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1) TIME OF EXPOSURE - 720 seconds

MATERIAL USED

Density = 1600 kg/m3 Thermal Conductivity = 1-10 W/mR

> & So, thermal conductivity of K=1 [chosen]

Cp = 500 Kg/m3 Melting Point = 2200 K

Max temperature < 2200 K.

The Material above is "Re-inforced Carbon Carbon" It is chosen because, we have conditions of Tambient = 310 K on beckerd & On the exposed Side we have the surface radiating to tempoj OK, => Outer-Space is Assumed.

This MATERIAL !

These conditions, suggest that it might be used for Heat Sheild of "Re-entry vehicle"

<=1-25χ10-6

For Heat Shield, the best moterial is RCC as petter material Li 900 cas used, the maximum temperature exceeded [2100°C Cappear) its Melting Temp[1264°C] Assumption: Let thickness is 10 cm, discretization into lo division; DC=0.01m XDt < = = time step < 40 seconds dt = 1 cessumed Hu Boundary Condition at exposed end $O(\sqrt{t}) - \varepsilon \sigma [T(0,t)]^{4} = \rho c \rho \frac{dT}{dt} - \kappa \frac{dT}{dt}$ (2) T(x=L) = T(x=L-dx) (All times) Initial Condition => T(2,0) = 300K

Governing Eq => fry = 1 du ; u=temp=T Applying Euler FTCS, Scheme

We get

After Simulation, we found that this thickness is optimum as the back end temp doesn't exceed 310 K.

It is between 307K < Tags 3lok

10 cm -> 5cm, 2.5 cm, 1.25 cm, 1 cm, 0.5 cm, 0.4 cm, 0-25 cm elements 2, 4, 8, 10, 20, 25, 40 From the plot estructed, we see, that grid independence is exchieved for 20 division, dx = 0.5 cm = 6.005m fa our longten of 10 cm.