# High Performance Computing for Aerospace Engineering

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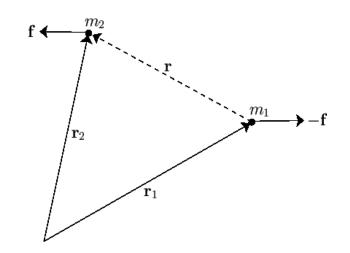
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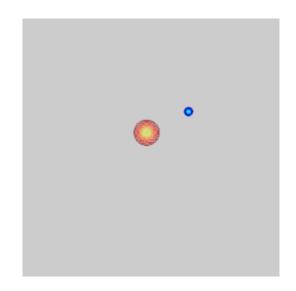
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#### Introduction

- The **N-body problem** is one of the most famous problems in mathematical physics, with its first complete mathematical formulation dating back to Newton's Principia [1].
- The N-Body Problem is the problem of predicting the motion of a system of N particles over time when the only forces with which they interact are the forces described in Newton's Law of Universal Gravitation.
- Currently, the *two body problem* (N = 2) has already been solved analytically and the solution is exact, in fact, they are conic sections.
- Nevertheless, until today, the analytical solution remains unsolved for N>2. The equations cannot be solved analytically and we must look at numerical solutions.
- For instance, the three body problem has been described as chaotic.



Forces in a two body problem [2].



The evolution of two bodies interaction [3].

<sup>[1]</sup> Aarseth, S. (2003). Gravitational N-Body Simulations: Tools and Algorithms (Cambridge Monographs on Mathematical Physics). Cambridge: Cambridge University Press. doi:10.1017/CBO9780511535246

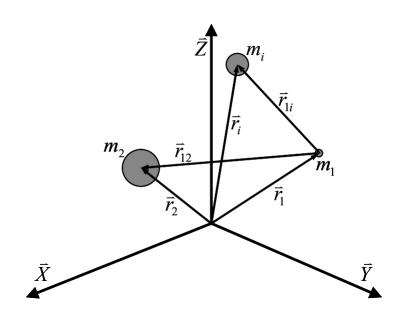
<sup>[2]</sup> Fitzpatrick, R. (2016).Two-body problem. <a href="https://farside.ph.utexas.edu/teaching/celestial/Celestial/node11.html">https://farside.ph.utexas.edu/teaching/celestial/Celestial/node11.html</a> (accessed: 11-04-2021)

## Introduction (cont.)

- A <u>chaotic system</u> can be described as a system that are deterministic for an initial state, meaning that no randomness is involved in the development of future states of the system but are highly sensitive on initial conditions [1].
- It is actually not possible to write down all the terms of a general formula that describes the motion of three or more gravitational objects. The issue lies in how many unknown variables a N body system contains.
- For N>2, there are more unknowns than equations describing them.

## Introduction (cont.)

- The **N-Body problem** can be stated as:
- "Given n bodies with masses and initial position and velocities, how will they evolve over time under gravitational interaction?"



Forces on a 3 Body system [1].

- For N = 3, solutions exist in special cases (i.e. restricted three-body problem).
- In general, numerical methods must be used to simulate such systems.

## Introduction (cont.)

• Given N bodies with an initial position  $\overrightarrow{x_i}$  and velocity  $v_i$  for  $1 \le i \le N$ , the force vector  $\overrightarrow{f_{ij}}$  on body i caused by its gravitational attraction to body j is given by the following [1]:

$$\overrightarrow{f_{ij}} = G \frac{m_i m_j}{|\overrightarrow{r_{ij}}|^2} \cdot \frac{\overrightarrow{r_{ij}}}{|\overrightarrow{r_{ij}}|}$$

- Where  $m_i$  and  $m_j$  are the masses of the bodies i and j respectively.
- Where  $\vec{r}_{ij} = \vec{x}_i \vec{x}_i$  is the vector from the body and G is the gravitational constant.

$$\vec{F}_i = \sum_{\substack{1 \le j \le N \\ j \ne i}} \overrightarrow{f_{ij}} = Gm_i \cdot \sum_{\substack{1 \le j \le N \\ j \ne i}} \frac{m_j \vec{r}_{ij}}{|\vec{r}_{ij}|^3}$$

