

# **EE 4302 ADVANCED CONTROL SYSTEM**

## **State-Feedback Controller Design**

### **CA1 Report**

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## 1. Open-Loop System Analysis

Consider the following plant described in the state-space notation:

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ -1.20 & -3.71 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} u$$
$$y = x_1 \quad (1.1)$$

and where both state-variables  $x_1$  and  $x_2$  are measurable.

Use MATLAB to obtain plots of frequency response from  $u$  to  $x_1$  and  $x_2$  separately.

We can obtain the open-loop system performance.

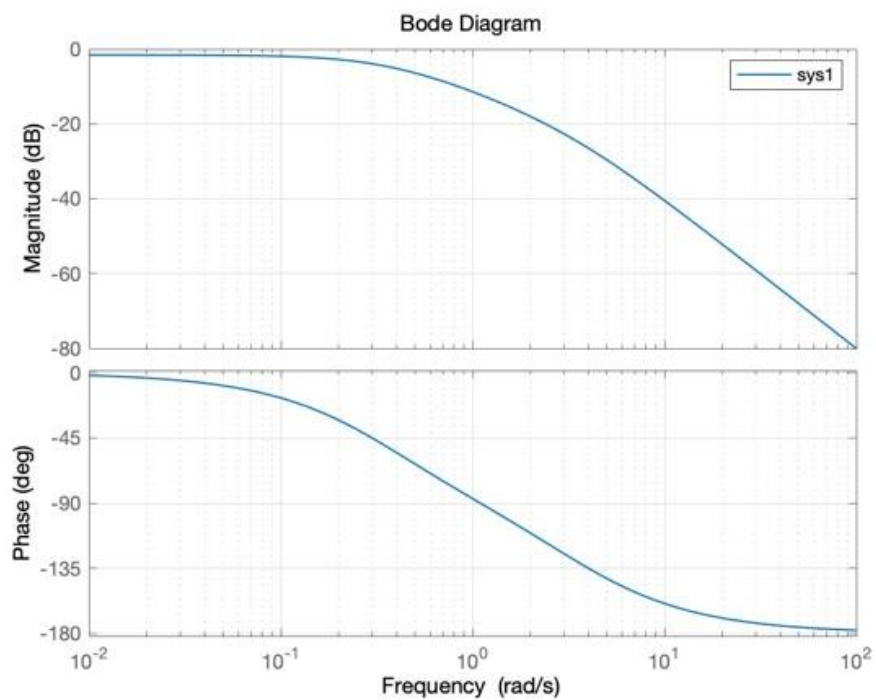


Figure 1 Frequency response from  $u$  to  $x_1$  of open-loop system

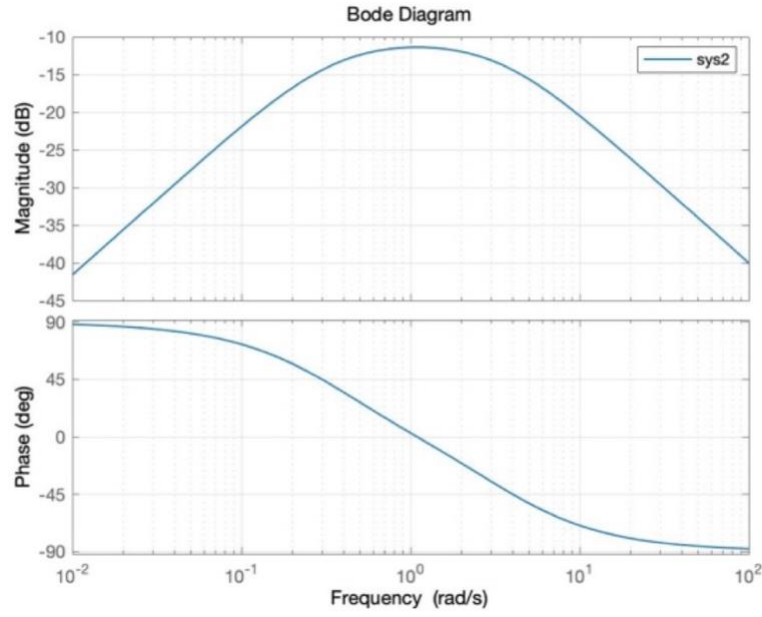


Figure 2 Frequency response from  $u$  to  $x_2$  of open-loop system

And calculate by MATLAB, the open-loop transfer function poles are located at:

$$p_1 = -0.3580 \quad p_2 = -3.3520$$

## 2. Design Using Ackermann's Formula

To apply Ackermann's Formula, we need to get the controllability matrix of the plant firstly.

The controllability matrix of the plant is:

$$\zeta = [G \quad FG] \quad (2.1)$$

i.e.

$$\zeta = \begin{bmatrix} 0 & 1 \\ 1 & -3.71 \end{bmatrix} \quad (2.2)$$

Calculate  $\det(\zeta)$  and check that the controllability matrix is nonsingular.

So, we can apply Ackermann's Formula to design the feedback controller.

Assume that the following control signal is applied to the plant

$$u = -Kx + K_S r \quad (2.3)$$

Where  $K$  is the controller gain;  $r$  is the reference signal;  $K_S$  is the scaling gain of  $r$ .

In order to apply the algorithm, we choose  $K = [0.1 \quad 0.1]$  and  $K_S = 1$  in the controller and apply it to the system. Then, use function FEEDBACK to obtain the closed-loop system.

It can be calculated that the closed-loop transfer function poles from  $r$  to  $y$  becomes:

$$p_1 = -0.3789, \quad p_2 = -3.4311$$

Plot the frequency response of the closed-loop system.

