pmatrix} = \mathbf{\Omega} u \qquad \begin{pmatrix} \vec{v}_1 \\ \vec{v}_2 \\ \vdots \\ \vec{v}_{Nb} \end{pmatrix} = \mathbf{V} u$$

Deja Vu All Over Again

- On the preceding slide, we defined Ω and V to compactly express the angular velocities and velocities of a collection of rigid bodies as linear combinations of a set of independent generalized speeds
- Looking at it another way, the elements of Ω and V project $\vec{\omega}$ and \vec{v} into the state-space dimension of each element of u .
- Ω contains the partial angular velocities. V contains the partial velocities.
- In the following slides, we will see how to construct Ω and V for a general tree-topology N-body spacecraft.

Example Problem



Constructions

Angular velocities are constructed like this:

$${}^N\vec{\omega}^2 = {}^N\vec{\omega}^1 + {}^1\vec{\omega}^2 = \mathbf{B}_1\omega_1 + \mathbf{B}_2\Gamma_1\sigma_1$$

Velocities are constructed like this:

$${}^N\vec{v}^2 = {}^N\vec{v}^1 + {}^N\vec{\omega}^1 \times \vec{r}_{11} - {}^N\vec{\omega}^2 \times \vec{r}_{21} = N v_1 + (\mathbf{B}_1\omega_1) \times (\vec{r}_{11} - \vec{r}_{21}) - (\mathbf{B}_2\Gamma_1\sigma_1) \times \vec{r}_{21}$$

Let $\vec{\beta}_i$ be the vector from the mass center of B_i to the mass center of B_l .

For example:

$$\vec{\beta}_4 = \vec{r}_{43} - \vec{r}_{33} + \vec{r}_{32} - \vec{r}_{12}$$

Let $\vec{\rho}_{ij}$ be the vector from the mass center of B_i to joint G_j . For example:

$$\vec{\rho}_{42} = \vec{r}_{43} - \vec{r}_{33} + \vec{r}_{32}$$

Path Tables

Is this body
in the path of
that body?

	B1	B2	B3	B4	B5
B1	x				
B2	x	x			
B3	x		x		
B4	x		x	x	
B5	x		x		x

Body Path Table

Is this joint
in the path of
that body?

	(B1)	G1	G2	G3	G4
B1	x				
B2	x	x			
B3	x		x		
B4	x		x	x	
B5	x		x		x

Joint Path Table

Warning: The Path Tables look the same for this example, but they differ in general. (To see this, swap the labels for G3 and G4.)

B1 is not a joint, but its mass center does serve as one, so it is included in the joint path table.

“I do vector calculus just for fun.”

-Weird Al Yankovic,
“White and Nerdy”

Useful Notation and Theorems

Cross product matrix: $\tilde{x} = \begin{bmatrix} 0 & -x_3 & x_2 \\ x_3 & 0 & -x_1 \\ -x_2 & x_1 & 0 \end{bmatrix}, \quad x \times y = \tilde{x} y$

Double-cross: $\bar{x} y = x \times (x \times y)$

General differentiation of a vector: ${}^A \frac{d}{dt} \vec{v} = {}^B \frac{d}{dt} \vec{v} + {}^A \vec{\omega}^B \times \vec{v}$

Apply to find angular acceleration, with a joint:

$${}^N \vec{\alpha}^B = {}^A \dot{\omega}_A + {}^B (\Gamma \dot{\sigma} + \dot{\Gamma} \sigma) + {}^B (\omega_B \times \Gamma \sigma)$$

Linear acceleration, with a joint:

$${}^N \vec{a}^B = {}^N \vec{a}^A + {}^N \vec{\omega}^A \times ({}^N \vec{\omega}^A \times \vec{r}_A) + {}^N \vec{\alpha}^A \times \vec{r}_A - {}^N \vec{\omega}^B \times ({}^N \vec{\omega}^B \times \vec{r}_B) - {}^N \vec{\alpha}^B \times \vec{r}_B$$

