

AN OXIDE DISPERSION STRENGTHENED Ni-W-Al ALLOY  
WITH SUPERIOR HIGH TEMPERATURE STRENGTH

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ABSTRACT

An experimental oxide dispersion strengthened (ODS) alloy, WAZ-D, derived from the WAZ-20 composition was produced by the mechanical alloying process. Cast WAZ-20 is strengthened by both a high refractory metal content, and 70 volume percent of gamma prime. The ODS alloy WAZ-D, was responsive to variables of alloy content, of attritor processing, of consolidation by extrusion, and of heat treatment. The best material produced had large highly elongated grains. It exhibited tensile strengths generally superior to a comparable cast alloy. The ODS alloy exhibited high temperature stress rupture life considerably superior to any known cast superalloy; for example, the stress rupture life of WAZ-D determined in vacuum at 1150° C (2100° F) and 102 MPa (15 ksi) was approximately 1000 hours. Tensile and rupture ductility were low, as was intermediate temperature rupture life. Very low creep rates were noted and some specimens failed with essentially no third stage creep. The alloy showed some potential for low stress post-extrusion forming. Comparison of WAZ-D to conventionally cast WAZ-20 and to directionally solidified WAZ-20 indicates that oxide dispersion strengthening may be beneficially added to even the strongest of superalloy compositions. Also the benefit derived from the oxide dispersion, in conjunction with the elongated microstructure, far out-weighed that from the elongated microstructure alone.

## INTRODUCTION

Advances in the design and utilization of gas turbine engines are closely tied to advances in material capability. Gas turbine blades in particular, demand extraordinary strength at high temperatures for prolonged periods of time. Among the classes of materials which may allow a substantial increase in gas turbine operating temperature are the oxide dispersion strengthened (ODS) superalloys. By applying ODS to a superalloy composition, a material may be fabricated combining strength derived from the  $\gamma'$  phase at intermediate temperatures and strength derived from the oxide dispersion at high temperatures, where gamma prime strengthening normally becomes ineffective.

In previous studies, alloys having low gamma prime fractions<sup>(1)</sup> or higher fractions up to about 60 volume percent<sup>(2,3)</sup> have been dispersion strengthened. These alloys have had gamma prime solvus temperatures generally well below their incipient melting temperatures. The objective of this study was to determine whether an alloy, WAZ-20,<sup>(4)</sup> having an even higher volume fraction of gamma prime, approximately 70 percent, and a very high gamma prime solvus, above the cast material's incipient melting temperature, could be effectively dispersion strengthened. To this end, oxide dispersion strengthened alloys based on the WAZ-20 composition were prepared and evaluated. In this report, the results of microstructural examinations and mechanical property determinations are discussed. Alloy content and processing variables were examined only to that extent necessary to define suitable powder and bar microstructures; results of these examinations are given here because of their possible interest to other investigators.

## MATERIALS AND PROCEDURE

Materials for this investigation were prepared at NASA-Lewis by the mechanical alloying process described by Benjamin.<sup>(5)</sup> Elemental Ni, W, and Zr powders were blended with  $ZrB_2$ , argon atomized  $Ni_3Al$  and preground  $Y_2O_3$  in a high energy attrition mill. In one trial, the Al was also added as an elemental powder. This resulted in failure as the alloyed powder could not be removed from the balls. Powder proportions were chosen to yield a nickel-based alloy of approximately 7 percent Al, 1 percent Zr, 1.1 percent  $Y_2O_3$ , 0.015 percent B, and either 8 or 17 percent W. In the latter case, the base alloy without  $Y_2O_3$  would resemble the cast alloy WAZ-20 noted for its high strength in the 1100° to 1200° C range.