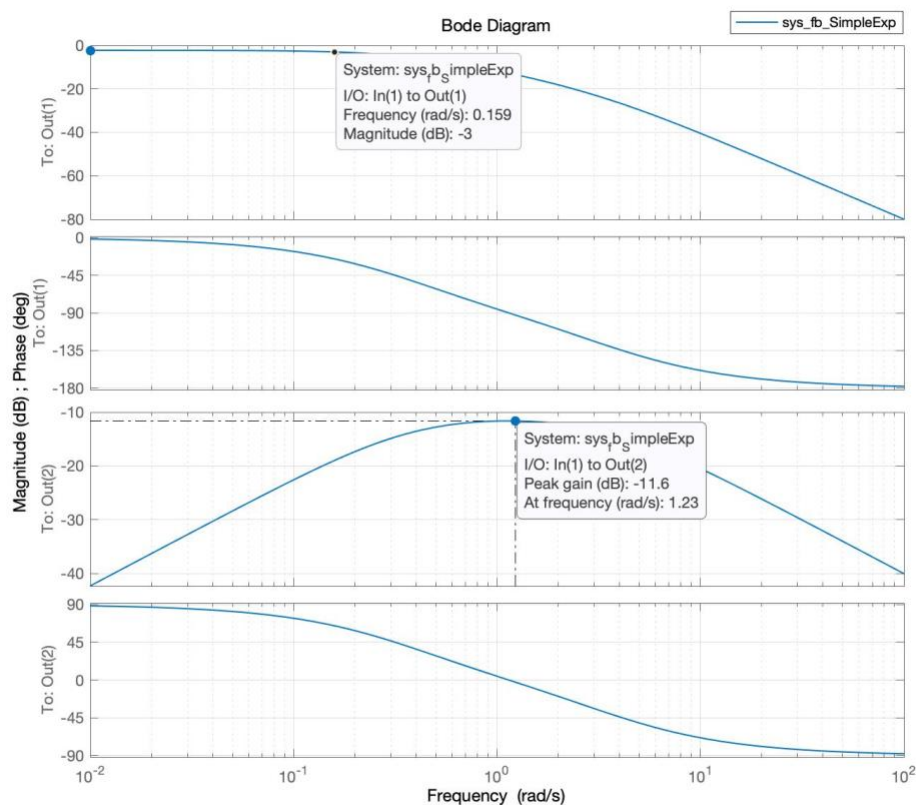


Figure 3 The frequency response from  $r$  to  $x_1$  and  $x_2$

The first two lines are frequency response from  $r$  to  $x_1$ . Since  $y = x_1$ , we can get the bandwidth of the closed-loop system is 0.159.

So, by changing the value of  $K$  and  $K_S$ , we can change the properties of the open-loop system and design the system as we desired.

(In the Bode diagram of closed-loop system, the intersection of the gain curve with the -3dB line is called the closed-loop cut-off frequency, also known as the closed-loop bandwidth frequency, denoted by  $\omega_b$ .)

## 2.1 Apply ITAE criterion

### 2.1.1 System Design

The specified requirements of the designs are:

Closed-loop bandwidth	not lower than 3.5 rad/s
Resonant peak $M_r$	not larger than 2dB
Steady-state gain between $r$ and $y$	0

Table 1 Design requirements

To meet the desired requirements, we choose the closed-loop poles based on ITAE table.

The bandwidth criterion is

$$\text{bandwidth} \geq 3.5 \text{ dB}$$

Refer to prototype response poles table for ITAE<sup>[1]</sup>, the system is second order and  $\omega_0$  is 4 rad/s, so the closed-loop poles are chosen as follows.

$$\begin{bmatrix} p_1 \\ p_2 \end{bmatrix} = 4 \begin{bmatrix} -0.7071 + 0.7071i \\ -0.7071 - 0.7071i \end{bmatrix} \quad (2.4)$$

With determined pole position, the desired closed-loop pole polynomial can be calculated as

$$\alpha_c(s) = s^2 + \alpha_1 s + \alpha_2 \quad (2.5)$$

and the controller gain  $K$  can be obtained using

$$K = [0 \quad 1] \alpha_c \zeta^{-1}(F) \quad (2.6)$$

However, using MATLAB function ‘acker’ and ‘dcgain’, we can easily get the controller gain  $K_{ITAE}$  and the scaling gain  $K_{S_{ITAE}}$ .

$$K_{ITAE} = 14.7997 \quad 1.9468 \quad K_{S_{ITAE}} = 15.9997 \quad (2.7)$$

i.e. the feedback controller is:

$$u = -[14.7997 \quad 1.9468]x + 15.9997r \quad (2.8)$$

MATLAB function acker and dcgain:

1. ACKER : ‘Pole placement gain selection using Ackermann's formula’.  
 $K = \text{acker}(A, B, P)$  calculates the feedback gain matrix  $K$  such that the single input system  $\dot{x} = Ax + Bu$  with a feedback law of  $u = -Kx$  has closed loop poles at the values specified in vector  $P$ , i.e.  $P = \text{eig}(A - B * K)$ .
2. DCGAIN:  $k = \text{dcgain}(\text{sys})$  computes the DC gain  $k$  of the LTI model  $\text{sys}$ .  
The continuous-time DC gain is the transfer function value at the frequency  $s=0$ . For state-space models with matrices  $(A, B, C, D)$ , the value is  $K = D - CA^{-1}B$ .

### 2.1.2 Closed-loop System Analysis

Then we can plot the frequency response and step response for closed loop system.

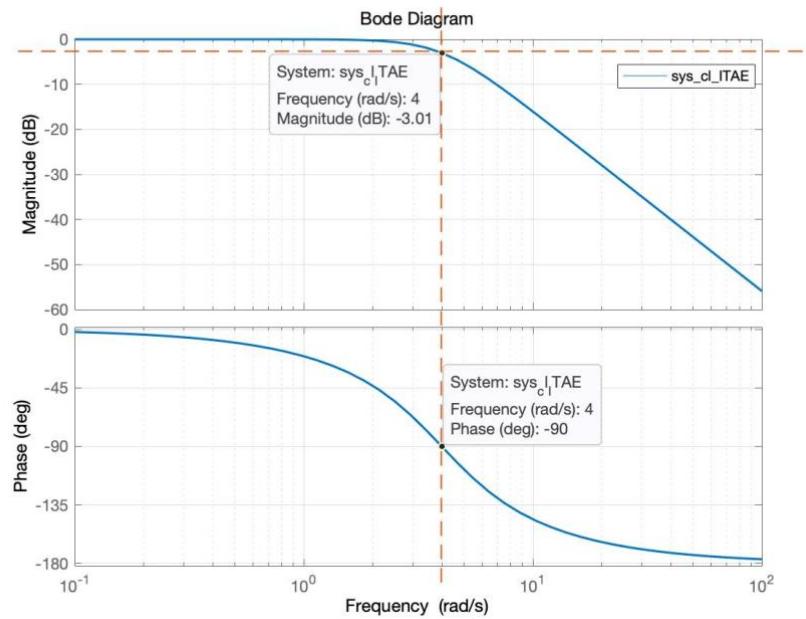


Figure 4 Frequency response for closed loop system using ITEA and Ackermann's formula