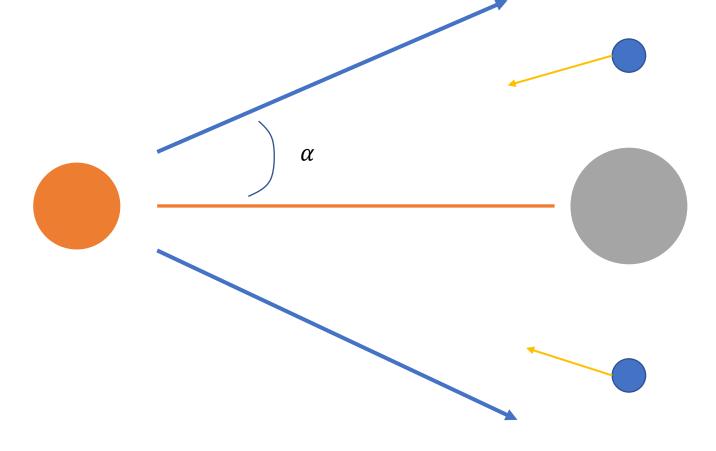
For N = 5, the relationships between particles follows the subsequent pattern:

12 23 34 45
13 24 35
14 25
15

Relations: 
$$\frac{N(N-1)}{2}$$

#### **Centre of Mass**

$$\overrightarrow{f_{ij}} = G \frac{m_i m_j}{|\vec{r}_{ij}|^2}$$



• Using the centre of mass of a group of planets was firstly considered to reduce the amount of communications between cores. The idea was demonstrated to induce little error only for  $\alpha \approx 0$  because of the distance values for each  $\overrightarrow{f_{ij}}$  and it was therefore discarded.

#### What have we programmed?

- 1. Adapted the Runge Kutta's method from C++ to C.
- 2. Programmed the Twobody() problem in series in C.
- 3. Programmed the Twobody() problem in parallel in C.
- 4. Programmed the Nbody() problem in series using All Pairs algorithm.
- 5. Programmed the Nbody() problem in parallel using All Pairs algorithm.
- 6. The resultant plot animations in MATLAB.

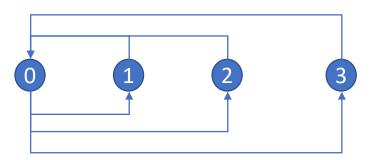
#### **All Pairs Algorithm**

- The **All Pairs algorithm** is one of the simplest N-body computation. It is known as the direct sum or brute force method and it involves calculating the force between each pair of elements and then adding up the result forces on each element [1].
- The computation has two steps:
  - 1. Compute the forces on each element
  - 2. Move each element a bit based on this force, and then repeat.
- Since each object computes the forces on it from each other object (N objects). Thus, the work complexity of this problem is a  $N^2$  algorithm. Each object must compute its interaction with each other object so each of N-objects has to compute N-1 forces.

$$N_{particles} \cdot (N-1)_{\underline{forces}} \approx O(N^2)$$

#### **All Pairs Algorithm**

 The pseudocode is shown here:

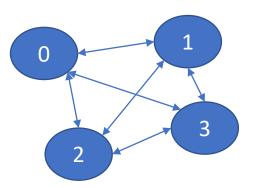


#### Algorithm 1 Parallelization

```
1: Initialization;
 2: Define the number of bodies (NBODY) and their parameters (mass and
   position)
 3: worksplit function
 4: Define a global variable containing myend for each processor
   (planet2proc)
 5: Data calculations (velocity and period for each body)
 6: Declare integration bounds and initial solution
 7: Define global initial conditions for all particles
 8: Set the Runge-Kutta parameters
 9: procedure Launch the integrator
       for Each time step do
          All processors calculate its e
11:
          all reduce function to calculate the minimum e
12:
          if proc! = 0 then
13:
             Send positions to master processor (proc == 0)
14:
          else Receive local position from other processors
15:
             Assembly the global position vector
16:
             Send the global position vector to other processors
17:
          end if
18:
          for Each planet of the actual processor do
19:
             Initialize velocities and accelerations
20:
21:
             for For all relations with other particles do
                 Ignore the planet itself
                 Calculate distances
23:
                 Calculate accelerations
24:
             end for
25:
             Calculate velocities
26:
          end for
27:
       end for
29: end procedure
30: Print error and steps
31: Save positions and velocities as a function of time in an array (.csv)
32: Finalization
```