The attrition mill was sealed by a simple inverted cup, rotating in a fluid filled annulus. The mill charge consisted of 5.6 Kg of powder and 110 Kg of 0.6 cm diameter alloy steel balls. Milling was conducted in a 0.4 m diameter stirred mill at 150 rpm under a flowing atmosphere of 0, 3.5, or 7 ml/min  $O_2$  added to 190 ml/min  $N_2$ . When oxygen was omitted the powder could not be removed from the balls. After 40 hours, or for one run of 16 hours, the resultant powder was discharged from the mill without exposure to air for examination, and further processing. The mechanically alloyed powder was loaded into 7.6 cm O.D. extrusion cans, outgassed under vacuum at 800° C, and consolidated by hot extrusion. Extrusion ratios of 16:1 and 20:1 and extrusion temperatures of 1095°, 1120°, 1175°, and 1230° C were used. (The extension ram speed was 0.3 m/s.) The extrusion dies were conical with an included angle of 90°. After extrusion, samples were annealed at various temperatures from 1290° to 1380° C to determine their recrystallization response. (Heat up rate was also varied.) Materials for mechanical tests were annealed at 1320° C in a thermal gradient provided by lowering the extruded rod into a salt bath. The entrance rate was 0.6 m per hour. After annealing, button head specimens having a 25 mm gauge length and 2.8 mm gauge diameter were ground for mechanical tests.

Specimens were tested in tension at 0.02 per minute strain rate at temperatures ranging from ambient to 1320° C (Table 2) and in stress rupture at

760°, 1040°, 1095°, 1150°, and 1205° C (Table 3). All stress rupture tests were conducted in vacuum, as were tensile tests above 1205° C. Specimens were prepared for optical and electron microscopy by standard metallographic techniques. For optical microscopy, an etchant of 2.5gFeCl<sub>3</sub>, 2.5gCuCl<sub>2</sub>, 10mlHNO<sub>3</sub>, 30mlHCl, and 30 ml ethyl alcohol was used. For electron microscopy, ion etching was used. Fracture surfaces were examined by scanning electron microscopy.

## RESULTS AND DISCUSSION

### Microstructures

The most uniform mechanically alloyed powder (Fig. 1) was produced using the lower oxygen flow rate and longer processing time. The most uniform highly elongated microstructures, (Fig. 2), resulted from extrusion of this powder at 1095° C and a 20:1 reduction ratio. Less attrition processing time and higher extrusion temperatures were less favorable to complete recrystallization while the use of 16:1 extrusion ratio produced a less elongated grain structure. Only materials which were extruded at 20:1 were mechanically tested. Some of the material contained inclusions of either oxide stringers or prealloyed powder particles which escaped the mechanical alloying process unscathed; because these inclusions affected rupture lives, the tabulated rupture data (Table 3) include a microstructural rating. Oxide stringers may be avoided by protecting the powder from air exposure while at elevated temperature; unmilled particles, particularly deleterious to stress rupture life, may be avoided by control of the milling process, and may be readily detected by optical microscopy examination of the mechanically alloyed powder prior to extrusion.

The recrystallization process was quite temperature sensitive; the degree of recrystallization was 5 percent at 1290° C, 100 percent at 1315° C, 80 percent at 1340° C, and 0 at 1380° C. No evidence of melting in WAZ-D was observed at 1380° C. Recrystallized grains were always elongated for rapidly heated specimens (e.g. to temp. in 5 min.) while slow heating (from 1200° to 1320° C in one hour) resulted in large, but more nearly equiaxed grains. Typical grain dimensions for optimally heat-treated WAZ-D were 0.6 mm in the longitudinal direction and 0.045 mm in the transverse direction, while the as-extruded material had equiaxed grains of only about 0.6  $\mu$ m diameter. Thus, the growth of each large grain consumed on the order of ten million small grains.

The oxide dispersion observed in the recrystallized product (Fig. 3) was characterized by a mixture of fine and coarse particles. The largest were on the order of 1  $\mu$ m, most were small, less than 0.1  $\mu$ m. The interparticle spacing determined by lineal analysis was 0.4  $\mu$ m. Calculations based on the analysis of the products (Table 1) indicate a total volume fraction of 4.5 percent hard phases, primarily oxides. This is consistent with previous observation in mechanically alloyed material<sup>(3)</sup> in which the total hard phase content exceeded the intentional additions by almost 3 percent. Chemical analysis also revealed a pickup during the milling process of about 4 percent iron, considerably more than is usual in superalloys, either conventional or mechanically alloyed.