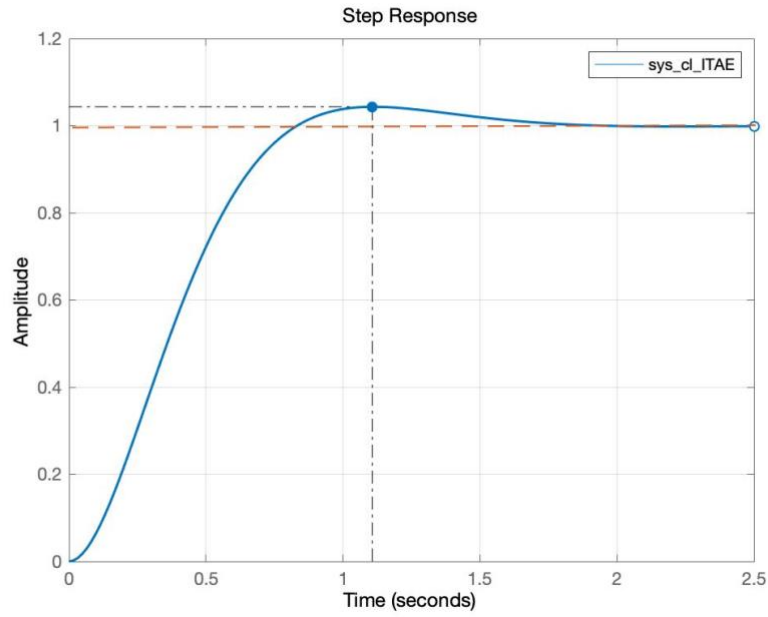


Figure 5 Step response for closed loop system using ITAE and Ackermann's formula

From figure 4 above we can draw the -3dB line and get the bandwidth frequency is 4 rad/s and also find the resonant peak  $M_r$  is equal to 0.

From figure 5, we can easily read that the steady-state gain between  $r$  and  $y$  is 1, which means the input and output are same, so the steady-state gain is 0dB.

The system meets all the design requirements.

## 2.2 Apply Bessel criterion

### 2.2.1 System Design

Similar to 2.1, we choose the closed-loop poles based on Bessel table.

The bandwidth criterion is

$$\text{bandwidth} \geq 3.5 \text{ dB}$$

Refer to prototype response poles table for Bessel<sup>[2]</sup>, the system is second order and  $\omega_0$  is 4 rad/s, so the closed-loop poles are chosen as follows.

$$\begin{bmatrix} p_1 \\ p_2 \end{bmatrix} = 4 \begin{bmatrix} -0.8660 + 0.5000i \\ -0.8660 - 0.5000i \end{bmatrix} \quad (2.9)$$

Use the same method as ITEA criterion, we can get the

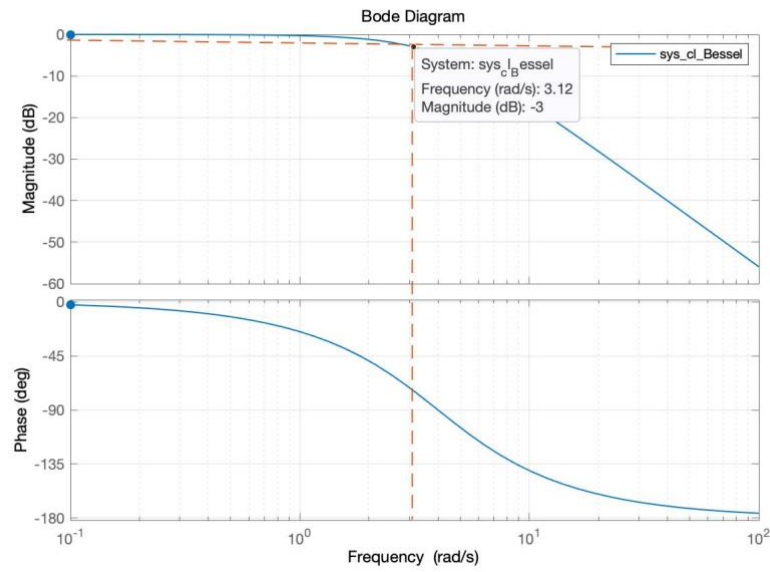
$$K_{Bessel} = [14.7993 \quad 3.2180] \quad K_{S_{Bessel}} = 15.9993 \quad (2.10)$$

Then the feedback controller can be obtained

$$u = -[14.7993 \quad 3.2180]x + 15.9993r \quad (2.11)$$

### 2.2.2 Closed-loop System Analysis

Plot the frequency response and step response for closed loop system.