## **All Pairs Algorithm**

 Another parallelization method:



```
Algorithm 2 Improved parallelization

    Initialization;

    Define the number of bodies (NBODY) and their parameters (mass and

3: worksplit function
4: Define a global variable containing myend for each processor
   (planet2proc)
5: Data calculations (velocity and period for each body)
6: Declare integration bounds and initial solution
7: Define global initial conditions for all particles
 8: Set the Runge-Kutta parameters
9: procedure Launch the integrator
       for Each time step do
           All processors calculate its e
11:
          all reduce function to calculate the minimum e
12:
          Each processor creates its own position vector and assigns within
   the global position vector
          for The actual processor (i) to the last processor do
14:
              for All processors (j) do
15:
                 Ignore the processor itself
16:
                 if i < j then
17:
                    Send positions to forward processors
18:
19:
                    Receive positions from forward processors and assem-
   bly the global position vector
                 else
                     Receive positions from preceding processors and assem-
21:
   bly the global position vector
                     Send positions to preceding processors
                 end if
              end for
24:
             for Each planet of the actual processor do
                 Initialize velocities and accelerations
                 for For all relations with other particles do
27:
                     Ignore the planet itself
                     Calculate distances
                     Calculate accelerations
30:
                 end for
31:
                 Calculate velocities
              end for
          end for
       end for
36: end procedure
37: Print error and steps
38: Save positions and velocities as a function of time in an array (.csv)
39: Finalization
```

20:

22:

23:

25:

26:

28:

29:

32:

33:

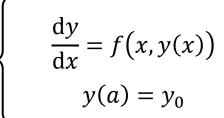
34:

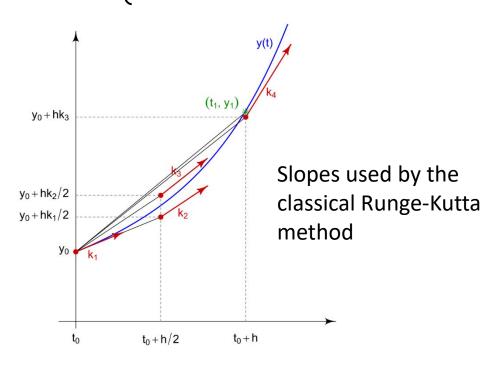
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## 4th order Runge Kutta numerical method

- Runge Kutta is a iterative method which includes the Euler method used in temporal discretization for the approximate solutions of ordinary differential equations.
- Considering a differential equation with its initial condition:

$$\begin{cases} k_1^n = f(x_n, y_n) \\ k_2^n = f\left(x_n + \frac{h}{2}, y_n + \frac{h}{2}k_1^n\right) \\ k_3^n = f\left(x_n + \frac{h}{2}, y_n + \frac{h}{2}k_2^n\right) \\ k_4^n = f(x_n + h, y_n + hk_3^n) \\ y_{i+1}^n = y_i^n + \frac{1}{6}(k_1 + 2k_2 + 2k_3 + k_4) \end{cases}$$





# 12<sup>th</sup> order Runge Kutta numerical method by Hiroshi Ono

- The Hiroshi Ono Runge Kutta method adapts the classical Runge Kutta but with a 12<sup>th</sup> order approximation. [1] [2]
- Besides, the advantage of this method is that it uses an adaptive step size depending on function and increasing its speed.
- An optimal way to implement the code is to represent the method by using the buthcler's table which encompasses all the RK coefficients [2]:

Optimal dtImportant region

Adaptive dt

If the method is explicit, by the simplified tableau



#### **Parallelization**

- What if we divide the calculation into parts? Here's an overview:
- Each processor is responsible for N/cores part of the velocity and position update using a worksplit().
- Master core reads initial data and broadcasts the local initial conditions to the other cores taking into account the particles within the core.
- Each processor is in charge of running the RK for all the particles of its core.
- Each processor has a different dt. Consider two processors:

- We have 2 routines:
  - A integration temporal routine (Runge Kutta)
  - The computation of the N Body
- Using MPI\_ALLGATHER, the valid dt is selected by setting the minimum of dt to all processors.
- Afterwards, the processors calculates the position and velocities of all its particles.
- Each processor receives the updated positions and updates its conditions.
- Finally, each processor creates an output.csv with the positions and velocities of its particles.
- In order to view the results, the output file is ported to MATLAB so as to plot the results

