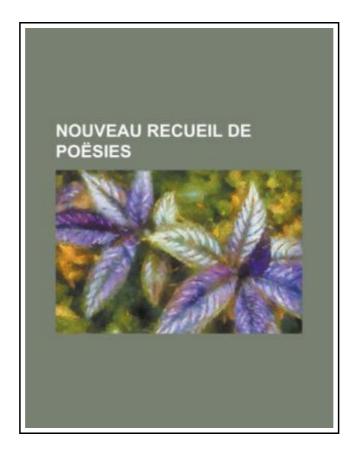
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RareBooksClub. Paperback. Book Condition: New. This item is printed on demand. Paperback. 80 pages. Original publisher: Fort Collins, CO: Colorado State University, Department of Mechanical Engineering; Springfield, Va.: For sale by the National Technical Information Service, 1985 OCLC Number: (OCoLC)708253122 Excerpt: . . . region on the grid surface where ion beamlet deflections exceed the useful range. Over this region high accelerator grid impingement currents and rapid grid erosion are predicted. THEANNULARFLOWELECTROTHERMRAALMJET Ben D. Shaw Theoretical analysis of the annular flow, electrothermal, plug ramjet was completed and the results are described in Ref. 3 so the results of this work will also simply be summarized here. In Ref. 3 the ramjet is examined as a possible means of achieving rapid projectile acceleration to velocities for such applications as direct launch of spacebound payloads. The perform-ance of this ramjet device operating with hydrogen propellant is examined for cases where this working fluid is treated (I) as a perfect gas and (2) as o a gas that is allowed to dissociate and ionize and then recombine with finite reaction rates in the nozzle. Performance results For these cases are compared to the performance of a conventional ramjet operating with perfect gas hydrogen propellant. It is shown that the performance of the conventional ramjet is superior to that of the annular flow, electrothermal ramjet. However, it is argued that the mechanical complexities associated with the conventional ramjet would make it impractical, and for this reason the annular flow, electro-thermal ramjet is more desirable as a launch system. Models are presented which describe both electrothermalplug ramjet and conventional ramjet opera-tion, and it is shown that for a given flight velocity there is a rate of heat addition per unit propellant mass for which ramjet operation is opti-mized....



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