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Influence of the alloying elements Re, Co and W on the propagation of the Ni/Ni₃Al interface crack

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The influence of the alloying elements Re, Co and W on the propagation of the (010)[101] Ni/Ni₃Al interface crack has been investigated by molecular dynamics simulations and the discrete-variational method. The simulation results show that the interface crack propagates in a brittle manner at low temperature (5 K), but in a ductile manner at high temperature (1273 K), both with and without the addition of alloying elements. Owing to the scientific and technological importance of superalloys, the effects of the chemical bonding behavior between the alloying element X (X = Re, Co, or W) and Ni atoms on the crack shape, crack propagation velocity, and dislocation emission were investigated. At low temperature, the alloying elements Re and W inhibit the propagation of the Ni/Ni₃Al interface crack, while at high temperature Re, Co and W can improve the ductility of Ni-based single-crystal superalloys. Furthermore, the adhesion work of the interface, surface energy, and unstable stacking energy were calculated to understand the propagation mechanism of the Ni/Ni₃Al interface crack because of alloying element addition. The results of this study may provide useful information for the design of Ni-based superalloys.

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1. Introduction

Because of their outstanding high-temperature mechanical properties, Ni-based single-crystal (SC) superalloys are important high-temperature materials, and they are used in turbine blades of advanced aircraft engines.^{1,2} The typical microstructure of these alloys is composed of the ordered L1₂ γ'-Ni₃Al phase coherently embedded in a solid solution matrix of the γ-Ni phase. Because of the chemical composition and structural character, Ni-based SC superalloys are much stronger than pure γ or γ' single-phase materials at high temperature.^{3–5} The properties of the γ/γ' interface have a great influence on the performance of Ni-based superalloys. Thus, it is important to study the γ-Ni/γ'-Ni₃Al interface.

The strength of the γ/γ' interface can be significantly improved by the addition of alloying elements, such as Re, Ta, W, Mo, Ru, and Co. In the past few years, the Ni/Ni₃Al interface has been investigated using density functional theory. Chen *et al.*⁶ investigated the strengthening effect of alloying elements on the Ni/Ni₃Al interface by calculating the bond orders, and showed that the rupture strength of Ni-based SC superalloys can be improved by Re, Ta, W, and Mo. Other studies^{7–11} have found that the binding strength, ductility, rupture strength, and toughness of the γ/γ' interface can be improved by the addition of the refractory

elements Re and Ru. Moreover, it has been reported that Co has little effect on improving the strength of the γ/γ' interface.⁸ Recently, Liu *et al.*¹² investigated the influence of Re on the propagation of the Ni/Ni₃Al interface crack by the molecular dynamics (MD) method, and they found that Re is able to inhibit the crack propagation at low temperature and improve the ductility of superalloys because of the strong bond strength of Ni–Re.

In general, the final fracture of materials is caused by the breaking of atomic bonds, and the rupture strength can be improved by forming stronger atomic bonds during crack propagation. Improvement of the atomic bonds can be achieved by certain alloying elements. At present, in view of the construction of Ni–Al–X (X = Re, Co, or W) embedded-atom method (EAM) potentials^{13–15} and the important influence of Re, Co, and W on the mechanical properties of superalloys,^{16–19} it is essential to investigate the influence of these three alloying elements on the propagation of the Ni/Ni₃Al interface crack at the atomic scale.

In the present study, 1 or 2 at% X (X = Re, Co, or W) atoms were randomly doped in the Ni matrix of the Ni/Ni₃Al interface system to investigate the influence of alloying element X on the propagation of the Ni/Ni₃Al interface crack at low (5 K) and high (1273 K) temperatures. Here, we investigate the crack propagation velocity, crack tip shape, and dislocation emission. The surface energy, unstable stacking energy, adhesion work of the interface, and bonding strength between atoms were also calculated to analyze the reasons for the influences of the alloying elements.

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2. Model and simulation conditions

In the present study, the simulation model is divided into two parts by the (010) plane. The upper part is Ni_3Al and the lower part is Ni, and the lattice constants for both parts are kept the same for the assumption of coherence. A sharp crack in the Ni/ Ni_3Al interface is obtained according to the anisotropic elastic displacement field.²⁰ As shown in Fig. 1, the crack surface is the (010) plane, and the crack front is oriented along the $[101]$ direction. The crack model consists of 200 atomic layers along the $x[10\bar{1}]$ direction, 160 atomic layers along the $y[010]$ direction, and 60 atomic layers along the $z[101]$ direction, including 480 000 atoms. Two larger simulation boxes (one with 200 atomic layers along the x and y directions, respectively, and 60 atomic layers along the z direction; and another with 200 atomic layers along the x direction, 160 atomic layers along the y direction, and 80 atomic layers along the z direction) were tested. It was found that the simulation results, such as the configuration evolution, elastic constants, surface energy, and crack propagation velocity, were insensitive to the system size. Thus, the smaller system size was used for the simulations. The initial crack length was about $l = 50$ Å. In Ni-based SC superalloys, it is known that the alloying elements Re and Co tend to distribute in the γ phase, while W uniformly distributes in both the γ and γ' phases.^{21,22} In the present study, it was confirmed that the crack propagates in the γ -Ni matrix phase near the interface (as shown in Sections 3.2.1 and 3.3.1), and the path of the crack propagation did not change even the X atoms were randomly distributed in the Ni_3Al phase of the Ni/ Ni_3Al superalloy system. Thus, 1 or 2 at% X ($X = \text{Re}, \text{Co}, \text{or W}$) atoms were randomly doped into the Ni matrix of the Ni/ Ni_3Al interface model to compare the effects of the three alloying elements.