### Tensile properties

The results of individual tensile tests of large elongated grain WAZ-D are listed in Table 2 and are shown graphically in Figure 4. Both contain comparable data from reference 4 for cast WAZ-20. The tensile strength of elongated grain ODS WAZ-D was not highly affected by the inclusions appearing in some powder blends.

From Table 2 and Figure 4 it may be noted that the tensile strength of WAZ-D was generally above that of cast WAZ-20, coinciding only at 1095° C. The improvement at low temperature is typical of a powder metallurgy product

and is thought to be caused more by a finer distribution of gamma prime than by the added hard phase dispersion. The improvement at higher temperatures is more significant, especially since WAZ-20 is one of the strongest nickel based alloys at 1200° C. Also significant is the 83 MPa (12 ksi) tensile strength of WAZ-D at 1315° C, (2400° F) at or above the incipient melting temperature of cast WAZ-20. The tensile ductility values (Table 2) for WAZ-D are generally below those of directionally solidified WAZ-20 and similar to those of conventionally cast WAZ-20. It is interesting to note, that the room temperature (R.T.) reduction in area and elongation were both increased by prior thermal exposure under creep conditions. This suggests that the properties of WAZ-D may be affected by post recrystallization heat treatment, a possibility not fully investigated here.

### Workability

Another interesting feature not represented in the listed tensile data was that the as-extruded material, though brittle at R.T., had the high plasticity (e.g. greater than 100 percent elongation) at elevated temperature associated with very fine grains. It could be strained in tension under low stress (e.g., 27 MPa, 4 ksi) at 1040° or 1095° C and then converted by 1320° C annealing to a large elongated grain material. This indicates that WAZ-D and perhaps other ODS alloys may be worked at high temperatures and low stresses prior to recrystallization to a creep resistant microstructure. The working would best be carried out in compression to avoid the porosity and cracks which occur during tensile straining of the unrecrystallized material.

### Stress rupture properties

From the stress rupture data (Table 3 and Fig. 5) it may be observed that the large elongated grain dispersion strengthened alloy WAZ-D provides a considerable stress rupture life advantage over the comparable cast alloy. And since comparison was made to the cast alloy of even larger and more elongated grains prepared by directional solidification, most of the advantage may be ascribed to the hard phase dispersion. A comparison at 103 MPa (15 ksi) and 1150° C (2100° F) indicates a rupture life improvement of two orders of magnitude from 5 hours for the cast material to approximately 1000 hours for WAZ-D. Alternatively, this may be considered as in Figure 6 as a use temperature advantage of 130° C over the strongest conventional cast alloys<sup>(6)</sup> or 90° C over the gamma-gamma prime-delta directionally solidified eutectic,<sup>(7)</sup> itself a candidate material for advanced gas turbine blades.

It may also be observed that the stress rupture life was strongly affected by the microstructure, unprocessed particles being particularly deleterious. Fracture of WAZ-D was primarily transgranular, even for long life rupture specimens. The voids which formed at transverse grain boundaries during high temperature testing, either tensile or rupture, appeared to be too few, and too scattered to provide a continuous fracture path. Fracture surfaces were generally quite rough. Longitudinal cracks were evident along grain boundaries in specimens fractured at high temperature (Fig. 7), and in specimens tested at R.T. after high temperature creep exposure. From the formation of these cracks, it is surmised that transverse properties, not measured in this investigation, would be low.

Stress rupture curves, recorded by monitoring a dial gauge affixed to the load train, showed total strains generally in the range of 1.5 percent. About 1 percent occurred soon after loading, for example, within the first 5 percent of sample life. Strain rate continued to decrease through most of the sample life. The curves turned upward only briefly before fracture. The decrease in strain rate with increasing time was in contrast to the increasing rate shown for MA-753,<sup>(8)</sup> and may be related to the formation during creep exposure of very large (600 μm wide, 100 μ thick) gamma prime platelets (Fig. 7). As in reference 9 these gamma prime plates grew transverse to the applied tensile

stress. The growth of some appeared to have increased the grain boundary roughness, and may have hindered grain boundary sliding. Approximate strain rates, covering the latter 80 percent of sample life and listed in Table 3, increased by an order of magnitude as the temperature increased from 1095° C (2000° F) to 1205° C (2200° F) and by two orders of magnitude as stress at 1095° C increased from 103 MPa (15 ksi) to 172 MPa (25 ksi). The strain during this period (the latter 80 percent of life and prior to any third stage) ranged from 0.16 percent to 0.51 percent but for most specimens was approximately 0.25 percent..

