REPORT

25.10.02 목 과제

CHEONGJU UNIVERSITY



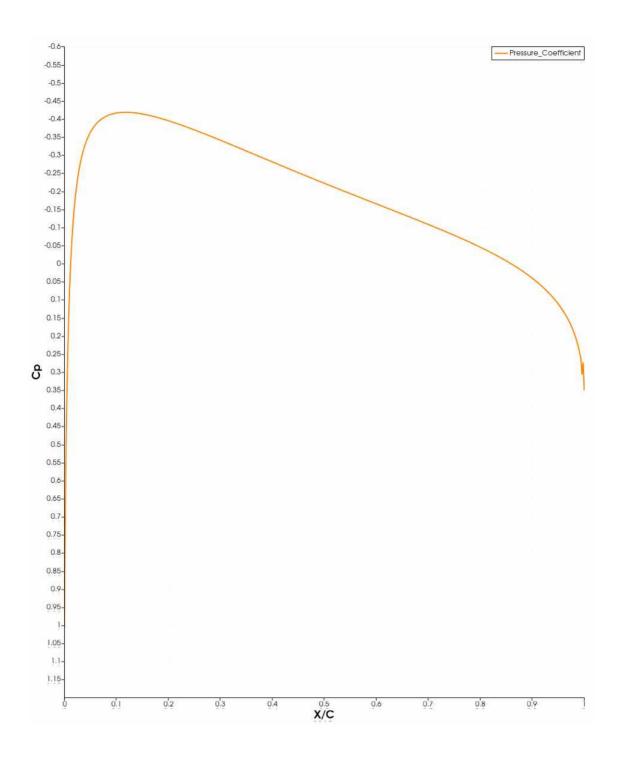
청주대학교

과 목 명 :	전산유체해석실습	
지도교수 :	임동균	
학 과:	무인항공기학과	
학 번:	2021010599	
이 름:	박원빈	
제 출 일 :	2025년10월3일	

```
------ DIRECT, ADJOINT, AND LINEARIZED PROBLEM DEFINITION -------%
% Physical governing equations (EULER, NAVIER STOKES,
                             WAVE_EQUATION, HEAT_EQUATION, FEM_ELASTICITY,
                             POISSON EQUATION, RANS)
SOLVER= RANS
% Specify turbulence model (NONE, SA, SA_NEG, SST)
KIND TURB MODEL= SA
% Mathematical problem (DIRECT, CONTINUOUS_ADJOINT)
MATH_PROBLEM= DIRECT
% Restart solution (NO, YES)
RESTART_SOL= NO
% Mach number (non-dimensional, based on the free-stream values)
MACH NUMBER= 0.150
% Angle of attack (degrees, only for compressible flows)
AOA= 0
% Side-slip angle (degrees, only for compressible flows)
SIDESLIP ANGLE= 0.0
% Init option to choose between Reynolds (default) or thermodynamics quantities
% for initializing the solution (REYNOLDS, TD_CONDITIONS)
INIT_OPTION= REYNOLDS
% Free-stream option to choose between density and temperature (default) for
% initializing the solution (TEMPERATURE_FS, DENSITY_FS)
FREESTREAM OPTION= TEMPERATURE FS
% Free-stream temperature (288.15 K by default)
FREESTREAM_TEMPERATURE= 288.15
% Reynolds number (non-dimensional, based on the free-stream values)
REYNOLDS NUMBER= 6.0E6
% Reynolds length (1 m by default)
REYNOLDS_LENGTH= 1.0
```

-RANS 방정식을 풀도록 지정

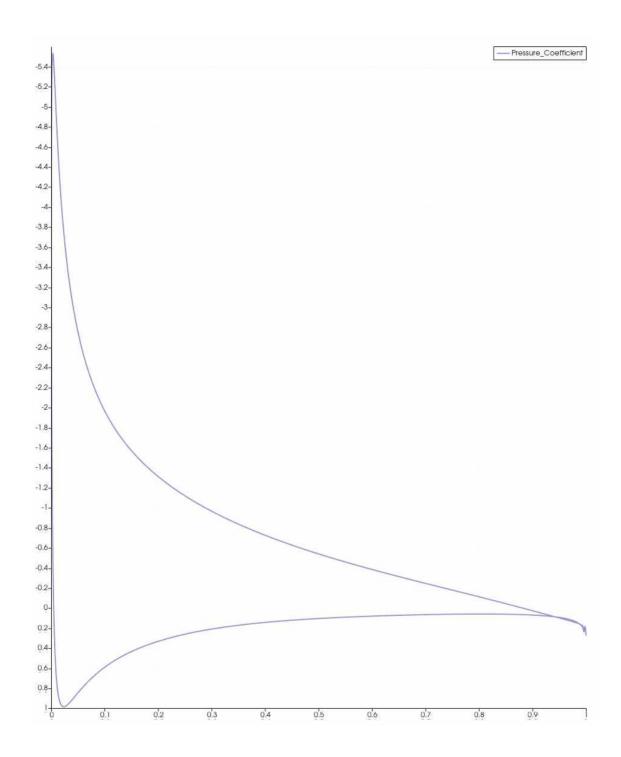
-마하수 0.15, 받음각 0°, 레이놀즈수 6.0*10⁶, 시위선 1.0m



```
% ------ DIRECT, ADJOINT, AND LINEARIZED PROBLEM DEFINITION ------%
% Physical governing equations (EULER, NAVIER_STOKES,
                              WAVE_EQUATION, HEAT_EQUATION, FEM_ELASTICITY,
%
%
                               POISSON EQUATION, RANS)
SOLVER= RANS
% Specify turbulence model (NONE, SA, SA NEG, SST)
KIND TURB MODEL= SA
% Mathematical problem (DIRECT, CONTINUOUS_ADJOINT)
MATH PROBLEM= DIRECT
% Restart solution (NO, YES)
RESTART SOL= NO
% ------ COMPRESSIBLE FREE-STREAM DEFINITION ------%
% Mach number (non-dimensional, based on the free-stream values)
MACH NUMBER= 0.150
% Angle of attack (degrees, only for compressible flows)
% Side-slip angle (degrees, only for compressible flows)
SIDESLIP ANGLE= 0.0
% Init option to choose between Reynolds (default) or thermodynamics quantities
% for initializing the solution (REYNOLDS, TD CONDITIONS)
INIT_OPTION= REYNOLDS
% Free-stream option to choose between density and temperature (default) for
% initializing the solution (TEMPERATURE_FS, DENSITY_FS)
FREESTREAM OPTION= TEMPERATURE FS
% Free-stream temperature (288.15 K by default)
FREESTREAM TEMPERATURE= 288.15
% Reynolds number (non-dimensional, based on the free-stream values)
REYNOLDS NUMBER= 6.0E6
%
% Reynolds length (1 m by default)
REYNOLDS_LENGTH= 1.0
```