- We have 3 files:
- main\_NBP\_MPI.c which is the main file.
- *Nbodylib.h* which contains all the *#define* and *MACROS*.
- *Nbodylib.c* which contains the functions *worksplit()* and *Nbody()*.

```
// Main
int main(int argc, char* argv[]) {
    int r; // For error checking
    // Start MPI
    r = MPI Init(&arac, &arav);
    checkr(r, "Initiate");
    MPI Status st;
    // Check rank and size of the processors
    int rank, size;
    rank = proc();
    size = nproc();
    array size = size;
    planet2proc = (int*)malloc(sizeof(int)*array size);
    // If the processor is higher than the number of processors, exit
   if (rank >= size) {exit(-1);}
    // Declare worksplit variables
    int mystart,mylocstart;
    int myend, mylocend;
    int bodies;
    for (int xproc=0; xproc<size; xproc++) {</pre>
     worksplit(&mystart, &myend, xproc, size, 0, NBODY - 1);
      if (xproc == rank) {
       bodies = myend - mystart + 1;
       mylocstart = mystart;
        mylocend = myend;
      planet2proc[xproc] = myend;
```

```
// Data calculations
   double V1 = Vx(M0,R1); // Mercury velocity
  double T1 = Tx(M0,R1); // Mercury period
  double V2 = Vx(M0,R2); // Venus velocity
  double T2 = Tx(M0,R2); // Venus period
  double V3 = Vx(M0,R3); // Earth velocity
  double T3 = Tx(M0,R3); // Earth period
  double V4 = Vx(M0,R4); // Mars velocity
  double T4 = Tx(M0,R4); // Mars period
  double V5 = Vx(M0,R5); // Jupiter velocity
  double T5 = Tx(M0,R5); // Jupiter period
  double V6 = Vx(M0,R6); // Saturn velocity
  double T6 = Tx(M0,R6); // Saturn period
  double V7 = Vx(M0,R7); // Uranus velocity
  double T7 = Tx(M0,R7); // Uranus period
  double V8 = Vx(M0,R8); // Neptune velocity
  double T8 = Tx(M0,R8); // Neptune period
  double V9 = Vx(M0,R9); // Pluto velocity
  double T9 = Tx(M0,R9); // Pluto period
// Integration bounds and initial solution
double tspan[2] = \{0.,T9\};
// All initial conditions
double *y0 loc;
v0 loc = (double *)malloc(bodies*6*sizeof(double));
// Global initial conditions for all particles
for(int planet p = 6*mylocstart; planet p < (6*(mylocend + 1)); planet_p++) {</pre>
    double y0_temp[N] = {0., 0., 0., 0., 0., 0., /* Sun */
                        R1, 0., 0., 0., V1, 0., /* Mercury */
                        0., R2, 0., -V2, 0., 0., /* Venus */
                        0., R3, 0., -V3, 0., 0., /* Earth */
                        0., R4, 0., -V4, 0., 0., /* Mars */
                        0., R5, 0., -V5, 0., 0., /* Jupiter */
                        0., R6, 0., -V6, 0., 0., /* Saturn */
                        0., R7, 0., -V7, 0., 0., /* Uranus */
                        0., R8, 0., -V8, 0., 0., /* Neptune */
                        0., R9, 0., -V9, 0., 0. /* Pluto */
    y0_loc[planet_p-6*mylocstart] = y0_temp[planet_p];
```

```
// Runge-Kutta parameters
RK_PARAM rkp = rkparams(tspan);
rkp.h0 = 24.*3600.;
rkp.eps = 1e-11;
// Launch the integrator
RK_OUT rko = odeRK("hiroshi912", NBody, tspan, y0_loc, 6*bodies, &rkp);
// Finish
printf("error = %.2e with %d steps\n", rko.err, rko.n);
char str[15+nproc()];
for(int xproc = 0; xproc < nproc(); xproc++) {</pre>
  if(xproc == proc()) {
    sprintf(str, "Output%d.csv", xproc);
    writerkout(str, &rko, 6*bodies);
freerkout(&rko);
free(y0_loc);
free(planet2proc);
// Finalise MPI
MPI_Finalize();
// End of the program
exit(0);
```

