

- Any LTI system is algebraically equivalent to the following form:

$$\begin{bmatrix} x_c \\ x_u \end{bmatrix} = \begin{bmatrix} A_c & A_{c2} \\ 0 & A_u \end{bmatrix} \begin{bmatrix} x_c \\ x_u \end{bmatrix} + \begin{bmatrix} B_c \\ 0 \end{bmatrix} u \quad x_c \in \mathbb{R}^{\bar{n}}, x_u \in \mathbb{R}^{n-\bar{n}}$$

$$u \in \mathbb{R}^k, y \in \mathbb{R}^m$$

$$y = [C_c \ C_u] \begin{bmatrix} x_c \\ x_u \end{bmatrix} + D u$$

\bar{n} = dim of controllable subsystem.

Def 14.1: The system (A, B) is ~~not~~ stabilizable if $n = \bar{n}$ or A_u is a stability matrix.

↙
So either:

- ① The system is fully controllable + A_u doesn't exist.
- ② A_u is a stability matrix and the states x_u will converge to zero exp. fast.

(~~Remember if (A, B) is controllable \rightarrow A_u is a stability matrix~~)
~~(A, B controllable \rightarrow A_u Hurwitz)~~

Eigenvector Test for Stabilizability:

Theorem 14.1 : $\dot{x} = Ax + Bu$ is stabilizable iff every e-vector of A^T corresponding to an e-value w/ positive or zero real part, is not in the kernel of B^T .

Discrete Systems : The D-LTI System is stabilizable iff every e-vector of A^T corresponding to an e-value w/ magnitude larger or equal to 1 is not in the kernel of B^T .

Proof : Use similarity transform to transform to controllable decomposition.

$$\bar{A} = \begin{bmatrix} A_c & A_{12} \\ 0 & A_u \end{bmatrix} = T^T A T \quad \bar{B} = \begin{bmatrix} \bar{B}_c \\ 0 \end{bmatrix} = T^{-1} B$$

① Show that if the ~~sys~~ (A, B) system is stabilizable then every "unstable" e-vector of A^T is not in the ~~ker~~ of B^T .

(prove by contradiction)

Assume (A, B) stabilizable \rightarrow Show unstable e-vector in null space of B^T .

i.e. $(\lambda, x) = \text{e-pair}$ s.t. $A^T x = \lambda x + B^T x = 0 \quad x \neq 0$

$$\Rightarrow (T^{-1} A T)^T x = \lambda x + (T^{-1} \bar{B})^T x = 0$$

$$\Rightarrow \begin{bmatrix} A_c^T & \\ A_{12}^T & A_u^T \end{bmatrix} \begin{bmatrix} x_c \\ x_u \end{bmatrix} = \lambda \begin{bmatrix} x_c \\ x_u \end{bmatrix} + \begin{bmatrix} \bar{B}_c^T & 0 \end{bmatrix} \begin{bmatrix} x_c \\ x_u \end{bmatrix} = 0$$

where $\begin{bmatrix} x_c \\ x_u \end{bmatrix} = T^{-1} x = \bar{x} \neq 0$

Since (A_c, B_c) - controllable +

$$A_c^T x_c = \lambda x_c \text{ and } B_c^T x_c = 0$$

Then $x_c = 0$ (not an e-vector)

From previous theorem,
if (A_c, B_c) is controllable
then can't have a e-vector
in $\ker B_c^T$

$\Rightarrow A_u^T x_u = \lambda x_u \Rightarrow \lambda$ is an unstable e-value
which contradicts stabilizability
assumption.

② Suppose system is not stabilizable

- Assume every unstable e-vector of A^T is not in
 $\ker B^T$

Not stabilizable $\Rightarrow \exists \lambda \in \text{RHP}$ s.t. $A_u^T x_u = \lambda x_u$
 $x_u \neq 0$

But

$$\begin{bmatrix} A_c^T & 0 \\ A_{12}^T & A_u^T \end{bmatrix} \begin{bmatrix} 0 \\ x_u \end{bmatrix} = \lambda \begin{bmatrix} 0 \\ x_u \end{bmatrix}$$

and $\bar{B}^T \begin{bmatrix} 0 \\ x_u \end{bmatrix} = [B_c^T \ 0] \begin{bmatrix} 0 \\ x_u \end{bmatrix} = 0$ for $\begin{bmatrix} 0 \\ x_u \end{bmatrix} \neq 0$

$\Rightarrow \begin{bmatrix} 0 \\ x_u \end{bmatrix}$ is an unstable e-vector of \bar{A}^T

and $\begin{bmatrix} 0 \\ x_u \end{bmatrix} \in \ker \bar{B}^T$

\rightarrow We found an "unstable" e-vector of \bar{A}^T in the
 \ker of \bar{B}^T so (\bar{A}, \bar{B}) cannot be stabilizable.

