

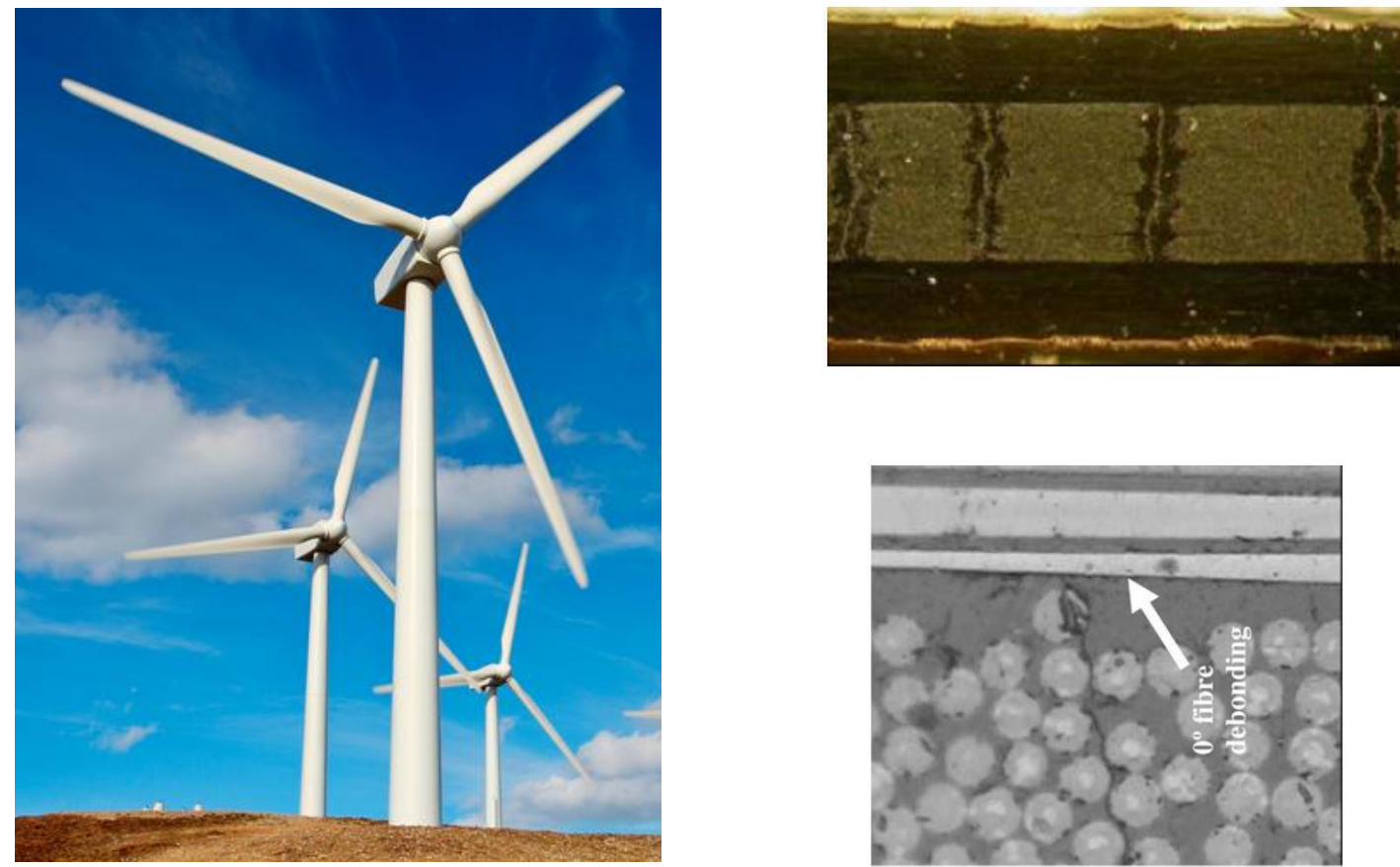
Damage Evolution in Cross-Ply Laminates Revisited via CZM-Based Finite-Volume Homogenization

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Introduction:

Q: Damage types in composite laminates?

A: Matrix cracking, fiber fracture, fiber/matrix debonding and delamination.