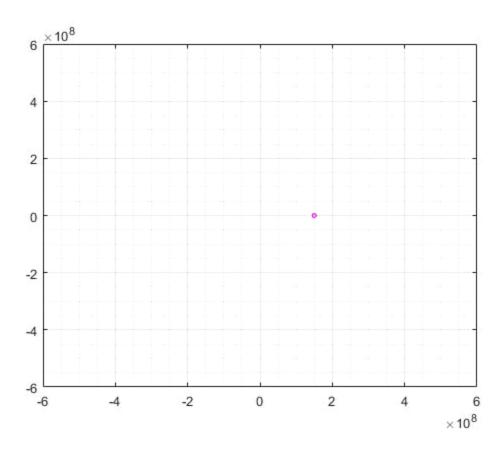
```
// Worksplit
void worksplit(int *mystart, int *myend, int proc, int nproc, int start, int end)
   // Number of tasks
   int ntask = end - start + 1;
   // Number of tasks per processor
    int interval = ntask / nproc;
    // Tasks Left
    int remainder = ntask % nproc;
    if (ntask < nproc)</pre>
        printf("Less tasks than processors\n");
        exit(-1);
    if (remainder != 0)
        if (proc < remainder)</pre>
            *mystart = start + proc * (interval + 1);
            *myend = *mystart + interval;
        else
            *mystart = start + remainder * (interval + 1) + (proc - remainder) * interval;
            *myend = *mystart + interval - 1;
    else
        *mystart = start + proc * interval;
        *myend = *mystart + (interval - 1);
```

```
void NBody(double t, double* var, int n, double* varp) {
      Body 1: Perturbated
          var[0] = rx \quad var[3] = vx
          var[1] = ry \quad var[4] = vy
          var[2] = rz \quad var[5] = vz
      Body 2: Perturber
          var[6] = rx \quad var[9] = vx
          var[7] = ry \quad var[10] = vy
          var[8] = rz \quad var[11] = vz
      Body 3: Perturber
          var[12] = rx \quad var[15] = vx
          var[13] = ry \quad var[16] = vy
          var[14] = rz \quad var[17] = vz
   // varp = prime (derivative of var | var prime)
   int r;
                // for error checking
   int start = 0;
   MPI Status st;
   if(proc() != 0) {
        start = planet2proc[proc()-1] + 1;
   double* y;
   y = (double*)malloc(3*NBODY*sizeof(double));
   if (proc() != 0) {
        double* pos; // vectors of positions
        pos = (double*)malloc(3*(BODIES(proc()))*sizeof(double));
        for (int body = 0; body < (BODIES(proc())); body++)</pre>
            pos[3*body]
                            = VAR(body,0);
            pos[3*body + 1] = VAR(body, 1);
            pos[3*body + 2] = VAR(body, 2);
       r = MPI_Ssend(pos, 3*(BODIES(proc())), MPI_DOUBLE, 0, proc() + nproc(), MPI_COMM_WORLD);
       checkr(r, "send");
       free(pos);
       r = MPI_Recv(y, 3*NBODY, MPI_DOUBLE, 0, proc(), MPI_COMM_WORLD, &st);
       checkr(r, "receive");
```

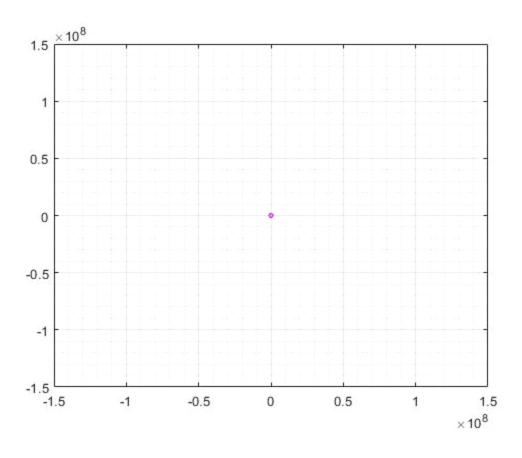
```
else { // If rank == 0
        double* var loc;
        // var loc = (double*)malloc(3*(BODIES(1))*sizeof(double));
        for (int y_counter = 0; y_counter <= planet2proc[0]; y_counter++) {</pre>
            Y(y_{counter,0}) = VAR(y_{counter,0});
            Y(y counter,1) = VAR(y counter,1);
            Y(y_counter,2) = VAR(y_counter,2);
        for (int xproc = 1; xproc < nproc(); xproc++) {</pre>
            var loc = (double*)malloc(3*(BODIES(xproc))*sizeof(double));
           // var_loc = realloc(var_loc, 3*(BODIES(xproc))*sizeof(double));
            r = MPI Recv(var loc, 3*(BODIES(xproc)), MPI DOUBLE, xproc, xproc + n
proc(), MPI_COMM_WORLD, &st);
            checkr(r, "receive");
            for (int p = 0; p < (3*(BODIES(xproc))); p++) {
                Y(planet2proc[xproc - 1] + 1,p) = var loc[p];
            free(var_loc);
            r = MPI Ssend(y, 3*NBODY, MPI DOUBLE, xproc, xproc, MPI COMM WORLD);
            checkr(r, "send");
```

```
else { // If rank == 0
       double* var loc;
       // var loc = (double*)malloc(3*(BODIES(1))*sizeof(double));
       for (int y counter = 0; y counter <= planet2proc[0]; y counter++) {</pre>
            Y(y_counter,0) = VAR(y_counter,0);
            Y(y counter,1) = VAR(y counter,1);
            Y(y_counter,2) = VAR(y_counter,2);
        for (int xproc = 1; xproc < nproc(); xproc++) {</pre>
            var_loc = (double*)malloc(3*(BODIES(xproc))*sizeof(double));
            // var_loc = realloc(var_loc,3*(BODIES(xproc))*sizeof(double));
            r = MPI Recv(var loc, 3*(BODIES(xproc)), MPI DOUBLE, xproc, xproc + n
proc(), MPI_COMM_WORLD, &st);
            checkr(r, "receive");
            for (int p = 0; p < (3*(BODIES(xproc))); p++) {
                Y(planet2proc[xproc - 1] + 1,p) = var loc[p];
            free(var_loc);
            r = MPI_Ssend(y, 3*NBODY, MPI_DOUBLE, xproc, xproc, MPI_COMM_WORLD);
            checkr(r, "send");
```