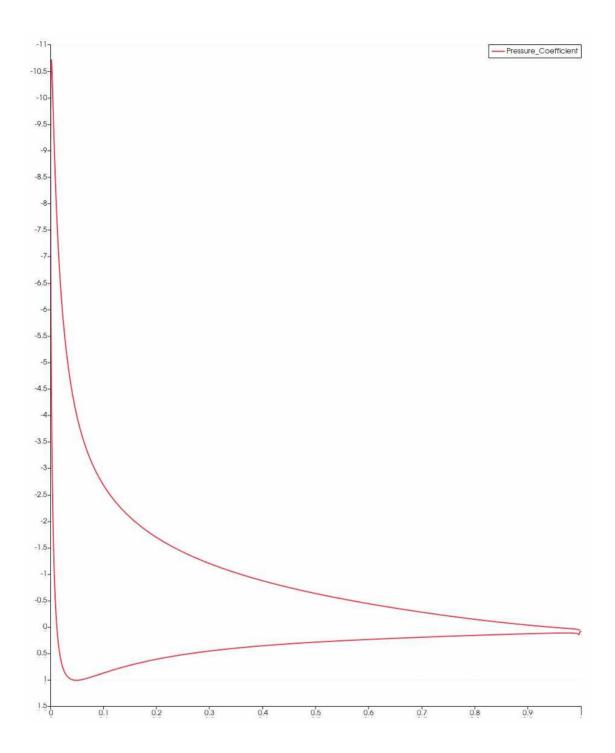
-받음각 10°, 나머지 조건은 0°일 때와 동일

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% Physical governing equations (EULER, NAVIER_STOKES,
                             WAVE_EQUATION, HEAT_EQUATION, FEM_ELASTICITY,
                             POISSON_EQUATION, RANS)
SOLVER= RANS
% Specify turbulence model (NONE, SA, SA_NEG, SST)
KIND TURB MODEL= SA
% Mathematical problem (DIRECT, CONTINUOUS_ADJOINT)
MATH_PROBLEM= DIRECT
% Restart solution (NO, YES)
RESTART SOL= NO
% ------ COMPRESSIBLE FREE-STREAM DEFINITION ------%
%
% Mach number (non-dimensional, based on the free-stream values)
MACH NUMBER= 0.150
% Angle of attack (degrees, only for compressible flows)
AOA= 15
% Side-slip angle (degrees, only for compressible flows)
SIDESLIP_ANGLE= 0.0
%
% Init option to choose between Reynolds (default) or thermodynamics quantities
% for initializing the solution (REYNOLDS, TD_CONDITIONS)
INIT OPTION= REYNOLDS
% Free-stream option to choose between density and temperature (default) for
% initializing the solution (TEMPERATURE_FS, DENSITY FS)
FREESTREAM OPTION= TEMPERATURE FS
% Free-stream temperature (288.15 K by default)
FREESTREAM_TEMPERATURE= 288.15
% Reynolds number (non-dimensional, based on the free-stream values)
REYNOLDS NUMBER= 6.0E6
%
% Reynolds length (1 m by default)
REYNOLDS LENGTH= 1.0
```

-받음각 15°, 나머지 조건은 위와 동일



받음각 (α)	0°	10°	15°	
특징 대칭 분포		상ㆍ하부 압력	상ㆍ하부 압력 최대	
	내성 군포	뚜렷하게 분리	분리	
CL	≈ 0	높은 양력 발생	C _{Lmax} 근접	
날개 윗면 최저 C _p	≈ -0.45	≈ -5	≈ -11	
	-받음각이 0°일 때는 상하 분포가 대칭적이라 양력이 거의 없다			
	-10°에서는 윗면 압력이 크게 낮아지고 아랫면 압력이 높아져 CL이			
Cp 분포변화	크게 증가.			
	-15°에서는 윗면 압력이 매우 낮아지지만 뒷전 유동이 불안정해져 박			
	리가 발생하고 CL이 감소하면서 실속 징후가 나타남.			