In experiments, the γ/γ' (Ni/ Ni_3Al) interface of Ni-based SC superalloys is a diffuse interface with a width of about 1–3 nm.^{22,23} There are two interfacial widths, one corresponding to an order–disorder transition and the other to compositional transition across the interface.²³ In this work, our purpose is focused on the doping effect of alloying elements on the Ni/ Ni_3Al interface. Thus, the Ni/ Ni_3Al interface model without transition layers was adopted in the present study.

As shown in Fig. 2, three different types of mode I interface cracks were originally constructed as follows. (i) The upper crack surface consists of Ni atoms, and the lower crack surface

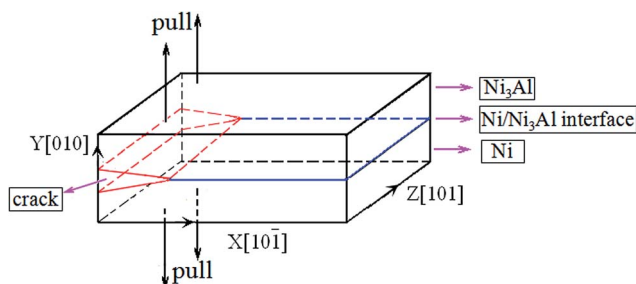


Fig. 1 Structural model of the Ni/ Ni_3Al interface crack. The blue line is the interface and the red lines are the upper and lower crack surfaces.

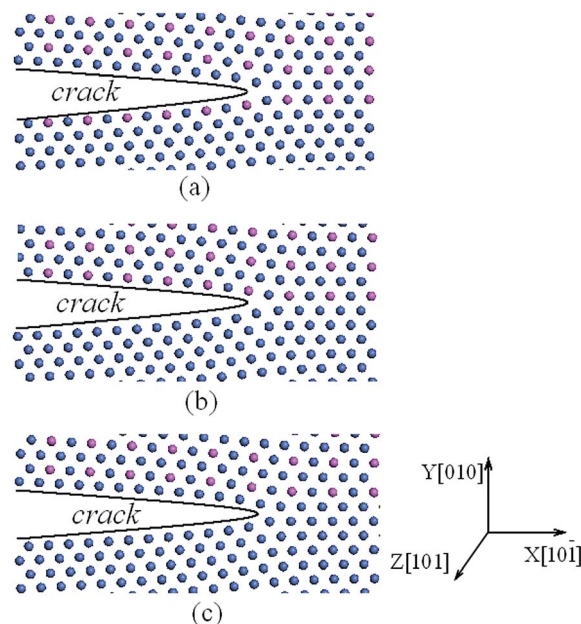


Fig. 2 Three different types of mode I interface cracks: (a) the upper crack surface consists of Ni atoms, and the lower crack surface consists of Ni and Al atoms; (b) the upper crack surface consists of Ni and Al atoms, and the lower crack surface consists of Ni atoms; and (c) both crack surfaces consist of Ni atoms and no Al atoms.

consists of Ni and Al atoms (Fig. 2(a)). (ii) The upper crack surface consists of Ni and Al atoms, and the lower crack surface consists of Ni atoms (Fig. 2(b)). (iii) Both of the crack surfaces consist of Ni atoms (Fig. 2(c)). In the present study, the MD simulations showed that the crack always propagates along the Ni–Ni layers near the Ni/ Ni_3Al interface, as shown in Fig. 2(c), which is consistent with experiments.^{24–26} Therefore, the type (iii) interface crack model was used to perform the following simulations. Fig. 3 shows the atomic configuration of the Ni/ Ni_3Al interface systems used in this study with 2 at% X ($X = \text{Re}, \text{Co}, \text{or W}$) atoms at the crack tip before relaxation.

The crack system was loaded in mode I, and strain loading was performed in the $y[010]$ direction. The strain rate was $\dot{\epsilon} = 1 \times 10^9 \text{ s}^{-1}$. The periodic boundary condition was applied in the direction parallel to the crack front, and fixed boundary conditions (the outermost four atomic layers were fixed) were used in the $x[10\bar{1}]$ and $y[010]$ directions. In the MD simulations, Newton's equations of motion were solved with the Gear algorithm.²⁷ The time steps were 5×10^{-15} and 1×10^{-15} s at 5 and

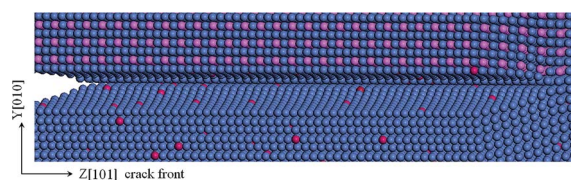


Fig. 3 Atomic configuration of the Ni/ Ni_3Al interface system with 2 at% X ($X = \text{Re}, \text{Co}, \text{or W}$) at the crack tip before relaxation. The blue, pink, and red balls represent Ni, Al, and X atoms, respectively.

1273 K, respectively. In the simulation process, the temperature was kept constant with the velocity renormalization technique. The Ni–Al–X (X = Re, Co, or W) embedded-atom method (EAM) potentials^{13–15} and the XMD program²⁸ were used.

3. Results and discussion

3.1 Mechanical parameters of the Ni matrixes with and without X (X = Re, Co, or W) addition

To examine the reliability of the Ni–Al–X (X = Re, Co, or W) EAM potentials, we calculated the three independent elastic constants C_{11} , C_{12} , and C_{44} , the bulk modulus B , the shear modulus G , Young's modulus E , and the lattice constant a for the perfect Ni matrixes with 0, 1, and 2 at% X atoms.