Angular and Linear Velocities

$$\begin{pmatrix} \vec{\omega}_1 \\ \vec{\omega}_2 \\ \vec{\omega}_3 \\ \vec{\omega}_4 \\ \vec{\omega}_5 \end{pmatrix} = \underbrace{\begin{bmatrix} \mathbf{B}_1 & 0 & 0 & 0 & 0 & 0 \\ \mathbf{B}_1 & \mathbf{B}_2 \Gamma_1 & 0 & 0 & 0 & 0 \\ \mathbf{B}_1 & 0 & \mathbf{B}_3 \Gamma_2 & 0 & 0 & 0 \\ \mathbf{B}_1 & 0 & \mathbf{B}_3 \Gamma_2 & \mathbf{B}_4 \Gamma_3 & 0 & 0 \\ \mathbf{B}_1 & 0 & \mathbf{B}_3 \Gamma_2 & 0 & \mathbf{B}_5 \Gamma_4 & 0 \end{bmatrix}}_{\Omega} \underbrace{\begin{pmatrix} \omega_1 \\ \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ v_1 \end{pmatrix}}_u$$

$$\begin{pmatrix} \vec{v}_1 \\ \vec{v}_2 \\ \vec{v}_3 \\ \vec{v}_4 \\ \vec{v}_5 \end{pmatrix} = \underbrace{\begin{bmatrix} 0 & 0 & 0 & 0 & 0 & N \\ \tilde{\beta}_2 \mathbf{B}_1 & \tilde{\rho}_{21} \mathbf{B}_2 \Gamma_1 & 0 & 0 & 0 & N \\ \tilde{\beta}_3 \mathbf{B}_1 & 0 & \tilde{\rho}_{32} \mathbf{B}_3 \Gamma_2 & 0 & 0 & N \\ \tilde{\beta}_4 \mathbf{B}_1 & 0 & \tilde{\rho}_{42} \mathbf{B}_3 \Gamma_2 & \tilde{\rho}_{43} \mathbf{B}_4 \Gamma_3 & 0 & N \\ \tilde{\beta}_5 \mathbf{B}_1 & 0 & \tilde{\rho}_{52} \mathbf{B}_3 \Gamma_2 & 0 & \tilde{\rho}_{54} \mathbf{B}_5 \Gamma_4 & N \end{bmatrix}}_V \underbrace{\begin{pmatrix} \omega_1 \\ \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ v_1 \end{pmatrix}}_u$$

Note resemblance to joint path table. Ω and V may be obtained by inspection.

Clearing Basis Dyads

- Let's “convert” vectors and dyads into arrays and matrices.
- Choose which frames to express equations in, and multiply through by the appropriate basis dyad.

Clearing Basis Dyads (2)

$$\begin{bmatrix} \mathbf{B}_1^T & 0 & 0 & 0 & 0 \\ 0 & \mathbf{B}_2^T & 0 & 0 & 0 \\ 0 & 0 & \mathbf{B}_3^T & 0 & 0 \\ 0 & 0 & 0 & \mathbf{B}_4^T & 0 \\ 0 & 0 & 0 & 0 & \mathbf{B}_5^T \end{bmatrix} \begin{pmatrix} \vec{\omega}_1 \\ \vec{\omega}_2 \\ \vec{\omega}_3 \\ \vec{\omega}_4 \\ \vec{\omega}_5 \end{pmatrix} = \begin{bmatrix} \mathbf{B}_1^T & 0 & 0 & 0 & 0 \\ 0 & \mathbf{B}_2^T & 0 & 0 & 0 \\ 0 & 0 & \mathbf{B}_3^T & 0 & 0 \\ 0 & 0 & 0 & \mathbf{B}_4^T & 0 \\ 0 & 0 & 0 & 0 & \mathbf{B}_5^T \end{bmatrix} \underbrace{\begin{bmatrix} \mathbf{B}_1 & 0 & 0 & 0 & 0 & 0 \\ \mathbf{B}_1 & \mathbf{B}_2 \Gamma_1 & 0 & 0 & 0 & 0 \\ \mathbf{B}_1 & 0 & \mathbf{B}_3 \Gamma_2 & 0 & 0 & 0 \\ \mathbf{B}_1 & 0 & \mathbf{B}_3 \Gamma_2 & \mathbf{B}_4 \Gamma_3 & 0 & 0 \\ \mathbf{B}_1 & 0 & \mathbf{B}_3 \Gamma_2 & 0 & \mathbf{B}_5 \Gamma_4 & 0 \end{bmatrix}}_{\Omega} \underbrace{\begin{pmatrix} \omega_1 \\ \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ \nu_1 \end{pmatrix}}_u$$