#### Effect of alloy content

Decreasing the tungsten in WAZ-D from 17 to 8 percent decreased the 1095° C (2000° F) tensile strength from 303 MPa (44 ksi) to 207 MPa (30 ksi). The R.T. tensile strength was decreased from 1570 MPa (227 ksi) to 1240 MPa (180 ksi) with both 10 percent elongation and reduction in area. After 700 hour 1095° C (2000° F) 103 MPa (15 ksi) exposure, the R.T. tensile strength was 104 MPa (151 ksi) and reduction in area was 19 percent. The stress rupture life at 1095° C (2000° F) was at least an order of magnitude improved over cast WAZ-20 placing the 8 percent W ODS alloy above all conventional cast alloys, but below the 17 percent W WAZ-D. Rupture ductility was 1 percent or less. Thus, decreasing the tungsten content decreased both tensile and stress rupture strengths and increased tensile but not rupture ductility.

#### CONCLUDING REMARKS

It was shown that a high  $\gamma'$  fraction alloy having a high gamma prime solvus temperature can be effectively dispersion strengthened. The strengths obtained were outstanding, especially at 1150° and 1205° C.

The strength was derived from the highly alloyed matrix, the elongated grain structure, and the hard phase dispersion all working in conjunction. The demonstrated tensile and rupture ductilities were low; however, it was shown that the tensile ductility can be improved by post-recrystallization heat treatment. It was also shown that the material could be strained at low stress at elevated temperature in the fine grained state, and then converted to a creep resistant microstructure. If this feature is applicable to other ODS alloys, it may provide a valuable tool for forming the complex shapes required for gas turbine hardware.

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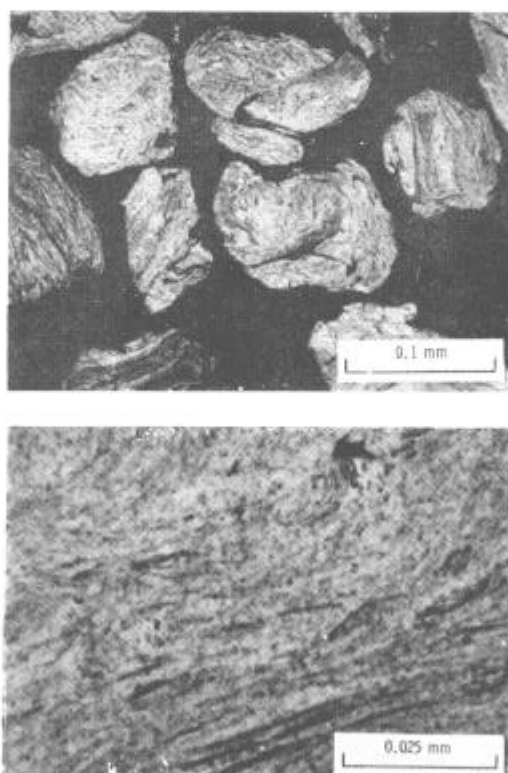


Figure 1. - Structure of mechanically alloyed WAZ-D powder.

TABLE 1. - COMPOSITION<sup>1</sup> (%) AND DENSITY (g/cc) OF ALLOYS

Alloy	Ni	W	Al	Fe	Zr	B	C	O	N	Y	$\rho$
WAZ-D (17W)	BAL	16.6	7.3	4.3	0.6 <sup>2</sup>	N.D. <sup>3</sup>	0.05	0.56 <sup>4</sup>	0.09	0.8	8.5
WAZ-D (8W)	BAL	7.7	6.8	4.5	0.5 <sup>2</sup>	N.D. <sup>3</sup>	0.08	0.51 <sup>4</sup>	0.02	0.8	8.2
WAZ-20 <sup>5</sup>	BAL	1.8	6.0	---	1.5	0.0003	0.11	-----	----	---	8.9