- And ~~$x = (-T)^{-1} \begin{bmatrix} 0 \\ x_u \end{bmatrix}$~~

\rightarrow contradiction since similarity transforms preserve controllability
properties.

PBT Test for Stabilizability:

Thm 14.2 -

1. The cont-time LTI system is stabilizable iff
 $\text{rank}[(A-\lambda I \ B)] = n \quad \forall \lambda \in \mathbb{C} : \text{Re}(\lambda) \geq 0$

→ Only difference is that we are looking @ the "unstable" e-values.

→ So, if unstable e-values are controllable, then we are okay.
The uncontrollable e-values will all be neg and go to zero
after some transient time. (notes)

Lyapunov Test for Stabilizability:

Thm 14.3 - The LTI system is stabilizable iff

there exists a pos-def P s.t.

$$AP + PA^T - BB^T < 0$$

↑ opposite sign from controllability Lyap test

- (A) Show that if P has pos-def solution then system is stabilizable.

(use e-vector test + assume $AP + PA^T - BB^T < 0$)
w/ P being pos-definite

Let $A^T x = \lambda x$ (where λ = unstable e-value)

$$\text{Then } x^*(AP + PA^T)x < x^*BB^Tx = \|B^Tx\|^2$$

$$(A^T x^*)^* Px + x^*PA^T x = \lambda^* x^* Px + \lambda x^* Px$$

$$= 2\text{Re}\{\lambda\}(x^*Px) = \|B^Tx\|^2$$

P is pos-def and $\text{Re}\{\lambda\} \geq 0$
(assumed unstable)

~~assume~~ ~~P~~

$$0 \leq 2\text{Re}\{\lambda\} x^*Px < \|B^Tx\|^2$$

$$\Rightarrow x \notin \ker B^T \quad (\text{because } B^Tx > 0)$$

∴ System is stabilizable (by e-vector test)

(B) Assume AB -LTI is stabilizable - show that P is pos-def solution.

- Let T be the similarity transform that creates controllable decomposition.

$$\bar{A} = \begin{bmatrix} A_c & A_{12} \\ 0 & A_u \end{bmatrix} = T^{-1}AT$$

$$\bar{B} = \begin{bmatrix} B_c \\ 0 \end{bmatrix} = T^{-1}B$$

(A_c, B_c) = controllable $\Rightarrow \exists P_c$ s.t.

$$A_c P_c + P_c A_c^T - B_c B_c^T = -Q_c < 0$$

(from feedback stabilization based off Lyap. test)

A_u - stable $\Rightarrow \exists P_u$ s.t.

$$A_u P_u + P_u A_u^T = -Q_u < 0 \quad (\text{std. Lyap. eq. for stabilit.})$$

Let $\bar{P} = \begin{bmatrix} P_c & 0 \\ 0 & \rho P_u \end{bmatrix}$ for $\rho > 0$

$\bar{A} \bar{P} + \bar{P} \bar{A}^T - \bar{B} \bar{B}^T$ (will show this equal ~~to~~ a neg. def. matrix)

$$= \begin{bmatrix} A_c & A_{12} \\ 0 & A_u \end{bmatrix} \begin{bmatrix} P_c & 0 \\ 0 & \rho P_u \end{bmatrix} + \begin{bmatrix} P_c & 0 \\ 0 & \rho P_u \end{bmatrix} \begin{bmatrix} A_c^T & 0 \\ A_{12}^T & A_u^T \end{bmatrix} - \begin{bmatrix} B_c \\ 0 \end{bmatrix} [B_c^T \ 0]$$

$$= \begin{bmatrix} A_c P_c & \rho A_{12} P_u \\ 0 & \rho A_u P_u \end{bmatrix} + \begin{bmatrix} P_c A_c^T & 0 \\ \rho P_u A_{12}^T & \rho P_u A_u^T \end{bmatrix} - \begin{bmatrix} B_c B_c^T & 0 \\ 0 & 0 \end{bmatrix}$$

$$\begin{aligned}
 &= \underbrace{\begin{bmatrix} A_c P_c + P_c A_c^T - B_c B_c^T \\ \rho P_u A_{12}^T \end{bmatrix}}_{-\rho Q_c} + \underbrace{\begin{bmatrix} \rho A_{12} P_u + P_c A_c^T \\ \rho A_u P_u + \rho P_u A_{12}^T \end{bmatrix}}_{-\rho Q_u}
 \end{aligned}$$

$$= \begin{bmatrix} -Q_c & \rho A_{1,2} P_u \\ \rho P_u A_{1,2}^T & \rho(-Q_u) \end{bmatrix} = - \begin{bmatrix} Q_c & -\rho A_{1,2} P_u \\ -\rho P_u A_{1,2}^T & \rho Q_u \end{bmatrix}$$

→ If ρ is small enough, this will be ~~be~~ neg-definite.