**Figure 6** Frequency response for closed loop system using Bessel



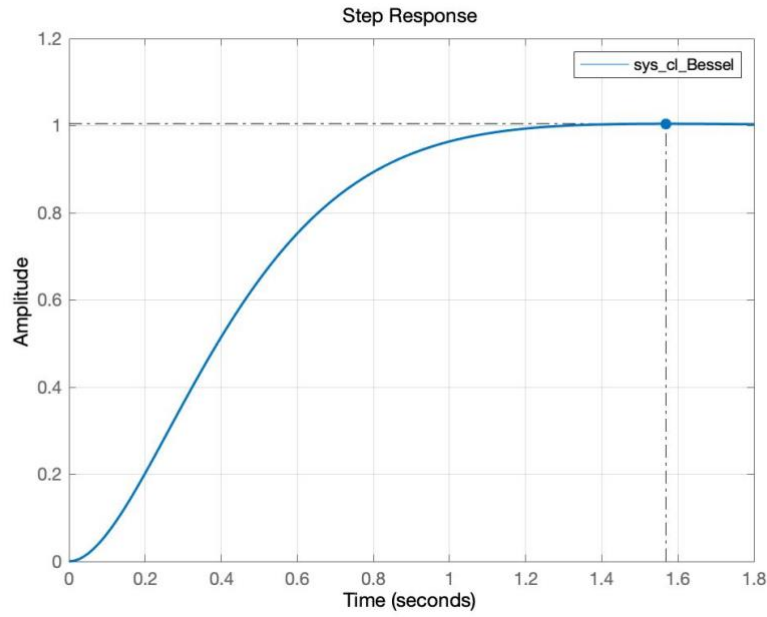


Figure 7 Step response for closed loop system using Bessel

From figure 6 above we can draw the -3dB line and get the bandwidth frequency is 3.12 rad/s, which is less than the requirement that the bandwidth  $\geq 3.5$  dB

And we can also get the resonant peak  $M_r$  is equal to 0.

From figure 7, we can easily read that the steady-state gain between  $r$  and  $y$  is 1, which means the input and output are same, so the steady-state gain is 0dB.

The system seems do not meet the design requirements.

## 2.3 Apply second-order dominant poles methodology

### 2.3.1 System Design

First of all, we need to design a second-order reference model for the closed-loop system.

$$H(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2} \quad (2.12)$$

As it often assumed that  $\zeta = 0.707$ .

Then choose  $\omega_n = 3.6$  and substitute them to property expressions.

$$\omega_b = \omega_n \left( (1 - 2\zeta^2) + \sqrt{(1 - 2\zeta^2)^2 + 1} \right)^{\frac{1}{2}} \quad (2.13)$$

$$\omega_r = \omega_n \sqrt{1 - 2\zeta^2}, \quad M_r = \frac{1}{2\zeta\sqrt{1 - \zeta^2}} \quad (2.14)$$

Then we can calculate  $\omega_b = 3.6005, M_r = 1.0000$ , which meets the design requirements.

We can also calculate the poles of system

$$P_1 = -2.5452 + 2.5460i, \quad p_2 = -2.5452 - 2.5460i \quad (2.15)$$

Still use Ackermann's Formula and MATLAB we can easily get the controller gain  $K_{SOD}$  and the scaling gain  $Ks_{SOD}$ .

$$K_{SOD} = [11.7600 \quad 1.3804], \quad Ks_{SOD} = 12.9600 \quad (2.16)$$

Then the feedback controller can be obtained

$$u = -[11.7600 \quad 1.3804]x + 12.9600r \quad (2.11)$$

### 2.3.2 Closed-loop System Analysis

Plot the frequency response and step response for closed loop system.

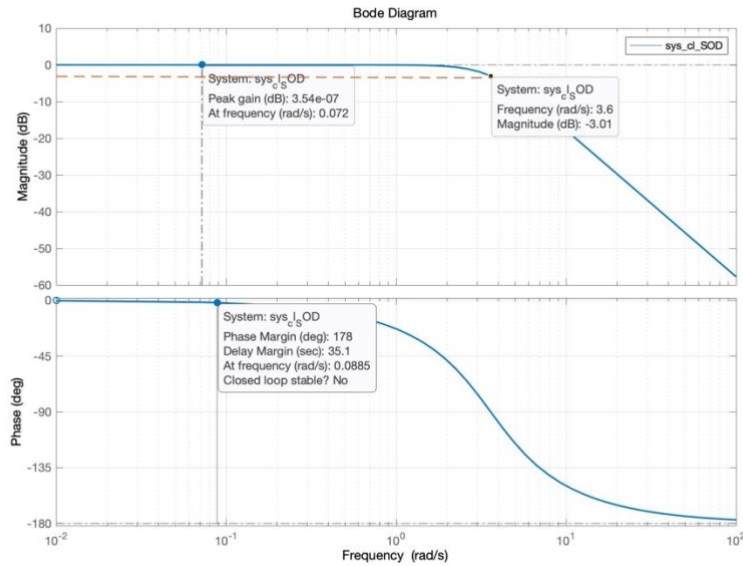
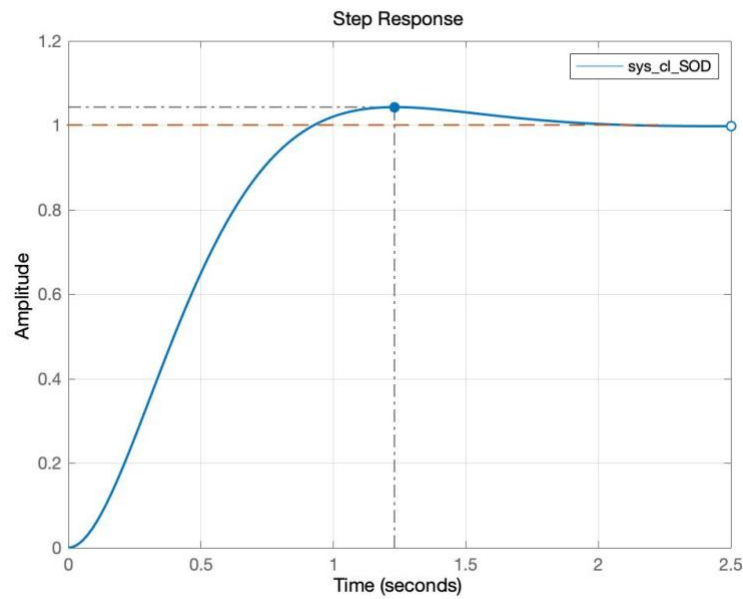


Figure 8 Frequency response for closed loop system



**Figure 9** Step response for closed loop system

From figure 8 above we can draw the -3dB line and get the bandwidth frequency is about 3.6 rad/s and we can also get the resonant peak  $M_r$  is equal to 0.