# 2 Body Sun-Earth (in Serie)



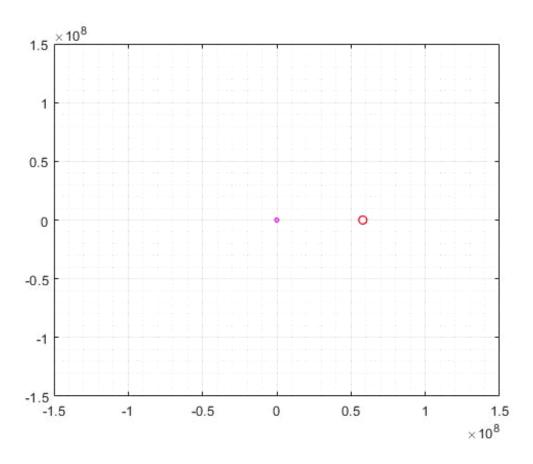
## 3 Body Sun-Venus-Earth (in Serie)



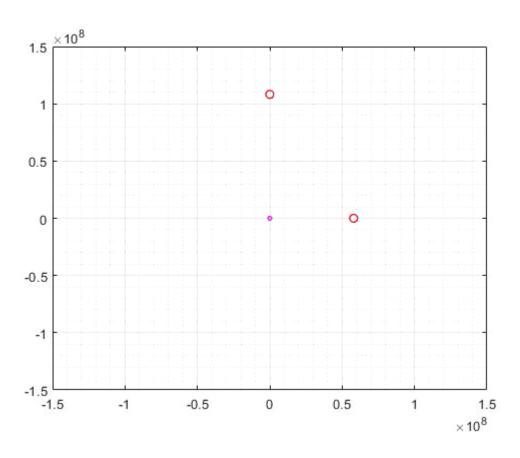
# 2 Body Sun-Mercury (in Parallel with 2 cores)