The simulated results are listed in Table 1. Herein, the bulk modulus $B = (C_{11} + 2C_{12})/3$ and the shear modulus G were calculated as the arithmetic Hill average;²⁹ $G = (G_V + G_R)/2$, where $G_V = (C_{11} - C_{12} + 3C_{44})/5$ and $G_R = 5/(4S_{11} - 4S_{12} + 3S_{44})$ (S_{11} , S_{12} , and S_{44} are the elastic compliances³⁰) are the Voigt and Reuss bounds, respectively. Young's modulus E was obtained as $E = 9GB/(G + 3B)$. In Table 1, the lattice constant of pure Ni is 3.52 Å, which is in agreement with previous theoretical and experimental results.^{31,32} With the addition of Re or W atoms, the lattice constant of the system increases, while it does not change with the addition of Co atoms. This can be explained by the different atomic radii of the host and alloying atoms. The radii of Re (1.37 Å) and W (1.37 Å) are larger than that of the host atom Ni (1.25 Å), and the radius of Co (1.25 Å) is comparable with that of the host atom. Furthermore, except for Co, the values of C_{11} , C_{12} , C_{44} , B , E , and G increase with the addition of Re or W atoms. The higher the concentration of Re or W atoms, the larger the values of C_{ij} , B , E , and G , which reveals that Re and W can improve the mechanical strength of superalloys. These results are consistent with previous studies.^{33,34} The ductile/brittle behavior of the material is related to the ratio of G/B or Poisson's ratio ν ($\nu = (3B - 2G)/(3B + 2G)$). The lower the value of G/B or the higher the value of ν , the more ductile the material. From Table 1, with increasing concentration of Re or W, the ratio of G/B decreases and the value of ν increases. However, the addition of Co has no effect on the values of G/B and ν . This means that Re and W can increase the

ductility of superalloys at high temperature. It is clear from Table 1 that different elements have different effects on the elastic properties of the Ni matrix.

In addition, for pure Ni, the difference of the elastic constants between the present study and previous theoretical and experimental results^{31,32} is negligible. Thus, it can be concluded that the Ni–Al–X (X = Re, Co, or W) EAM potentials well describe the systems in the present study.

3.2 Influence of alloying element on crack propagation at low temperature

To investigate the influence of alloying element X (X = Re, Co, or W) on brittle crack propagation of the Ni/Ni₃Al interface at low temperature, the structure evolution of the crack tip and the crack propagation velocity at 5 K were simulated. In addition, the adhesion work (W_{ad}) of the interface was calculated to investigate the reason for the influence of the alloying elements on the crack propagation velocity at low temperature.

3.2.1 Crack-tip shape of the Ni/Ni₃Al interface. Fig. 4 shows the interface crack-tip shapes at 20 ps under the strain rate of $\dot{\epsilon} = 1 \times 10^9 \text{ s}^{-1}$ with 2 at% X (X = Re, Co, or W). From the evolution configurations of the crack tip in Fig. 4, it can be seen that the interface crack propagates in a cleavage manner and the crack tip retains the original sharp shape (as shown in Fig. 2) at low temperature with and without the alloying element addition. During the process of crack propagation, there is no dislocation near the crack tip even with the addition of alloying atoms. Moreover, the crack path does not change and the crack propagates along the Ni/Ni₃Al interface, which is in agreement with experimental studies.^{24–26}

3.2.2 Crack propagation velocity of the Ni/Ni₃Al interface. Table 2 lists the calculated values of the interface crack velocity under the strain rate $\dot{\epsilon} = 1 \times 10^9 \text{ s}^{-1}$ at 5 K with and without the addition of alloying element X (X = Re, Co, or W). From Table 2, the crack propagation velocity clearly decreases with the addition of 2 at% Re or W atoms, but not with the addition of Co. For example, with the addition of 2 at% Re and W, the crack propagation velocities were 274.3 and 229.7 m s^{−1} at 30 ps, which are 8.8% and 23.7% less than the crack propagation velocity without alloying element addition (300.9 m s^{−1}), respectively. In contrast, the crack propagation velocity

Table 1 Mechanical parameters of the Ni systems with and without X (X = Re, Co, or W) addition, including the elastic constants C_{11} , C_{12} , and C_{44} , bulk modulus B , shear modulus G , Young's modulus E , G/B ratio, Poisson's ratio ν and lattice constant a

System		C_{11} (GPa)	C_{12} (GPa)	C_{44} (GPa)	B (GPa)	G (GPa)	E (GPa)	G/B	ν	a (Å)
Ni	This work	237	150	127	179.0	82.7	215.4	0.462	0.299	3.520
	EAM ^a	241	151	127	181.0	83.8	217.9	0.463	0.299	3.52
	Exp ^b	246	147	125	180.0	86.2	223.0	0.478	0.293	3.52
Ni(Re)	1 at% Re	239	152	128	181.0	83.1	216.2	0.459	0.301	3.524
	2 at% Re	241	154	129	183.0	83.5	217.5	0.456	0.302	3.527
Ni(Co)	1 at% Co	236	149	126	178.0	82.3	213.9	0.462	0.299	3.519
	2 at% Co	234	148	125	176.7	81.6	212.0	0.462	0.299	3.519
Ni(W)	1 at% W	238	152	128	180.7	82.7	215.3	0.458	0.301	3.524
	2 at% W	239	154	129	182.3	82.8	215.6	0.454	0.303	3.528

^a Ref. 31. ^b Ref. 32.