$$\begin{pmatrix} \omega_1 \\ \omega_2 \\ \omega_3 \\ \omega_4 \\ \omega_5 \end{pmatrix} = \begin{bmatrix} U & 0 & 0 & 0 & 0 & 0 \\ {}^2C^1 & \Gamma_1 & 0 & 0 & 0 & 0 \\ {}^3C^1 & 0 & \Gamma_2 & 0 & 0 & 0 \\ {}^4C^1 & 0 & {}^4C^3 \Gamma_2 & \Gamma_3 & 0 & 0 \\ {}^5C^1 & 0 & {}^5C^3 \Gamma_2 & 0 & \Gamma_4 & 0 \end{bmatrix} \underbrace{\begin{pmatrix} \omega_1 \\ \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ \nu_1 \end{pmatrix}}_u$$

Clearing Basis Dyads (3)

$$\begin{bmatrix} N^T & 0 & 0 & 0 & 0 \\ 0 & N^T & 0 & 0 & 0 \\ 0 & 0 & N^T & 0 & 0 \\ 0 & 0 & 0 & N^T & 0 \\ 0 & 0 & 0 & 0 & N^T \end{bmatrix} \begin{pmatrix} \vec{v}_1 \\ \vec{v}_2 \\ \vec{v}_3 \\ \vec{v}_4 \\ \vec{v}_5 \end{pmatrix} = \begin{bmatrix} N^T & 0 & 0 & 0 & 0 \\ 0 & N^T & 0 & 0 & 0 \\ 0 & 0 & N^T & 0 & 0 \\ 0 & 0 & 0 & N^T & 0 \\ 0 & 0 & 0 & 0 & N^T \end{bmatrix} \underbrace{\begin{bmatrix} 0 & 0 & 0 & 0 & 0 & N \\ \tilde{\beta}_2 \mathbf{B}_1 & \tilde{\rho}_{21} \mathbf{B}_2 \Gamma_1 & 0 & 0 & 0 & N \\ \tilde{\beta}_3 \mathbf{B}_1 & 0 & \tilde{\rho}_{32} \mathbf{B}_3 \Gamma_2 & 0 & 0 & N \\ \tilde{\beta}_4 \mathbf{B}_1 & 0 & \tilde{\rho}_{42} \mathbf{B}_3 \Gamma_2 & \tilde{\rho}_{43} \mathbf{B}_4 \Gamma_3 & 0 & N \\ \tilde{\beta}_5 \mathbf{B}_1 & 0 & \tilde{\rho}_{52} \mathbf{B}_3 \Gamma_2 & 0 & \tilde{\rho}_{54} \mathbf{B}_5 \Gamma_4 & N \end{bmatrix}}_V \underbrace{\begin{pmatrix} \omega_1 \\ \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ v_1 \end{pmatrix}}_u$$

$$\begin{pmatrix} v_1 \\ v_2 \\ v_3 \\ v_4 \\ v_5 \end{pmatrix} = \underbrace{\begin{bmatrix} 0 & 0 & 0 & 0 & 0 & U \\ \tilde{\beta}_2 {}^N C^1 & \tilde{\rho}_{21} {}^N C^2 \Gamma_1 & 0 & 0 & 0 & U \\ \tilde{\beta}_3 {}^N C^1 & 0 & \tilde{\rho}_{32} {}^N C^3 \Gamma_2 & 0 & 0 & U \\ \tilde{\beta}_4 {}^N C^1 & 0 & \tilde{\rho}_{42} {}^N C^3 \Gamma_2 & \tilde{\rho}_{43} {}^N C^4 \Gamma_3 & 0 & U \\ \tilde{\beta}_5 {}^N C^1 & 0 & \tilde{\rho}_{52} {}^N C^3 \Gamma_2 & 0 & \tilde{\rho}_{54} {}^N C^5 \Gamma_4 & U \end{bmatrix}}_V \underbrace{\begin{pmatrix} \omega_1 \\ \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ v_1 \end{pmatrix}}_u$$

