



#### Chapter 5

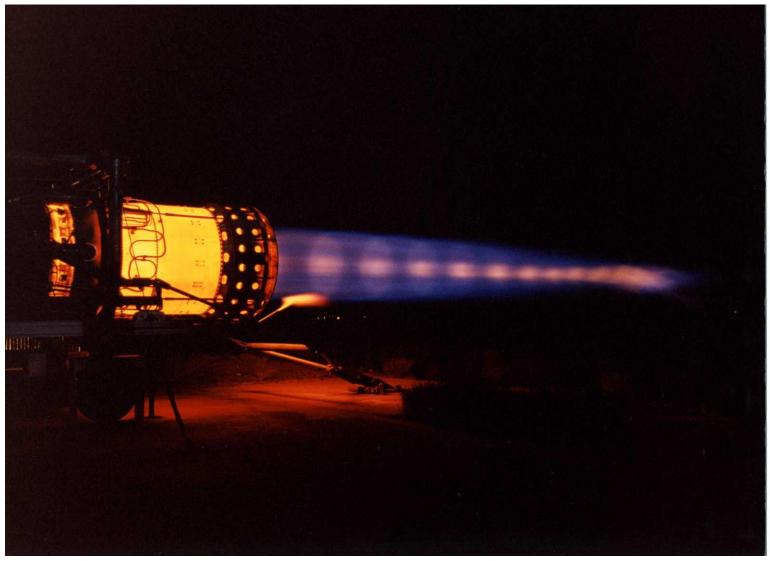
## The Turbojet Engine with an Afterburner

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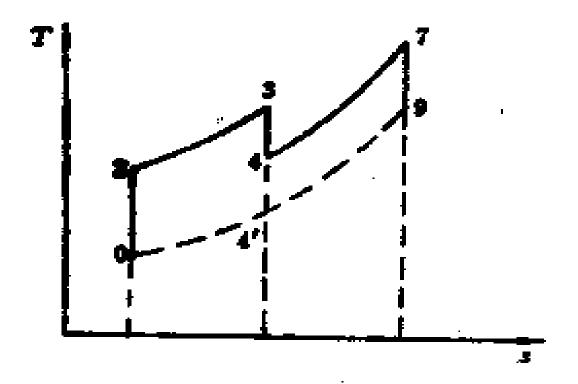
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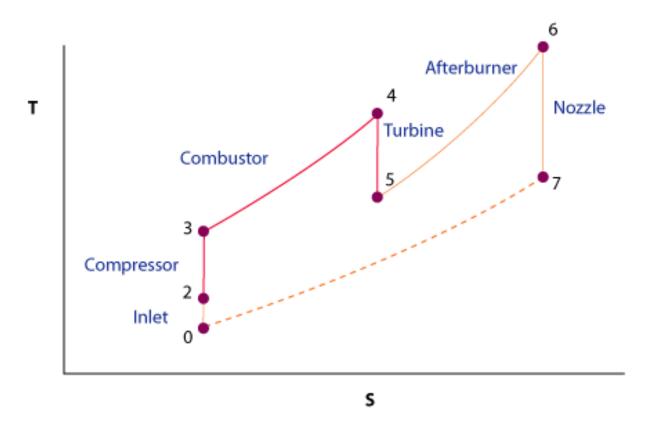
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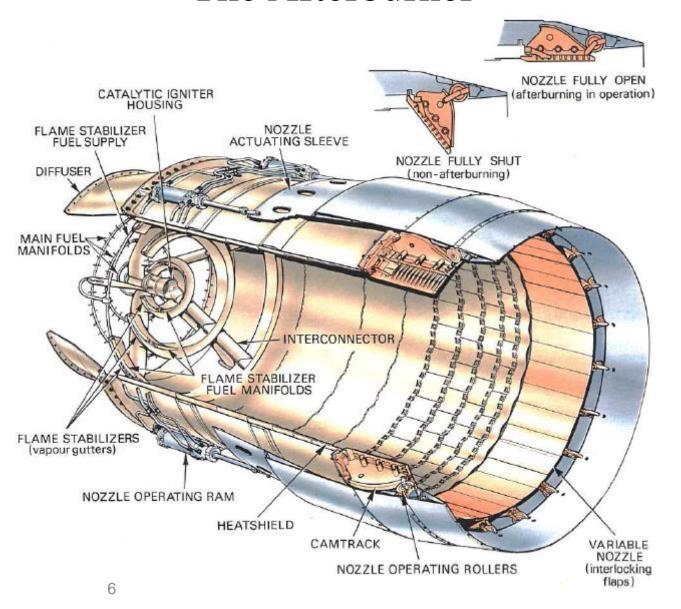