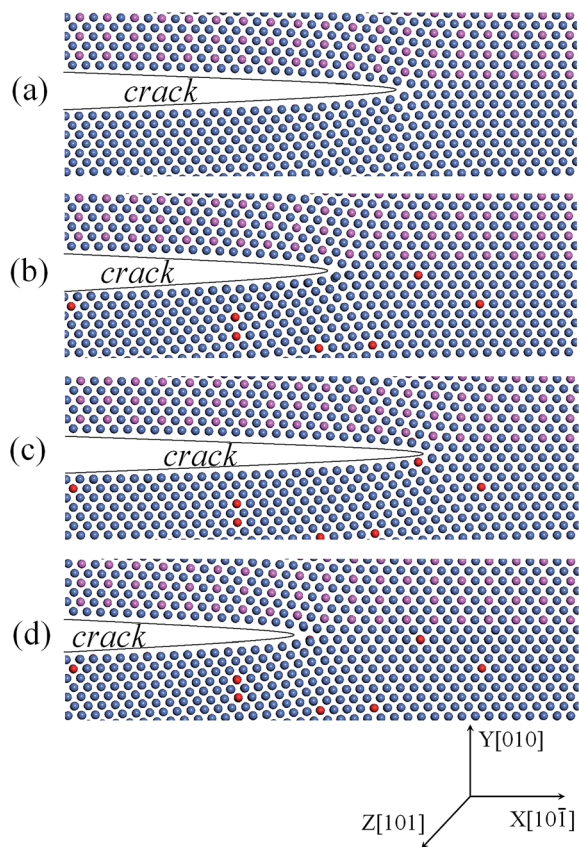


Fig. 4 Crack-tip shapes at 20 ps under the strain rate $\dot{\epsilon} = 1 \times 10^9 \text{ s}^{-1}$ at 5 K. The blue, pink, and red balls represent Ni, Al, and X (X = Re, Co, or W) atoms, respectively. (a) Without the addition of alloying element, and (b)–(d) with the addition of 2 at% Re, Co, and W atoms, respectively.

increases by 4.2% with 2 at% Co addition. The order of the crack propagate velocity with the alloying element addition is $\text{Co} > \text{No} > \text{Re} > \text{W}$ (“No” means “the interface system without the addition of alloying element”). This indicates that Re and W can reduce the brittle crack propagation velocity of the Ni/Ni₃Al interface at low temperature, and W has the greatest effect. This may be related to the bonding strength between atoms, which will be discussed in Section 3.4.

3.2.3 Adhesion work of the Ni/Ni₃Al interface. During the process of interface separation, the atomic bonds at the crack tip are broken and crack surfaces are generated. The formation of new crack surfaces is related to the crack propagation

Table 2 Crack propagation velocity (V_{crack}) (m s^{-1}) at different simulation times for the Ni/Ni₃Al interface systems with and without X (X = Re, Co, or W) additions under $\dot{\epsilon} = 1 \times 10^9 \text{ s}^{-1}$ at 5 K. Ni (2 at% X) means 2 at% X randomly doped in the Ni matrix

	Ni/ Ni ₃ Al	Ni (2 at% Re)/ Ni ₃ Al	Ni (2 at% Co)/ Ni ₃ Al	Ni (2 at% W)/ Ni ₃ Al
V_{crack} (at 30 ps)	300.9	274.3	313.5	229.7
V_{crack} (at 35 ps)	334.2	311.7	347.7	267.6
V_{crack} (at 40 ps)	401.1	374.1	411.3	336.3

velocity. It is known that the adhesion work (W_{ad}) is the reversible energy change to form free surfaces from interfaces.³⁵ Accordingly, the brittle crack propagation velocity should be inversely proportional to the adhesion work. Thus, to investigate the reason for the change of the crack propagation velocity with the addition of alloying elements, it is necessary to determine the adhesion work of the Ni/Ni₃Al interface by the MD method.

As shown in Fig. 5, the simulation model contained 8000 atoms, and had dimensions of $50.4 \times 71.3 \times 25.2 \text{ \AA}^3$ in the x [101], y [010], and z [101] directions, respectively. Periodic boundary conditions were used in the x and z directions, and the free boundary condition was used in the y direction. The W_{ad} of the interface can be obtained by the following expression:^{5,12,36}

$$W_{\text{ad}} = (E_{\text{Ni}} + E_{\text{Ni}_3\text{Al}} - E_{\text{Ni/Ni}_3\text{Al}})/S \quad (1)$$

where E_{Ni} and $E_{\text{Ni}_3\text{Al}}$ are the energies of fully relaxed Ni and Ni₃Al (as shown in Fig. 5), respectively. $E_{\text{Ni/Ni}_3\text{Al}}$ is the total energy of the fully relaxed Ni/Ni₃Al interface system, and S is the interfacial area.

