

①

$$\dot{\omega} \xrightarrow{SV} J \dot{\omega} + \omega [J \omega]_x = \sum_i B M_i$$

$$\dot{R} = R \cdot [\omega]_x$$

$$R = C_{IB}$$

$$\dot{R} = C_{EI}$$

$$[a]_x = \begin{pmatrix} 0 & -a_z & a_y \\ a_z & 0 & -a_x \\ -a_y & a_x & 0 \end{pmatrix}$$

$$\begin{aligned} \dot{I} \dot{V} &= \sum_i I F_i \\ B \dot{V} &= m \cdot \dot{B} V + \omega \cdot [m \cdot B V]_x = \sum_i B F_i \\ I \dot{P} &= I V = C_{IE} E V \end{aligned}$$

Bx : x w.r.t (E): body
 Ix : x w.r.t (I): inertial
 (E): earth

J : inertia
 ω : angular rate
 M : Torque
 R : rotation matrix
 m : mass
 F : force
 v : velocity
 p : position

$$\frac{1}{m} \sum_i B F_i = \underbrace{C_{BI} \cdot g}_{\text{gravity w.r.t body}} - \underbrace{\begin{pmatrix} k_x & k_y & 0 \\ 0 & k_x & k_y \end{pmatrix}}_{\text{drag (normalized with m)}} \cdot E V + \underbrace{\begin{pmatrix} 0 \\ 0 \\ \frac{T}{m} \end{pmatrix}}_{\text{thrust}}$$

(no wind)

Acc measurement (IMU), no noise, no bias, no wind

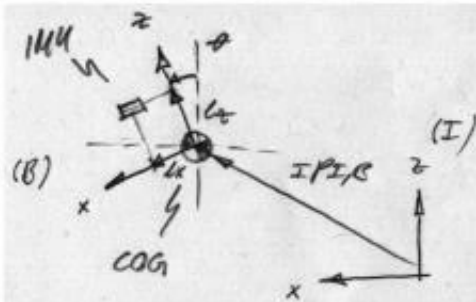
$$B a = \frac{1}{m} \sum_i B F_i - C_{BI} \cdot g = \begin{pmatrix} -k_x \cdot B v_x \\ -k_y \cdot B v_y \\ \frac{T}{m} \end{pmatrix} \quad k_z = 0$$

acceleration measured from the quad w.r.t body frame

gravity gets canceled, while hovering we measure $\frac{T}{m} \approx 9.81 \text{ m/s}^2$

2D (x, z)

$$B a = \begin{pmatrix} -k_x \cdot B v_x \\ \frac{T}{m} \end{pmatrix}$$



$$I P_{I, IMU} = I P_{I, B} + I P_{B, IMU}$$

$$= \begin{pmatrix} x \\ z \end{pmatrix} + C_{EB} \cdot \begin{pmatrix} l_x \\ l_z \end{pmatrix}$$

$$C_{EB} = R(\theta) = \begin{pmatrix} \cos \theta & \sin \theta \\ -\sin \theta & \cos \theta \end{pmatrix}$$

offset of IMU wrt. to COG w.r.t. body

$$= \begin{pmatrix} x \\ z \end{pmatrix} + \begin{pmatrix} l_x \cos \theta + l_z \sin \theta \\ -l_x \sin \theta + l_z \cos \theta \end{pmatrix}$$

$$\dot{I P}_{I, IMU} = \begin{pmatrix} \dot{x} \\ \dot{z} \end{pmatrix} + \begin{pmatrix} -l_x \sin \theta \dot{\theta} + l_z \cos \theta \dot{\theta} \\ -l_x \cos \theta \dot{\theta} - l_z \sin \theta \dot{\theta} \end{pmatrix}$$

$$\ddot{I P}_{I, IMU} = \begin{pmatrix} \ddot{x} \\ \ddot{z} \end{pmatrix} + \begin{pmatrix} -l_x \cos \theta \dot{\theta}^2 - l_z \sin \theta \dot{\theta}^2 & -l_x \sin \theta \ddot{\theta} + l_z \cos \theta \ddot{\theta} \\ l_x \sin \theta \dot{\theta}^2 - l_z \cos \theta \dot{\theta}^2 & -l_x \cos \theta \ddot{\theta} - l_z \sin \theta \ddot{\theta} \end{pmatrix}$$

$$= \begin{pmatrix} \ddot{x} \\ \ddot{z} \end{pmatrix} + \begin{pmatrix} \cos \theta & \sin \theta \\ -\sin \theta & \cos \theta \end{pmatrix} \cdot \begin{pmatrix} -l_x \\ -l_z \end{pmatrix} \cdot \ddot{\theta}^2 + \begin{pmatrix} \cos \theta & \sin \theta \\ -\sin \theta & \cos \theta \end{pmatrix} \begin{pmatrix} l_z \\ -l_x \end{pmatrix} \cdot \ddot{\theta}$$

$$= \begin{pmatrix} \ddot{x} \\ \ddot{z} \end{pmatrix} + C_{EB} \cdot \left(\begin{pmatrix} -l_x \\ -l_z \end{pmatrix} \ddot{\theta}^2 + \begin{pmatrix} l_z \\ -l_x \end{pmatrix} \ddot{\theta} \right)$$

→ Acc measurement (IMU), no noise, no bias, no wind
+ offset to COG

this is acceleration due to the offset w.r.t. body

$$B \gamma a = \begin{pmatrix} -l_x \cdot \ddot{\theta}^2 - l_z \ddot{\theta} \\ \frac{T}{m} - l_z \ddot{\theta}^2 - l_x \ddot{\theta} \end{pmatrix}$$