



MA3005 Cheat Sheet

Control Theory (Nanyang Technological University)

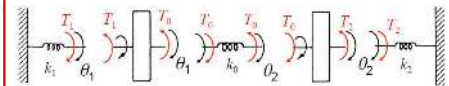


Scan to open on Studocu

1. Final value theorem: $\lim_{t \rightarrow \infty} f(t) = \lim_{s \rightarrow 0} sF(s)$. Poles of $F(s)$ lie in left half s-plane and at most 1 pole on imaginary axis.

[Steady state behaviour]

2. Initial value theorem: $f(0^+) = \lim_{s \rightarrow \infty} sF(s)$. No restriction on poles location, applicable for sinusoidal functions.



$f(t)$	$F(s) = \mathcal{L}[f(t)]$	
$f(t) = 1$	$F(s) = \frac{1}{s}$	$s > 0$
$f(t) = e^{at}$	$F(s) = \frac{1}{(s-a)}$	$s > a$
$f(t) = t^n$	$F(s) = \frac{n!}{s^{n+1}}$	$s > 0$
$f(t) = \sin(at)$	$F(s) = \frac{a}{s^2 + a^2}$	$s > 0$
$f(t) = \cos(at)$	$F(s) = \frac{s}{s^2 + a^2}$	$s > 0$
$f(t) = \sinh(at)$	$F(s) = \frac{a}{s^2 - a^2}$	$s > a $
$f(t) = \cosh(at)$	$F(s) = \frac{s}{s^2 - a^2}$	$s > a $
$f(t) = t^n e^{at}$	$F(s) = \frac{n!}{(s-a)^{n+1}}$	$s > a$
$f(t) = e^{at} \sin(bt)$	$F(s) = \frac{b}{(s-a)^2 + b^2}$	$s > a$
$f(t) = e^{at} \cos(bt)$	$F(s) = \frac{(s-a)}{(s-a)^2 + b^2}$	$s > a$
$f(t) = e^{at} \sinh(bt)$	$F(s) = \frac{b}{(s-a)^2 - b^2}$	$s - a > b $
$f(t) = e^{at} \cosh(bt)$	$F(s) = \frac{(s-a)}{(s-a)^2 - b^2}$	$s - a > b $

Routh-Hurwitz (Closed Loop CE):

C.E.: $a_0 s^n + a_1 s^{n-1} + a_2 s^{n-2} + \dots + a_n = 0$

None of the coefficient is missing

From Characteristic Eqn.