Feedback Stabilization Based on Lyapunov Test:

Assume $(AB - LTI)$ is stabilizable.

Then $AP + PA^T - BB^T < 0$ (from last ~~test~~ theorem)

~~Let $K = \frac{1}{2} B^T P^{-1}$~~

∴ $AP + PA^T - \frac{1}{2} BB^T P^{-1}P - \frac{1}{2} PP^T BB^T < 0$

$$(A - \frac{1}{2} BB^T P^{-1})P + P(A - \frac{1}{2} BB^T P^{-1})^T < 0$$

$$(A - BK)P + P(A - BK)^T < 0$$

w/ $K = \frac{1}{2} B^T P^{-1}$

$$\Rightarrow PP^T(A - BK)P + P(A - BK)^T P^{-1}P < 0$$

$$P \left[P^{-1}(A - BK) + (A - BK)^T P^{-1} \right] P < 0$$

Let $Q = P^{-1}$ + mult both right + left side by Q

$$\Rightarrow Q^T(A - BK) + (A - BK)^T Q < 0$$

* $(A - BK)$ is a stability matrix by Lyap. theorem.

* state feedback control $u = -Kx$ asymptotically stabilizes the system.

Thm 14.4 - When the ~~stabilizable~~ system (A, B) is stabilizable one always find ~~such~~ $u = -kx$ that makes closed loop system $\dot{x} = (A - BK)x$ asymptotically stable.

→ This is a necessary & sufficient condition.

(A, B) - stabilizable iff $\exists k$ s.t. $u = -kx$ makes $\dot{x} = (A - BK)x$ asymp. stable.

14.6 : Eigenvalue assignment:

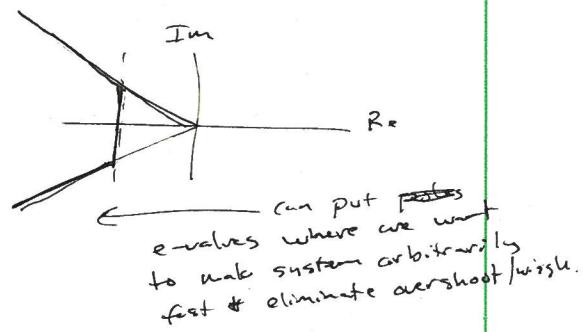
If $\dot{x} = Ax + Bu$ is controllable

Given any set of n complex #'s (appearing in complex conj. pairs), there exists a state feedback matrix K

s.t. the closed-loop system ~~system~~

$\dot{x} = (A - BK)x$ has eigenvalues equal to λ_i

(i.e. can put ^{eigenvalues} pretty much anywhere)



controllable

E-vector Test for controllability

- (A, B) is controllable iff there is no e-vector of A^T that is in the ker of B^T

Lyapunov Test for controllability

- A - stability matrix then (A, B) controllable iff \exists a unique, pos-def

Sol. $W = \int_0^\infty e^{At} B B^T e^{A^T t} dt$ to

eq. $AW + WAT = -B\Omega^T$

PBH Test for controllability

(A, B) is controllable iff $\text{rank}[A - \lambda I \ B] = n$
 $\forall \lambda \in \mathbb{C}$

stabilizable

Eigenvector Test for stabilizability

- (A, B) is stabilizable iff every e-vector of A^T corresponding to an e-value w/ $\text{Re}\{\lambda\} \geq 0$, is not in ker of B^T

Lyapunov Test for Stabilizability

- LTI system is stabilizable iff \exists a unique, pos-def solution P

~~P~~

to eq. $AP + P A^T - B B^T < 0$

PBH Test for Stabilizability

- The LTI system is stabilizable iff $\text{rank}[A - \lambda I \ B] = n$

$\forall \lambda \in \mathbb{C} : \text{Re}\{\lambda\} \geq 0$

↑ complex plane

→ only look at "unstable" e-values to see if they are controllable

Pole Placement

(A, B) - controllable $\Leftrightarrow A_{cc} = (A - BK)$ can have any "legal" set of eigenvalues.

$$P_{cc}(s) = \prod_{i=1}^n (s - \mu_i) = s^n + \gamma_{n-1}s^{n-1} + \dots + \gamma_1s + \gamma_0$$

Find k so that $\det(sI_n - A_{cc}) = \det(sI_n - A + BK) = P_{cc}(s)$

↓
desired closed-loop polynomial.