1. Analyses performed in house on extruded and heat treated bar.

2. Analysis believed in error; added Zr was 0.95%.

3. Not determined, 0.015 added.

4. For 3.5 mL/min O<sub>2</sub>, 40 hour runs.

5. Reference 4.

TABLE 2. - RESULTS OF TENSILE TESTS OF LARGE ELONGATED GRAIN ODS WAZ-D (17W) AND COMPARABLE DATA FOR CAST WAZ-20

Temperature, °C	Ultimate tensile strength MPa (ksi)	Reduction in area %	Elongation %
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## ODS-WAZ-D (17W)

Room temp.	1572	(228)	3	7
"	1565	(227)	3	5
"	1358	(197) <sup>1</sup>	7.7	7.3
"	1331	(193) <sup>2</sup>	9	11
760°	903	(131)	2	1
760°	917	(133)	5	4
1095°	290	(42)	2	2
1095°	303	(44)	0	5
1205°	221	(32)	2	3
1205°	186	(27)	0	3
1260°	124	(18)	0	1
1315°	80	(12)	0	1

## WAZ-20, conventionally cast (ref. 4)

Room temp.	745	(108)	N.R. <sup>3</sup>	4
760°	752	(109)	N.R. <sup>3</sup>	4
1095°	303	(44)	N.R. <sup>3</sup>	4
1205°	138	(20)	N.R. <sup>3</sup>	5

## WAZ-20, directionally solidified (ref. 4)

Room temp.	896	(130)	N.R.	13
760°	827	(120)	N.R.	4
1095°	303	(44)	N.R.	12
1205°	138	(20)	N.R.	8

1. Determined after 360 hour (20 ksi), 1095° C creep exposure.

2. Determined after 1220 hour (15 ksi), 1150° C creep exposure.

3. Not reported.

TABLE 3. - RESULTS OF VACUUM STRESS-RUPTURE TESTS OF ODS WAZ-D (17W)  
AND COMPARABLE DATA FOR CAST WAZ-20 TESTED IN AIR

Temp. °C	Microstructure <sup>1</sup>	Stress MPa (ksi)	Rupture life, hrs	R.A. %	Elong. %	Strain rate, sec <sup>-1</sup>
ODS-WAZ-D (17W), vacuum test						
760	B	572 (83)	9	0	1	N.D.
760	C	572 (83)	3	0	1	N.D.
1040	D	103 (15)	0	77	126	N.D.
1040	C	103 (15)	147	0		N.D.
1095	A	103 (15)	1125 <sup>2</sup>	N.A. <sup>3</sup>	N.A.	8.6·10 <sup>-10</sup>
1095	B	103 (15)	995	1	1	5.6·10 <sup>-10</sup>
1095	C	103 (15)	68	0	1	N.D.
1095	A	138 (20)	357 <sup>2</sup>	N.A.	N.A.	2.7·10 <sup>-9</sup>
1095	B	138 (20)	73	0	1	1.4·10 <sup>-8</sup>
1095	A	172 (25)	41	1	1	3.1·10 <sup>-8</sup>
1095	B	172 (25)	3	1	1	N.D.
1150	A	103 (15)	95 <sup>4</sup>	0	2	N.D.
1150	A	103 (15)	674	7.8	0	1.4·10 <sup>-9</sup>
1150	A	103 (15)	706	0	0	1.5·10 <sup>-9</sup>
1150	B	103 (15)	1273 <sup>2</sup>	N.A.	N.A.	1.3·10 <sup>-9</sup>
1205	A	103 (15)	95	5	1.4	2.0·10 <sup>-8</sup>
1205	A	103 (15)	112	6.4	3	8.5·10 <sup>-9</sup>
Conventionally cast WAZ-20, air test (ref. 4)						
1040	E	103 (15)	100	NR <sup>5</sup>	NR	NR
1095	E	103 (15)	20	NR	NR	NR
1150	E	103 (15)	5	NR	NR	NR
Directionally solidified WAZ-20, air test (ref. 4)						
760	F	572 (83)	75 <sup>6</sup>	3	4	NR
1040	F	103 (15)	190	NR	NR	NR
1095	F	103 (15)	42	NR	NR	NR
1150	F	103 (15)	4	NR	NR	NR

1. Microstructures:

- A. Recrystallized, large elongated, very few inclusions.
- B. Recrystallized, large elongated, some few inclusions, mainly stringers.
- C. Recrystallized, large elongated, many inclusions, mainly unprocessed particles.
- D. As extruded, fine grained, equiaxed.
- E. Conventionally cast, large equiaxed grains.
- F. Directionally solidified, large elongated grains.