Need to be computed

h₁ = a₁a₂ - a₀a₃, h₂ = a₂a₃ - a₁a₄

c₁ = h₁a₂ - a₁h₂, c₂ = h₂a₃ - a₂h₃

h₃ = a₂a₄ - a₁a₅, c₃ = h₃a₃ - a₂h₄

h₄ = a₃a₄ - a₂a₅, c₄ = h₄a₄ - a₃h₅

h₅ = a₄a₅ - a₃a₆, c₅ = h₅a₅ - a₄h₆

h₆ = a₅a₆ - a₄a₇, c₆ = h₆a₆ - a₅h₇

h₇ = a₆a₇ - a₅a₈, c₇ = h₇a₇ - a₆h₈

h₈ = a₇a₈ - a₆a₉, c₈ = h₈a₈ - a₇h₉

h₉ = a₈a₉ - a₇a₁₀, c₉ = h₉a₉ - a₈h₁₀

h₁₀ = a₉a₁₀ - a₈a₁₁, c₁₀ = h₁₀a₁₀ - a₉h₁₁

h₁₁ = a₁₀a₁₁ - a₉a₁₂, c₁₁ = h₁₁a₁₁ - a₁₀h₁₂

h₁₂ = a₁₁a₁₂ - a₁₀a₁₃, c₁₂ = h₁₂a₁₂ - a₁₁h₁₃

h₁₃ = a₁₂a₁₃ - a₁₁a₁₄, c₁₃ = h₁₃a₁₃ - a₁₂h₁₄

h₁₄ = a₁₃a₁₄ - a₁₂a₁₅, c₁₄ = h₁₄a₁₄ - a₁₃h₁₅

h₁₅ = a₁₄a₁₅ - a₁₃a₁₆, c₁₅ = h₁₅a₁₅ - a₁₄h₁₆

h₁₆ = a₁₅a₁₆ - a₁₄a₁₇, c₁₆ = h₁₆a₁₆ - a₁₅h₁₇

h₁₇ = a₁₆a₁₇ - a₁₅a₁₈, c₁₇ = h₁₇a₁₇ - a₁₆h₁₈

h₁₈ = a₁₇a₁₈ - a₁₆a₁₉, c₁₈ = h₁₈a₁₈ - a₁₇h₁₉

h₁₉ = a₁₈a₁₉ - a₁₇a₂₀, c₁₉ = h₁₉a₁₉ - a₁₈h₂₀

h₂₀ = a₁₉a₂₀ - a₁₈a₂₁, c₂₀ = h₂₀a₂₀ - a₁₉h₂₁

h₂₁ = a₂₀a₂₁ - a₁₉a₂₂, c₂₁ = h₂₁a₂₁ - a₂₀h₂₂

h₂₂ = a₂₁a₂₂ - a₂₀a₂₃, c₂₂ = h₂₂a₂₂ - a₂₁h₂₃

h₂₃ = a₂₂a₂₃ - a₂₁a₂₄, c₂₃ = h₂₃a₂₃ - a₂₂h₂₄

h₂₄ = a₂₃a₂₄ - a₂₂a₂₅, c₂₄ = h₂₄a₂₄ - a₂₃h₂₅

h₂₅ = a₂₄a₂₅ - a₂₃a₂₆, c₂₅ = h₂₅a₂₅ - a₂₄h₂₆

h₂₆ = a₂₅a₂₆ - a₂₄a₂₇, c₂₆ = h₂₆a₂₆ - a₂₅h₂₇

h₂₇ = a₂₆a₂₇ - a₂₅a₂₈, c₂₇ = h₂₇a₂₇ - a₂₆h₂₈

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h₃₁ = a₃₀a₃₁ - a₂₉a₃₂, c₃₁ = h₃₁a₃₁ - a₃₀h₃₂

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h₃₄ = a₃₃a₃₄ - a₃₂a₃₅, c₃₄ = h₃₄a₃₄ - a₃₃h₃₅

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h₄₀ = a₃₉a₄₀ - a₃₈a₄₁, c₄₀ = h₄₀a₄₀ - a₃₉h₄₁

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h₅₀ = a₄₉a₅₀ - a₄₈a₅₁, c₅₀ = h₅₀a₅₀ - a₄₉h₅₁

h₅₁ = a₅₀a₅₁ - a₄₉a₅₂, c₅₁ = h₅₁a₅₁ - a₅₀h₅₂

h₅₂ = a₅₁a₅₂ - a₅₀a₅₃, c₅₂ = h₅₂a₅₂ - a₅₁h₅₃

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h₈₂ = a₈₁a₈₂ - a₈₀a₈₃, c₈₂ = h₈₂a₈₂ - a₈₁h₈₃

h₈₃ = a₈₂a₈₃ - a₈₁a₈₄, c₈₃ = h₈₃a₈₃ - a₈₂h₈₄

h₈₄ = a₈₃a₈₄ - a₈₂a₈₅, c₈₄ = h₈₄a₈₄ - a₈₃h₈₅

h₈₅ = a₈₄a₈₅ - a₈₃a₈₆, c₈₅ = h₈₅a₈₅ - a₈₄h₈₆

h₈₆ = a₈₅a₈₆ - a₈₄a₈₇, c₈₆ = h₈₆a₈₆ - a₈₅h₈₇

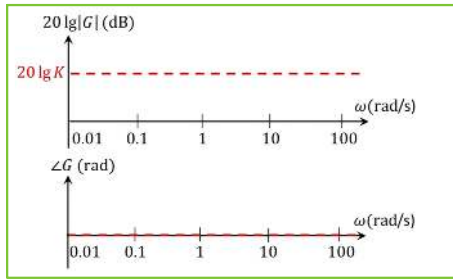
h₈₇ = a₈₆a₈₇ - a₈₅a₈₈, c₈₇ = h₈₇a₈₇ - a₈₆h₈₈

