
Modeling Multi-Body Spacecraft Using Momentum State Dynamics (MSD)

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Objectives

- To formulate momentum state dynamics for Freespace, adding these capabilities
 - Tree topology
 - Flexible modes
 - 2-DOF and 3-DOF gimbal joints
 - Slider joints
 - Lockable joints
- By request, we seek to preserve the decoupling between the differential equations of motion

$F = \dot{p}$ The rest is just accounting.

Rigid Bodies

Linear and Angular Momentum

Linear Momentum: ${}^N P^{B^*} = m {}^N v^{B^*} = m {}^N v^R + {}^N \omega^B \times c$

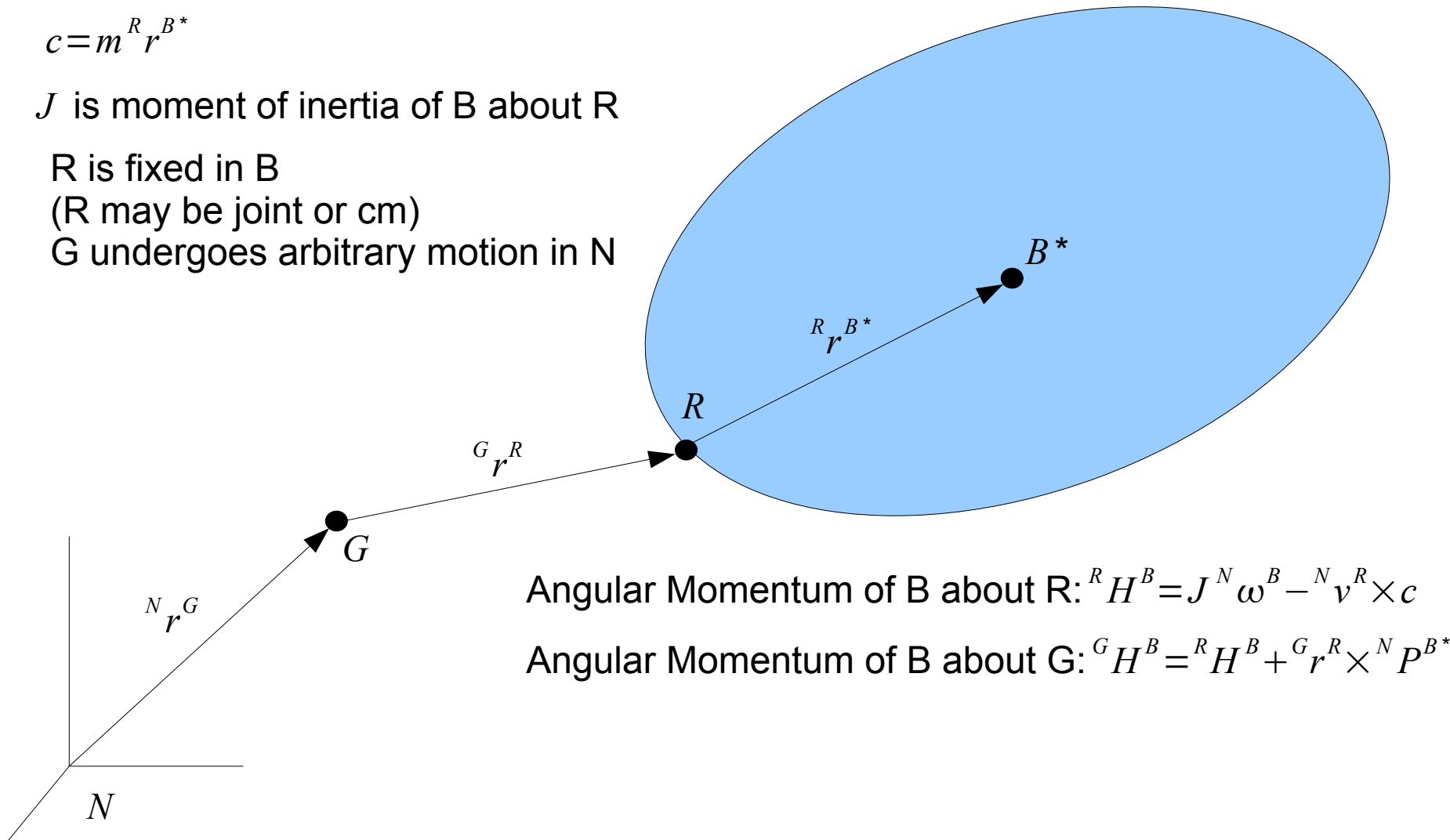
$$c = m {}^R r^{B^*}$$

J is moment of inertia of B about R

R is fixed in B

(R may be joint or cm)

G undergoes arbitrary motion in N



Angular Momentum of B about R: ${}^R H^B = J {}^N \omega^B - {}^N v^R \times c$

Angular Momentum of B about G: ${}^G H^B = {}^R H^B + {}^G r^R \times {}^N P^{B^*}$