Table 3 lists the W_{ad} values of the systems with and without the addition of alloying element X (X = Re, Co, or W) in interface layers i and j (as shown in Fig. 5). From Table 3, the addition of Re or W atoms can effectively increase the adhesion work of the Ni/Ni₃Al interface. For example, compared with the Ni/Ni₃Al interface system without alloying element addition, the values of W_{ad} increase by 1.43% and 2.69% with the addition of 2 at% Re and W atoms, respectively, but W_{ad} decreases by 0.23% with 2 at% Co addition. The magnitude of the change of the W_{ad} value with alloying elements is in the order: $\text{W} > \text{Re} > \text{Co}$, which is the opposite trend to the influence of the alloying elements on the interface brittle crack propagation velocity (as discussed in Section 3.2.2). That is, the larger the value of W_{ad} , the smaller the crack propagation velocity of the interface. Therefore, it can be predicted that an alloying element with a large W_{ad} value will reduce the brittle crack propagation velocity of the Ni/Ni₃Al interface. Moreover, the value of W_{ad} increases with the increasing atomic concentration of Re or W, but decreases for

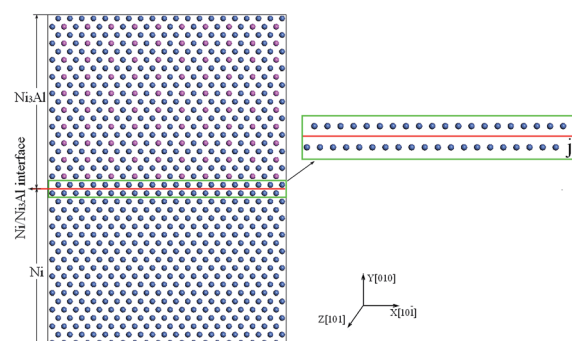


Fig. 5 Model of the adhesion work of the Ni/Ni₃Al interface. The blue and pink balls represent Ni and Al atoms, respectively. The red line represents the interface. Alloying X (X = Re, Co, or W) atoms can be doped in the interface atomic layers i and j .

Co. This indicates that at low temperature Re and W can inhibit brittle crack propagation of the Ni/Ni₃Al interface, and W has the greatest effect, while Co can accelerate interface crack propagation, which agrees well with the results discussed in Section 3.2.2.

3.3 Influence of alloying element on the crack propagation at high temperature

To investigate the influence of the alloying element on the crack propagation of the Ni/Ni₃Al interface at high temperature, we gave the crack-tip shapes at 1273 K. The surface energy and unstable stacking energy were calculated to evaluate the competition between crack cleavage and dislocation emission at the crack tip.

3.3.1 Crack-tip bluntness of the Ni/Ni₃Al interface. Fig. 6 shows the crack-tip shapes of the Ni/Ni₃Al interface with and without X (X = Re, Co, or W) addition at 8 ps under the strain rate $\dot{\epsilon} = 1 \times 10^9 \text{ s}^{-1}$ at 1273 K. From Fig. 6, at high temperature the cracks advance in a ductile manner in the Ni matrix, which is consistent with experiments,^{24–26} and the crack tips become blunt with the addition of the alloying elements.

In this section, to evaluate the bluntness of crack tip, the curvature radius (R) was given by drawing an inscribed circle at the crack tip. The curvature radius of the crack tip increases when adding 2 at% Re ($R = 0.35 \text{ \AA}$ in Fig. 6(b)) or W ($R = 0.42 \text{ \AA}$ in Fig. 6(d)). However, when adding 2 at% Co ($R = 0.23 \text{ \AA}$ in Fig. 6(c)), the curvature radius is the same as without alloying element addition ($R = 0.23 \text{ \AA}$ in Fig. 6(a)). This may be because the Ni–Re and Ni–W bonds are stronger than the Ni–Ni bond while the Ni–Co bond is weaker, which will be confirmed in Section 3.4. The larger curvature radius means that bond breaking of the crack tip is more difficult and the dislocation emission near the crack tip becomes easy. Obviously, the above discussion shows that Re and W can prevent the bond breaking in the cleavage plane. Furthermore, W has the greatest strengthening effect on the interface crack at high temperature.

3.3.2 Dislocation emission near the crack tip. At high temperature (1273 K), there is dislocation emission near the crack tip of the Ni/Ni₃Al interface with the addition of alloying element X (X = Re, Co, or W) under the strain rate $\dot{\epsilon} = 1 \times 10^9 \text{ s}^{-1}$. When 2 at% W atoms were added in the Ni matrix, the $a/6\langle 112 \rangle\{111\}$ partial dislocation appeared near the crack tip at 45 ps (as shown by the red atoms in Fig. 7), while at the same time there was no dislocation with the addition of 2 at% Re or

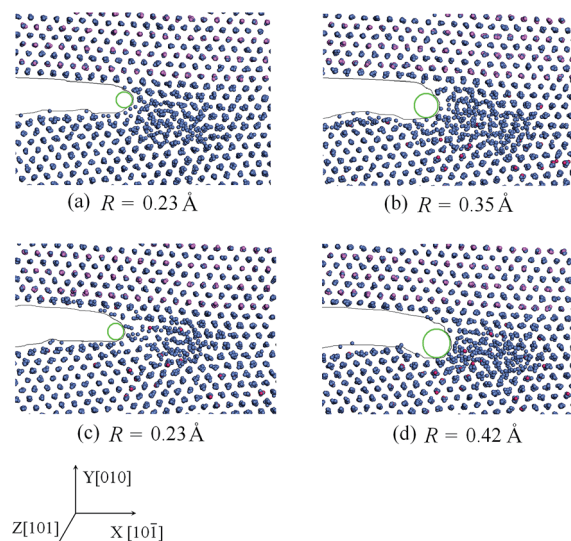


Fig. 6 Crack-tip shapes at 8 ps under the strain rate $\dot{\epsilon} = 1 \times 10^9 \text{ s}^{-1}$ at 1273 K. The blue, pink, and red balls represent Ni, Al, and X (X = Re, Co, or W), respectively. The green circles represent the curvature radius (R) of the crack tips. (a) Without the addition of an alloying element. (b)–(d) With the addition of 2 at% Re, Co, and W atoms, respectively.