Sun						
0	0	0	0	0	0	0
86400	0.0245110000000000	0.000584000000000000	0	1.0000000000000e-06	0	0
250682.142843000	0.205685000000000	0.0142310000000000	0	2.0000000000000e-06	0	0
458056.891552000	0.681010000000000	0.086384000000000	0	3.0000000000000e-06	1.0000000000000e-06	0
677541.526709000	1.4689130000000	0.277192000000000	0	4.0000000000000e-06	1.0000000000000e-06	0
900692.956362000	2.5439060000000	0.64338400000000	0	5.0000000000000e-06	2.0000000000000e-06	0
1125253.78642900	3.86774100000000	1.23517000000000	0	6.0000000000000e-06	3.0000000000000e-06	0
1350550.43727600	5.3942240000000	2.09511500000000	0	7.0000000000000e-06	4.0000000000000e-06	0
1576340.24726600	7.0703580000000	3.2567390000000	0	8.0000000000000e-06	6.0000000000000e-06	0
1802517.17034800	8.8378230000000	4.74327000000000	0	8.0000000000000e-06	7.0000000000000e-06	0
2029046.25641200	10.6350090000000	6.56691100000000	0	8.000000000000e-06	9.0000000000000e-06	0
t	X	У	Z	VX	vy	VZ
t Mercury	X	У	Z	VX	vy	٧Z
t Mercury °	<b>X</b> 57909227	<b>y</b> 0	<b>Z</b> 0	<b>VX</b>	<b>VY</b> 47.8792710000000	<b>V</b> Z
t Mercury 0 86400		<b>y</b> 0 4133251.54644200	<b>Z</b> 0 0		•	
0	57909227	<b>y</b> 0 4133251.54644200 11916728.3596100	<b>Z</b> 0 0 0	0	47.8792710000000	
0 86400	57909227 57761533.9424860		<b>Z</b> 0 0 0 0	0 -3.41736700000000	47.8792710000000 47.7571580000000	
0 86400 250682.142843000	57909227 57761533.9424860 56669834.6280460	11916728.3596100	<b>Z</b> 0 0 0 0 0 0	0 -3.4173670000000 -9.85273500000000	47.8792710000000 47.7571580000000 46.8545430000000	
0 86400 250682.142843000 458056.891552000	57909227 57761533.9424860 56669834.6280460 53805683.6141540	11916728.3596100 21410907.9831080	<b>Z</b> 0 0 0 0 0 0 0	0 -3.41736700000000 -9.85273500000000 -17.7025100000000	47.8792710000000 47.7571580000000 46.8545430000000 44.4864660000000	
0 86400 250682.142843000 458056.891552000 677541.526709000	57909227 57761533.9424860 56669834.6280460 53805683.6141540 49058030.2004960	11916728.3596100 21410907.9831080 30769924.2059480	<b>Z</b> 0 0 0 0 0 0 0 0 0	0 -3.41736700000000 -9.85273500000000 -17.7025100000000 -25.4405320000000	47.8792710000000 47.7571580000000 46.854543000000 44.4864660000000 40.5611130000000	
0 86400 250682.142843000 458056.891552000 677541.526709000 900692.956362000	57909227 57761533.9424860 56669834.6280460 53805683.6141540 49058030.2004960 42580473.3400290	11916728.3596100 21410907.9831080 30769924.2059480 39247698.4597760	<b>Z</b> 0 0 0 0 0 0 0 0 0 0	0 -3.4173670000000 -9.8527350000000 -17.702510000000 -25.440532000000 -32.4499450000000	47.8792710000000 47.7571580000000 46.8545430000000 44.4864660000000 40.5611130000000 35.2054770000000	
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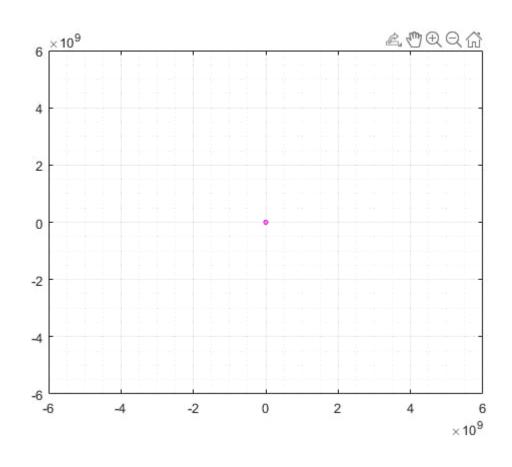
# 2 Body Sun-Mercury (in Parallel with 2 cores)



# 3 Body Sun-Mercury-Venus (in Parallel with 3 cores)



# Solar System (in Parallel\*)



#### Validation of the results

- An orbiting particle has two important forms of energy:  $gravitational\ potential\ energy,\ U$ , and  $kinetic\ energy,\ K$ .
- The sum of the kinetic and gravitational potential energy is known as the total mechanical energy of the system, E.
- The sum of these two energies give us what is known as the total mechanical energy of the system, E. The equation for the total mechanical energy of an orbiting body is derived by combining the centripetal force of a satellite being equal to the force of gravity, with the magnitude of the force from Newton's 2nd law and remembering the equation for the centripetal acceleration [1]:

$$F = rac{GMm}{r^2} = rac{mv^2}{r}$$
  $E = K + U = rac{GMm}{2r} - rac{GMm}{r} = -rac{GMm}{2r}$   $mv^2 = rac{GMm}{r}$   $E = K + U = rac{GMm}{2r} - rac{GMm}{r} = -rac{GMm}{2r}$   $E = -rac{GMm}{2r}$ 

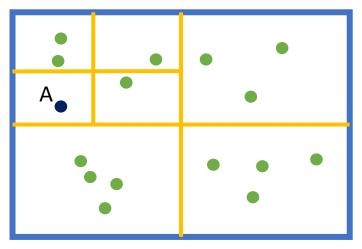
Kinetic energy: 
$$K = \frac{1}{2} \sum_{i=1}^{N} m_i v_i^2$$
  
