



Chapter 5

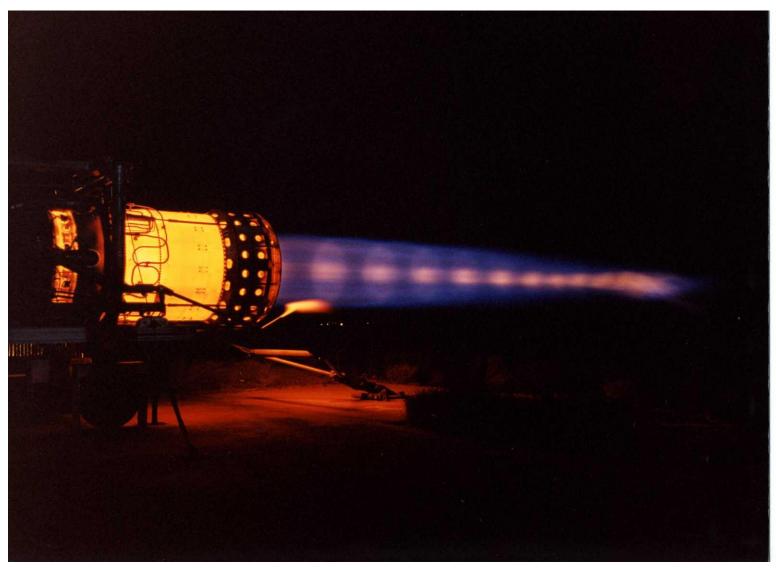
The Turbojet Engine with an Afterburner

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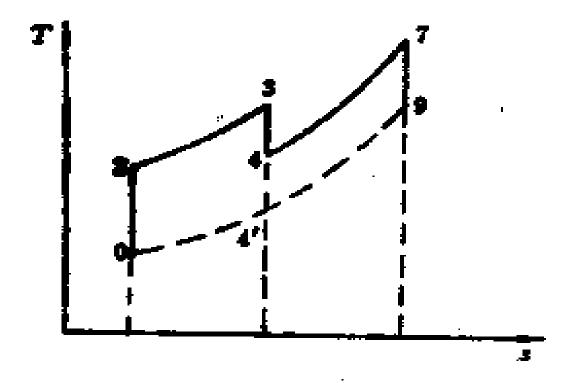
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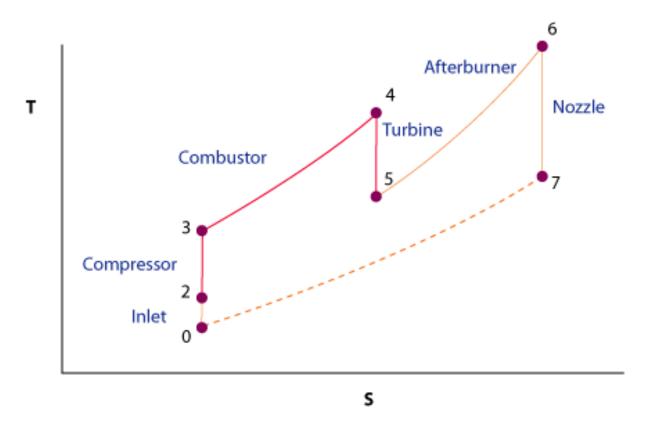
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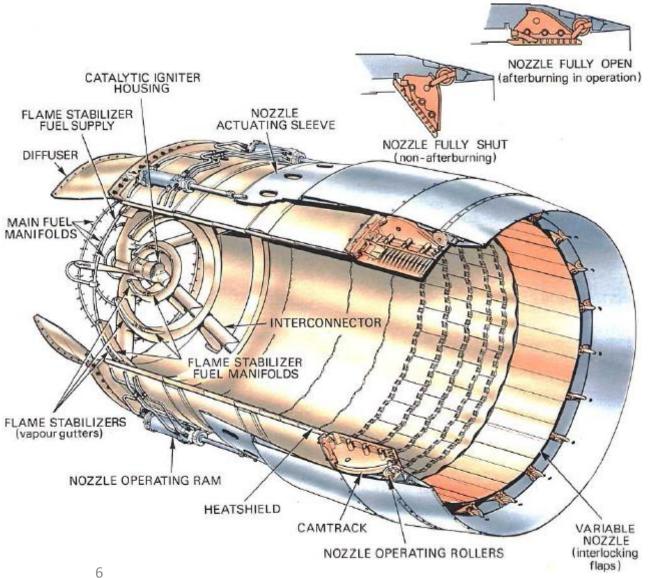