h₈₈ = a₈₇a₈₈ - a₈₆a₈₉, c₈₈ = h₈₈a₈₈

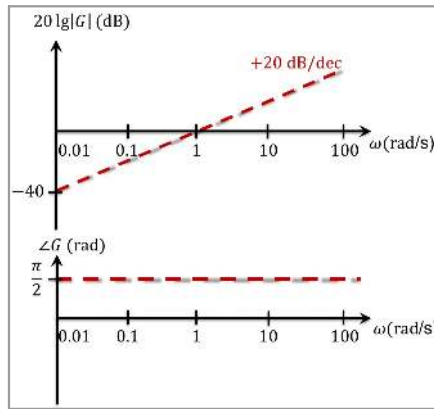
Bode Plot Sketching

1. Separate transfer function into various terms (E.g Constant, Differentiator, Integrator, 1st order pole etc)

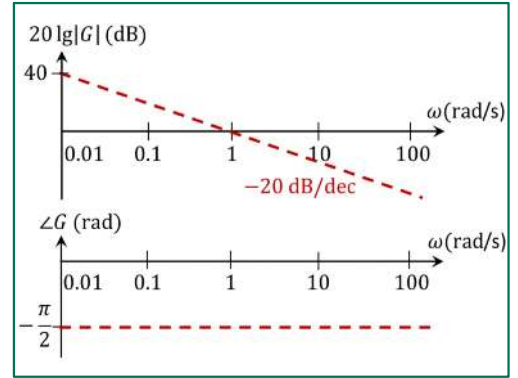
Plotting constant K



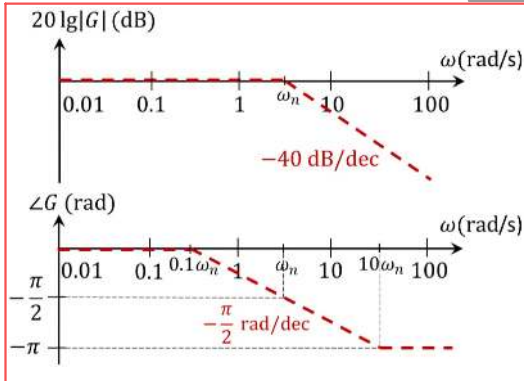
Plotting Differentiator s



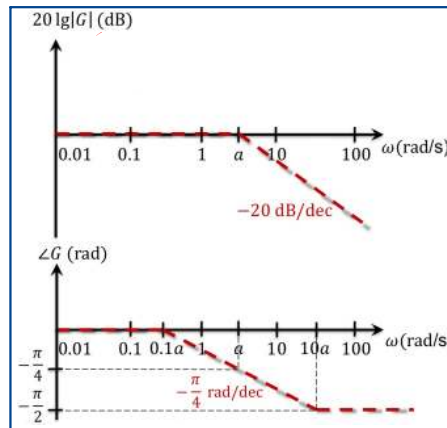
Plotting integrator $\frac{1}{s}$



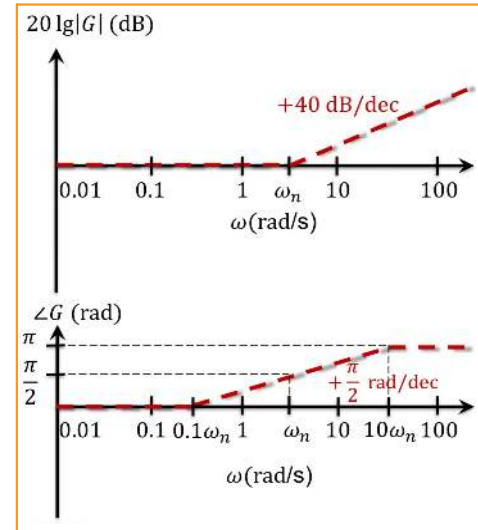
Plotting second order pole $\frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$