Potential energy:  $W = -\frac{1}{2} G \sum_{i=1}^{N} \sum_{j=1,\neq i}^{N} \frac{m_i m_j}{|\boldsymbol{r}_i - \boldsymbol{r}_j|}$   
Total energy:  $E = K + W$ 

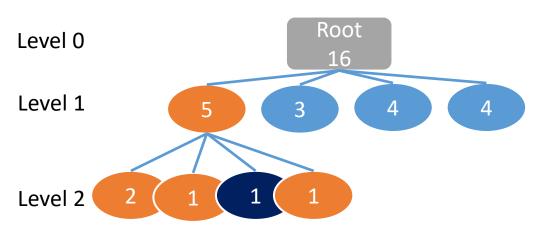
Energy conservation: E = constant

### More optimized Algorithms

- Barnes-Hut Oct-tree algorithm: O(NlogN) [1] [2] [3]
  - In 1996, Josh Barnes and Piet Hut developed an approximation algorithm for performing an N Body simulation that reduces the amount of computational time from  $O(N^2)$  to O(NlogN).
  - The main idea is that force calculations don't need to be exact for particles that are far away from each other. They proposed to group particles together and approximate them by their centre of mass. A tree is used to represent the hierarchical subdivision of space known as octants (quadrants in 2D).
  - The division continues until each particle has a cell itself.
  - Consider the highlighted particle A: At the first level of the tree, three cells (in blue) are far away and can be approximated by its centre of mass but within level it cannot be approximated expect the first quadrant of this level.

16 particles in a 2D Domain





<sup>[1]</sup> Barnes, J., & Hut, P. (1986). A hierarchical O(N log N) force-calculation algorithm. Nature, 324(6096), 446–449. doi:10.1038/324446a0

<sup>[2]</sup> Greengard, L. (1990). The Numerical Solution of the N-Body Problem. Computers in Physics, 4 (142).

<sup>[3]</sup> Liu, Pangfeng & Bhatt, Sandeep. (2001). Experiences with parallel N-body simulation. Parallel and Distributed Systems, IEEE.

#### **Problems encountered**

- Runge Kutta is strongly <u>non parallelizable</u>.
- Adaptative vs. constant time step.
- Difficult to verify the results.
- Enormous amount of send and receives which results in a slower computational time unless N is considerably high and a decent amount of computer cores.
- Several alternative methods where analysed for optimizing sends and receives.
- Overflow of data when sending data.

#### **Further improvements**

- Set sphere of influence depending on the mass and distance.
- Initial conditions must be written in situ, consider import from txt.
- Implement the Barnes-Hut approach.
- Implement a 3D plot of the motion.
- As bodies approach each other, the force between them grows without bound, which is an undesirable situation for numerical integration. In astrophysical simulations, collisions between bodies are generally precluded; this is reasonable if the bodies represent galaxies that may pass right through each other. Therefore, a softening factor  $\varepsilon^2 > 0$  is added, and the denominator is rewritten as follows [1] [2]:

$$\overrightarrow{F_i} \approx Gm_i \cdot \sum_{1 \leq j \leq N} \frac{m_j \overrightarrow{r}_{ij}}{\left(|\overrightarrow{r}_{ij}|^2 + \varepsilon^2\right)^{3/2}}$$

• To integrate over time, we need the acceleration  $a_i = F_i/m_i$  to update the position and velocity of body i, and so we simplify the computation to this:

$$\overrightarrow{a_i} \approx G \cdot \sum_{1 \leq j \leq N} \frac{m_j \overrightarrow{r}_{ij}}{\left(|\overrightarrow{r}_{ij}|^2 + \varepsilon^2\right)^{3/2}}$$

#### Conclusions

- All pairs N-body is a brute-force, simple, high performance method. Nevertheless, there are other more efficient algorithms.
- There is a trade-off between:
  - Maximizing parallelism
  - More work per processor
- The most efficient way is to increase the amount of work on each processor and reducing the amount of parallelism as the speed of the communications between a single processor is faster than communicating between processors.
- The problem is implemented in 3D but the resultant plots are 2D.
- We learned how to version control our codes and work in a repository as a group using git.
- We have struggled a lot but we have been able to view programming from another point of view. Besides, we been able to think differently as we learned how memory is structured in a computer.
- We also learned how to debug effectively, for instance, we used *valgrind* tool to view where memory had conflicts.
- After this, we aim to develop code using MPI in our future problems.