Co. This reveals that dislocation emission becomes easy with the addition of W atoms at high temperature.

3.3.3 Surface energy and unstable stacking energy. The competition between cleavage and dislocation emission can be evaluated by the ratio of the surface energy (γ_s) to the unstable stacking energy (γ_{us}).^{12,37} In the present study, the dislocation emission ability with the addition of alloying element was investigated using the γ_s/γ_{us} ratio.¹²

The crack propagates in the Ni matrix at high temperature, as shown in Fig. 6, and γ_s for the (100) plane of the Ni matrix was calculated using the MD method. The calculation model contained 240 000 atoms, including 200 atomic layers along the $x[10\bar{1}]$ direction, 80 atomic layers along the $y[010]$ direction, and

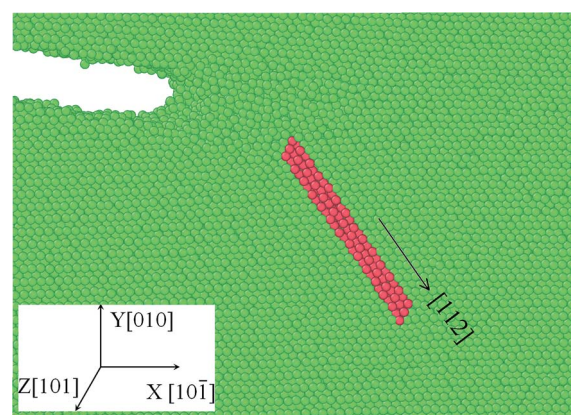


Fig. 7 Atomic configuration of the crack tip for the system with 2 at% W atoms at 45 ps and 1273 K. The green atoms represent the area without dislocation, and red atoms represent the area with the $a/6\langle 112 \rangle\{111\}$ partial dislocation.

Table 3 Adhesion work (W_{ad}) of the Ni/Ni₃Al interface with and without X (X = Re, Co, or W)

System		$W_{ad} (\text{J m}^{-2})$
Ni/Ni ₃ Al		3.016
Ni(Re)/Ni ₃ Al	1 at% Re	3.038
	2 at% Re	3.059
Ni(Co)/Ni ₃ Al	1 at% Co	3.013
	2 at% Co	3.009
Ni(W)/Ni ₃ Al	1 at% W	3.056
	2 at% W	3.097

60 atomic layers along the $z[101]$ direction. The surface energy γ_s is defined as^{12,38}

$$\gamma_s = (E_{\text{off}} - E_{\text{on}})/(2S) \quad (2)$$

where E_{off} is the potential energy of the system with periodic boundary conditions in the x and z directions, and a free boundary condition in the y direction, E_{on} is the potential energy of the system with periodic boundary conditions in the x , y , and z directions, and S is the area of the (010) plane.

Table 4 lists the calculated values of γ_s for the systems with and without alloying element X (X = Re, Co, or W) in the Ni matrix. Compared with the pure Ni matrix, the value of γ_s increases with the addition of Re or W atoms, and decreases with the addition of Co atoms. Moreover, the higher the concentration of Re or W atoms, the larger the value of γ_s , while the opposite trend is observed for Co. For example, the calculated values of γ_s with 1 and 2 at% Re atoms were 1.602 and 1.616 J m⁻², while for Co they were 1.589 and 1.586 J m⁻², respectively. Comparing the values of γ_s for the same concentrations of Re and W atoms, W has a greatest effect on the surface energy, which may be related to the strengths of the Ni-X bonds, which will be discussed in Section 3.4.

The energy barrier for partial dislocation nucleation can be approximately evaluated by the unstable stacking energy (γ_{us}), which was introduced by Rice.³⁷ As shown in Fig. 7, the $a/6\langle 112 \rangle$ Shockley partial dislocation occurs in the Ni matrix with the addition of alloying element W. Here, the γ_{us} values of the $a/6\langle 112 \rangle$ Shockley partial dislocation for the systems with and without X (X = Re, Co, or W) atoms were calculated by the MD method. The simulation model included 201 600 atoms, with 168 atomic layers along the $x[112]$ direction, 120 atomic layers along the $y[11\bar{1}]$ direction, and 60 atomic layers along the $z[\bar{1}10]$ direction. Periodic boundary conditions were applied in the x and z directions, and a free boundary condition was used in the y direction. The simulation model was divided into two equal parts by the (11 $\bar{1}$) plane: the lower half part was fixed and the upper half part was gradually slipped along the [112] direction in the (11 $\bar{1}$) plane. Then, the relative energy difference of the system before and after slipping was calculated by^{12,39}

$$E_{\text{POT}}(R,0) = (E_{\text{POT}}(R) - E_{\text{POT}}(0))/S \quad (3)$$

Table 4 Surface energy (γ_s), unstable stacking energy (γ_{us}), and $\gamma_s/\gamma_{\text{us}}$ ratio for Ni matrixes with and without X (X = Re, Co, or W) atoms

System		γ_s (J m ⁻²)	γ_{us} (J m ⁻²)	$\gamma_s/\gamma_{\text{us}}$
Ni	This work	1.592	0.2778	5.731
	Previous ^a	1.592	0.2775	5.737
Ni(Re)	1 at% Re	1.602	0.2773	5.777
	2 at% Re	1.616	0.2763	5.849
Ni(Co)	1 at% Co	1.589	0.2755	5.768
	2 at% Co	1.586	0.2742	5.784
Ni(W)	1 at% W	1.614	0.2582	6.251
	2 at% W	1.643	0.2499	6.575

^a Ref. 12.

where R is the relative slip distance between the lower and upper parts, $E_{\text{POT}}(R)$ and $E_{\text{POT}}(0)$ are the potential energies of the systems at relative slip distances R and 0, respectively, and S is the total area of the (11 $\bar{1}$) plane. When $R = 0.5b$ (b is the magnitude of the Burgers vector of the $a/6[112]$ Shockley partial dislocation), the γ_{us} of the system can be approximately obtained.