MSD Equations of Motion

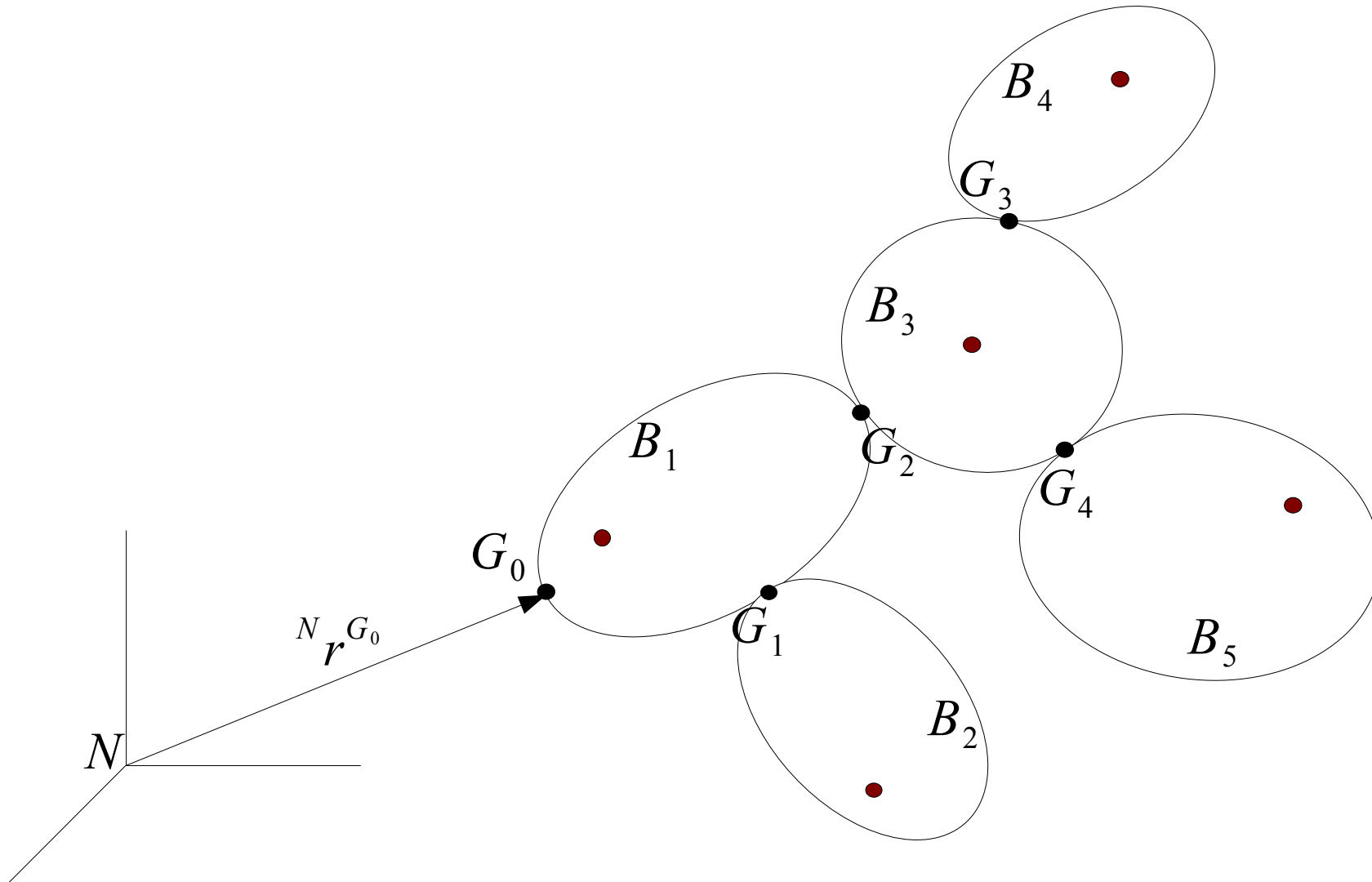
$${}^N \dot{P}^{B*} \equiv \frac{{}^N d {}^N P^{B*}}{dt} = F$$

$${}^G \dot{H}^B \equiv \frac{{}^B d {}^G H^B}{dt} = T_G - {}^N v^G \times {}^N P^{B*} - {}^N \omega^B \times {}^G H^B$$

Since \dot{H} and \dot{P} appear alone on the lefthand side, the EOM may be integrated without solving a system of equations for them.

However, since H and P depend explicitly on ω and v , a system of equations must still be solved at each evaluation of the EOM.

Example Problem



Composite Momenta

Let P_k be the linear momentum of the assembly of bodies outboard of the joint G_k

$$\begin{Bmatrix} P_k \end{Bmatrix} = \begin{Bmatrix} P_0 \\ P_1 \\ P_2 \\ P_3 \\ P_4 \end{Bmatrix} = \begin{Bmatrix} {}^N P^{B_1*} + {}^N P^{B_2*} + {}^N P^{B_3*} + {}^N P^{B_4*} + {}^N P^{B_5*} \\ {}^N P^{B_2*} \\ {}^N P^{B_3*} + {}^N P^{B_4*} + {}^N P^{B_5*} \\ {}^N P^{B_4*} \\ {}^N P^{B_5*} \end{Bmatrix}$$

Let H_k be the angular momentum of all bodies outboard of joint G_k , about G_k .

$$\begin{Bmatrix} H_k \end{Bmatrix} = \begin{Bmatrix} H_0 \\ H_1 \\ H_2 \\ H_3 \\ H_4 \end{Bmatrix} = \begin{Bmatrix} {}^{G_0} H^{B_1} + {}^{G_0} H^{B_2} + {}^{G_0} H^{B_3} + {}^{G_0} H^{B_4} + {}^{G_0} H^{B_5} \\ {}^{G_1} H^{B_2} \\ {}^{G_2} H^{B_3} + {}^{G_2} H^{B_4} + {}^{G_2} H^{B_5} \\ {}^{G_3} H^{B_4} \\ {}^{G_4} H^{B_5} \end{Bmatrix}$$

$[\{H_k\}, P_0]$ are the momentum states chosen by Hughes in his text, and used in Freespace.