Assume a SISO system: (initially)

→ Put in controllable canonical form, if (A, B) is controllable can put in this form.

$$A = A_c = \begin{bmatrix} 0 & 1 & \dots & 0 \\ 0 & 0 & 1 & \vdots \\ \vdots & & \ddots & 1 \\ -\alpha_0 & -\alpha_1 & \dots & -\alpha_{n-1} \end{bmatrix}$$

char. poly. of A matrix

$$\text{where } \det(sI_n - A) = P_A(s) = s^n + \alpha_{n-1}s^{n-1} + \dots + \alpha_1s + \alpha_0$$

$$B = B_c = \begin{bmatrix} 0 \\ \vdots \\ 0 \\ 1 \end{bmatrix}$$

w/ single input the gain matrix k will be a row vector

$$k = [k_0 \ k_1 \ \dots \ k_{n-1}]$$

$$A_{cc} = A_c - B_c k = \begin{bmatrix} 0 & 1 & \dots & 0 \\ 0 & 0 & 1 & \ddots & \vdots \\ \vdots & & \ddots & 1 \\ -\alpha_0 & -\alpha_1 & \dots & -\alpha_{n-1} \end{bmatrix} - \underbrace{\begin{bmatrix} 0 \\ 0 \\ \vdots \\ 1 \end{bmatrix}}_{[k_0 \ k_1 \ \dots \ k_{n-1}]} =$$

$$\begin{bmatrix} 0 & \dots & 0 \\ \vdots & \ddots & \vdots \\ 0 & \dots & 0 \\ k_0 & k_1 & \dots & k_{n-1} \end{bmatrix}$$

$$A_{cc} = \begin{bmatrix} 0 & 1 & \cdots & 0 \\ 0 & 0 & 1 & \cdots & 0 \\ \vdots & & \ddots & & \vdots \\ 0 & & \cdots & 1 \\ -(\alpha_0 + k_0) & & \cdots & -(\alpha_{n-1} + k_{n-1}) \\ \alpha & -(\alpha_1 + k_1) \end{bmatrix}$$

→ This matrix is also in controller canonical form
 → This means char polynomial

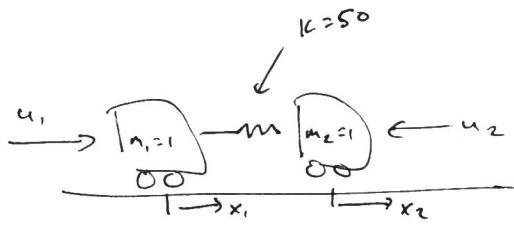
$$P_{A_{cc}}(s) = s^n + (\alpha_{n-1} + k_{n-1})s^{n-1} + \dots + (\alpha_1 + k_1)s + (\alpha_0 + k_0)$$

choose:

$$\begin{cases} \gamma_i = \alpha_i + k_i \\ k_i = \gamma_i - \alpha_i \end{cases}$$

↑
What we want the e-value to be — coefficients of the desired polynomial.

↑
↑
Subtract off part don't want.
Put in part want.



→ want poles at $(-10, -10, 5\sqrt{2}(-1 \pm j))$

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_4 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -50 & 50 & 0 & 0 \\ 50 & -50 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 1 & 0 \\ 0 & -1 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \end{bmatrix}$$

→ controllability matrix $\text{rank}(C) = 4$

→ Put into Multiple Input Controller Form

$$A_c = \left(\begin{array}{cc|cc} 0 & 1 & 0 & 0 \\ -50 & 0 & -50 & 0 \\ \hline 0 & 0 & 0 & 1 \\ -50 & 0 & -50 & 0 \end{array} \right) \quad B_c = \left(\begin{array}{cc} 0 & 0 \\ 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{array} \right)$$

$$K = \begin{bmatrix} K_{11} & K_{12} & K_{13} & K_{14} \\ K_{21} & K_{22} & K_{23} & K_{24} \end{bmatrix}$$

$$A_c + B_c K_c = \left(\begin{array}{cc|cc} 0 & 1 & 0 & 0 \\ -50 + K_{11} & K_{12} & -50 + K_{13} & K_{14} \\ \hline 0 & 0 & 0 & 1 \\ -50 + K_{21} & K_{22} & -50 + K_{23} & K_{24} \end{array} \right) \quad \rightarrow \text{single block structure}$$

$K_{11} = 50$
 $K_{13} = 51$
 $K_{12} = K_{14} = 0$

$$A_c + B_c K_c = \left(\begin{array}{cccc} 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -50 + K_{21} & K_{22} & -50 + K_{23} & K_{24} \end{array} \right)$$

desired char eq:

$$P_{des}(s) = s^4 + 34s^2 + 483s^2 + 3414s + 10000$$

→ current char eq.

$$P_{act}(s) = s^4 - k_{24}s^3 + (+50 - k_{23})s^2 - k_{22}s + (+50 - k_{21})$$

$$k_{21} = -9950$$

~~$$k_{22} = 3414$$~~

$$k_{22} = -3414$$

$$k_{23} = -433$$

$$k_{24} = -34$$