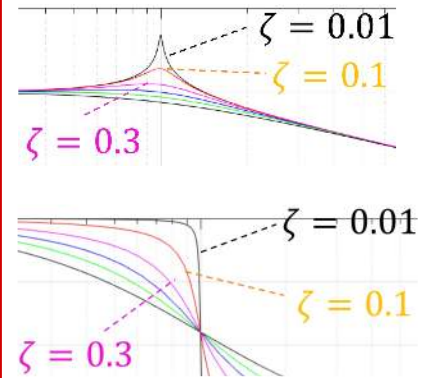
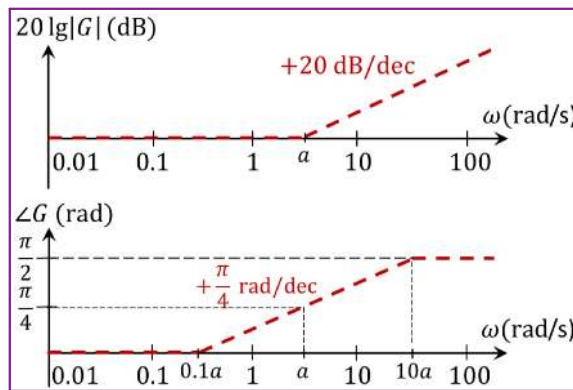
Plotting First Order Pole $\frac{a}{s+a}$



Plotting Second Order Zero $\frac{s^2 + 2\zeta\omega_n s + \omega_n^2}{\omega_n^2}$



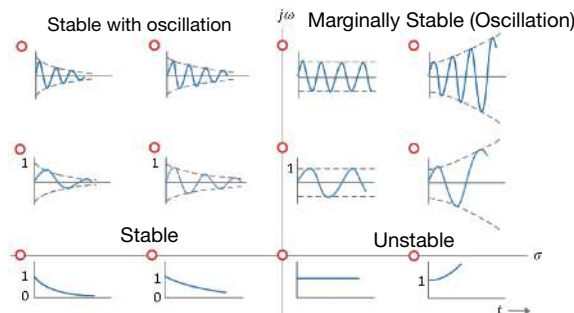
Plotting first order zero $\frac{s+a}{a}$



Bode Plot Analysis

Steady State Response (Look at magnitude plot)

- Type 0 (No initial slope)**
For step input, starting Y axis value = $20\log K_p$. Find K_p and sub into formula for $E_{ss} = 1/(1+K_p)$
- Type 1 (Initial slope is -20db/dec)**
For ramp input, draw -20db/dec line, intersecting 0db line. ω_n value at point of intersection = K_v . Sub K_v back into $E_{ss} = 1/K_v$
- Type 2 (Initial slope is -40db/dec)**
For parabolic inputs, draw -40db/dec line, intersecting 0db line. ω_n value at point of intersection = $\sqrt{K_a}$. Find K_a and sub back into $E_{ss} = 1/K_a$



Bode Plot Analysis

Approximating Transfer Function

- Check initial slope for Differentiators (+20db/dec) Integrators (-20db/dec)
- Find the Corner Frequencies ($0.1\omega_n$, ω_n and $10\omega_n$), and assume Damping Factor (0.01 for sharp, 0.3 for smooth), if theres resonance theres 2nd order pole
- Look at slope after $10\omega_n$, if slope changes, might have 1st order zero (E.g -60db/dec to -40db/dec)
- Look at slope right after ω_n , if it is consistent, if not somewhere before resonance theres another term (E.g first order pole)

Example to find K

Changes depending on starting X axis value

Low Frequency Gain Analysis

$$20 \log |\lim_{s \rightarrow j} G(s)| = 16.5 \rightarrow |\lim_{s \rightarrow j} G(s)| = 10^{16.5/20}$$

$$\rightarrow \left| \frac{K}{j} \frac{161}{-1 + 7.6j + 161} \right| = 10^{16.5/20} \rightarrow K = 6.6$$

Stability

Unity feedback is stable if and only if:

$$\gamma > 0 \text{ \& } GM > 0$$

If entire Magnitude plot below 0db $\gamma = \infty$
If entire Phase plot above 180 deg, $GM = \infty$

$$\lim_{s \rightarrow j} = 10$$