Handling Non-spherical Joints

When all joints are spherical, the momentum states $[\{H_k\}, P_0]$ form a minimum-dimension set of dynamical variables.

When some joints are 1-DOF or 2-DOF gimbals, it is advisable to reduce the order of the dynamical state accordingly.

Queen, London, and Gonzalez handled 1-DOF gimbals with scalar momentum states, which are the projection of H_k along the hinge axis.

We extend this technique to 2-DOF and 3-DOF gimbals using the idea of joint partials, as introduced in our development of multibody dynamics using Kane's method.

Joint Partial

Consider two bodies connected by a gimballed joint. The relative angular velocity of the outer body, B , with respect to the inner body, A , may be written as a function of the gimbal angles, θ , and the gimbal rates, $\sigma \equiv \dot{\theta}$. For example, a Body-3, 2-1-3 Euler rotation through $(\theta_1, \theta_2, \theta_3)$ gives:

$${}^A\omega^B = \underbrace{\begin{bmatrix} c_2 s_3 & c_3 & 0 \\ c_2 c_3 & -s_3 & 0 \\ -s_2 & 0 & 1 \end{bmatrix}}_I \underbrace{\begin{bmatrix} \sigma_1 \\ \sigma_2 \\ \sigma_3 \end{bmatrix}}_\sigma$$

where $s_2 = \sin(\theta_2)$, $c_2 = \cos(\theta_2)$, etc, and the components of ω are expressed in the frame of the outer body, B .

The matrix I is called the matrix of joint partials. Each Euler sequence, including one- and two-DOF gimbals, has its own I . The dimensions of I are 3 x NDOF.

For a spherical joint, $\sigma = \omega$, and $I = U$, the identity matrix.

Joint Partials and Momentum States

Noting that the columns of Γ are the components of the gimbal axes expressed in the joint's outer body frame, we define the scalar momentum states

$$h_{kj} \equiv H_k \cdot \Gamma_{kj},$$

or the 1-, 2-, or 3-DOF

$$h_k = \Gamma_k^T H_k$$

The minimum-dimension angular momentum state vector then becomes:

$$\begin{Bmatrix} h_k \end{Bmatrix} = \begin{Bmatrix} h_0 \\ h_1 \\ h_2 \\ h_3 \\ h_4 \end{Bmatrix} = \begin{Bmatrix} G_0 H^{B_1} + G_0 H^{B_2} + G_0 H^{B_3} + G_0 H^{B_4} + G_0 H^{B_5} \\ \Gamma_1^T (G_1 H^{B_2}) \\ \Gamma_2^T (G_2 H^{B_3} + G_2 H^{B_4} + G_2 H^{B_5}) \\ \Gamma_3^T (G_3 H^{B_4}) \\ \Gamma_4^T (G_4 H^{B_5}) \end{Bmatrix}$$

Construction Preliminaries

Using joint partials, we can systematically construct expressions for $[\{H_k\}, P_0]$ as functions of the minimum-dimension state vector, $u \equiv [\sigma_k, v_0]$.

To begin, let's define some terms:

$$\omega_i \equiv {}^N \omega^{B_i}$$

$$v_k \equiv {}^N v^{G_k}$$

${}^i C^j$ is the direction cosine matrix between B_i and B_j ,

so the components of a vector may be transformed thus: ${}^i v = {}^i C^{jj} v$

ρ_{ik} is the vector from R_i to G_k . It is expressed in N .

The cross-product matrix is denoted by: $\tilde{x} = \begin{bmatrix} 0 & -x_3 & x_2 \\ x_3 & 0 & -x_1 \\ -x_2 & x_1 & 0 \end{bmatrix}, \quad x \times y = \tilde{x} y$

Assembly Notation

As we assemble the system, we adopt a not-uncommon notation to distinguish system-level properties from body-level properties.

In cases where a common symbol is used for both, the system-level vectors are enclosed in braces, $\{\cdot\}$, and system-level matrices are enclosed in brackets, $[\cdot]$ to reduce confusion.

For example, we use J to denote the inertia tensor of a generic single body. J_i denotes the inertia tensor of the i th body, and $[J]$ denotes a system matrix formed by assembling the single-body inertia tensors:

$$[J] \equiv \text{diag}[J_1 \quad J_2 \quad J_3 \quad \dots]$$

Linear and Angular Velocities

$$\begin{pmatrix} \omega_1 \\ \omega_2 \\ \omega_3 \\ \omega_4 \\ \omega_5 \end{pmatrix} = \underbrace{\begin{bmatrix} U & 0 & 0 & 0 & 0 & 0 \\ {}^2C^1 & \Gamma_1 & 0 & 0 & 0 & 0 \\ {}^3C^1 & 0 & \Gamma_2 & 0 & 0 & 0 \\ {}^4C^1 & 0 & {}^4C^3\Gamma_2 & \Gamma_3 & 0 & 0 \\ {}^5C^1 & 0 & {}^5C^3\Gamma_2 & 0 & \Gamma_4 & 0 \end{bmatrix}}_{\Omega} \underbrace{\begin{pmatrix} \sigma_0 \\ \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ v_0 \end{pmatrix}}_u$$