From figure 9, we can easily read that the steady-state gain between  $r$  and  $y$  is 1, which means the input and output are same, so the steady-state gain is 0dB.

The system meets the design requirements.

## 2.4 Part Conclusion

In a short, I think the second-order dominant poles methodology and ITAE criterion are much better methods among those three. And I prefer the second-order dominant poles methodology more, because we can design a system by designing its natural ratio and damping ratio and also meet the requirements precisely.

Besides, Bessel prototype methodology will have a lag in the bandwidth  $\omega_0$  of the real system to the designed one. For example, we assume  $\omega_0$  as 4 rad/s at the beginning but the real bandwidth is less, that 3.6 rad/s.

### 3. Design Using Linear Quadratic Regulator (LQR)

#### 3.1 System Design

The LQR approach minimizes the following cost function (or performance index) during the regulation.

$$J = \int_0^{\infty} (x^T Q x + u^T r u) \quad (3.1)$$

We fixed the value of  $r$ , and optimized the function by changing the value of  $Q$ . When there is only one variable,  $r$  becomes a scalar.

So, in the experiment Eq (3.1) can be stated as

$$J = \int_0^{\infty} (x^T Q x + r u^2) \quad (3.2)$$

To meet the design requirements, finally we choose

$$Q_1 = \begin{bmatrix} 500 & 0 \\ 0 & 0.01 \end{bmatrix}, r = 1 \quad (3.3)$$

Then the feedback controller gain  $K_{LQR}$  and the scaling gain  $K_{S_{LQR}}$  can be calculated.

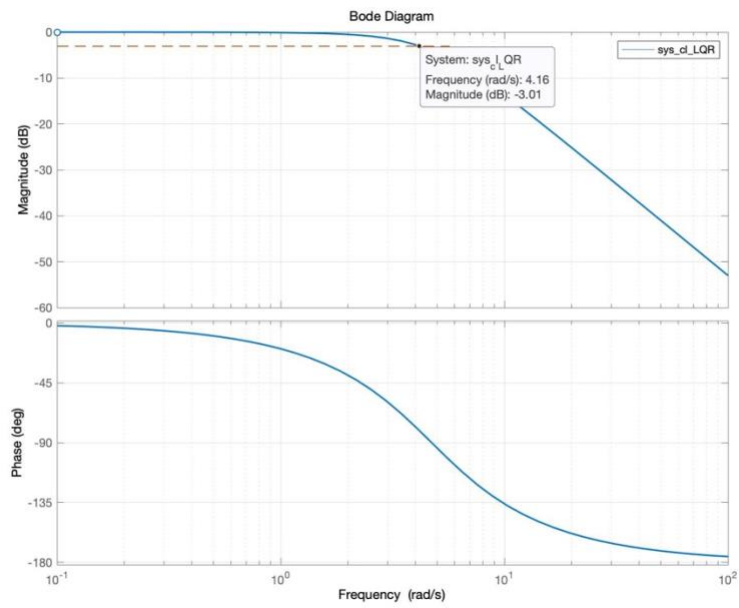
$$K_{LQR} = [21.1929 \quad 3.7840], \quad K_{S_{LQR}} = 22.3929 \quad (3.4)$$

Thus, the feedback controller can be obtained as

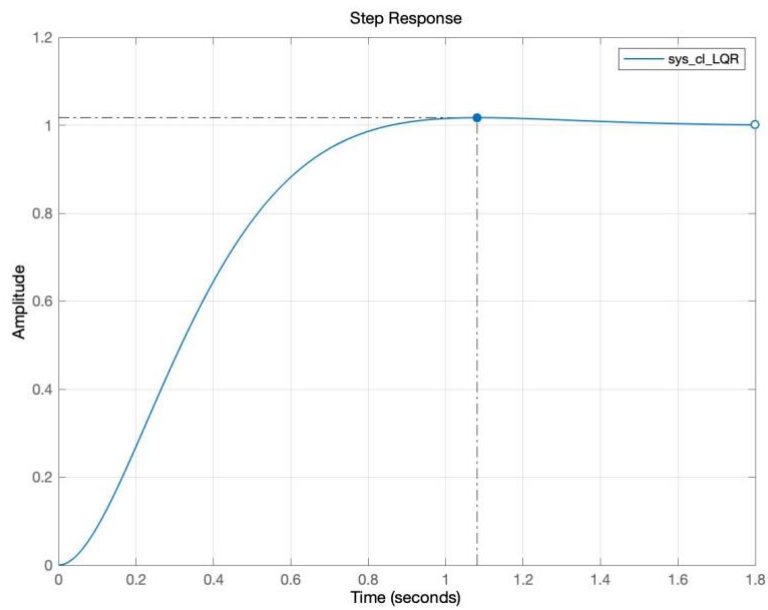
$$u = -[21.1929 \quad 3.7840]x + 22.3929r \quad (3.5)$$

#### 3.2 Closed-loop System Analysis

Plot the frequency response and step response for closed loop system.



**Figure 10** Frequency response for closed loop system



**Figure 11** Step response for closed loop system

From figure 10 above we can draw the -3dB line and get the bandwidth frequency is about 4.16 rad/s and we can also get the resonant peak  $M_r$  is equal to 0.

From figure 11, we can easily read that the steady-state gain between  $r$  and  $y$  is 1, which means the input and output are same, so the steady-state gain is 0dB.

The system meets the design requirements.

### 3.3 The effect of Matrix $Q$

Since  $Q = \text{diag}(q_1, q_2)$  with  $q_1, q_2, r > 0$ . We adjust these parameters to verify how the penalties setting would influence the closed-loop system performance.