Table 4 lists the calculated γ_{us} values for the systems with (1 and 2 at%) and without X (X = Re, Co, or W) atoms. From Table 4, the value of γ_{us} decreases with the addition of alloying elements, and the higher the concentration of X atoms, the lower the values of γ_{us} . The ratio of $\gamma_s/\gamma_{\text{us}}$ is also listed in Table 4. The $\gamma_s/\gamma_{\text{us}}$ ratios all increase when 1 or 2 at% X (X = Re, Co, or W) atoms are randomly doped into the Ni matrix. As an example, the $\gamma_s/\gamma_{\text{us}}$ ratio increases by 2.06%, 0.93%, and 14.73% with the addition of 2 at% Re, Co, and W, respectively. This reveals that at high temperature, dislocation emission becomes much easier and cleavage is more difficult when alloying atoms are randomly doped into the pure Ni matrix. Doping with W results in the largest increase of $\gamma_s/\gamma_{\text{us}}$, which suggests that the dislocation more easily occurs with the addition of W atoms than with Re or Co atoms. This was confirmed in Section 3.3.2. Moreover, the $a/6\langle 112 \rangle$ Shockley partial dislocation occurs in the Ni matrix with 6 at% Re (or Co) addition at 45 ps, which is consistent with a previous study.¹² Accordingly, we predict that a dislocation will occur in the Ni matrix of a Ni/Ni₃Al interface system with a large $\gamma_s/\gamma_{\text{us}}$ ratio when doped with an appropriate amount of an alloying element.

3.4 Bonding strength of Ni-X (X = Re, Co, or W)

From the above results, the effects of the three alloying elements Re, Co, and W on the crack propagation of the Ni/Ni₃Al interface at low or high temperature are different. The reason may be related to the interatomic interactions of Ni-X (X = Re, Co, or W). Thus, it is necessary to calculate and compare the bonding strengths of Ni-X and Ni-Ni.

3.4.1 Bonding strength estimated from the well depth of the pair potential. It is well known that the bonding strength between solute and matrix atoms can be approximately estimated from the well depth of the pair potential in the EAM potentials.^{12,38,40} Fig. 8 shows the variation of the Ni-Ni and Ni-X (X = Re, Co, or W) pair potentials with distance in the EAM potential. The greater the depth of the potential well means the stronger the bond. From Fig. 8, the depths of the potential wells for Ni-Re and Ni-W are greater than that of Ni-Ni, which indicates that it is more difficult to break the Ni-Re and Ni-W bonds. The bonding strength of Ni-W is the highest, while the Ni-Co bond is weaker than the Ni-Ni bond.

3.4.2 Bonding strength calculated by the discrete-variational method (DVM). The bonding strength between atoms in a local region can be accurately determined by the interatomic energy (IE)⁴¹ using *ab initio* calculations. The bonding strengths of Ni-Re and Ni-Ni atoms in the Ni matrix and Ni₃Al have been investigated by this method.^{38,42} The IE is expressed as^{43–45}

$$E_{ij} = \sum_n N_n \sum_{\alpha\beta} a_{n\alpha}^* a_{n\beta} H_{\beta\alpha l} \quad (4)$$

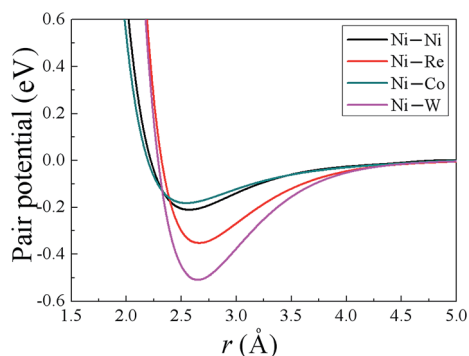


Fig. 8 Variation of the pair potentials of Ni–Ni and Ni–X (X = Re, Co, or W) with distance (r) in the EAM potential.

where N_n is the electron occupation number for molecular orbital ψ_n and $H_{\beta\alpha l}$ is the Hamiltonian matrix element that connects the atomic orbital β of atom j to the atomic orbital α of atom l . The coefficient $a_{n\alpha l}$ is obtained by $a_{n\alpha l} = \langle \phi_{\alpha l} | \psi_n(r) \rangle$. The IE can be used to evaluate the bonding strength of two adjacent atoms because it is related to the Hamiltonian matrix element and the wave vector matrix in the representation. Generally, a larger absolute value of IE means a stronger interatomic interaction. The IE can be obtained using the DVM, which has been successfully used to describe the electronic structures of metals and alloys.^{46,47}

Because both the upper and lower crack surfaces consist of Ni atomic layers, we only considered the interatomic energies between alloying element X (X = Re, Co, or W) and Ni atoms. First, two interface crack models were constructed to calculate the IE: one with X on the upper surface of the crack tip (Fig. 9(a)) and the other with X on the lower surface (Fig. 9(b)). Both of the crack models consisted of 8000 atoms, with 20 atomic layers along the $z[101]$ direction and 40 atomic layers along the $x[10\bar{1}]$ and $y[010]$ directions. The alloying X (X = Re, Co, or W) atom was at the center of the crack tip along the crack front. Then, each model was relaxed by MD simulation with the Ni–Al–X (X = Re, Co, or W) EAM potential.^{13–15} Finally, a cluster containing 150 atoms located at the crack tip of the Ni/Ni₃Al interface was selected to perform the DVM calculation. Here, spin-