$$\begin{pmatrix} v_0 \\ v_1 \\ v_2 \\ v_3 \\ v_4 \end{pmatrix} = \underbrace{\begin{bmatrix} 0 & 0 & 0 & 0 & 0 & U \\ \tilde{\rho}_{20}^N C^1 & \tilde{\rho}_{21}^N C^2 \Gamma_1 & 0 & 0 & 0 & U \\ \tilde{\rho}_{30}^N C^1 & 0 & \tilde{\rho}_{32}^N C^3 \Gamma_2 & 0 & 0 & U \\ \tilde{\rho}_{40}^N C^1 & 0 & \tilde{\rho}_{42}^N C^3 \Gamma_2 & \tilde{\rho}_{43}^N C^4 \Gamma_3 & 0 & U \\ \tilde{\rho}_{50}^N C^1 & 0 & \tilde{\rho}_{52}^N C^3 \Gamma_2 & 0 & \tilde{\rho}_{54}^N C^5 \Gamma_4 & U \end{bmatrix}}_V \underbrace{\begin{pmatrix} \sigma_0 \\ \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ v_0 \end{pmatrix}}_u$$

Body Momenta

Using:

$${}^N P^{B*} = m {}^N v^R - c \times {}^N \omega^B,$$

$${}^R H^B = J {}^N \omega^B + c \times {}^N v^R,$$

Let

$$[m] \equiv \text{diag} \begin{bmatrix} m_1 U & m_2 U & m_3 U & m_4 U & m_5 U \end{bmatrix}$$

$$[J] \equiv \text{diag} \begin{bmatrix} J_1 & J_2 & J_3 & J_4 & J_5 \end{bmatrix}$$

$$[\tilde{c}] \equiv \text{diag} \begin{bmatrix} {}^N C^1 \tilde{c}_1 & {}^N C^2 \tilde{c}_2 & {}^N C^3 \tilde{c}_3 & {}^N C^4 \tilde{c}_4 & {}^N C^5 \tilde{c}_5 \end{bmatrix}$$

yielding

$$\{ {}^N P^{B*} \} = [[m] V - [\tilde{c}] \Omega] u$$

$$\{ {}^R H^B \} = [[J] \Omega - [\tilde{c}]^T V] u$$

Composite Momenta

Using

$${}^G H^B = {}^R H^B + {}^G r^R \times {}^N P^{B*}$$

Let

$$S \equiv \begin{bmatrix} U & {}^1C^2 & {}^1C^3 & {}^1C^4 & {}^1C^5 \\ & U & & & \\ & & U & {}^3C^4 & {}^3C^5 \\ & & & U & \\ & & & & U \end{bmatrix}, \quad R \equiv \begin{bmatrix} 0 & {}^1C^N \tilde{\rho}_{20} & {}^1C^N \tilde{\rho}_{30} & {}^1C^N \tilde{\rho}_{40} & {}^1C^N \tilde{\rho}_{50} \\ & 0 & & & \\ & & 0 & {}^3C^N \tilde{\rho}_{42} & {}^3C^N \tilde{\rho}_{52} \\ & & & 0 & \\ & & & & 0 \end{bmatrix}$$

yielding

$$\{H_k\} = S \{ {}^R H^B \} - R \{ {}^N P^{B*} \}$$

or

$$\begin{aligned} \{H_k\} &= [(S[J] + R[\tilde{c}])\Omega - (S[\tilde{c}]^T + R[m])V]u \\ \{h_k\} &= [\Gamma_k]^T \{H_k\} = [\Gamma_k]^T [(S[J] + R[\tilde{c}])\Omega - (S[\tilde{c}]^T + R[m])V]u \end{aligned}$$

Similarly,

$$\Sigma \equiv \begin{bmatrix} U & U & U & U & U \\ & U & & & \\ & & U & U & U \\ & & & U & \\ & & & & U \end{bmatrix} \text{ gives } \{P_k\} = \Sigma [m]V - [\tilde{c}]\Omega]u$$

Partial Momenta

Define

$$A = (S[J] + R[\tilde{c}])\Omega - (S[\tilde{c}]^T + R[m])V$$

$$\Lambda = \begin{bmatrix} U & U & U & U & U \end{bmatrix} \begin{bmatrix} [m]V - [\tilde{c}]\Omega \end{bmatrix}$$

so

$$\{H_k\} = Au, \quad \{h_k\} = [\Gamma_k]^T Au$$

$$P_0 = \Lambda u$$

In perfect analogy to Ω and V , A and Λ are, respectively, the partial angular momenta and partial linear momenta relating $\{H_k\}$ and P_0 to the dynamic states, u .

MSD Equations of Motion

Integrate:

$$\begin{Bmatrix} \dot{P}_0 \\ \{\dot{h}_k\} \end{Bmatrix} = \begin{Bmatrix} F_0 \\ [\Gamma_k]^T \left\{ T_k - v_k \times P_k - \sum_i \omega_i \times^{G_k} H^{B_i} \right\} \end{Bmatrix}$$

(where F_0 is the total resultant force on the spacecraft, applied at the center of mass, and T_k is the resultant torque about G_k of all forces and moments acting on bodies outboard of G_k)

Solve for u :

$$\begin{Bmatrix} P_0 \\ \{h_k\} \end{Bmatrix} = \begin{bmatrix} \Lambda \\ [\Gamma_k]^T A \end{bmatrix} u$$

Construct $\{\omega_i\}, \{v_k\}$:

$$\begin{aligned} \{\omega_i\} &= \Omega u \\ \{v_k\} &= V u \end{aligned}$$

Repeat

Locking a Single DOF

To lock the j th DOF of joint G_k , simply set the corresponding element of u to zero, and eliminate the corresponding \dot{h}_{kj} from the EOM to be integrated.