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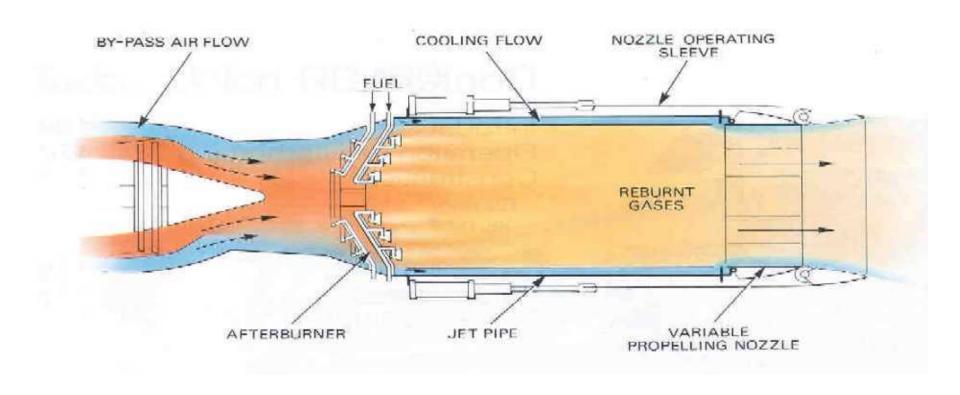


#### Basic requirements of an Afterburner

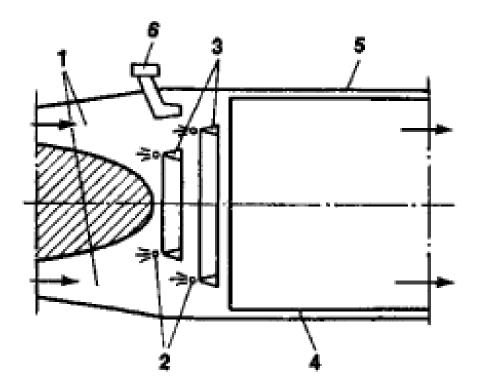
- ➤ Reliable ignition
  - ➤ Compared to main CC, advantage of A/B is high *T* (better for chemical reaction), disadvantages are low total pressure, high flow speed, and less oxygen.
  - ➤ In general, it's still easier to ignite except at high altitude.
- ➤ Complete combustion
  - ➤ Because of low total pressure, high velocity, and high fuel flow, burning efficiency is lower (0.85~0.90).
  - Especially at high altitudes, it decreases significantly even under rich fuel conditions.
- ➤ No oscillation
  - ➤ Burning in afterburner might be oscillating, with low frequency of 50~200Hz. This might cause flameout or breaking of parts.
  - ➤ Heat shields are used for structural protection.

#### **Basic requirements of an Afterburner**

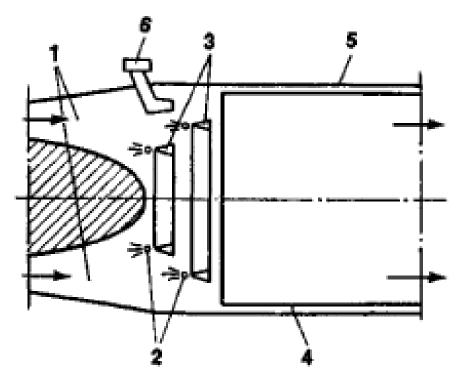
- > Low total pressure losses
  - ➤ Although velocity decreases in the diffuser, it is still quite high.
  - Afterburner does not work during most part of the flight time. It causes drag for air flow (due to its length, flame stabilizers, fuel manifolds).



