From the polar eg'n [r(0)], determine directions where r->0 ie when Itecos 0'=0 e cos0 =-1 cos0'=-1 Attractive case [K<O; LHS of previtig.]  $\theta' = \cos^{-1}(-\frac{1}{2}) = \pm \left( \pi - \cos^{-1}(\frac{1}{2}) \right)$ Repulsive case [K>0; RHS of fig.]

O'= ± cos'(\frac{1}{2}) The scattering angle a, through which it has been deflected from its original motion is  $\omega = \pi - 20' = \pi - 2\pi + 2\cos'(\frac{1}{e})$  $=-\left(\pi-2\cos^{-1}\left(\frac{1}{e}\right)\right)$ Take abs-value = @=11-2cos-1(1) Can relate @, b ; V: @- 7 = - 2cos ( (2)  $-\frac{1}{2}(\Theta-\pi)=\cos^{-1}(\frac{1}{e})$  $\cos\left(-\frac{1}{2}(\Theta-\pi)\right)=\cos\left(\frac{1}{2}(\pi-\Theta)\right)=\frac{1}{e}$ 

 $Cos(\frac{1}{2}(\pi-\pi)) = cos(\frac{1}{2}(\pi-G)) = \frac{1}{2}$   $e = sec(\frac{1}{2}(\pi-G))$   $Vse the identity <math>b^2 = a^2(e^2-1) = a^2(sec^2(\frac{1}{2}(\pi-G))-1) = a^2cot\frac{1}{2}G$   $but a = \frac{1}{2}Kl = \frac{1}{2}Kl so |b = \frac{1}{2}Kl cot^2(G)|$ 

but  $a = \frac{|\mathcal{K}|}{2E} = \frac{|\mathcal{K}|}{mv^2}$  so  $b = \frac{|\mathcal{K}|}{mv^2} \cot^2(\frac{\omega}{2})$ e.g. Orbit of an unbound comet The min. distance of a comet from the Sun (perihelion distance) is I the radius of the Earth's orbit (assumed circular) its velocity at that point is twice the orbital velocity of the Earth. Find the velocity when the comet crosses Earth's orbit of the angle at which the orbits cross, Will the counce escape from The solar system? What Kind of orbit does it follow? At perihelion,  $V = a_E$ ,  $V = 2V_{c,E}$ , V = 0Sim  $a_E$  or  $a_E$  of the comet at that location =  $a_E$ KE of the comet at that location =  $\lim_{k \to \infty} (2k, \epsilon)^2 = 2mk_{\rm g}^2$ but, by definition,  $V_{c,E} = \frac{GM}{q_E}$ , so KE = 2mGME = 2GMm - 2GMm = 0 So, parabolic orbit;

a E GE GE Comet will escape. This E is the same at all locations, so at Earth's radius as  $\frac{GMm = 1 mV^2 \Rightarrow V^2 = 2GM = 2V_{C,E}^2 \Rightarrow V = \sqrt{2}V_{C,E}}{a_E}$ To find &, consider ang. now at Earth's orbit.  $\int_{E} = m \left( V \cos \alpha \right) a_{E} = m \sqrt{2} V_{SE} \cos \alpha a_{E}$ At perihelion, Jp = MVaE = m 2Vc, EaE = m Ve, EaE Equate hose two expressions

MJZ VO, E COSX OR = MVO, E OR