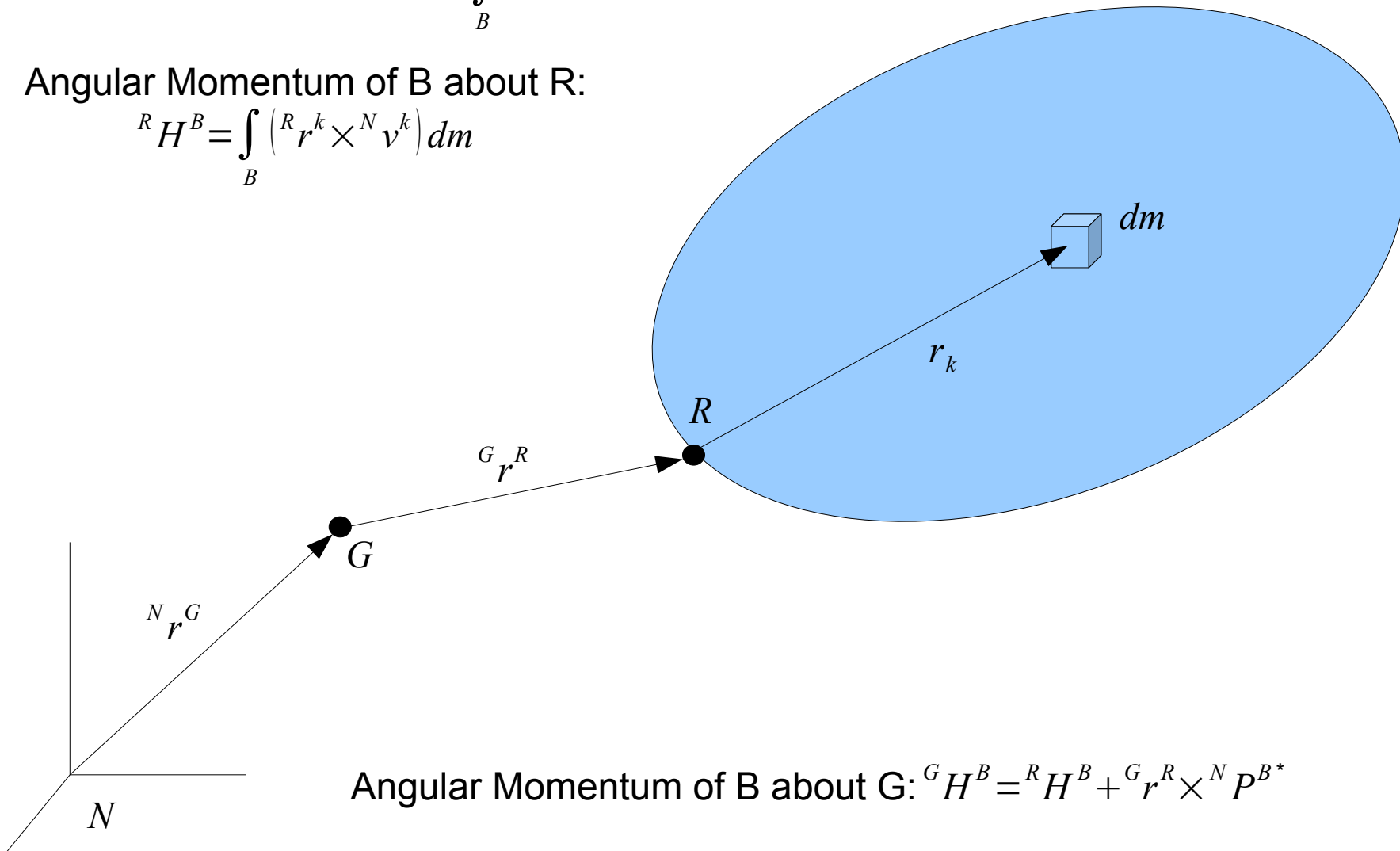
Flexible Bodies

Linear and Angular Momentum

Linear Momentum: ${}^N P^{B*} = \int_B {}^N v^k dm$

Angular Momentum of B about R:

$${}^R H^B = \int_B ({}^R r^k \times {}^N v^k) dm$$



Angular Momentum of B about G: ${}^G H^B = {}^R H^B + {}^G r^R \times {}^N P^{B*}$

Linear and Angular Momentum

Linear Momentum:

$$\begin{aligned}
 {}^N P^{B*} &= \int_B {}^N v^k dm \\
 &= m {}^N v^R + {}^N \omega^B \times (c + P_\Psi \eta) + P_\Psi \xi
 \end{aligned}$$

Angular Momentum of B about R:

$$\begin{aligned}
 {}^R H^B &= \int_B ({}^R r^k \times {}^N v^k) dm \\
 &= \int_B (r_k + \Psi_k \eta) \times \left[{}^N v^R + {}^N \omega^B \times (r_k + \Psi_k \eta) + \Psi_k \xi \right] dm \\
 &= J {}^N \omega^B - {}^N v^R \times (c + P_\Psi \eta) + (H_\Psi + Q_\Psi \eta) \xi - {}^N \omega^B \cdot (R_\Psi + R_\Psi^T) \eta
 \end{aligned}$$