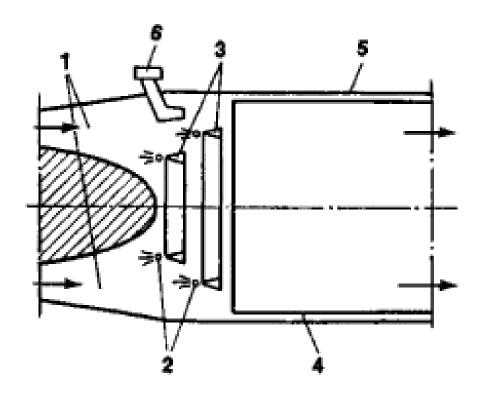
- ➤ Diffuser
  - Reduce flow speed from 400 m/s to 150m/s.
  - Redress the core flow to the axial direction.



- > Fuel injection and vaporization
  - > Injection direction is against flow direction.
  - Fuel supply can be divided in 2-3 zones.

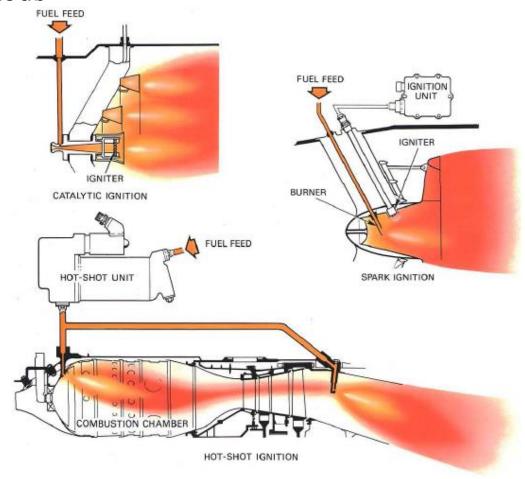


- > Ignition methods
  - > Spark
  - ➤ Catalytic
  - ➤ Hot-shot



## **Afterburning Process and Main Parts**

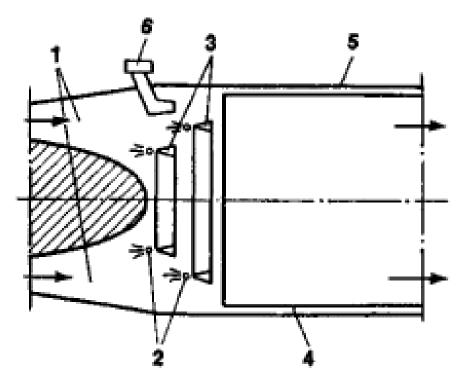
> Ignition methods



- > Flame stabilizers
  - Form V (Radial, circumferential).
  - ➤ Sand dune (Chinese invention).



End-on view of the V- gutter



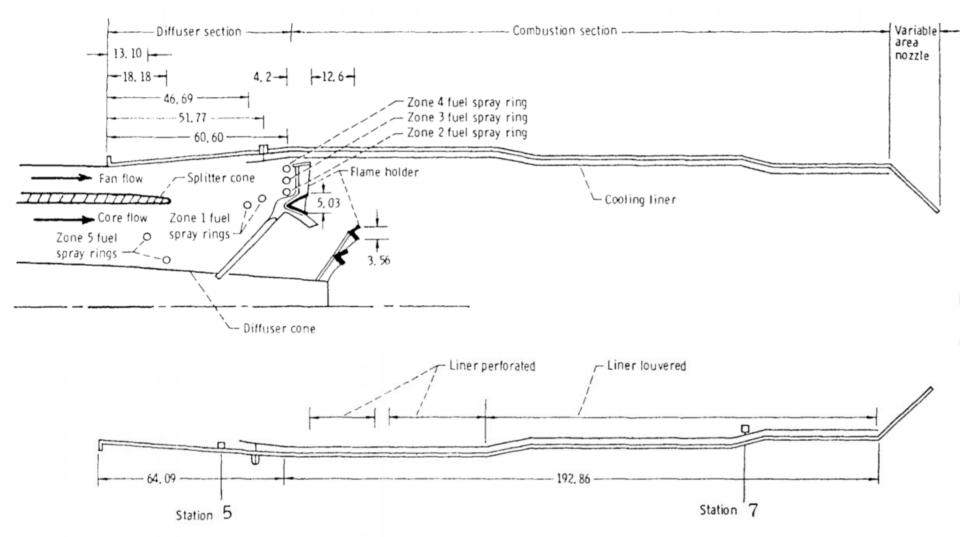


Fig. 2.38 Afterburner for TF30-P-3 augmented turbofan engine [all dimensions are in centimeters (from Ref. 29)].