Accelerations across a Joint

Consider the k th joint, its inner body, B_i , and outer body, B_o . The angular acceleration of B_o is related to the angular acceleration of B_i and the motion of the joint:

$${}^N\vec{\alpha}^o = {}^N\vec{\alpha}^i + \mathbf{B}_o(\Gamma_k \dot{\sigma}_k + \dot{\Gamma}_k \sigma_k) + \mathbf{B}_o(\omega_o \times \Gamma_k \sigma_k)$$

The acceleration of B_o 's mass center is:

$${}^N\vec{a}^o = {}^N\vec{a}^i + {}^N\vec{\omega}^i \times ({}^N\vec{\omega}^i \times \vec{r}_{ik}) + {}^N\vec{\alpha}^i \times \vec{r}_{ik} - {}^N\vec{\omega}^o \times ({}^N\vec{\omega}^o \times \vec{r}_{ok}) - {}^N\vec{\alpha}^o \times \vec{r}_{ok}$$

As we construct these expressions, we find that some terms contain \dot{u} , and some don't.

We group the latter together, and coin the term 'remainder accelerations', so that we may write

$$\begin{aligned} [\alpha] &= \Omega \dot{u} + [\alpha_r] \\ [a] &= V \dot{u} + [a_r] \end{aligned}$$

The remainder accelerations may be constructed recursively in the same way as α and a .

$$\begin{aligned} {}^N\vec{\alpha}_r^o &= {}^N\vec{\alpha}_r^i + \mathbf{B}_o(\dot{\Gamma}_k \sigma_k + \omega_o \times \Gamma_k \sigma_k) \\ {}^N\vec{a}_r^o &= {}^N\vec{a}_r^i + {}^N\vec{\omega}^i \times ({}^N\vec{\omega}^i \times \vec{r}_{ik}) + {}^N\vec{\alpha}_r^i \times \vec{r}_{ik} - {}^N\vec{\omega}^o \times ({}^N\vec{\omega}^o \times \vec{r}_{ok}) - {}^N\vec{\alpha}_r^o \times \vec{r}_{ok} \end{aligned}$$

Assembling Kane's Equation

Let: $m = \begin{bmatrix} m_1 U & & & & \\ & m_2 U & & & \\ & & m_3 U & & \\ & & & m_4 U & \\ & & & & m_5 U \end{bmatrix}, \quad I = \begin{bmatrix} I_1 & & & & \\ & I_2 & & & \\ & & I_3 & & \\ & & & I_4 & \\ & & & & I_5 \end{bmatrix}, \quad T = \begin{Bmatrix} T_1 \\ T_2 \\ T_3 \\ T_4 \\ T_5 \end{Bmatrix}, \quad F = \begin{Bmatrix} F_1 \\ F_2 \\ F_3 \\ F_4 \\ F_5 \end{Bmatrix}, \quad \omega \times H = \begin{Bmatrix} (\omega \times H)_1 \\ (\omega \times H)_2 \\ (\omega \times H)_3 \\ (\omega \times H)_4 \\ (\omega \times H)_5 \end{Bmatrix}$

Substituting into Kane's equation:

$$\Omega^T [T - I(\Omega \dot{u} + \alpha_r) - \omega \times H] + V^T [F - m(V \dot{u} + a_r)] = 0$$

And rearranging:

$$\underbrace{(\Omega^T I \Omega + V^T m V)}_{COEF(N_u \times N_u)} \dot{u} = \underbrace{\Omega^T (T - I \alpha_r - \omega \times H) + V^T (F - m a_r)}_{RHS(N_u \times 1)}$$

Conclusions

- Kane's equation has been rearranged to a form suitable for numerical solution.
 - A system of equations linear in \dot{u}
- For a tree-topological spacecraft, partial angular velocity and partial velocity matrices may be deduced by inspection of the path tables.
- The hard part of Kane's method is differentiating vectors
 - Vector theorems help.
 - Notation that handles multiple frames helps, too.