Linear and Angular Velocities

$$\begin{pmatrix} \omega_1 \\ \omega_2 \\ \omega_3 \\ \omega_4 \\ \omega_5 \end{pmatrix} = \underbrace{\begin{bmatrix} U & 0 & 0 & 0 & 0 & 0 \\ {}^2C^1 & \Gamma_1 & 0 & 0 & 0 & 0 \\ {}^3C^1 & 0 & \Gamma_2 & 0 & 0 & 0 \\ {}^4C^1 & 0 & {}^4C^3\Gamma_2 & \Gamma_3 & 0 & 0 \\ {}^5C^1 & 0 & {}^5C^3\Gamma_2 & 0 & \Gamma_4 & 0 \end{bmatrix}}_{\Omega_u} \underbrace{\begin{pmatrix} \sigma_0 \\ \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ v_0 \end{pmatrix}}_u + \underbrace{\begin{bmatrix} 0 & 0 & 0 & 0 & 0 \\ {}^2C^1\Theta_{11} & -\Theta_{21} & 0 & 0 & 0 \\ {}^3C^1\Theta_{12} & 0 & -\Theta_{32} & 0 & 0 \\ {}^4C^1\Theta_{12} & 0 & {}^4C^3(\Theta_{33}-\Theta_{32}) & -\Theta_{43} & 0 \\ {}^5C^1\Theta_{12} & 0 & {}^5C^3(\Theta_{34}-\Theta_{32}) & 0 & -\Theta_{54} \end{bmatrix}}_{\Omega_\xi} \underbrace{\begin{pmatrix} \xi_1 \\ \xi_2 \\ \xi_3 \\ \xi_4 \\ \xi_5 \end{pmatrix}}_\xi$$

$$\begin{pmatrix} v_0 \\ v_1 \\ v_2 \\ v_3 \\ v_4 \end{pmatrix} = \underbrace{\begin{bmatrix} 0 & 0 & 0 & 0 & 0 & U \\ \tilde{\rho}_{20} {}^NC^1 & \tilde{\rho}_{21} {}^NC^2\Gamma_1 & 0 & 0 & 0 & U \\ \tilde{\rho}_{30} {}^NC^1 & 0 & \tilde{\rho}_{32} {}^NC^3\Gamma_2 & 0 & 0 & U \\ \tilde{\rho}_{40} {}^NC^1 & 0 & \tilde{\rho}_{42} {}^NC^3\Gamma_2 & \tilde{\rho}_{43} {}^NC^4\Gamma_3 & 0 & U \\ \tilde{\rho}_{50} {}^NC^1 & 0 & \tilde{\rho}_{52} {}^NC^3\Gamma_2 & 0 & \tilde{\rho}_{54} {}^NC^5\Gamma_4 & U \end{bmatrix}}_{V_u} \underbrace{\begin{pmatrix} \sigma_0 \\ \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ v_0 \end{pmatrix}}_u +$$

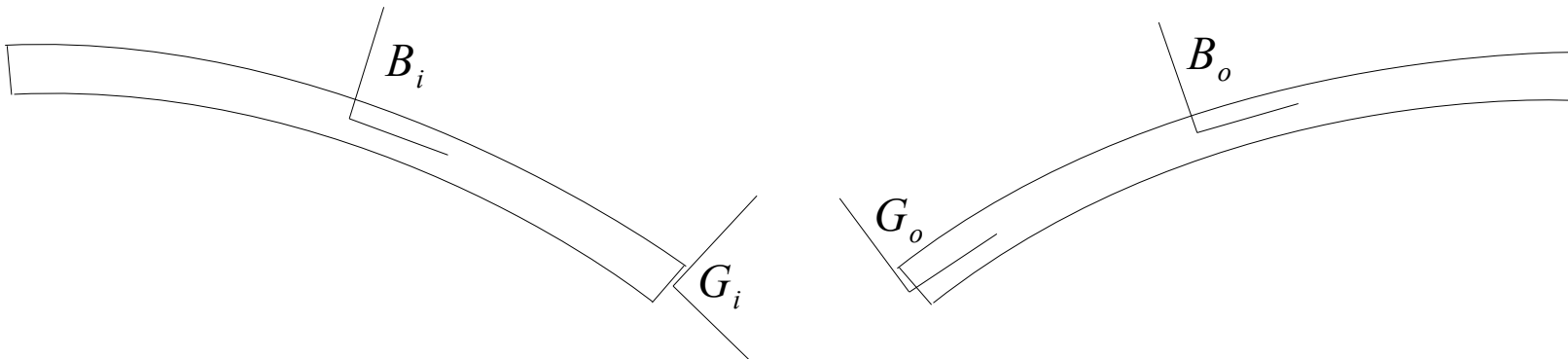
$$\underbrace{\begin{bmatrix} 0 & 0 & 0 & 0 & 0 \\ {}^NC^1\Psi_{11} + \tilde{\rho}_{21} {}^NC^1\Theta_{11} & -{}^NC^2\Psi_{21} - \tilde{\rho}_{21} {}^NC^2\Theta_{21} & 0 & 0 & 0 \\ {}^NC^1\Psi_{12} + \tilde{\rho}_{32} {}^NC^1\Theta_{12} & 0 & -{}^NC^3\Psi_{32} - \tilde{\rho}_{32} {}^NC^3\Theta_{32} & 0 & 0 \\ {}^NC^1\Psi_{12} + \tilde{\rho}_{42} {}^NC^1\Theta_{12} & 0 & {}^NC^3(\Psi_{33} - \Psi_{32}) + \tilde{\rho}_{43} {}^NC^3\Theta_{33} - \tilde{\rho}_{42} {}^NC^3\Theta_{32} & -{}^NC^4\Psi_{43} - \tilde{\rho}_{43} {}^NC^4\Theta_{43} & 0 \\ {}^NC^1\Psi_{12} + \tilde{\rho}_{52} {}^NC^1\Theta_{12} & 0 & {}^NC^3(\Psi_{34} - \Psi_{32}) + \tilde{\rho}_{54} {}^NC^3\Theta_{34} - \tilde{\rho}_{52} {}^NC^3\Theta_{32} & 0 & -{}^NC^5\Psi_{54} - \tilde{\rho}_{54} {}^NC^5\Theta_{54} \end{bmatrix}}_{V_\xi} \underbrace{\begin{pmatrix} \xi_1 \\ \xi_2 \\ \xi_3 \\ \xi_4 \\ \xi_5 \end{pmatrix}}_\xi$$