Still let  $r = 1$  and we changed the coefficient related to  $x_1$ , and observe the effect on  $x_1$ . i.e.

$$Q_1 = \begin{bmatrix} i & 0 \\ 0 & 1 \end{bmatrix} \quad (3.6)$$

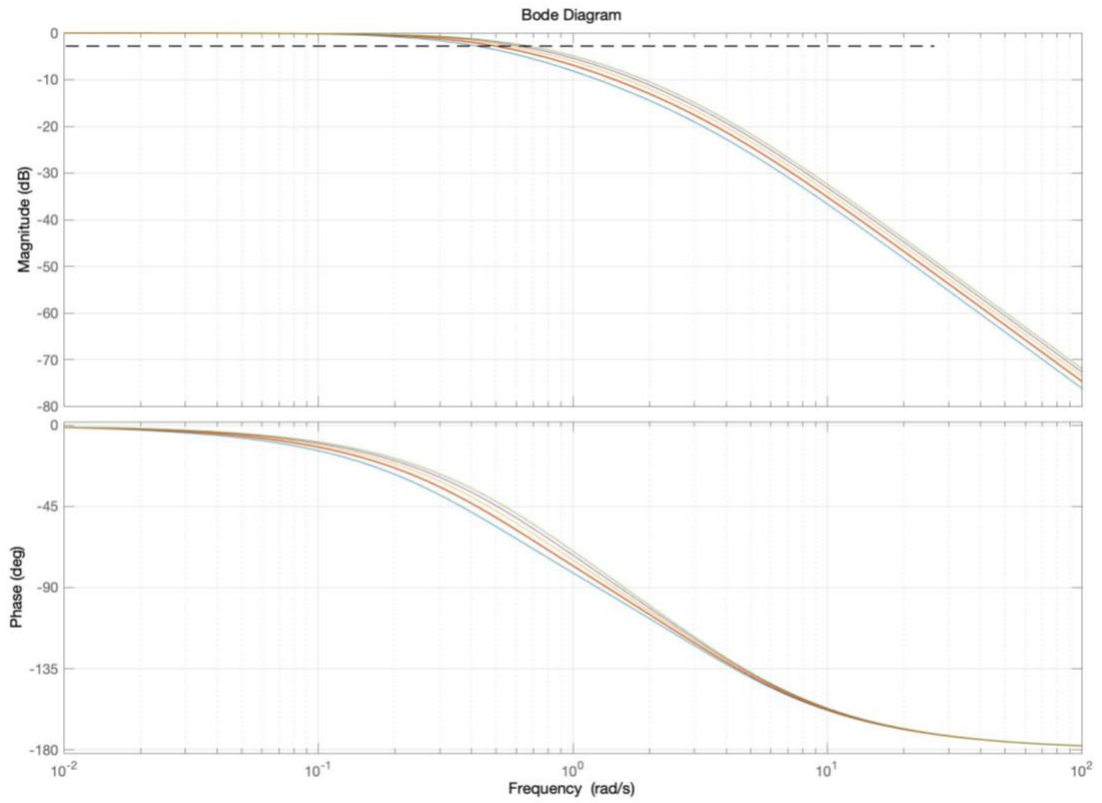
Change the value of  $i$  from 1 to 5.

We can get controller gain  $K_{LQR}$  and the scaling gain  $K_{S_{LQR}}$  of each controller.

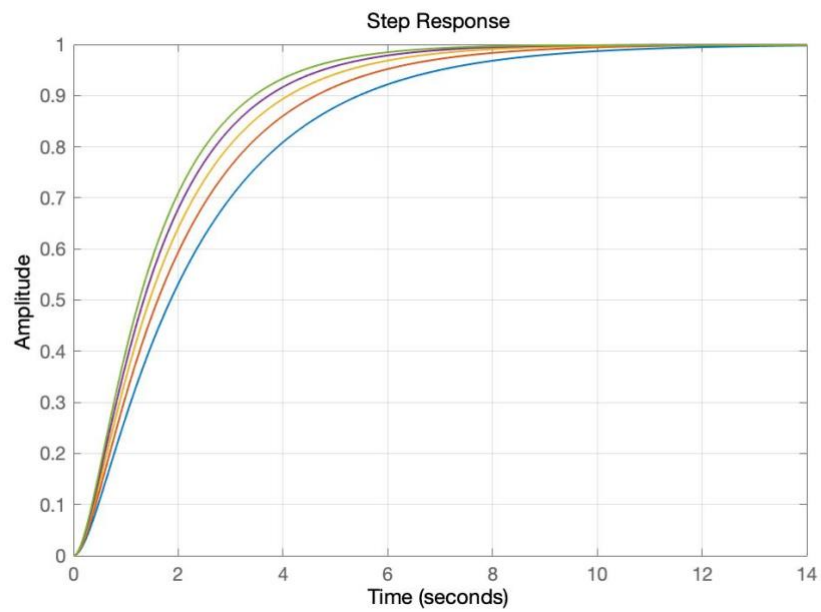
controller gain $K_{LQR}$	scaling gain $K_{S_{LQR}}$
[0.3620    0.2255]	1.5620
[0.6547    0.2992]	1.8547
[0.9071    0.3617]	2.1071
[1.1324    0.4166]	2.3324
[1.3377    0.4661]	2.5377

Table 2

Plot the Bode Diagram of each controller, we can get the figure below.



**Figure 12** Frequency response for closed loop system



**Figure 13** Step response for closed loop system

By increasing the related coefficient to  $x1$ , we can see in figure10 that the bandwidths of the closed loop system become larger, and the step response figures in figure11 move

up with  $i$  increasing, which means the control force on  $x_1$  is more and more strengthened.

Then, we changed the coefficient related to  $x_2$ , and observe the effect on  $x_1$  and  $x_2$ .

i.e.

$$Q_1 = \begin{bmatrix} 1 & 0 \\ 0 & i \end{bmatrix} \quad (3.7)$$

Change the value of  $i$  from 1 to 5. We can get controller gain  $K_{LQR}$  and the scaling gain  $Ks_{LQR}$  of each controller.

controller gain $K_{LQR}$	scaling gain $Ks_{LQR}$
[0.3620    0.2255]	1.5620
[0.3620    0.3506]	1.5620
[0.3620    0.4719]	1.5620
[0.3620    0.5898]	1.5620
[0.3620    0.7045]	1.5620

Table 3

Plot the Bode Diagram of each controller, we can get the figure below.