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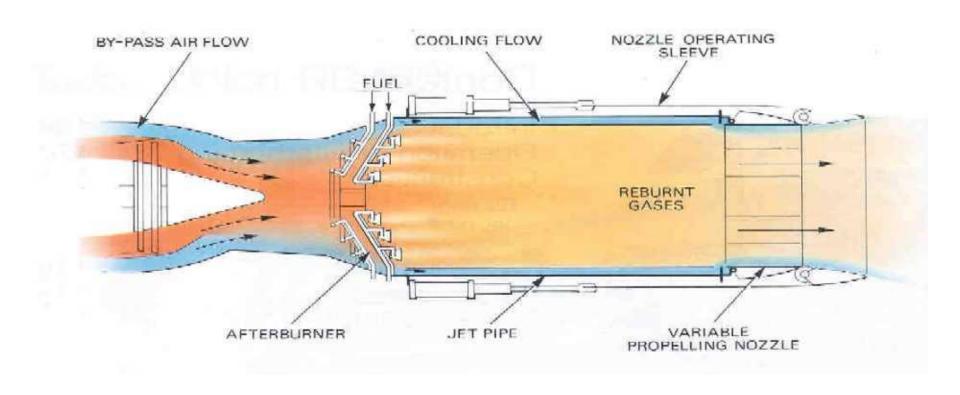


Basic requirements of an Afterburner

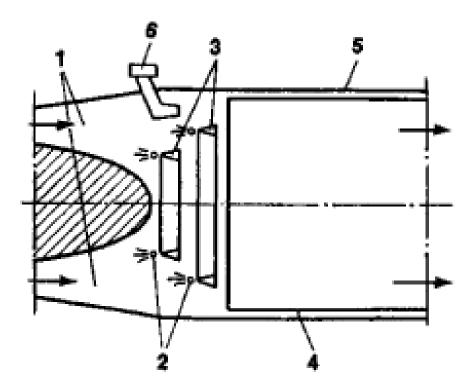
- ➤ Reliable ignition
 - ➤ Compared to main CC, advantage of A/B is high *T* (better for chemical reaction), disadvantages are low total pressure, high flow speed, and less oxygen.
 - ➤ In general, it's still easier to ignite except at high altitude.
- Complete combustion
 - ➤ Because of low total pressure, high velocity, and high fuel flow, burning efficiency is lower (0.85~0.90).
 - Especially at high altitudes, it decreases significantly even under rich fuel conditions.
- ➤ No oscillation
 - ➤ Burning in afterburner might be oscillating, with low frequency of 50~200Hz. This might cause flameout or breaking of parts.
 - ➤ Heat shields are used for structural protection.

Basic requirements of an Afterburner

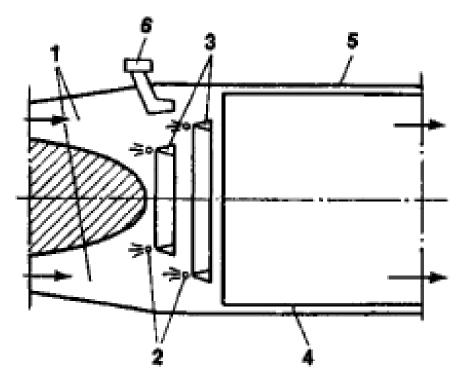
- > Low total pressure losses
 - ➤ Although velocity decreases in the diffuser, it is still quite high.
 - Afterburner does not work during most part of the flight time. It causes drag for air flow (due to its length, flame stabilizers, fuel manifolds).



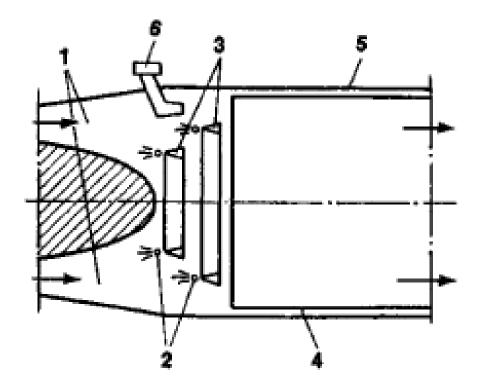
- ➤ Diffuser
 - Reduce flow speed from 400 m/s to 150m/s.
 - > Redress the core flow to the axial direction.



- > Fuel injection and vaporization
 - > Injection direction is against flow direction.
 - Fuel supply can be divided in 2-3 zones.

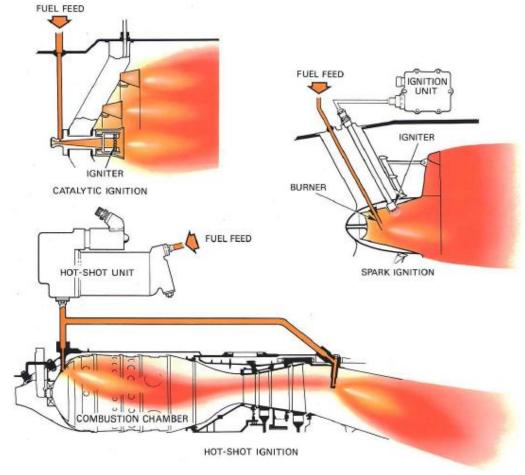


- > Ignition methods
 - > Spark
 - ➤ Catalytic
 - ➤ Hot-shot



Afterburning Process and Main Parts

> Ignition methods



- > Flame stabilizers
 - Form V (Radial, circumferential).
 - > Sand dune (Chinese invention).