2. Discontinued test

3. Not applicable

4. After 1031 hours at 103 MPa, 1095° C

5. Not reported

6. Single test, this work



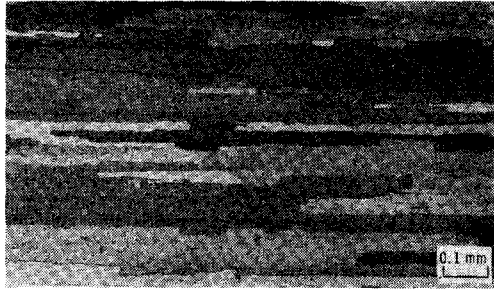


Figure 2. - Gradient annealed WAZ-D (17 W).

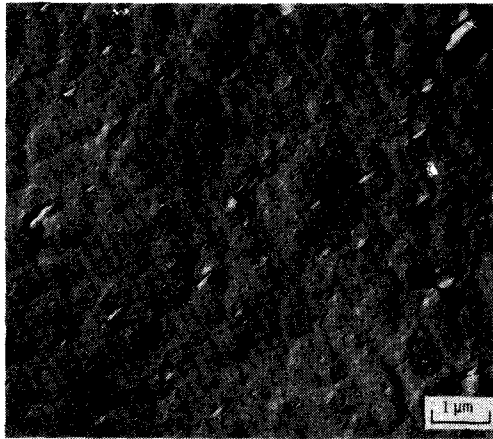


Figure 3. - Fine and coarse hard phase particles in WAZ-D after 1315°C (2400 ft) anneal. As ion etched.

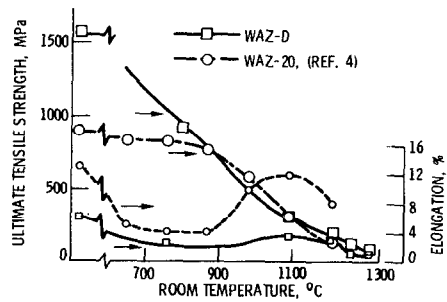


Figure 4. - Ultimate tensile strength and elongation of ODS WAZ-D and directionally solidified WAZ-20.

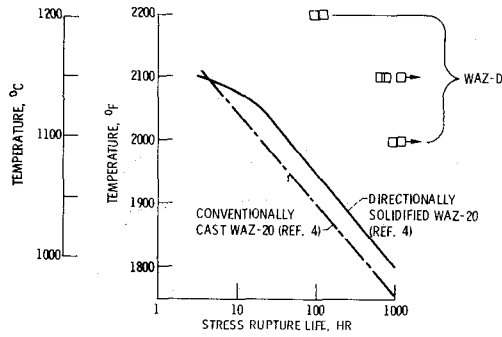


Figure 5. - Stress rupture lives of WAZ-D and WAZ-20.

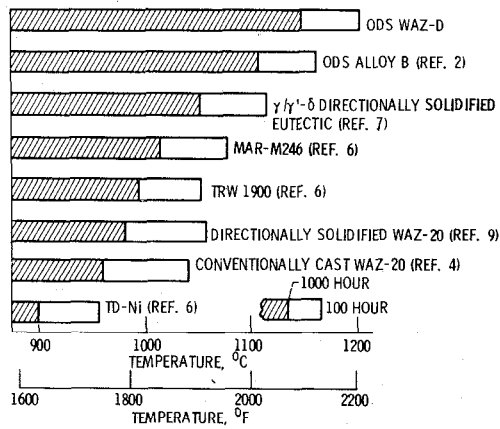


Figure 6. - Temperature to produce rupture at 103 MPa (15 ksi) in 100 and 1000 hours for advanced and conventional alloys.



Figure 7. - WAZ-D showing development of gamma prime plates and transgranular fracture after 100 hour, 1205°C, 103 MPa (15 ksi) rupture life.