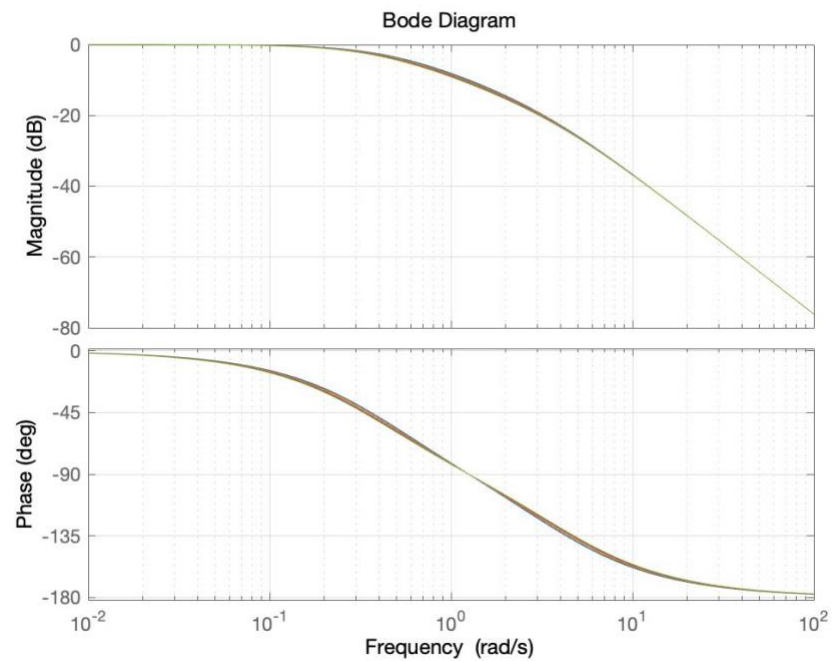
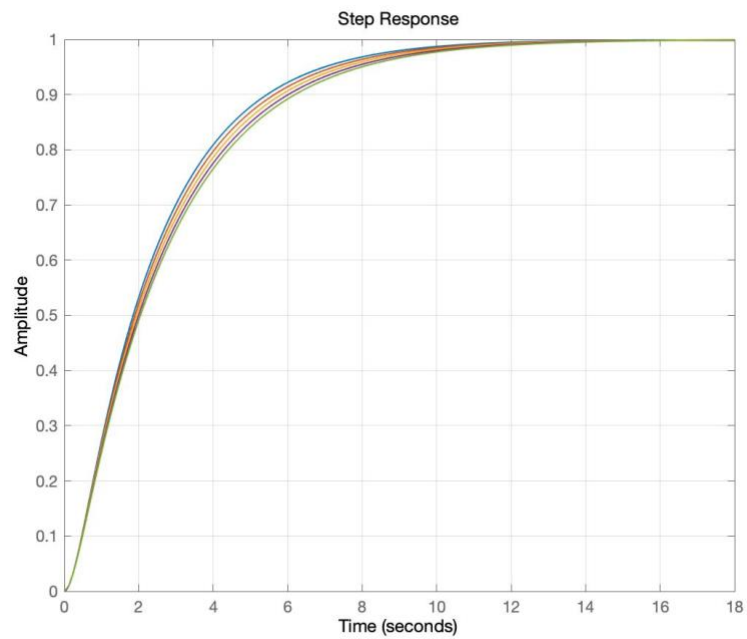


Figure 14 Frequency response for closed loop system





**Figure 15 Step response for closed loop system**

By increasing related coefficient to  $x_2$ , the frequency response figures in Figure 12 stay basically the same, and the step response figures in Figure 13 move down with  $i$  increasing, which means the control force on  $x_1$  is more and more weakened.

## Appendix 1

[1] ITAE criterion

[2] Bessel criterion

<b>TABLE 6.1</b>		
<b>Prototype Response Poles</b>		
(a) ITAE transfer functions	$k$	Pole locations for $\omega_0 = 1 \text{ rad/s}^\dagger$
	1	$s + 1$
	2	$s + 0.7071 \pm j0.7071^\ddagger$
	3	$(s + 0.7081)(s + 0.5210 \pm j1.068)$
	4	$(s + 0.4240 \pm j1.2630)(s + 0.6260 \pm j0.4141)$
	5	$(s + 0.8955)(s + 0.3764 \pm j1.2920)(s + 0.5758 \pm j0.5339)$
	6	$(s + 0.3099 \pm j1.2634)(s + 0.5805 \pm j0.7828)(s + 0.7346 \pm j0.2873)$
(b) Bessel transfer functions	$k$	Pole locations for $\omega_0 = 1 \text{ rad/s}^\dagger$
	1	$s + 1$
	2	$s + 0.8660 \pm j0.5000)^\ddagger$
	3	$(s + 0.9420)(s + 0.7455 \pm j0.7112)$
	4	$(s + 0.6573 \pm j0.8302)(s + 0.9047 \pm j0.2711)$
	5	$(s + 0.9264)(s + 0.5906 \pm j0.9072)(s + 0.8516 \pm j0.4427)$
	6	$(s + 0.5385 \pm j0.9617)(s + 0.7998 \pm j0.5622)(s + 0.9093 \pm j0.1856)$

<sup>†</sup>Pole locations for other values of  $\omega_0$  can be obtained by substituting  $s/\omega_0$  for  $s$  everywhere.

<sup>‡</sup>The factors  $(s + a + jb)(s + a - jb)$  are written as  $(s + a \pm jb)$  to conserve space.