Damage simulation approaches based on FEA:

- 1) Re-meshing of analysis domain
- 2) Cohesive Zone Model (CZM)

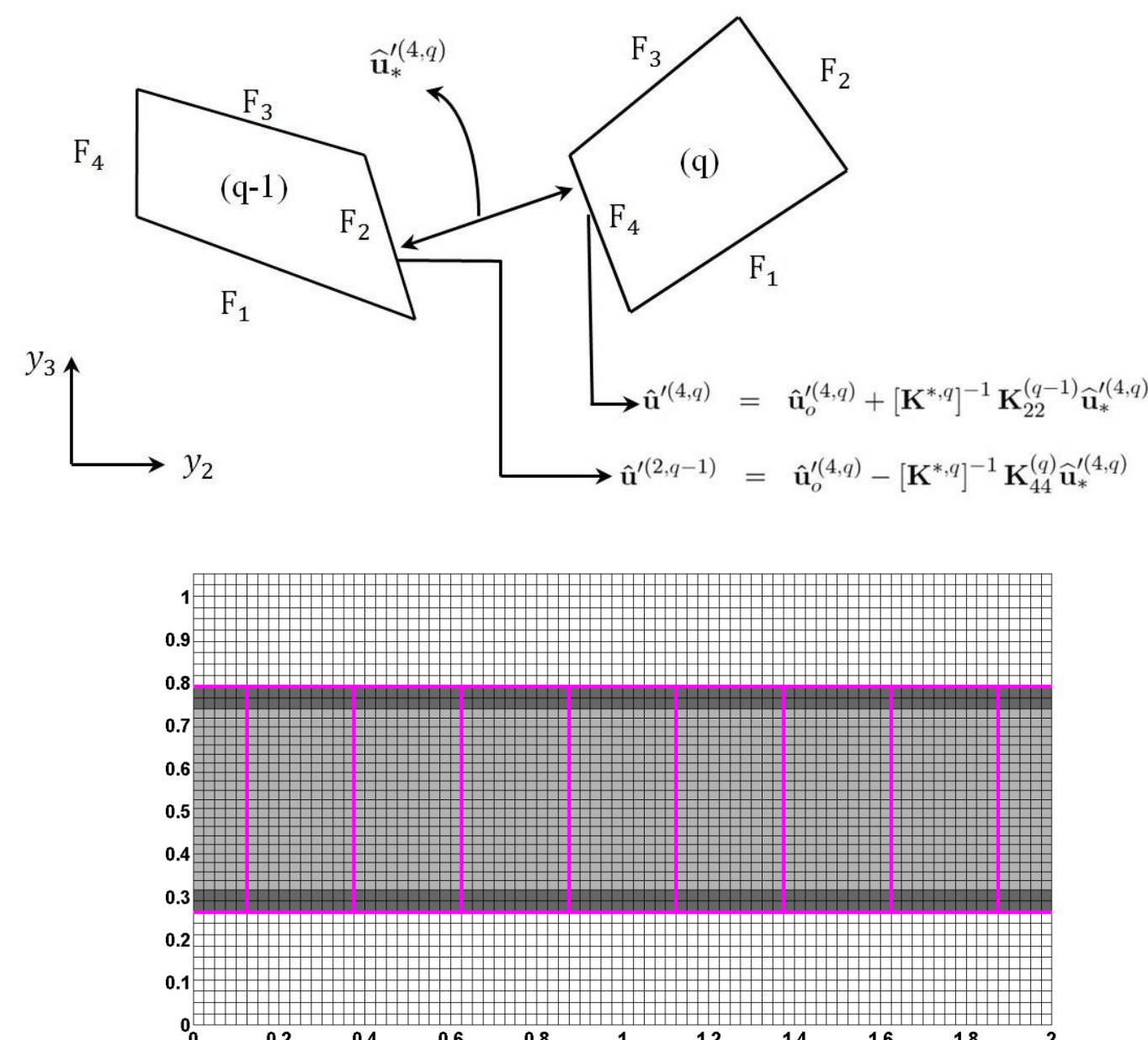
Challenge in heterogeneous materials:

large modulus mismatch produces large stress gradient, especially in the presence of cracks.