Effect of Flexibility on Rotation at Joints

At joint G_k , the relative attitude of the outer body, B_o , with respect to the inner body, B_i , must account for the rotation due to the flexibility of both bodies at the joint, as well as the "rigid-body" rotation due to the gimballed or spherical joint.

Introducing the intermediate frames G_o and G_i affixed, respectively, to the interfaces between the "rigid" joint and the outer and inner bodies,

$${}^oC^i = \underbrace{{}^{Bo}C^{Go}}_{C(\Theta_{ko} \eta_o)} \underbrace{{}^{Go}C^{Gi}}_{C(\theta)} \underbrace{{}^{Gi}C^{Bi}}_{C(\Theta_{ki} \eta_i)}$$



Effect of Flexibility on $\tilde{\rho}_{ik}$

Recall that $\tilde{\rho}_{ik}$ denotes the vector from the reference point R_i (fixed in the i th body, B_i) to the k th joint, G_k .

We keep the same notation, but keep in mind that this vector now includes deflections due to flexibility (Ψ_η) as appropriate over the traversal from R_i to G_k .

Body Momenta

Using:

$${}^N P^{B*} = m {}^N v^R - (c + P_\Psi \eta) \times {}^N \omega^B + P_\Psi \xi,$$

$${}^R H^B = J {}^N \omega^B - {}^N v^R \times (c + P_\Psi \eta) + (H_\Psi + Q_\Psi \eta) \xi - {}^N \omega^B \cdot (R_\Psi + R_\Psi^T) \eta,$$

Let

$$\underline{J}_i \equiv (J - \eta^T (R_\Psi + R_\Psi^T))_i$$

$$[m] \equiv \text{diag} [m_1 U \quad m_2 U \quad m_3 U \quad m_4 U \quad m_5 U]$$

$$[\underline{J}] \equiv \text{diag} [\underline{J}_1 \quad \underline{J}_2 \quad \underline{J}_3 \quad \underline{J}_4 \quad \underline{J}_5]$$

$$[\tilde{\underline{C}}] \equiv \text{diag} [{}^N C^1 (c + P_\Psi \eta)_1^\times \quad {}^N C^2 (c + P_\Psi \eta)_2^\times \quad {}^N C^3 (c + P_\Psi \eta)_3^\times \quad {}^N C^4 (c + P_\Psi \eta)_4^\times \quad {}^N C^5 (c + P_\Psi \eta)_5^\times]$$

$$[P_\Psi] \equiv \text{diag} [{}^N C^1 P_{\Psi 1} \quad {}^N C^2 P_{\Psi 2} \quad {}^N C^3 P_{\Psi 3} \quad {}^N C^4 P_{\Psi 4} \quad {}^N C^5 P_{\Psi 5}]$$

$$[\underline{H}_\Psi] \equiv \text{diag} [(H_\Psi + Q_\Psi \eta)_1 \quad (H_\Psi + Q_\Psi \eta)_2 \quad (H_\Psi + Q_\Psi \eta)_3 \quad (H_\Psi + Q_\Psi \eta)_4 \quad (H_\Psi + Q_\Psi \eta)_5]$$

yielding

$$\{{}^N P^{B*}\} = [m] V_u - [\tilde{\underline{C}}] \Omega_u u + [m] V_\xi - [\tilde{\underline{C}}] \Omega_\xi + [P_\Psi] \xi$$

$$\{{}^R H^B\} = [\underline{J}] \Omega_u - [\tilde{\underline{C}}]^T V_u u + [\underline{J}] \Omega_\xi - [\tilde{\underline{C}}]^T V_\xi + [\underline{H}_\Psi] \xi$$