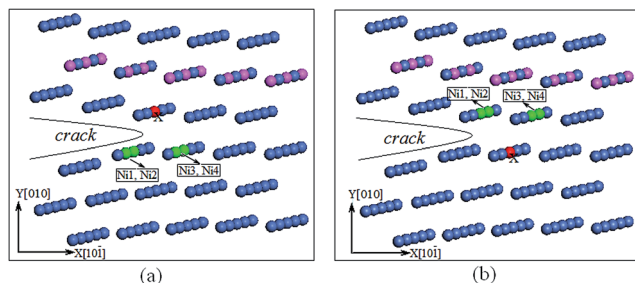


Fig. 9 Computational configurations of the DVM: (a) with a single X (X = Re, Co, or W) atom on the upper surface of the crack tip, and (b) with a single X atom on the lower surface of the crack tip. The blue, pink, and red balls represent Ni, Al, and X atoms, respectively. The green balls (Ni1–Ni4) represent the nearest-neighbor atoms of X.

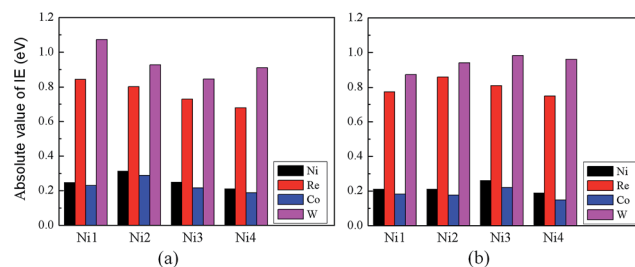


Fig. 10 Absolute values of the interatomic energy (IE) between alloying element X (X = Re, Co, or W) and its four nearest-neighbor atoms (Ni1–Ni4) by DVM calculations. (a) Absolute values of the IE for the configuration with a single alloying atom X on the upper surface of the crack tip. (b) Absolute values of the IE for the configuration with an X atom on the lower surface of the crack tip.

polarization calculations were performed, and more calculation details of the IE can be found in the literature.^{45,48}

Fig. 10 shows the absolute values of the IE between alloying atom X (X = Re, Co, or W) and its four nearest-neighbor atoms. In Fig. 10(a) and (b), the absolute values of IE between Re (or W) and the four nearest-neighbor atoms (Ni1–Ni4) are all larger than that of Ni–Ni. Ni–W has the largest absolute value of IE, which is more than three times larger than the IE of Ni–Ni. This reveals that there is a stronger bond between Re (and W) and Ni atoms than between Ni atoms, and the Ni–W bond is the strongest. Co exhibits the smallest absolute IE with Ni1–Ni4, which is even less than the IE of Ni–Ni. This suggests that there is a weaker bond between Co and Ni atoms than between Ni and Ni atoms. From Fig. 10, the order of the interaction between alloying atom X (X = Re, Co, or W) and Ni atoms in the Ni/Ni₃Al interface crack is $W > Re > Co$, which agrees with the result from the well depth of the pair potential, as discussed in Section 3.4.1.

From the above discussion, we can conclude that the Ni–Re and Ni–W bonds are indeed stronger than Ni–Ni. Therefore, at low temperature, Re and W can prevent brittle propagation of the Ni/Ni₃Al interface crack, while at high temperature the crack tip becomes blunter with the addition of Re or W atoms. In contrast, the trend for Co is the opposite because the Co–Ni bond is weaker than the Ni–Ni bond. Moreover, W forms the strongest bonds with Ni atoms. Thus, at high temperature dislocation begins to appear from the crack tip with 2 at% W doping in the Ni matrix.

4. Conclusions

In summary, the Ni/Ni₃Al interface crack at the (010)[101] orientation was constructed to investigate the influence of the alloying elements Re, Co, and W on the propagation of the interface crack at low and high temperatures. The results are as follows:

(1) The alloying elements Re and W can improve the mechanical properties of superalloys. Moreover, the calculated results, such as the ratio of G/B and Poisson's ratio ν , show that Re and W can increase the ductility of superalloys at high temperature, and W has a greater effect than Re.

(2) The Ni/Ni₃Al interface crack propagates in a brittle manner at low temperature, but in a ductile manner at high temperature, both with and without the addition of alloying elements.

(3) The alloying elements Re and W can restrict brittle propagation of the Ni/Ni₃Al interface crack because Ni–Re and Ni–W bonds are stronger than Ni–Ni bonds. This means that Re and W can decrease the brittleness of superalloys, and W has a greater effect. Moreover, the propagation velocity of the brittle crack is inversely proportional to the adhesion work of interface, which can be used to predict that an alloying element with large W_{ad} is able to prevent brittle crack propagation.

(4) At high temperature, the crack tip of the Ni/Ni₃Al interface becomes blunt with the addition of Re or W atoms, and dislocation occurs near the crack tip when 2 at% W atoms are added. This reveals that the alloying elements Re and W can improve the ductility of superalloys at high temperature, and W has the greater effect.

(5) The competition between cleavage and dislocation emission of the interface crack can be evaluated by the ratio γ_s/γ_{us} , the higher the ratio, the easier dislocation emission and the more difficult cleavage.

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