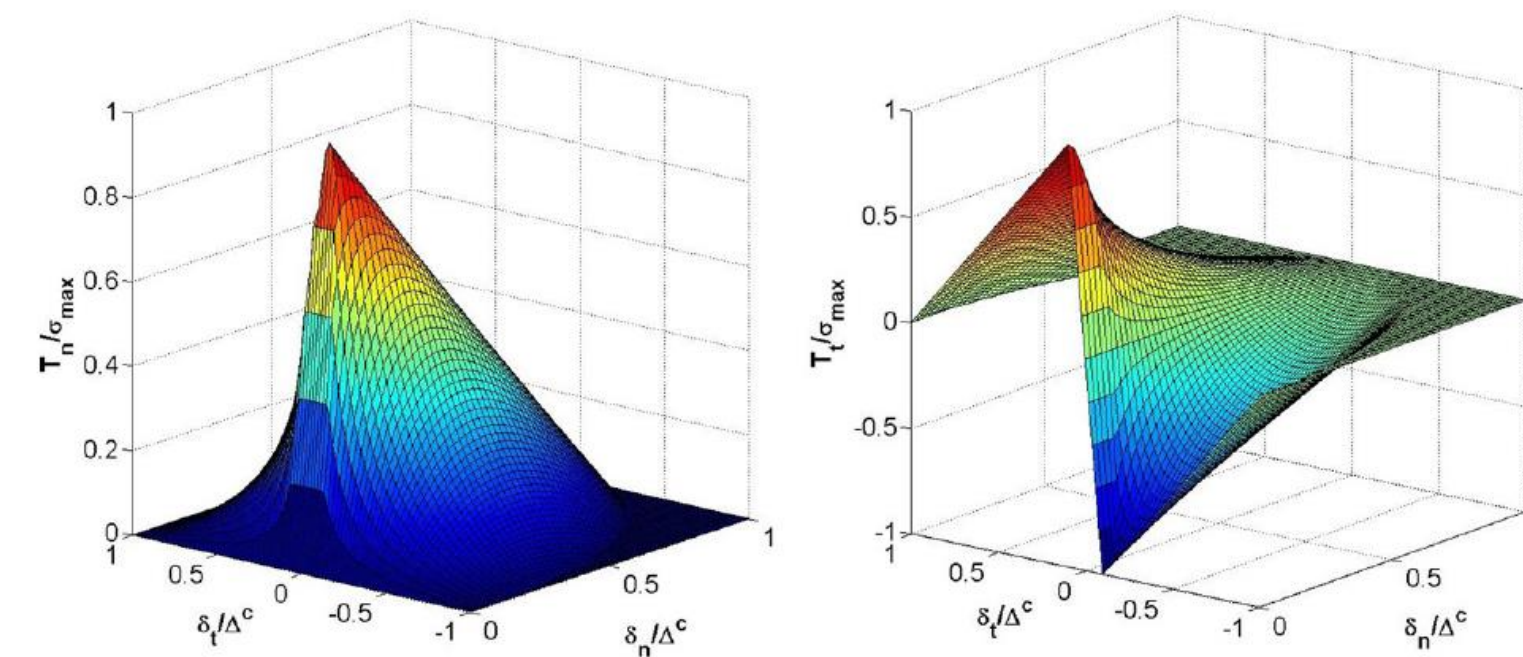
Finite Volume Direct Averaging Micromechanics (FVDAM) → an attractive alternative to FEA.

CZM-Based Parametric FVDAM Theory:

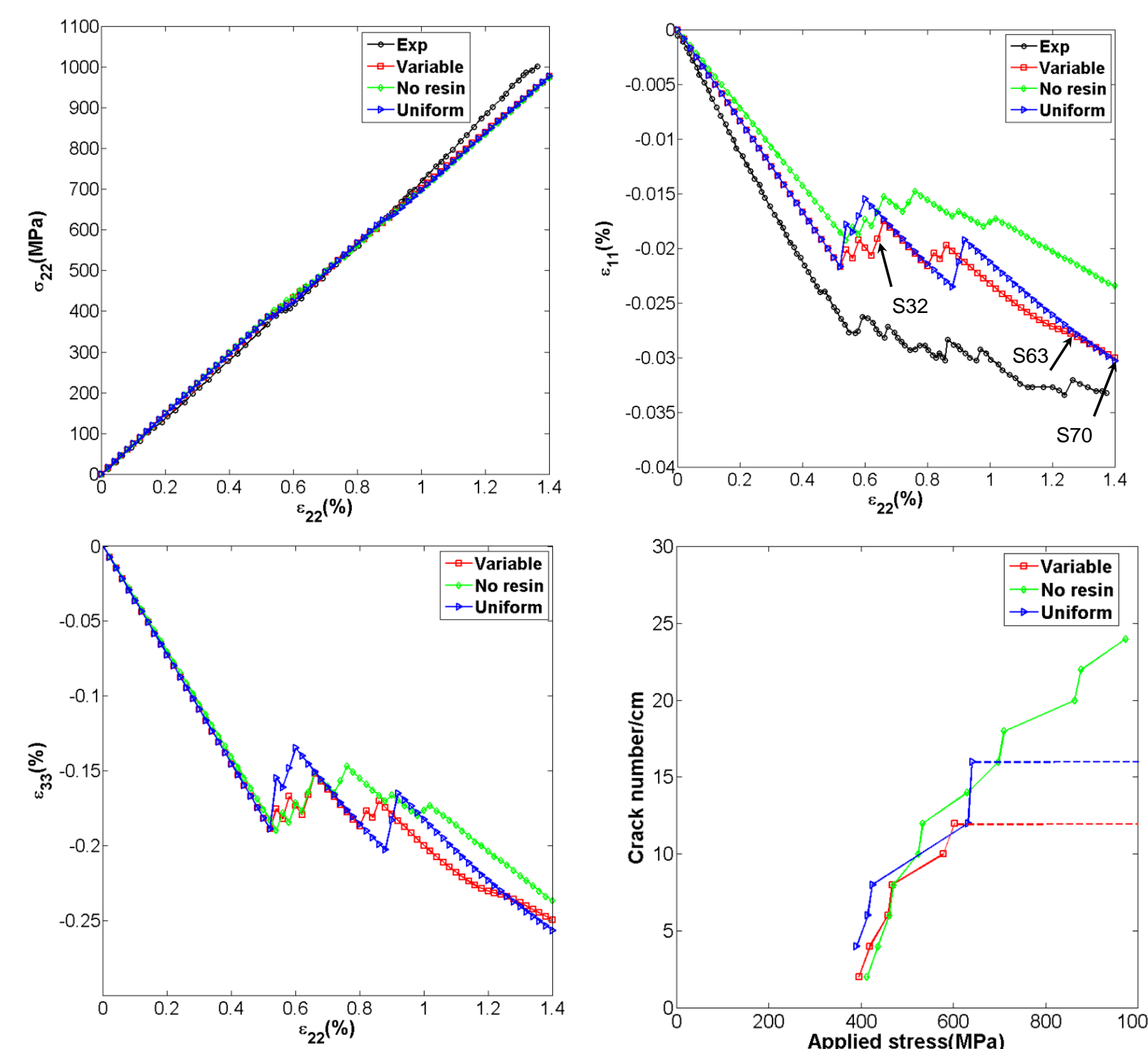
Parametric mapping and discontinuity:



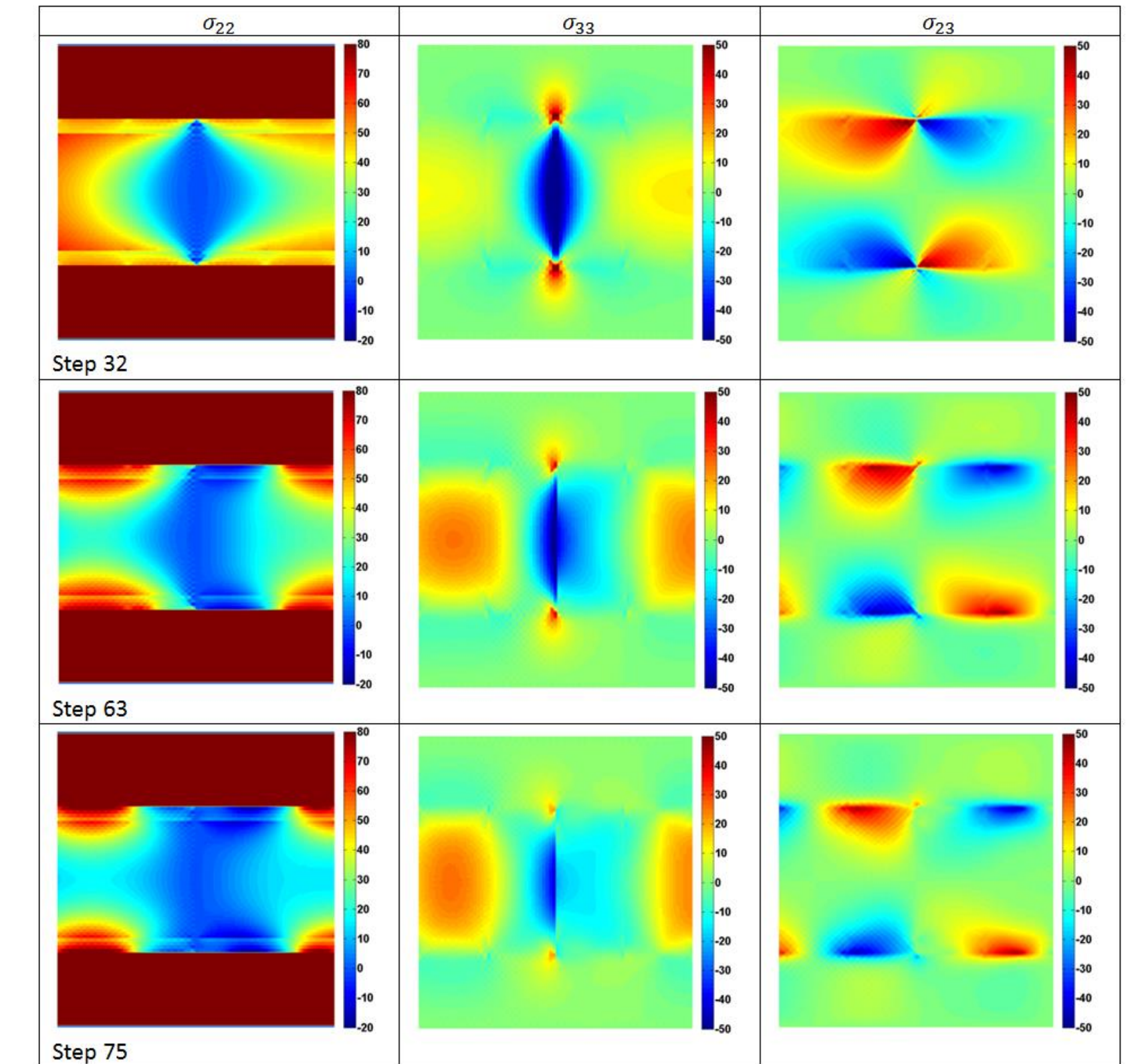
- Displacement field : $u_i^{(q)}(\mathbf{x}, \mathbf{y}(\eta, \xi)) = \bar{\varepsilon}_{ij}x_j + u_i^{(q)}(\eta, \xi)$
- Fluctuating displacement: $u_i^{(q)} = W_{i(00)}^{(q)} + \eta W_{i(10)}^{(q)} + \xi W_{i(01)}^{(q)} + \frac{1}{2}(3\eta^2 - 1)W_{i(20)}^{(q)} + \frac{1}{2}(3\xi^2 - 1)W_{i(02)}^{(q)}$
- Primary global system of equations: $\mathbb{K}\hat{\mathbf{U}}'_o = \Delta\mathbf{C}\bar{\boldsymbol{\varepsilon}} + \Delta\mathbf{\Gamma}\Delta T - \mathbf{L}\hat{\mathbf{U}}'_*$
- Auxiliary system of equations: $\mathbf{T} = \mathbf{K}^*\hat{\mathbf{U}}'_* + \mathbf{K}_o\hat{\mathbf{U}}'_o + \mathbf{C}\bar{\boldsymbol{\varepsilon}} + \mathbf{\Gamma}\Delta T$
- Homogenized Hooke's Law: $\bar{\boldsymbol{\sigma}} = \frac{1}{V} \int \boldsymbol{\sigma}(x) dV = \sum_{q=1}^{N_q} v_{(q)} \bar{\boldsymbol{\sigma}}^{(q)} = \mathbf{C}^* \bar{\boldsymbol{\varepsilon}} - (\bar{\boldsymbol{\sigma}}^{th} + \bar{\boldsymbol{\sigma}}^d)$
- Cohesive Law



Results: Homogenized response



Stress field with evolving damage:



Conclusions:

- The newly developed framework offers a unified methodology for simulating damage in cross-ply composite laminates due to cracking or progressive interfacial degradation.
- Transverse and through-thickness Poisson's ratios are sensitive to different damage modes, which can serve as excellent damage detection indicators.

Future Work:

- Incorporation of out of plane loading capability to simulate the dissipative response of unidirectional composites with two brittle constituents.

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References

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- 2) F. Paris et al., Characterization and evolution of matrix and interface related damage in [0/90]S laminates under tension. Part I: Numerical predictions, Composites Science and Technology, 2010, 70: 1168–1175.