Composite Momenta

Using

$${}^G H^B = {}^R H^B + {}^G r^R \times {}^N P^{B*}$$

Let

$$S \equiv \begin{bmatrix} U & {}^1C^2 & {}^1C^3 & {}^1C^4 & {}^1C^5 \\ & U & & & \\ & & U & {}^3C^4 & {}^3C^5 \\ & & & U & \\ & & & & U \end{bmatrix}, \quad R \equiv \begin{bmatrix} 0 & {}^1C^N \tilde{\rho}_{20} & {}^1C^N \tilde{\rho}_{30} & {}^1C^N \tilde{\rho}_{40} & {}^1C^N \tilde{\rho}_{50} \\ & 0 & & & \\ & & 0 & {}^3C^N \tilde{\rho}_{42} & {}^3C^N \tilde{\rho}_{52} \\ & & & 0 & \\ & & & & 0 \end{bmatrix}$$

yielding

$$\{H_k\} = S \{ {}^R H^B \} - R \{ {}^N P^{B*} \}$$

or

$$\{H_k\} = \left[(S[\underline{J}] + R[\underline{\tilde{c}}]) \Omega_u - (S[\underline{\tilde{c}}]^T + R[m]) V_u \right] u + \left[(S[\underline{J}] + R[\underline{\tilde{c}}]) \Omega_\xi - (S[\underline{\tilde{c}}]^T + R[m]) V_\xi + S[\underline{H}_\Psi] - R[P_\Psi] \right] \xi$$

$$\{h_k\} = [\Gamma_k]^T \{H_k\}$$

Similarly,

$$\Sigma \equiv \begin{bmatrix} U & U & U & U & U \\ & U & & & \\ & & U & U & U \\ & & & U & \\ & & & & U \end{bmatrix} \text{ gives } \{P_k\} = \Sigma \left\{ \left[[m] V_u - [\underline{\tilde{c}}] \Omega_u \right] u + \left[[m] V_\xi - [\underline{\tilde{c}}] \Omega_\xi + [P_\Psi] \right] \xi \right\}$$

Partial Momenta

Define

$$A_\Psi = \begin{bmatrix} (S[\underline{J}] + R[\tilde{\underline{c}}])\Omega_u - (S[\tilde{\underline{c}}]^T + R[m])V_u & (S[\underline{J}] + R[\tilde{\underline{c}}])\Omega_\xi - (S[\tilde{\underline{c}}]^T + R[m])V_\xi + S[\underline{H}_\Psi] - R[P_\Psi] \end{bmatrix}$$

$$\Lambda_\Psi = \begin{bmatrix} U & U & U & U & U \end{bmatrix} \begin{bmatrix} [m]V_u - [\tilde{\underline{c}}]\Omega_u & [m]V_\xi - [\tilde{\underline{c}}]\Omega_\xi + [P_\Psi] \end{bmatrix}$$

so

$$\{H_k\} = A_\Psi \begin{Bmatrix} u \\ \xi \end{Bmatrix}, \quad \{h_k\} = [\Gamma_k]^T A_\Psi \begin{Bmatrix} u \\ \xi \end{Bmatrix}$$

$$P_0 = \Lambda_\Psi \begin{Bmatrix} u \\ \xi \end{Bmatrix}$$

Flexible Body Momenta

The momentum states discussed so far are extensions of the rigid-body momentum states. We still need some equations of motion for the flexible modes of each body. Ken London suggests we use the momentum states:

$$\Pi \equiv \int_B \Psi_k^T {}^N v^k dm$$

with the equations of motion:

$$\dot{\Pi} = \int_B \Psi_k^T df$$

Expressing ${}^N v^k$ as:

$${}^N v^k = {}^N v^R + {}^N \omega^B \times ({}^R r^k + \Psi_k \eta) + \Psi_k \xi$$

and referring to modal integrals defined in Kane_NBody_Flex slides, we immediately obtain

$$\Pi = P_\Psi^T {}^N v^R + (H_\Psi + Q_\Psi \eta)^T {}^N \omega^B + M_\Psi \xi$$

or

$$\{\Pi\} = \begin{bmatrix} [H_\Psi]^T \Omega_u + [P_\Psi]^T V_u & [H_\Psi]^T \Omega_\xi + [P_\Psi]^T V_\xi + [M_\Psi] \end{bmatrix} \begin{Bmatrix} u \\ \xi \end{Bmatrix}$$

For compactness, we introduce Φ_Ψ ,

$$\{\Pi\} = \Phi_\Psi \begin{Bmatrix} u \\ \xi \end{Bmatrix}$$

MSD Equations of Motion

Integrate:

$$\begin{pmatrix} \dot{P}_0 \\ \{\dot{h}_k\} \\ \{\dot{\Pi}_i\} \end{pmatrix} = \begin{pmatrix} F_0 \\ [\Gamma_k]^T \left\{ T_k - v_k \times P_k - \sum_i \omega_i \times^{G_k} H^{B_i} \right\} \\ \left\{ \int_{B_i} \Psi^T df \right\} \end{pmatrix}$$

(where F_0 is the total resultant force on the spacecraft, applied at the center of mass, and T_k is the resultant torque about G_k of all forces and moments acting on bodies outboard of G_k).

As discussed in the Kane_NBody_Flex slides, $\int_{B_i} \Psi^T df$ comprises spring and damping forces, $-C_\Psi \xi - K_\Psi \eta$, as well as any external forces or moments that have zero resultants on the body, but non-zero projections into the modal space.

Solve for u, ξ :

$$\begin{pmatrix} P_0 \\ \{h_k\} \\ \{\Pi\} \end{pmatrix} = \begin{bmatrix} \Lambda_\Psi \\ [\Gamma_k]^T A_\Psi \\ \Phi_\Psi \end{bmatrix} \begin{pmatrix} u \\ \xi \end{pmatrix}$$

Construct $\{\omega_i\}, \{v_k\}$:

$$\begin{aligned} \{\omega_i\} &= \Omega_u u + \Omega_\xi \xi \\ \{v_k\} &= V_u u + V_\xi \xi \end{aligned}$$

Repeat