## Appendix 2 MATLAB Part Code

### 1. Mr, wr and bandwidth(-3dB) frequency calculate.

```
function [ Mr,Wr,Wr_3db] = mr(S_close)
% The closed-loop resonant amplitude Mr And resonant frequency Wr of the system
[mag,phase,w] = bode(S_close);
c = size(mag,3);
mag1 = zeros(c,1);
for i=1:c
    mag1(i) = 20*log10(mag(1,1,i));%Get the amplitude of each point
end
[M,i] = max(mag1);% Get the maximum amplitude and corresponding frequency
%show resonant amplitude Mr And resonant frequency Wr
Mr = M;
Wr = w(i);
%show the -3dB cutoff frequency
Wr_3db = interp1(mag1,w,-3,'Newton');%Interpolation
c = size(phase,3);
phal = zeros(c,1);
for i=1:c
    phal(i) = phase(1,1,i);%Get the frequency of each point
end
end
```

### 2. Open-loop system analysis

```
F_Plant = [0,1;-1.20,-3.71]; % Process Matrix
G_Plant = [0;1]; % Input Matrix
H_Plant_State1 = [1,0]; % Output Matrix
H_Plant_State2 = [0,1]; % Output Matrix Assuming x2 is the Output
H_Plant_FullState = [1,0;0,1]; % Output Matrix Assuming Both States are Outputs
sys1 = ss(F_Plant,G_Plant,H_Plant_State1,0);%function 'ss' to establish state-space
model
sys2 = ss(F_Plant,G_Plant,H_Plant_State2,0);
figure(1) % Open a new figure window named 'figure 1'
bode(sys1) % Bode plot from u to x1; Use function 'bode' to plot bode graph
grid on
figure(2)
bode(sys2) % Bode plot from u to x2
grid on
Poles = eig(F_Plant); % The open-loop poles are the eigenvalue of the original plant
process matrix
% Use function 'eig' to calculate the eigenvalue of a square matrix
disp('The open-loop system transfer function poles are located at:')
display(Poles)
```

```

ControllabilityMatrix = [G_Plant,F_Plant*G_Plant];
if (det(ControllabilityMatrix)==0)
    error('The original plant is not controllable.')
% An 'error' command will stop the MATLAB program and pop-up an error message
% If the controllability matrix has determinant of zero, the plant is uncontrollable.
else
    disp('The original plant is controllable.')
end

```

### 3. Sample system

```

sys_fullstateplant = ss(F_Plant,G_Plant,H_Plant_FullState,[0;0]);
% The original plant with full state vector as output
K_SimpleExp = [0.1,0.1];
Signal_KX_SimpleExp = ss(0,[0,0],0,K_SimpleExp); % Dummy system representing the value
of K*x
sys_fb_SimpleExp = feedback(sys_fullstateplant,Signal_KX_SimpleExp,-1); % Form a
feedback system.
% Use fcn 'feedback' to establish a closed-loop system with a feedback controller
% Note that this system is 1-input-2-output system. The input is r (with
% Ks=1 by default) and the outputs are x1 and x2.
figure(3)
bode(sys_fb_SimpleExp) % Bode plot from r to x1 and x2
F_ClosedLoop = F_Plant-G_Plant*K_SimpleExp;
Poles = eig(F_ClosedLoop); % The closed-loop poles are the eigenvalue of the closed-loop
process matrix
display(Poles)

```

### 4. ITEA

```

ITAE_Poles = 4*[-0.7071+0.7071*i;-0.7071-0.7071*i]; % Poles as a column vector
K_ITAE = acker(F_Plant,G_Plant,ITAE_Poles)
% Use function 'acker' to calculate the controller gain using Ackermann's Formula
Signal_KX_ITAE = ss(0,[0,0],0,K_ITAE)
sys_fb_ITAE = feedback(sys_fullstateplant,Signal_KX_ITAE,-1)
Ks_ITAE = 1/dcgain(sys_fb_ITAE) % Calculate the steady state gain of the feedback system
% Use function 'dcgain' to calculate the steady state gain of a system
sys_cl_ITAE = Ks_ITAE*sys_fb_ITAE;
% The feedback system together with the scaling gain Ks forms the final design of closed-
loop system
% Note that sys_cl_ITAE is SINGLE output system with output y=x1

```

## 6. Bessel

```
% Similarly for Bessel Prototype
BesselPoles = 4*[-0.8660+0.5000*1i;-0.8660-0.5000*1i];
K_Bessel = acker(F_Plant,G_Plant,BesselPoles)
Signal_KX_Bessel = ss(0,[0,0],0,K_Bessel);
sys_fb_Bessel = feedback(sys_fullstateplant,Signal_KX_Bessel,-1);
Ks_Bessel = 1/dcgain(sys_fb_Bessel)
sys_cl_Bessel = Ks_Bessel*sys_fb_Bessel;
```

## 7. Second order

```
DampRatio = 0.707; % Define Damping Ratio for Reference Model
NatruaFreq = 3.6; % Define Natural Frequency for Reference Model
BandWidthDesire = NatruaFreq*sqrt((1-2*DampRatio^2)+sqrt((1-2*DampRatio^2)^2+1))
% Calculate the Bandwidth of the reference model
SODPoles = roots([1,2*DampRatio*NatruaFreq,NatruaFreq^2])
% Calculate the pole positions of the reference model, the root of pole polynomial
% Use function 'roots' to calculate the roots of a polynomial
K_SOD = acker(F_Plant,G_Plant,SODPoles);
Signal_KX_SOD = ss(0,[0,0],0,K_SOD);
sys_fb_SOD = feedback(sys_fullstateplant,Signal_KX_SOD,-1);
Ks_SOD = 1/dcgain(sys_fb_SOD);
sys_cl_SOD = Ks_SOD*sys_fb_SOD;
```

## 8. LQR(Loop part)

```
% Design Using LQR
for i = 1:5
    Q_LQR = [1,0;0,i];
    R_LQR = 1; % Define penalty matrix R (in this case, a scalar)
    [K_LQR,~,~] = lqr(F_Plant,G_Plant,Q_LQR,R_LQR)
    Signal_KX_LQR = ss(0,[0,0],0,K_LQR);
    sys_fb_LQR = feedback(sys_fullstateplant,Signal_KX_LQR,-1);
    Ks_LQR = 1/dcgain(sys_fb_LQR)
    sys_cl_LQR = Ks_LQR*sys_fb_LQR;
    figure(1)
    bode(sys_cl_LQR)
    hold on
    grid on
    figure(2)
    step(sys_cl_LQR)
    grid on
    hold on
end
```